SMAP ® - 3D

Structure Medium Analysis Program

3-D Static, Consolidation and Dynamic
Analysis for Dry, Saturated and
Partially Saturated Soils
and Rock Mass

User's Manual Version 7.06

COMTEC RESEARCH

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Introduction

1.1 Overview

SMAP-3D is an advanced three-dimensional finite element computer program developed for the geometric and material nonlinear structure-medium interaction analysis. The program can be a powerful tool for the geomechanical analysis since it can solve static, consolidation and dynamic problems in dry, partially saturated or fully saturated soils and porous rock mass. The program has been designed to integrate the pre-, main-, and post-processors as shown at the end of this section.

1.2 Features

Features of SMAP-3D include:

- Three-dimensional isoparametric continuum element
  - Models soils, rocks and concrete media
  - Allows yielding and tension cut-off
  - Models dry, saturated and partially saturated porous media
- Joint element
  - Models faults, joints, and interfaces
  - Allows sliding and debonding

- Shell element
  - Models slabs, shear walls, tunnel linings, etc.
  - Considers membrane deformation, bending and torsional resistance
- Beam element
  - Models structural frames
  - Considers axial deformation, bending and torsional resistance
- Truss element
  - Models rock bolts and anchor bar
  - Allows yielding, buckling and post-buckling
- External loads
  - Pressure time history
  - Displacement/Velocity/Acceleration time history
  - Initial velocity
  - Gravity load
  - Base acceleration time history
- Special boundary condition
  - Skew boundary (Not Available)
  - Transmitting boundary
- Simulation of a sequence of excavation and construction
- Nonlinear material model
  - Von Mises model
  - Mohr-Coulomb model
  - In Situ Rock model
  - JWL Explosive Source model
  - Modified Cam-Clay model
  - Hyperbolic model
  - Engineering model
  - User defined model
- Large deformations
  - Use updated Lagrangian

1.3 Applications

Applications of SMAP-3D include:

- Dynamic analysis
  - Wave propagation
  - Ground motions due to tunnel blasting
  - Blast-induced liquefaction
  - Earthquake analysis
- Consolidation analysis
  - Foundation settlement
  - Earth dam stability during construction
  - Ground water flow through tunnel liner
- Rock-structure interaction analysis
  - Underground power plant chamber
  - Lined or unlined shafts and tunnels subjected to internal water pressures as well as external earth pressures
- Shallow and deep foundation analysis
- Slope stability analysis
- Framed structural analysis
- Plate or shell structural analysis

Overview of SMAP-3D Program Structure

USER INPUT User prepares Mesh, Main, and Post Files according to SMAP-3D User's Manual as described in Section 4.

PRESMAP Pre-processors to automatically generate Mesh File which contains nodal coordinates, boundary constraints, and element indexes.

SMAP-3D Main-processor executing Mesh and Main Files to compute displacements, stresses and strains. Output files include:

CONTSS.DAT	Stresses/strains in continuum
SHELSF.DAT	Shell member end forces
SHELSM.DAT	Shell stresses/moments
SHELRB.DAT	Shell rebar stresses/total moments
BEAMSF.DAT	Section forces in beam
TRUSS.DAT	Stresses/strains in truss
DISPLT.DAT	Nodal displacements, velocities and accelerations

PLOT-XY
PLOT-2D
PLOT-3D

Post-processors executing Post File for graphical output:

- Finite element mesh
- Deformed shape
- Principal stress distribution
- Section forces in beam elements
- Axial force/stress/strain in truss element
- Contours of stresses and factor of safety
- 3D iso surface of stresses and strains
- Time histories of displacements/stresses/strains

Installing SMAP -3D

2.1 Minimum System Requirements

✓ Windows 64 bit operating system
✓ Intel Pentium 4 or AMD processors
✓ 4 GB Ram with 30 GB free space in Drive C
✓ SVGA monitor

2.2 Installation Procedure

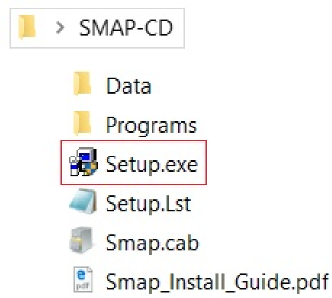
1. Uninstall if there are pre-existing SMAP programs.
   To uninstall SMAP programs, remove following program using Add/Remove in Control Panel:
   SMAP
   Delete following files if they are existing:
   C:\Program Files\Smap
   C:\Windows\Setup1.exe
   Rename or delete following folders if they are existing:
   C:\SMAP
   C:\SmapKey

2. Download SMAP-CD.exe from the Download section of www.ComtecResearch.com

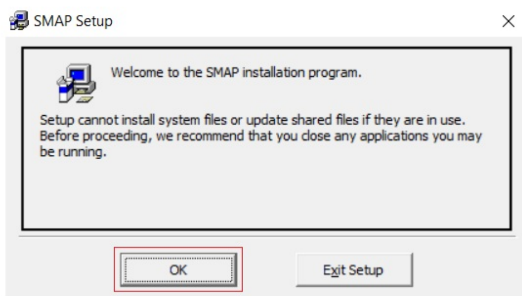
3. Run SMAP-CD.exe
   SMAP-CD folder will be created with SMAP installation programs

2-2 Installing SMAP-3D

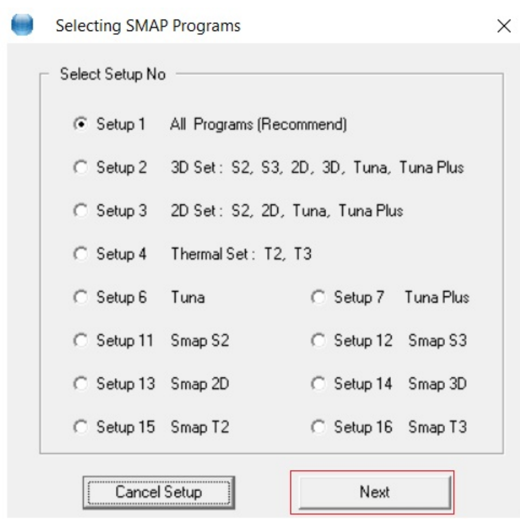
4. Double-click **Setup.exe**



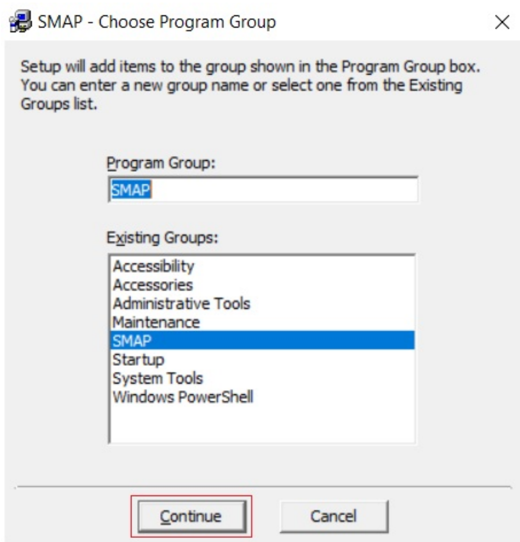
5. Click **OK**



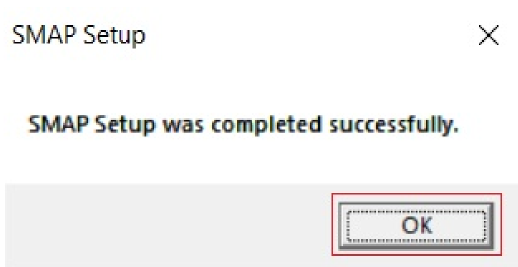
6. Click **Next**
   It will take few minutes.
   Wait until next step.



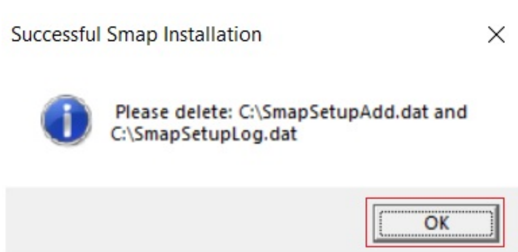
7. Click **Continue**



8. Click **OK**



9. Click **OK**



Note:

Following two log files will be generated once finished:
C:\SmapSetupAdd.dat
C:\SmapSetupLog.dat

If Smap Installation is successful, delete these two files.

If Smap Installation is not successful,
follow the instruction in SmapSetupAdd.dat.

If you still have problems with Smap Installation,
send these two files to info@ComtecResearch.com

Running Programs

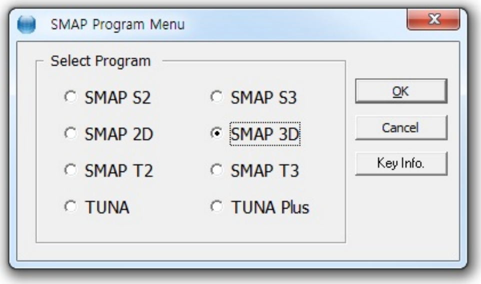
3.1 Introduction

Generally, SMAP-3D consists of pre-, main-, and post-processing programs. Pre-processing programs are mainly used to automatically generate Mesh Files which will contain nodal coordinates, boundary conditions, and element indexes. Main-processing program of SMAP-3D is the one which computes static, consolidation and dynamic response of three-dimensional problems. Post-processing programs are used to show graphically the results from the main-processing program.

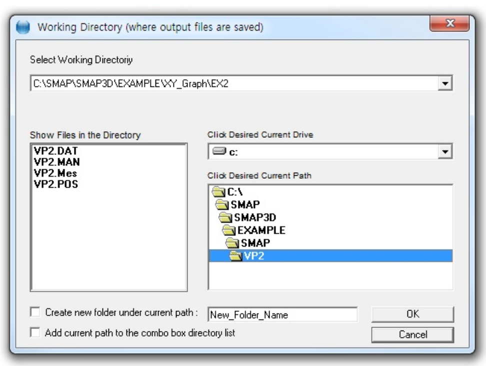
Accessing SMAP-3D Programs

1. When it is the first time, you copy Smap.exe in C:\Ct \Ctmenu and setup a Shortcut to SMAP Icon on your computer desktop. Then You simply double-click SMAP Shortcut.

2. Select **SMAP-3D** radio button and then click **OK** button.

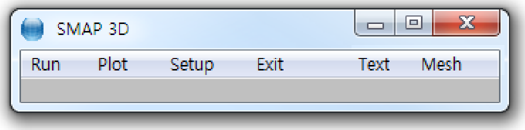


3. Next, you need to select **Working Directory**. Working Directory should be the existing directory where all the output files are saved. It is a good idea to have all your input files for the current project in this Working Directory. Click the disk drive, double-click the directory, and then **OK** button. Note that when you select **Working Directory**, a sub directory **Temp** is created automatically. All intermediate scratch files are saved in this sub directory **Temp**.

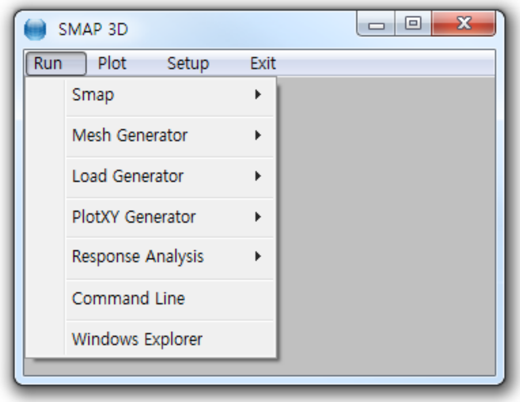


SMAP-3D Menu

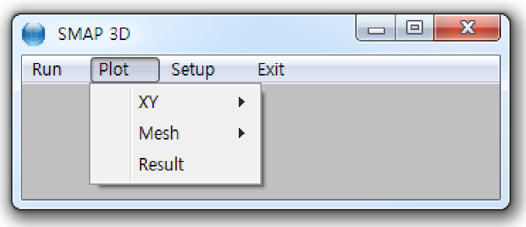
SMAP-3D provides following Main Menus; Run, Plot, Setup, Exit, Text and Mesh.



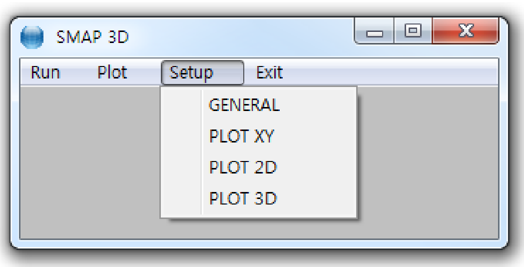
RUN executes main- and pre-processing programs and has following Sub Menus;
Smap,
Mesh Generator,
Load Generator,
PlotXY Generator,
Response Analysis,
Command Line and
Windows Explorer.



PLOT executes XY, Mesh, and Result. Result is associated with post-processing programs to show graphically the computed results.



SETUP is mainly used to set plotting control parameters for PLOT-XY, PLOT-2D, and PLOT-3D and has the following Sub Menus; General, PLOT-XY, PLOT-2D and PLOT-3D.



EXIT is used to end SMAP-3D.

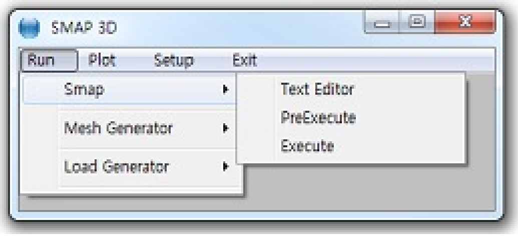
TEXT is used to edit Text files. **MESH** is used to plot F. E. Mesh files.

3.2 RUN Menu

3.2.1 SMAP

Once you have prepared the input files (Mesh, Main, and Post) according to the SMAP-3D User's Manual in Section 4, you are ready to execute SMAP-3D main-processing program.

SMAP Menu has the following Sub Menus; Text Editor, PreExecute, and Execute.



TEXT EDITOR is used to create or modify the input file using Notepad.

PRE EXECUTE is used either to check the input file or to generate plotting information files. **PRE EXECUTE** is especially useful when you want to check input data to see whether there is any input error. It is also useful when you have finished **EXECUTE** but you want to add or modify the Post File for plot. In this case, you edit the Post File as you want, run **PRE EXECUTE**, and then run post-processing programs in **PLOT** menu.

EXECUTE executes SMAP-3D main-processing program.

SMAP-3D Output Files

Once you execute SMAP-3D, generally you can obtain following output files:

CONTSS.DAT	Contains stresses/strains in continuum element
SHELSF.DAT	Contains shell member end section forces
SHELSM.DAT	Contains shell stresses/moments
SHELRB.DAT	Contains shell reinforcing bar axial stresses
BEAMSF.DAT	Contains section forces in beam element
TRUSS.DAT	Contains stresses/strains in truss element
DISPLT.DAT	Contains nodal displacements

It should be noted that all of your output files are saved in the Working Directory that you specified at the beginning.

SMAP-3D Graphical Output

SMAP-3D Post-processing programs can generate the following graphical output:

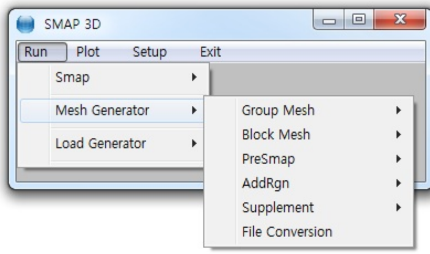
- Finite element mesh
- Deformed shape
- Principal stress distribution
- Section forces in beam element
- Extreme fiber stresses/strains in beam elements
- Axial force/stress/strain in truss element
- Contours of stresses, strains and factor of safety
- 3D iso surface of stresses and strains
- Time histories of displacements/stresses/strains

Graphical output can be followed by running RESULT from PLOT Menu.

3.2.2 MESH GENERATOR

MESH GENERATOR is mainly used to model two and three dimensional finite element meshes that are used as Mesh File.

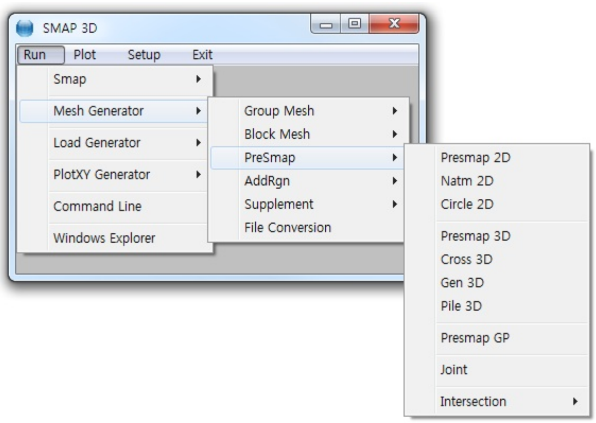
MESH GENERATOR Menu has the following Sub Menus; Group Mesh, Block Mesh, PreSmap, AddRgn, Supplement, and File Conversion.



GROUP MESH is a two-dimensional CAD program specially designed to build group mesh which can be used to generate finite element mesh with the aid of program ADDRGN-2D. Section 5 in SMAP-3D Example Problems describes in detail about running Group Mesh.

BLOCK MESH is a three-dimensional CAD program specially designed to build block mesh which can be used to generate finite element mesh with the aid of program PRESMAP-GP. Section 6 in SMAP-3D Example Problems describes in detail about running Block Mesh.

PRESMAP menu includes two and three dimensional pre-processing programs to generate finite element meshes: Section 7 in SMAP-3D Example Problems describes in detail about running PRESMAP Programs.



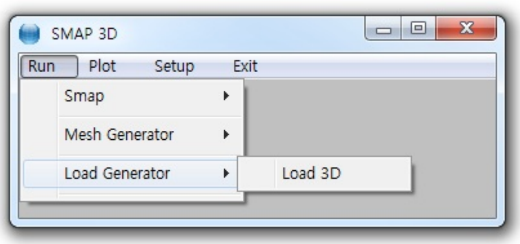
ADDRGN is the pre-processing program which has the following two basic functions: Combine two different meshes and modify existing meshes. Section 8 in SMAP-3D Example Problems describes in detail about running ADDRGN programs.

SUPPLEMENT contains supporting programs which are useful to prepare input data for pre- and main-processing programs. Section 9 in SMAP-3D Example Problems describes in detail about running SUPPLEMENT programs.

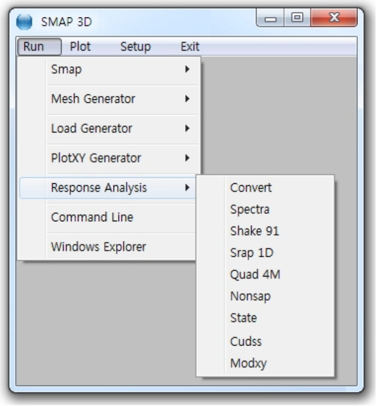
FILE CONVERSION is to convert Mesh File formats between different programs. IGES or FEMAP (Version 4.1- 4.5) can be converted to SMAP Mesh File format. Section 10 in SMAP-3D User's Manual describes in detail about running FILE CONVERSION program.

3.2.3 LOAD GENERATOR

LOAD GENERATOR includes the pre-processing program **LOAD-3D** which generates nodal values of external forces, specified velocities, initial velocities, accelerations and transmitting boundaries.
Section 10 in SMAP-3D Example Problems describes in detail about running **LOAD-3D** program.



3.2.4 PlotXY GENERATOR



PlotXY GENERATOR is the graphical user interface which is mainly used to generate or edit Simplified Time History and Simplified Snapshot of

Card Group 12 in SMAP Post File.

Section 12.7 in SMAP-3D User's Manual describes in detail about running PlotXY Generator program.

3.2.5 RESPONSE ANALYSIS

RESPONSE ANALYSIS runs following programs for seismic analysis:

Convert	Changing format of input earthquake acceleration data
Spectra	Constructing response spectra from acceleration history
Shake 91	Solving 1D seismic response by frequency domain analysis
Srap 1D	Solving 1D seismic response by finite element analysis
Quad 4M	Solving 2D seismic response by finite element analysis
Nonsap	Solving static and dynamic response of nonlinear systems
State	Plotting stress state on p-q space and octahedral plane
Cudss	Solving cyclic undrained direct simple shear for PM4Sand
Modxy	Modifying each XY data curve separately for PLOT-XY
	All Examples enclosed in the directory C:\Smap\Response

3.2.6 COMMAND LINE

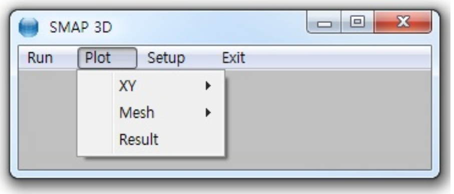
COMMAND LINE opens Windows Command Prompt at the current Working Directory. You can use a keyboard to navigate, access, and modify files and folders by entering commands. For example, COMMAND LINE is used when executing manually SMAP main solvers.

3.2.7 WINDOWS EXPLORER

WINDOWS EXPLORER opens Windows File Explorer at the current Working Directory. You can use a mouse to navigate and manage the drives, folders and files on your computer.

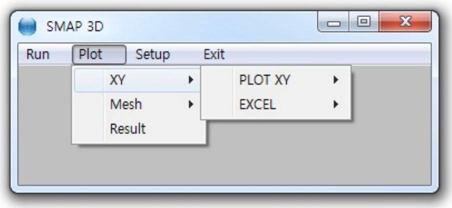
3.3 PLOT Menu

PLOT Menu is to show graphically XY graph, Mesh and Computed Result.



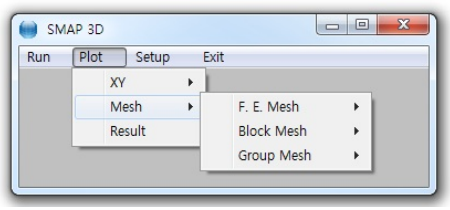
3.3.1 XY

XY graph can be displayed by PLOT-XY or EXCEL. Section 11 in SMAP-3D Example Problems describes in detail about running XY graph.



3.3.2 MESH

MESH has following Sub Menus; F. E. Mesh, Block Mesh and Group Mesh.



F. E. Mesh is used to open or create Finite Element Mesh File.

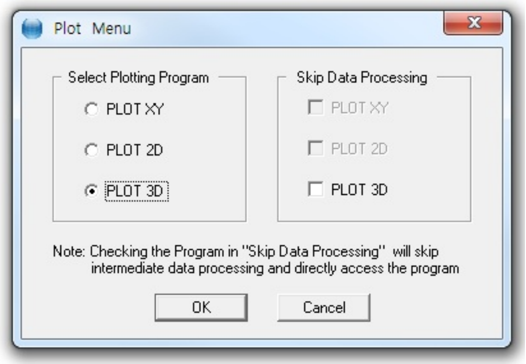
Block Mesh is used to open or build Block Mesh. Section 6 in SMAP-3D Example Problems describes in detail about running Block Mesh.

Group Mesh is used to open or build Group Mesh. Section 5 in SMAP-3D Example Problems describes in detail about running Group Mesh.

3.3.3 RESULT

Once you finished executing SMAP-3D main-processing program, you need to run post-processing programs to show graphically numerical results.

PLOT Menu contains PLOT-XY, PLOT-2D, and PLOT-3D.



PLOT-XY reads Card 12 in Post File and plots time histories of stress/strain/displacement and snapshots of stress/strain/displacement vs. distance. Refer to PLOT-XY User's Manual in Section 13.

PLOT-2D reads Card 11 in Post File and plots contours of continuum stress/strain, beam section forces, truss axial force/stress/strain, principal stress vectors, and deformed shapes. Refer to PLOT-2D User's Manual in Section 14.

PLOT-3D reads Mesh File and Smap Output Files and with no input for Post File, plots contours of stress/strain/displacement, iso surface, principal stress vectors, and deformed shapes. Refer to PLOT-3D User's Manual in Section 15.

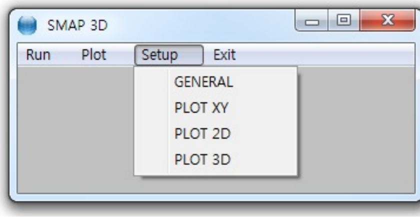
Note: When you first plot results, do not check the check box in Skip Data Processing. When you replot results, however, you can check the check box to skip intermediate data processing. This will save time and keep modified output data.

3.4 SETUP Menu

You need to run SETUP Menu

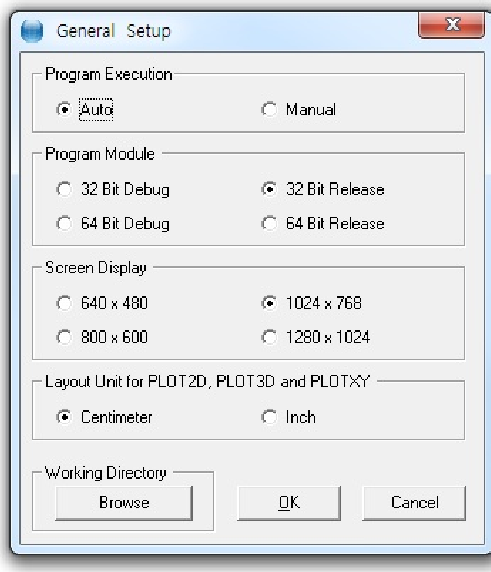
- To specify SMAP-3D main-processing program module.
- To adjust scales of graphical outputs from PLOT-XY, PLOT-2D, and PLOT-3D

SETUP Menu has four Sub Menus; General, PLOT-XY, PLOT-2D, and PLOT-3D



3.4.1 General Setup

General Setup has five different items; Program Execution, Program Module, Screen Display, Layout Unit, and Working Directory.



Program Execution has two options; Auto and Manual.
For Manual Execution, refer to Section 3.5 in User's Manual.

Program Module has four options. 32 Bit Debug, 32 Bit Release, 64 Bit Debug, and 64 Bit Release. Debug program modules run slower but gives more detailed information when run time errors occur. For most cases, 32 Bit Release is recommended. 64 Bit Modules are designed to run large problems.

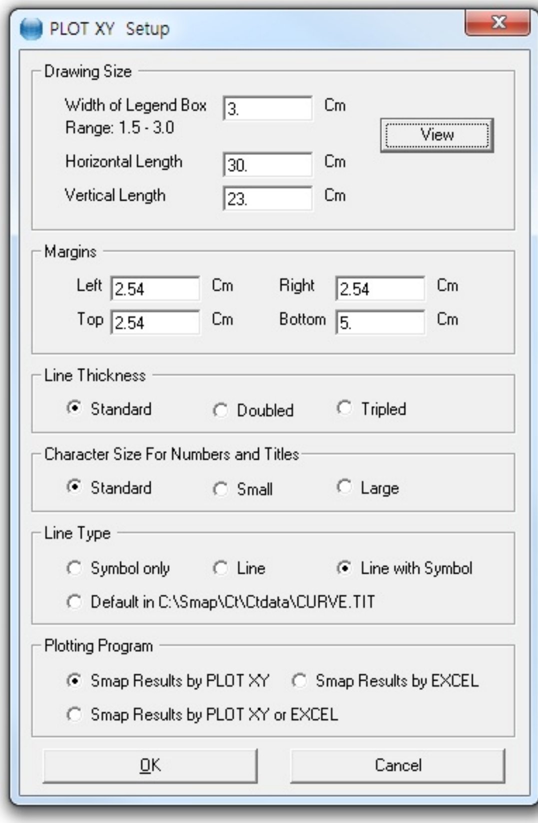
Screen Display has four options; 640x480, 800x600, 1024x768, and 1280x1024. This will affect the size of child window in PLOT-XY and PLOT-2D.

Layout Unit is used for PLOT-XY, PLOT-2D, and PLOT-3D. You can select either Centimeter or Inch in specifying plot scales and dimensions.

Working Directory is to change the current working directory. When you click the Browse button, Working Directory dialog will be shown so that you can select new directory.

3.4.2 PLOT-XY Setup

PLOT-XY Setup is mainly used to specify scales and dimensions of post processing program PLOT-XY. It has six different items; Drawing Size, Margins, Line Thickness, Character Size, Line Type, and Plotting Program.



Drawing Size controls the size of output. Once you specify Legend Box Width, Horizontal and Vertical Length, you can click **View** button to see the scaled layout.

Margins is used to shift the drawing area. Left margin is the distance from the left edge of printer page to the left frame line. In the similar way, you can specify Top, Right, and Bottom margins.

Line Thickness specifies the thickness of lines. This option is not used.

Character Size for Numbers and Titles specifies the size of characters for numbers and titles. It has three options; Standard, Small, and Large.

Line type is used to specify default line type and has four options; Symbol only, Line, Line with Symbol, and Default in C:\ Smap\Ct\Ctdata\Curve.tit.

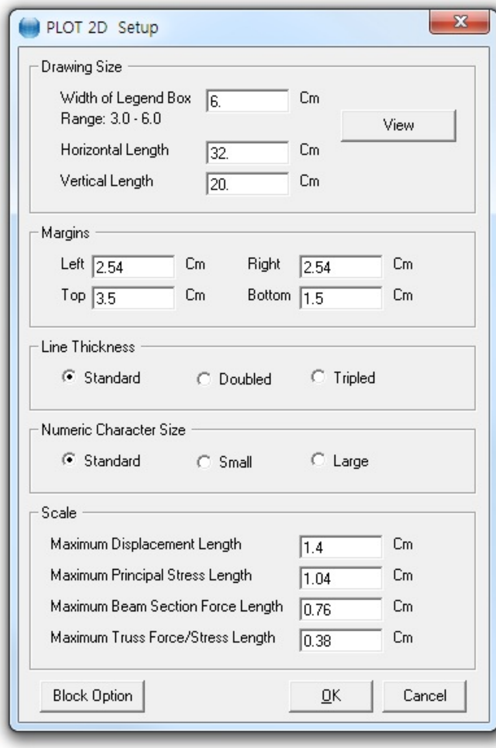
Plotting Program is used to specify default program to plot Smap results. It has three options; PLOT-XY, EXCEL, and PLOT-XY or EXCEL. Last option is to select either PLOT-XY or EXCEL at the time you plot results.

3.4.3 PLOT-2D Setup

PLOT-2D Setup is mainly used to specify scales and dimensions of post processing program PLOT-2D. It has six different items; Drawing Size, Margins, Line Thickness, Numeric Character Size, Scale and Block Option. The first four items are much similar to those described in PLOT-XY Setup.

Scale specifies Maximum Displacement Length, Maximum Principal Stress Length, Maximum Beam Section Force Length, and Maximum Truss Force/Stress Length, which will be shown on PLOT-2D.

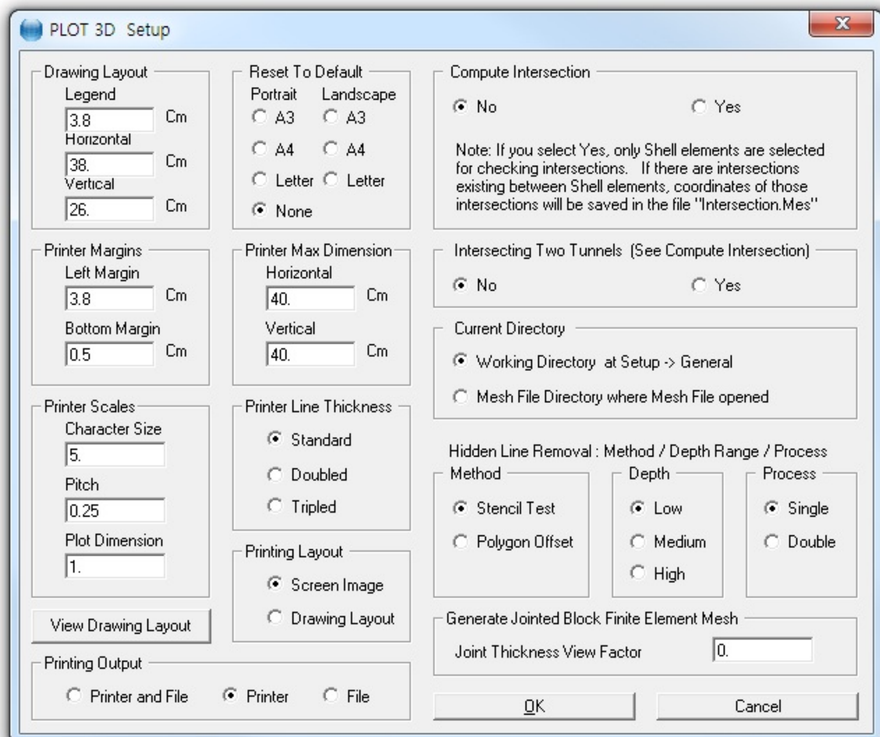
Block Option specifies options to generate either PRESMAP Output or Block Diagram.



3.4.4 PLOT-3D Setup

PLOT-3D Setup is mainly used to specify scales and dimensions of post processing program PLOT-3D. In addition, it can compute intersection of shell elements and intersecting two tunnels. And It can also generate jointed block finite element meshes. It has thirteen different items; Drawing Layout, Printer Margins, Printer Scales, Reset To Default, Printer Max Dimension, Printer Line Thickness, Printing Layout, Printing Output, Compute Intersection, Intersecting Two Tunnels, Current Directory, Hidden Line Removal and Generate Jointed Block Finite Element Mesh.

Refer to descriptions shown in the PLOT-3D Setup dialog.



3.5 Manual Procedure to Run SMAP-3D

Occasionally, you need to execute SMAP-3D main-processing program manually to see what is going on each step, specially when terminated due to some errors.

Method 1

1. Select Setup -> General -> Manual in Program Execution
2. Select Run -> Smap -> Execute
3. Select Smap project file when displaying file open dialog
4. Now Smap is running on Windows Command Line
5. Type Enter key to continue to next step or Control C to stop

Method 2

1. Select Run -> Command Line
2. Change to Temp sub directory

   Create Temp sub directory if not existing.

   Type MD Temp

   Then change to this sub directory.

   Type CD Temp

   Now, the files in the Working Directory can be accessed by prefixing "..\" to the file name.
3. Type C:\Smap\Ct\Ctbat\Smap3D
4. Type ..\VP1.Dat to access input file in Working Directory, for example
5. Type Enter key to continue to next step or Control C to stop

3.6 Debugging SMAP-3D Main-Processing Program

Debug information would be helpful in the following cases:

- Having run time errors
- Extracting convergence
- Checking elapsed time

In order to get debug information, you need to modify the file "Smap_3D.dat" in the directory C:\Smap\Ct\Ctdata\Debug

```
     1,     11,       1,       1,      1,    100,         90
IDEBUG, NCLDEB, IOUTDEB, ICONVER, NELDEB, NO_MAX, NO_RESTART

   This "DEBUG.DAT" file allows listing of status with
   elapsed time information while running main process of
   SMAP programs.   This is the very useful features to see
   where it spends most time and where it stops.

   IDEBUG   =  0 : Do not print debug information.
               1 : Print debug information. Refer to IOUTDEB.
               2 : Print debug information in each individual
                   files based on NO_MAX and NO_RESTART and
                   save  in C:\SMAP\SMAP2D\DEBUG for SMAP-2D
                   and   in C:\SMAP\SMAP3D\DEBUG for SMAP-3D

   NCLDEB        : Ending cycle number.
                   No printing debug information after NCLDEB.

   IOUTDEB =   0 : Debug information on screen.
               1 : Debug information on file,
                   Smap_3D.deb in Working Directory\Temp

   ICONVER =   0 : Do not print convergence information.
               1 : Print the ratio of displacement increment
                   to current displacement (DU/U)

   NELDEB   = -1 : Do not print element information in element
                   level operation.
            =  0 : Print current element number in element
                   level operation.
            >  0 : Print debug information for the element
                   number NELDEB in element level operation.

   NO_MAX        : Maximum number of individual files.
                   Used for IDEBUG = 2.

   NO_RESTART    : Restart number for individual file
                   once it reaches NO_MAX.
                   Used for IDEBUG = 2.
```

SMAP-3D
User's Manual

4.1 Introduction

To run SMAP-3D main-processing program, you need to prepare a Project File which contains Mesh File name, Main File name, and Post File name.

Mesh File contains nodal coordinates, boundary conditions, element indexes and material property numbers. This Mesh File is normally generated by Mesh Generator programs.

Main File contains all the other data required for the three-dimensional numerical analysis of static, consolidation, or dynamic problems.

Post File contains information which is used to show graphically the results from the main-processing program.

4.2 Project File

Project File is a collection of names of Mesh, Main, and Post Files with the following text format:

```
Mesh File Name
   Full path of Mesh File
Main File Name
   Full path of Main File
Post File Name
   Full path of Post File
```

As an example, a Project File **VP2.Dat** can be written as:

```
Mesh File Name
   D:\Example\VP2.Mes
Main File Name
   D:\Example\VP2.Man
Post File Name
   D:\Example\VP2.Pos
```

4.3 Mesh File

Mesh File contains nodal coordinates, boundary conditions, element indexes and material property numbers. This Mesh File is normally generated by Mesh Generator programs.

To plot Mesh File, select Mesh in Plot menu.

Mesh File

Card Group	Input Data and Definitions (Mesh File)
1 General Information	1.1 TITLE [Character string] TITLE — Project title
	1.2 LABEL1 [Character string] LABEL1 — Label for Card 1.3
	1.3 NUMNP, NCONT, NBEAM, NTRUSS NUMNP — Total number of nodal points NCONT — Total number of continuum elements NBEAM — Total number of beam elements NTRUSS — Total number of truss elements

<table>
<tr><th>Card Group</th><th>Input Data and Definitions (Mesh File)</th></tr>
<tr><td rowspan="2">2

Coordinate</td><td>2.1

LABEL2A [Character string]
LABEL2B [Character string]

LABEL2A Label for coordinate
LABEL2B Label for Card 2.2</td></tr>
<tr><td>2.2

NUMNP Cards
NODE, ISX, ISY, ISZ, IFX, IFY, IFZ, IRX, IRY,
IRZ, IEX, IEY, IEZ, XA, YA, ZA
- - - - - -
- - - - - -

NODE Node Number

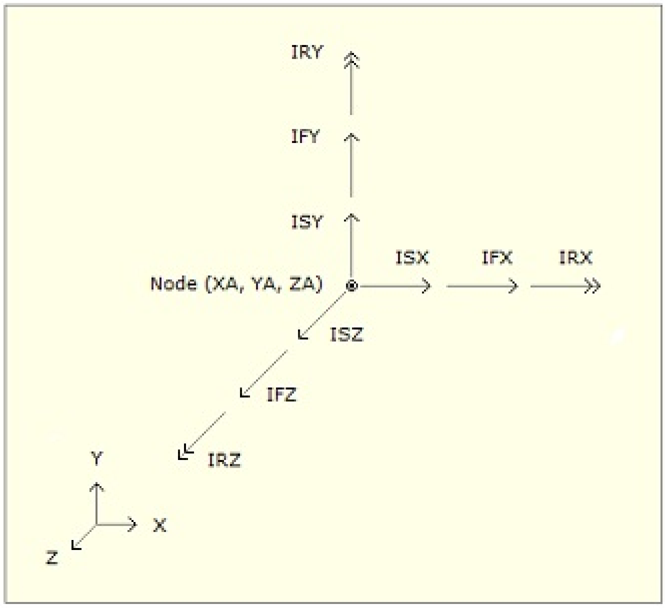
ISX Skeleton X DOF (Degree of Freedom)
ISY Skeleton Y DOF
ISZ Skeleton Z DOF

IFX X DOF for relative pore fluid motion
IFY Y DOF for relative pore fluid motion
IFZ Z DOF for relative pore fluid motion

IRX Rotational DOF about X axis for bending
IRY Rotational DOF about Y axis for bending
IRZ Rotational DOF about Z axis for bending

IEX Slip X DOF
IEY Slip Y DOF
IEZ Slip Z DOF

ISX, ISY, ISZ, IFX, IFY, IFZ, IRX, IRY, IRZ, IEX, IEY, IEZ
= 0 Free to move in specified direction
= 1 Fixed in specified direction</td></tr>
</table>

Card Group	Input Data and Definitions (Mesh File)
2 Coordinate	2.2 

Card Group	Input Data and Definitions (Mesh File)
3	3.1

Continuum Element (If NCONT = 0, skip this card group)

3.1

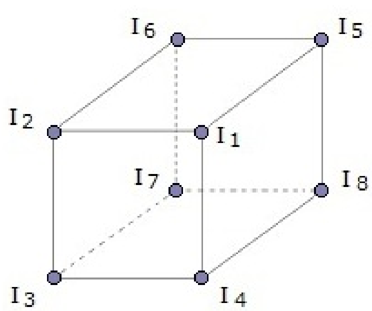
LABEL3A [Character string]
LABEL3B [Character string]

LABEL3A	Label for continuum element
LABEL3B	Label for Card 3.2

3.2

NCONT Cards

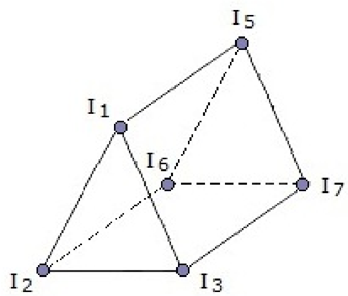
NEL, I_1, I_2, I_3, I_4, I_5, I_6, I_7, I_8, MATC, KS, KF,
INTR, INTS, INTT, TBJWL
\- - - - - - - -



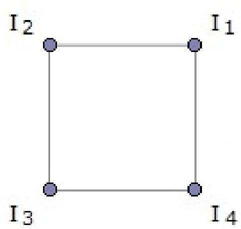
NEL	Element number
$I_1,I_2,I_3,I_4,I_5,I_6,I_7,I_8$	Element corner node numbers
MATC	Material property number

Card Group	Input Data and Definitions (Mesh File)
3 Continuum Element	3.2 KS = -1 Element has high explosive solid phase = 0 Element has solid phase 3D continuum = 1-6 Element has joint and KS represents face designation number. Refer to description in the following page. = 15 Element has SHELL element KF = 0 Element has fluid phase = 1 Element has no fluid phase INTR Use INTR = 2 INTS Use INTS = 2 INTT Use INTT = 2 TBJWL Detonation time (Required for KS = -1) Time from initial detonation to the detonation of this element.

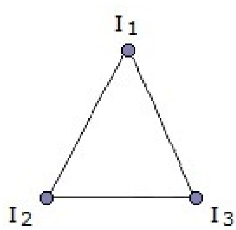
Wedge Element (I_4 = I_8 = 0)



Quadrilateral Shell Element ($I_5 = I_6 = I_7 = I_8 = 0$)



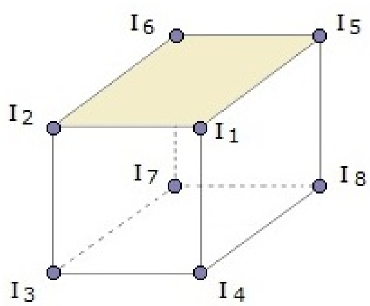
Triangular Shell Element ($I_4 = I_5 = I_6 = I_7 = I_8 = 0$)



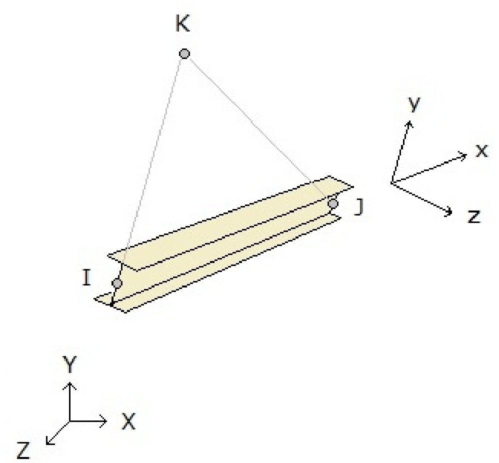
Joint Element Face Designation

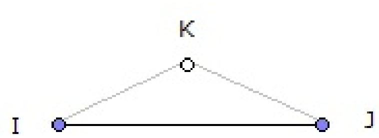
KS	I_1'	I_2'	I_3'	I_4'
1	5	6	2	1
2	6	7	3	2
3	7	8	4	3
4	8	5	1	4
5	1	2	3	4
6	6	5	8	7

For KS = 1



It should be noted that the thickness of joint element is determined Not by gap between two faces ($I_5\ I_6\ I_2\ I_1$ and $I_8\ I_7\ I_3\ I_4$), But by joint thickness (t) specified in Card 5.3.2.4.11 in Main File input The nodal coordinates of I_5, I_6, I_2, I_1 represent the Location of Joint Face but nodal coordinates of I_8, I_7, I_3, I_4 are used only For Plotting Purpose

Card Group	Input Data and Definitions (Mesh File)
4	4.1 LABEL4A [Character string] LABEL4B [Character string] LABEL4A Label for beam element LABEL4B Label for Card 4.2
Beam Element (If NBEAM = 0, skip this card group)	4.2 NBEAM Cards ┌ NEL, I, J, K, MSEC │ - - - - - └ - - - - - NEL Beam element number I, J Node number at beam end points K Reference node number MSEC Material section number  Local beam reference system Axis x and y are in plane defined by plane I, J, K nodes Axis x is defined by line I - J Axis z is perpendicular to the I, J, K plane

Card Group	Input Data and Definitions (Mesh File)
5 Truss Element (If NTRUSS = 0, skip this card group)	5.1 LABEL5A [Character string] LABEL5B [Character string] LABEL5A Label for truss element LABEL5B Label for Card 5.2
	5.2 NTRUSS Cards [NEL, I, J, MATT, K, NELPI, NELPJ - - - - - - - - - - - - - -] NEL Truss element number I, J Node number at truss end points MATT Material property number K Reference node number NELPI Parent continuum element number for embedded truss node I NELPJ Parent continuum element number for embedded truss node J 

4.4 Main File

Mesh File in the previous section 4.3 contains the geometrical data of the structure to be analyzed.

Main File contains all the other data required for the three-dimensional numerical analysis of static, consolidation, or dynamic problems.

Main File consists of ten different card groups:

- System Control and Title
- Analysis Type
- Computational Parameters
- Coordinate
- Continuum Element
- Beam Element
- Truss Element
- Element Activity
- Loads
- Requested Output

Card Group	Input Data and Definitions (Main File)
Version No, System Control and Title	1.0 VERSION VERSION Version No (Current Version = 7.05)
	1.1 IBATCH, IVMDK, IOPTDB, ISYMSOL IBATCH = 0 Interactive terminal job = 1 Batch job (not available) = 2 Generate Mesh File PlotMesh.Mes (This will not execute input) = -1 Terminal interactive job with beep sound when the calculation is finished = -11 Same as IBATCH =-1 except long beep sound and character based screen display < -11 Same as IBATCH =-11 except no display IVMDK = 0 Use hard disk to store internal variables = 1 Use addressable memory to store internal variables IOPTDB = 0 Use single precision to solve equation = 1 Use double precision to solve equation ISYMSOL = 0 Program determines solution scheme = 1 Impose symmetric solution scheme = 2 Impose unsymmetric solution scheme
	1.2 LTITLE LTITLE Main title (Max = 80 characters)
	1.3 LSUBTL LSUBTL Subtitle (Max = 80 characters)

<table>
<tr><th>Card Group</th><th>Input Data and Definitions (Main File)</th></tr>
<tr><td>2

Analysis Type</td><td>2.1

NTCSF, NLNR, NGEN, IQUAD, NTEMP, ITDIS, MODAL

NTCSF = 1 Static analysis
= 2 Consolidation analysis
= 3 Dynamic analysis (Implicit method)
= 4 Dynamic analysis (Explicit method)
= 5 Mode superposition analysis
For NTCSF = -5, computes only natural frequencies and mode shapes

NLNR = 0 Linear elastic material
= 1 Nonlinear material

NGEN = 0 Small displacement
= 1 Large displacement (Updated Lagrangian)

IQUAD = 0 No automatic generation
= 1 Automatic generation of quadratic elements
All linear continuum elements are automatically transformed into quadratic elements. For IQUAD = 16, transformed into 16 node hexahedral elements.

NTEMP = 0 Thermal expansion is not considered
= 1 Thermal properties and element temperatures are read from input file ELTEMP.DAT that should be located in working directory.
See Table in the next page

Output motions when base acceleration applied
ITDIS = 0 Relative displacement & Relative velocity
= 1 Total displacement & Total velocity

Modal analysis options for NTCSF = 5 or -5
MODAL = 0 Subspace iteration method
= 1 Determinant search method
= 2 Jacobi iteration method</td></tr>
</table>

Input File ELTEMP.DAT

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Main File)</th></tr>
<tr><td rowspan="4">Thermal Property</td><td colspan="2">1.1
TITLE

TITLE Project title (Max = 80 characters)</td></tr>
<tr><td colspan="2">2.1
LABEL 1 [Character string]
-
LABEL 6 [Character string]

LABEL 1-6 Labels for Card 2.2</td></tr>
<tr><td rowspan="2">2.2
For Each Material</td><td>2.2.1
MATNO, MODEL

MATNO Material property number
If MATNO = -1, end of Card 2.2

MODEL = 1 Constant thermal expansion
= 2 Step thermal expansion
= 3 Porosity rate dependent expansion

Note
MODEL = 2 and 3 are not available</td></tr>
<tr><td>2.2.2
T_o, E_da

T_o Freezing temperature (Degree C)
E_da Anisotropic expansion parameter (ξ)</td></tr>
</table>

Input File ELTEMP.DAT

Card Group	Input Data and Definitions (Main File)	
Thermal Property	2.2 For Each Material	2.2.3 E_u, V_u, E_f, V_f E_u Unfrozen Young's modulus V_u Unfrozen Poisson's ratio E_f Frozen Young's modulus V_f Frozen Poisson's ratio
		2.2.4 Required only for MODEL = 1 Alpha_c Alpha_c Coefficient of thermal expansion (L/L/Temperature)
		2.2.5 Required only for MODEL = 2 Strain_m, dT_o Strain_m Maximum expansive strain dT_o Strain_m distributed over dT_o (Deg C)
		2.2.6 Required only for MODEL = 3 RateN_m, T_m, g_T, Z_eta RateN_m Maximum porosity rate T_m Temperature (Deg C) at RateN_m g_T Temperature gradient (Deg C/m) at RateN_m Z_eta Stress parameter, ζ, in stress unit (Mpa) used for reducing porosity rate

Input File ELTEMP.DAT

<table>
<tr><th>Card Group</th><th>Input Data and Definitions (Main File)</th></tr>
<tr><td rowspan="4">3

Temperature Profile, Can be repeated for each $TIME_i$</td><td>3.1

LABEL 2 [Character string]

LABEL 1   Label for Card 3.2</td></tr>
<tr><td>3.2

$TIME_i$

$TIME_i$ Time. $TIME_i$ should be 0.0 for initial state
If $TIME_i$ = -1.0, end of data</td></tr>
<tr><td>3.3

LABEL 3 [Character string]

LABEL 3   Label for Card 3.4</td></tr>
<tr><td>3.4

NELNO,   MATNO T_{top} T_{bot} $T_{,x}$ $T_{,y}$ $T_{,z}$
-   -   -   -   -   -   -
-   -   -   -   -   -   -

NELNO   Element number
If $NELNO_i$ = -1, end of Card 3.4

MATNO   Material property number.
T_{top} Temperature on top surface
T_{bot} Temperature on bottom surface

$T_{,x}$ Temperature gradient in x direction
$T_{,y}$ Temperature gradient in y direction
$T_{,z}$ Temperature gradient in z direction</td></tr>
</table>

Card Group	Input Data and Definitions (Main File)
3 Computational Parameters	3.1 Cycles and Time Step NCYCL, DT, NDTGR, NITER, MNEWRP, TOLER, IRANGE, KRANGE NCYCL — Number of total solution cycles DT — Global time step: Duration of each cycle NDTGR — Number of time step group (Max=100) If NDTGR = 0, constant time steps are used. For NDTGR > 0, NLNR=1 should be specified in Card Group 2.1 NITER — Number of maximum iteration (Iteration is available for NTCSF = 1) MNEWRP = 0 Modified Newton-Raphson method = 1 Newton-Raphson method =-1 Newton-Raphson method with first iteration as trial guess For specified velocity, use MNEWRP = 0 TOLER — Tolerance for convergence, defined as the ratio of displacement increment to current displacement (Default TOLER=0.001) IRANGE = 0 NITER is applied throughout NCYCL = 1 NITER is applied based on Cycle No = 2 NITER is applied based on Time KRANGE = 0 Stiffness update option is not used = 1 Stiffness update option based on Cycle No = 2 Stiffness update option based on Time

Card Group	Input Data and Definitions (Main File)	
3 Computational Parameters	3.1.1 If NDTGR = 0, go to Card Group 3.1.3 ICYCLTIME ICYCLTIME = 0 Selection of time step is based on Cycle No = 1 Selection of time step is based on Time	
	3.1.2 For Each Time Step Group	3.1.2.1 STIME, ITYPE STIME Starting Cycle No for ICYCLTIME = 0 Starting Time (t_o) for ICYCLTIME = 1 For the first time group, use STIME = 0 ITYPE = 0 Constant time step = 1 Constant log time step = 2 Arbitrary specified time step <u>If ITYPE = 0</u> DT DT Time step <u>If ITYPE = 1</u> DT_1, CLDT DT_1 Starting time step CLDT Constant log time step CLDT = $\log_{10}(t_{i+1}-t_o) - \log_{10}(t_i-t_o)$ <u>If ITYPE = 2</u> NUMDT $DT_1, \ldots, DT_{NUMDT}$ NUMDT Number of time step $DT_1, \ldots, DT_{NUMDT}$ Listing of specified time steps

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Main File)</th></tr>
<tr><td rowspan="2">3

Computational Parameters</td><td colspan="2">3.1.3

If IRANGE = 0, go to Card Group 3.1.5
NRANGE

NRANGE   Number of specified ranges where NITER is applied (Max=100)</td></tr>
<tr><td>3.1.4

For Each Range</td><td>3.1.4.1

SFTIME, SLTIME

SFTIME
Starting Cycle No for IRANGE = 1
Starting Time for IRANGE = 2

SLTIME
Ending Cycle No for IRANGE = 1
Ending Time for IRANGE = 2</td></tr>
</table>

Card Group	Input Data and Definitions (Main File)	
3 Computational Parameters	3.1.5	If KRANGE = 0, go to Card Group 3.2 NRANGE NRANGE Number of specified ranges where stiffness update option is applied (Max=100)
	3.1.6 For Each Range	3.1.6.1 SFTIME, SLTIME, NST SFTIME Starting Cycle No for KRANGE = 1 Starting Time for KRANGE = 2 SLTIME Ending Cycle No for KRANGE = 1 Ending Time for KRANGE = 2 NST Number of time steps for which the global stiffness matrix is assumed to be constant. Ex. For NST = 2, stiffness matrix is updated every other step during the specified ranges from SFTIME to SLTIME

Card Group	Input Data and Definitions (Main File)
3 Computational Parameters	3.2 Numerical Time-Integration and Artificial Viscosity If NTCSF < 3, go to Card Group 3.3 TETA, BETA, GAMA, CQ, CL, F1, F3, RD, NTMODE TETA θ See Table 1 BETA β See Table 1 GAMA γ See Table 1 CQ Quadratic artificial viscosity coefficient CL Linear artificial viscosity coefficient F1 First natural frequency F3 Third natural frequency or Predominant frequency of input motion RD For NTCSF = 5, RD as critical damping ratio. For NTCSF = 3 and RD > 0, program first computes F1 and F3 and then automatically reruns. Actual RD is defined in element material input sections. Note: Both F1 and F3 are used to compute Rayleigh mass and stiffness proportional damping coefficients. NTMODE Number of mode shapes to be considered Note: If NTCSF = 4, only CQ and CL are used

Card Group			Input Data and Definitions (Main File)
3 Computational Parameters	3.3 Calculational Mode Change	3.3.1	If NTCSF = 4, go to Card Group 4 NCLMCH NCLMCH = 0 Do not change calculation mode > 0 Change calculation mode at cycle NCLMCH
		3.3.2	If NCLMCH = 0, go to Card Group 4 NTCNEW, DTNEW TETANEW, BETANEW, GAMANEW, CQNEW, CLNEW, F1NEW, F3NEW, RDNEW, NTMODENEW NTCNEW New value of NTCSF after NCLMCH DTNEW New value of DT after NCLMCH TETANEW, BETANEW, GAMANEW, CQNEW, CLNEW, F1NEW, F3NEW, RDNEW, NTMODENEW are new values of Card 3.2 after NCLMCH, respectively

Table 1 Values of β and θ for $\gamma = 1/2$ *

Integration Method	β	θ
Explicit second central difference	0	1.0
Fox-Goodwin	1/12	1.0
Linear acceleration	1/6	1.0
Newmark's constant acceleration	1/4	1.0
Wilson	1/6	2.0
Stiff linear acceleration	1/6	1.5

*$\gamma = 1/2$ indicates no damping

$\gamma > 1/2$ introduces numerical damping and $\beta = (\gamma + 1/2)^2 / 4$

For more information, refer to
Ghaboussi and Wilson, "Variational Formulation of Dynamic of Fluid Saturated Porous Elastic Solids," ASCE Engineering Mechanics Journal, August 1972

Card Group	Input Data and Definitions (Main File)
4 Coordinate	4.1 NUMNP NUMNP Total number of nodal points
	4.2 CMFAC, SCFP CMFAC Coordinate multiplication factor (Use CMFAC = 1.0) SCFP Stress conversion factor for converting pressure units to Pascals Note SCFP is used for nonlinear pore fluid and JWL model

Stress Unit	SCFP
kg/cm^2	98066.5
t/m^2	9806.65
kg/m^2	9.807
$Newton/cm^2$	10000
bar	100000
psi	6895
ksi	6.895×10^6
psf	47.88
MPa	1000000

Card Group		Input Data and Definitions (Main File)
4 Coordinates	4.4 Boundary Code Change	4.4.1 NBNODE, NCLBCH, IFLCOD NBNODE — Number of nodes where boundary codes are changed NCLBCH — Cycle No where boundary codes are changed IFLCOD = 0 — Read Card 4.4.2 here = 1 — Read Card 4.4.2 from file NewBcode.dat starting with NBNODE as first card If NBNODE = 0, go to next Card Group 4.5
		4.4.2 NBNODE Cards: $NODE_1$, ISX_1, ISY_1, ISZ_1, IFX_1, IFY_1, IFZ_1, IRX_1, IRY_1, IRZ_1 - - - - - - - - - - - - - - Refer to Card Group 2.2 in Mesh File in page 4-5 for description

Card Group		Input Data and Definitions (Main File)
4 Coordinates	4.5 Repeating Nodes	4.5.1 NREPEAT NREPEAT Number of repeating nodes If NREPEAT = 0, go to next Card Group 5.1
		4.5.2 NREPEAT Cards: NODER, NODEP - - - - - - NODER Repeating node NODEP Parent node Note : Repeating node NODER shares the same degrees of freedom as those of the corresponding parent node NODEP

<table>
<tr><th>Card Group</th><th>Input Data and Definitions (Main File)</th></tr>
<tr><td rowspan="2">5
Continuum Element</td><td>5.1

NCONT

NCONT   Total number of continuum element

If NCONT = 0, go to next Card Group 6</td></tr>
<tr><td>5.2

IPFORM, NSPTC, IEDOF

IPFORM   Use IPFORM = 0

NSPTC = 0   Compute stresses and strains at integration points
= 1   Compute stresses and strains at center of element

For shell elements, use NSPTC = 0

IEDOF = 0   Do not include incompatible extra DOF
= 1   Include incompatible extra DOF</td></tr>
</table>

Card Group				Input Data and Definitions (Main File)
5 Continuum Element	5.3 Material Property Data	5.3.1		NTNC NTNC — Number of material property set for continuum element
		For Each Material Property Set	5.3.2.0	MATNO, MATNP MATNO — Material number MATNP — Parent material number MATNO will duplicate MATNP If MATNP > 0, go to next property set.
			5.3.2.1	TITLE TITLE — Material name (Max 80 characters)
			5.3.2.2	POR, GW, G, PFMIN, DAMP, ICST POR — Initial porosity (n_o) GW — Unit weight of water at ( )$^{\circ}$ c G — Gravity constant (g) PFMIN — Minimum fluid pressure (Not used) DAMP — Initial critical damping ratio ICST = 0 : Lumped mass = 1 : Consistent mass (Default)
			5.3.2.3	NF NF = 0 Linear fluid and solid grain = 1 Nonlinear fluid and solid grain

Card Group		Input Data and Definitions (Main File)
5 Continuum Element	5.3 Fluid and Solid Grain Property (NF = 0)	5.3.2.3.1

For NF = 0 (Linear Fluid and Solid Grain)

RK_1, BKG, SGG, BKF, SGF, NK, RK_1FAC, NPHNO

RK_1	Darcy's coefficient of permeability
BKG	Bulk modulus of grain
SGG	Specific gravity of solid grain
BKF	Bulk modulus of pore fluid
SGF	Specific gravity of pore fluid
NK = 0	Isotropic permeability
= 1	Anisotropic permeability
RK_1FAC	Multiplication factor for RK_1, applied during NGSTEP
NPHNO	Permeability intensity history number in Card Group 9.2.3

For NK = 1

a_{xx}, a_{yy}, a_{zz}, a_{xy}, a_{xz}, a_{yz}

a_{ij}	Permeability component ($k_{ij} = a_{ij} \cdot RK_1$)

Card Group		Input Data and Definitions (Main File)
5 Continuum Element	5.3 Fluid and Solid Grain Property (NF = 1)	5.3.2.3.2 <u>For NF = 1 (Nonlinear Fluid and Solid Grain)</u> Permeability Property NP, RK1, RK2, RK3, NK, RK_1FAC, NPHNO NP = 0 Constant permeability Nonlinear permeability = 1 Function of porosity = 2 Function of flow velocity = 3 Function of porosity and flow velocity RK1, RK2, RK3 Permeability constants dependent on NP See Table 2 NK = 0 Isotropic permeability = 1 Anisotropic permeability RK_1FAC Multiplication factor for RK_1, applied during NGSTEP NPHNO Permeability intensity history number in Card Group 9.2.3 For NK = 1 α_{xx}, α_{yy}, α_{zz}, α_{xy}, α_{xz}, α_{yz} α_{ij} Permeability component ($k_{ij} = \alpha_{ij} \cdot k$)

Card Group		Input Data and Definitions (Main File)
5 Continuum Element	5.3 Fluid and Solid Grain Property (NF = 1)	5.3.2.3.2 Solid Grain Property NG, BKG, SGG, CO, VO, S, PB NG = 0 Constant grain modulus = 1 Nonlinear grain modulus BKG Initial bulk modulus of grain SGG Initial specific gravity CO Initial wave velocity at relatively low pressure[*] VO Initial Poisson's ratio[*] S Experimentally determined constant relating loading wave velocity to peak particle velocity. Generally equal to about 1.5 for most rocks and minerals[*] PB Threshold pressure beyond which material tends to behave like a fluid ([*]) Not used for NG = 0

Table 2 Permeability Constants

NP	Equivalent Permeability k (length/time)	Input Variables
0	$k = RK_1$	$RK_1 =$ Darcy's coefficient of permeability (length/time) RK_2, RK_3 not used
1	$k = 10^{RK_1 (n - RK_2)}$	$RK_1 =$ Slope of n vs. log k line in units log (length/time). $RK_2 =$ Porosity corresponding to k=1.0 $RK_3 =$ Not used
2	$k = \dfrac{RK_1}{1 + \dfrac{RK_3}{\gamma_f}\sqrt{RK_1}\,\lvert\dot{w}_i\rvert}$	$RK_1 =$ Darcy's coefficient of permeability (length/time) $= \dfrac{\gamma_f}{a}$ $RK_2 =$ Not used. $RK_3 =$ Ward's coeff. for turbulent flow $\beta_f = b k^{1/2}$
3	$k = \dfrac{K_l}{1 + \dfrac{RK_3}{\gamma_f}\sqrt{K_l}\,\lvert\dot{w}_i\rvert}$ $K_l = 10^{RK_1 (n-RK_2)}$	RK_1 See NP = 1 RK_2 See NP = 1 RK_3 See NP = 2

Card Group		Input Data and Definitions (Main File)
5 Continuum Element	5.3 Fluid and Solid Grain Property (NF = 1)	5.3.2.3.2

Pore Fluid Property

NW, BKF, SGF, SO, GAMMA, PAO, T

NW	= 0	Constant fluid modulus
	= 1	Nonlinear modulus (Fresh water)
	= 2	Nonlinear modulus (Sea water)
BKF		Initial bulk modulus of pore fluid
SGF		Initial specific gravity of pore fluid
SO		Initial degree of saturation* SO $\neq$1.0 invokes partial saturation model
GAMMA		Ratio of heat capacity *, $\gamma = C_p / C_v$
PAO		Initial pore air pressure (Absolute)*
T		Not used
		(*) Not used for NW = 0

Card Group		Input Data and Definitions (Main File)
5 Continuum Element	5.3 Material Property	5.3.2.4 MODELNO, DSRNMAX, MAXCYCL, Ko, NEHNO, NRHNO MODELNO = 1 Elastic Model = 2 Von Mises Model = 3 Mohr-Coulomb Model = 4 In Situ Rock Model = 5 Generalized Hoek and Brown Model = 6 Advanced Elasto-Plastic Model (N.A.) = 7 Single Hardening Plastic Model (N.A.) = 8 JWL High Explosive Model = 9 Modified Cam Clay Model = 10 Engineering Model = 11 Joint Model = 12 Duncan and Chang Hyperbolic Model = 13 Elastic Model for SHELL element = 14 User Defined Model = 15 User Defined Model = 16 User Defined Model = 17 User Defined Model = 18 User Defined Model = 21 PM4Sand Model (N.A.) DSRNMAX = 0.0 Do not apply strain sub cycling > 0.0 Maximum strain sub increment MAXCYCL Maximum number of strain sub cycling Ko Coefficient of earth pressure at rest NEHNO Young□s modulus multiplication factor history number in Card Group 9.2.3 NRHNO Element volume multiplication factor history number in Card Group 9.2.3 Note: Ko, NEHNO, NRHNO are applicable only for MODELNO =1, 2, 3, 4, 5, 10, 12

<table>
<tr><th>Card Group</th><th colspan="3">Input Data and Definitions (Main File)</th></tr>
<tr><td>5
Continuum Element</td><td>5.3
Material Property Data</td><td>Skeleton Property for MODELNO = 1 (Elastic Model)</td><td>5.3.2.4.1
<u>For MODELNO = 1 [Elastic Model]</u>

E, v

E Young's modulus
v Poisson's ratio</td></tr>
</table>

Card Group	Input Data and Definitions (Main File)		
5 Continuum Element	5.3 Material Property Data	Skeleton Property for MODELNO = 2 (Von Mises Model)	5.3.2.4.2 For MODELNO = 2 [Von Mises Model] E, ν σ E Young's modulus ν Poisson's ratio σ Shear strength in triaxial compression

Card Group	Input Data and Definitions (Main File)		
5 Continuum Element	5.3 Material Property Data	Skeleton Property for MODELNO = 3 (Mohr-Coulomb Model)	5.3.2.4.3 For MODELNO = 3 [Mohr-Coulomb Model] E, v φ, c, K, T, ST_n, ST_s E Young's modulus v Poisson's ratio φ Internal frictional angle (°) $C = \frac{(1 - \sin\varphi)}{2\cos\varphi}\sigma_c$ C Cohesion K The ratio of the shear strength in triaxial extension to the shear strength in triaxial compression at the same pressure T Tensile strength ST_n Factor used to divide stiffness normal to tensile crack ST_s Factor used to divide shear modulus for the cracked zone Note : To ignore stiffness reduction associated with tensile crack, use ST_n=ST_s=1.0

Card Group	Input Data and Definitions (Main File)

5 Continuum Element

5.3 Material Property Data

Skeleton Property for MODELNO = 4 (In Situ Rock Model)

5.3.2.4.4

For MODELNO = 4 [In Situ Rock Model]

E, v
m, s, σ_c, K, T, ST_n, ST_s

E	Young's modulus
v	Poisson's ratio
φ	Internal frictional angle (°)
C	Cohesion $C = \frac{(1 - \sin\varphi)}{2 \cos\varphi} \sigma_c$
K	The ratio of the shear strength in triaxial extension to the shear strength in triaxial compression at the same pressure
T	Tensile strength
ST_n	Factor used to divide stiffness normal to tensile
ST_s	Factor used to divide shear modulus for the cracked zone
m,s	Hoek and Brown material parameters See Table 3
σ_c	Unconfined compressive strength

Note :
To ignore stiffness reduction associated with tensile crack, use ST_n = ST_s =1.0

Table 3 Hoek and Brown Material Parameters (m, s)

Rock Type / Rock Quality	Dolomite, Limestone & Marble	Mudstone, Siltstone, Shale and Slate (normal to cleavage)	Sandstone and Quartzite	Andesite, Dolerite & Rhyolite	Amphibolite, Gabbro, Gneiss, Norite and Quartz-Diorite
Intact CSIR rating = 100 NGI rating = 150	m = 7 s = 1	10.0 1.0	15.0 1.0	17.0 1.0	25.0 1.0
Very Good Quality CSIR rating = 85 NGI rating = 100	3.5 0.1	5.0 0.1	7.5 0.1	8.5 0.1	12.5 0.1
Good Quality CSIR rating = 65 NGI rating = 10	0.7 0.004	1.0 0.004	1.5 0.004	1.7 0.004	2.5 0.004
Fair Quality CSIR rating = 44 NGI rating = 1	0.14 0.001	0.20 0.0001	0.3 0.0001	0.34 0.0001	0.5 0.0001
Poor Quality CSIR rating = 23 NGI rating = 0.1	0.04 0.00001	0.05 0.00001	0.08 0.00001	0.09 0.00001	0.13 0.00001
Very Poor Quality CSIR rating = 3 NGI rating = 0.01	0.007 0.0	0.01 0.0	0.015 0.1	0.017 0.0	0.025 0.0

Table 3 Continued

Description of Rock Quality

Intact Rock Samples	Laboratory size specimens free from joints
Very Good Quality Rock Mass	Tightly interlocking undisturbed rock with unweathered joints at 1 to 3m
Good Quality Rock Mass	Fresh to slightly weathered rock, slightly disturbed with joints at 1 to 3m
Fair Quality Rock Mass	Several sets of moderately weathered joints spaced at 0.3 to 1m
Poor Quality Rock Mass	Numerous weathered joints at 30 to 500mm with sane gouge. Clean compacted waste rock
Very Poor Quality Rock Mass	Numerous heavily weathered joints spaced < 50m with gouge. Waste rock with fines

Card Group			Input Data and Definitions (Main File)
5 Continuum Element	5.3 Material Property	Skeleton Property for MODELNO = 5 (Generalized Hoek and Brown Model)	5.3.2.4.5

For MODELNO = 5 [Generalized Hoek & Brown Model]

Elastic Parameters
E, v

E Young's Modulus
v Poisson's Ratio

Tensile Strength Parameters
NTCUT

NTCUT = 0 No tension cut-off
= 1 Tension cut-off

For NTCUT = 1, otherwise go to next Card
T, St_n, St_s

T Tensile strength
ST_n Factor used to divide stiffness normal to tensile crack
ST_s Factor used to divide shear modulus for cracked zone

Note: To ignore stiffness reduction associated with tensile crack, use ST_n=ST_s=1.0

Strength Parameters
A_1, A_2, A_3, A_4, A_5, A_6, A_7, A_8

1. Von Mises (A_1 = 0.0)
   $F = q - A_4\ R(\theta)$
   $A_2 = A_3 = 0.0$
   $A_4 = A_6 = q_{VM} = \sigma$
   Refer to Card 5.3.2.4.2

Card Group	Input Data and Definitions (Main File)		
5 Continuum Element	5.3 Material Property	Skeleton Property for MODELNO = 5 (Generalized Hoek and Brown Model)	5.3.2.4.5 2. Hoek and Brown (A_1 = 0.5) $F = q - ((A_2 + A_3 p)^? + A_4) R(\theta)$ $A_2 = (m^2 / 36 + s) \sigma_c^2$ $A_3 = m \sigma_c$ $A_4 = m \sigma_c / 6$ Refer to Card 5.3.2.4.4 3. Mohr-Coulomb (A_1 = 1.0) $F = q - ((A_2 + A_4) + A_3 p) R(\theta)$ $A_2 + A_4 = 3 \sigma_c (1 - \sin\varphi) / (3 - \sin\varphi)$ $A_3 = 6 \sin\varphi / (3 - \sin\varphi)$ Refer to Card 5.3.2.4.3 4. Quadratic (A_1 = 2.0) $F = q - (A_2 + A_3 p + A_4 p^2) R(\theta)$ 5. Elliptic (A_1 = 3.0) $F = q - (A_3 + (A_6 - A_3) (1 - ((p - A_2)/A_4)^2)^{1/2}) R(\theta)$ $A_5 = K$ (See notes in previous page) $A_6 = q_{VM}$ (Von Mises limit stress) The mean pressure (p_o) at which it reaches Von Mises limit stress (q_{VM}) is given by: For A1 = 0.0, $p_o = \infty$ For A1 = 0.5, $p_o = ((A_6 - A_4)^2 - A_2)/A_3$ For A1 = 1.0, $p_o = (A_6 - (A_2 + A_4))/A_3$ For A1 = 2.0, $p_o = (-A_3 + (A_3^2 - 4A_4 (A_2 - A_6))^{1/2})/(2A_4)$ For A1 = 3.0, $p_o = A_2$

Card Group	Input Data and Definitions (Main File)

5 Continuum Element

5.3 Material Property

Skeleton Property for MODELNO = 5 (Generalized Hoek and Brown Model)

5.3.2.4.5

$A_7 = p_{BD}$ Brittle-Ductile transition pressure
$A_8 = r_i$ Initial dilatancy parameter

Dilatancy parameter r is calculated as

For $p_{BD} > 0.0$ and $p < p_{BD}$
$r = r_i \ (1 - p / p_{BD})$

For $p_{BD} > 0.0$ and $p \geq p_{BD}$
$r = 0.0$

For $p_{BD} = 0.0$
$r = r_i$

Note: Derivatives of potential function (Q) are related to the yield function (F) as
$\partial Q / \partial p = (\partial F / \partial p) \ r$
$\partial Q / \partial q = (\partial F / \partial q)$
$\partial Q / \partial \theta = (\partial F / \partial \theta)$

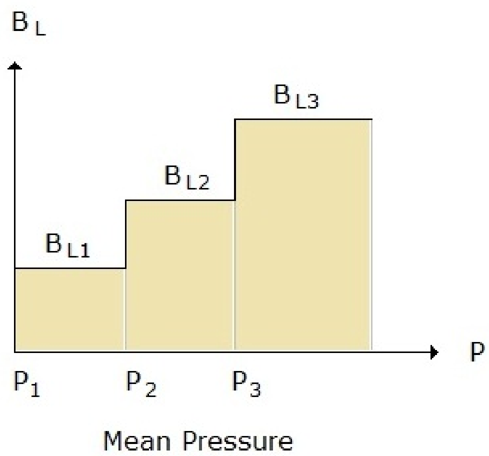
For associated flow rule
use $A_7 = 0.0$, $A = 1.0$
and set ISYMSOL=1 in Card 1.1

For no plastic volume change
use $A_7 = 0.0$, $A_8 = 0.0$
and set ISYMSOL=2 in Card 1.1

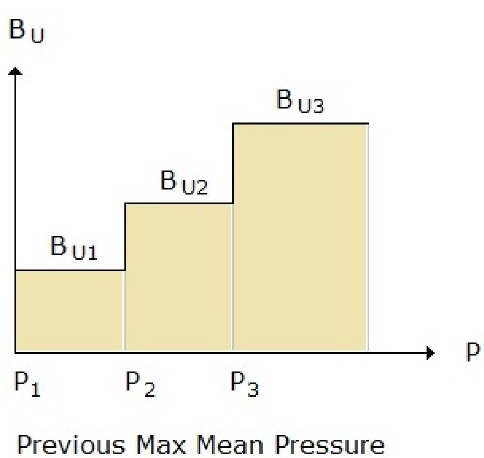
For all non associated cases
set ISYMSOL=2 in Card 1.1

Card Group			Input Data and Definitions (Main File)
5 Continuum Element	5.3 Material Property	Skeleton Property for MODELNO = 5 (Generalized Hoek and Brown Model)	5.3.2.4.5 Pressure - Dependent Moduli IBULK, ISHEAR IBULK = 0 Constant bulk modulus = 1 Nonlinear bulk modulus ISHEAR = 0 Constant shear modulus = 1 Constant Poisson's ratio Loading Bulk Modulus Definition NLPC NLPC Number of volumetric pressure/modulus pairs describing the virgin loading bulk modulus NLPC Cards: P_1, B_{L1} P_2, B_{L2} - - P_n, B_{Ln} P_i, B_{Li} Pressure and bulk modulus pairs Unloading Bulk Modulus Definition NUPC NUPC Number of volumetric pressure/modulus pairs describing unloading bulk modulus NUPC Cards: P_1, B_{U1} P_2, B_{U2} - - P_n, B_{Un} P_i, B_{Ui} Pressure and bulk modulus pairs

Loading Bulk Modulus as a Function of Mean Pressure



Unloading Bulk Modulus as a Function of Previous Max Pressure



Card Group	Input Data and Definitions (Main File)		
5 Continuum Element	5.3 Material Property	Skeleton Property for MODELNO = 6 (Advanced Plastic Model)	5.3.2.4.6 For MODELNO = 6 [Advanced Elasto-plastic Model] Not Available

Card Group	Input Data and Definitions (Main File)		
5 Continuum Element	5.3 Material Property	Skeleton Property for MODELNO = 7 (Single Hardening Elasto-Plastic Model)	5.3.2.4.7

For MODELNO = 7 [Single Hardening Elasto-Plastic Model]

Not Available

Precision Parameters

NDVMIN, NDVMAX, DEEMAX, NUNLOAD, NDRIFT

NDVMIN		Min number of plastic strain sub increment
NDVMAX		Max number of plastic strain sub increment
DEEMAX		Max plastic strain sub increment
NUNLOAD	= 0	Smooth initial unloading
	= 1	No smooth unloading
NDRIFT	= 0	Drift correction
	= 1	No drift correction

Tensile Strength

APEX, ATMO

APEX	Tensile strength T
ATMO	Atmospheric pressure P_a

Elastic Constant

AKUR, AN, APOI

AKUR	Elastic Young's modulus constant Kur
AN	Elastic Young's modulus exponent n
APOI	Elastic Poisson's ratio v

Isotropic Hardening

NACRV

AACC(I), AAPC(I), ABRK(I) I = 1, NACRV

NACRV	Number of segments for isotropic hardening function
AACC	Isotropic hardening constant C

Card Group	Input Data and Definitions (Main File)		
5 Continuum Element	5.3 Material Property	Skeleton Property for MODELNO = 7 (Single Hardening Elasto-Plastic Model)	5.3.2.4.7

AAPC Isotropic hardening constant P
ABRK Break point in terms of W_p/P

Failure Constant
AK, AMY, AETA1

AK The ratio of triaxial extensive to compressive strength at a given pressure
AMY Failure exponent m
AETA1 Failure constant η_1

Yield Constant
AY1, AH, ALPHA

AY1 Yield constant ψ_1
AH Yield constant h
ALPHA Yield constant α

Potential Constant
AY2, AMU

AY2 Potential constant ψ_2
AMU Potential constant μ

Unload/Reload Constant
AHLAM, AHGAM, AHBET, APCO

AHLAM (λ), AHGAM (γ), AHBET (β)
These unload/reload constants are not used

APCO Effective mean pressure at which yielding begins

Card Group	Input Data and Definitions (Main File)

5 Continuum Element

5.3 Material Property

Skeleton Property for MODELNO = 8 (JWL High Explosive Model)

5.3.2.4.8

For MODELNO = 8 [JWL High Explosive Model]

Elastic Constant

E, v

Note: When using JWL model, specify NLNR = 1 and NGEN = 1 in Card 2

JWL Model Parameters

A, B, R_1, R_2, ω, E_v

A	JWL material constant (Megabar)
B	JWL material constant (Megabar)
R_1	JWL material constant (Dimensionless)
R_2	JWL material constant (Dimensionless)
ω	JWL material constant (Dimensionless)
E_v	Chemical energy density of explosive (Megabar cc/cc)

Burn Fraction Parameters

C_d, B_s, XL

C_d	Detonation velocity
B_s	Constant used to spread the detonation front [Usually set B_s = 2.5]
XL	Characteristic length of element If XL = 0.0, program computes XL Note: If C_d = 0 and B_s = 0, XL represents LHNO (Pressure Load History Number) specified in Cards 9.2.3.1 through 9.2.3.5 and above JWL parameters are ignored

Card Group			Input Data and Definitions (Main File)
5 Continuum Element	5.3 Material Property Data	Skeleton Property for MODELNO = 9 (Modified Cam Clay Model)	5.3.2.4.9 For MODELNO = 9 [Modified Cam Clay Model] Cam-Clay Material Parameters P_c, e_o, v, C_c, C_r, M, G_o P_c Preconsolidation pressure e_o Initial void ratio v Poisson's ratio C_c Virgin compression index C_r Swelling/recompression index M Strength parameter G_o Initial elastic shear modulus at P_c When G_o = 0, shear modulus is computed based on v Tensile Strength Parameters NTCUT NTCUT = 0 No tension cut-off = 1 Tension cut-off For NTCUT = 1, otherwise go to next Card T, ST_n, ST_s T Tensile strength ST_n Factor used to divide stiffness normal to tensile crack ST_s Factor used to divide shear modulus for cracked zone Note: To ignore stiffness reduction associated with tensile crack, use ST_n = ST_s = 1.0

Card Group	Input Data and Definitions (Main File)

5 Continuum Element

5.3 Material Property Data

Skeleton Property for MODELNO =9 (Modified Cam Clay Model)

5.3.2.4.9

Creep Option

NCREEP

NCREEP = 0 No creep
= 1 Only volumetric creep
= 2 Only deviatoric creep
= 3 Both volumetric and deviatoric creep

Volumetric Creep Parameters (For NCREEP = 1 or 3)

t_{vi}, C_{α}

t_{vi} Initial volumetric age
C_{α} Secondary compression coefficient

Deviatoric Creep Parameters (For NCREEP = 2 or 3)

t_{di}, A, α, m

t_{di}, Initial deviatoric age
A Sing-Mitchell creep parameter
α Sing-Mitchell creep parameter
m Sing-Mitchell creep parameter

Note: Deviatoric creep is not available

Card Group	Input Data and Definitions (Main File)

5 Continuum Element

5.3 Material Property Data

Skeleton Property for MODELNO = 10 (Engineering Model)

5.3.2.4.10

<u>For MODELNO = 10 [Engineering Model]</u>

<u>Strength Parameters</u>

NSTYPE
ST1, Y1, S1, VM1

NSTYPE	= 1	Single failure surface
	= 2	Double falling failure Surface

	<u>Peak</u>
ST1	Tensile failure limit
Y1	Yield stress intercept
S1	Slope
VM1	Von Mises limit

<u>For NSTYPE = 2</u>

FSRATE
ST2, Y2, S2, VM2

FSRATE	Rate of deviatoric plastic strain at which failure surface drops to residual level
	<u>Residual</u>
ST2	Tensile failure limit
Y2	Yield stress intercept
S2	Slope
VM2	Von Mises limit

<u>Loading Modulus</u>

NLS
EBL(i), BKL(i), POL(i) i = 1, NLS

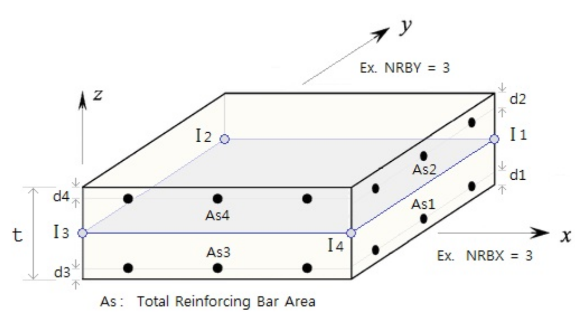
NLS	Number of loading slopes
EBL(i)	Volume strain breakpoint between loading slopes i and i+1
BKL(i)	Bulk modulus for loading slope i
POL(i)	Poisson's ratio for loading slope i

Card Group	Input Data and Definitions (Main File)		
5 Continuum Element	5.3 Material Property Data	Skeleton Property for MODELNO = 10 (Engineering Model)	5.3.2.4.10 <u>Unloading Modulus</u> NUS PBU(i), BKU(i), POU(i) i = 1, NUS NUS — Number of unloading slopes PBU(i) — Pressure breakpoint between unloading slopes i and i+1 BKU(i) — Bulk modulus for unloading slope i POU(i) — Poisson's ratio for unloading slope i Note: Special case for NLS = 1 1. Loading and unloading modulus are assumed to be the same Input data for unloading Modulus is not considered 2. Tension cutoff is based on individual principal stress. The limit of tensile stress is equal to ST1 / 3

Card Group			Input Data and Definitions (Main File)
5 Continuum Element	5.3 Material Property Data	Skeleton Property for MODELNO = 11 (Joint Model)	5.3.2.4.11 For MODELNO = 11 [Joint Model] Elastic Modulus and Thickness NM E, G, t, v NM = 0 Linear elastic joint = 1 Nonlinear joint = 2 Lumped nonlinear joint = 3 Contact nonlinear joint = 4 Thin Layer Element E Elastic Young's modulus G Elastic shear modulus t Joint thickness v Poisson's ratio (Used for NM = 4) Strength Parameters (Only for NM > 0) C, φ, r C Cohesion φ Friction angle (°) r = -1 Decoupled volume and shear = 0 No plastic volume change (N.A.) = 1 Associated flow rule (N.A.) = -2 Decoupled shear (N.A.)

Card Group			Input Data and Definitions (Main File)
5 Continuum Element	5.3 Material Property Data	Skeleton Property for MODELNO = 11 (Joint Model)	5.3.2.4.11 <u>Normal Stress-Strain Relation (Only for NM = 1,2,3)</u> ε_1, σ_1 ε_2, σ_2 ε_3, σ_3 ε_4, σ_4 ε_i, σ_i Pair of strain and stress to define normal stress-strain relation (Tension is positive) <u>Tensile Strength (Only for NM = 4)</u> TENSTR TENSTR Tensile strength Note: 1. For t > 0.0, coordinateso of joint element is adjusted based on t 2. For t < 0.0, no adjustment of coordinates. Users input mesh should represent joint thickness t 3. For t = 0.0 and NM = 4, joint thickness by user's input coordinate 4. Lumped nonlinear joint (NM=2) has better performance than nonlinear joint (NM=1). Contact nonlinear joint (NM=3) has no resistance in shear.

Card Group			Input Data and Definitions (Main File)
5 Continuum Element	5.3 Material Property Data	Skeleton Property for MODELNO = 12 (Duncan and Chang Hyperbolic Model)	5.3.2.4.12 For MODELNO = 12 [Duncan and Chang Hyperbolic Model] A_1, A_2, A_3, A_4, A_5, R_f A_1 = 1.0 A_2 = 1000. A_3 = 6 sinφ / (3 - sinφ) A_4 = 6 cosφ C / (3 - sinφ) - 1000 A_5 = 1.0 R_f = 0.7 ~ 0.9 Loading Bulk Modulus Definition NLPC NLPC Number of volumetric strain/modulus/ Poisson's ratio pairs describing the virgin loading NLPC Cards EBL_1, BKL_1, POL_1 EBL_2, BKL_2, POL_2 - - - EBL_n, BKL_n, POL_n EBL, BKL, POL Refer to Card 5.3.2.4.10 Unloading Bulk Modulus Definition NUPC NUPC Number of volumetric pressure/modulus /Poisson's ratio pairs describing the unloading NUPC Cards PBU_1, BKU_1 , POU_1 PBU_2, BKU_2, POU_2 - - - PBU_n, BKU_n, POU_n PBU, BKU, POU Refer to Card 5.3.2.4.10

Card Group	Input Data and Definitions (Main File)		
5 Continuum Element	5.3 Material Property Data	MODELNO = 13 (Elastic Model for SHELL Element)	5.3.2.4.13 For MODELNO = 13 [Elastic Model for SHELL Element] E, ν, t, FACIN, FACBD, MR_{12}, MR_{23}, MR_{34}, MR_{41}, IRB For IRB =1 [Include Reinforcing Bars] E_S, NRBX, NRBY, d_1, As_1, d_2, As_2, d_3, As_3, d_4, As_4 E, ν — Young's modulus, Poisson's ratio t — Shell thickness FACIN — Multiplication factor for in-plane stiffness. FACBD — Multiplication factor for bending stiffness. $E_{in\text{-}plane}$ = FACIN • E, $E_{bending}$ = FACBD • E. For only in-plane deformation, FACBD = 0.0 MR_{ij} — Moment release flag along the edge I_i - I_j MR = 0: No hinge MR = 1: Hinge IRB — Include Reinforcing Bars. 0: No, 1: Yes E_S — Young's modulus of reinforcing bars NRBX — Number of reinforcing bars in x direction NRBY — Number of reinforcing bars in y direction d_i, As_i — Cover depth, reinforcing bar area 

Card Group	Input Data and Definitions (Main File)		
5 Continuum Element	5.3 Material Property Data	MODELNO = 14 (User Defined Model)	5.3.2.4.14 For MODELNO = 14 [User Defined Model] 60 Cards: PROP (41) PROP (42) - - PROP (100) PROP (41) - PROP (100): Material constants related to the User's Model. Note: 1. Users can use their own material model by modifying file MODEL14.FOR in the directory C:\SMAP\SMAP3D\PROGRAM\USER\MODEL-14. Input material constants and state variables to the User's Material Model are described in detail in source file MODEL14.FOR. 2. MODEL14.FOR can be compiled by Microsoft Fortran PowerStation 4.0 using the batch file MAKE14.BAT. 3. Text file LABEL14.DAT can be modified appropriately. 4. Dynamic Link Library file MODEL14.DLL can be obtained once compiled. MODEL14.DLL should be saved in the directory C:\SMAP\SMAP3D\PROGRAM.

Card Group			Input Data and Definitions (Main File)
5 Continuum Element	5.3 Material Property Data	MODELNO = 15 (User Defined Model)	5.3.2.4.15 For MODELNO = 15 [User Defined Model] 60 Cards ┌ PROP (41) │ PROP (42) │ - │ - └ PROP (100) PROP (41) - PROP (100): Material constants related to the User's Model. Note: 1. Users can use their own material model by modifying file MODEL15.FOR in the directory C:\SMAP\SMAP3D\PROGRAM\USER\MODEL-15. Input material constants and state variables to the User's Material Model are described in detail in source file MODEL15.FOR. 2. MODEL15.FOR can be compiled by Microsoft Fortran PowerStation 4.0 using the batch file MAKE15.BAT. 3. Text file LABEL15.DAT can be modified appropriately. 4. Dynamic Link Library file MODEL15.DLL can be obtained once compiled. MODEL15.DLL should be saved in the directory C:\SMAP\SMAP3D\PROGRAM.

Card Group			Input Data and Definitions (Main File)
5 Continuum Element	5.3 Material Property Data	MODELNO = 16 (User Defined Model)	5.3.2.4.16 For MODELNO = 16 [User Defined Model] 60 Cards: PROP (41) PROP (42) - - PROP (100) PROP (41) - PROP (100): Material constants related to the User's Model. Note: 1. Users can use their own material model by modifying file MODEL16.FOR in the directory C:\SMAP\SMAP3D\PROGRAM\USER\MODEL-16. Input material constants and state variables to the User's Material Model are described in detail in source file MODEL16.FOR. 2. MODEL16.FOR can be compiled by Microsoft Fortran PowerStation 4.0 using the batch file MAKE16.BAT. 3. Text file LABEL16.DAT can be modified appropriately. 4. Dynamic Link Library file MODEL16.DLL can be obtained once compiled. MODEL16.DLL should be saved in the directory C:\SMAP\SMAP3D\PROGRAM.

Card Group			Input Data and Definitions (Main File)
5 Continuum Element	5.3 Material Property Data	MODELNO = 17 (User Defined Model)	5.3.2.4.17 For MODELNO = 17 [User Defined Model] 60 Cards PROP (41) PROP (42) - - PROP (100) PROP (41) - PROP (100): Material constants related to the User's Model. Note: 1. Users can use their own material model by modifying file MODEL17.FOR in the directory C:\SMAP\SMAP3D\PROGRAM\USER\MODEL-17. Input material constants and state variables to the User's Material Model are described in detail in source file MODEL17.FOR. 2. MODEL17.FOR can be compiled by Microsoft Fortran PowerStation 4.0 using the batch file MAKE17.BAT. 3. Text file LABEL17.DAT can be modified appropriately. 4. Dynamic Link Library file MODEL17.DLL can be obtained once compiled. MODEL17.DLL should be saved in the directory C:\SMAP\SMAP3D\PROGRAM.

Card Group			Input Data and Definitions (Main File)
5 Continuum Element	5.3 Material Property Data	MODELNO = 18 (User Defined Model)	5.3.2.4.18 For MODELNO = 18 [User Defined Model] 60 Cards: PROP (41) PROP (42) - - PROP (100) PROP (41) - PROP (100): Material constants related to the User's Model. Note: 1. Users can use their own material model by modifying file MODEL18.FOR in the directory C:\SMAP\SMAP3D\PROGRAM\USER\MODEL-18. Input material constants and state variables to the User's Material Model are described in detail in source file MODEL18.FOR. 2. MODEL18.FOR can be compiled by Microsoft Fortran PowerStation 4.0 using the batch file MAKE18.BAT. 3. Text file LABEL18.DAT can be modified appropriately. 4. Dynamic Link Library file MODEL18.DLL can be obtained once compiled. MODEL18.DLL should be saved in the directory C:\SMAP\SMAP3D\PROGRAM.

Card Group		Input Data and Definitions (Main File)
5 Continuum Element	5.3 Skeleton Property for MODELNO = 21 (PM4Sand Model)	5.3.2.4.21

For MODELNO = 21 [PM4Sand Model]

D_R G_o h_{po} p_a N_s S_{cheme} T_{antyp}

Secondary Parameters (Skip these cards for N_s = 1)

h_o e_{max} e_{min} n^b n^d A_{do}
z_{max} c_z C_e ϕ_{cv} v_o C_{GD}
C_{DR} c_{kaf} Q R m $F_{sed.min}$ p_{sed}

D_R Apparent relative density (Fraction)
G_o Shear modulus coefficient
h_{po} Contraction rate parameter
p_a Atmospheric pressure (10.33 for stress unit t/m^2)
N_s Secondary parameter specification: 0 = Yes, 1 = No
S_{cheme} Integration scheme (Use S_{cheme} = 0)
T_{antyp} Drift correction method (Use T_{antyp} = 0)

h_o Control parameter for ratio of plastic to elastic modulus
e_{max} Maximum void ratio (Default 0.8)
e_{min} Minimum void ratio (Default 0.5)
n^b Control parameter for dilatancy & peak friction angle
n^d Control parameter for transition from contr. to dilation
A_{do} Bolton's dilatancy parameter

z_{max} Maximum allowable fabric dilatancy tensor z
c_z Control parameter when fabric effects get important
C_e Control parameter for adjusting strain accumulation rate
ϕ_{cv} Critical state effective friction angle (Default 33˚)
v_o Poisson's ratio (Default 0.3)
C_{GD} Factor for shear modulus degradation (Default 2.0)

C_{DR} Control parameter for rotated dilatancy surface
c_{kaf} Control parameter for effects of sustained shear stress
Q, R Parameters for Bolton's empirical critical state line
m Parameter defining size of yield stress (Default 0.01)
$F_{sed.min}$ Parameter for post-shaking elastic modulus reduction
p_{sed} Mean effective stress for post-shaking reconsolidation

Set -1 for default values of secondary model parameters.

For description, refer to Boulanger, R. W. And ziotopoulou, k. PM4Sand (Version 3.1): A Sand Plasticity Model for Earthquake Engineering Applications, Report No UCD/CGM-17/01, Dept. of Civil & Env. Eng., U. of Cal., Davis, CA, 109 pp.

Card Group		Input Data and Definitions (Main File)
5 Continuum Element	5.5 Skew Boundary	5.5.1 NSKEW NSKEW Number of element sides on a skew boundary. Not available (Use NSKEW = 0)

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Main File)</th></tr>
<tr><td rowspan="4">5

Continuum Element</td><td rowspan="4">5.6

Initial Stress</td><td>5.6.1
IEFST

IEFST = 0 Zero initial effective stress
= 1 Specified initial effective stress</td></tr>
<tr><td>5.6.2
If IEFST =1, list initial effective stresses for each element

SXX, SYY, SZZ (NCONT Cards)

SXX σ_x' (Normal stress in x direction)
SYY σ_y' (Normal stress in y direction)
SZZ σ_z' (Normal stress in z direction)

Note: For joint element (KS > 0), SZZ represents joint nomal stress and SXX = SYY = 0.0.</td></tr>
<tr><td>5.6.3
IPOFP

IPOFP = 0 Zero initial pore fluid pressure
= 1 Specified initial pore fluid pressure</td></tr>
<tr><td>5.6.4
If IPOFP =1, list initial pore fluid pressure for each element

PRF (NCONT Cards)

PRF List initial pore fluid pressures for each element, specified sequentially from 1 to NCONT</td></tr>
</table>

Card Group				Input Data and Definitions (Main File)
5 Continuum Element	5.7 Element Surface	5.7.1		NUMEST, MATEST NUMEST — Number of material & element surface traction MATEST — Number of material surface traction If NUMEST = 0, go to Card Group 6
		For Each Material / Element Surface	5.7.2.1	(MATEST) Cards MAT, KP, KH, KD, a_0, a_1, a_2, a_3 (NUMEST - MATEST) Cards NEL, KP, KH, KD, a_0, a_1, a_2, a_3 MAT — Material number NEL — Element number KP — Element surface designation number KH — Load history number specified in Cards 9.2.3.1 through 9.2.3.5. If KH=0, constant static pressure/traction vector is acting all the time. KD = 0 — Uniformly distributed traction vector defined in local coordinate system $P'_n = a_0$ $P_x=a_1$ $P_y=a_2$ $P_z=a_3$ = 1 — Uniformly distributed traction vector defined in global coordinate system $P'_n = a_0$ $P_X=a_1$ $P_Y=a_2$ $P_Z=a_3$ P'_n is static normal pressure. (Compression is positive) = 2 — Linearly distributed static normal pressure $P_{n4}= a_0$ at I_4' $P_{n1}= a_1$ at I_1' $P_{n2}= a_2$ at I_2' $P_{n3}= a_3$ at I_3'

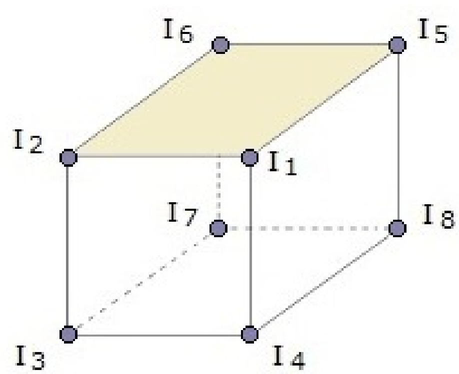
Card Group			Input Data and Definitions (Main File)
5 Continuum Element	5.7 Element Surface	For Each Material / Element Surface	5.7.2.1 Linearly distributed surface tractions defined in global coordinate system = 3 q_X $q_{X4} = a_0$ at I_4' $q_{X1} = a_1$ at I_1' $q_{X2} = a_2$ at I_2' $q_{X3} = a_3$ at I_3' = 4 q_Y $q_{Y4} = a_0$ at I_4' $q_{Y1} = a_1$ at I_1' $q_{Y2} = a_2$ at I_2' $q_{Y3} = a_3$ at I_3' = 5 q_Z $q_{Z4} = a_0$ at I_4' $q_{Z1} = a_1$ at I_1' $q_{Z2} = a_2$ at I_2' $q_{Z3} = a_3$ at I_3' = 6 Static normal pressure given as functions of global X, Y and Z coordinates $P'_n = a_0 + a_1 X + a_2 Y + a_3 Z$ Global surface traction given as functions of global X, Y and Z coordinates = 7 q_X $q_X = a_0 + a_1 X + a_2 Y + a_3 Z$ = 8 q_Y $q_Y = a_0 + a_1 X + a_2 Y + a_3 Z$ = 9 q_Z $q_Z = a_0 + a_1 X + a_2 Y + a_3 Z$ Note1: Element traction is not available for KS = -1 (High Explosive Solid Element) Note2: (NEL1, -NEL2) generates the same surface traction from NEL1+1 to NEL2. This also applies to material based traction. Refer to description in next page

Element Surface Designation and Local Axes

KP	8-Node Brick				6-Node Wedge			
	I_1'	I_2'	I_3'	I_4'	I_1'	I_2'	I_3'	I_4'
1	5	6	2	1	5	6	2	1
2	6	7	3	2	6	7	3	2
3	7	8	4	3	7	5	1	3
4	8	5	1	4	0	0	0	0
5	1	2	3	4	1	2	3	0
6	6	5	8	7	6	5	7	0

For Shell element, use KP = 1.

For KP = 1



Card Group	Input Data and Definitions (Main File)
6 Beam Element	6.1 NBEAM NBEAM — Total number of beam elements If NBEAM = 0, go to Card Group 7
	6.2 NBMST NBMST Use NBMST = 1
	6.3 NTNB NTNB — Number of material property set for beam element
	6.4 For Each Msterial 6.4.1 MATNO, MR, NEHNO, NFSHR, CTSy, CTSz, DAMP MATNO — Material number MR — Moment Release = 0 — No hinge = 1 — Hinge at node I = -1 — Hinge at node J = 2 — Hinge at node I and J = 3 — Joint spring element Spring Element at Node I = 11 — Axial spring ($Kx = E A / L$) = 12 — Shear(y) spring ($Ky = 12 E Iz / L^3$) = 13 — Shear(z) spring ($Kz = 12 E Iy / L^3$) = 14 — Torsional spring ($Kt = G J / L$) = 15 — Rotational(y) spring ($Kry = 4 E Iy / L$) = 16 — Rotational(z) spring ($Krz = 4 E Iz / L$) Use Negative for Spring Element at Node J NEHNO — Young's modulus multiplication factor history number in Card Group 9.2.3 NFSHR = 0 — Neglect shear deformation = 1 — Include shear deformation CTSy, CTSz — Shear coefficient for Iy, Iz DAMP — Critical damping ratio

Card Group			Input Data and Definitions (Main File)
6 Beam Element	6.4 For Each Msterial	6.4.2	

For MR ≠ 3

A, WL, RHO, E, G, J, I_y, I_z

A	Cross section area
WL	Weight per unit length of beam
RHO	Mass density
E	Young's modulus
G	Shear modulus
J	Torsional moment of inertia
I_y	Moment of inertia about member y axis
I_z	Moment of inertia about member z axis

For MR = 3

K_X, K_Y, K_Z, K_T, K_{RY}, K_{RZ}

K_X	Axial spring stiffness
K_Y	Shear(y) spring stiffness
K_Z	Shear(z) spring stiffness
K_T	Torsional spring stiffness
K_{RY}	Rotational(y) spring stiffness
K_{RZ}	Rotational(z) spring stiffness

Card Group	Input Data and Definitions (Main File)
7 Truss Element	7.1 NTRUSS NTRUSS Total number of truss elements If NTRUSS = 0, go to Card Group 8
	7.2 NTRST NTRST Use NTRST = 1
	7.3 NTNT, $MATP_1$, $MATP_2$, $MATP_3$ NTNT Number of material property set for truss element MATP Material number of parent continuum element which is not allowed to embed truss element

Card Group			Input Data and Definitions (Main File)
7 Truss Element	7.4 For Each Material	7.4.1	MATNO, ME, MS MATNO — Material number ME = 0 — No embedment = 1 — Embedded with auto subdivision = 2 — Embedded with no subdivision = 3 — Embedded using input NELPI and NELPJ. See Card 5.2 in mesh file description =-N — Embedded with N equal subdivision MS = 0 — No slip = 1 — Monotonic loading path = 2 — Arbitrary loading path = n — (n > 2) Plastic stiffness = Kslip x 10^{-n} Note: For ME = 1, 2, and -N, input files of mesh and main are automatically updated
		7.4.2	A, WL, RHO, E, STRSI, DAMP A — Cross section area WL — Weight per unit length of truss RHO — Mass density (Used for dynamic analysis) To lump all mass at node J, use -RHO E — Young's modulus STRSI — Initial stress. Tension is positive For constant initial stress, use E = 0 DAMP — Critical damping ratio Negative for viscous damping constant

Card Group		Input Data and Definitions (Main File)
7 Truss Element	7.4 For Each Material	7.4.3 If NLNR = 0 and NGEN = 0, skip this Card σ_{yc}, σ_{yt}, ε_f, I, y_{max} σ_{yc} Yield stress in compression σ_{yt} Yield stress in tension ε_f Strain at rupture For $\varepsilon_f \le \sigma_y/E$, ε_f represents Yield strain at tension I Moment of inertia (Minimum) y_{max} Distance from neutral axis to extreme fiber (Maximum) $\sigma_{yc} = \sigma_{yt} = 0$: Linear elastic material $\sigma_{yc} = 0$: No compression (Cable) $\sigma_{yt} = 0$: No tension (Strut) $I = 0$: No buckling $y_{max} = 0$: No yield on buckling
		7.4.4 If MS = 0, skip this Card Kslip, Cmax, Cres, Umax, Ures, Dslip Kslip Stiffness for shear stress - slip displacement Cmax Maximum cohesion Cres Residual cohesion (N.A.) Umax Slip at the end of Cmax (N.A.) Ures Slip at the beginning of Cres (N.A.) Dslip Diameter of slip surface

Card Group	Input Data and Definitions (Main File)
8 Element Activity	8.1 NFAD, MCFAD, MBFAD, MTFAD NFAD Number of materials / elements with activity MCFAD Number of continuum materials with activity MBFAD Number of beam materials with activity MTFAD Number of truss materials with activity If NFAD = 0, go to Card Group 9
	8.2 (MCFAD) Cards [MATC, NAC, NDAC (MBFAD) Cards [MATB, NAC, NDAC- (MTFAD) Cards [MATT, NAC, NDAC (NFAD - MCFAD - MBFAD - MTFAD) Cards [NEL, NAC, NDAC MATC Continuum material number MATB Beam material number MATT Truss material number NEL Element number NAC Load step at which an element is activated NDAC Load step at which an element is deactivated Note: If initially active and deactivated at step 5: NAC = 0, NDAC = 5 If active permanently from step 20: NAC = 20, NDAC > NCYCL (NEL1, -NEL2) generates the same activity from NEL1+1 to NEL2. This also applies to material based activity.

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Main File)</th></tr>
<tr><td rowspan="2">9

Loads</td><td rowspan="2">9.1

Gravity Load</td><td>9.1.1

NGSTEP

NGSTEP = 0 No gravity load is applied
> 0 Number of load steps (Cycles) through which the gravity load is applied incrementally

Note:
During gravity load step, inertia forces are not active

To print time history output after NGSTEP, use negative value of NGSTEP.
Output times would be relative to the time at NGSTEP</td></tr>
<tr><td>9.1.2

If NGSTEP = 0, go to Card Group 9.1.3
IRELD, FRX, FRY, FRZ, NHFRX, NHFRY, NHFRZ

IRELD = 0 Displacements/strains include gravity load
= 1 Displacements/strains after NGSTEP are relative to gravity load

FRX X component of unit gravity load
FRY Y component of unit gravity load
FRZ Z component of unit gravity load

NHFRX Intensity history number in X direction
NHFRY Intensity history number in Y direction
NHFRZ Intensity history number in Z direction

Note: Intensity is specified through Card 9.2.3. Intensity Times Distribution Factor will be additive to FRX, FRY, or FRZ</td></tr>
</table>

Card Group		Input Data and Definitions (Main File)
9 Loads	9.1 Gravity Load	9.1.2.1 If NHFRX = 0, skip this card A_0, A_1, A_2, A_3, Y_1, Y_2 A_i Distribution factor Y_i Global Y coordinate For $Y < Y_1$ $A_i = A_0$ For $Y > Y_2$ $A_i = A_3$ For others $A_i = A_1 + (Y - Y_1) * (A_2 - A_1) / (Y_2 - Y_1)$
		9.1.2.2 If NHFRY = 0, skip this card A_0, A_1, A_2, A_3, Y_1, Y_2 A_i Distribution factor Y_i Global Y coordinate
		9.1.2.3 If NHFRZ = 0, skip this card A_0, A_1, A_2, A_3, Y_1, Y_2 A_i Distribution factor Y_i Global Y coordinate

Card Group		Input Data and Definitions (Main File)
9 Loads	9.1 Specified Displacement	9.1.3 NUMDIS NUMDIS — Total number of degrees of freedom at which input displacement time histories are specified If NUMDIS = 0, go to Card Group 9.2.1
		9.1.4 For each of NUMDIS where displacement is specified NODE, IDOF, LHNO, DINT NODE — Node number Skeleton displacement IDOF = 1 x-direction = 2 y-direction = 3 z-direction Apparent relative fluid displacement = 4 x-direction = 5 y-direction = 6 Z-direction LHNO — Displacement history number corresponding to sequence of displacement specifications given in Card Group 9.1.5.3 DINT — Displacement intensity factor

Card Group			Input Data and Definitions (Main File)
9 Loads	9.1 Specified Displacement		9.1.5.1 NUMDH, NUMDTP, TDSTART, TDFAC NUMDH — Number of different input displacement time histories NUMDTP — Number of displacement-time pairs TDSTART — Starting time TDFAC — Time scale factor for TD
		For Each Load History	9.1.5.2 TD_1, TD_2, ..., TD_{NUMDTP} TD_i — Specified times
			9.1.5.3 $SDIS_1$, $SDIS_2$, ..., $SDIS_{NUMDTP}$ $SDIS_i$ — Displacement magnitude at corresponding time TD_i

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Main File)</th></tr>
<tr><td rowspan="2">9
Loads</td><td rowspan="2">9.2
Concentrated Nodal Force</td><td>9.2.1

NUMCON

NUMCON Total number of degrees of freedom at which input concentrated force time histories are specified

If NUMCON = 0, skip the rest of this Card Group</td></tr>
<tr><td>9.2.2

For each of the NUMCON where load is applied

NODE, IDOF, LHNO, CINT

NODE Node number

Total force acting on a given node
IDOF = 1 x-direction
= 2 y-direction
= 3 z-direction

Fluid force acting on a given node
= 4 x-direction
= 5 y-direction
= 6 z-direction

LHNO Load history number corresponding to sequence of load specifications given in Card Group 9.2.3.4 or 9.2.3.5

CINT Load intensity factor</td></tr>
</table>

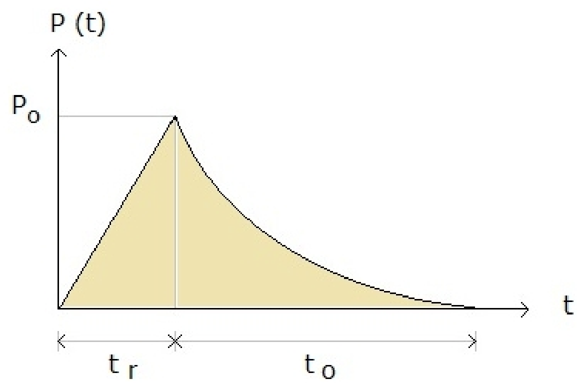
Card Group					Input Data and Definitions (Main File)
9 Loads	9.2 Concentrated Nodal Force				9.2.3.1 NTFNC, NUMCH NTFNC = 0 User-specified arbitrary force = 1 Force is specified by math functions NUMCH Number of different force time histories
		NTFNC = 0 (User-Specified Arbitrary Force)			9.2.3.2 NUMCTP, NCTYPE, DTXC, TCSTART, TCFAC NUMCTP Number of force-time pairs NCTYPE = 0 Constant time increment = 1 Specified times for all time histories = 2 Specified times for each time history DTXC Constant time interval for NCTYPE = 0 TCSTART Starting time TCFAC Time scale factor for TC
			For Each Load History		9.2.3.3 If NCTYPE = 0, go to next Card TC_1, TC_2, ..., TC_{NUMCTP} TC_i Specified times For NCTYPE =1, specify only once for the first load history
					9.2.3.4 $SCON_1$, $SCON_2$, ..., $SCON_{NUMCTP}$ $SCON_i$ Force magnitude at time TC_i

Card Group	Input Data and Definitions (Main File)		
9 Loads	9.2 Concentrated Nodal Force	NTFNC = 1 (Math Function)	9.2.3.5 For each of NUMCH loading time histories NFNC, a_1, a_2, a_3, a_4, a_5 NFNC = 1 Polynomial decaying load = 2 Exponential decaying load = 3 Trigonometric load a_1, a_2, a_3, a_4 Force function coefficients defined in the next page a_5 Starting time

Polynomial Decaying (NFNC = 1)

$a_1 = P_o \quad a_2 = t_r \quad a_3 = t_o \quad a_4 = n$

For $t_r \le t \le (t_r + t_o)$ $$P(t) = P_o \left[1 - \frac{(t - t_r)}{t_o}\right]^n$$



Exponential Decaying (NFNC = 2)

$$P(t) = a_1 + a_2 e^{a_3 t}$$

Trigonometric (NFNC = 3)

$t \le a_4 \quad P(t) = a_1 \sin(a_2 t) + a_3 \cos(a_2 t)$

$t > a_4 \quad P(t) = 0$

Card Group			Input Data and Definitions (Main File)
9 Loads	9.3 Specified Velocity	9.3.1	NUMVEL NUMVEL — Total number of degrees of freedom at which velocity histories are specified If NUMVEL= 0, skip the rest of this Card Group
		9.3.2	For each of the NUMVEL where velocity is specified NODE, IDOF, LHNO, VINT NODE — Node number IDOF — Skeleton velocity = 1 x - direction = 2 y - direction = 3 z - direction Apparent relative fluid velocity = 4 x-direction = 5 y-direction = 6 z-direction LHNO — Velocity history number corresponding to sequence of velocity specifications given in Card Group 9.3.3.4 or 9.3.3.5 VINT — Velocity intensity factor
		9.3.3.1	NTFNV, NUMVH NTFNV = 0 User-specified arbitrary velocity = 1 Velocity specified by math function NUMVH Number of different input velocity time histories

Card Group				Input Data and Definitions (Main File)
9 Loads	9.3 Specified Velocity	NTFNV = 0 (User-Specified Arbitrary Velocity)		9.3.3.2 NUMVTP, NVTYPE, DTXV, TVSTART, TVFAC NUMVTP — Number of velocity-time pairs NVTYPE = 0 — Constant time increment = 1 — Specified times for all time histories = 2 — Specified times for each time history DTXV — Constant interval for NVTYPE = 0 TVSTART — Starting time TVFAC — Time scale factor for TV
			For Each Load History	9.3.3.3 If NVTYPE = 0, go to next Card TV_1, TV_2, ..., TV_{NUMVTP} TV_i — Specified times For NVTYPE =1, specify only once for the first load history
				9.3.3.4 $SVEL_1$, $SVEL_2$, ..., $SVEL_{NUMVTP}$ $SVEL_i$ — Velocity magnitude at time TV_i

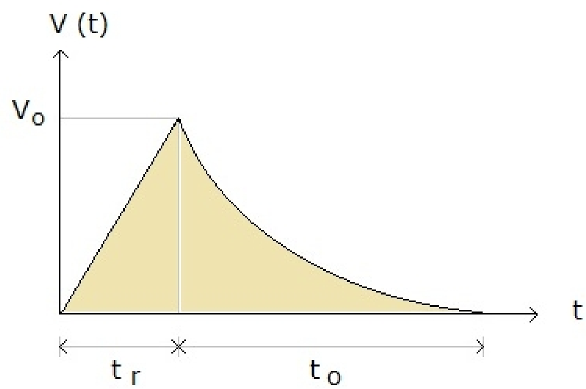
Card Group	Input Data and Definitions (Main File)		
9 Loads	9.3 Specified Velocity	NTFNV = 1 (Math Function)	9.3.3.5 For each of NUMVH velocity time histories NFNV, a_1, a_2, a_3, a_4, a_5 NFNV = 1 Polynomial decaying velocity = 2 Exponential decaying velocity = 3 Trigonometric velocity a_1,a_2,a_3,a_4 Velocity function coefficients defined in the next page a_5 Starting time

Polynomial Decaying (NFNV = 1)

$a_1 = V_o \quad a_2 = t_r \quad a_3 = t_o \quad a_4 = n$

$$V(t) = V_o \left[1 - \frac{(t - t_r)}{t_o} \right]^n$$

For $t_r \le t \le (t_r + t_o)$



Exponential Decaying (NFNV = 2)

$$V(t) = a_1 + a_2 e^{a_3 t}$$

Trigonometric (NFNV = 3)

$t \le a_4 \qquad V(t) = a_1 \sin(a_2 t) + a_3 \cos(a_2 t)$

$t > a_4 \qquad V(t) = 0$

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Main File)</th></tr>
<tr><td rowspan="2">9

Loads</td><td rowspan="2">9.4

Initial Velocity</td><td>9.4.1

NINVEL

NINVEL Number of degrees of freedom where initial velocity is applied.

If NINVEL= 0, skip the rest of this Card Group</td></tr>
<tr><td>9.4.2

For each of the NINVEL where velocity is applied
NODE, IDOF, VEL

NODE Node number

Skeleton velocity
IDOF = 1 x-direction
= 2 y-direction
= 3 z-direction

Apparent relative fluid velocity
= 4 x-direction
= 5 y-direction
= 6 z-direction

VEL Initial velocity</td></tr>
</table>

Card Group		Input Data and Definitions (Main File)
9 Loads	9.5 Specified Acceleration	9.5.1 NUMACC, MOTION, EYB, EDEN, ECP, ECS NUMACC — Total number of directions at which input acceleration time histories are specified MOTION = 0 No EHS (Elastic Half Space) = 1 EHS with base acceleration applied = 2 EHS with base shear force applied = 3 EHS with conventional base accel. applied EYB, EDEN, ECP, ECS : Half space top y coordinate, Unit weight, Compression and Shear wave speeds used for elastic half space if MOTION is not zero If NUMACC = 0, Skip the rest of this Card Group
		9.5.2 For each of NUMACC where acceleration is specified NODE, IDOF, LHNO, AINT NODE — Node number IDOF = 1 x-direction skeleton acceleration = 2 y-direction skeleton acceleration = 3 z-direction skeleton acceleration LHNO — Acceleration history number corresponding to sequence of acceleration specifications given in Card Group 9.5.3.4 or 9.5.3.5 AINT — Acceleration intensity factor Note: For uniformly distributed acceleration, set all node numbers to zero. Output motions are relative to base motion for MOTION = 0 or 1
		9.5.3.1 NTFNA, NUMAH NTFNA = 0 User-specified arbitrary acceleration = 1 Acceleration specified by math function NUMAH — Number of different input time histories

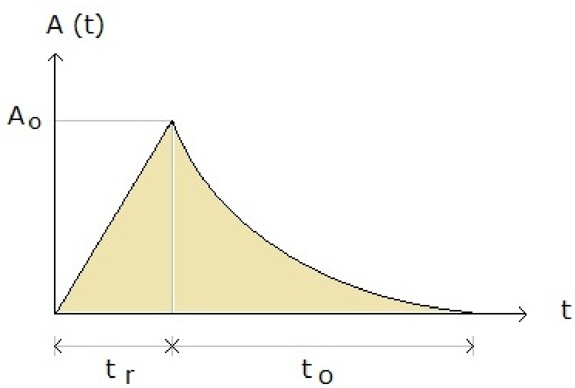
Card Group	Input Data and Definitions (Main File)
9 Loads / Specified Acceleration / NTFNA = 0 (User-Specified Arbitrary Acceleration)	9.5.3.2 NUMATP, NATYPE, DTXA, TASTART, TAFAC, IACCM NUMATP — Number of acceleration-time pairs NATYPE = 0 Constant time increment = 1 Specified times for all time histories = 2 Specified times for each time history DTXA — Constant time interval for NATYPE = 0 TASTART — Starting time TAFAC — Time scale factor for TA IACCM = 0 Input histories from Main File = 1 Input histories from External Files First 3 lines represent headers
For Each Load History	9.5.3.3 If NATYPE = 0, go to next Card TA_1, TA_2, ..., TA_{NUMATP} TA_i — Specified times For NATYPE =1, specify only once for the first load history For IACCM = 1, specified times read from Acc_Time_1.dat, Acc_Time_2.dat ...
	9.5.3.4 $SACC_1$, $SACC_2$, ..., $SACC_{NUMATP}$ $SACC_i$ — Acceleration magnitude at time TA_i For IACCM = 1, specified histories read from Acc_History_1.dat, Acc_History_2.dat ...

Card Group	Input Data and Definitions (Main File)		
9 Loads	Specified Acceleration	NTFNA = 1 (Math Function)	9.5.3.5 For each of NUMAH acceleration time histories NFNA, a_1, a_2, a_3, a_4, a_5 NFNA = 1 Polynomial decaying acceleration = 2 Exponential decaying acceleration = 3 Trigonometric acceleration a_1,a_2,a_3,a_4 Acceleration function coefficients defined in the next page a_5 Starting time

Polynomial Decaying (NFNA = 1)

$a_1 = A_o \quad a_2 = t_r \quad a_3 = t_o \quad a_4 = n$

For $t_r \le t \le (t_r + t_o)$ $\quad A(t) = A_o \left[1 - \frac{(t - t_r)}{t_o}\right]^n$



Exponential Decaying (NFNA = 2)

$$A(t) = a_1 + a_2 e^{a_3 t}$$

Trigonometric (NFNA = 3)

$t \le a_4$	$A(t) = a_1 \sin(a_2 t) + a_3 \cos(a_2 t)$
$t > a_4$	$A(t) = 0$

Card Group		Input Data and Definitions (Main File)
9 Loads	9.6 Transmitting Boundary	9.6.1 NODVIS, NELVIS, NOSVIS NODVIS Number of transmitting degrees of freedom NELVIS Number of continuum element surfaces NOSVIS Number of outer boundary surfaces (Max = 6)
		9.6.2 If NODVIS = 0, go to Card Group 9.6.3 NODE, IDOF, VISC For each of NODVIS NODE Node number IDOF = 1 Damping in x-direction = 2 Damping in y-direction = 3 Damping in z-direction VISC Constant which is proportional to the force on a given node (ρCA_c), equal to impedence times contributing area on the node $C = C_p$ for IDOF normal to the boundary $C = C_s$ for IDOF parallel to the boundary $C_{p,}$ C_s : Compression & shear wave speed
		9.6.3 If NELVIS = 0, go to next Card Group NEL, KT For each of the NELVIS NEL Element number KT Element surface designation number. Same as KP in Card Group 5.7.2.1
		9.6.4 If NOSVIS = 0, go to next Card Group NOS, VC For each of the NOSVIS NOS Outer surface number VC NOS = 1: VC = Y_{TOP} NOS = 2: VC = X_{LEFT} NOS = 3: VC = Y_{BOTOM} NOS = 4: VC = X_{RIGHT} NOS = 5: VC = Z_{FRONT} NOS = 6: VC = Z_{BACK}

Card Group	Input Data and Definitions (Main File)
10 Requested Output	10.1 NTPRNT NTPRNT — Number of cycles between output data print
	10.2.1 NHPEL NHPEL — Number of elements at which stress/strain time histories are requested
	10.2.2 If NHPEL = 0, skip the following Card NEL_1, NEL_2, ..., NEL — Element number to be printed
	10.3.1 NHPMT NHPMT — Number of nodes at which motion time histories are requested
	10.3.2 If NHPMT = 0, skip the following Card $NODE_1$, $NODE_2$, ..., $NODE_{NHPMT}$ NODE — Node numbers to be printed
	10.4.1 NTIME NTIME — Number of times at which stress/strain/motion profiles are requested
	10.4.2 If NTIME = 0, skip the following Card $TIME_1$, $TIME_2$, ..., $TIME_{NTIME}$ TIME — Time to be printed

4.5 Post File

Post File contains information which are used to show graphically the results from the main-processing program.

Post File consists of three different card groups:

- Card Group 11 (PLOT-2D)
- Card Group 12 (PLOT-XY)
- Card Group 13 (FEMAP)

Card Group 11 contains the input data which are used to plot the following snapshots in two dimension:

- Finite element mesh/element/node number
- Principal stress distribution
- Deformed shape
- Beam section force/extreme fiber stress/strain
- Truss axial force/stress/strain
- Contours of continuum element data

Card Group 12 contains the input data for the following plots:
Time history

- Stress/strain/time
- Displacement/velocity/acceleration/time

Snapshot

- Stress/strain vs. distance
- Displacement/velocity/acceleration vs. distance

Card Groups 13 is no longer supported.
These plots can be performed automatically by using PLOT-3D.

PLOT-2D

Post-Processor

Card Group	Input Data and Definitions (Post File)
11 PLOT-2D Plot Information	11.1 NPTYPE, IHOR, IVER NPTYPE = 0 End of plotting output = 1 Finite element mesh / element number = 2 Principal stress distribution = 3 Deformed shape = 4 Beam section force / fiber stress / strain = 5 Truss axial force / stress / strain = 6 Contours of continuum element data = 7 Stress state in p-q space and octahedral plane. When NPTYPE = 7 is specified, all other cases of NPTYPE are not considered. IHOR, IVER Horizontal and Vertical coordinate flags (x=1, y=2, z=3, -x=-1, -y=-2, -z=-3) If NPTYPE = 0, Skip rest of Card Group 11

Card Group		Input Data and Definitions (Post File)
11 PLOT-2D Plot Information	11.2 For NPTYPE = 1 (Finite Element Mesh / Element Number)	11.2.1 TITLE TITLE Any title (Max = 70 characters)
		11.2.2 IUNIT IUNIT = 1 Inch = 2 Cm = 3 User-specified unit
		11.2.3 For IUNIT = 3 NCHR LABEL NCHR Number of characters for mesh unit LABEL Name of mesh unit

Card Group	Input Data and Definitions (Post File)

11

PLOT-2D Plot Information

For NPTYPE = 1 (Finite Element Mesh / Element Number)

11.2.4

IMODE

IMODE	=	1	Plot finite element mesh
	=	-1	Plot element and node numbers
	=	2	Plot element numbers
	=	-2	Plot node numbers
	=	3	Plot skeleton boundary codes
	=	-3	Plot fluid boundary codes
	=	4	Plot rotational boundary codes

11.2.5

NGROUP

NGROUP	=	0	Plot all elements
	>	0	Plot specified groups (Max=1000)

11.2.6

<u>If NGROUP = 0, Skip this Card</u>

	NSS,	NEE,	NIC,	NNN
NGROUP Cards	-	-	-	-
	-	-	-	-

NSS	Starting element number in a row
NEE	Number of elements in a row
NIC	Element number increment for next row
NNN	Total number of rows

10	11	12	13
20	21	22	23
30	31	32	33

Example

NSS = 10
NEE = 4
NIC = 10
NNN = 3

Card Group		Input Data and Definitions (Post File)
11 PLOT-2D Plot Information	11.3 For NPTYPE = 2 (Principal Stress Distribution)	11.3.1 TITLE TITLE Any title (Max = 70 characters)
		11.3.2 IUNIT IUNIT = 1 In, Psi = 2 Cm, Kg/cm^2 = 3 User-specified unit
		11.3.3 For IUNIT = 3 NCHR LABEL NCHRC LABELC NCHR Number of characters for mesh unit LABEL Name of mesh unit NCHRC Number of characters for stress unit LABELC Name of stress unit

Card Group	Input Data and Definitions (Post File)	
11 PLOT-2D Plot Information	For NPTYPE = 2 (Principal Stress Distribution)	11.3.4 NLTIME, $TIME_{REF}$ $TIME_1$, $TIME_2$, ..., $TIME_{NLTIME}$ NLTIME Number of specified times (Max=1000) $TIME_{REF}$ Reference time TIME Specified time If $TIME_{REF}$ is not equal to 0.0, Stress at $TIME_i$ are relative to $TIME_{REF}$
		11.3.5 NGROUP, IAVG, ISCRIN, IMESH, IPSTRS NGROUP = 0 Plot stresses at all elements > 0 Plot stresses at specified groups (Max=1000) IAVG = 0 Do not plot averages = 1 Plot average stresses ISCRIN = 0 Do not screen the data = 1 Screen the data IMESH = 0 Do not plot meshes = 1 Plot meshes IPSTRS = 0 Do not store principal stresses = 1 Store principal stresses on file PSTRS.DAT

Card Group		Input Data and Definitions (Post File)
11 PLOT-2D Plot Information	For NPTYPE = 2 (Principal Stress Distribution)	11.3.6 If NGROUP = 0, Skip this Card NGROUP Cards: NSS, NEE, NIC, NNN - - - - - - - - Refer to Card Group 11.2.6
		11.3.7 NRL NRL — Number of nodes to be connected by a solid line (Max=5000)
		11.3.8 If NRL = 0, Skip this Card $NODE_1$, $NODE_2$, ..., $NODE_{NRL}$ NODE — Reference node numbers. If $NODE_i$ has negative sign, a New Line is drawn

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Post File)</th></tr>
<tr><td rowspan="4">11
PLOT-2D Plot Information</td><td rowspan="4">11.4
For NPTYPE = 3 (Deformed Shape)</td><td>11.4.1
TITLE

TITLE Any title of up to 70 characters</td></tr>
<tr><td>11.4.2
IUNIT

IUNIT = 1 In
= 2 Cm
= 3 User-specified unit</td></tr>
<tr><td>11.4.3
<u>For IUNIT = 3</u>
NCHR
LABEL

NCHR Number of characters for mesh and displacement unit.
LABEL Name of mesh and displacement unit</td></tr>
<tr><td>11.4.4
NLTIME, $TIME_{REF}$
$TIME_1$, $TIME_2$, ..., $TIME_{NLTIME}$

NLTIME Number of specified times (Max=1000)
$TIME_{REF}$ Reference time
TIME Specified time

If $TIME_{REF}$ is not equal to 0.0,
Displacement at $TIME_i$ are relative to $TIME_{REF}$</td></tr>
</table>

Card Group		Input Data and Definitions (Post File)
11 PLOT-2D Plot Information	11.4 For NPTYPE = 3 (Deformed Shape)	11.4.5

11.4.5

Row and Line Plots (Repeat in any order)

For Row Plot --> 1, IDISP
NSR, JCR, NJR, ICR, NIR

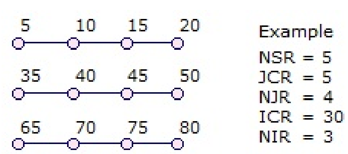
For Line Plot --> 2, IDISP
NPT
$NODE_1$, $NODE_2$, .., $NODE_{NPT}$

For End Plot --> 0, 0

IDISP = 0 Undeformed shape
= 1 Deformed shape
= 2 Displacement vector

For Row Plot (Max = 1000)

NSR	Starting node number of row plot
JCR	Node number increment in a row
NJR	Number of nodes in a row
ICR	Node number increment for next row
NIR	Total number of rows



For Line Plot (Max = 160)

NPT	Number of nodes (Max=1000)
NODE	Node number

Card Group			Input Data and Definitions (Post File)
11 PLOT-2D Plot Information	11.5 For NPTYPE = 4 (Beam Section Force / Extreme Fiber Stress / Strain)		11.5.1 TITLE TITLE Any title (Max = 70 characters)
			11.5.2 IUNIT IUNIT = 1 In, Psi = 2 Cm, Kg/cm^2 = 3 User-specified unit
			11.5.3 For IUNIT = 3 NCHR LABEL NCHRB LABELB NCHR Number of characters for mesh unit LABEL Name of mesh unit NCHRB Number of characters for section force / extreme fiber stress LABELB Name of section force / fiber stress

Card Group		Input Data and Definitions (Post File)
11 PLOT-2D Plot Information	11.5 For NPTYPE = 4 (Beam Section Force / Extreme Fiber Stress / Strain)	11.5.4 NLTIME, $TIME_{REF}$ $TIME_1$, $TIME_2$, ..., $TIME_{NLTIME}$ NLTIME — Number of specified times (Max=1000) $TIME_{REF}$ — Reference time TIME — Specified time If $TIME_{REF}$ is not equal to 0.0, Section force / Stress / Strain plots at $TIME_i$ are relative to $TIME_{REF}$
		11.5.5 NBTS NBTS = 1 Thrust = 2 Shear in member y direction = 3 Shear in member z direction = 4 Torque = 5 Bending moment about y axis = 6 Bending moment about z axis See Figure PL-4 for Sign Convention
		11.5.6 NBGROUP NBGROUP — Number of beam groups (Max=280)

Card Group			Input Data and Definitions (Post File)
11 PLOT-2D Plot Information	11.5 For NPTYPE = 4 (Beam Section Force / Extreme Fiber Stress / Strain)	For Each Beam Group	11.5.7.1 NBLIST NBLIST = 0 Elements from NFBEAM to NLBEAM = 1 Listing of individual elements
			11.5.7.2 For NBLIST = 0 NFBEAM, NLBEAM NFBEAM Starting beam element number NLBEAM Ending beam element number
			11.5.7.3 For NBLIST = 1 MBEAM N_1, N_2, ..., N_{MBEAM} MBEAM Number of beam element (Max=280) N_1, N_2, N_{MBEAM} List of element number
		11.5.8 NRL NRL Number of nodes to be connected by a Solid Line (Max=280)	
		11.5.9 If NRL = 0, Skip this Card $NODE_1$, $NODE_2$, ..., $NODE_{NRL}$ NODE Reference node numbers If $NODE_i$ has negative sign, a New Line is drawn	

Card Group	Input Data and Definitions (Post File)	
11 PLOT-2D Plot Information	11.6 For NPTYPE = 5 (Truss Axial Force / Stress / Strain)	11.6.1 TITLE TITLE Any title (Max = 70 characters)
		11.6.2 IUNIT IUNIT = 1 In, Pound = 2 Cm, Kg = 3 User-specified unit
		11.6.3 For IUNIT = 3 NCHR LABEL NCHRT LABELT NCHR Number of characters for mesh unit LABEL Name of mesh unit NCHRT Number of characters for axial data LABELT Name of axial force / stress / strain

Card Group		Input Data and Definitions (Post File)
11 PLOT-2D Plot Information	11.6 For NPTYPE = 5 (Truss Axial Force / Stress / Strain)	11.6.4 NLTIME, $TIME_{REF}$ $TIME_1$, $TIME_2$, ..., $TIME_{NLTIME}$ NLTIME Number of specified times (Max=1000) $TIME_{REF}$ Reference time TIME Specified times If $TIME_{REF}$ is not equal to 0.0, Force / Stress / Strain at $TIME_i$ are relative to $TIME_{REF}$
		11.6.5 NTTS NTTS = 1 Axial force = 2 Axial stress = 3 Axial strain
		11.6.6 NTGROUP NTGROUP Number of truss groups (Max=100)

Card Group	Input Data and Definitions (Post File)
11 PLOT-2D Plot Information For NPTYPE = 5 (Truss Axial Force / Stress / Strain) For Each Truss	11.6.7.1 NTLIST NTLIST = 0 Elements from NFTRUS to NLTRUS = 1 Listing of individual elements
	11.6.7.2 <u>For NTLIST = 0</u> NFTRUS, NLTRUS NFTRUS Starting truss element number NLTRUS Ending truss element number
	11.6.7.3 <u>For NTLIST = 1</u> MTRUS N_1, N_2, ..., N_{MTRUS} MTRUS Number of element (Max=280) N_1, N_2, N_{MTRUS} List of element number
	11.6.8 NRL NRL Number of nodes to be connected by a Solid Line (Max=280)
	11.6.9 <u>If NRL = 0, Skip this Card</u> $NODE_1$, $NODE_2$, ..., $NODE_{NRL}$ NODE Reference node numbers If $NODE_i$ has negative sign, a New Line is drawn

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Post File)</th></tr>
<tr><td rowspan="4">11

PLOT-2D Plot Information</td><td rowspan="4">For NPTYPE = 6 (Contours of Continuum Element Data)</td><td>11.7.1

TITLE

TITLE Any title (Max = 70 characters)</td></tr>
<tr><td>11.7.2

IUNIT

IUNIT = 1 In, Pound
= 2 Cm, Kg
= 3 User-specified unit</td></tr>
<tr><td>11.7.3

<u>For IUNIT = 3</u>

NCHR
LABEL
NCHRC
LABELC

NCHR Number of characters for mesh unit
LABEL Name of mesh unit
NCHRC Number of characters for contouring data
LABELC Name of contouring data</td></tr>
<tr><td>11.7.4

NLTIME, $TIME_{REF}$
$TIME_1$, $TIME_2$, ..., $TIME_{NLTIME}$

NLTIME Number of specified times (Max=1000)
$TIME_{REF}$ Reference time
TIME Specified time

If $TIME_{REF}$ is not equal to 0.0,
Contour plots at $TIME_i$ are relative to $TIME_{REF}$</td></tr>
</table>

Card Group		Input Data and Definitions (Post File)
11 PLOT-2D Plot Information	For NPTYPE = 6 (Contours of Continuum Element Data)	11.7.5 NCTS NCTS Variable to be plotted. Select from Table PL-1
		11.7.6 DELTA, IRES, IRGP, IENL, R_x, R_y DELTA = -DELTA Line contour, absolute value of DELTA is desired contour interval = 0 Color-filled contour = 2 Smoothed color-filled contour IRES = 0 Draft copy = 1 Fine copy IRGP = 0 Values at ref. grid points are not added = 1 Values at ref. grid points are added IENL = 0 Standard view = 2 Laplacian & spline interpolation scheme = 3 Davis distance to a power interpolation For IENL= 2 R_x Weight factor applied to spline function If R_x =0.0, only Laplacian interpolation is used R_y is not used For IENL= 3 R_y Power applied to 1/(distance **power) interpolation scheme. Recommended starting value is 4.0. R_x is not used Reference [Davis, J.c., 1986, Statistics and Data Analysis in Geology, page 356]

Card Group			Input Data and Definitions (Post File)
11 PLOT-2D Plot Information	11.7 For NPTYPE = 6 (Contours of Continuum Element Data)		11.7.7 NGROUP NGROUP = 0 Plot at all elements > 0 Plot at specified groups (Max=1000)
			11.7.8 If NGROUP = 0, Skip this Card NGROUP Cards: NSS, NEE, NIC, NNN - - - - - - - - Refer to Card Group 11.2.6
			11.7.9 NRL NRL Number of nodes to be connected by a Solid Line (Max=5000)
			11.7.10 If NRL = 0, Skip this Card $NODE_1$, $NODE_2$, ..., $NODE_{NRL}$ NODE Reference node numbers If $NODE_i$ has negative sign, a New Line is drawn

Card Group		Input Data and Definitions (Post File)
11 PLOT-2D Plot Information	11.8 For NPTYPE = 7 (Stress State in p-q Space and Octahedral Plane)	11.8.1 TITLE TITLE Any title of up to 70 characters
		11.8.2 LABELC LABELC Label for stress unit
		11.8.3 NLTIME $TIME_1$, $TIME_2$, ..., $TIME_{NLTIME}$ NLTIME Number of specified times (Max=10) TIME Specified time
		11.8.4 NUMNEL NEL_1, NEL_2, ..., NEL_{NUMNEL} NUMNEL Number of specified elements (Max=10) NEL Element number

Table PL-1 Continuum Contour Plot

NCTS	Legend	Description	
		Continuum Element (See Fig. PL-1)	
2	STRESS-XX	Normal XX stress	(σ_x')
3	STRESS-YY	Normal YY stress	(σ_y')
4	STRESS-ZZ	Normal ZZ stress	(σ_z')
5	STRESS-XY	Shear XY stress	(τ_{xy})
6	STRESS-YZ	Shear YZ stress	(τ_{yz})
7	STRESS-XZ	Shear XZ stress	(τ_{xz})
8	PRESSURE	Mean pressure	(P')
9	FLUID-PRES	Fluid pressure	(π)
10	TSTRESS-XX	Normal XX total stress	($\sigma_x = \sigma_x' + \pi$)
11	TSTRESS-YY	Normal YY total stress	($\sigma_y = \sigma_y' + \pi$)
12	TSTRESS-ZZ	Normal ZZ total stress	($\sigma_z = \sigma_z' + \pi$)
13	TPRESSURE	Total mean pressure	($P = P' + \pi$)
14	D.STRES	Deviatoric stress	($Q = (3/\sqrt{2})\ \tau_{oct}$)
15	STRAIN-XX	Normal XX strain	(ε_x)
16	STRAIN-YY	Normal YY strain	(ε_y)
17	STRAIN-ZZ	Normal ZZ strain	(ε_z)
18	STRAIN-XY	Shear XY strain	(γ_{xy})
19	STRAIN-YZ	Shear YZ strain	(γ_{yz})
20	STRAIN-XZ	Shear XZ strain	(γ_{xz})
21	VOL-STRAIN	Volumetric strain	(ε_v)
22	GAMMA-OCT	Octahedral shear strain	(γ_{oct})
23	TAU-OCT	Octahedral shear stress	(τ_{oct})
24	FS	Safety factor	(Fig. PL-2)
25	YIELD-FLAG	Yield flag	(Fig. PL-3)
26	STRESS - 1	Major principal stress	(σ_1')
27	STRESS - 2	Inter. principal stress	(σ_2')
28	STRESS - 3	Minor principal stress	(σ_3')

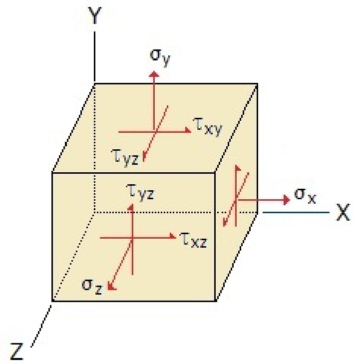


Figure PL-1 Sign Conventions for Continuum Stress

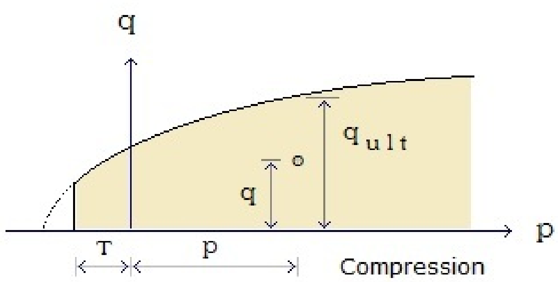


Figure PL-2 Definition of Safety Factor

Factor of Safety (FS) is defined as:

For elastic material FS = 10

For elasto-plastic material FS = q_{ult} / q

FS is limited to $1 \leq$ F.S. ≤ 10

For p $\leq$ -T FS = 1

$P = (\sigma_x + \sigma_y + \sigma_z) / 3$

$q = (3 / \sqrt{2})\, \tau_{oct}$

Yield Flag for Beam and Continuum Elements

Yield Flag	Stress Status
0	Stress point is in elastic
1	Stress point is in plastic
2	Stress Point develops crack

Stress Status for Truss Element

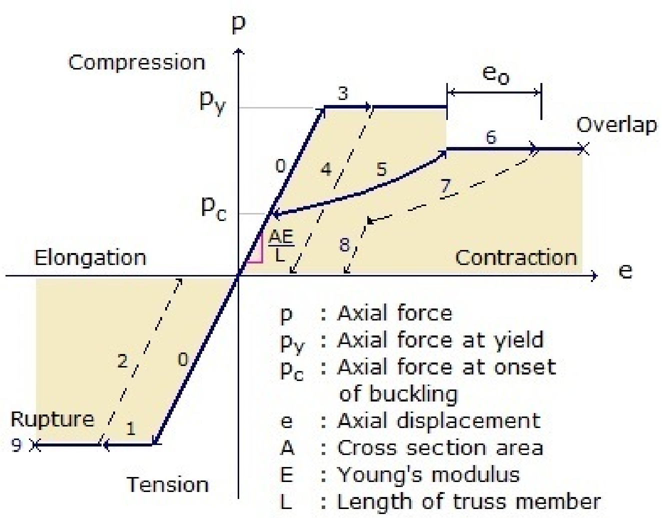


Figure PL-3 Description of Stress Status

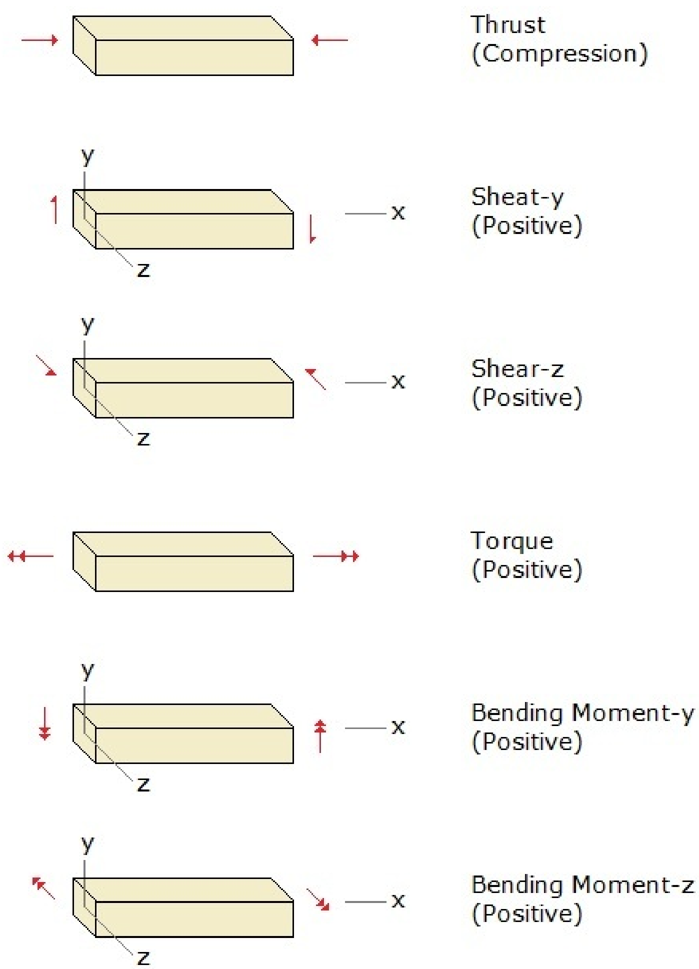


Figure PL-4 Sign Conventions for Beam

PLOT-XY

Post-Processor

Card Group	Input Data and Definitions (Post File)
12 PLOT-XY Information	12.1 IPTYPE IPTYPE 0 End of plotting output Standard Time history 1 Stress/Strain/Time 2 Displacement/Velocity/Accel./Time Standard Snapshot 3 Stress/Strain vs. Distance 4 Displacement/Velocity/Accel. vs. Distance Simplified Time history 5 Stresses/Strains for a Given Element 6 Stress/Strain Pair for Different Elements 7 Displacements/Velocities/Accel. for a Given Node 8 Displacement/Velocity/Accel. Pair for Different Nodes Simplified Snapshot 9 Stresses/Strains for a Given Time 10 Stress/Strain for Different Times 11 Displacements/Velocities/Accel. for a Given Time 12 Displacement/Velocity/Accel. for Different Times Note: Simplified plots (IPTYPE 5 to 12) should be specified after standard plots. You can edit simplified plots using PlotXY Generator in SMAP Run Menu.

Card Group		Input Data and Definitions (Post File)
12 PLOT-XY Information	12.2 For IPTYPE = 1 (Stress / Strain / Time History)	12.2.1 IPLOT IPLOT = 0 For each specified element, Number of different pair of variables = 1 For each specified pair of variables, Number of different element data
		12.2.2 NOEL NOEL Number of elements (Max 10)
		12.2.3 LIST (I) I = 1, NOEL LIST (I) List element numbers
		12.2.4 NDPQ NDPQ Number of different pair of variables

Card Group		Input Data and Definitions
12 PLOT-XY Information	12.2 For IPTYPE = 1 (Stress / Strain / Time History)	12.2.5 NDPQ Cards: K_{x1}, K_{y1} K_{x2}, K_{y2} - - - - K_x,K_y Select from Table PL-1 Note: <u>Shell rebar axial stresses & total bending moments</u> K values in Table PL-1 are to select average values. To select values at corner i, assign $K = 10\ K_{Table} + i$ For top rebar x-dir. axial stress at element corner I_3 assign K = 10 (95) + 3 = 953
		12.2.6 TMFAC, STFAC, SNFAC Multiplication factor TMFAC Time STFAC Stress SNFAC Strain
		12.2.7 IPLOT = 0: For each element IPLOT = 1: For each pair of variables TITLE (50 characters) X - LABEL (50 characters) Y - LABEL (50 characters)

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Post File)</th></tr>
<tr><td rowspan="3">12
PLOT-XY Information</td><td rowspan="3">12.3
For IPTYPE = 2 (Displacement / Velocity / Acceleration / Time History)</td><td>12.3.1
IPLOT

IPLOT = 0 For each specified node,
Number of different pair of variables

= 1 For each specified pair of variables,
Number of different node data</td></tr>
<tr><td>12.3.2
NODE

NODE Number of nodes (Max 10)</td></tr>
<tr><td>12.3.3
LIST (I), I = 1, NODE

LIST (I) List node numbers</td></tr>
</table>

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Post File)</th></tr>
<tr><td rowspan="4">12

PLOT-XY Information</td><td rowspan="4">12.3

For IPTYPE = 2 (Displacement / Velocity / Acceleration / Time History)</td><td>12.3.4

NDPQ

NDPQ Number of different pair of variables</td></tr>
<tr><td>12.3.5

NDPQ Cards:
K_{x1}, K_{y1}
K_{x2}, K_{y2}
- -
- -

K_x, K_y Select from Table PL-2</td></tr>
<tr><td>12.3.6

TMFAC, SND, SNV, SNA, NC, ANGLE

Multiplication factor
TMFAC Time
SND Displacement
SNV Velocity
SNA Acceleration

NC = 0 No transfer
= 1 Transfer from X-Y to polar coordinate
= 2 Transfer from polar to X-Y coordinate

ANGLE Rotation angle (Degree)</td></tr>
<tr><td>12.3.7

IPLOT = 0: For each node
IPLOT = 1: For each pair of variables

TITLE (50 characters)
X-LABEL (50 characters)
Y-LABEL (50 characters)</td></tr>
</table>

Card Group		Input Data and Definitions (Post File)
12 PLOT-XY Information	12.4 For IPTYPE = 3 (Stress / Strain vs. Distance Snapshot)	12.4.1 IPLOT IPLOT = 0 For each specified time, Number of different variables = 1 For each specified variable, Number of different time data
		12.4.2 NOTM NOTM Number of times (Max 10)
		12.4.3 TLIST (I), I = 1, NOTM TLIST (I) List times in sequential order
		12.4.4 NDPQ NDPQ Number of different variables
		12.4.5 NDPQ Cards: K_{y1}, K_{y2}, -, - K_y Select from Table PL-1 For K > 94, see Note in Card Group 12.2.5

Card Group			Input Data and Definitions (Post File)
12 PLOT-XY Information	12.4 For IPTYPE = 3 (Stress / Strain vs. Distance Snapshot)		12.4.6 ISCALD, ILTNUM, XSTART ISCALD = 0 Unscaled distance = 1 Scaled distance ILTNUM = 0 Do not list element numbers = 1 List Element No vs Value in PlotXy.Lin XSTART Reference starting X-coordinate Note: If ISCALD = 1 and ILTNUM = 1, X-LABEL is used for distance unit
			12.4.7 Element Number Specification (Max 800 Elements) For arbitrary order > 1 NRL N_1, N_2, N_{NRL} For sequential order > 2 NSTAR, NINCR, NPONT For end of generation > 0 NRL Number of elements N_1,N_2,..,N_{NRL} Element numbers NSTAR Starting element numbers NINCR Element number increment NPONT Number of element

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Post File)</th></tr>
<tr><td rowspan="2">12

PLOT-XY Information</td><td rowspan="2">12.4

For IPTYPE = 3 (Stress / Strain vs. Distance Snapshot)</td><td>12.4.8

STFAC, SNFAC, SDFAC

Multiplication factor
STFAC Stress
SNFAC Strain
SDFAC Distance</td></tr>
<tr><td>12.4.9

IPLOT = 0: For each specified time
IPLOT = 1: For each variable

TITLE (50 characters)
X-LABEL (50 characters)
Y-LABEL (50 characters)</td></tr>
</table>

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Post File)</th></tr>
<tr><td rowspan="5">12

PLOT-XY Information</td><td rowspan="5">12.5

For IPTYPE = 4 (Displacement / Velocity / Acceleration vs. Distance Snapshot)</td><td>12.5.1

IPLOT

IPLOT = 0 For each specified time,
Number of different variables

= 1 For each specified variable,
Number of different time data</td></tr>
<tr><td>12.5.2

NOTM

NOTM Number of times (Max 10)</td></tr>
<tr><td>12.5.3

TLIST (I), I = 1, NOTM

TLIST (I) List times in sequential order</td></tr>
<tr><td>12.5.4

NDPQ

NDPQ Number of different variables</td></tr>
<tr><td>12.5.5

NDPQ Cards: K_{y1}, K_{y2}, -, -

K_y Select from Table PL-2</td></tr>
</table>

Card Group		Input Data and Definitions (Post File)
12 PLOT-XY Information	12.5 For IPTYPE = 4 (Displacement / Velocity / Acceleration vs. Distance Snapshot)	12.5.6 ISCALD, ILTNUM, XSTART ISCALD = 0 Unscaled distance = 1 Scaled distance ILTNUM = 0 Do not list node numbers = 1 List Node No vs Value in PlotXy.Lin XSTART Reference starting X-coordinate Note: If ISCALD = 1 and ILTNUM = 1, X-LABEL is used for distance unit

Card Group			Input Data and Definitions (Post File)
12 PLOT-XY Information	12.5 For IPTYPE = 4 (Displacement / Velocity / Acceleration vs. Distance Snapshot)	12.5.7	Node Number Specification (Max 800 nodes) For Arbitrary Order > 1 NRL N_1, N_2, .., N_{NRL} For Sequential Order > 2 NSTAR, NINCR, NPONT For End Generation > 0 NRL: Number of nodes N_1,N_2,..,N_{NRL}: Node numbers NSTAR: Starting node numbers NINCR: Node number increment NPONT: Number of nodes
		12.5.8	SND, SNV, SNA, NC, ANGLE, SDFAC Multiplication factor SND: Displacement SNV: Velocity SNA: Acceleration NC = 0: No transfer = 1: Transfer from X-Y to polar coordinate = 2: Transfer from polar to X-Y coordinate ANGLE: Rotation angle (Degree) SDFAC: Multiplication factor for distance
		12.5.9	IPLOT = 0: For each specified time IPLOT = 1: For each variable TITLE (50 characters) X-LABEL (50 characters) Y-LABEL (50 characters)

Card Group		Input Data and Definitions
12 PLOT-XY Information	For IPTYPE = 5 (Time History of Stresses/Strains for a Given Element)	12.6.1 NEL NEL Element number
		12.6.2 NDQ NDQ Number of different quantities
		12.6.3 NDQ Cards: K_{y1}, K_{y2}, -, - K_y Select from Table PL-1
		12.6.4 TMFAC, STFAC, SNFAC Multiplication factor TMFAC Time STFAC Stress SNFAC Strain
		12.6.5 TITLE (50 characters) X - LABEL (50 characters) Y - LABEL (50 characters)

Card Group		Input Data and Definitions
12 PLOT-XY Information	For IPTYPE = 6 (Time History of Stress/Strain Pair for Different Elements)	12.7.1 NOEL NOEL Number of elements (Max 10)
		12.7.2 LIST (I) I = 1, NOEL LIST (I) List element numbers
		12.7.3 K_x, K_y K_x, K_y Select from Table PL-1
		12.7.4 TMFAC, STFAC, SNFAC Multiplication factor TMFAC Time STFAC Stress SNFAC Strain
		12.7.5 TITLE (50 characters) X - LABEL (50 characters) Y - LABEL (50 characters)

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions</th></tr>
<tr><td rowspan="5">12

PLOT-XY Information</td><td rowspan="5">For IPTYPE = 7 (Time History of Displacements/Vel./Accel. for a Given Node)</td><td>12.8.1

NOD
NOD Node number</td></tr>
<tr><td>12.8.2

NDQ
NDQ Number of different quantities</td></tr>
<tr><td>12.8.3

NDQ Cards: K_{y1}, K_{y2}, -, -

K_y Select from Table PL-2</td></tr>
<tr><td>12.8.4

TMFAC, SND, SNV, SNA

Multiplication factor
TMFAC Time
SND Displacement
SNV Velocity
SNA Acceleration</td></tr>
<tr><td>12.8.5

TITLE (50 characters)
X - LABEL (50 characters)
Y - LABEL (50 characters)</td></tr>
</table>

Card Group		Input Data and Definitions
12 PLOT-XY Information	For IPTYPE = 8 (Time History of Displ./Vel./Accel. Pair for Different Nodes)	12.9.1 NODE NODE — Number of nodes (Max 10)
		12.9.2 LIST (I) I = 1, NODE LIST (I) — List node numbers
		12.9.3 K_x, K_y K_x, K_y — Select from Table PL-2
		12.9.4 TMFAC, SND, SNV, SNA Multiplication factor TMFAC — Time SND — Displacement SNV — Velocity SNA — Acceleration
		12.9.5 TITLE (50 characters) X - LABEL (50 characters) Y - LABEL (50 characters)

Card Group		Input Data and Definitions
12 PLOT-XY Information	For IPTYPE = 9 (Snap Shot of Stresses/Strains for a Given Time)	12.10.1 TIME TIME Specified time
		12.10.2 NDQ NDQ Number of different quantities
		12.10.3 NDQ Cards: K_{y1}, K_{y2}, - K_y Select from Table PL-1
		12.10.4 XSTART XSTART Reference starting X-coordinate
		12.10.5 Element Number Specification (Max 800 Elements) NRL N_1, N_2, N_{NRL} NRL Number of elements N_1, N_2, .., N_{NRL} Element numbers N_i, $-N_{i+1}$, N_{i+2} From N_i to N_{i+1} with increment N_{i+2}
		12.10.6 STFAC, SNFAC, SDFAC Multiplication factor STFAC Stress SNFAC Strain SDFAC Distance
		12.10.7 TITLE (50 characters) X - LABEL (50 characters) Y - LABEL (50 characters)

Card Group		Input Data and Definitions
12 PLOT-XY Information	For IPTYPE = 10 (Snap Shot of a Stress/Strain for Different Times)	12.11.1 NOTM NOTM Number of times (Max 10)
		12.11.2 TLIST (I), I = 1, NOTM TLIST (I) List times in sequential order
		12.11.3 K_y K_y Select from Table PL-1
		12.11.4 XSTART XSTART Reference starting X-coordinate
		12.11.5 Element Number Specification (Max 800 Elements) NRL N_1, N_2, N_{NRL} NRL Number of elements N_1, N_2, .., N_{NRL} Element numbers N_i, $-N_{i+1}$, N_{i+2} From N_i to N_{i+1} with increment N_{i+2}
		12.11.6 STFAC, SNFAC, SDFAC Multiplication factor STFAC Stress SNFAC Strain SDFAC Distance
		12.11.7 TITLE (50 characters) X - LABEL (50 characters) Y - LABEL (50 characters)

Card Group		Input Data and Definitions
12 PLOT-XY Information	For IPTYPE = 11 (Snap Shot of Displacements/Vel./Accel for a Given Time)	12.12.1 TIME TIME Specified time
		12.12.2 NDQ NDQ Number of different quantities
		12.12.3 NDQ Cards: K_{y1}, K_{y2}, - K_y Select from Table PL-2
		12.12.4 XSTART XSTART Reference starting X-coordinate
		12.12.5 Node Number Specification (Max 800 Nodes) NRL N_1, N_2, N_{NRL} NRL Number of nodes $N_1, N_2, .., N_{NRL}$ Node numbers $N_i, -N_{i+1}, N_{i+2}$ From N_i to N_{i+1} with increment N_{i+2}
		12.12.6 SND, SNV, SNA, SDFAC Multiplication factor SND Displacement SNV Velocity SNA Acceleration SDFAC Distance
		12.12.7 TITLE (50 characters) X - LABEL (50 characters) Y - LABEL (50 characters)

Card Group		Input Data and Definitions
12 PLOT-XY Information	For IPTYPE = 12 (Snap Shot of a Displ./Vel./Accel. for Different Times)	12.13.1 NOTM NOTM Number of times (Max 10)
		12.13.2 TLIST (I), I = 1, NOTM TLIST (I) List times in sequential order
		12.13.3 K_y K_y Select from Table PL-2
		12.13.4 XSTART XSTART Reference starting X-coordinate
		12.13.5 Node Number Specification (Max 800 Nodes) NRL N_1, N_2, N_{NRL} NRL Number of nodes N_1, N_2, .., N_{NRL} Node numbers N_i, $-N_{i+1}$, N_{i+2} From N_i to N_{i+1} with increment N_{i+2}
		12.13.6 SND, SNV, SNA, SDFAC Multiplication factor SND Displacement SNV Velocity SNA Acceleration SDFAC Distance
		12.13.7 TITLE (50 characters) X - LABEL (50 characters) Y - LABEL (50 characters)

Table PL-1 (IPTYPE = 1, 3, 5, 6, 9, 10)

K_x, K_y	Legend	Description	
1	TIME	Time	(t)
		Continuum Element (See Fig. PL-1)	
2	STRESS-XX	Normal XX stress	(σ_x')
3	STRESS-YY	Normal YY stress	(σ_y')
4	STRESS-ZZ	Normal ZZ stress	(σ_z')
5	STRESS-XY	Shear XY stress	(τ_{xy})
6	STRESS-YZ	Shear YZ stress	(τ_{yz})
7	STRESS-XZ	Shear XZ stress	(τ_{xz})
8	PRESSURE	Mean pressure	(P')
9	FLUID-PRES	Fluid pressure	(π)
10	TSTRESS-XX	Normal XX total stress	($\sigma_x = \sigma_x' + \pi$)
11	TSTRESS-YY	Normal YY total stress	($\sigma_y = \sigma_y' + \pi$)
12	TSTRESS-ZZ	Normal ZZ total stress	($\sigma_z = \sigma_z' + \pi$)
13	TPRESSURE	Total mean pressure	($P = P' + \pi$)
14	D.STRES	Deviatoric stress	($Q = (3/\sqrt{2})\ \tau_{oct}$)
15	STRAIN-XX	Normal XX strain	(ε_x)
16	STRAIN-YY	Normal YY strain	(ε_y)
17	STRAIN-ZZ	Normal ZZ strain	(ε_z)
18	STRAIN-XY	Shear XY strain	(γ_{xy})
19	STRAIN-YZ	Shear YZ strain	(γ_{yz})
20	STRAIN-XZ	Shear XZ strain	(γ_{xz})
21	VOL-STRAIN	Volumetric strain	(ε_v)
22	GAMMA-OCT	Octahedral shear strain	(γ_{oct})
23	TAU-OCT	Octahedral shear stress	(τ_{oct})
24	FS	Safety factor	(Fig. PL-2)
25	YIELD-FLAG	Yield flag	(Fig. PL-3)
26	STRESS - 1	Major principal stress	(σ_1')
27	STRESS - 2	Inter. principal stress	(σ_2')
28	STRESS - 3	Minor principal stress	(σ_3')

Table PL-1 continued

K_x, K_y	Legend	Description	
		Beam Element (See Fig. PL-4)	
35	THRUST	Thrust	(F_x)
36	SHEAR-Y	Shear in y direction	(F_y)
37	SHEAR-Z	Shear in z direction	(F_z)
38	TORQUE	Torque	(T)
39	MOMENT-Y	Moment about y axis	(M_y)
40	MOMENT-Z	Moment about z axis	(M_z)
41	STRAIN-FT	Top fiber strain	(ε_{ft})
42	STRESS-FT	Top fiber stress	(σ_{ft})
43	STRAIN-RT	Top reinf. bar strain	(ε_{rt})
44	STRESS-RT	Top reinf. bar stress	(σ_{rt})
45	STRAIN-RB	Bot. reinf. bar strain	(ε_{rb})
46	STRESS-RB	Bot. reinf. bar stress	(σ_{rb})
47	STRAIN-FB	Bot. fiber strain	(ε_{fb})
48	STRESS-FB	Bot. fiber stress	(σ_{fb})
49	STRAIN-FL	Left fiber strain	(ε_{fl})
50	STRESS-FL	Left fiber stress	(σ_{fl})
51	STRAIN-RL	Left reinf. bar strain	(ε_{rl})
52	STRESS-RL	Left reinf. bar stress	(σ_{rl})
53	STRAIN-RR	Right reinf. bar strain	(ε_{rr})
54	STRESS-RR	Right reinf. bar stress	(σ_{rr})
55	STRAIN-FR	Right fiber strain	(ε_{fr})
56	STRESS-FR	Right fiber stress	(σ_{fr})
		Truss Element	
61	FORCE-XX	Axial force	(F_x)
62	STRESS-XX	Axial stress	(σ_x)
63	STRAIN-XX	Axial strain	(ε_x)

Table PL-1 continued

K_x, K_y	Legend	Description	
		Shell element section forces and stresses	
71	MOMENT-XX	Bending moment	(M_{xx})
72	MOMENT-YY	Bending moment	(M_{yy})
73	MOMENT-XY	Twisting moment	(M_{xy})
74	M-MAX	Max bending moment	(M_{max})
75	M-MIN	Min bending moment	(M_{min})
76	MXY-MAX	Max twisting moment	($M_{xy\ max}$)
		Mid-surface stress	
77	SMID-XX	Normal xx stress	($\sigma_{xx\ mid}$)
78	SMID-YY	Normal yy stress	($\sigma_{yy\ mid}$)
79	SMID-XY	Shear xy stress	($\sigma_{xy\ mid}$)
80	SM-MAX	Max normal xx stress	($\sigma_{max\ mid}$)
81	SM-MIN	Min normal yy stress	($\sigma_{min\ mid}$)
82	SMXY-MAX	Max shear xy stress	($\sigma_{xy\ max\ mid}$)
		Top-surface stress	
83	STOP-XX	Normal xx stress	($\sigma_{xx\ top}$)
84	STOP-YY	Normal yy stress	($\sigma_{yy\ top}$)
85	STOP-XY	Shear xy stress	($\sigma_{xy\ top}$)
86	ST-MAX	Max normal xx stress	($\sigma_{max\ top}$)
87	ST-MIN	Min normal yy stress	($\sigma_{min\ top}$)
88	STXY-MAX	Max shear xy stress	($\sigma_{xy\ max\ top}$)
		Bottom-surface stress	
89	SBOT-XX	Normal xx stress	($\sigma_{xx\ bot}$)
90	SBOT-YY	Normal yy stress	($\sigma_{yy\ bot}$)
91	SBOT-XY	Shear xy stress	($\sigma_{xy\ bot}$)
92	SB-MAX	Max normal xx stress	($\sigma_{max\ bot}$)
93	SB-MIN	Min normal yy stress	($\sigma_{min\ bot}$)
94	SBXY-MAX	Max shear xy stress	($\sigma_{xy\ max\ bot}$)
		Rebar axial stress and total moment	
95	ASTRES-XT	Top rebar x direction axial stress	($\sigma_{xx\ top}$)
96	ASTRES-YT	Top rebar y direction axial stress	($\sigma_{yy\ top}$)
97	ASTRES-XB	Bot. rebar x direction axial stress	($\sigma_{xx\ bot}$)
98	ASTRES-YB	Bot. rebar y direction axial stress	($\sigma_{yy\ bot}$)
99	TMOMENT-XX	Total bending moment about x axis	($M_{xx\ total}$)
100	TMOMENT-YY	Total bending moment about y axis	($M_{yy\ total}$)
		Note: Moments per unit width (See Fig. PL-5)	

Table PL-2 (IPTYPE = 2, 4, 7, 8, 11, 12)

K_x, K_y	Legend	Description	
1	TIME	Time	(t)
		Skeleton displacement	
2	X-DIS.	X-displacement	(u_x)
3	Y-DIS.	Y-displacement	(u_y)
4	Z-DIS.	Z-displacement	(u_z)
5	X-VEL.	X-velocity	(u_x)
6	Y-VEL.	Y-velocity	(u_y)
7	Z-VEL.	Z-velocity	(u_z)
8	X-ACC.	X-acceleration	(u_x)
9	Y-ACC.	Y-acceleration	(u_y)
10	Z-ACC.	Z-acceleration	(u_z)
		Relative fluid displacement	
11	R.FL.X-DIS	X-displacement	($w_x = n (U_x - u_x)$)
12	R.FL.Y-DIS	Y-displacement	(w_y)
13	R.FL.Z-DIS	Z-displacement	(w_z)
14	R.FL.X-VEL	X-velocity	(w_x)
15	R.FL.Y-VEL	Y-velocity	(w_y)
16	R.FL.Z-VEL	Z-velocity	(w_z)
17	R.FL.X-ACC	X-acceleration	(w_x)
18	R.FL.Y-ACC	Y-acceleration	(w_y)
19	R.FL Z-ACC	Z-acceleration	(w_z)

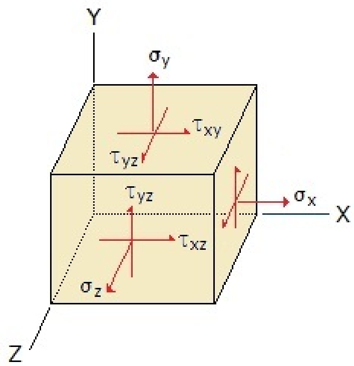


Figure PL-1 Sign Conventions for Continuum Stress

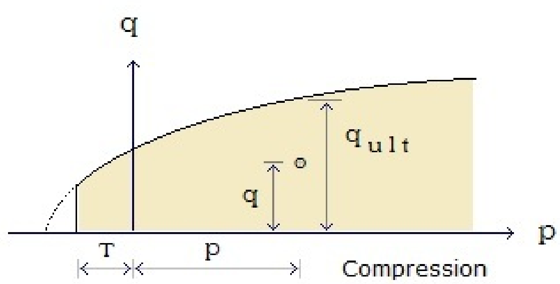


Figure PL-2 Definition of Safety Factor

<u>Factor of Safety (FS) is defined as:</u>

For elastic material FS = 10

For elasto-plastic material FS = q_{ult} / q
FS is limited to $1 \leq$ F.S. ≤ 10
For p $\leq$ -T FS = 1

$$P = (\sigma_x + \sigma_y + \sigma_z) / 3$$
$$q = (3 / \sqrt{2})\, \tau_{oct}$$

Yield Flag	Stress Status for Beam & Continuum Element
0	Stress point is in elastic
1	Stress point is in plastic
2	Stress Point develops crack

Stress Status for Truss Element

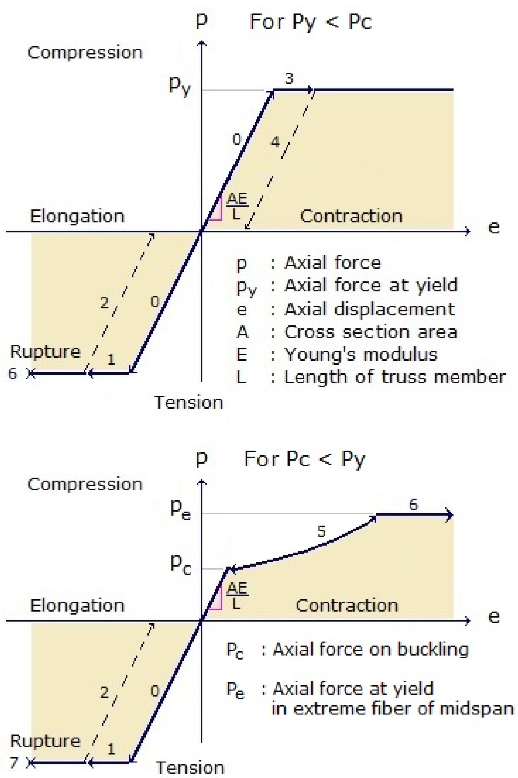


Figure PL-3 Description of Stress Status

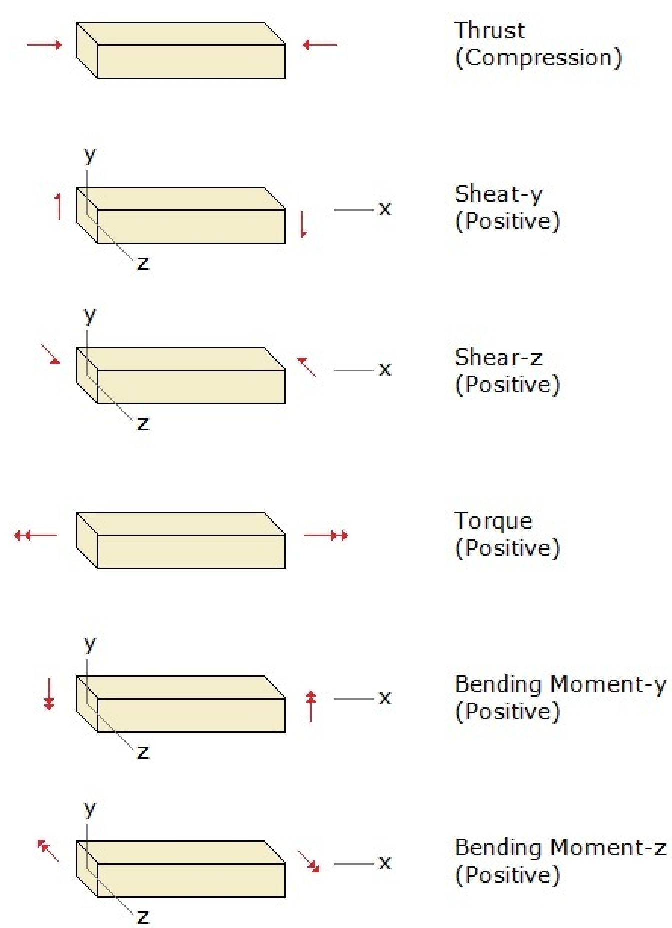
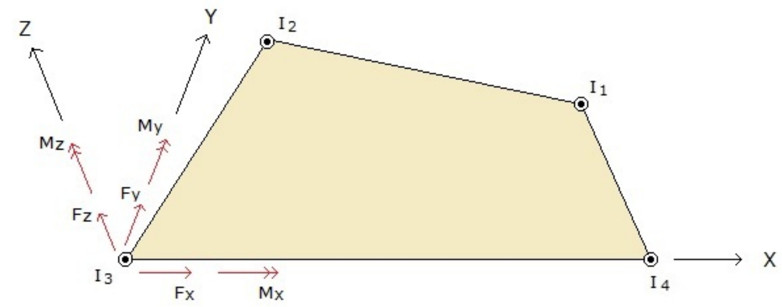
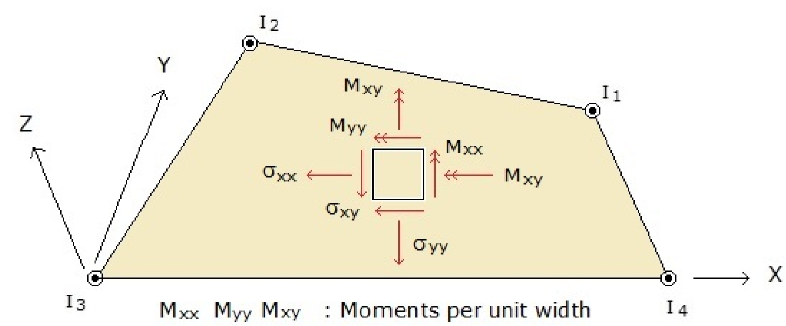


Figure PL-4 Sign Conventions for Beam



Shell Member End Forces at Element Nodes
[Output File SHELMEF.DAT]



Shell Stresses and Moments at Element Center
[Output File: SHELSM.DAT]

Figure PL-5 Sign Conventions for Shell

Group Mesh
User's Manual

5.1 Introduction

Group Mesh Generator is a two-dimensional CAD program specially designed to build group mesh which can be used to generate finite element mesh with the aid of program ADDRGN-2D.

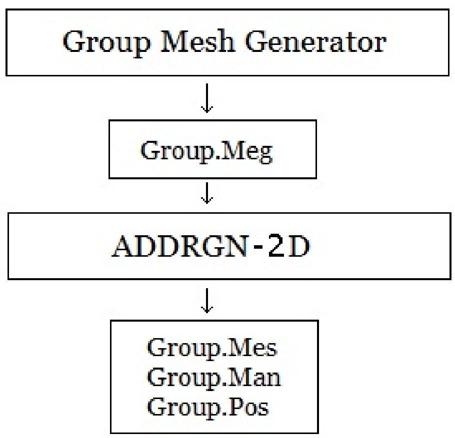


Figure 5.1 Flow diagram of group mesh generation

Group.Meg contains group mesh data that can be generated or modified by Group Mesh Generator. The file Group.Meg is used as input to the program ADDRGN-2D, thereby generating finite element mesh file Group.Mes along with the main file Group.Man for element activity and the post file Group.Pos for PLOT-2D plot.

Group Mesh Generator can be accessed through SMAP menu
Run or Plot as explained in Section 5.2.

ADDRGN-2D can be accessed from SMAP menu:
Run → Mesh Generator → AddRgn → Addrgn 2D.
This program can also be accessed indirectly by executing
F. E. Mesh Plot in Group dialog as explained in Section 5.3.8.

5.2 Group Mesh Generator

Group Mesh Generator can be accessed by selecting the following menu items in SMAP:

Run → Mesh Generator → Group Mesh or
Plot → Mesh → Group Mesh

When you build new group mesh, you can select either Built-in Base Mesh or Existing Finite Element Mesh. Built-in Base Mesh is explained in detail in Section 5.4.

Once you click OK button in Group Input dialog, PLOT-2D program is displayed along with group menu which is the main access to Group Mesh Generator.

When click Group menu in PLOT-2D, Group dialog is displayed.

5.3 Group

Group dialog in Figure 5.2 is the main dialog associated with group mesh generation or modification. Group dialog consists of following eight parts:

- Group Identity
- MTYPE and Material Parameter
- Line Options
- Coordinate Constraint
- Element Activity
- PLOT-2D Plot
- Translation
- Command Buttons

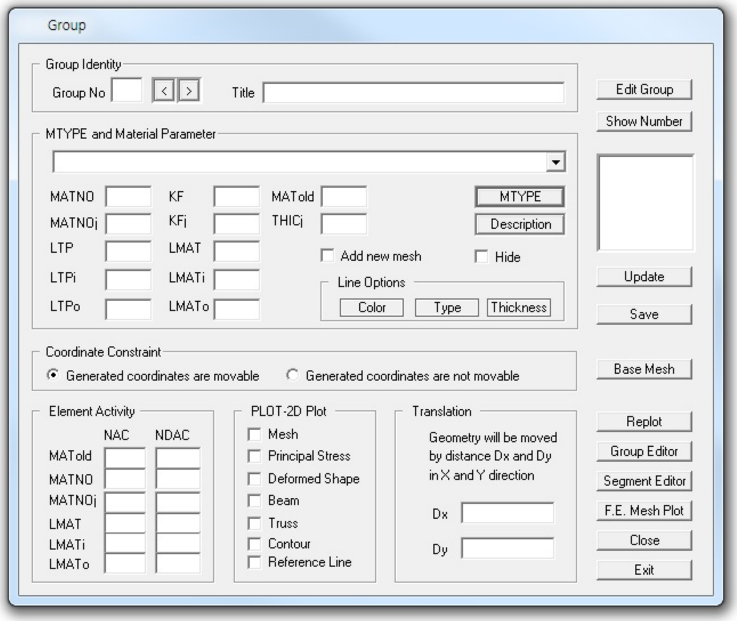


Figure 5.2 Group dialog.

5.3.1 Group Identity

Here, you type Group No and Title.

When you add a new group, first select an appropriate MTYPE and change all default parameters as you want. Then click Add Group button to build the geometry of new group.

When you type the existing Group No, all parameters of that group are shown on the screen. Click Edit Group button to modify the geometry of the group.

It should be noted that Add Group and Edit Group buttons share the same position in the Group dialog. And Add Group for new group and Edit Group for existing group will appear.

5.3.2 MTYPE and Material Parameter

MTYPE dialog with icons and MTYPE list box with brief explanations are shown in Figures 5.3 and 5.4, respectively.

You can select MTYPE from the list box or by clicking MTYPE button which opens MTYPE dialog with icons.
Selection of proper MTYPE is the most important to model the desired group. Once you select MTYPE, all input variables and options available for that MTYPE will be shown on the screen along with default values.

Figure 5.3
MTYPE dialog

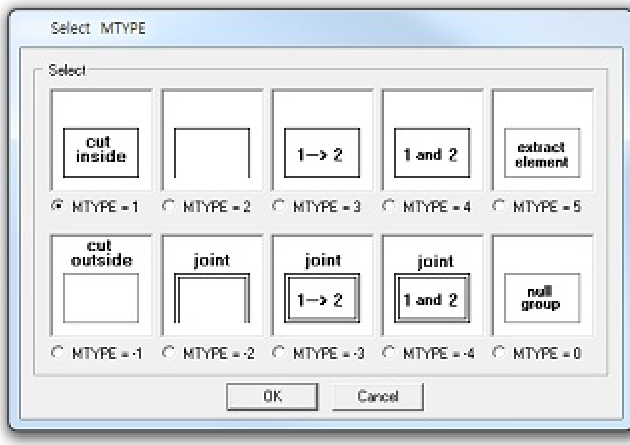
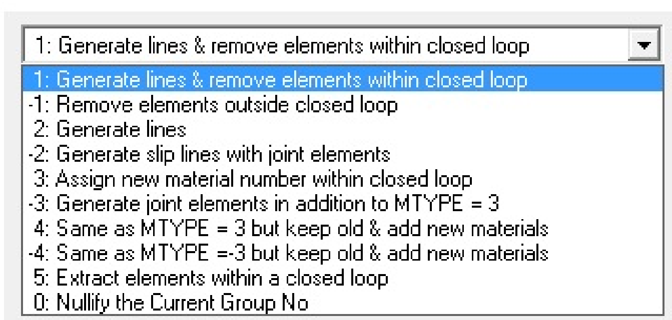


Figure 5.4
MTYPE list box



Click Description button to see description of material parameters and element activity as shown in Figure 5.5.

Add new mesh check box is available only for MTYPE = 3. When checked, new group is formed without interfering with the other groups.

Hide check box is to hide the current group geometry on the screen when checked.

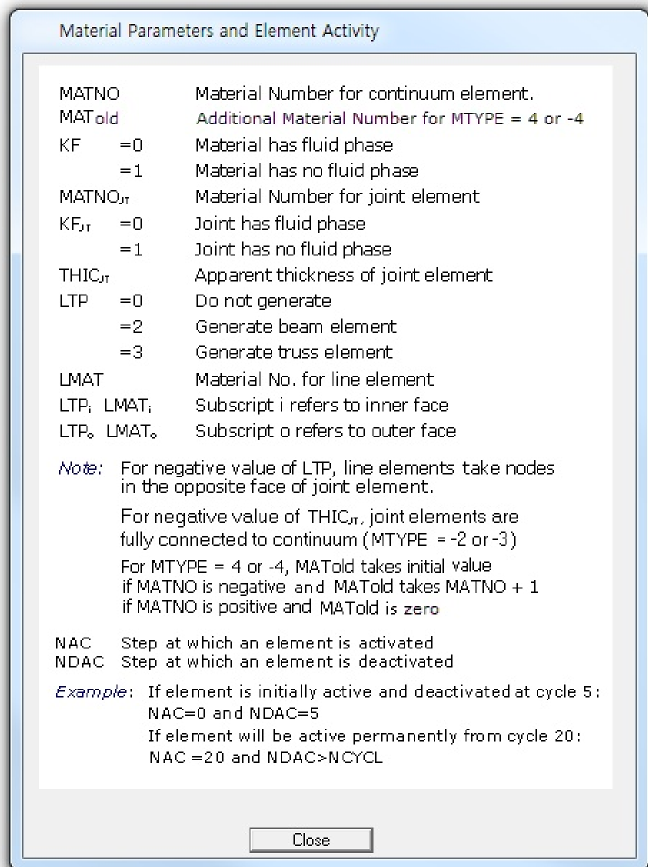


Figure 5.5 Material parameters & element activity (SMAP-2D)

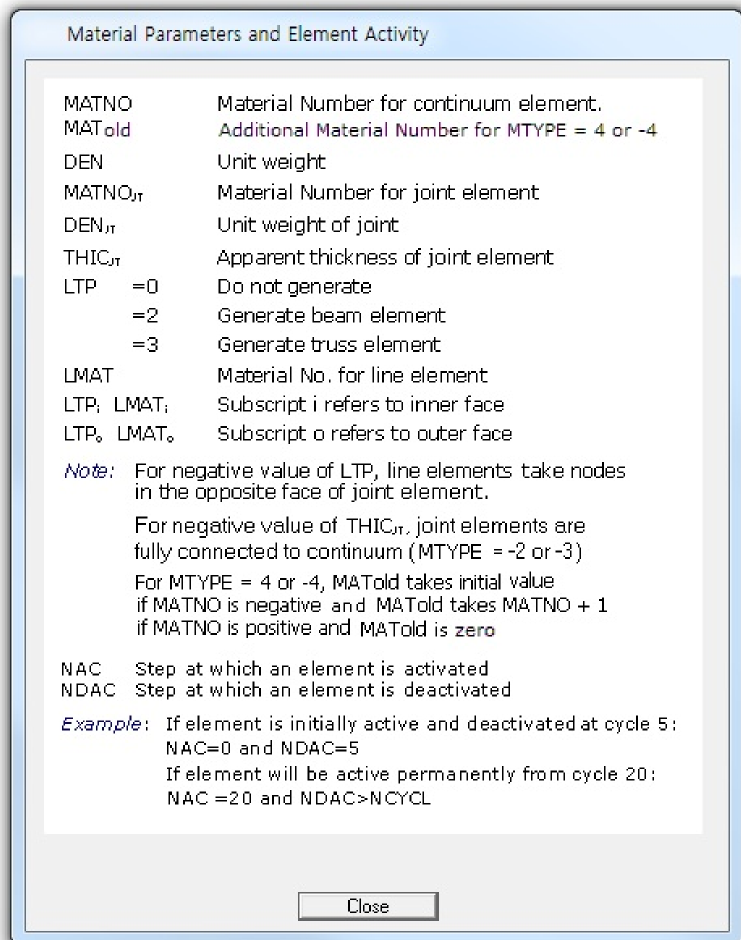


Figure 5.5 Material parameters & element activity (SMAP-S2)

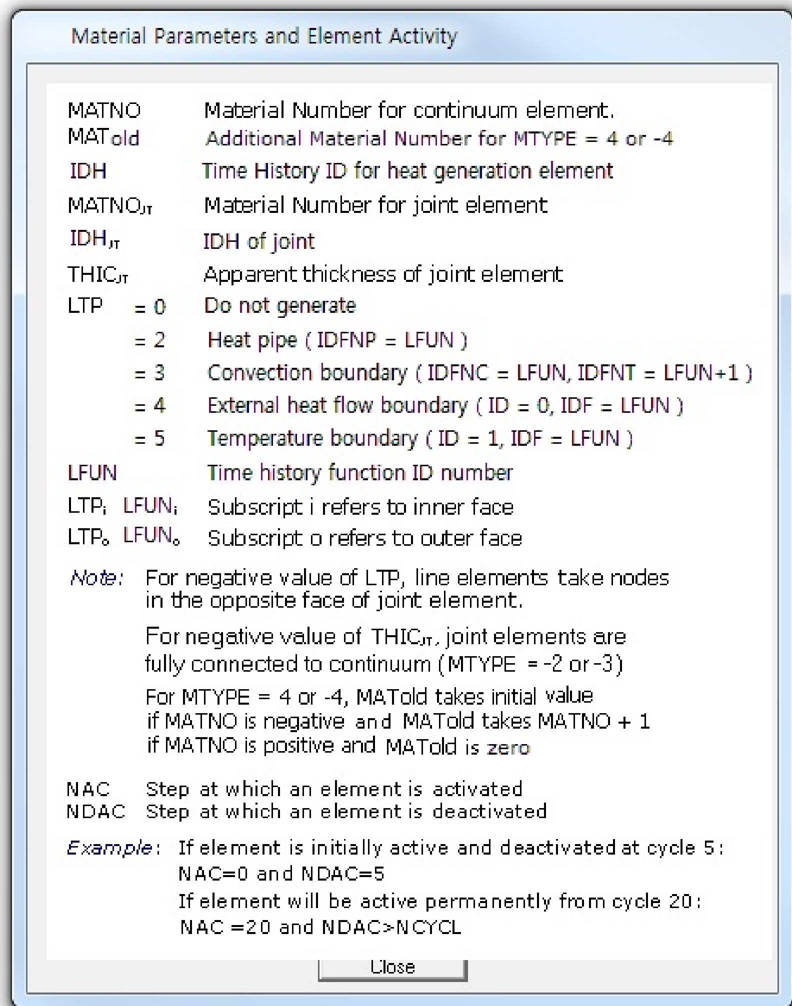


Figure 5.5 Material parameters & element activity (SMAP-T2)

5.3.3 Line Options

Line options are provided to distinguish the outline of the group from the other groups. Figure 5.6 shows available line color, line type and line thickness.

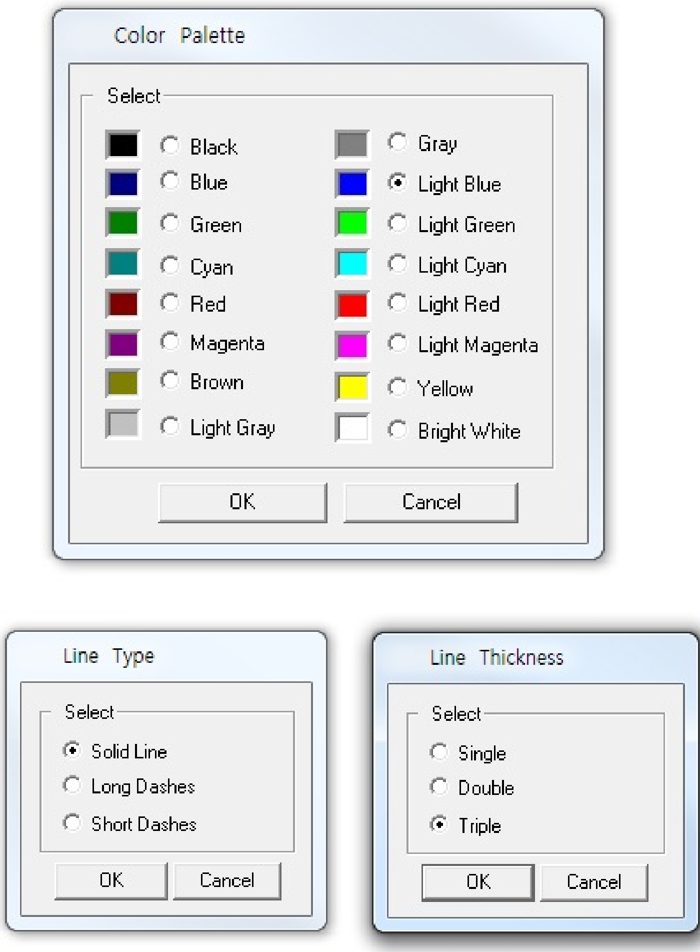


Figure 5.6 Line options.

5.3.4 Coordinate Constraint

Finite element meshes are generated when you click F. E. Mesh Plot button.

Normally, finite element nodal coordinates associated with the current group are adjusted to get the overall optimum meshes by selecting Generated coordinates are movable.

However, you can make such generated coordinates not movable by selecting Generated coordinates are not movable.

5.3.5 Element Activity

Element activity data is used in SMAP main program Card group 8. Elements in current group is to activate at step NAC and deactivate at step NDAC. Such activity data is generated and saved in Group.Man when executing group mesh, that is, by clicking F. E. Mesh Plot button.

Examples of element activity are shown at bottom of Fig. 5.5.

5.3.6 PLOT-2D Plot

PLOT-2D Plot data is used in SMAP post processing program PLOT-2D to plot computed results available for the current group. Such plot information is generated and saved in Group.Pos when executing group mesh.

It should be noted that SMAP post processing program PLOT-3D can automatically produce all such plots.

5.3.7 Translation

Translation is mainly used to move the geometry of the current group in x and y directions. Here D_x and D_y represent relative distances from the current position of the group to the new position.

Once you type in D_x and D_y, you need to click Update and then Replot buttons to confirm the translation of the current group.

5.3.8 Command Buttons

Command buttons are shown on the right side of Group dialog.

Add Group

This is used to build the geometry of the new group.
Line Segment dialog in Figure 5.14 will be displayed.

Edit Group

This is used to modify the geometry of the existing group.
Edit Segment dialog in Figure 5.7 will be displayed.

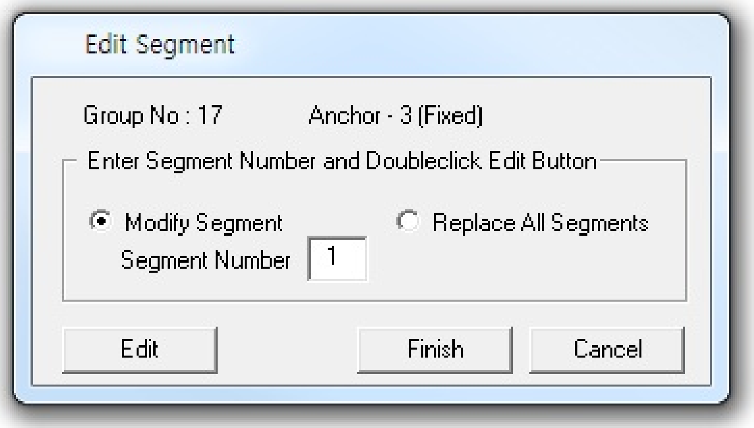


Figure 5.7 Edit segment dialog.

Show Number

This is used to show group and segment numbers.
Plot Group / Segment No dialog in Figure 5.8 will be displayed.

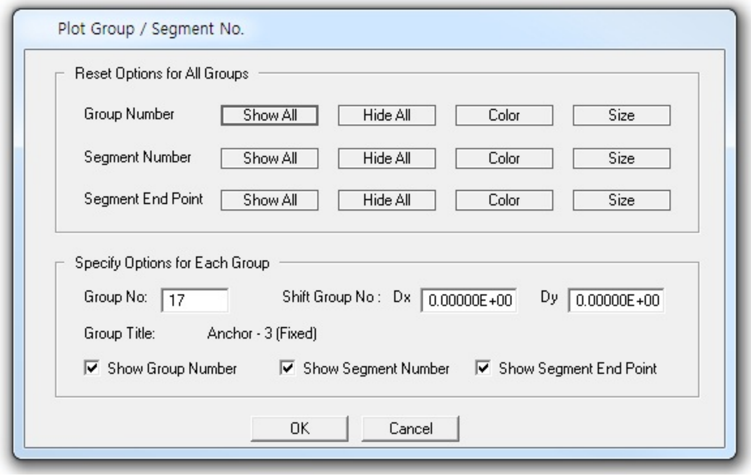


Figure 5.8 Plot Group / Segment No dialog.

Update

This is used to update the current group parameters shown on the screen. It should be noted that you need to click Update button before leaving the current group. Leaving the current group without clicking Update will not update all the changes you made on the current group.

Save

This is used to save all the works you have done . This includes updating the current group parameters shown in the Group dialog.

Base Mesh

This is used to edit Built-in Base Mesh which is explained in detail in Section 5.4. Base Mesh dialog in Figure 5.13 will be displayed.

Replot

This is used to show the geometry of groups you have updated so far.

Group Editor

This is used to delete, cut and paste, or copy and paste specified groups. Group Editor dialog in Figure 5.9 will be displayed.

Figure 5.9 Group editor dialog.

Segment Editor

This is used to add or modify the segments of the existing group based on text input. Segment Editor dialog in Figure 5.10 will be displayed.

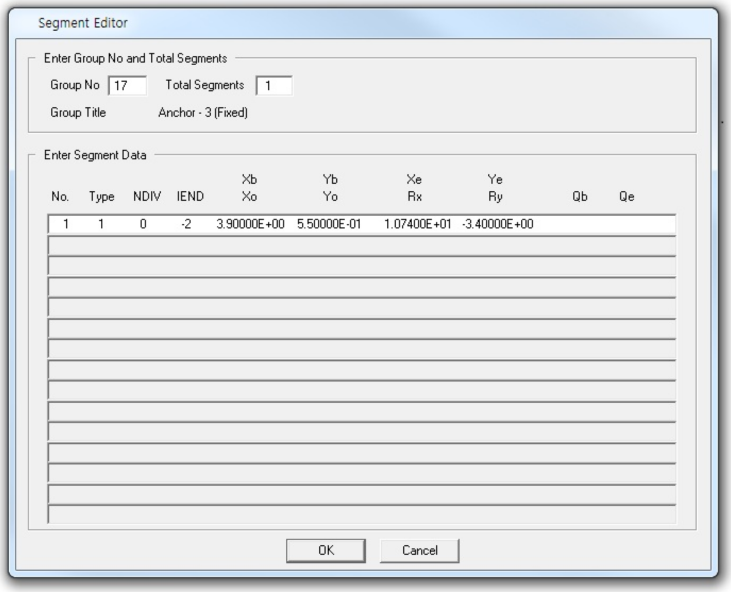


Figure 5.10 Segment editor dialog.

F. E. Mesh Plot

This is used to execute the group mesh and then plot the generated finite element mesh. It should be noted that you need to click Save button before executing F. E. Mesh Plot.

Once executed, new sub directory Plot_Mesh under working directory will be created along with following files:

Group.Mes Mesh file with finite element.
Group.Man Main file with element activity.
Group.Pos Post file with PLOT-2D plot data.

Close

This is used to close the Group dialog.

Exit

This is used to exit from the Group Mesh Generator.
Exit dialog in Figure 5.11 will be displayed.

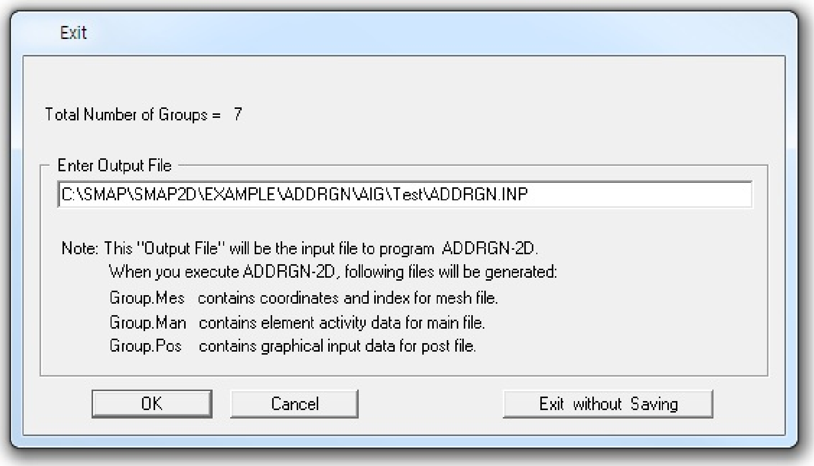


Figure 5.11 Exit dialog.

5.4 Base Mesh

Base Mesh is the finite element mesh where you build group meshes. You can select either Built-in Base Mesh or Existing Finite Element Mesh at the time when you first build new group mesh as discussed in Section 5.2.

5.4.1 Built-in Base Mesh

Figure 5.12 shows layout of Built-in Base Mesh which consists of rectangular blocks that will be filled with finite elements.

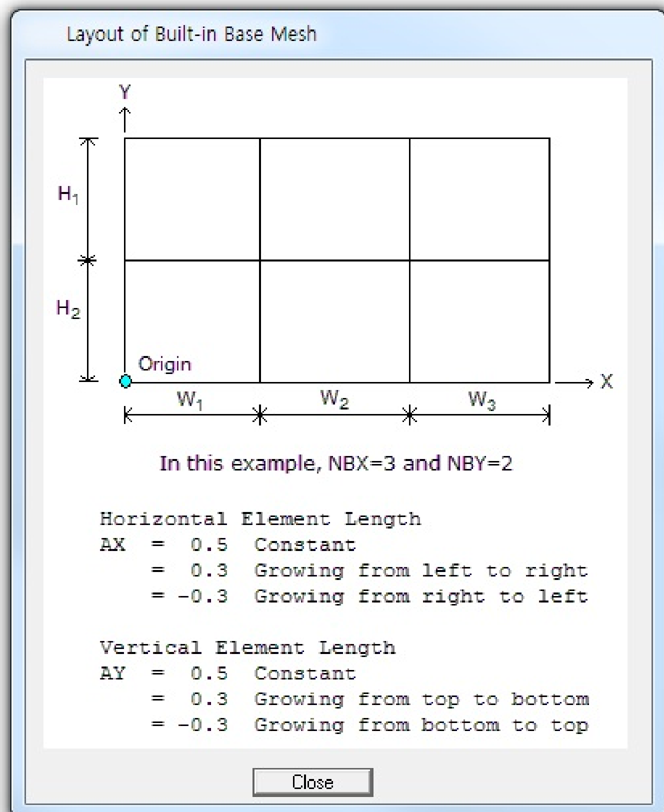


Figure 5.12 Layout of built-in base mesh.

Figure 5.13 shows Built-in Base Mesh dialog which is used to edit block dimensions, element sizes and boundary conditions.

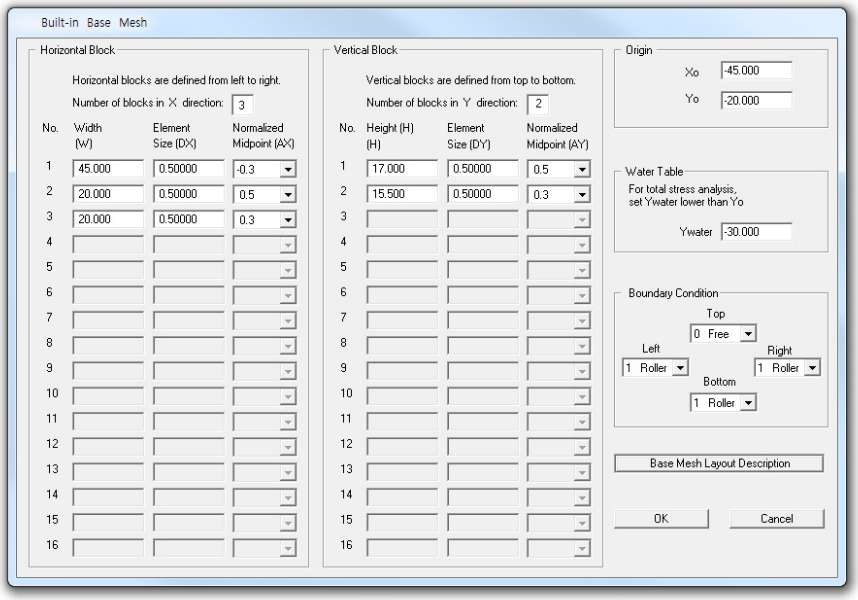


Figure 5.13 Built-in base mesh dialog.

5.4.2 Existing Finite Element Mesh

You can provide existing finite element mesh file to be used as base mesh. Group meshes will be built on this finite element mesh.

5.5 Segment

There are two types of segments, Line and Arc Segments which are used to build a group. Segment dialog will be displayed when you click Add Group or Edit Group button on the Group dialog screen.

5.5.1 Line Segment

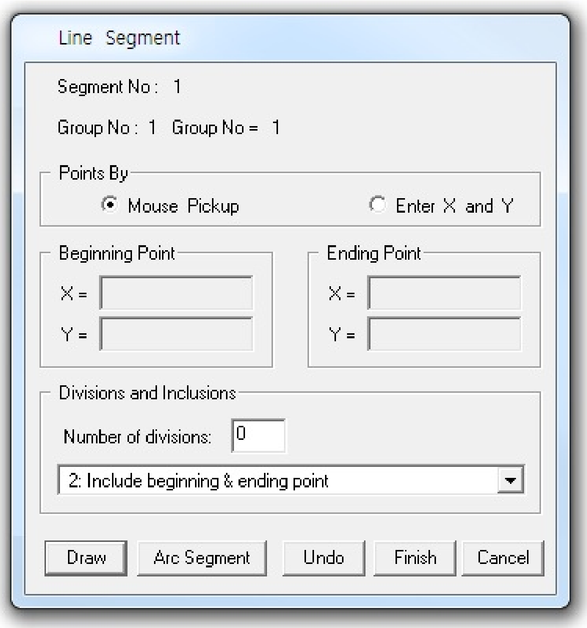


Figure 5.14
Line segment dialog.

Line Segment dialog is shown in Figure 5.14.

Segment No
Current segment number will be displayed automatically.

Group No & Title
Current group number and title will be displayed automatically.

Point By
Select Mouse Pickup or Enter X and Y.

Beginning & Ending Point
Coordinates of beginning and ending points are required when Enter X and Y is selected.

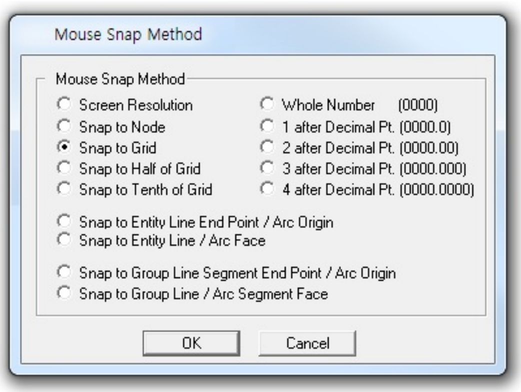
Divisions and Inclusions
Use following default values.
Number of divisions 0
Combo box selection 2: Include beginning & ending point

Draw
Draw line segment.

For Mouse Pickup,
1. Click Draw button.
2. Move the mouse to the point and click left mouse button. Or hold down left mouse button, move the mouse and release the button at the point.

Note 1:
It is important to choose an appropriate mouse snap method before drawing by mouse. Mouse snap dialog in Figure 5.15 can be opened by clicking Mouse-Snap menu in PLOT-2D.
For example, when you choose Snap to Grid, mouse cursor will automatically move to the nearest grid point.

Figure 5.15
Mouse snap dialog



For Enter X and Y,

1. Type in the coordinates of beginning and ending points.
2. Click Draw button.

Note 2:
You can draw many segments continuously by repeating above Mouse Pickup or Enter X and Y procedure.

Arc Segment
Switch to arc segment.

Undo
Undo the changes you just made for line segment.

Finish
Finish and exit from drawing the current group.

Cancel
Cancel and exit from drawing the current group.

5.5.2 Arc Segment

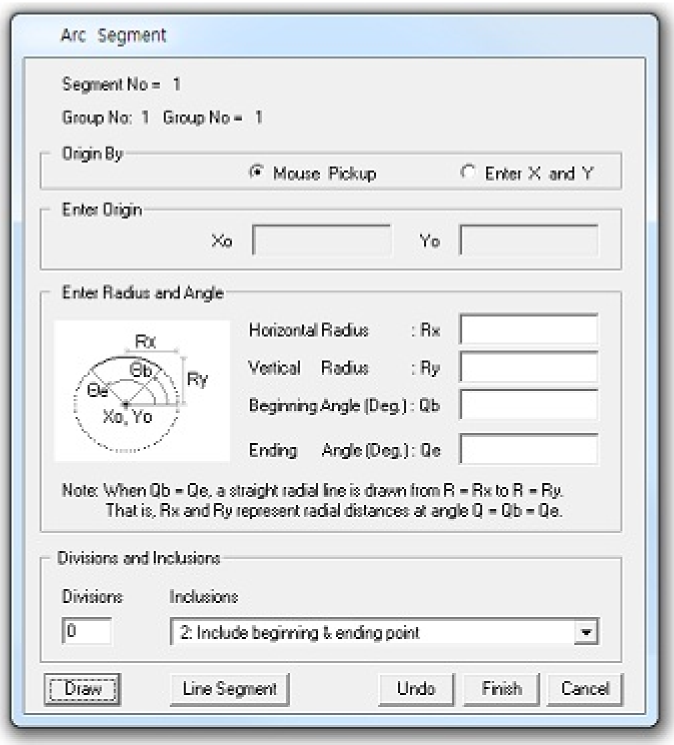


Figure 5.16 Arc segment dialog.

Arc Segment dialog is shown in Figure 5.16.

Segment No
Current segment number will be displayed automatically.

Group No & Title
Current group number and title will be displayed automatically.

Origin By
Select Mouse Pickup or Enter X and Y.

Enter Origin
Coordinates of origin are required for Enter X and Y.

Enter Radius and Angle
Enter Horizontal & vertical radii, and beginning & ending angles.

Divisions and Inclusions
Use following default values.
Number of divisions 0
Combo box selection 2: Include beginning & ending point

Draw
Draw arc segment.

For Mouse Pickup,
1. Type in R_x, R_y, Θ_b, Θ_e
2. Click Draw button
3. Move the mouse to the origin and click left mouse button. Or hold down left mouse button, move the mouse and release the button at the origin.

For Enter X and Y,
1. Type in X_o, Y_o, R_x, R_y, Θ_b, Θ_e
2. Click Draw button

Refer to Note 1 & 2 in Section 5.5.1.

Line Segment
Switch to line segment.

Undo
Undo the changes you just made for arc segment.

Finish
Finish and exit from drawing the current group.

Cancel
Cancel and exit from drawing the current group.

5.6 Modifying Finite Element Meshes

Group Mesh Generator can be used to directly modify finite element meshes.

When you open input file, Mesh Generator reads the extension of the input file name and it assumes that the input file is the finite element mesh file if the extension is .Mes.

Editing finite element meshes has three parts: Nodal Boundary, Nodal Coordinate and Element Material. These editing modes can be accessed from Mesh menu in PLOT-2D as shown in Figure 5.17.

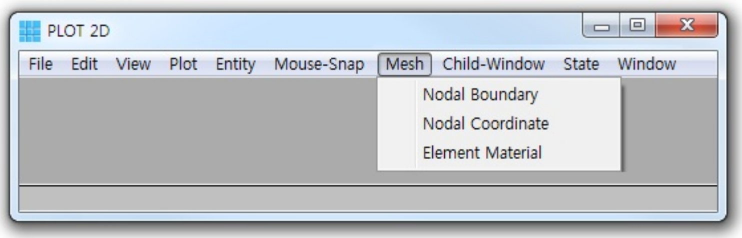


Figure 5.17 Menu for editing finite element mesh

It should be noted that once you edited the finite element meshes, modified finite element mesh is saved as MeshFile.Mes in the current working directory. The original input mesh file is not changed.

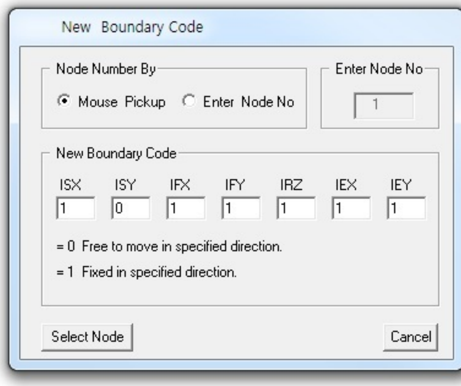
5.6.1 Edit Nodal Boundary

When you click Nodal Boundary from the Mesh menu, Edit Boundary dialog will be displayed.

5.6.1.1 Mouse Pickup

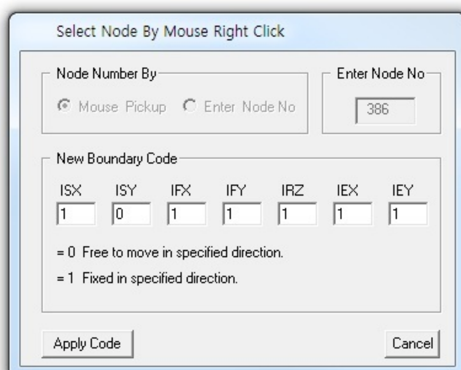
When you select Mouse Pickup mode as in Figure 5.18, you are supposed to select node number by mouse click. Click Select Node button.

Figure 5.18
Edit boundary
(Mouse Pickup)



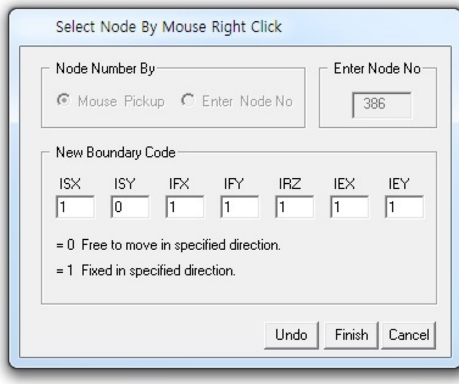
Click the node by Mouse Right Click, edit boundary codes and then click Apply Code button in Figure 5.19.

Figure 5.19
Edit boundary
(Apply Code)



You can repeat the same procedure many times for other nodes. Once finished, click Finish button in Figure 5.20.

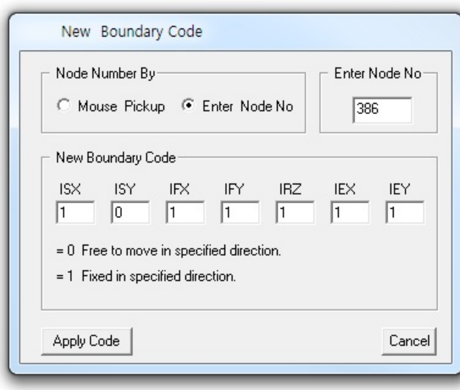
Figure 5.20
Edit boundary (Finish)



5.6.1.2 Enter Node No

When you select Enter Node No mode as in Figure 5.21, you are supposed to type in node number. Edit boundary codes and then click Apply Code button.

Figure 5.21
Edit boundary (Enter Node No)



You can repeat the same procedure many times for other nodes. Once finished, click Finish button.

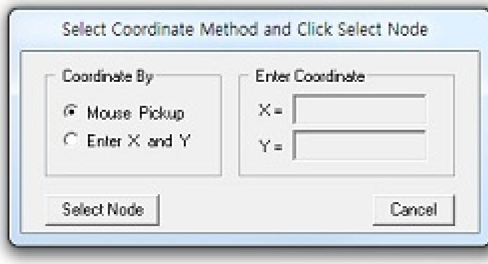
5.6.2 Edit Nodal Coordinate

When you click Nodal Coordinate from the Mesh menu, Edit Coordinate dialog will be displayed.

5.6.2.1 Mouse Pickup

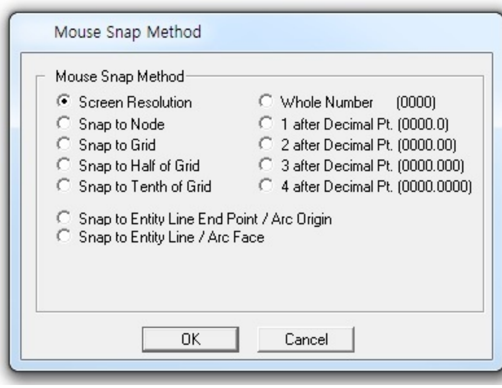
When you select Mouse Pickup mode as in Figure 5.22, you are supposed to select node number by mouse click. Click Select Node button.

Figure 5.22
Edit coordinate
(Mouse Pickup)



Select the node number by Mouse Right Click and then move the coordinate by Mouse Left Click. It is convenient to select an appropriate Mouse-Snap method in Figure 5.23 before moving the coordinates.

Figure 5.23
Mouse snap method



You can repeat the same procedure many times for other nodes. Once finished, click Finish button in Figure 5.24.

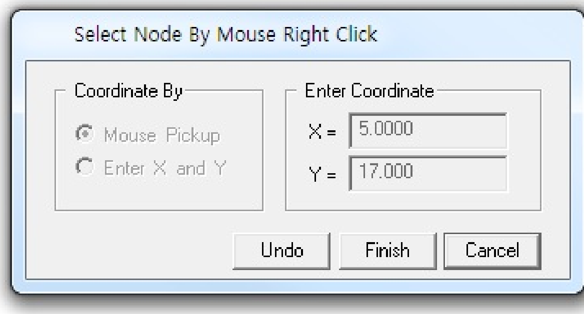


Figure 5.24 Edit coordinate (Finish)

5.6.2.2 Enter X and Y

When you select Enter X and Y mode as in Figure 5.25, you are supposed to type in nodal coordinates. Type in X and Y coordinates and then click Apply button.

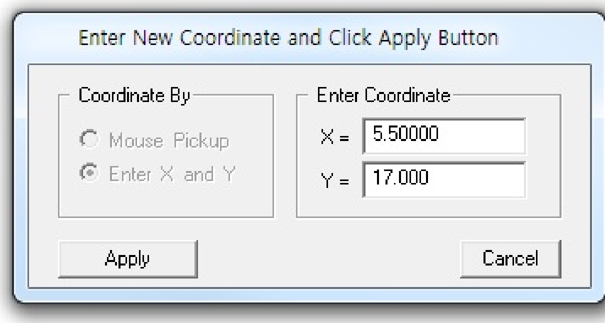


Figure 5.25 Edit coordinate (Enter X and Y)

You can repeat the same procedure many times for other nodes. Once finished, click Finish button.

5.6.3 Edit Element Material

When you click Element Material from the Mesh menu, Edit Element Material dialog will be displayed.

5.6.3.1 Mouse Pickup

When you select Mouse Pickup mode as in Figure 5.26, you are supposed to select element number by mouse click.
Click Select Element button.

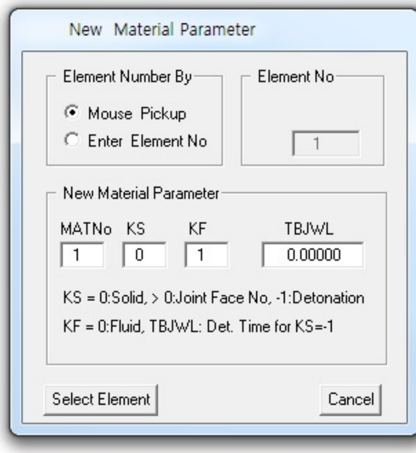


Figure 5.26
Edit element material
(Mouse Pickup)

Click the element by Mouse Right Click, edit material parameters and then click Apply button in Figure 5.27.

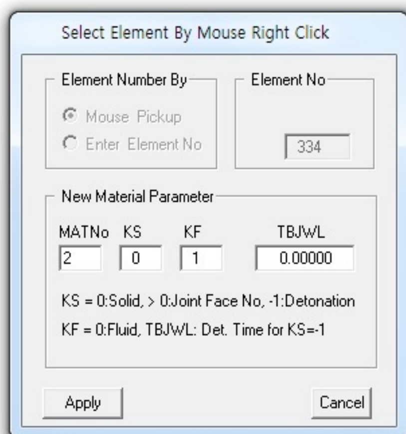


Figure 5.27
Edit element material
(Apply)

You can repeat the same procedure many times for other elements. Once finished, click Finish button in Figure 5.28.

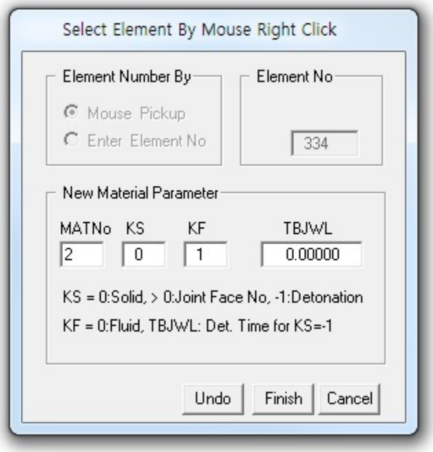


Figure 5.28
Edit element material
(Finish)

5.6.3.2 Enter Element No

When you select Enter Element No mode as in Figure 5.29, you are supposed to type in element number. Edit material parameters and then click Apply button.

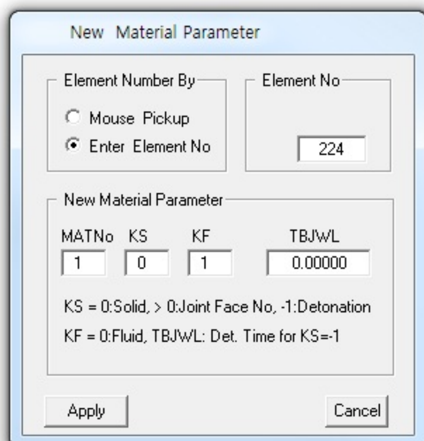


Figure 5.29
Edit element material
(Enter Element No)

You can repeat the same procedure many times for other elements. Once finished, click Finish button.

5.7 Entities

Entities are graphical objects which are mainly used to assist editing the geometry of groups and elements.

There are four types of entities: Mark, Line, Arc, and Text.

Entities can be accessed from Entity menu in PLOT-2D as shown in Figure 5.30.

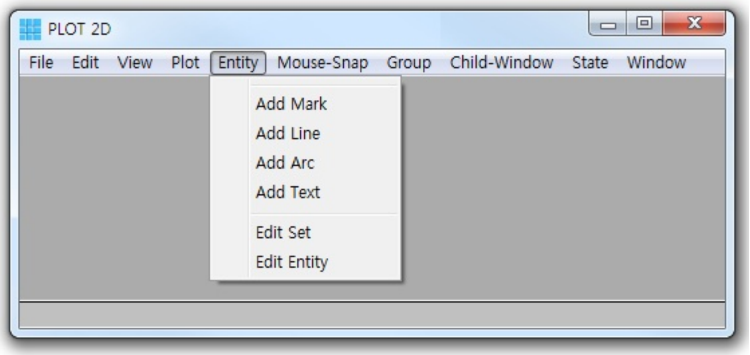


Figure 5.30 Entity menu

Entity menu has six parts:
Add Mark, Add Line, Add Arc, Add Text, Edit Set and Edit Entity.

First four Add Entities are to build new entities.

Edit Set is to assign entity set so that each plot number can include only selected entities.

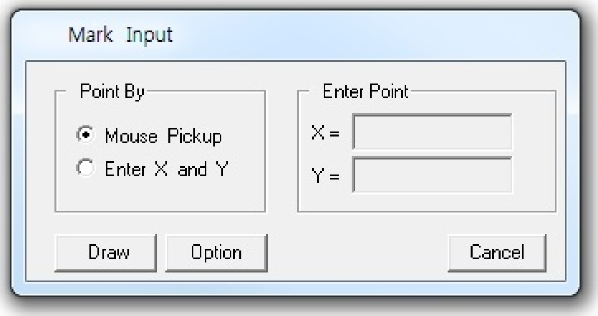
Edit Entity is to modify, delete or replace the selected entity.

5.7.1 Add Mark

Marks are graphical symbols which are mainly used to assist editing the geometry of groups and elements.

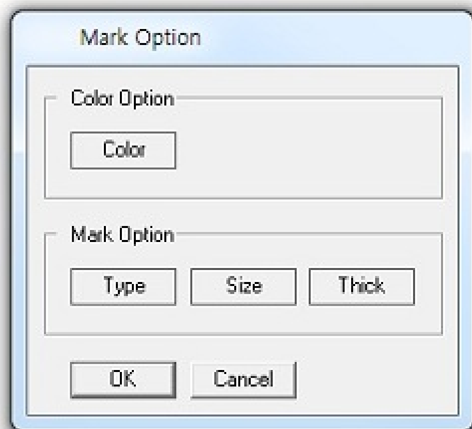
When you select Add Mark submenu, Mark Input dialog in Figure 5.31 is displayed.

Figure 5.31
Mark input
(Mouse Pickup)



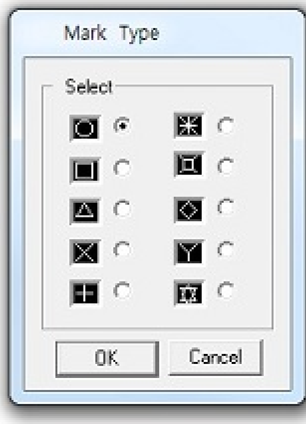
Option button is to show Mark Option in Figure 5.32.

Figure 5.32
Mark option dialog



Available Mark Types are shown in Figure 5.33.

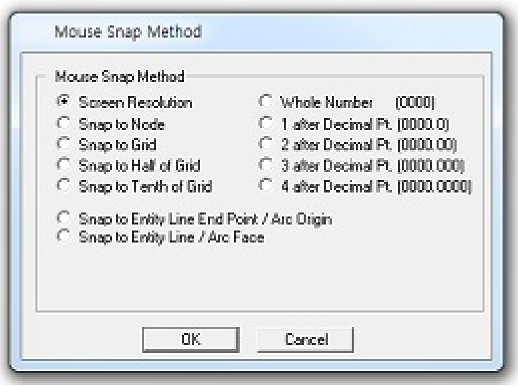
Figure 5.33 Mark type dialog



5.7.1.1 Mouse Pickup

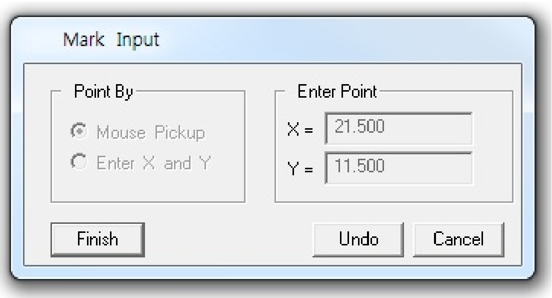
When you select Mouse Pickup mode as in Figure 5.31, you are supposed to select the mark center position by mouse click. Click Draw button and then move the position by Mouse Left Click. It is convenient to select an appropriate Mouse-Snap method in Figure 5.34 before moving the position.

Figure 5.34
Mouse snap method



Once finished, click Finish button in Figure 5.35.

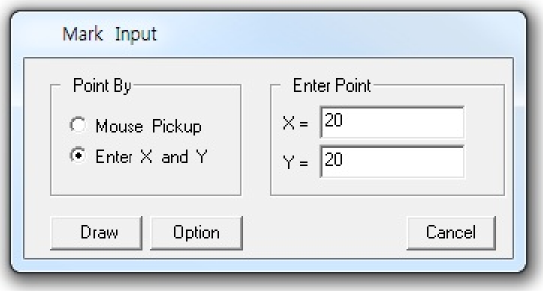
Figure 5.35
Mark input
(Finish)



5.7.1.2 Enter X and Y

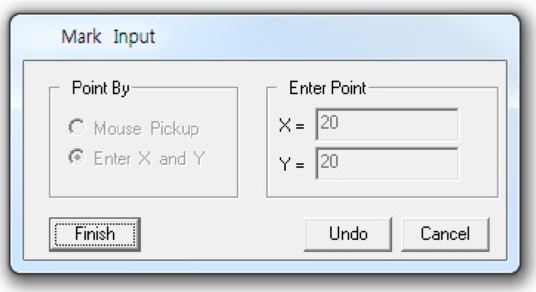
When you select Enter X and Y mode as in Figure 5.36, you are supposed to type in the coordinates of the mark center position. Click Draw button.

Figure 5.36
Mark input
(Enter X and Y)



Once finished, click Finish button in Figure 5.37.

Figure 5.37
Mark input
(Finish)

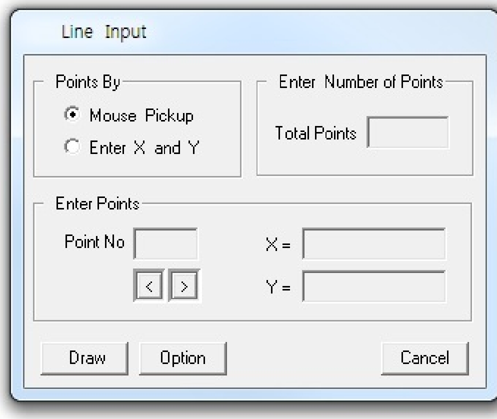


5.7.2 Add Line

Lines are graphical objects which are mainly used to assist editing the geometry of groups and elements.

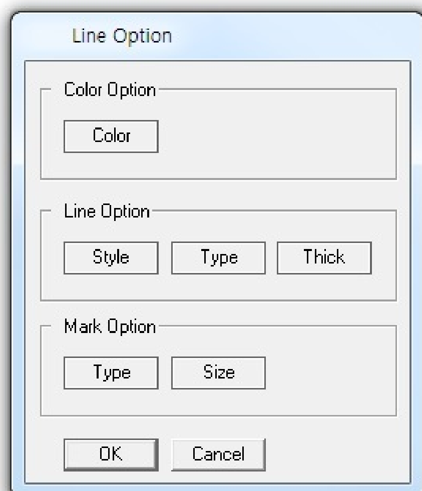
When you select Add Line submenu, Line Input dialog in Figure 5.38 is displayed.

Figure 5.38
Line input
(Mouse Pickup)



Option button is to show Line Option in Figure 5.39.

Figure 5.39
Line option dialog



Available Line Styles are shown in Figure 5.40.

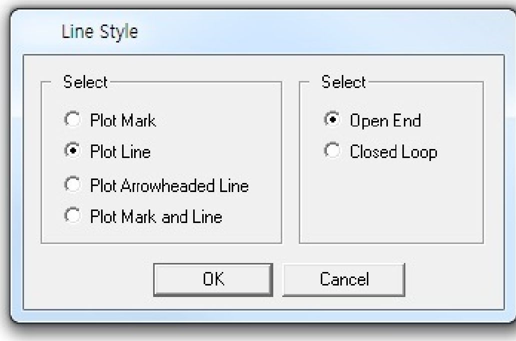


Figure 5.40 Line style dialog

Available Line Types are shown in Figure 5.41.

Figure 5.41 Line type dialog

5.7.2.1 Mouse Pickup

When you select Mouse Pickup mode as in Figure 5.38,
you are supposed to select the line end point by mouse click.
Click Draw button and then select the point by Mouse Left Click.

It is convenient to select an appropriate Mouse-Snap method
in Figure 5.34 before moving the coordinate.

You can click many points to build continuous lines.
Once finished, click Finish button in Figure 5.42.

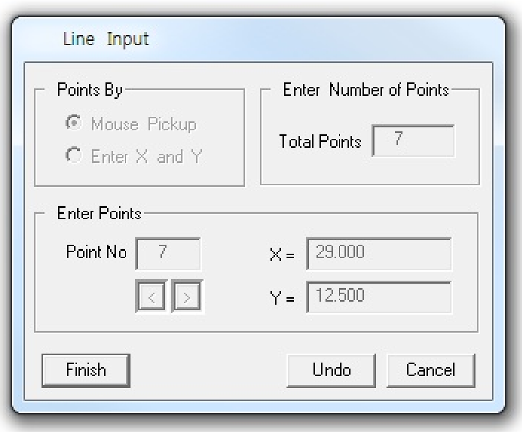
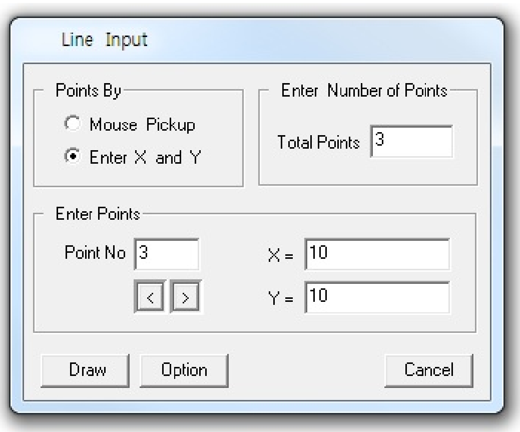


Figure 5.42 Line input (Finish)

5.7.2.2 Enter X and Y

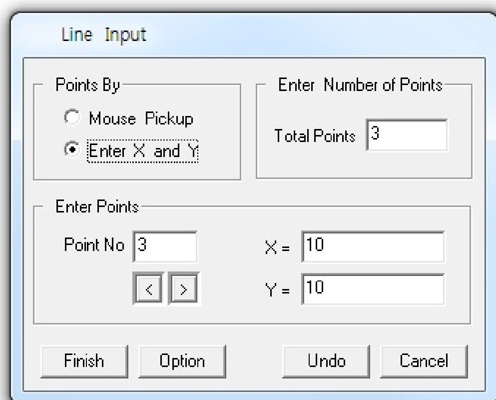
When you select Enter X and Y mode as in Figure 5.43,
you are supposed to type the coordinates of the line.
Click Draw button.

Figure 5.43
Line input
(Enter X and Y)



And then click Finish button in Figure 5.44.

Figure 5.44
Line input
(Finish)

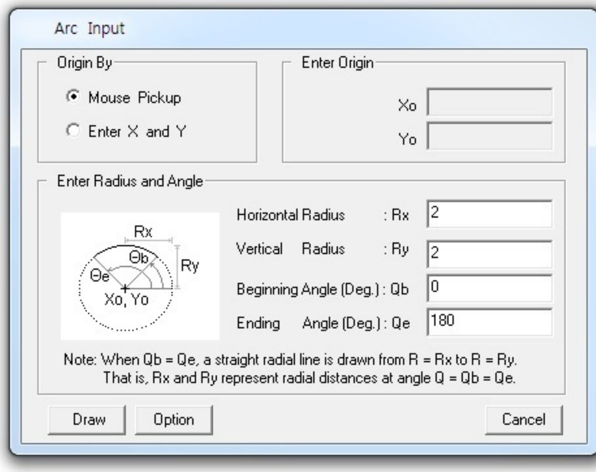


5.7.3 Add Arc

Arcs are graphical objects which are mainly used to assist editing the geometry of groups and elements.

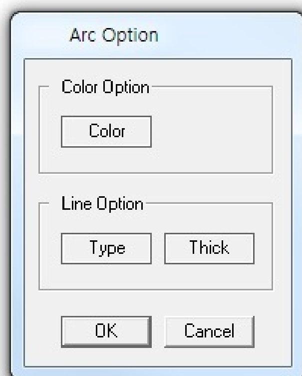
When you select Add Arc submenu, Arc Input dialog in Figure 5.45 is displayed.

Figure 5.45
Arc input
(Mouse Pickup)



Option button is to show Arc Option in Figure 5.46.

Figure 5.46 Arc option dialog



5.7.3.1 Mouse Pickup

When you select Mouse Pickup mode as in Figure 5.45,
you are supposed to select the arc origin by mouse click.

Type in Horizontal Radius, Vertical Radius, Beginning Angle
and Ending Angle.

Click Draw button and then select the origin by Mouse Left Click.
It is convenient to select an appropriate Mouse-Snap method
in Figure 5.34 before moving the coordinate.

Once finished, click Finish button in Figure 5.47.

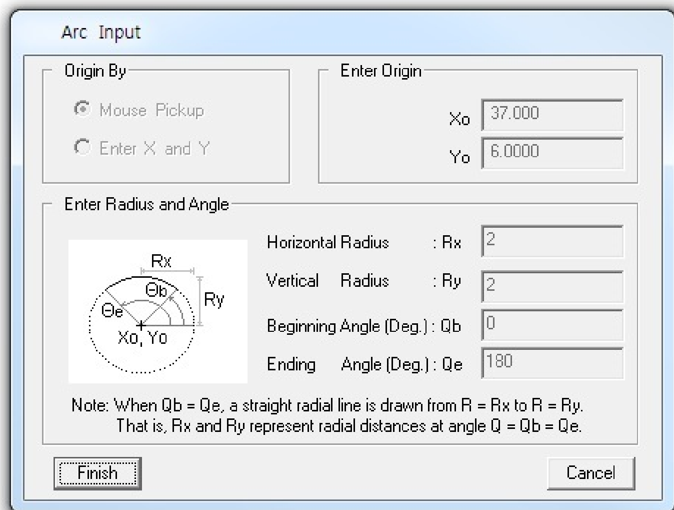


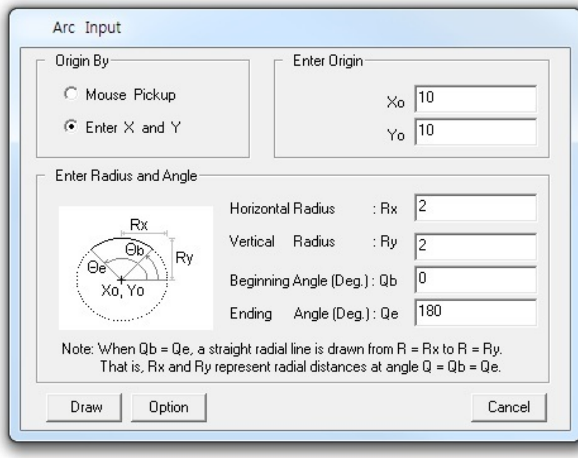
Figure 5.47 Arc input (Finish)

5.7.3.2 Enter X and Y

When you select Enter X and Y mode as in Figure 5.48, you are supposed to type in the coordinates of the arc origin.

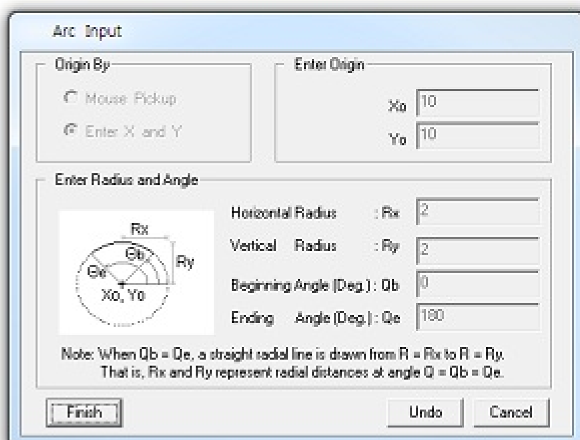
Type in Horizontal Radius, Vertical Radius, Beginning Angle and Ending Angle. And then click Draw button.

Figure 5.48
Arc input
(Enter X and Y)



Once finished, click Finish button in Figure 5.49.

Figure 5.49
Arc input
(Finish)

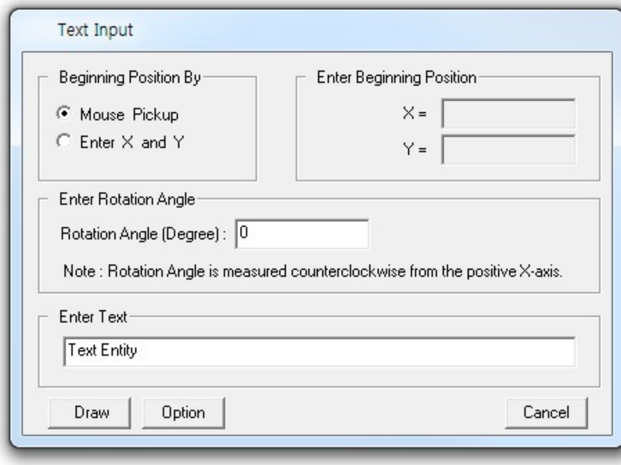


5.7.4 Add Text

Texts are characters which are mainly used to assist describing the geometry of groups and elements.

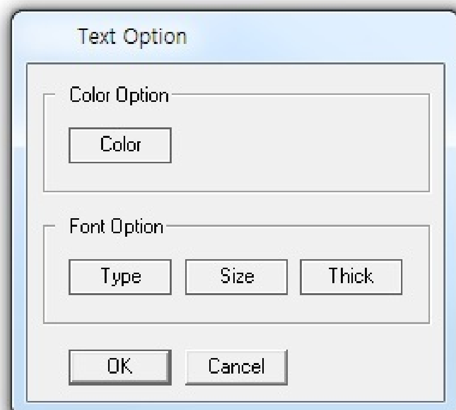
When you select Add Text submenu, Text Input dialog in Figure 5.50 is displayed.

Figure 5.50
Text input
(Mouse Pickup)



Option button is to show Text Option in Figure 5.51.

Figure 5.51
Text option dialog



Available Font Sizes are shown in Figure 5.52.

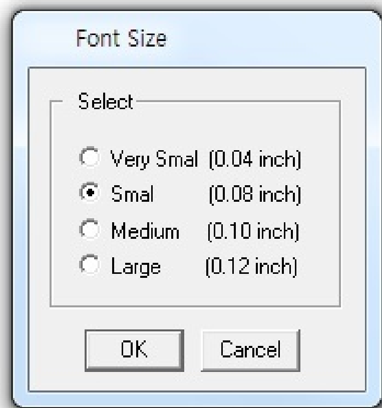


Figure 5.52 Font size dialog

5.7.4.1 Mouse Pickup

When you select Mouse Pickup mode as in Figure 5.50,
you are supposed to select the beginning position of text by mouse click.

Type in Rotation Angle and Text.

Click Draw button and then select the beginning position of the text by Mouse Left Click. It is convenient to select an appropriate Mouse-Snap method in Figure 5.34 before moving the coordinate.

Once finished, click Finish button in Figure 5.53.

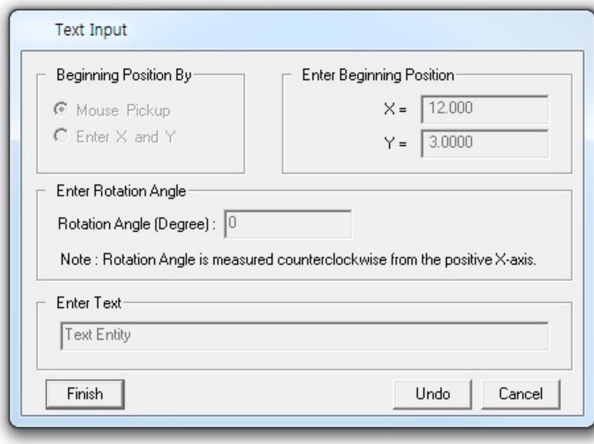


Figure 5.53 Text input (Finish)

5.7.4.2 Enter X and Y

When you select Enter X and Y mode as in Figure 5.54,
you are supposed to type in the coordinates of beginning position of text.

Type in Rotation Angle and Text. And then click Draw button.

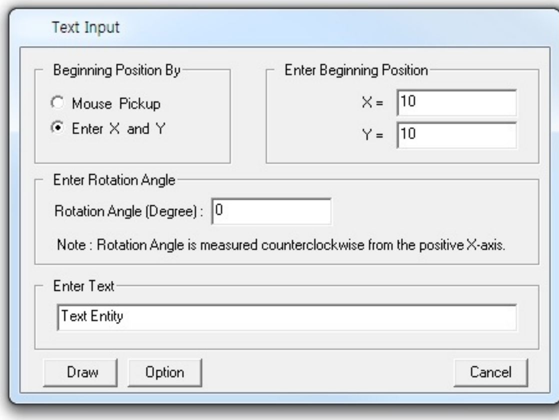


Figure 5.54
Text input
(Enter X and Y)

Once finished, click Finish button in Figure 5.55.

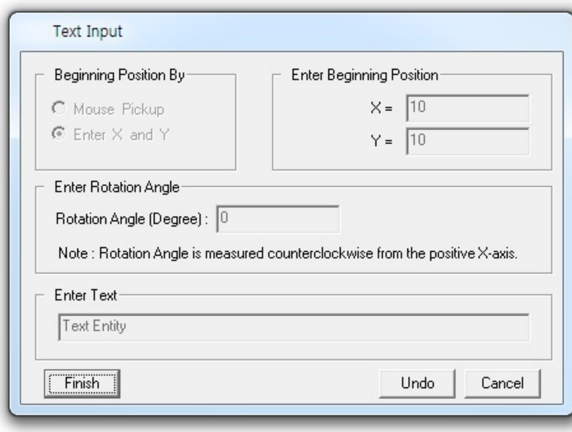


Figure 5.55
Text input
(Finish)

5.7.5 Edit Set

Edit Set is to assign Entity Set as shown in Figure 5.56.

Edit Set consists of two parts:

1. Enter Entity Set Number and assign Entity Numbers.
2. Enter Plot Number and assign Entity Set Number.

Every time Enter Set Number or Plot Number is changed, click Update button. When finished, click Finish button.

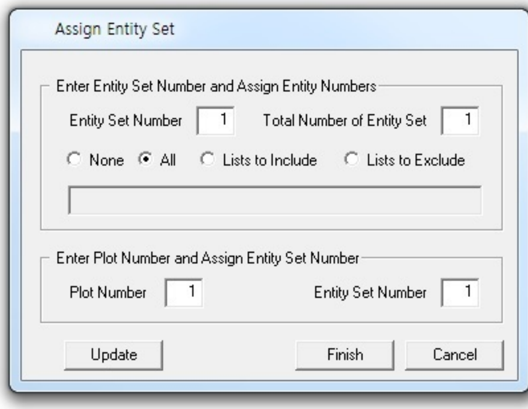


Figure 5.56
Assign entity set dialog

5.7.5.1 Enter Entity Set No & Assign Entity No

Here, you enter Entity Set Number, Total Number of Entity Set and then select Option for the current set.

When Lists to Include option is selected, type in entity numbers to be included in the current set.

When Lists to Exclude option is selected, type in entity numbers to be excluded in the current set.

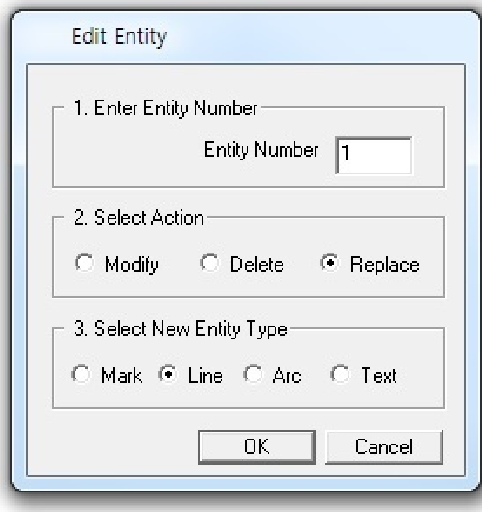
5.7.5.2 Enter Plot No & Assign Entity Set No

Here, you enter Plot Number and assign Entity Set Number.

5.7.6 Edit Entity

Edit Entity is to modify, delete or replace the selected entity as shown in Figure 5.57.

Figure 5.57
Edit entity dialog



5.7.6.1 Modify

Modify is to modify the current entity.

When OK button is clicked, Entity Input dialog corresponding to the current entity is displayed. Follow the same procedure as described in Add Entity.

5.7.6.2 Delete

Delete is to delete the current entity.

5.7.6.3 Replace

Replace is to replace the current entity by new entity type.

When OK button is clicked, Entity Input dialog corresponding to the new entity type is displayed. Follow the same procedure as described in Add Entity.

Block Mesh
User's Manual

6.1 Introduction

Block Mesh Generator is a three-dimensional CAD program specially designed to build block mesh which can be used to generate finite element mesh with the aid of program PRESMAP-GP.

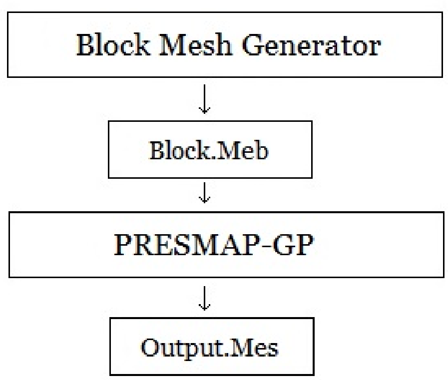


Figure 6.1 Flow diagram of block mesh generation

Block.Meb contains block mesh data that can be generated or modified by Block Mesh Generator. The file Block.Meb is used as input to the program PRESMAP-GP, thereby generating finite element mesh file Output.Mes.

Block Mesh Generator can be accessed through SMAP menu
Run or Plot as explained in Section 6.2.

PRESMAP-GP can be accessed from SMAP menu:
Run → Mesh Generator → PreSmap → Presmap GP.
This program can also be accessed indirectly by executing
Show F. E. Mesh in Block Editor dialog in Section 6.5.8.

6.2 Block Mesh Generator

Block Mesh Generator can be accessed by selecting the following menu items in SMAP:
Run → Mesh Generator → Block Mesh or
Plot → Mesh → Block Mesh

When you build new block mesh, PLOT-3D program in Figure 6.2 is displayed along with Work Plane Editor in Figure 6.3.
Click Block Editor toolbar in Figure 6.4. Building new block is discussed in detail in Section 6.5.8.

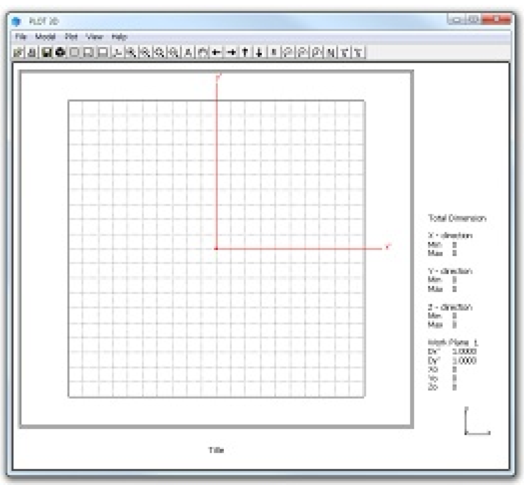


Figure 6.2 Prebuilt work plane on PLOT-3D

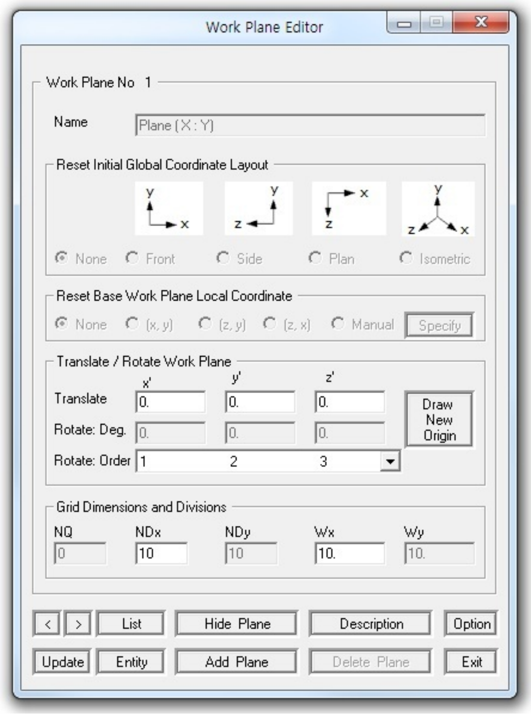


Figure 6.3 Prebuilt work plane editor

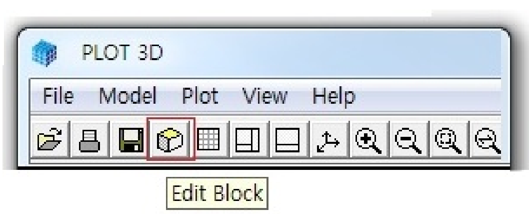


Figure 6.4 Block editor toolbar

When you open existing block mesh, Select Open in SMAP menu as shown in Figure 6.5 and then select the input file. Block mesh will be displayed on PLOT-3D as in Figure 6.6.

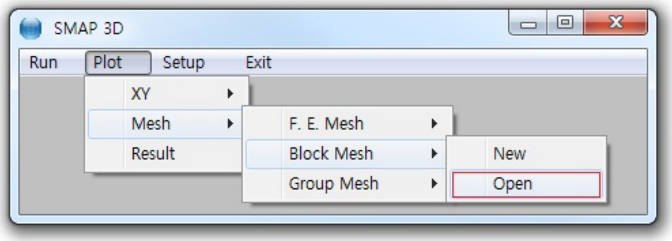


Figure 6.5 Open input file dialog

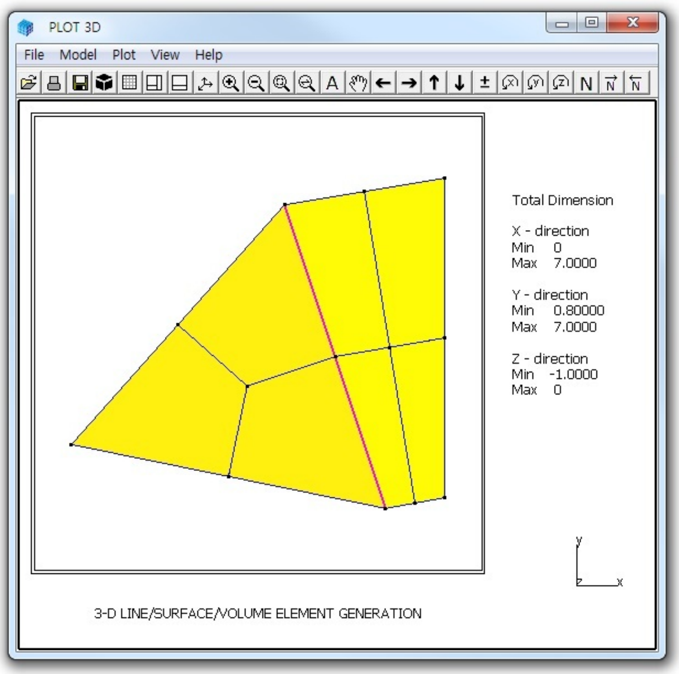


Figure 6.6 Block mesh on PLOT-3D

6.3 Work Plane

Work Planes are rectangular planes with grid lines and local coordinate axes, which are mainly used to assist editing the geometry of blocks and elements.

Work Plane Editor can be accessed by selecting the following menu items in PLOT-3D:
Model → Work Plane → Show Editor
or by clicking Work Plane toolbar as shown in Figure 6.7.

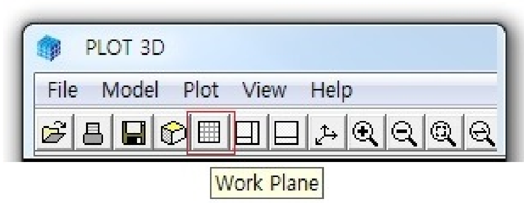


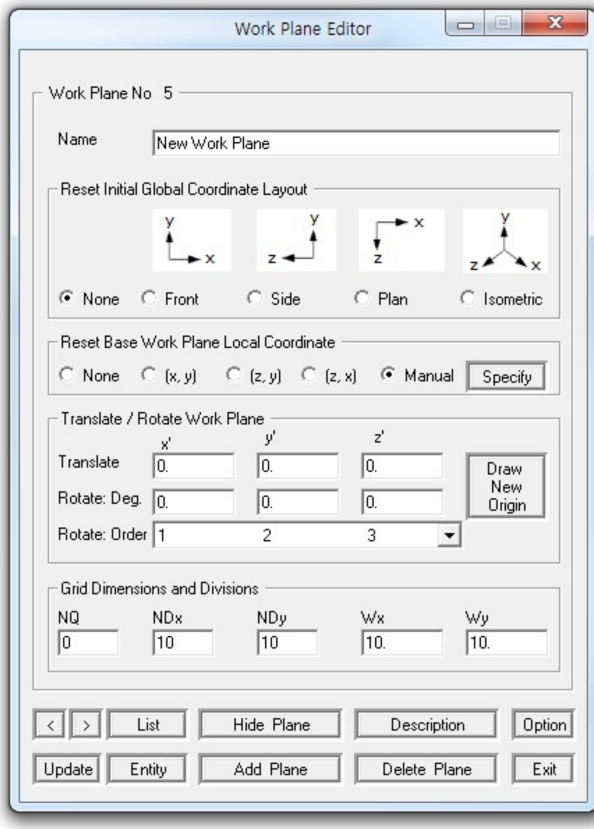
Figure 6.7 Work plane toolbar

Work Plane Editor dialog in Figure 6.8 consists of following six parts:

- Name
- Reset Initial Global Coordinate Layout
- Reset Base Work Plane Local Coordinate
- Translate / Rotate Work Plane
- Grid Dimensions and Divisions
- Command Buttons

First three work planes are prebuilt work planes:
(X : Y), (Z : Y) and (Z : X) planes. New work planes can be added by copying one of these prebuilt planes.

Figure 6.8
Work plane editor



6.3.1 Name

Name is work plane name you can specify for identification.

6.3.2 Reset Initial Global Coordinate Layout

This is used to reset initial global coordinate layout. You can select Front, Side, Plan or Isometric views. Once selected, click Update button to see the selected layout.

6.3.3 Reset Base Work Plane Local Coordinate

This is used to reset base work plane local coordinate.
You can select (x, y), (z, y), (z, x) or Manual.

For Manual, click Specify button to display Base Work Plane Local Coordinate dialog in Figure 6.9. Base work plane can be specified either by coordinates of three points or by three node numbers.

Once selected, click Update button to see the selected local coordinate.

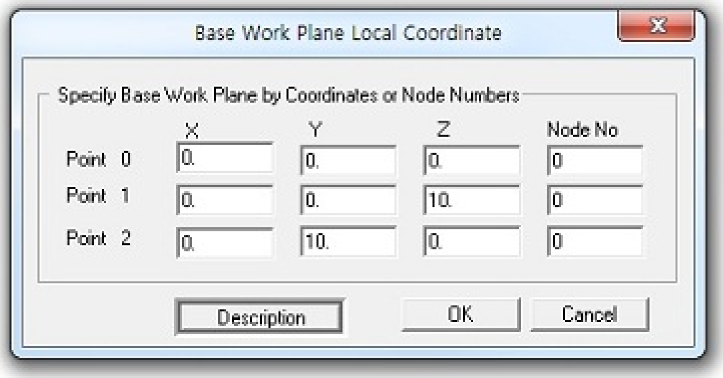


Figure 6.9 Base work plane local coordinate dialog

6.3.4 Translate / Rotate Work Plane

This is used to translate and rotate work plane.
When you rotate about more than one axis, select appropriate rotation order from the list box.

Click Draw New Origin button in Figure 6.8 to display Work Plane Origin dialog in Figure 6.10. This is a convenient way of moving the work plane origin.

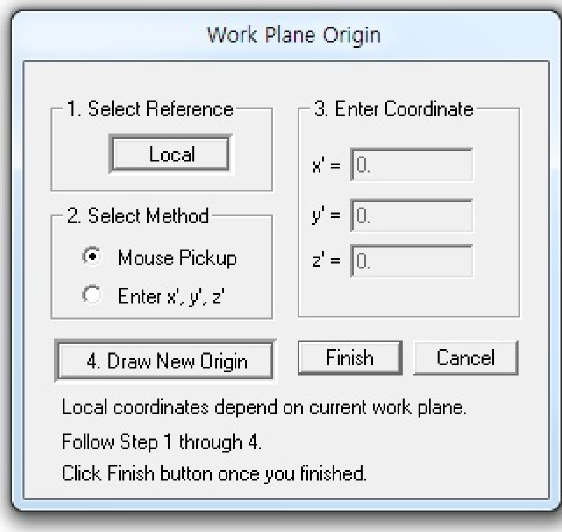


Figure 6.10 Work plane origin dialog

6.3.5 Grid Dimensions and Divisions

You can specify quadrant (NQ), grid divisions (NDx, NDy), and grid dimensions (Wx, Wy) as shown in work plane description in Figure 6.11.

Normally, you set the grid dimensions such that they include all blocks.

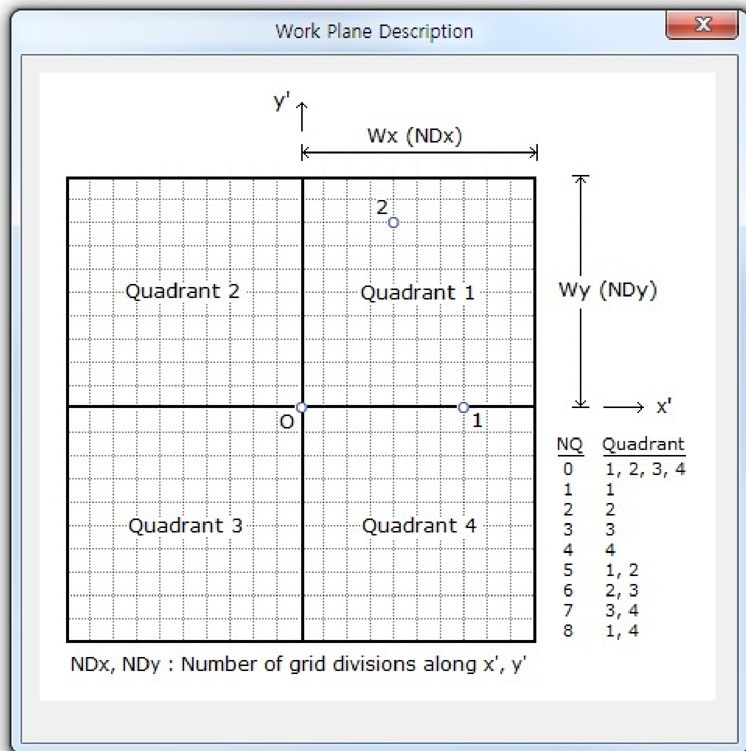


Figure 6.11 Work plane description

6.3.6 Command Buttons

Command buttons are shown on the bottom of Work Plane Editor dialog.

List

This is used to list all available work planes in Figure 6.12.
When you click OK button, selected work plane will be displayed as the current work plane.

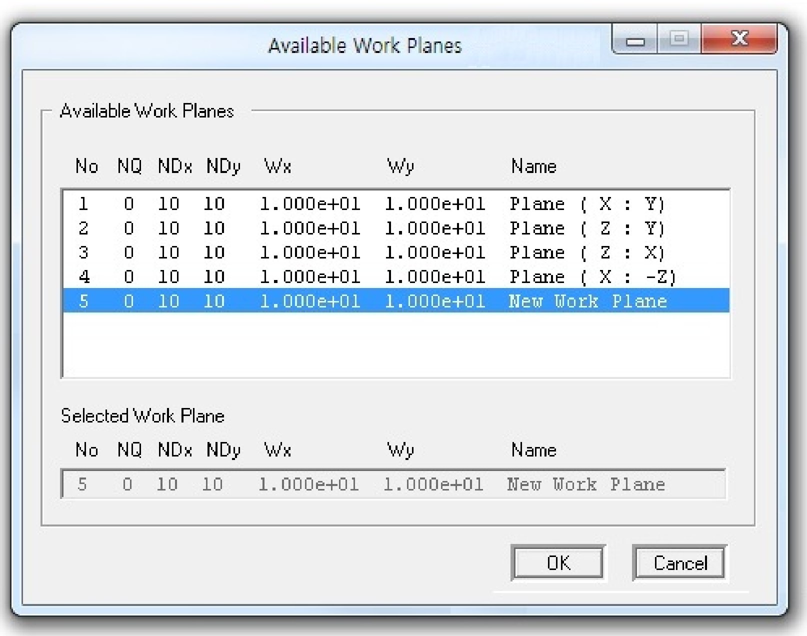


Figure 6.12 Work plane list

Hide Plane

This is used to hide the work plane and entities on the screen.

Description

This is used to show the description of work plane as shown in Figure 6.11.

Option

This is used to open work plane option dialog in Figure 6.13.
Click Update button on this dialog to see the changes made by selected options.

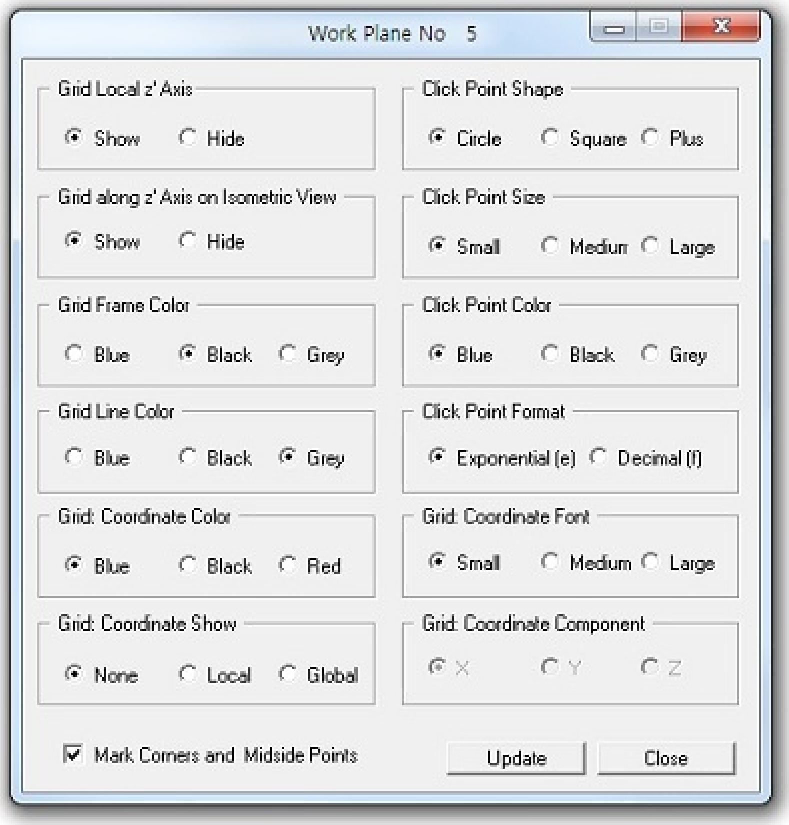


Figure 6.13 Work plane option dialog

Update

This is used to update the current work plane parameters shown on the Work Plane Editor dialog.

Entity

This is used to show Entity Editor dialog in Figure 6.17.

Entities are geometric objects under the current work plane, which are mainly used to assist editing the geometry of blocks and elements. Section 6.4 discusses entities in detail.

Add Plane

This is used to add new work plane.

New work plane is made by copying the work plane shown on the dialog. Once you edit work plane parameters, click Update button in the Work Plane Editor dialog to see the changes.

Delete Plane

This is used to delete the current work plane.

Exit

This is used to hide the work plane and exit from the dialog.

6.3.7 Prebuilt Work Planes

First three work planes are prebuilt work planes:
(x : y), (z : y) and (z : x) planes.

These Prebuilt Work Planes can be accessed by selecting the following menu items in PLOT-3D as shown in Figure 6.14:
Model → Work Plane

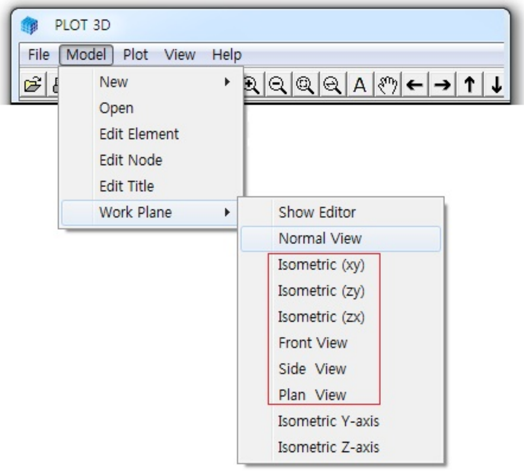


Figure 6.14 Prebuilt work plane menus

or by clicking Axis toolbar as shown in Figure 6.15.

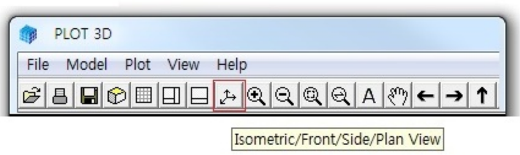


Figure 6.15 Axis toolbar

There are six different views associated with these prebuilt work planes as shown in Figure 6.16: Isometric (xy), Isometric (zy), Isometric (zx), Front, Side and Plan views.

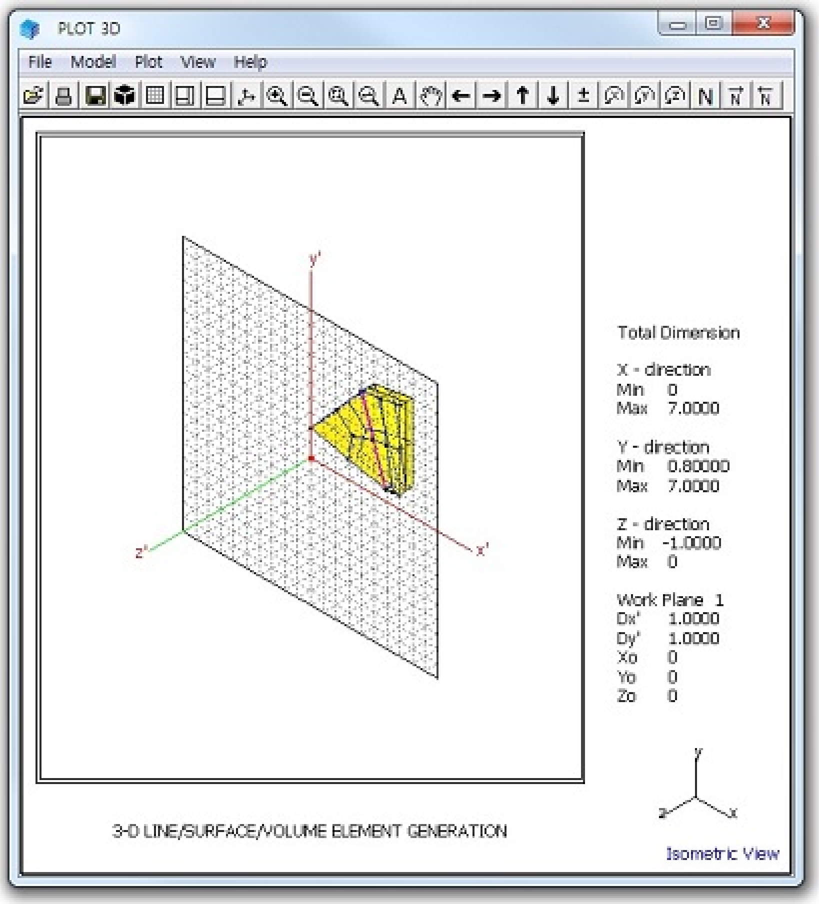


Figure 6.16 Prebuilt work plane: Isometric (xy) View

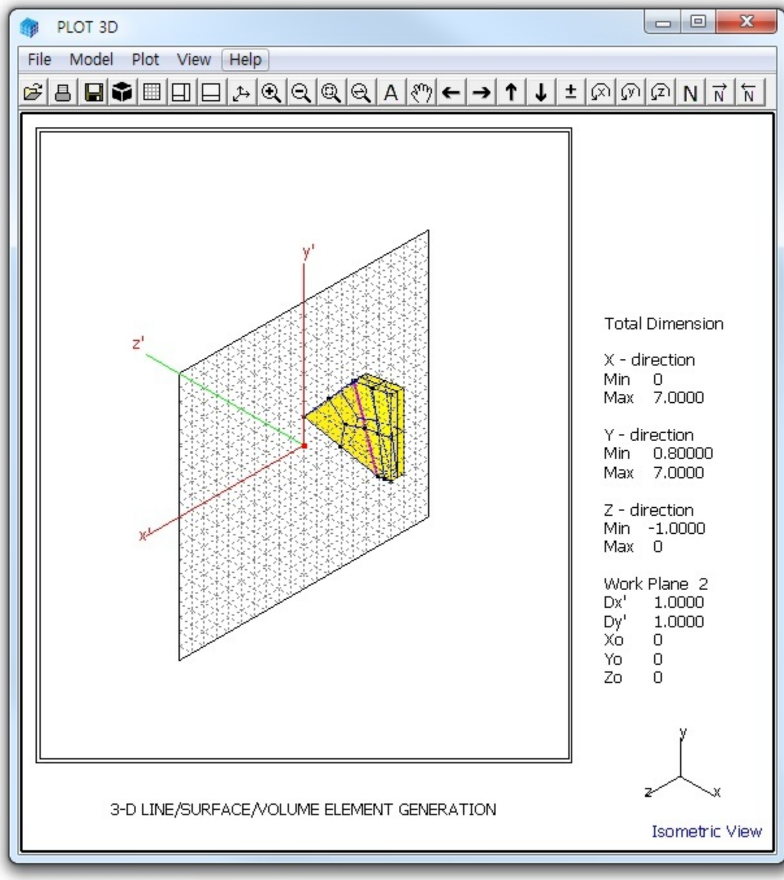


Figure 6.16 Prebuilt work plane: Isometric (zy) View

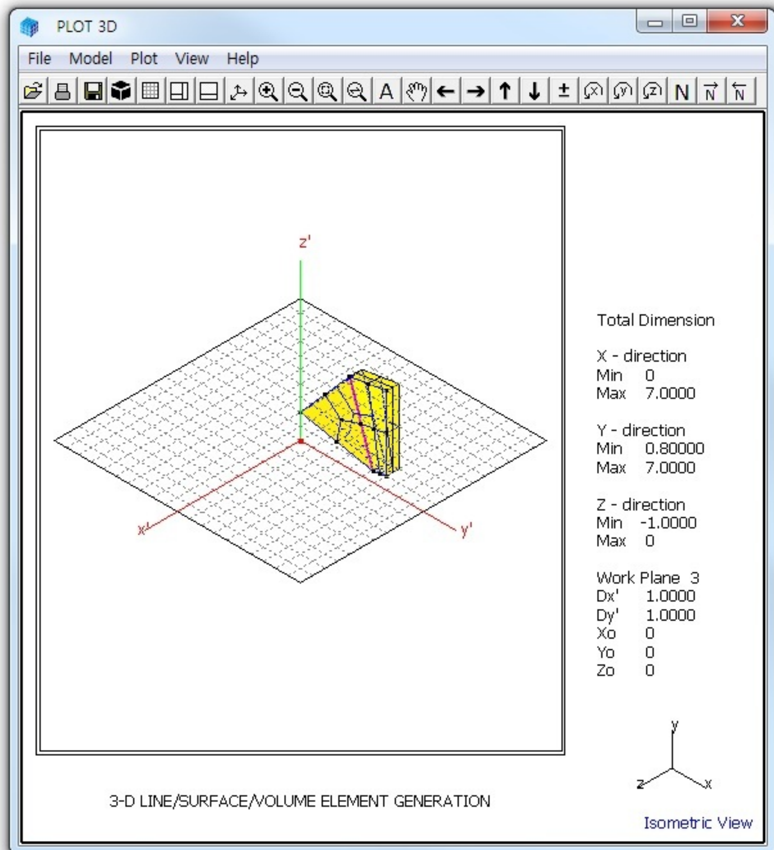


Figure 6.16 Prebuilt work plane: Isometric (zx) View

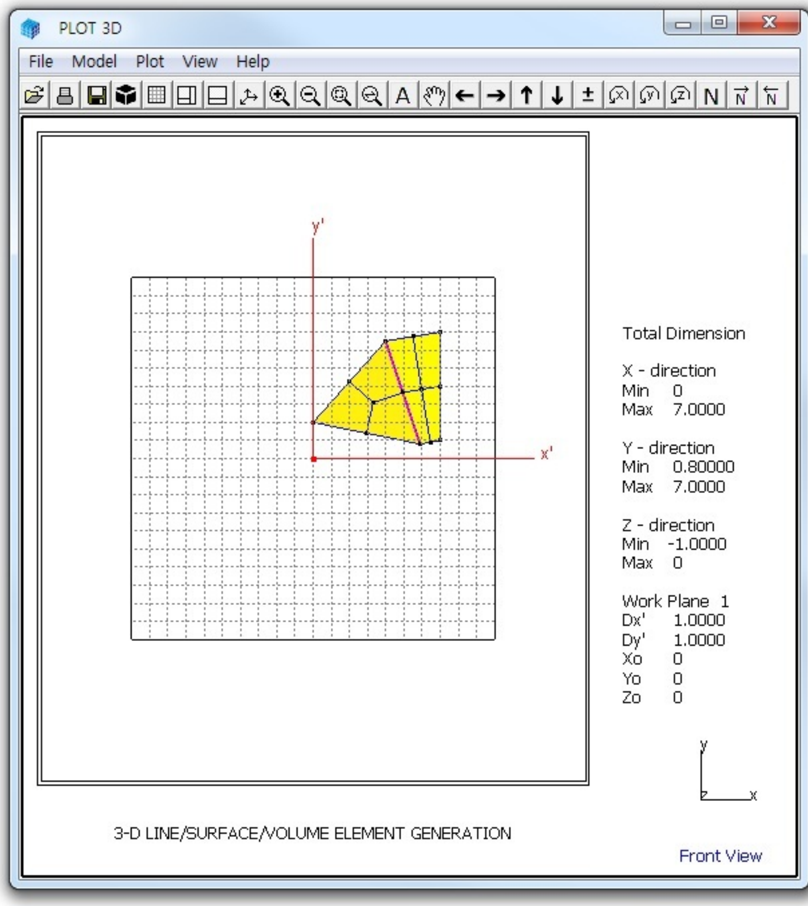


Figure 6.16 Prebuilt work plane: Front View

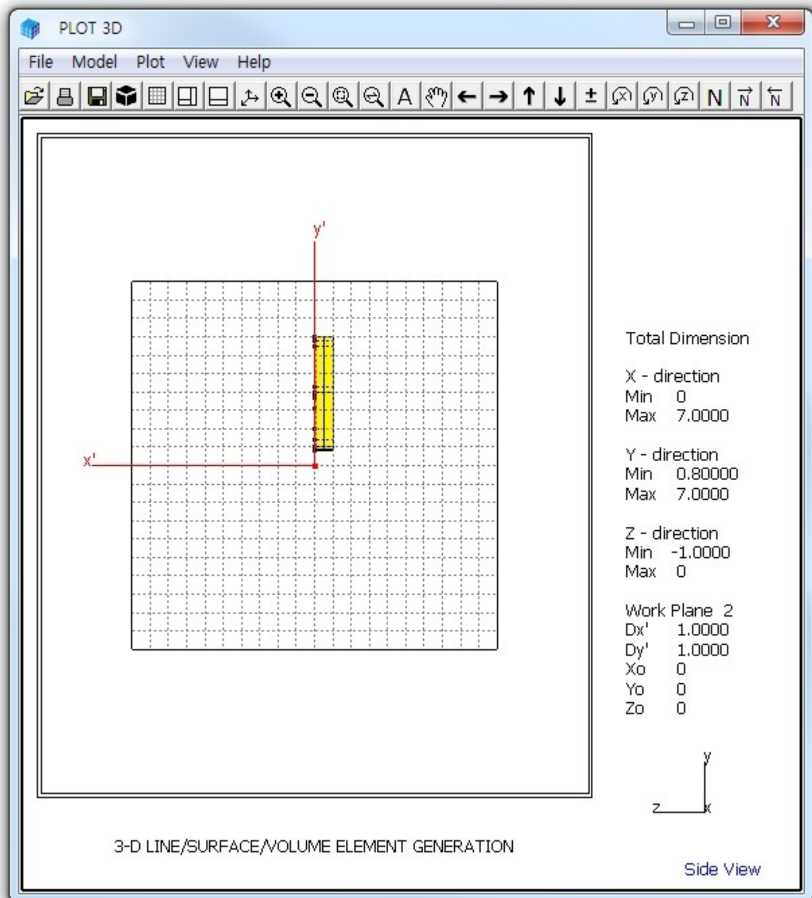


Figure 6.16 Prebuilt work plane: Side View

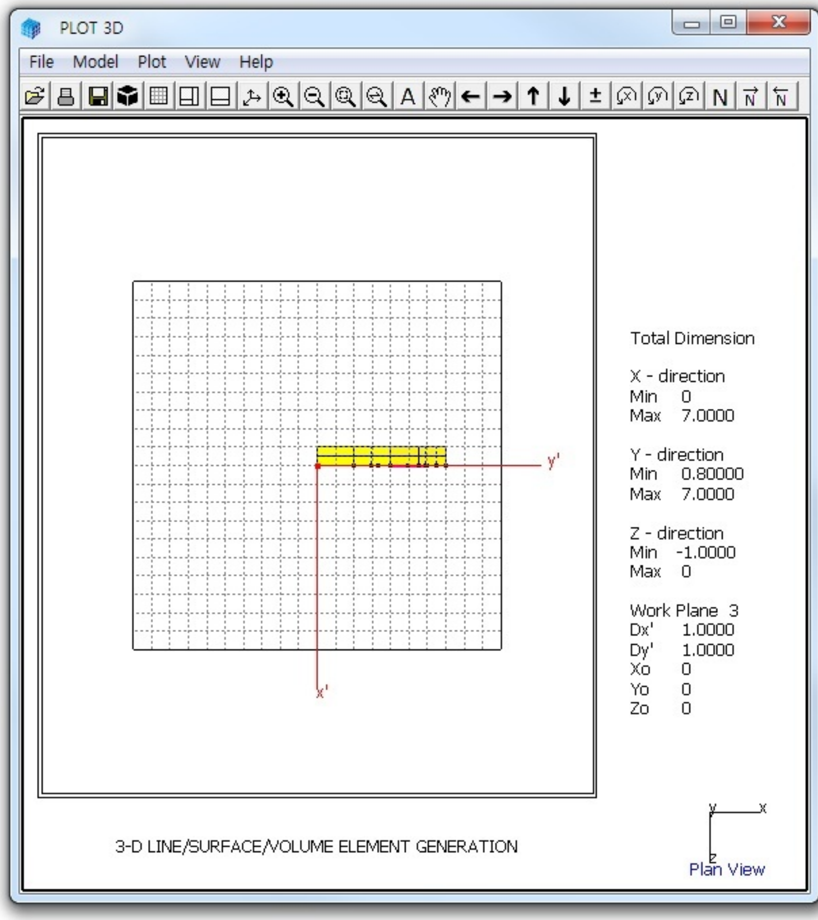


Figure 6.16 Prebuilt work plane: Plan View

6.4 Entities

Entities are geometric objects under the work plane, which are mainly used to assist editing geometry of blocks and elements.
There are five types of entities: Line, Arc, Cube, Ellipsoid, and Cylinder.

Entity Editor dialog in Figure 6.17 can be accessed by clicking Entity button on the Work Plane Editor dialog in Figure 6.8.

Entity Editor dialog consists of following seven parts:

- Entity Number
- Line Thickness
- Line Type
- Line Visibility
- Line Color
- Reference Coordinate
- Command Buttons

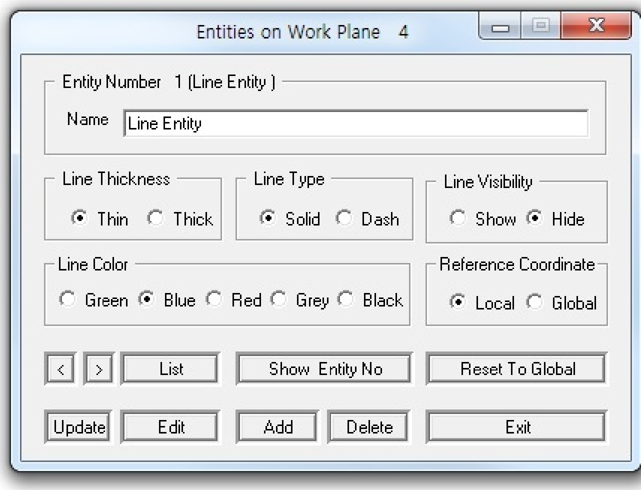


Figure 6.17 Entity editor dialog

6.4.1 Entity Number

Entity number and type are automatically displayed.
You can edit default entity name.

6.4.2 Line Thickness

Two options are available: Thin and Thick.

6.4.3 Line Type

Two options are available: Solid and Dash.

6.4.4 Line Visibility

Two options are available: Show and Hide.

6.4.5 Line Color

Five options are available: Green, Blue, Red, Grey, and Black.

6.4.6 Reference Coordinate

Two options are available: Local and Global.

6.4.7 Command Buttons

Command buttons are shown on Entity dialog in Figure 6.17.

List

This is used to list all available entities in current work plane.

When you click OK button, selected entity will be displayed as the current entity on the Entity Editor dialog.

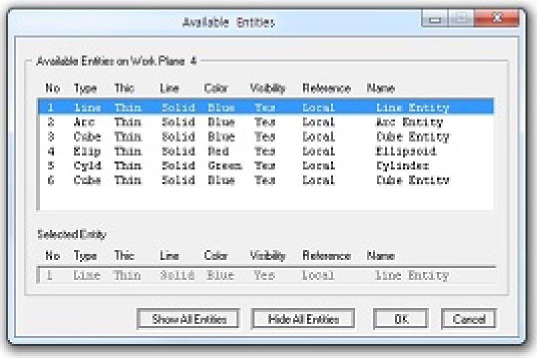


Figure 6.18
Entity list dialog

Show Entity No

This is used to show all entity numbers on the screen.

Reset To Global

This is used to reset the current entity global reference by the current local coordinate.

Update

This is used to update parameters of the current entity.

Edit

This is used to edit the geometry of the current entity.

Add

This is used to add new entity. Refer to Section 6.4.9

Delete

This is used to delete the current entity.

Exit

This is used to exit from the Entity Editor dialog.

6.4.8 Popup Menu for Entity

When Entity Editor dialog is opened, you can directly access an entity by Control + Right Click. Then the selected entity is displayed on the Entity Editor dialog along with Popup Menu as shown in Figure 6.19.

Popup Menu consists of eight submenus:
Edit, Copy, Add, Hide, Delete, List, Number and Exit.
These menus are essentially duplicates of command buttons on the Entity Editor dialog.

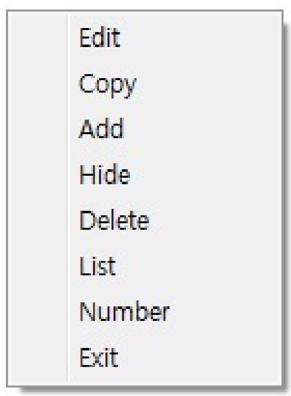


Figure 6.19 Popup menu for entity

6.4.9 Adding New Entity

To add a new entity, click Add button on Entity Editor dialog. Then Entity Type Selection dialog will be displayed as shown in Figure 6.20.

There are five types of entities: Line, Arc, Cube, Ellipsoid and Cylinder. You can also select Copy Existing Entity and then type Entity No.

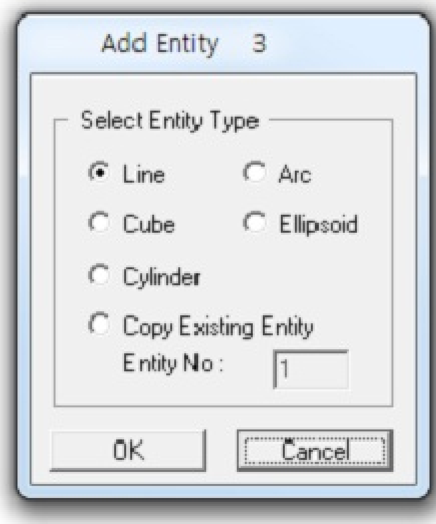


Figure 6.20 Entity type selection dialog

6.4.9.1 Line Entity

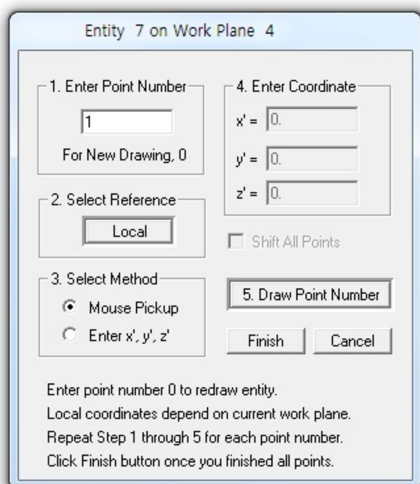
Line Entity dialog is shown in Figure 6.21.

To draw Line Entity, follow five steps:
1. Enter Point Number
2. Select Reference
3. Select Method
4. Enter Coordinate
5. Draw Point Number

For Mouse Pickup method, when clicking Draw Point Number button at step 5, Coordinates on Work Plane dialog in Figure 6.22 will be opened. Click Info button to see the notes on Mouse Actions on Work Plane as shown in Figure 6.23. Once finished, click Finish in Figure 6.22.

Finally, click Finish on Line Entity dialog in Figure 6.21.
Then you will be back to Entity Editor dialog where you can set the other parameters for the new entity.

Figure 6.21
Line entity dialog



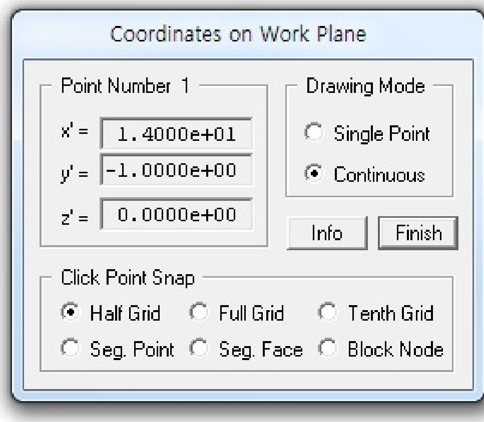


Figure 6.22 Coordinates on work plane

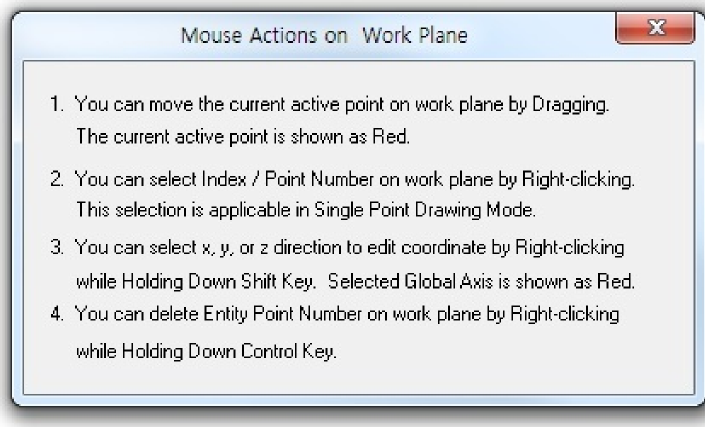


Figure 6.23 Mouse actions on work plane

6.4.9.2 Arc Entity

Arc Entity dialog is shown in Figure 6.24.

To draw Arc Entity, follow five steps:
1. Select Reference
2. Select Method
3. Enter Origin
4. Enter Dimensions
5. Draw Arc Entity

For Mouse Pickup method, when clicking Draw Arc Entity button at step 5, Coordinates on Work Plane dialog in Figure 6.22 will be opened. Click Info button to see the notes on Mouse Actions on Work Plane as shown in Figure 6.23. Once finished, click Finish in Figure 6.22.

Finally, click Finish on Arc Entity dialog in Figure 6.24. Then you will be back to Entity Editor dialog where you can set the other parameters for the new entity.

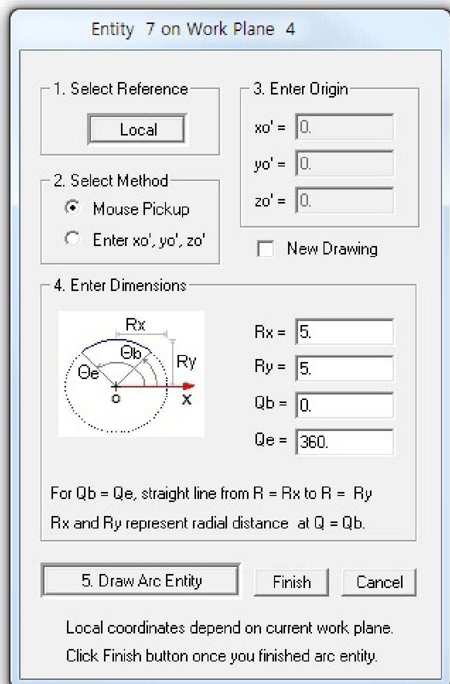


Figure 6.24
Arc entity dialog

6.4.9.3 Cube Entity

Cube Entity dialog is shown in Figure 6.25.

To draw Cube Entity, follow five steps:

1. Select Reference
2. Select Method
3. Enter Origin
4. Enter Dimensions
5. Draw Cube Entity

For Mouse Pickup method, when clicking Draw Cube Entity button at step 5, Coordinates on Work Plane dialog in Figure 6.22 will be opened. Click Info button to see the notes on Mouse Actions on Work Plane as shown in Figure 6.23. Once finished, click Finish in Figure 6.22.

Finally, click Finish on Cube Entity dialog in Figure 6.25. Then you will be back to Entity Editor dialog where you can set the other parameters for the new entity.

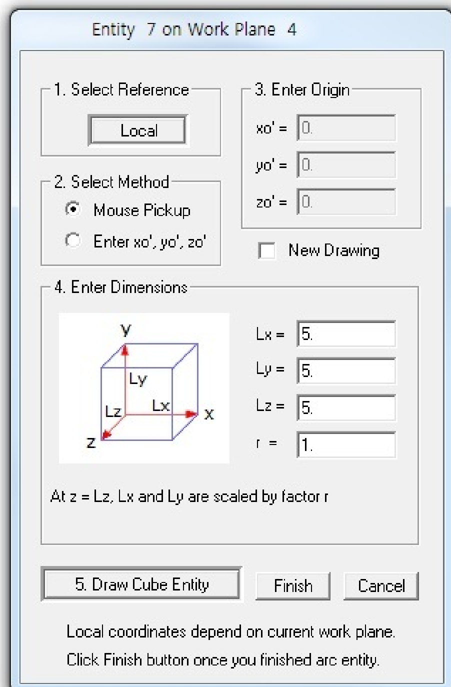


Figure 6.25
Cube entity dialog

6.4.9.4 Ellipsoid Entity

Ellipsoid Entity dialog is shown in Figure 6.26.

To draw Ellipsoid Entity, follow five steps:

1. Select Reference
2. Select Method
3. Enter Origin
4. Enter Dimensions
5. Draw Ellipsoid Entity

For Mouse Pickup method, when clicking Draw Ellipsoid Entity button at step 5, Coordinates on Work Plane dialog in Figure 6.22 will be opened. Click Info button to see the notes on Mouse Actions on Work Plane as in Figure 6.23. Once finished, click Finish in Figure 6.22.

Finally, click Finish on Ellipsoid Entity dialog in Figure 6.26. Then you will be back to Entity Editor dialog where you can set the other parameters for the new entity.

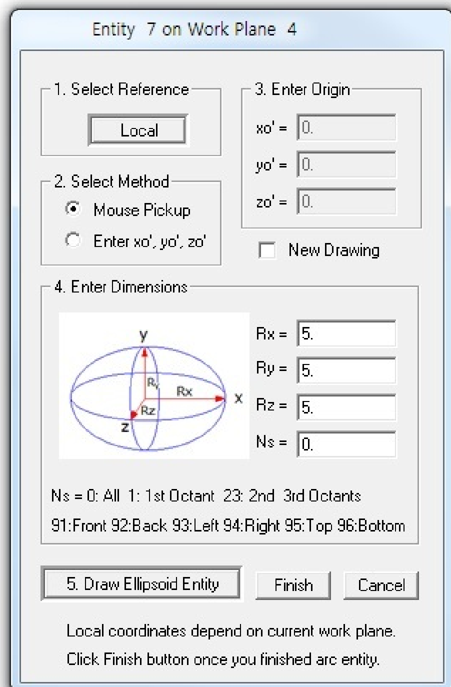


Figure 6.26
Ellipsoid entity dialog

6.4.9.5 Cylinder Entity

Cylinder Entity dialog is shown in Figure 6.27.

To draw Cylinder Entity, follow five steps:

1. Select Reference
2. Select Method
3. Enter Origin
4. Enter Dimensions
5. Draw Cylinder Entity

For Mouse Pickup method, when clicking Draw Cylinder Entity button at step 5, Coordinates on Work Plane dialog in Figure 6.22 will be opened. Click Info button to see the notes on Mouse Actions on Work Plane as in Figure 6.23. Once finished, click Finish in Figure 6.22.

Finally, click Finish on Cylinder Entity dialog in Figure 6.27. Then you will be back to Entity Editor dialog where you can set the other parameters for the new entity.

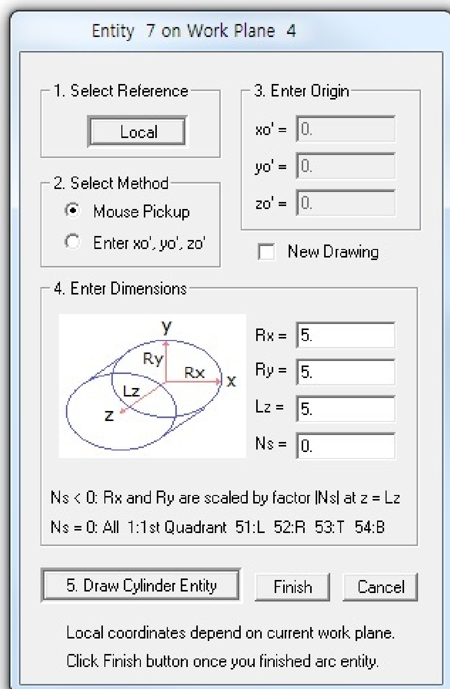


Figure 6.27
Cylinder entity dialog

6.5 Block

Blocks are groups of elements. Each block consist of the same type of finite elements.

Block Editor can be accessed by selecting the following menu items in PLOT-3D:
Model → Block Editor
or by clicking Block Editor toolbar as shown in Figure 6.28.

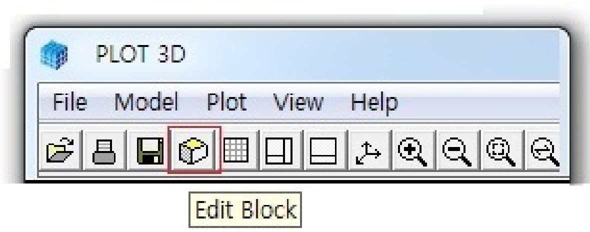


Figure 6.28 Block editor toolbar

Block Editor dialog in Figure 6.29 consists of following eight parts:

- Title
- Block Number
- Interpolation Coordinate System
- Coordinate Modification
- Interpolation Scheme / Element Type
- Reference Node Numbers
- Material and Element Generation Parameters
- Command Buttons

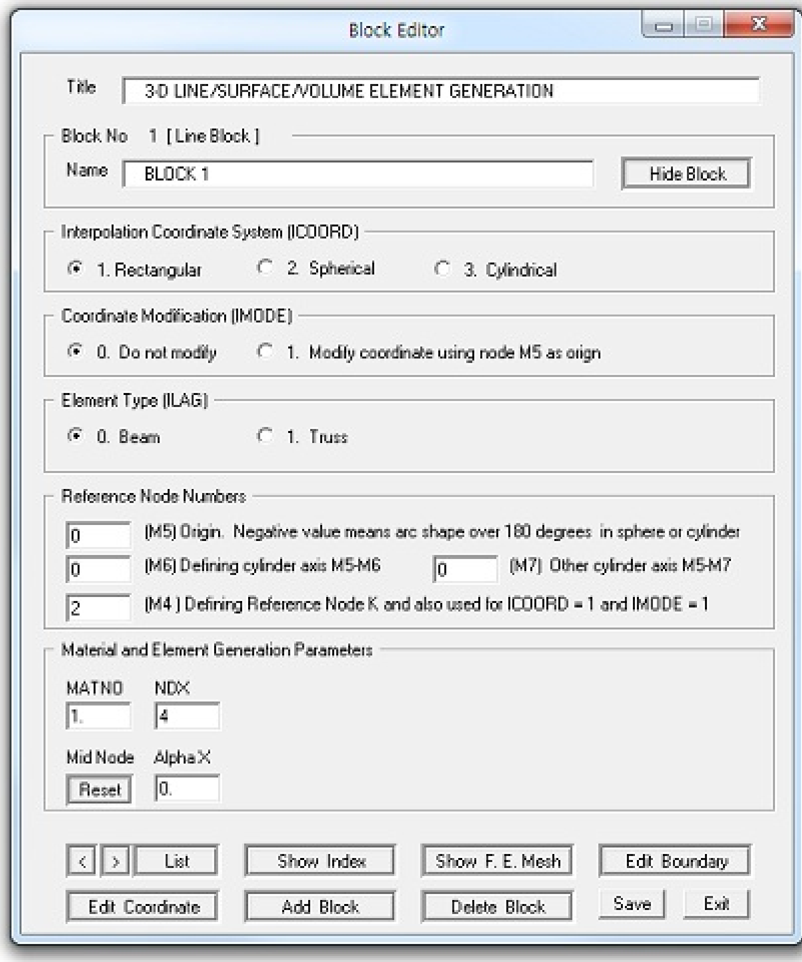


Figure 6.29 Block editor (Line Block)

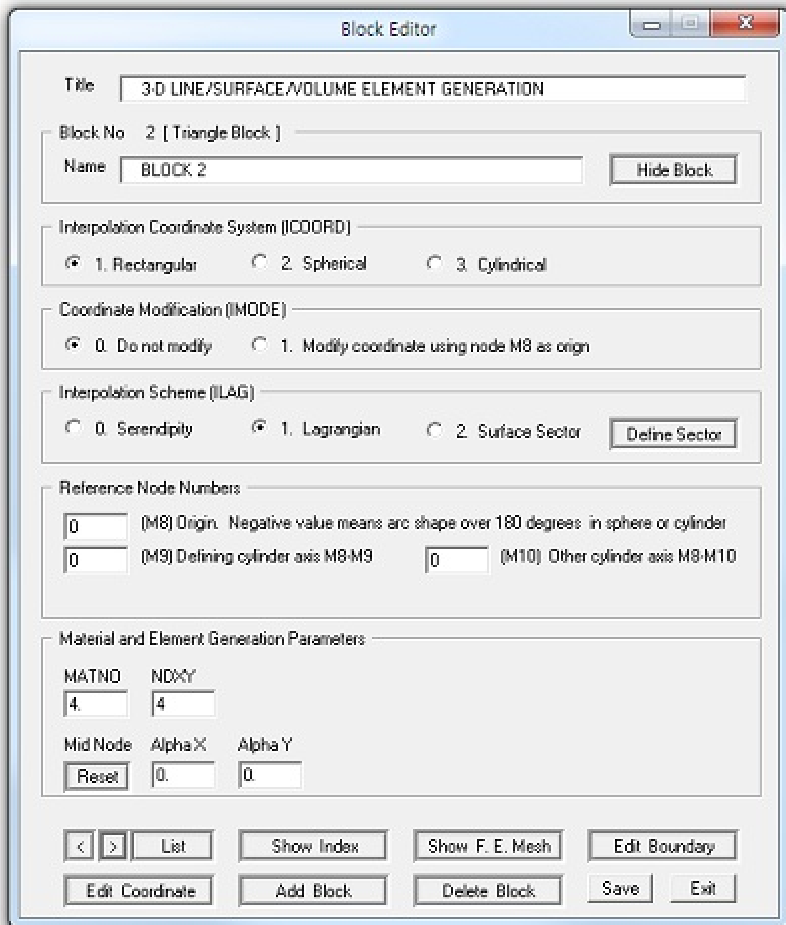


Figure 6.29 Block editor (Triangle Block)

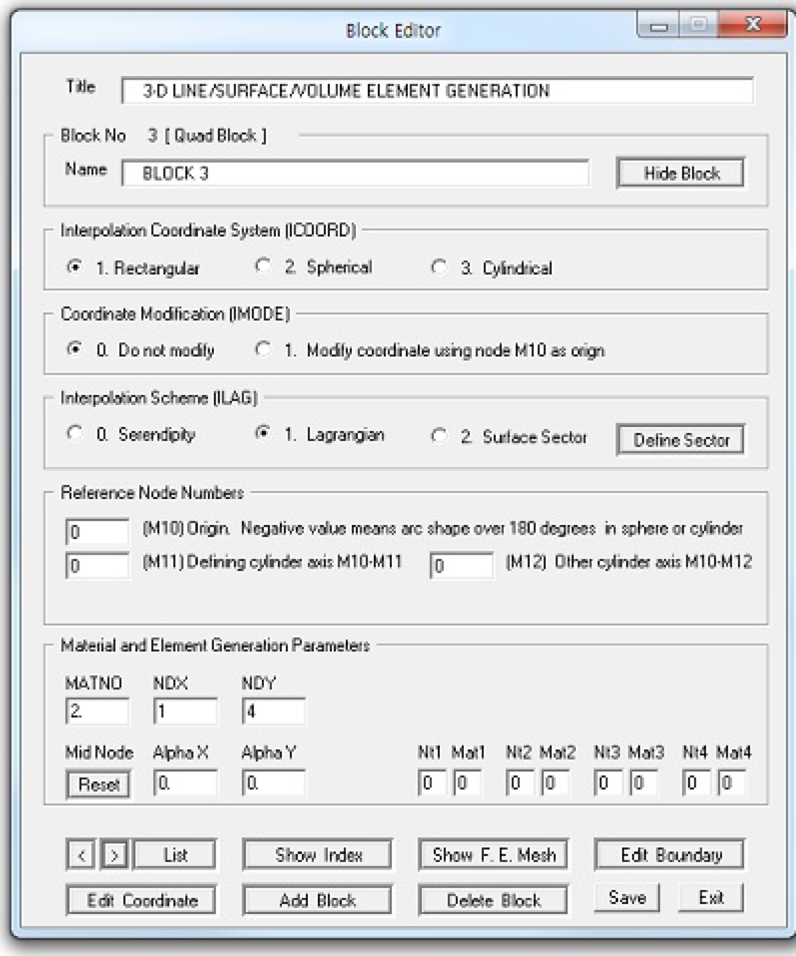


Figure 6.29 Block editor (Quad Block)

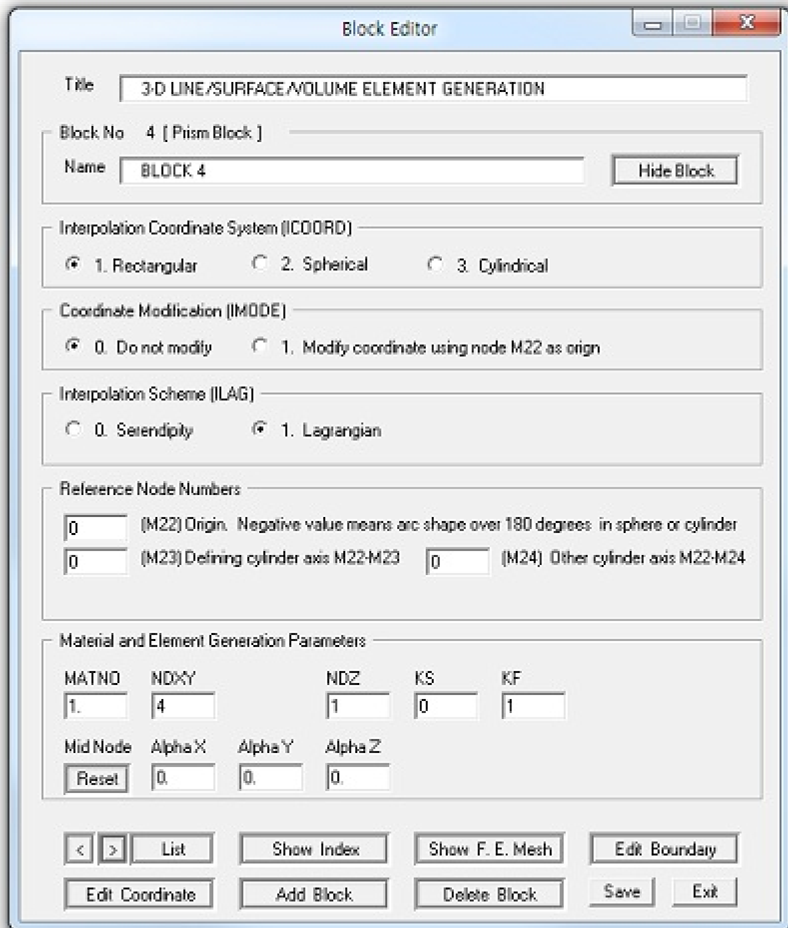


Figure 6.29 Block editor (Prism Block)

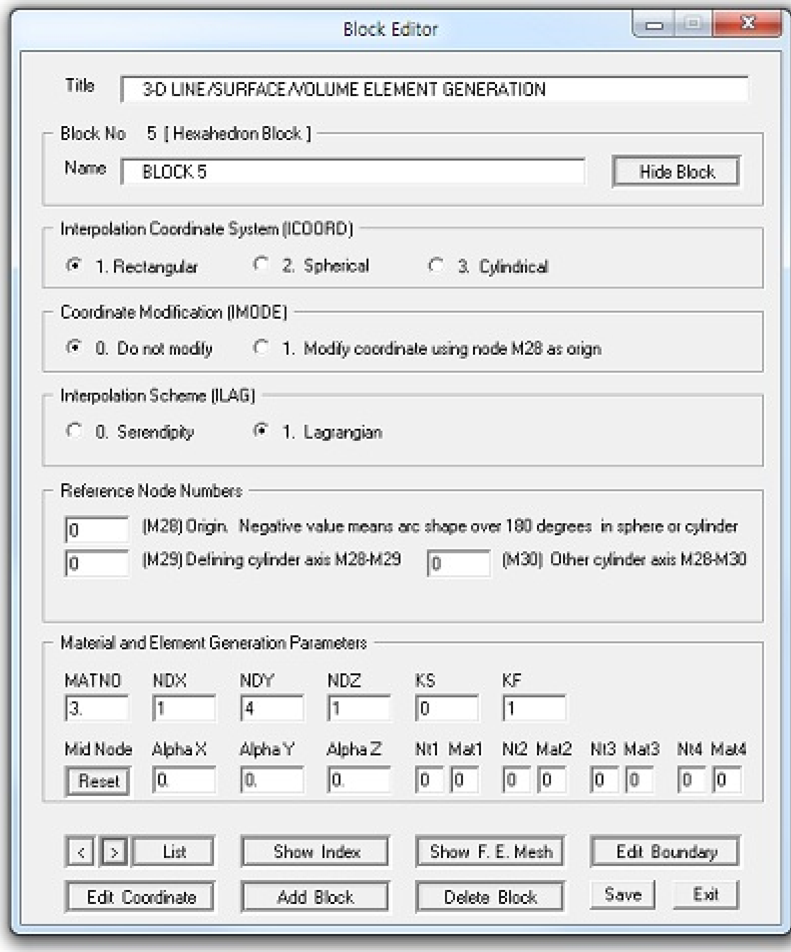


Figure 6.29 Block editor (Hexahedron Block)

6.5.1 Title

This is the title for the block mesh file.

6.5.2 Block Number

Block number and type are automatically displayed as the label of the frame. You can specify block name for identification.

Hide Block button is to hide the current block on the screen.

6.5.3 Interpolation Coordinate System

This is to select the coordinate system for interpolation. Three options are available: Rectangular, Spherical and Cylindrical.

6.5.4 Coordinate Modification

This is to modify generated coordinates based on the reference node as origin.

6.5.5 Interpolation Scheme / Element Type

For line blocks, two options are available for the type of line element: Beam and Truss.

For surface blocks, three options are available: Serendipity, Lagrangian and Surface Sector.

For volume blocks, two options are available: Serendipity and Lagrangian.

When you click Define Sector button, Surface Sector dialog is displayed to edit input parameters as shown in Figure 6.30.

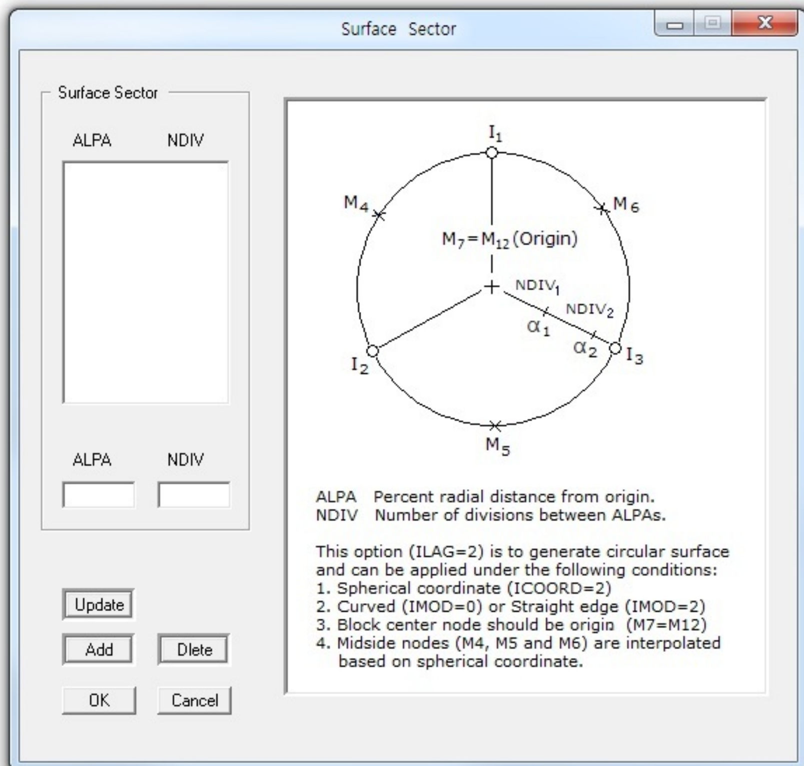


Figure 6.30 Surface sector (Triangle Block)

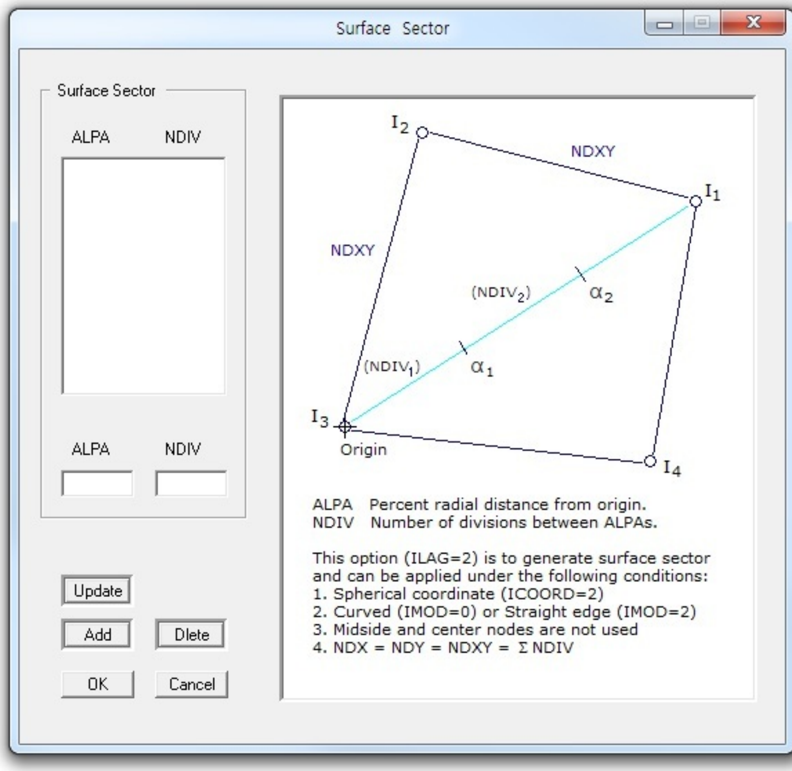


Figure 6.30 Surface sector (Quad Block)

6.5.6 Reference Node Numbers

This is to specify reference node numbers which are associated with block type.

6.5.7 Material & Element Generation Parameters

This is to specify material number and element generation parameters for the block.

6.5.8 Command Buttons

Command buttons are shown on the bottom of Block Editor dialog in Figure 6.29.

List

This is used to list all available blocks in the current block mesh as shown in Figure 6.31.

When you click OK button, selected block will be displayed as the current block on the Block Editor dialog.

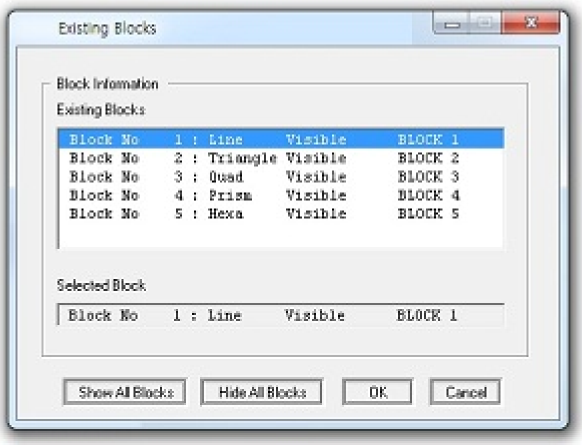


Figure 6.31 Block list

Show Index

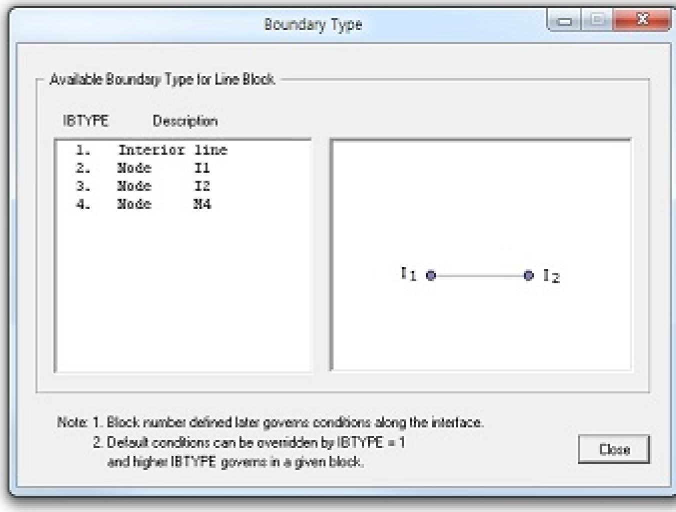
This is used to show block index numbers.

Show F. E. Mesh

This is used to execute block mesh and then plot the generated finite element mesh.

Edit Boundary

This is used to edit boundary conditions shown in Figure 6.32.



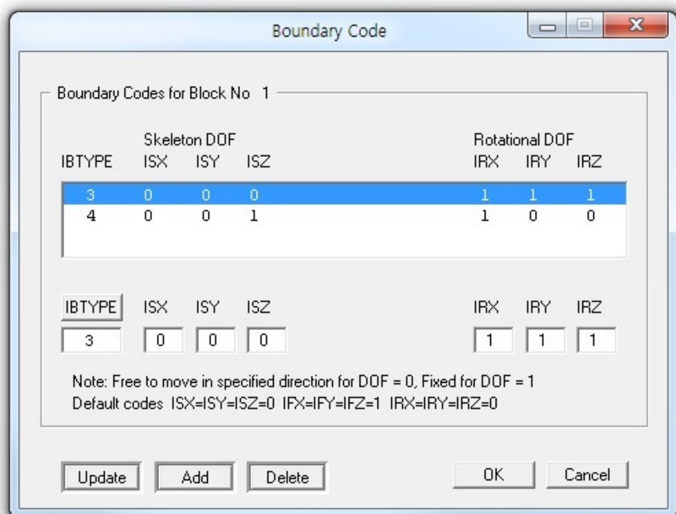


Figure 6.32 Boundary code (Line Block)

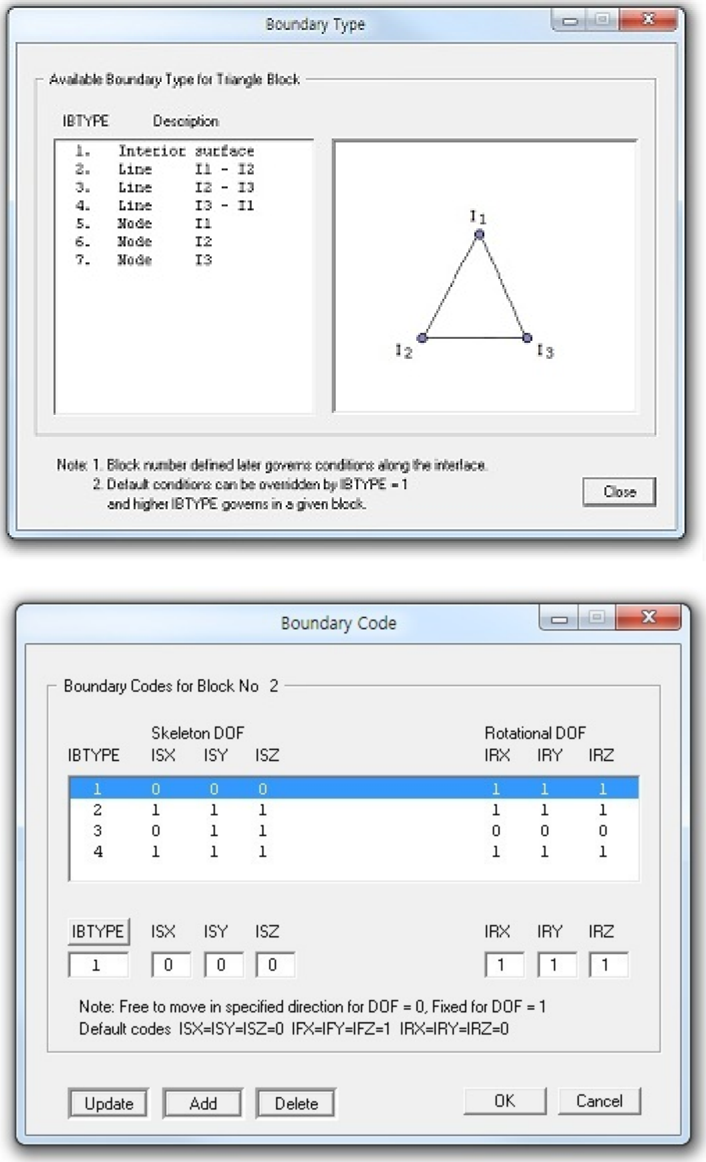
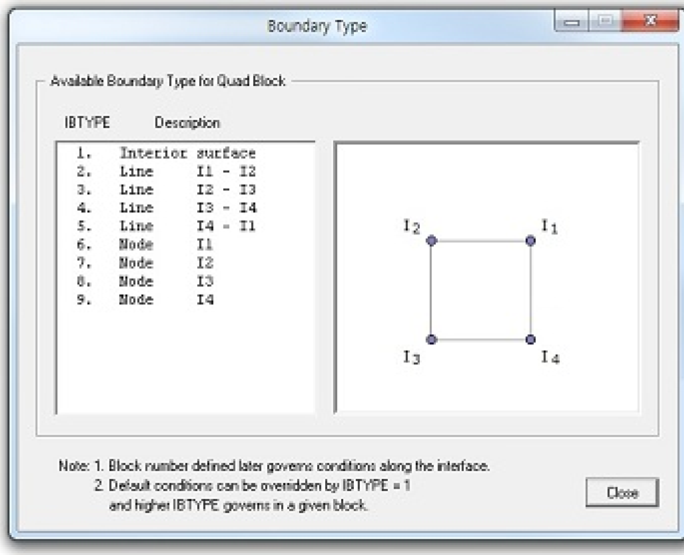


Figure 6.32 Boundary code (Triangle Block)



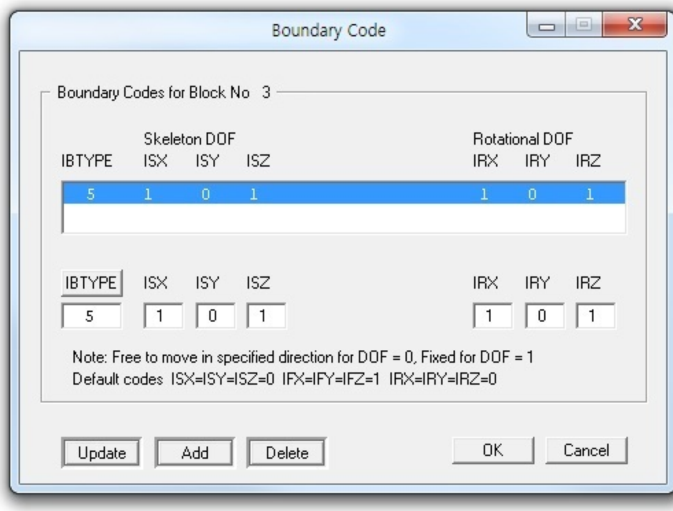


Figure 6.32 Boundary code (Quad Block)

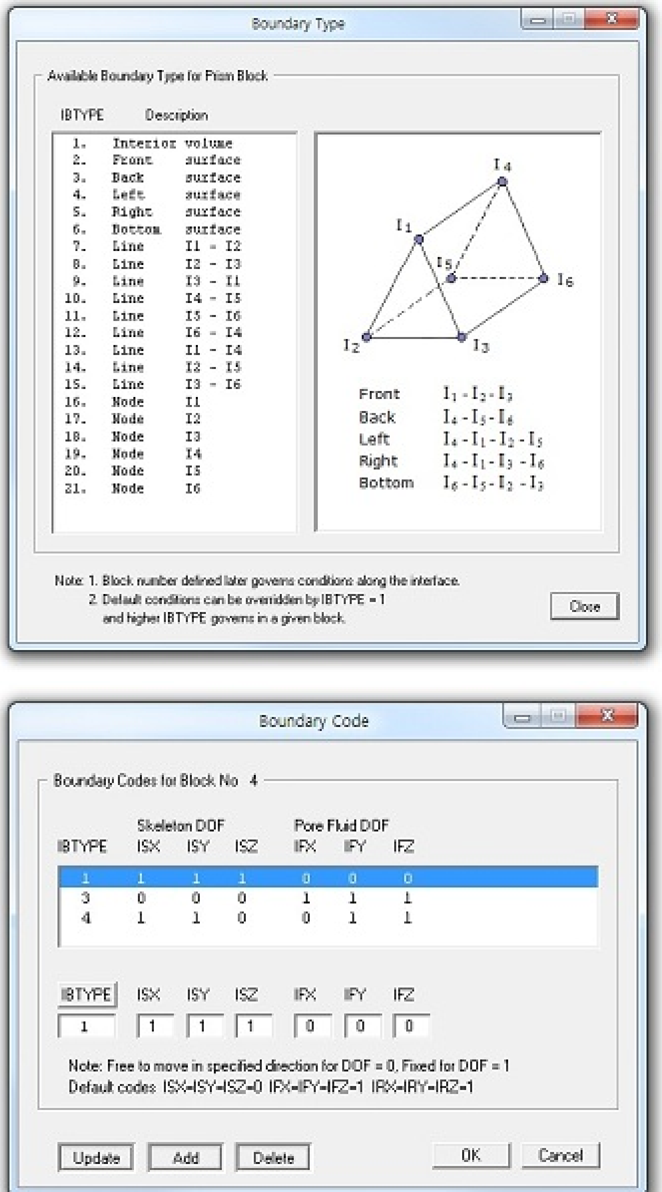


Figure 6.32 Boundary code (Prism Block)

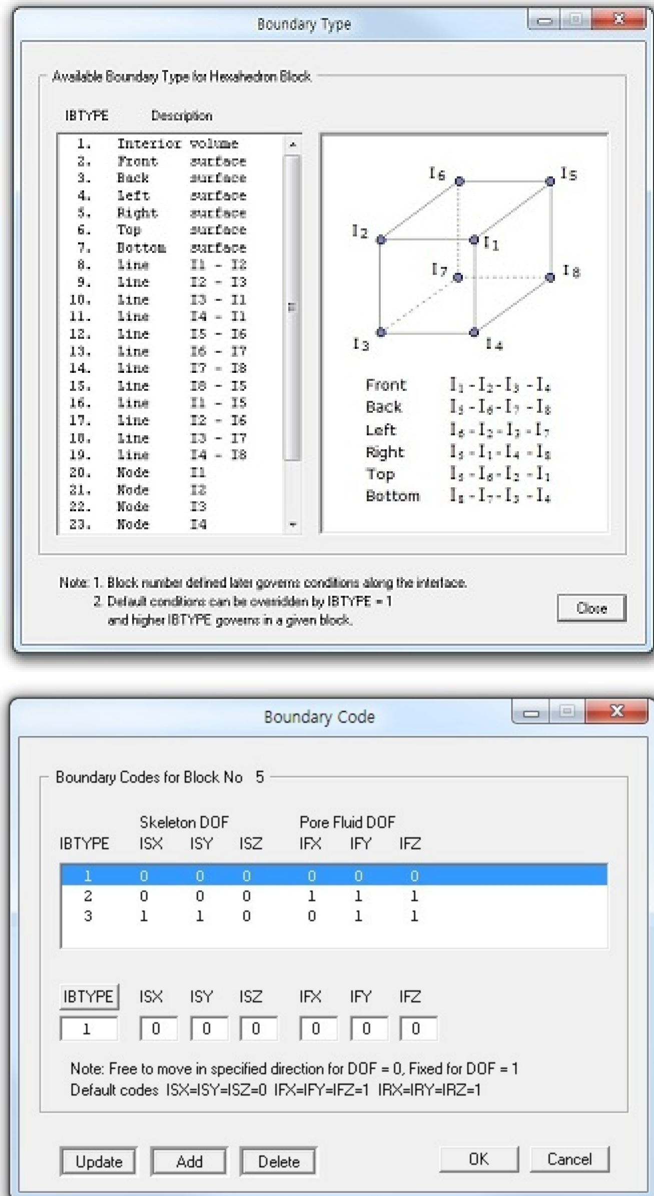


Figure 6.32 Boundary code (Hexahedron Block)

Edit Coordinate

This is used to edit the geometry of the block.
Before editing, work plane should be displayed on the screen.

Type Block No on Edit Current Block dialog in Figure 6.33
and then click OK button.

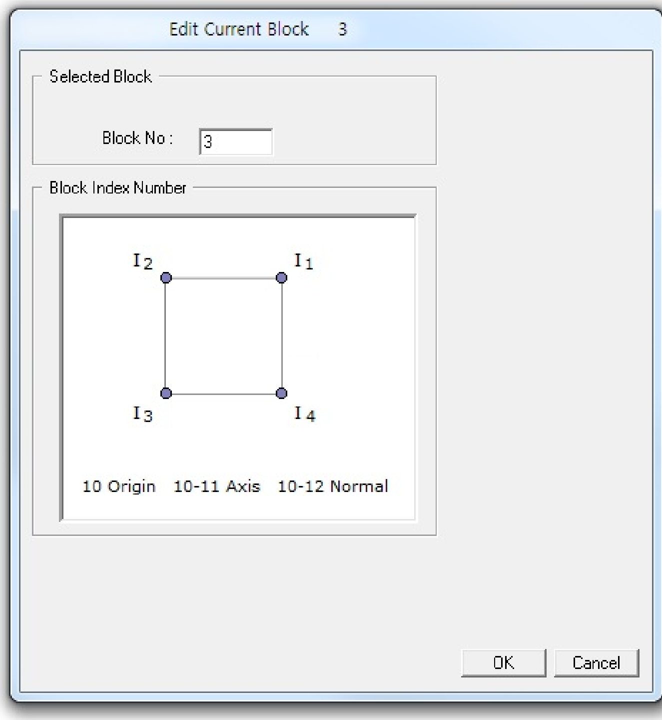


Figure 6.33 Edit current block (Selection Mode)

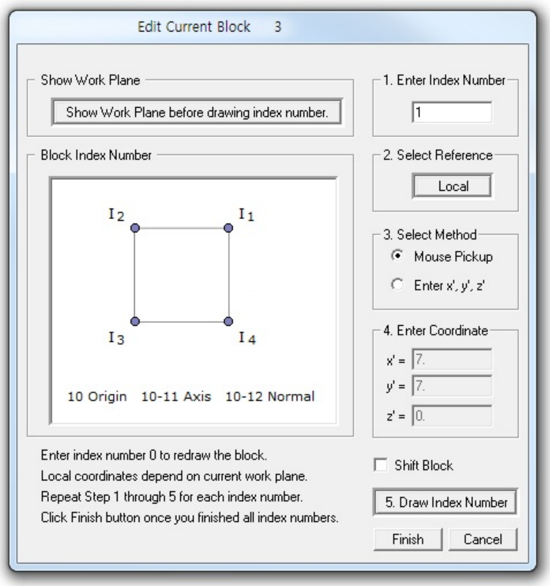
Edit Current Block dialog now shows input parameters required
to edit the geometry of the block as shown in Figure 6.34.

To edit block, follow five steps:
1. Enter Index Number
2. Select Reference
3. Select Method
4. Enter Coordinate
5. Draw Index Number

For Mouse Pickup method, when clicking Draw Index Number button at step 5, Coordinates on Work Plane dialog in Figure 6.35 will be opened. Click Info button to see the notes on Mouse Actions on Work Plane as shown in Figure 6.36. Once finished, click Finish in Figure 6.35.

Finally, click Finish on Edit Current Block dialog in Figure 6.34.
Then you will be back to Block Editor dialog where you can set the other parameters for the current block.

Figure 6.34
Edit current block
(Edit Mode)



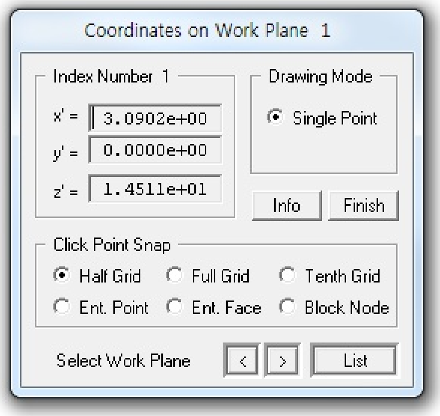


Figure 6.35 Coordinates on work plane

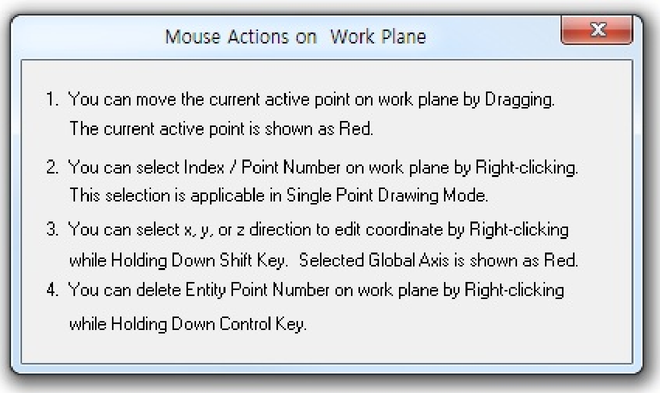


Figure 6.36 Mouse actions on work plane

Add Block

This is used to add the geometry of the new block.
Before building, work plane should be displayed on the screen.

Build New Block dialog in Figure 6.37 will be displayed.
Select Block Type, Interpolation Coordinate System
and then click OK button.

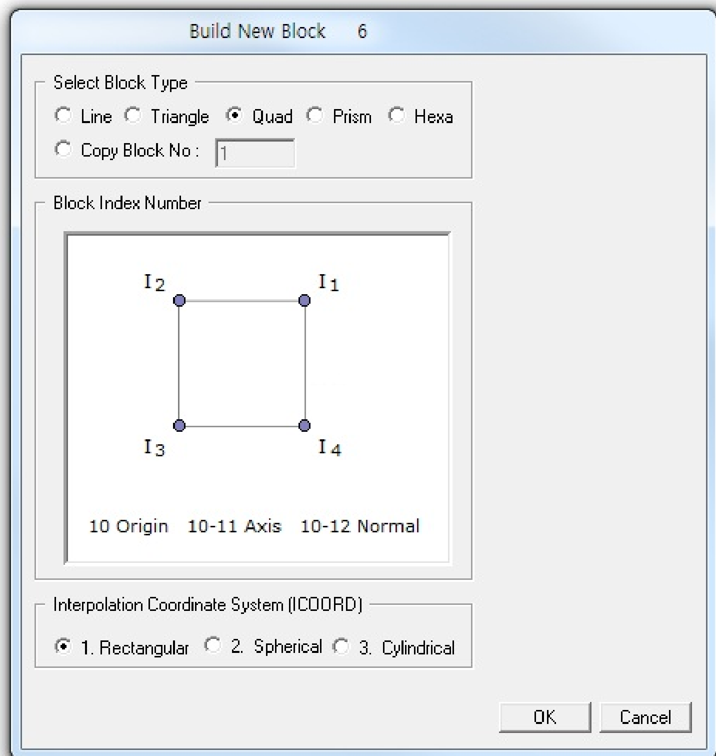


Figure 6.37 Build new block (Selection Mode)

Build New Block dialog now shows input parameters required to build the geometry of new block as shown in Figure 6.38.

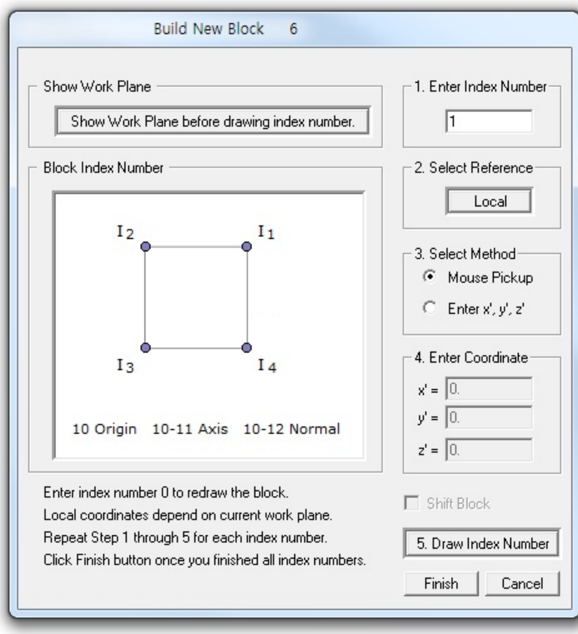
To build new block, follow five steps:

1. Enter Index Number
2. Select Reference
3. Select Method
4. Enter Coordinate
5. Draw Index Number

For Mouse Pickup method, when clicking Draw Index Number button at step 5, Coordinates on Work Plane dialog in Figure 6.39 will be opened. Click Info button to see the notes on Mouse Actions on Work Plane as shown in Figure 6.36. Once finished, click Finish in Figure 6.39.

Finally, click Finish on Build New Block dialog in Figure 6.38. Then you will be back to Block Editor dialog where you can set the other parameters for the new block.

Figure 6.38
Build new block
(Build Mode)



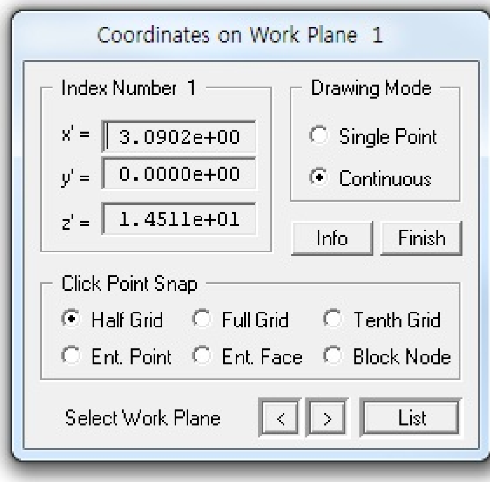


Figure 6.39 Coordinates on work plane

Delete Block

This is used to delete the current block.

Save

This is used to save all the works you have done.

Exit

This is used to exit from the block editor.

6.5.9 Popup Menu for Block

When Block Editor dialog is opened, you can directly access a block by Shift + Right Click. Then the selected block is displayed on the Block Editor dialog along with Popup Menu as shown in Figure 6.40.

Popup Menu consists of eleven submenus:
Edit, Copy, Add, Hide, Delete, List, Index, Boundary, F.E. Mesh, Save and Exit. These menus are essentially duplicates of command buttons on the Block Editor dialog.

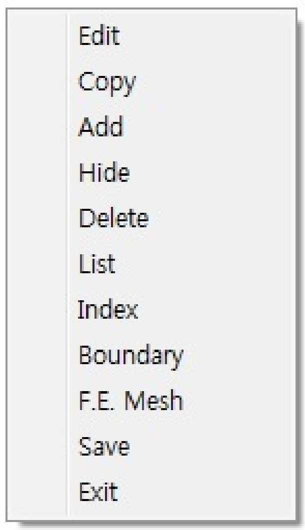


Figure 6.40 Popup menu for block

6.6 Modifying Finite Element Meshes

Block Mesh Generator can be used to directly modify finite element mesh.

When you open input file, Mesh Generator reads the format of the input file and automatically identifies whether it is block mesh file or finite element mesh file.

Editing finite element mesh has three parts: Edit Element, Edit Node and Edit Title. These editing modes can be accessed from Model menu in PLOT-3D as shown in Figure 6.41.

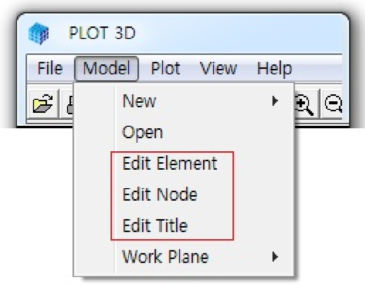


Figure 6.41 Menu for editing finite element mesh

You can check the current editing mode by moving the mouse on Editing Mode toolbar as shown in Figure 6.42.

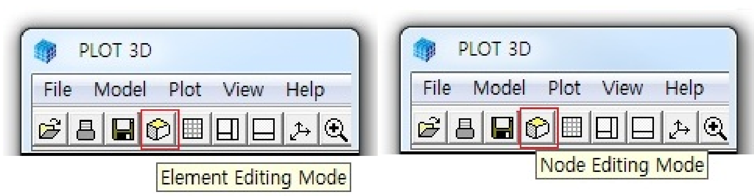
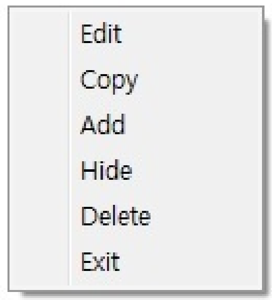


Figure 6.42 Toolbar for editing finite element mesh

6.6.1 Edit Element

When you are in Element Editing Mode, you can access popup menu for element in Figure 6.43 by Shift + Right Click.

Figure 6.43 Popup menu for element

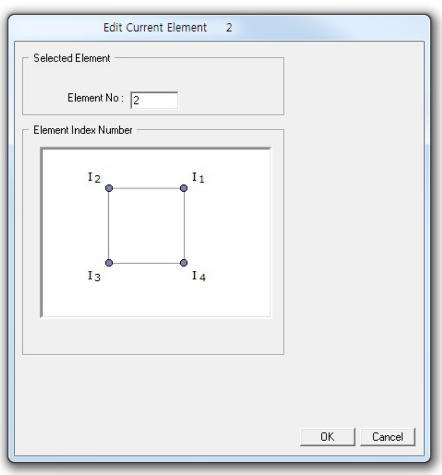


Element popup menu consists of six submenus:
Edit, Copy, Add, Hide, Delete and Exit.

Edit
This is used to edit the geometry of element.
Before editing, work plane should be displayed on the screen.

Edit Current Element dialog is displayed in Figure 6.44.
Type Element No and click OK button.

Figure 6.44
Edit current element
(Selection Mode)



Edit Current Element dialog now shows input parameters required to edit the geometry of element as shown in Figure 6.45.

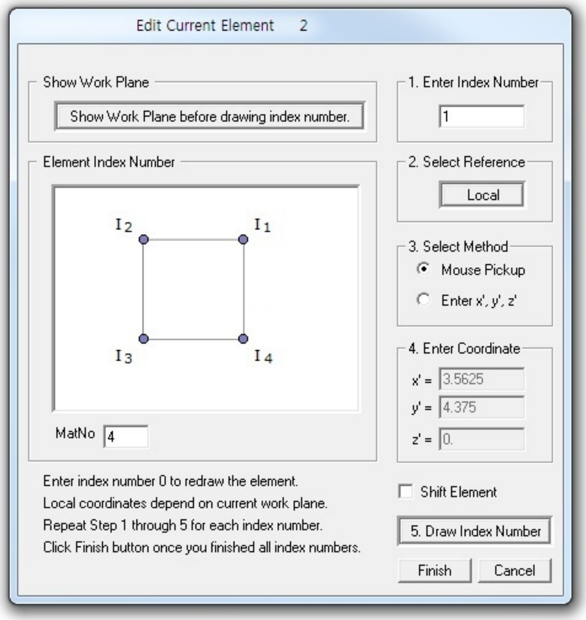
To edit element, follow five steps:

1. Enter Index Number
2. Select Reference
3. Select Method
4. Enter Coordinate
5. Draw Index Number

For Mouse Pickup method, when clicking Draw Index Number button at step 5, Coordinates on Work Plane dialog in Figure 6.46 will be opened. Click Info button to see the notes on Mouse Actions on Work Plane as shown in Figure 6.47.   Once finished, click Finish in Figure 6.46.

Finally, click Finish on Edit Current Element dialog in Figure 6.45.

Figure 6.45
Edit current element
(Edit Mode)



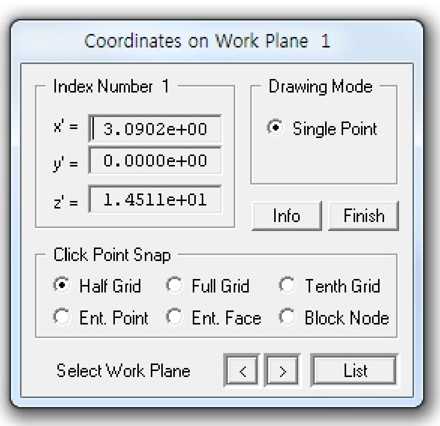


Figure 6.46 Coordinates on work plane (Edit Mode)

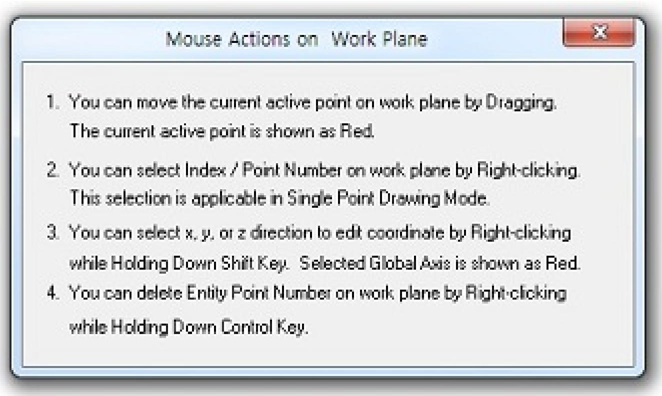


Figure 6.47 Mouse actions on work plane

Copy

This is used to copy the selected element and paste it as new element.

Edit Current Element dialog with new element number is displayed as shown in Figure 6.48. Shift Element check box should be checked to move this new element.

Follow the same procedure as in Edit.

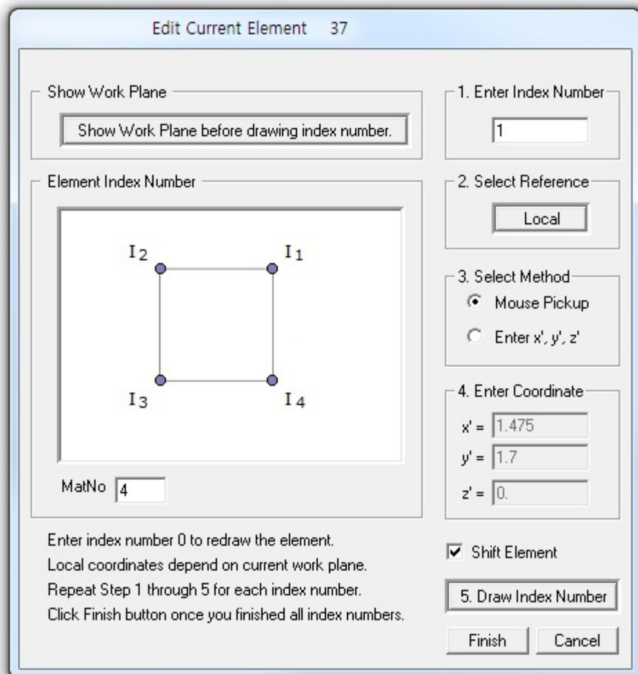


Figure 6.48 Edit current element (Copy Mode)

Add

This is used to add the geometry of the new element.
Before building, work plane should be displayed on the screen.

Build New Element dialog in Figure 6.49 will be displayed.
Select Element Type and then click OK button.

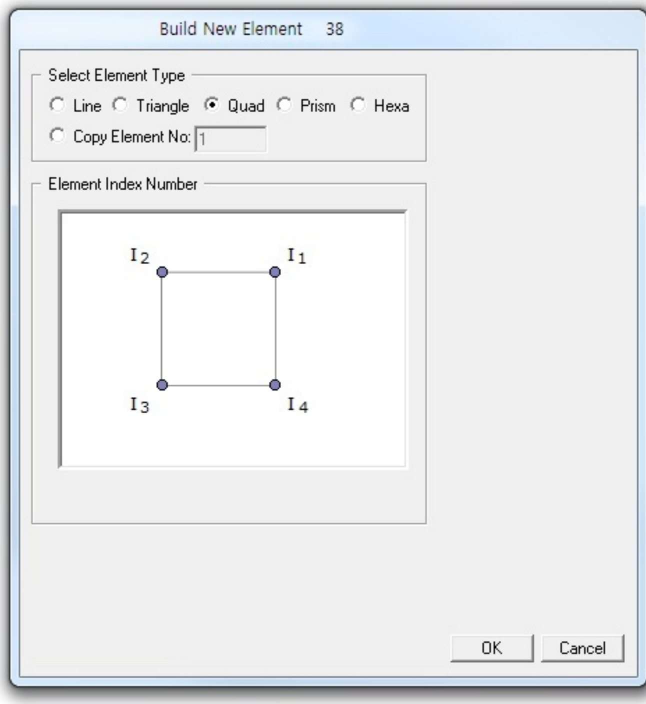


Figure 6.49 Build new element (Selection Mode)

Build New Element dialog now shows input parameters required to build the geometry of new element as shown in Figure 6.50.

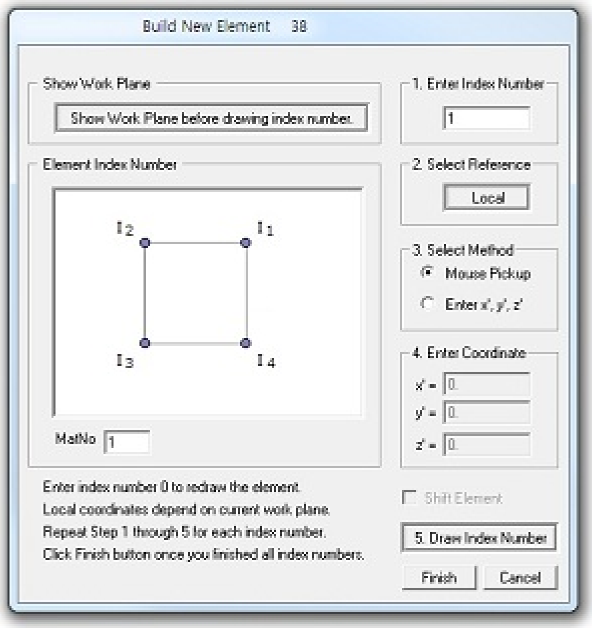
To build new element, follow five steps:

1. Enter Index Number
2. Select Reference
3. Select Method
4. Enter Coordinate
5. Draw Index Number

For Mouse Pickup method, when clicking Draw Index Number button at step 5, Coordinates on Work Plane dialog in Figure 6.51 will be opened. Click Info button to see the notes on Mouse Actions on Work Plane as shown in Figure 6.47.   Once finished, click Finish in Figure 6.51.

Finally, click Finish on Build New Element dialog in Figure 6.50.

Figure 6.50
Build new element
(Edit Mode)



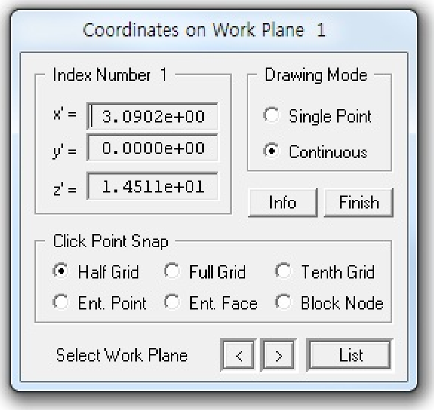


Figure 6.51 Coordinates on work plane (Add Mode)

Hide
This is used to hide the selected element from the screen.
To show the hidden element, follow instructions in Figure 6.52.

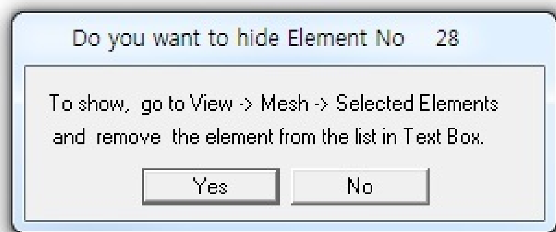


Figure 6.52 Instructions to show the hidden element

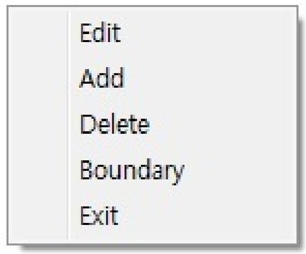
Delete
This is used to delete the selected element.

Exit
This is used to exit from the element editing mode.

6.6.2 Edit Node

When you are in Node Editing Mode, you can access popup menu for node in Figure 6.53 by Shift + Right Click.

Figure 6.53 Popup menu for node



Node popup menu consists of five submenus:
Edit, Add, Delete, Boundary and Exit.

Edit

This is used to edit the coordinates of node.
Before editing, work plane should be displayed on the screen.

Edit Current Node dialog is displayed in Figure 6.54.

To edit current node, follow five steps:

1. Enter Node Number
2. Select Reference
3. Select Method
4. Enter Coordinate
5. Draw Node Number

For Mouse Pickup method, when clicking Draw Node Number button at step 5, Coordinates on Work Plane dialog in Figure 6.55 will be opened. Click Info button to see the notes on Mouse Actions on Work Plane as shown in Figure 6.47. Once finished, click Finish in Figure 6.55.

Finally, click Finish on Edit Current Node dialog in Figure 6.54.

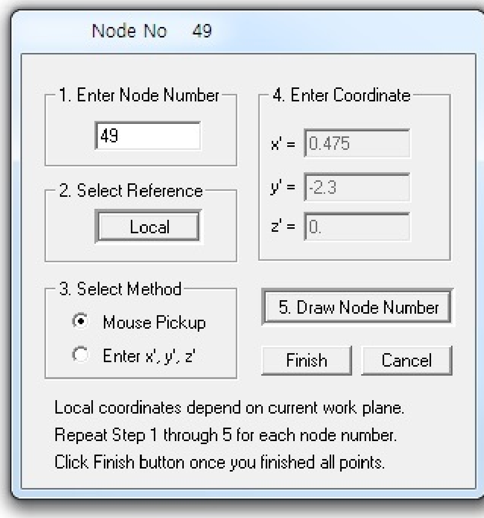


Figure 6.54 Edit current node dialog

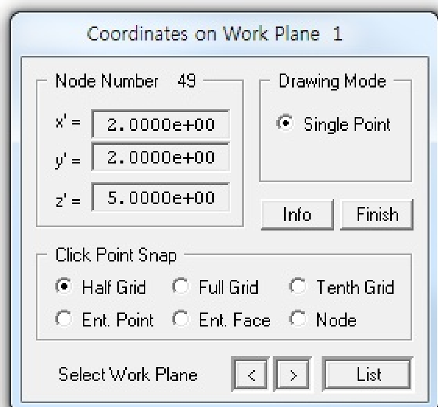


Figure 6.55 Coordinates on work plane (Edit Node)

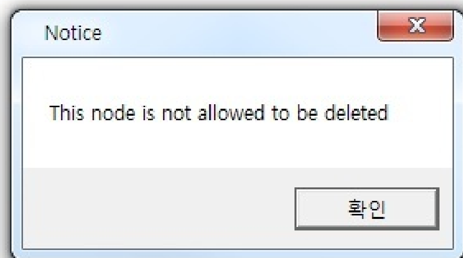
Add
This is used to add new node.
Before adding, work plane should be displayed on the screen.

Build New Node dialog similar to Figure 6.54 is displayed.
Follow the same procedure as in Edit.

Delete
This is used to delete the selected node.

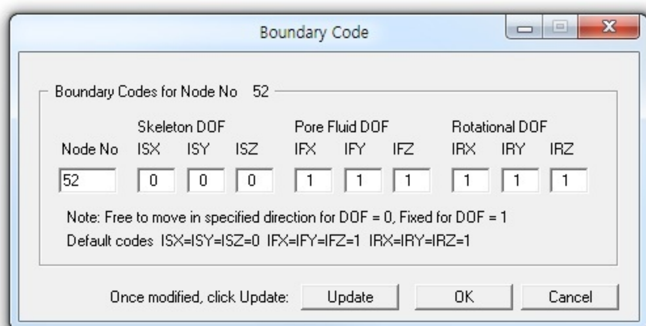
You can delete only standalone nodes which are not connected to elements. Refer to the notice in Figure 6.56.

Figure 6.56
Notice on deleting connected nodes



Boundary
This is used to edit boundary codes associated with the current selected node as shown in Figure 6.57.

Figure 6.57
Boundary editor dialog



Exit
This is used to exit from the node editing mode.

6.6.3 Edit Title

This is used to edit the title of the finite element mesh file as shown in Figure 6.58.

Figure 6.58 Mesh title editor dialog

PRESMAP
User's Manual

7.1 Introduction

PRESMAP programs are mainly used to model the geometry of the structures to be analyzed. Mesh File described in Section 4.3 can be created using PRESMAP programs.

Seven PRESMAP programs are provided in this manual; PRESMAP-2D, NATM-2D, CIRCLE-2D, PRESMAP-3D, CROSS-3D, GEN-3D, PILE-3D, PRESMAP-GP, JOINT-3D. and INTERSECTION.

PRESMAP-2D includes Model 1, 2, 3, and 4. Model 1 is basic pre-processor which can be applied to model various types of problem geometry. Model 2 is the special pre-processor developed to model near-fields around underground openings such as tunnels, culverts, etc. Model 3 is the special pre-processor developed to model triangular and rectangular shape geometry. Model 4 is the useful pre-processor to generate layered embankments having slope.

NATM-2D is the special pre-processing program developed to generate automatically two-dimensional finite element meshes and boundary conditions for NATM (New Austrian Tunneling Method) tunnels.

CIRCLE-2D is the special pre-processing program developed to generate automatically two-dimensional finite element meshes for circular cross section with joint interface.

PRESMAP-3D is the basic pre-processor which can be applied to model various types of three dimensional geometries.

CROSS-3D is the special pre-processing program developed to generate automatically three dimensional finite element meshes and boundary conditions for crossing tunnels. The intermediate output file with file extension *.TMP* from CROSS-3D contains finite element block coordinates, indexes, and boundary conditions which are essentially input data to PRESMAP-3D.

GEN-3D generates coordinates, element indexes, boundary codes, external loads, and transmitting boundaries in three dimensional coordinate system by extending typical two dimensional output files from PRESMAP-2D, NATM-2D or CIRCLE-2D.

PILE-3D is the special pre-processor which can be used to generate all input files required for pile foundation analysis. It can generate Concrete Pile with Anchor Bolts or Steel Pipe with Concrete Cap.

PRESMAP-GP is the general purpose pre-processing program which can be used to generate coordinates, element indexes and boundary codes for truss, beam, shell or continuum elements. Users can select rectangular, spherical or cylindrical coordinate for interpolation.

JOINT-3D is the special pre-processor which can be used to generate jointed continuum finite element meshes given the conventional continuum SMAP-3D Mesh File input. For the jointed continuum analysis, each continuum finite element is surrounded by joint elements which allow slippage along the joint when reaching shear strength and debonding normal to the joint face when exceeding tensile strength.

INTERSECTION programs are mainly used to compute the locations of the 3D surfaces crossing each other. These surfaces consist of Shell Elements with different materials. The computed coordinates of intersections can be used for the construction of complicated three-dimensional meshes. Two methods are available: Shell Element and Two Tunnels.

PRESMAP-2D

Model 1

User's Manual

<table>
<tr><th>Card Group</th><th>Input Data and Definitions (Model 1)</th></tr>
<tr><td rowspan="3">1
General Information</td><td>1.1
TITLE
TITLE — Any title of (Max = 60 characters)</td></tr>
<tr><td>1.2
IP
IP = 0 — Plane strain or plane stress
= 1 — Axisymmetry</td></tr>
<tr><td>1.3
NBLOCK, NBNODE, NSNEL, CMFAC (SMAP-S2/2D)
NBLOCK, NBNODE, NSNEL, CMFAC, TEMPI (SMAP-T2)

See Figure 7.1

NBLOCK — Number of blocks
NBNODE — Number of block nodes
NSNEL — Starting element number
CMFAC — Coordinate magnification factor
TEMPI — Initial temperature</td></tr>
</table>

Card Group	Input Data and Definitions (Model 1)
1 General Information	1.4 NBX, NBY, MIDX, MIDY, NF, NSNODE See Figure 7.2 NBX — Number of blocks in x-direction NBY — Number of blocks in y-direction MIDX = 0 — Element has no side nodes in x-direction = 1 — Element has side nodes in x-direction MIDY = 0 — Element has no side nodes in y-direction = 1 — Element has side nodes in y-direction NF = 0 — Element and node numbering sequence from top to bottom and left to right. = 1 — Element and node numbering sequence from left to right and top to bottom. NSNODE — Starting node number

Card Group	Input Data and Definitions (Model 1)
2 Block Coordinate	2.1 NBNODE Cards: $NODE_1$, X_1, Y_1 $NODE_2$, X_2, Y_2 - - - - - - NODE Node number X X-coordinate Y Y-coordinate

Card Group	Input Data and Definitions (Model 1)
3	3.1 BLNAME BLNAME Block name (up to 60 characters)
	3.2 IBLNO IBLNO Block number
	3.3 I_1, I_2, I_3, I_4, M_5, M_6, M_7, M_8, M_9 See Figure 7.1 I_1, I_2, I_3, I_4 Corner node number M_5, M_6, M_7, M_8 Side node number M_9 Center node number
Data for Each Block	3.4 IBASE, IB_1, IB_2, IB_3, IB_4, IB_5, IB_6, IB_7, IB_8 (SMAP-2D) IB_1, IB_2, IB_3, IB_4, IB_5, IB_6, IB_7, IB_8 (SMAP-S2) See Figure 7.3 IBASE Base boundary code IB_1, IB_2, IB_3, IB_4 Corner boundary code IB_5, IB_6, IB_7, IB_8 Edge boundary code

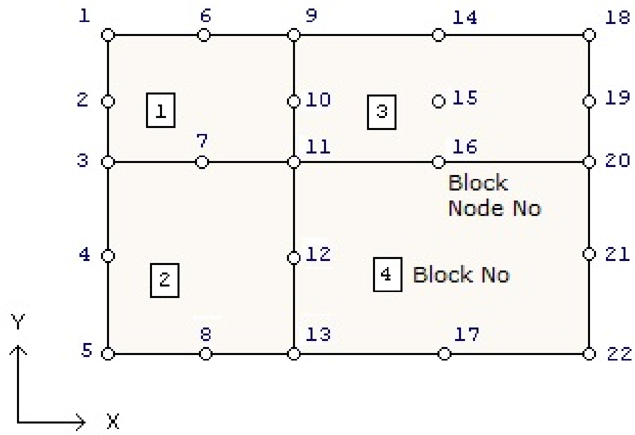
Card Group	Input Data and Definitions (Model 1)
3 Data for Each Block	3.5 MATNO, NDX, NDY, KS, KF (SMAP-2D) MATNO, NDX, NDY, THICK, DENSITY (SMAP-S2) MATNO, NDX, NDY, IDH (SMAP-T2) MATNO — Material property number If MATNO = 0, the block is void. NDX — Number of elements in x-direction NDY — Number of elements in y-direction KS = 0 — Has solid phase = 1 — No solid phase KF = 0 — Has fluid phase = 1 — No fluid phase THICK — Thickness of element. For plane strain, use THICK=1.0 DENSITY — Unit weight of element IDH — Heat generation history ID number

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Model 1)</th></tr>
<tr><td rowspan="3">3

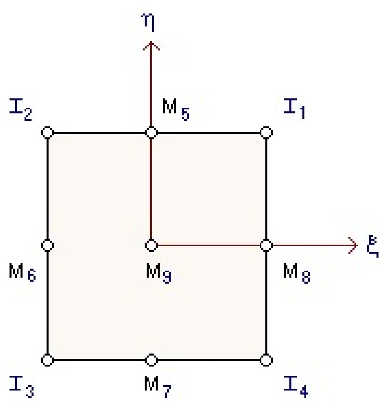
Data for Each Block</td><td colspan="2">3.6
NFSIDE
NFSIDE Number of block sides where boundary forces are specified</td></tr>
<tr><td rowspan="2">Force Data for Each Specified Side (see Figure 7.4)</td><td>3.7.1
IEDGE, LHNO, IBF
IEDGE Edge designation number
LHNO Load history number
IBF = 0 No applied force
= 1 Static fluid pressure
= 2 Horizontal force
= 3 Vertical force
= 4 Horizontal and vertical force</td></tr>
<tr><td>3.7.2
IBF = 1 > $IDIR_n$, q_{n1}, q_{n2}
= 2 > $IDIR_h$, q_{h1}, q_{h2}
= 3 > $IDIR_v$, q_{v1}, q_{v2}
= 4 > $IDIR_h$, q_{h1}, q_{h2}
$IDIR_v$, q_{v1}, q_{v2}

IDIR = 1 Pressure/force increases linearly with x
= 2 Pressure/force increases linearly with y

q_{n1}, q_{n2} Static pressure coefficient at edge ends
q_{h1}, q_{h2} Horizontal components of load coefficients at edge ends
q_{v1}, q_{v2} Vertical components of load coefficients at edge ends</td></tr>
</table>



NBLOCK = 4, NBNODE = 22
Block number should be in order from top to bottom and left to right



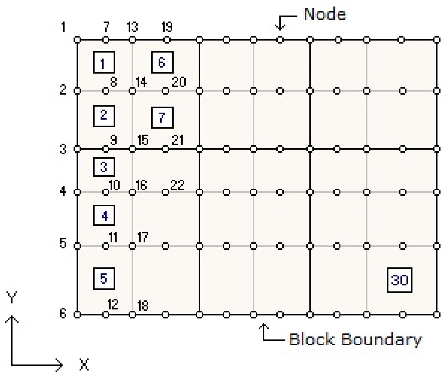
For Block Number 3

I_1 = 18
I_2 = 9
I_3 = 11
I_4 = 20
M_9 = 15

M_5 = 14
M_6 = 10
M_7 = 16
M_8 = 19

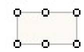
PRESMAP uses Serendipity interpolation if $M_9 = 0$
and Lagrangion interpolation if $M_9 \neq 0$

Figure 7.1 Block Specification and Block Index



NBX = 3
NBY = 2

MIDX = 1
MIDY = 0

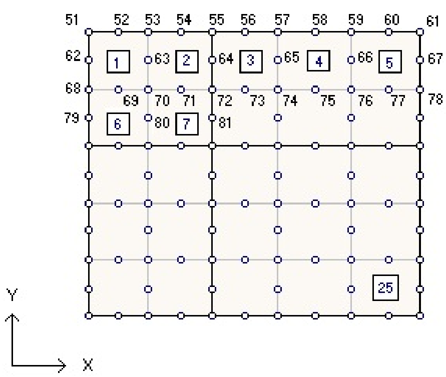


6 Node Element

NF = 0 (Element and node numbers generated from top to bottom and left to right)

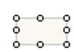
NSNODE = 1

□ Element Number



NBX = 2
NBY = 2

MIDX = 1
MIDY = 1



8 Node Element

NF = 1 (Element and node numbers generated from left to right and top to bottom)

NSNODE = 51

□ Element Number

Figure 7.2 Element and Node Numbering Sequence for Model 1 of PRESMAP-2D

Boundary Codes				
IBASE or IB	ISX	ISY	IFX	IFY
0	0	0	0	0
1	1	0	0	0
2	0	1	0	0
3	1	1	0	0
4	0	0	1	0
5	1	0	1	0
6	0	1	1	0
7	1	1	1	0
8	0	0	0	1
9	1	0	0	1
10	0	1	0	1
11	1	1	0	1
12	0	0	1	1
13	1	0	1	1
14	0	1	1	1
15	1	1	1	1

ISX Specifies skeleton X(radial) degree of freedom
ISY Specifies skeleton Y(axial) degree of freedom

IFX Specifies X(radial) degree of freedom for relative pore fluid motion.
IFY Specifies Y(axial) degree of freedom for relative pore fluid motion.

ISX, ISY, IFX, IFY = 0 Free to move in specified direction
= 1 Fixed in specified direction

Figure 7.3a Boundary Codes for SMAP-2D

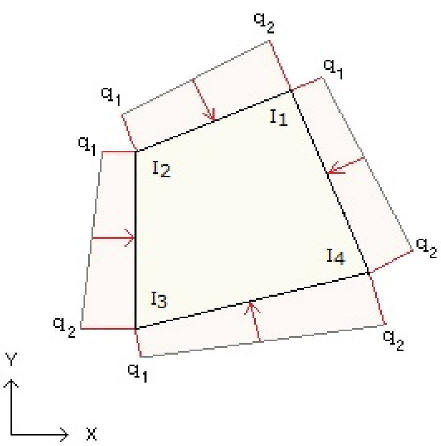
Boundary Type	Boundary Codes		
IB	IDX	IDY	IDT
0	0	0	1
1	1	0	1
2	0	1	1
3	1	1	1
4	0	0	0
5	1	0	0
6	0	1	0
7	1	1	0

IDX = 0 Displacement in x-direction is free
= 1 Displacement in x-direction is fixed

IDY = 0 Displacement in y-direction is free
= 1 Displacement in y-direction is fixed

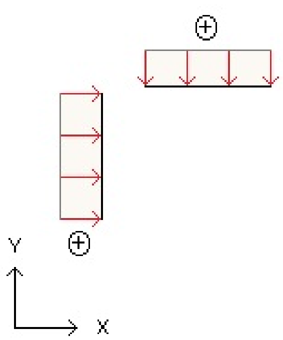
IDT = 0 Rotational degree of freedom is free
= 1 Rotational degree of freedom is fixed

Figure 7.3b Boundary Codes for SMAP-S2



BF = 1 Static fluid pressure
Compression is positive

I_2 I_1	IEDGE = 1
I_2 I_3	IEDGE = 2
I_3 I_4	IEDGE = 3
I_1 I_4	IEDGE = 4



Horizontal / Vertical Force
Sign Convention

IBF	= 1	Static fluid pressure
	= 2	Horizontal force
	= 3	Vertical force
	= 4	Horizontal / Vertical force

Figure 7.4 Boundary Force/Pressure Sign Conventions

PRESMAP-2D

Model 2

User's Manual

<table>
<tr><th>Card Group</th><th>Input Data and Definitions (Model 2)</th></tr>
<tr><td rowspan="4">1

General Information</td><td>1.1
TITLE
TITLE — Any title (Max = 60 characters)</td></tr>
<tr><td>1.2
IP
IP = 0 — Plane strain or plane stress
= 1 — Axisymmetry</td></tr>
<tr><td>1.3
NSNEL, NSNODE, NF, CMFAC (SMAP-S2/2D)
NSNEL, NSNODE, NF, CMFAC, TEMPI (SMAP-T2)

NSNEL — Starting element number
NSNODE — Starting node number

NF = 0 — Element and node numbering sequence from top to bottom and left to right
= 1 — Element and node numbering sequence from left to right and top to bottom

CMFAC — Coordinate magnification factor
TEMPI — Initial temperature</td></tr>
<tr><td>1.4
NSUBR, NDRF, NDRS, NDRT, DRF, DRS

See Figure 7.5

NSUBR — Number of subregions
NDRF — Number of divisions in the first row block
NDRS — Number of divisions in the second row block
NDRT — Number of divisions in the third row block
DRF — Length of the first row block
DRS — Length of the second row block</td></tr>
</table>

Card Group	Input Data and Definitions (Model 2)
2 Data for Each Subregion	2.1 SUBNAME SUBNAME Subregion name (up to 60 characters)
	2.2 ISUBNO ISUBNO Subregion number
	2.3 ISBTYPE, LSFTYPE, NSEG See Figure 7.6 and 7.7 ISBTYPE = 0 Column grids are normal to subregion surface = 1 Column grids are straight line LSFTYPE = 0 Straight line subregion surface = 1 Circular subregion surface NSEG Number of segments along subregion surface

<table>
<tr><th>Card Group</th><th colspan="3">Input Data and Definitions (Model 2)</th></tr>
<tr><td rowspan="2">2
Data for Each Subregion</td><td rowspan="2">2.4
Subregion Surface (Figure 7.6 and 7.7)</td><td>For LSFTYPE= 0</td><td>2.4.1
X_A, Y_A, X_B, Y_B

X_A, Y_A X and Y coordinate of point A
X_B, Y_B X and Y coordinate of point B</td></tr>
<tr><td>For LSFTYPE1=1</td><td>2.4.2
R, X_O, Y_O, θ_a, θ_B

R Radius of arc AB
X_O, Y_O X and Y coordinate of circle origin
θ_A, θ_B Polar angle (degree) of point A and B</td></tr>
</table>

<table>
<tr><th>Card Group</th><th colspan="4">Input Data and Definitions (Model 2)</th></tr>
<tr><td rowspan="4">2
Data for Each Subregion</td><td rowspan="4">2.5
Subregion Outer Edge</td><td rowspan="4">For ISBTYPE=0</td><td rowspan="2">Point C</td><td>2.5.1.1
LCTYPE
LCTYPE = 0 X_C and Y_C are specified
= 1 X_C is specified
= 2 Y_C is specified
= 3 DRT_C is specified</td></tr>
<tr><td>2.5.1.2
If LCTYPE = 0 --> X_C, Y_C
= 1 --> X_C
= 2 --> Y_C
= 3 --> DRT_C

X_C, Y_C X and Y coordinate of point C
DRT_C Length of third row block along the edge AC</td></tr>
<tr><td rowspan="2">Point D</td><td>2.5.2.1
LDTYPE
LDTYPE = 0 X_D and Y_D are specified
= 1 X_D is specified
= 2 Y_D is specified
= 3 DRT_D is specified</td></tr>
<tr><td>2.5.2.2
If LDTYPE = 0 --> X_D, Y_D
= 1 --> X_D
= 2 --> Y_D
= 3 --> DRT_D

X_D,Y_D X and Y coordinate of point D
DRT_D Length of third row block along the edge BD.</td></tr>
</table>

Card Group			Input Data and Definitions (Model 2)
2 Data for Each Subregion	2.5 Subregion Outer Edge	For ISBTYPE =1	2.5.3 X_C, Y_C, X_D, Y_D X_C, Y_C X and Y coordinate of point C X_D, Y_D X and Y coordinate of point D

Card Group	Input Data and Definitions (Model 2)
2 Data for Each Subregion	2.6 $IBASE_1$, $IBASE_2$, $IBASE_3$ (SMAP-2D) IB_B, IB_A, IB_C, IB_D, IB_{AB}, IB_{AC}, IB_{CD}, IB_{BD} (SMAP-2D/S2) See Figure 7.3 in Model 1 $IBASE_1$, $IBASE_2$, $IBASE_3$ — First, second, and third block base boundary code IB_B , IB_A , IB_C , IB_D — Corner boundary code IB_{AB}, IB_{AC}, IB_{CD}, IB_{BD} — Edge boundary code
	2.7 1st Block: $MATNO_1$, KS_1, KF_1 (SMAP-2D) $MATNO_1$, $DENSITY_1$ (SMAP-S2) $MATNO_1$, IDH_1 (SMAP-T2) 2nd Block: - - 3rd Block: - - $MATNO_1$ — Material property number of first block KS_1, KF_1 — Solid and fluid phase flag of first block $DENSITY_1$ — Unit weight of first block IDH_1 — Heat generation history ID of first block Note: For KS and KF, refer to Card Group 3.5 in PRESMAP-2D Model 1 User's Manual

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Model 2)</th></tr>
<tr><td rowspan="3">2

Data for Each Subregion</td><td colspan="2">2.8

NFSIDE

NFSIDE — Number of edge where boundary forces are specified</td></tr>
<tr><td rowspan="2">2.9

Force Data for Each Specified Edge (see Figure 7.8)</td><td>2.9.1

IEDGE, LHNO, IBF

IEDGE — Edge designation number
LHNO — Load history number

IBF = 0 — No applied force
= 1 — Static fluid pressure
= 2 — Horizontal force
= 3 — Vertical force
= 4 — Horizontal and vertical force</td></tr>
<tr><td>2.9.2

IBF = 1 --> $IDIR_n$, q_{n1}, q_{n2}
= 2 --> $IDIR_h$, q_{h1}, q_{h2}
= 3 --> $IDIR_v$, q_{v1}, q_{v2}
= 4 --> $IDIR_h$, q_{h1}, q_{h2}
$IDIR_v$, q_{v1}, q_{v2}

IDIR = 1 — Pressure/force increases linearly with x
= 2 — Pressure/force increases linearly with y

q_{n1}, q_{n2} — Static pressure coefficients
q_{h1}, q_{h2} — Horizontal load coefficients
q_{v1}, q_{v2} — Vertical load coefficients</td></tr>
</table>

Block numbers are in order from surface to outer edge and counterclockwise

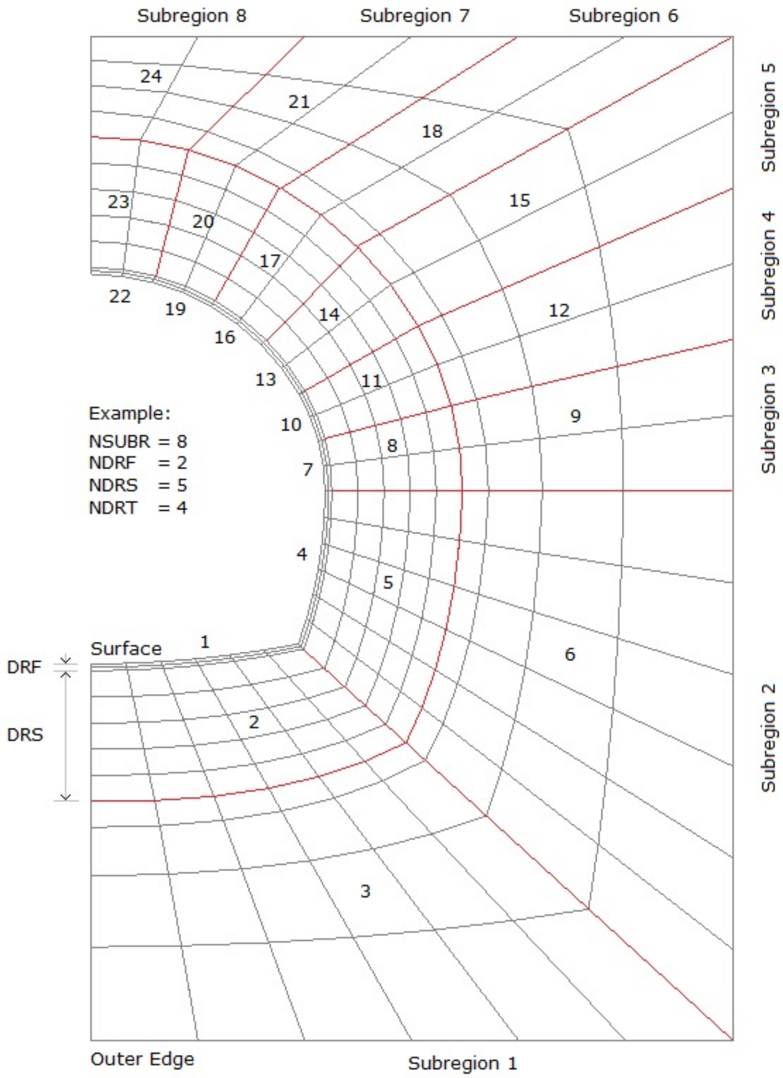


Figure 7.5 Subregion and block numbering sequence

Table 7.1 Subregion parameters in Example Figure 7.5

Subregion	ISBTYPE	LSFTYPE	NSEG
1	1	1	6
2	1	1	6
3	0	1	2
4	0	1	2
5	0	1	2
6	0	1	2
7	0	1	2
8	0	1	2

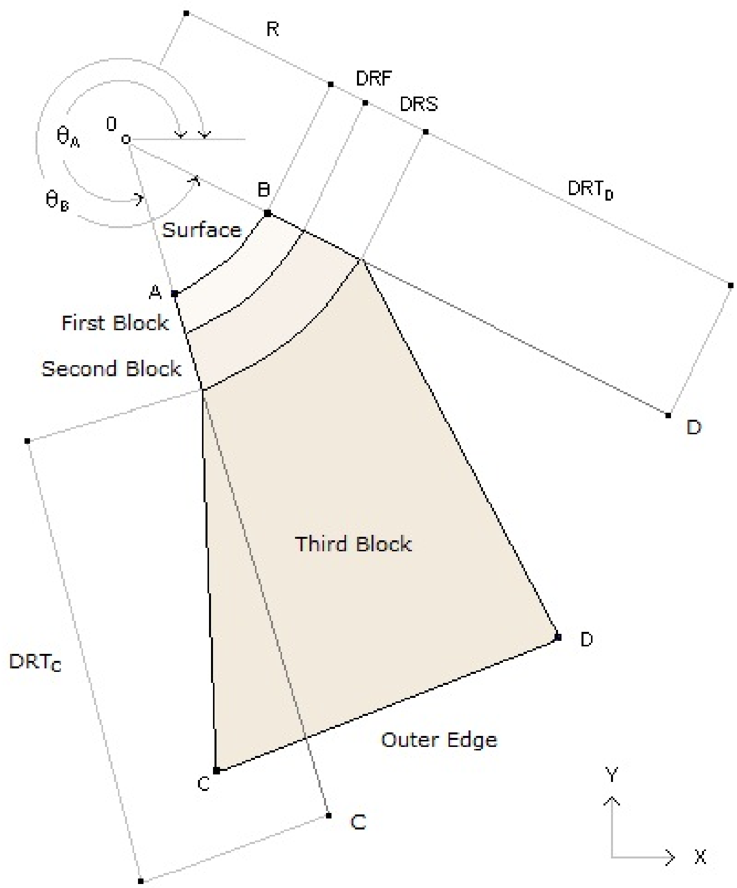


Figure 7.6 Typical Subregion for ISBTYPE = 0

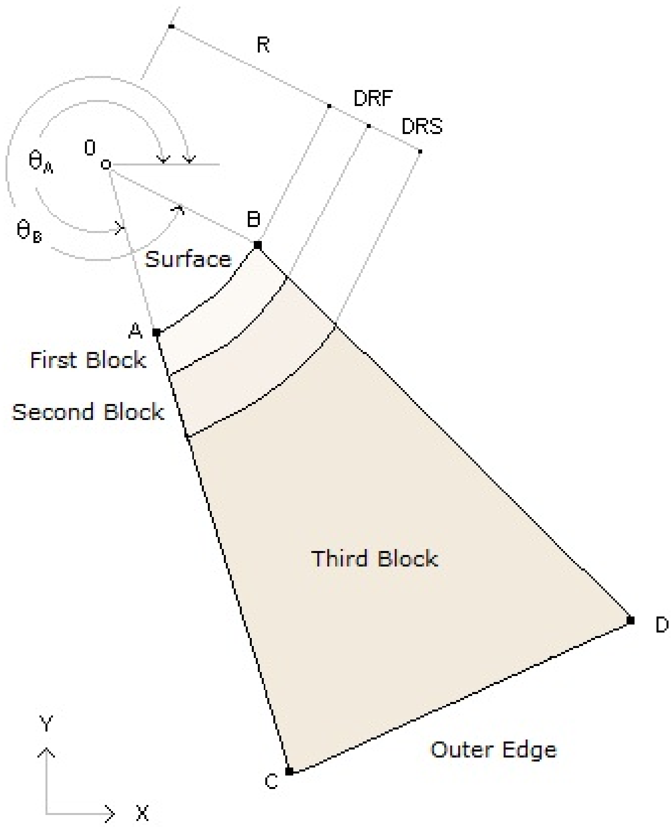
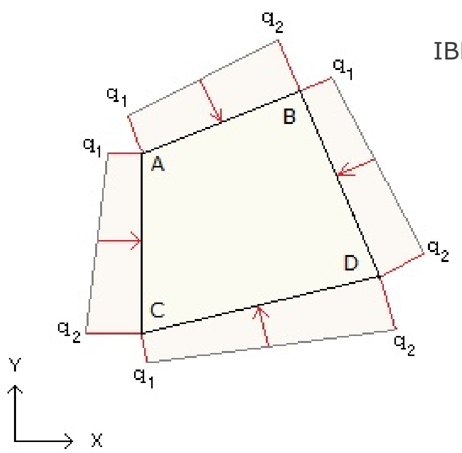
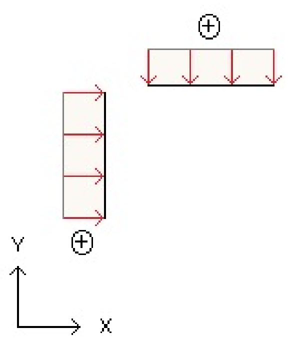


Figure 7.7 Typical Subregion for ISBTYPE = 1



IBF = 1 Static fluid pressure
Compression is positive

AB:	IEDGE = 1
AC:	IEDGE = 2
CD:	IEDGE = 3
BD:	IEDGE = 4



Horizontal / Vertical Force
Sign Convention

IBF	= 1	Static fluid pressure
	= 2	Horizontal force
	= 3	Vertical force
	= 4	Horizontal / Vertical force

Figure 7.8 Boundary Force/Pressure Sign Conventions

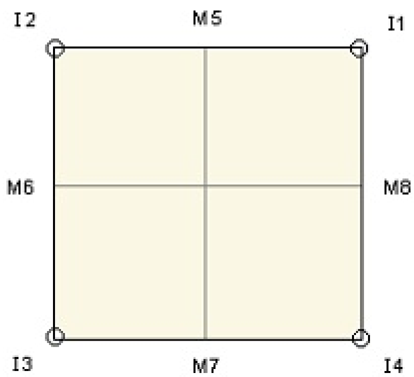
PRESMAP-2D

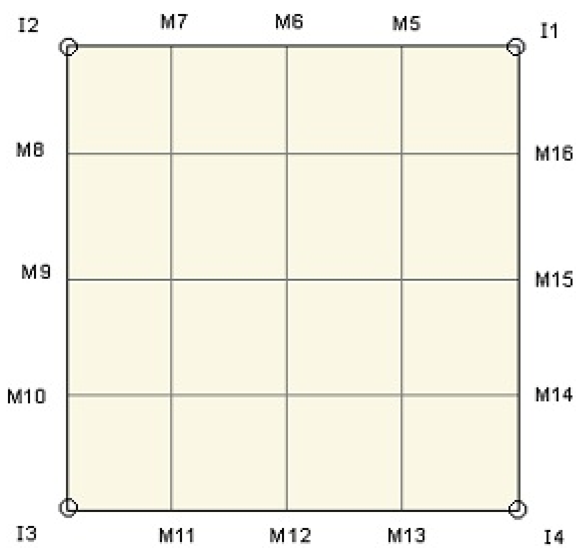
Model 3

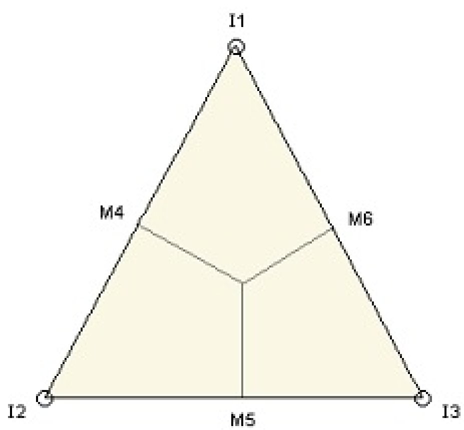
User's Manual

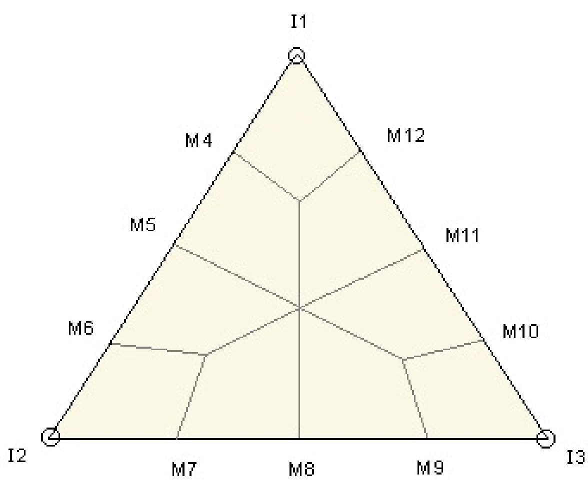
Card Group	Input Data and Definitions (Model 3)
1 General Information	1.1 TITLE TITLE — Any title (Max = 60 characters)
	1.2 IP IP = 0 — Plane geometry = 1 — Axisymmetry geometry
	1.3 NBLOCK, NBNODE, NSNEL, NSNODE, CMFAC See Figure 7.9 NBLOCK — Number of blocks NBNODE — Number of block nodes NSNEL — Starting element number NSNODE — Starting node number CMFAC — Coordinate magnification factor
2 Block Coordinates	2.1 NBNODE Cards: $NODE_1$, X_1, Y_1 $NODE_2$, X_2, Y_2 - - - - - - NODE — Node number X — X-coordinate Y — Y-coordinate

Card Group	Input Data and Definitions (Model 3)
3 Data for Each Block (see Figure 7.9)	3.1 IBLNO, IBLTYPE, MATNO, KS, KF (SMAP-2D) IBLNO, IBLTYPE, MATNO, DENSITY (SMAP-S2) IBLNO, IBLTYPE, MATNO, IDH (SMAP-T2) IBLNO Block number IBLTYPE Block type MATNO Material number KS = 0 Has solid phase = 1 No solid phase KF = 0 Has fluid phase = 1 No fluid phase DENSITY Unit weight IDH Heat generation history ID number

Card Group	Input Data and Definitions (Model 3)
3 Data for Each Block	3.2 For IBLTYPE = 1 I_1, I_2, I_3, I_4, M_5, M_6, M_7, M_8 I_1, I_2, I_3, I_4 Corner node number M_5, M_6, M_7, M_8 Side node number  Note: IBLTYPE = 1 generates 4 elements

Card Group	Input Data and Definitions (Model 3)
3 Data for Each Block	For IBLTYPE = 2 I_1, I_2, I_3, I_4, M_5, M_6, M_7, M_8, M_9, M_{10}, M_{11}, M_{12}, M_{13}, M_{14}, M_{15}, M_{16} I_1, I_2, I_3, I_4 Corner node number M_5, M_6,, M_{16} Side node number  Note: IBLTYPE = 2 generates 16 elements

Card Group	Input Data and Definitions (Model 3)
3 Data for Each Block	For IBLTYPE = 3 I_1, I_2, I_3, M_4, M_5, M_6 I_1, I_2, I_3 Corner node number M_4, M_5, M_6 Side node number  **Note:** IBLTYPE = 3 generates 3 elements

Card Group	Input Data and Definitions (Model 3)
3 Data for Each Block	For IBLTYPE = 4 I_1, I_2, I_3, M_4 M_5, M_6, M_7, M_8, M_9, M_{10} M_{11}, M_{12} I_1, I_2, I_3 Corner node number M_4 - M_{12} Side node number  **Note:** IBLTYPE = 4 generates 9 elements

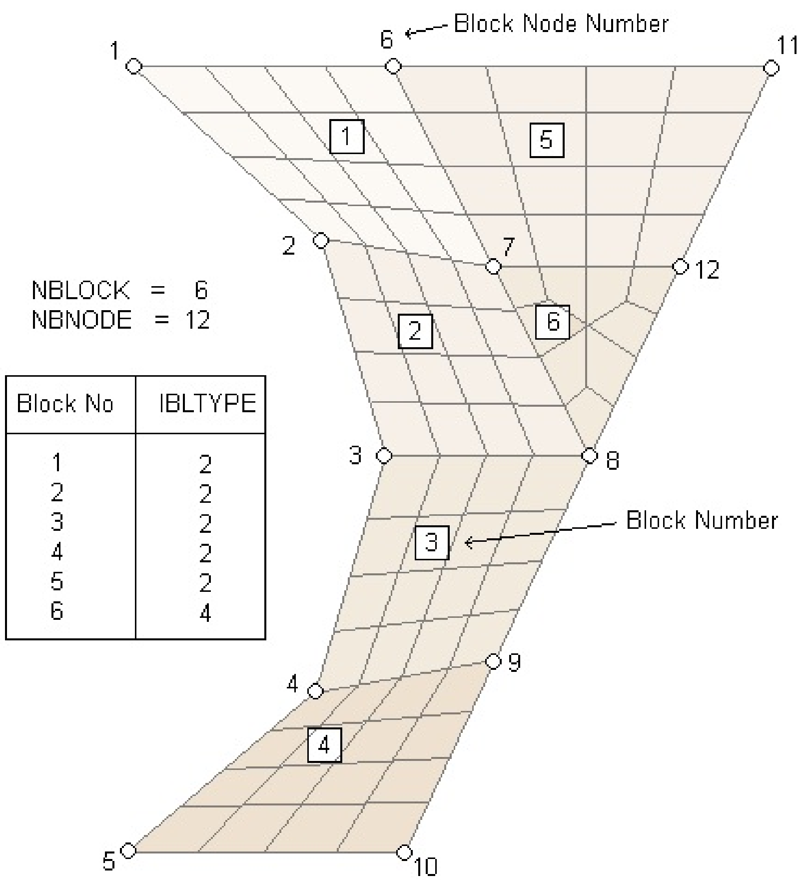


Figure 7.9 Block Node Number for Model 3 of PRESMAP-2D

PRESMAP-2D

Model 4

User's Manual

Card Group	Input Data and Definitions (Model 4)
1 General Information	1.1 TITLE TITLE Any title (Max = 60 characters)
	1.2 NLAYER, NDIV, ITRANGL See Figure 7.10 NLAYER Number of layer NDIV Number of elements in first layer ITRANGL = 0 Last element in each layer is rectangle = 1 Last element in each layer is triangle
	1.3 NSNEL, NSNODE, CMFAC NSNEL Starting element number NSNODE Starting node number CMFAC Coordinate magnification factor
2 Block Coordinates	2.1 XB1, YB1, YB2, XB3 See Figure 7.10 XB1, YB1 X, Y coordinate of block node 1 YB2 Y coordinate of block node 2 XB3 X coordinate of block node 3

Card Group	Input Data and Definitions (Model 4)
3 Material Parameters	3.1 MATNO, KS, KF (SMAP-2D) MATNO, DENSITY (SMAP-S2) MATNO, IDH (SMAP-T2) MATNO Material number KS = 0 Has solid phase = 1 No solid phase KF = 0 Has fluid phase = 1 No fluid phase DENSITY Unit weight IDH Heat generation history ID number

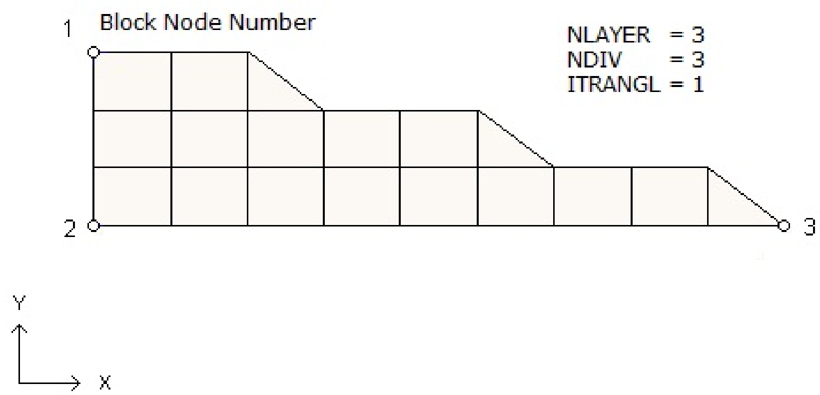


Figure 7.10 Block node number for Model 4 of PRESMAP-2D

NATM-2D

User's Manual

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td rowspan="3">1

General Information</td><td>1.1

TITLE
TITLE Any title (Max = 60 characters)</td></tr>
<tr><td>1.2

IUNIT
<table><tr><th>IUNIT</th><th>Length</th><th>Force</th><th>Pressure</th><th>Unit Weight</th></tr><tr><td>1</td><td>in</td><td>lb</td><td>lb/in^2</td><td>lb/in^3</td></tr><tr><td>2</td><td>m</td><td>ton</td><td>ton/m^2</td><td>ton/m^3</td></tr></table></td></tr>
<tr><td>1.3

MODEL, IGEN, IEXMESH, ILNCOUPL, IAUTO

MODEL = 1 Single tunnel (Half section)
= 2 Single tunnel (Full section)
= 3 Two tunnels (Symmetric)
= 4 Two tunnels (Unsymmetric)

IGEN = 0 Generate whole mesh
= 1 Generate core
= 2 Generate surrounding

IEXMESH = 0 No user supplied mesh
= 1 Add generated mesh to user supplied mesh

For Lining analysis
ILNCOUPL= 0 Surrounding rock by continuum element
= 1 Surrounding rock by spring element

IAUTO = 0 Generate Mesh file
= 1 Generate Mesh, Main and Post files
Available only for SMAP-S2

See Figure 7.11</td></tr>
</table>

Card Group	Input Data and Definitions
2 Tunnel Analysis Boundary	2.1 MODEL = 1: HT, HL, W, DX, DY, NY = 2: HT, HL, W, DX, DY, NY = 3: HT, HL, W, WP, DX, DY, NY = 4: HT, HL, W, WP, HP, DX, DY, NY HT Tunnel depth HL Depth from springline to bottom boundary W Horizontal distance from left to right boundary WP Horizontal distance from left tunnel center line to right tunnel center line HP Vertical distance from right tunnel springline to left tunnel springline. When HP is positive, left tunnel springline is above the right tunnel springline. DX Far-field horizontal element length DY Far-field vertical element length NY Maximum number of elements in the vertical direction See Figure 7.11

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td rowspan="2">3

Soil / Rock Layer Information</td><td>3.1

NLAYER

NLAYER — Total number of layers. Max = 10</td></tr>
<tr><td>3.2

NLAYER Cards:
LAYERNO$_1$, H$_1$, DD$_1$
LAYERNO$_2$, H$_2$, DD$_2$
- - -
- - -

LAYERNO — Soil/rock layer number
H — Thickness of soil/rock layer

DD = GAMA — SMAP-S2
= IDH — SMAP-T2
= KF — SMAP-2D

GAMA — Unit weight
IDH — Heat generation history ID number

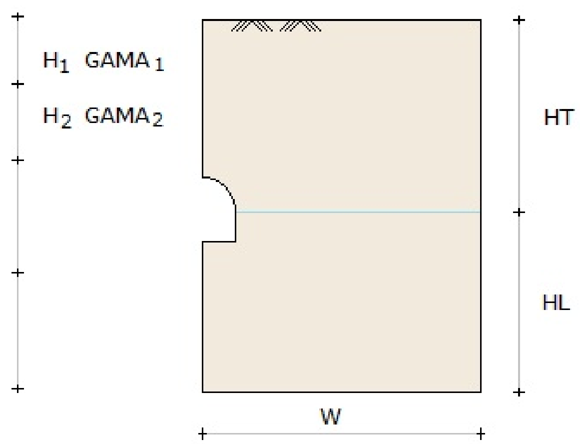
KF = 0 — Has fluid phase
= 1 — No fluid phase

See Figure 7.11</td></tr>
</table>

Card Group	Input Data and Definitions
4	4.1 R_1, A_1, R_2, A_2, R_3, A_3, R_4, GR, GA R_1, R_2, R_3, R_4 Radius as shown in Figure 7.12 A_1, A_2, A_3 Angle (˚) as shown in Figure 7.12 GR Growing rate for near-field element. Use GR = 1 GA Normalized mid length. Use GA= 0.5
Tunnel Dimension (Repeat this card group for the left tunnel when MODEL = 4)	4.2 INVSHOT, T_S, T_l INVSHOT = 0 No shotcrete at invert = 1 Shotcrete at invert T_S Thickness of shotcrete T_l Thickness of lining Note: For A_1+A_2 > 90, invert shotcrtete is always included
	4.3 NUMRB, L_{RB}, $L_{SPACING}$, $T_{SPACING}$, NSRB NUMRB Number of rock bolts Example: NUMRB = 11 in Figure 7.12 L_{RB} Length of rock bolt $L_{SPACING}$ Rock bolt spacing in longitudinal direction $T_{SPACING}$ Rock bolt spacing in tangential direction NSRB Number of elements between rock bolts Use NSRB = 2 or 3

Card Group	Input Data and Definitions
5 Water Pressure and Loosening Load	5.1 LDTYPE, DGW, GAMAW, HPRES, VPRES, SUBGK, ITSPR, NUMSJ LDTYPE = 0 No external load = 1 Water pressure only = 2 Loosening load only = 3 Water pressure and loosening load DGW Depth of ground water table from ground surface GAMAW Unit weight of water HPRES Horizontal pressure due to loosening load VPRES Vertical Pressure due to loosening load SUBGK Coefficient of subgrade reaction (ILCOUPL = 1) ITSPR = 0 No tangential spring = 1 Add tangential spring NUMSJ Number of segment joints Available for circular shape of MODEL 2
	5.2 Joint Locations If NUMSJ = 0, skip this card AJ_1, AJ_1, ... , AJ_{NUMSJ} AJ_i Angle (degrees) from crown top ($AJ_i \leq 180$)

MODEL = 1 Single Tunnel (Half Section)



MODEL = 2 Single Tunnel (Full Section)

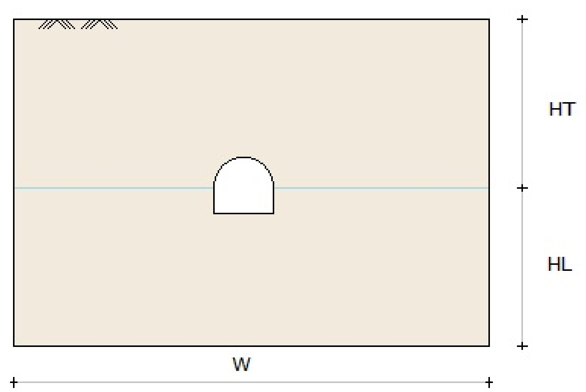
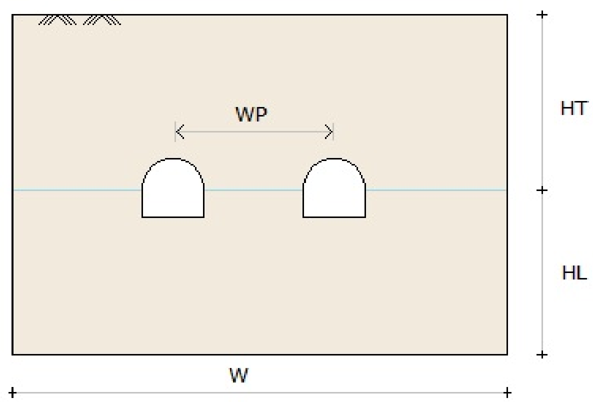


Figure 7.11 Schematic tunnel section view for MODEL = 1 and 2

MODEL = 3 Two Tunnel (Symmetric Section)



MODEL = 4 Two Tunnel (Unsymmetric Section)

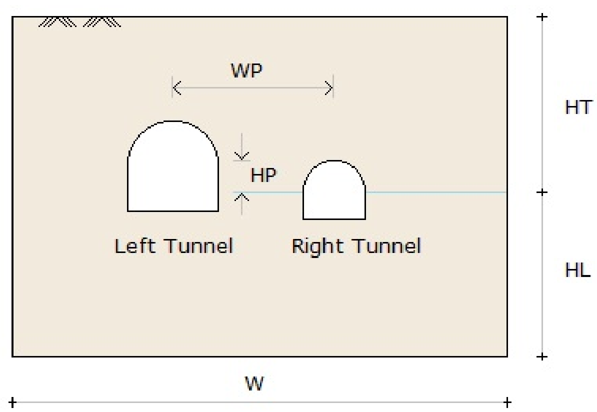
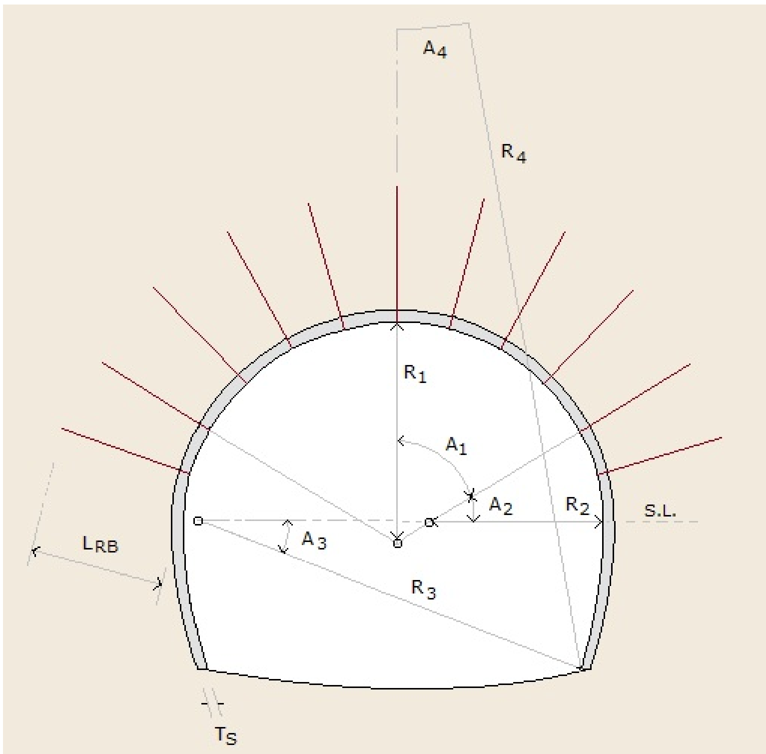


Figure 7.11 Schematic tunnel section view for MODEL = 3 and 4



R_4 = 0 : Invert is flat
R_4 < 0 : Invert depth is given as absolute value of R_4

Refer to Example problem MODEL 4-1 and 4-3

Figure 7.12 Tunnel dimension (A_1+A_2 = 90)

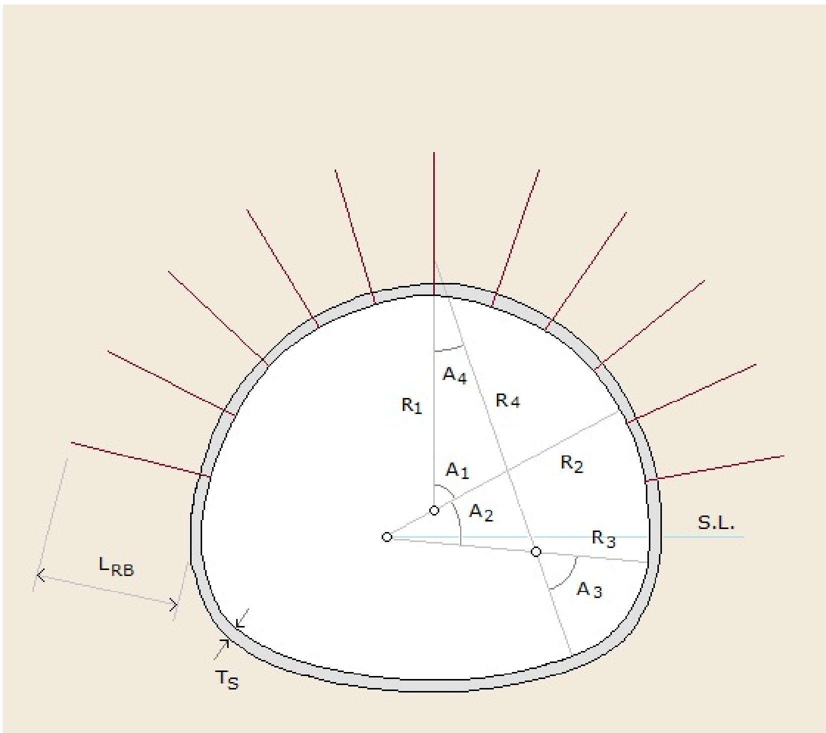


Figure 7.12 Tunnel dimension (A_1+A_2 > 90)

CIRCLE-2D

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<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td rowspan="2">1
General Information</td><td>1.1

TITLE
TITLE Any title (Max = 80 characters)</td></tr>
<tr><td>1.3

MODEL, NSNEL, NSNODE

MODEL = 1 Quarter Section
= 2 Half Section
= 3 Full Section

NSNEL Starting element number
NSNODE Starting node number

See Figure 7.13</td></tr>
<tr><td>2
Geometry</td><td>2.1

R, FINEMESH, NEARMESH, NDIV, BH, BV

R Radius of Circular Core

FINEMESH = 0 Coarse Mesh
= 1 Fine Mesh

NEARMESH = 0 All Quad Mesh
= 1 Quad and Triangle Mesh

NDIV Number of divisions for outer zone
BH, BV Horizontal and Vertical dimensions</td></tr>
</table>

Card Group	Input Data and Definitions
3 Material Number	3.1 $COREMAT_1$, $COREMAT_2$, $COREMAT_{2j}$, JOINTMAT, NEARMAT $COREMAT_1$ Material No for Core 1 $COREMAT_2$ Material No for Core 2 $COREMAT_{2j}$ Material No for Core 2 facing Joint JOINTMAT Material No for Joint NEARMAT Material No for Near Note $COREMAT_1$ and $COREMAT_2$ have the common interface with NEARMAT and JOINTMAT, respectively. When material number for $COREMAT_1$ or JOINTMAT is zero, meshes corresponding to that material will not be generated.

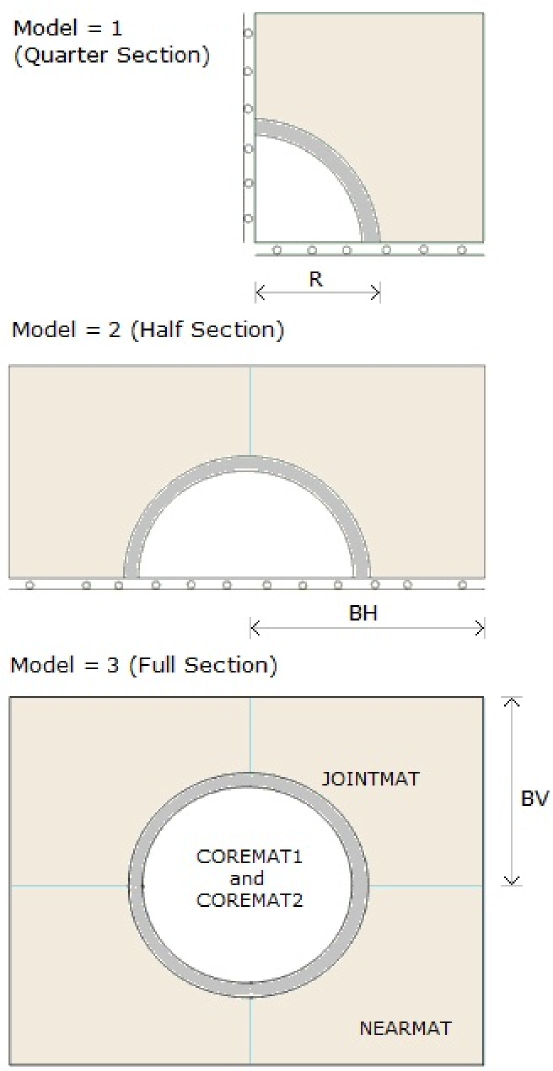

Figure 7.13 Model type for CIRCLE-2D

PRESMAP-3D

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<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td>1

General Information</td><td>1.1

TITLE
TITLE   Any title (Max = 80 characters)</td></tr>
<tr><td></td><td>1.3

NBLOCK, NBNODE, NSNODE, NSNEL, CMFAC

NBLOCK   Number of blocks
NBNODE   Number of block nodes
NSNODE   Starting node number
NSNEL   Starting element number
CMFAC   Coordinate magnification factor

Note: If NBLOCK is negative value, the output file contains plotting information for block diagram.</td></tr>
<tr><td>2

Block Coordinates</td><td>2.1

NBNODE Cards
$NODE_1$, X_1, Y_1, Z_1
$NODE_2$, X_2, Y_2, Z_2
- - - -
- - - -

NODE   Node number
X   X-coordinate
Y   Y-coordinate
Z   Z-coordinate</td></tr>
</table>

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td rowspan="3">3
Data for Each Block</td><td>3.1

BLNAME

BLNAME Block name (Max = 60 characters)</td></tr>
<tr><td>3.2

ILAG

ILAG = 0 Serendipity interpolation
= 1 Lagrangian interpolation</td></tr>
<tr><td>3.3

I_1, I_2, I_3, I_4, I_5, I_6, I_7, I_8
M_9, M_{10}, M_{11}, M_{12}, M_{13}, M_{14}, M_{15}, M_{16}, M_{17}, M_{18}, M_{19}, M_{20}
M_{21}, M_{22}, M_{23}, M_{24}, M_{25}, M_{26}, M_{27} (only for ILAG=1)

See Figure 7.13

I_1 - I_8 Corner node number of a block
M_9 - M_{20} Side node number of a block
M_{21} - M_{27} Side node number of a block required for Lagrangian interpolation.</td></tr>
</table>

Card Group		Input Data and Definitions
3	3.4	3.4.1 NBOUND NBOUND — Number of boundaries to be specified. If NBOUND=0, go to Card group 3.5
Data for Each Block	See Figure 7.14	3.4.2 NBOUND Cards For SMAP-3D IBTYPE, ISX, ISY, ISZ, IFX, IFY, IFZ For SMAP-T3 IBTYPE, ID, IDF IBTYPE = 1 Interior volume = 2 Front surface = 3 Back surface = 4 Left surface = 5 Right surface = 6 Top surface = 7 Bottom surface = 8 Line I_1 and I_2 = 9 Line I_2 and I_3 = 10 Line I_3 and I_4 = 11 Line I_4 and I_1 = 12 Line I_5 and I_6 = 13 Line I_6 and I_7 = 14 Line I_7 and I_8 = 15 Line I_8 and I_5 = 16 Line I_1 and I_5 = 17 Line I_2 and I_6 = 18 Line I_3 and I_7 = 19 Line I_4 and I_8

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions</th></tr>
<tr><td>3

Data for Each Block</td><td>See Figure 7.14</td><td>3.4.2

= 20 Node I_1
= 21 Node I_2
= 22 Node I_3
= 23 Node I_4
= 24 Node I_5
= 25 Node I_6
= 26 Node I_7
= 27 Node I_8

ISX Skeleton X DOF
ISY Skeleton Y DOF
ISZ Skeleton Z DOF

IFX Pore fluid X DOF relative to skeleton
IFY Pore fluid Y DOF relative to skeleton
IFZ Pore fluid Z DOF relative to skeleton

ISX, ISY, ISZ, IFX, IFY, IFZ
= 0 Free to move in specified direction
= 1 Fixed in specified direction

Note: Default boundary conditions are
ISX=ISY=ISZ=0 and IFX=IFY=IFZ=1

For SMAP-T3

ID = 0 Heat flow is specified
= 1 Temperature is specified

IDF = Time history identification number</td></tr>
</table>

Card Group	Input Data and Definitions
3 Data for Each Block	3.5 MATNO, NDX, NDY, NDZ, KS, KF For SMAP-S3/3D MATNO, NDX, NDY, NDZ, IDH For SMAP-T3 MATNO Material property number NDX Number of elements in x-direction NDY Number of elements in y-direction NDZ Number of elements in z-direction KS =-1 Element has high explosive solid phase = 0 Element has solid phase > 0 Element has joint and absolute value of KS represents face designation number KF = 0 Element has fluid phase = 1 Element has no fluid phase IDH Heat generation history ID number

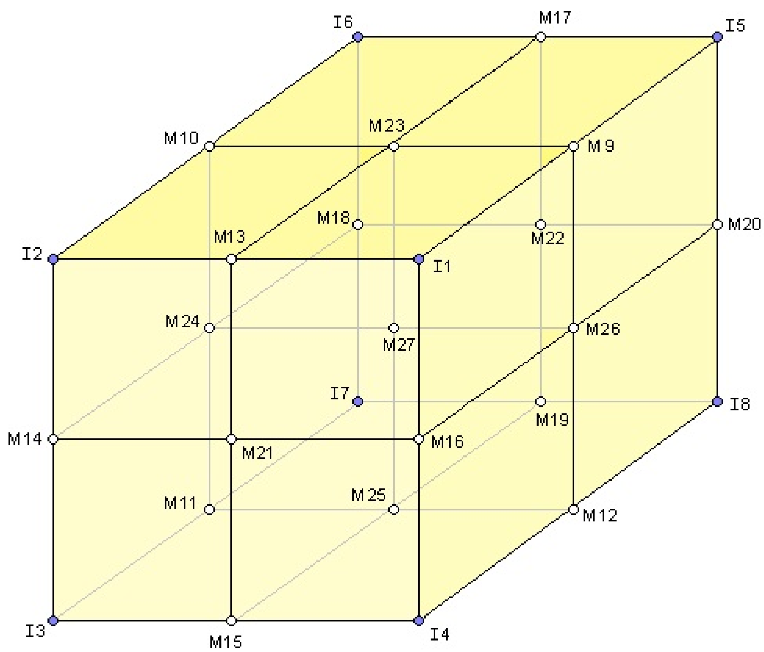

Figure 7.13 Block index for PRESMAP-3D

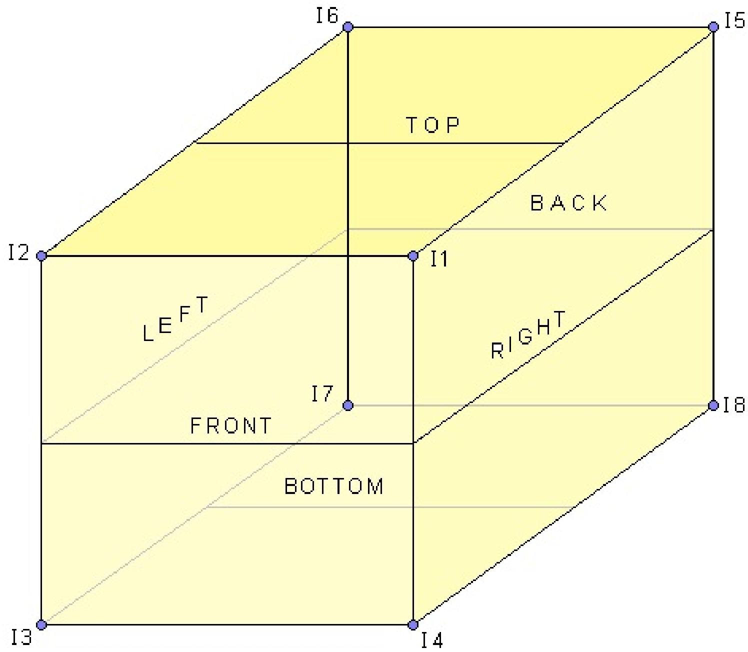

Figure 7.14 Boundary surface designation for PRESMAP-3D

CROSS-3D

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<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td rowspan="2">1

General Information</td><td>1.1

TITLE

TITLE Any title (Max = 80 characters)</td></tr>
<tr><td>1.2

MODELNO, KF, NSNODE, NSNEL, CMFAC (SMAP-3D)
MODELNO, IH, NSNODE, NSNEL, CMFAC (SMAP-T3)

MODELNO = 1 Identical size tunnels crossing at right angle at the same level. See Figure 7.15 and 7.16

= 2 Large and small tunnels crossing at right angle at the same level. See Figure 7.17 and 7.18

= 3 Lower and upper tunnels crossing at right angle with some clearance. See Figure 7.19 and 7.20

KF = 0 Element has fluid phase
= 1 Element has no fluid phase

IH Heat source ID number (SMAP-T3)

NSNODE Starting node number
NSNEL Starting element number
CMFAC Coordinate magnification factor</td></tr>
</table>

Card Group	Input Data and Definitions
2 For MODELNO =1 (Identical Two Crossing Tunnels, See Figures 7.15 & 7.16)	2.1.1 XL, YB, YT, ZL, t XL, YB, YT, ZL — Problem dimensions (See Figure 7.15) t — Radial distance from tunnel surface to the boundary of near region. Default value is 20% of the tunnel width. Example, t = liner thickness
	2.1.2 IPART, NDR, NTBND, NTOPN IPART = 0 Whole region (from Y =-YB to Y = YT) = 1 Upper region (from Y = 0.0 to Y = YT) = 2 Lower region (from Y =-YB to Y = 0.0) NDR — Number of elements along radial distance (t) NTBND — Number of elements along the length (XL+YB+YT+ZL) NTOPN — Number of elements along the perimeter of tunnel opening from node 1 to node 5. See Figure 7.16

Card Group	Input Data and Definitions
2	2.1.3
For MODELNO =1 (Identical Two Crossing Tunnels, see Figures 7.15 & 7.16)	NTNODE

NTNODE Cards

$NODE_1$, X_1, Y_1
$NODE_2$, X_2, Y_2
- - -
- - -

NTNODE	Number of nodes to specify tunnel shape
NODE	Node number
X	X-coordinate
Y	Y-coordinate

Note: Nodes from 1 to 5 are required

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td rowspan="2">2

For MODELNO =2 (Large and Small Crossing Tunnels, See Figures 7.17 & 7.18)</td><td>2.2.1

XL, YB, YT, ZL, t_l, t_s

XL, YB, YT, ZL — Problem dimensions (See Figure 7.17)

t_l, t_s — Radial distance from tunnel surface to the boundary of near region. t_l is for large tunnel and t_s for small tunnel ($t_l \geq t_s$). Default value is 20% of the tunnel width. Example, t = liner thickness</td></tr>
<tr><td>2.2.2

IPART, NDR, NTBND, NTOPNL, NTOPNS

IPART = 0 — Whole region (from Y = -YB to Y = YT)
= 1 — Upper region (from Y = 0.0 to Y = YT)
= 2 — Lower region (from Y = -YB to Y = 0.0)

NDR — Number of elements along the radial distance (t_l for large tunnel and t_s for small tunnel)

NTBND — Number of elements along the length (XL+YB+YT+ZL)

NTOPNL — Number of elements along the perimeter of large tunnel opening from node 1 to node 7 See Figure 7.18

NTOPNS — Number of elements along the perimeter of small tunnel opening from node 1 to node 5 See Figure 7.18</td></tr>
</table>

Card Group	Input Data and Definitions
2 For MODELNO =2 (Large and Small Crossing Tunnels, see Figures 7.17 & 7.18)	2.2.3 Large Tunnel Shape, See Figure 7.18 NTLNODE NTLNODE Cards: $NODE_1$, X_1, Y_1 $NODE_2$, X_2, Y_2 - - - - - - NTLNODE — Number of nodes to specify large tunnel NODE — Node number X — X-coordinate Y — Y-coordinate Note: Nodes from 1 to 7 are required
	2.2.4 Small Tunnel Shape, See Figure 7.18 NTSNODE NTSNODE Cards: $NODE_1$, Z_1, Y_1 $NODE_2$, Z_2, Y_2 - - - - - - NTSNODE — Number of nodes to specify small tunnel NODE — Node number Z — Z-coordinate Y — Y-coordinate Note: Nodes from 1 to 5 are required

Card Group	Input Data and Definitions
2 For MODELNO =3 (Crossing Tunnels with Clearance, See Figures 7.19 & 7.20)	2.3.1 XL, YB, YC, YT, ZL, t_l, t_u XL, YB, YC, YT, ZL — Problem dimensions (See Figure 7.19) t_l, t_u — Radial distance from tunnel surface to the boundary of near region. t_l is for lower tunnel and t_u for upper tunnel. Default value is 20% of the tunnel width. Example, t = liner thickness.
	2.3.2. NDRL, NDRU, NTBND, NTOPNL, NTOPNU NDRL — Number of elements along the radial distance (t_l) for lower tunnel NDRU — Number of elements along the radial distance (t_u) for upper tunnel NTBND — Number of elements along the length (XL+YB+YC+YT+ZL) NTOPNL — Number of elements along the perimeter of lower tunnel opening from node 1 to node 5. See Figure 7.20 NTOPNU — Number of elements along the perimeter of upper tunnel opening from node 1 to node 5. See Figure 7.20

Card Group	Input Data and Definitions
2 For MODELNO =3 (Crossing Tunnels with Clearance, See Figures 7.19 & 7.20)	2.3.3 **Lower Tunnel Shape, See Figures 7.20** NTLNODE NTLNODE Cards: $\left[\begin{array}{lll} NODE_1, & X_1, & Y_1 \\ NODE_2, & X_2, & Y_2 \\ - & - & - \\ - & - & - \end{array}\right.$ NTLNODE — Number of nodes to specify lower tunnel NODE — Node number X — X-coordinate Y — Y-coordinate Note: Nodes from 1 to 5 are required
	2.3.4 **Upper Tunnel Shape, See Figures 7.20** NTUNODE NTUNODE Cards: $\left[\begin{array}{lll} NODE_1, & Z_1, & Y_1 \\ NODE_2, & Z_2, & Y_2 \\ - & - & - \\ - & - & - \end{array}\right.$ NTUNODE — Number of nodes to specify upper tunnel NODE — Node number Z — Z-coordinate Y — Y-coordinate Note: Nodes from 1 to 5 are required

Card Group	Input Data and Definitions
3	3.1 NBOUND NBOUND Number of boundaries to be specified If NBOUND = 0, no data is required hereafter
Boundary Conditions	3.2 NBOUND Cards IBTYPE, ISX, ISY, ISZ, IFX, IFY, IFZ (SMAP-3D) IBTYPE, ID, IDF (SMAP-T3) IBTYPE = 1 Interior volume (overriding default) = 2 Front surface (Z = ZL) = 3 Back surface (Z = 0.0) = 4 Left surface (X = 0.0) = 5 Right surface (X = XL) = 6 Top surface For MODELNO = 1 or 2, Y = YT if IPART = 0 or 1 Y = 0.0 if IPART =2 For MODELNO = 3, Y = YT + YC = 7 Bottom surface For MODELNO = 1 or 2, Y = 0.0 if IPART = 1 Y =-YB if IPART = 0 or 2 For MODELNO = 3, Y =-YB ISX Skeleton X DOF ISY Skeleton Y DOF ISZ Skeleton Z DOF

Card Group	Input Data and Definitions
2 Boundary Conditions	3.2 IFX Pore fluiud X DOF relative to skeleton IFY Pore fluiud Y DOF relative to skeleton IFZ Pore fluiud Z DOF relative to skeleton ISX, ISY, ISZ, IFX, IFY, IFZ = 0 Free to move in specified direction = 1 Fixed in specified direction *Note:* Default boundary conditions are ISX=ISY=ISZ=0 and IFX=IFY=IFZ=1 For SMAP-T3 ID = 0 Heat flow is specified = 1 Temperature is specified IDF = Time history identification number

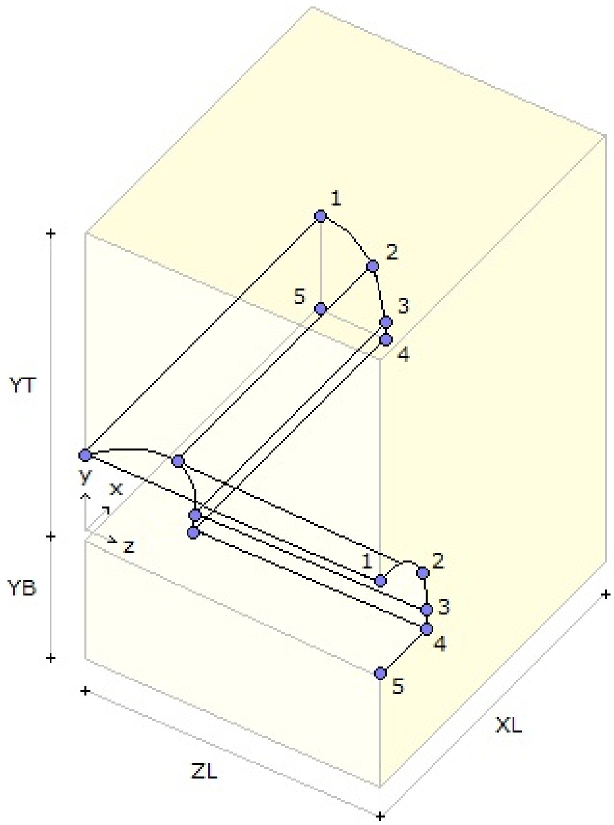


Figure 7.15 Schematic view of identical two crossing tunnels for MODELNO = 1

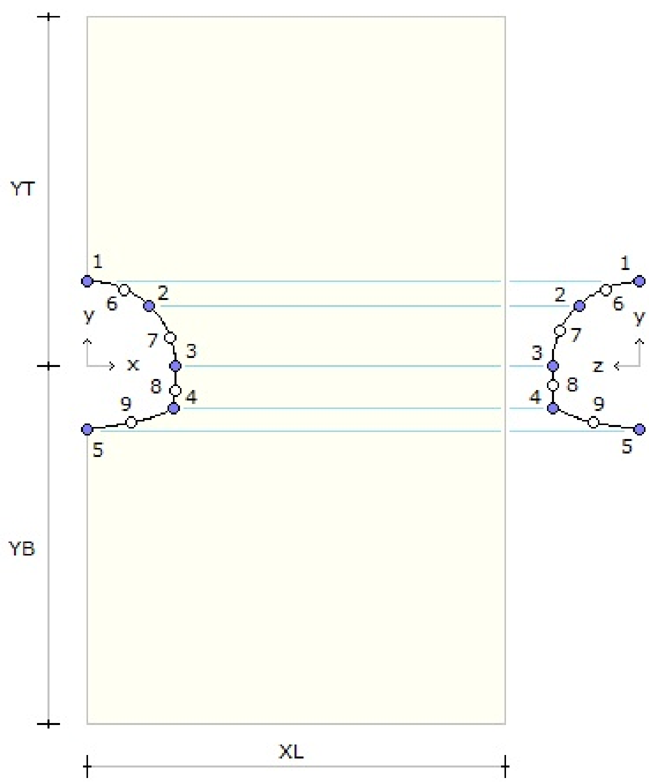


Figure 7.16 Node numbers defining tunnel shape for MODELNO = 1

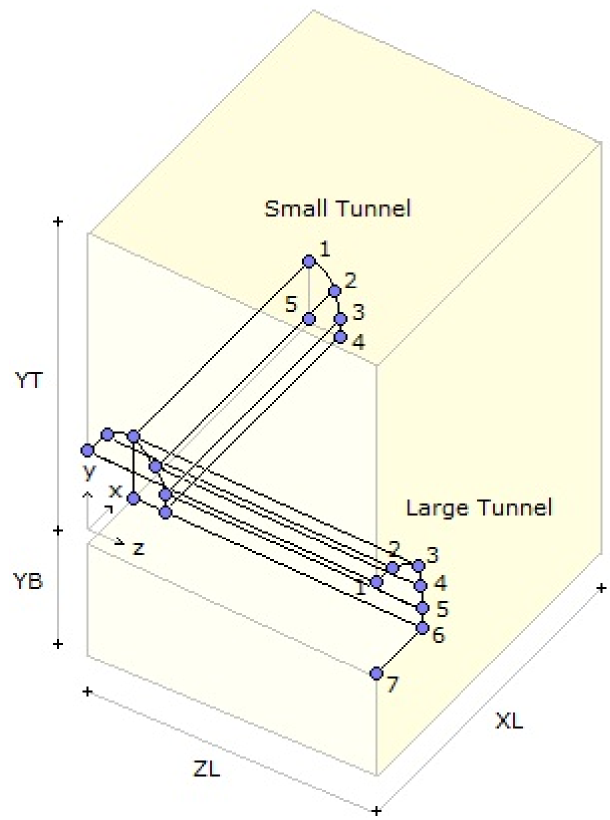


Figure 7.17 Schematic view of large and small crossing tunnels for MODELNO = 2

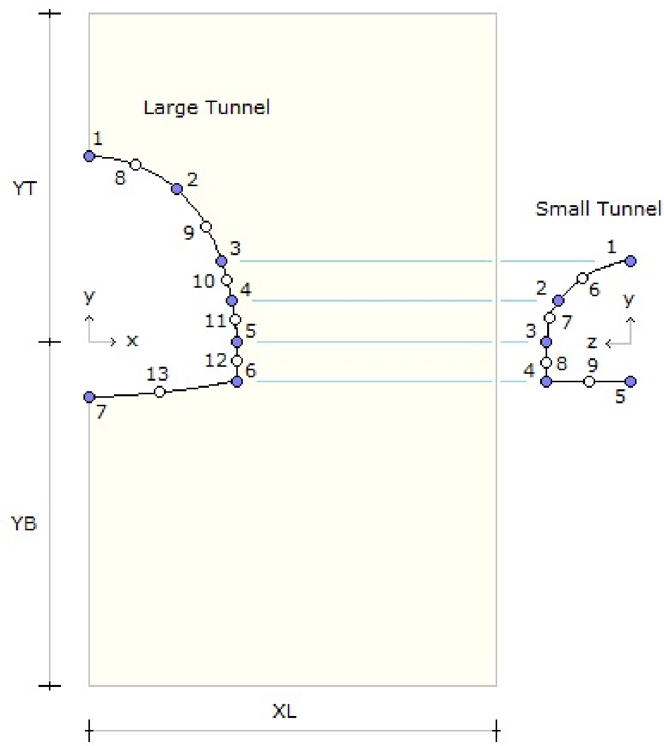


Figure 7.18 Node numbers defining tunnel shape for MODELNO = 2

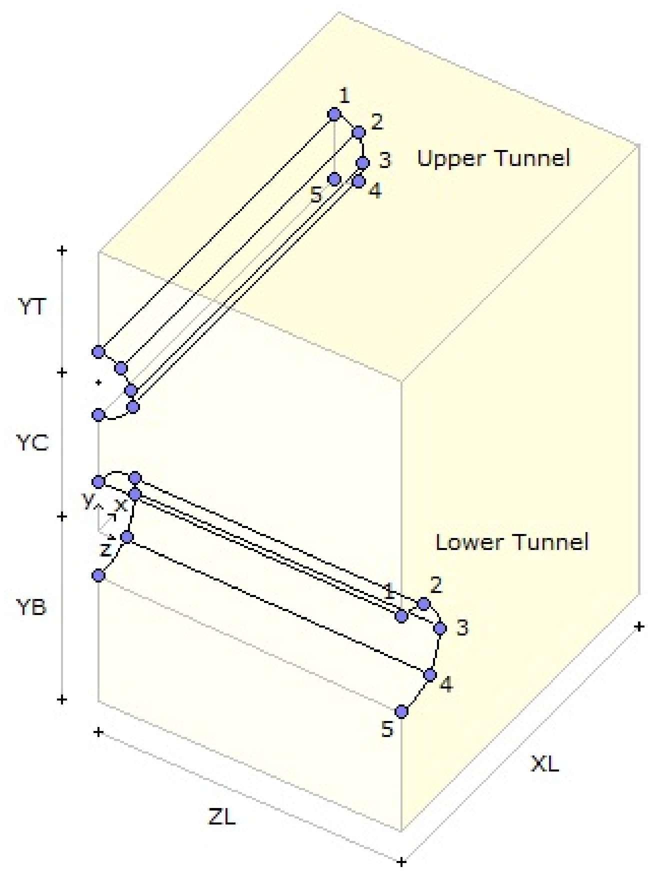


Figure 7.19 Schematic view of crossing tunnels with clearance for MODELNO = 3

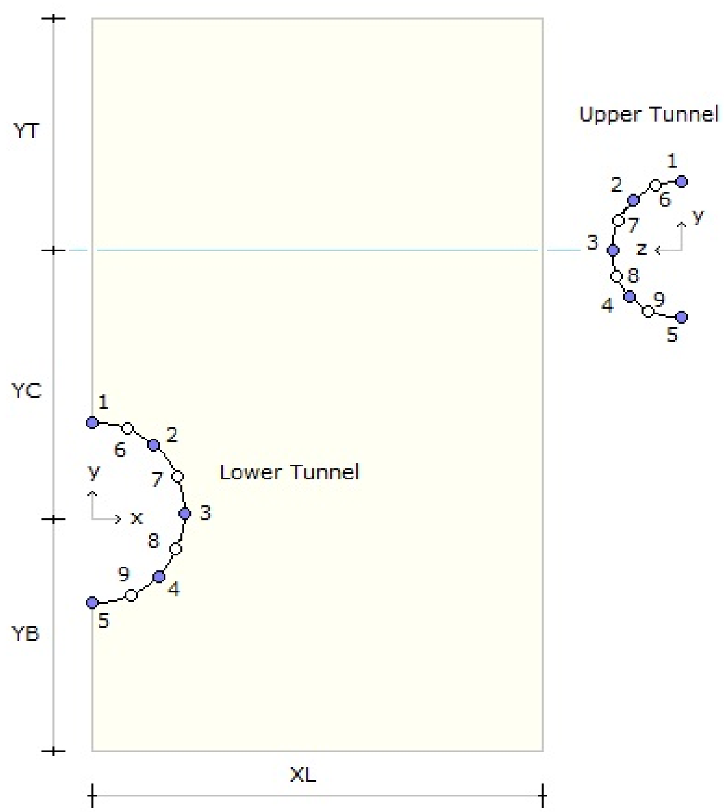


Figure 7.20 Node numbers defining tunnel shape for MODELNO = 3

GEN-3D

User's Manual

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td rowspan="3">1

General Information</td><td>1.1

TITLE
TITLE Any title (Max = 60 characters)</td></tr>
<tr><td>1.2

NBZ, NBNODE, NSNODE, NSNEL,
IBOUND, IPLANE, CLOSE, CMFAC

NBZ Number of blocks in z-direction
NBNODE Number of block nodes in z-direction
NSNODE Starting node number
NSNEL Starting element number

IBOUND = 0 Do not include control boundary (Default)
= 1 Include boundary as wire frame (Truss)
= 2 Include boundary as plane surface (Shell)
= 3 Include boundary as frame and surface

IPLANE = 0 Input 2D section in (X, Y) plane (Default)
= 1 Input 2D section in (-Z, Y) plane
= 2 Input 2D section in (X,-Z) plane
= 3 Input 2D section in specified plane

ICLOSE = 0 Open loop
= 1 Closed loop
First section represents last section

CMFAC Coordinate magnification factor for 2D sec.</td></tr>
<tr><td>1.2.1

If IBOUND = 0, skip this card
X_{LEFT}, X_{RIGHT}, Y_{BOTTOM}, Y_{TOP}, Z_{BACK}, Z_{FRONT}

X_{LEFT}, X_{RIGHT} X coordinates for left & right boundary
Y_{BOTTOM}, Y_{TOP} Y coordinates for bottom & top boundary
Z_{BACK}, Z_{FRONT} Z coordinates for back & front boundary</td></tr>
</table>

Card Group	Input Data and Definitions
1	1.2.2 Required only if IPLANE = 3 X_O, Y_O, Z_O X_a, Y_a, Z_a X_b, Y_b, Z_b X_O, Y_O, Z_O Coordinates defining local origin X_a, Y_a, Z_a Coordinates defining local x axis X_b, Y_b, Z_b Coordinates defining local y axis
General Information	1.3 IBZ_{BASE}, IBZ_{FRONT}, IBZ_{BACK} See Figure 7.21 IBZ_{BASE} Base boundary code IBZ_{FRONT} Front surface boundary code IBZ_{BACK} Back surface boundary code IBZ ISZ IFZ 0 0 0 1 0 1 2 1 0 3 1 1 ISZ Z DOF for skeleton motion IFZ Z DOF for relative pore fluid motion ISZ,IFZ = 0 Free to move in specified direction. = 1 Fixed in specified direction. For SMAP-T3 ID = ISZ and IDF = IFZ ID = 0 Heat flow is specified = 1 Temperature is specified IDF Time history identification number

Card Group	Input Data and Definitions
2 Block Coordinate	2.1 NBNODE Cards $NODE_1$, Z_1, X_1 $NODE_2$, Z_2, X_2 - - - - - - - - - NODE Node number Z Z coordinate X X coordinate Note: Z and X define the coordinates of center line

Card Group	Input Data and Definitions
3	3.1 BLNAME BLNAME Block name (up to 60 characters)
	3.2 IBLNO IBLNO Block number
Data for Each Block	3.3 I, J, LTYPE, IMATC, IMATB, IMATT, NIXCH (See Figure 7.21) I, J End node number of a block LTYPE = 0 Straight line = 1 Circular line IMATC Material number increment for Continuum IMATB Material number increment for Beam IMATT material number increment for Truss NIXCH Number of materials for index change
	3.4 NDZ, α, MC_1, MC_2, MC_3, MB, MT NDZ Number of elements in z-direction α = 0.5 Element length is constant = 0.3 Element length is growing from I to J = -0.3 Element length is growing from J to I MC Material number not to be modified for Continuum MB Material number not to be modified for Beam MT Material number not to be modified for Truss If MC/MB/MT has negative sign, that material will be removed
	3.5 Required only for LTYPE = 1 Z_o, X_o, R, θ_b, θ_e Z_o, X_o Coordinates of origin R Radius θ_b, θ_e Beginning and ending angle (°)

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions</th></tr>
<tr><td>3</td><td colspan="2">3.6
Required only for NIXCH > 0

NIXCH Cards MAT, NMAT, NI_1, NI_2, NI_3, NI_4, NI_5, NI_6, NI_7, NI_8

MAT Material number
NMAT New material number
NI_i Index number increment at NI_i

Note: Index change applied only for block first layer. If NMAT = -1, it assumes that new material property number 1 consists of joint elements whose joint face designates number KS = 6</td></tr>
<tr><td rowspan="3">4
Transmitting Boundary Generation</td><td colspan="2">4.1
ITRANB
ITRANB = 0 Do not generate transmitting boundary
= 1 Generate transmitting boundary
= 2 Generate element transmitting boundary

If ITRANB = 0, rest of Cards are not used
If ITRANB = 2, go to Card Group 4.4</td></tr>
<tr><td rowspan="2">4.2.
Material Property</td><td>4.2.1
NTNC
NTNC Number of material property set</td></tr>
<tr><td>4.2.21
NTNC Cards MAT, RHO, CP, CS

MAT Material number
RHO Mass density
CP Compression wave speed
CS Shear wave speed</td></tr>
</table>

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td>4

Transmitting Boundary Generation</td><td>4.3

Nodal Transmitting Boundary Generation
(Can be repeated in any order)

For surface whose normal is x-direction, 1
NPT
$N_1, N_2, \ldots, N_{NPT}$

For surface whose normal is y-direction, 2
NPT
$N_1, N_2, \ldots, N_{NPT}$

For surface whose normal is z-direction (Front Surface) 3

For surface whose normal is z-direction (Back Surface) 4
For end of transmitting boundary generation, 0

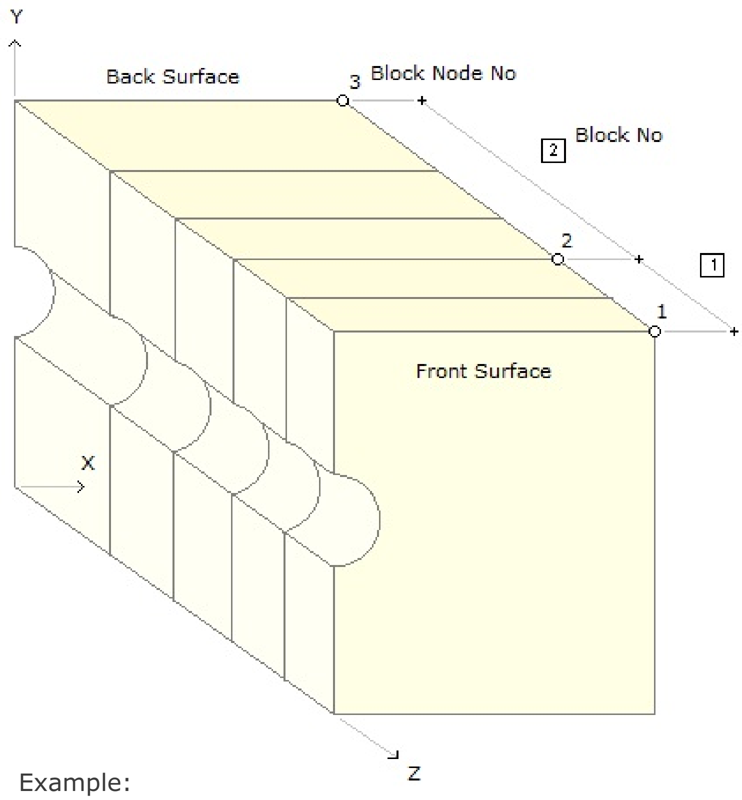
NPT Number of nodes
$N_1, N_2, \ldots, N_{NPT}$ Node numbers</td></tr>
<tr><td></td><td>4.4

Element Transmitting Boundary Generation
(Can be repeated in any order)

For surface whose normal is X-Y plane 1
NPT
$N_1, N_2, \ldots, N_{NPT}$
For front surface, 3

For back surface, 4
For end of transmitting boundary generation, 0

NPT Number of nodes
$N_1, N_2, \ldots, N_{NPT}$ Node numbers</td></tr>
</table>



Example:

IPLANE = 0 (Input 2D section in X-Y plane)

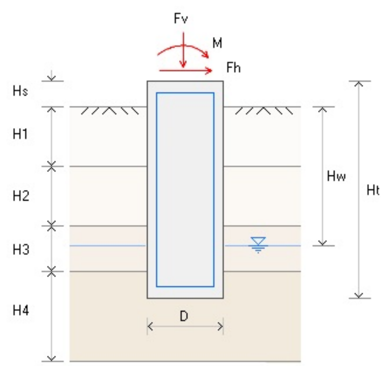
For Block No 1, I = 1 J = 2 NDZ = 2 α = 0.4
For Block No 2, I = 2 J = 3 NDZ = 3 α = 0.4

Total Number of Blocks, NBZ = 2
Total Number of Block Nodes, NBNODE = 3

Figure 7.21 Block index for GEN-3D

PILE-3D

User's Manual

Card Group	Input Data and Definitions
1 Title	1.1 Title Title Project title
	1.2 Stitle Stitle Project subtitle
2 Pile Properties	2.1 D, H_t, H_s, H_w, N_t D Pile diameter (m) H_t Pile length (m) H_s Pile length above ground surface (m) H_w Depth of water table (m) N_t Number of finite elements along the pile length  Fig. 1 Pile dimension
	2.2 E_p, v_p, t_p [Steel Pipe / Liner Plate] E_p Young's modulus (t/m^2) v_p Poisson's ratio t_p Thickness (mm) To exclude steel pipe, set t_p = 0.0

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td rowspan="5">2

Pile Properties</td><td>2.3
N_r, d_{top}, d_{bot}

N_r Number of reinforcing bar layers
d_{top} Top cover depth (cm)
d_{bot} Bottom cover depth (cm)
Note: Reinforcing bars are not considered, set $N_r = 0$</td></tr>
<tr><td>2.4
D_b, d_b, N_b [For Each Longitudinal Reinforcing Bar Layer]

D_b Diameter (mm)
d_b Cover depth (mm)
N_b Number of bars</td></tr>
<tr><td>2.5
E_c, v_c, φ, c, T, γ [Concrete]

E_c Young's modulus (t/m^2)
v_c Poisson's ratio
φ Internal friction angle ($^\circ$)
c Cohesion (t/m^2)
T Tensile strength (t/m^2)
γ Unit weight (t/m^3)</td></tr>
<tr><td>2.6
E_r, σ_y [Reinforcing Bar]

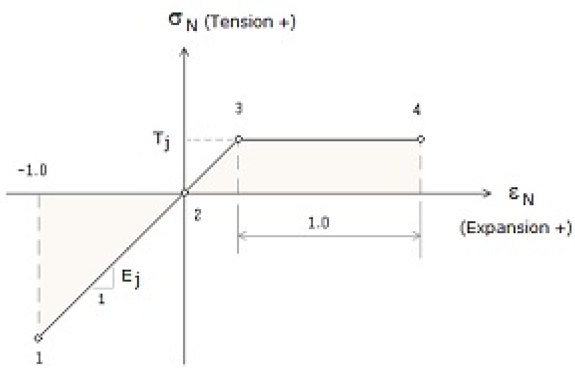
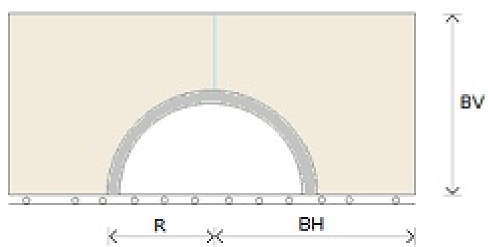
E_r Young's modulus (t/m^2)
σ_y Yield stress (t/m^2)</td></tr>
<tr><td>2.7
E_b, G_b, φ, c, T, t_b [Pile Base Interface]

E_b Young's modulus normal to the base (t/m^2)
G_b Shear modulus along the base (t/m^2)
φ Friction angle along the base ($^\circ$)
c Cohesion along the base (t/m^2)
T Tensile strength normal to the base (t/m^2)
t_b Thickness (mm)</td></tr>
</table>

Card Group		Input Data and Definitions
3	3.1	N_{LAYER} N_{LAYER} Total number of layers
Soil / Rock Layers	For Each Layer	3.2 L_{NO}, M_{NO}, H, γ_t, K_o L_{NO} Layer number M_{NO} = 1 Elastic model 3 Mohr-Coulomb model 4 In situ rock model 9 Modified Can-Clay model 12 Generalized decoupled hyperbolic model H Thickness of layer (m) γ_t Unit weight (t/m^3) K_o Coefficient of earth pressure at rest
		3.3 E_j, G_j, φ, c, T, t_j [Pile Side Interface, see Fig. 2] E_j Young's modulus normal to interface (t/m^2) G_j Shear modulus along the interface (t/m^2) φ Friction angle along the interface (°) c Cohesion along the interface (t/m^2) T Tensile strength normal to interface (t/m^2) t_j Thickness (mm)
		3.4.1 E, v [M_{NO} = 1: Elastic] E Young's modulus (t/m^2) v Poisson's ratio
		3.4.3 E, v, φ, c, T [M_{NO} = 3: Mohr-Coulomb] E Young's modulus (t/m^2) v Poisson's ratio φ Internal friction angle (°) c Cohesion (t/m^2) T Tensile strength (t/m^2)

Card Group		Input Data and Definitions
3 Soil / Rock Layers	For Each Layer	3.4.4 E, v, m, s, σ_c, T [M_{NO} = 4: In Situ Rock] E Young's modulus (t/m^2) v Poisson's ratio m,s Hoek & Brown material parameters σ_c Unconfined strength of intact rock (t/m^2) T Tensile strength (t/m^2)
		3.4.9 P_c, e_o, v, C_c, C_r, T [M_{NO} = 9: Modified Cam-Clay] P_c Preconsolidation pressure (t/m^2) e_o Initial void ratio v Poisson's ratio C_c Virgin compression index C_r Swelling / recompression index M Strength parameter
		3.4.12 E, v, φ, c, E_u, K [M_{NO} = 12: Generalized Decoupled] E Loading Young's modulus (t/m^2) v Poisson's ratio φ Internal friction angle (°) c Cohesion (t/m^2) E_u Unloading Young's modulus (t/m^2) K The ratio of shear strength in triaxial extension over triaxial compression at the same mean pressure

Card Group	Input Data and Definitions
4 External Loads	4.1 F_V, F_H, M, N_{STEP} F_V Vertical force (t) Compression is positive F_H Horizontal force (t) M Moment (t-m) N_{STEP} Number of computational steps through which external loads are applied
5 Anchor Bolt	5.1 D_a, d_a, L_a, N_{bolt}, E_a, σ_{ya} D_a Diameter (mm) d_a Cover depth (mm) L_a Length embedded in pile (m) N_{bolt} Number of bolts E_a Young's modulus (t/m^2) σ_{ya} Yield stress (t/m^2) Note: For N_{bolt} = 0, Concrete below GS is not active
6 F. E. Mesh on Plan View	6.1 FineMesh, NearMesh, N_{DIV}, B_H, B_V [See Fig. 3] FineMesh = 0 Coarse mesh 1 Fine mesh NearMesh = 0 All quad mesh 1 Quad and triangle mesh N_{DIV} Number of divisions for outer zone B_H Horizontal dimension (m) B_V Vertical dimension (m)

Card Group	Input Data and Definitions
Notes	 Fig. 2 Joint Material Normal Stress-Strain Relation  Fig. 3 Finite Element Mesh on Plan View

PRESMAP-GP

User's Manual

Card Group	Input Data and Definitions
1 General Information	1.1 TITLE TITLE Any title (Max = 80 characters) Note: Following two cards are required at the beginning StartPresmap VersionNo = 7.000
	1.2 NBLOCK, NBNODE, NSNODE, NSNEL, IGBND, ISMAP, CMFAC, ICOMP NBLOCK Number of blocks NBNODE Number of block nodes NSNODE Starting node number NSNEL Starting element number IGBND = 0 Do not generate = 1 Generate global boundary conditions based on Card 1.3 ISMAP = 1 Mesh generation for SMAP-S2 = 2 Mesh generation for SMAP-2D = -2 Mesh generation for SMAP-T2 = 3 Mesh generation for SMAP-3D & S3 = -3 Mesh generation for SMAP-T3 CMFAC Coordinate magnification factor ICOMP = 0 Do not impose = 1 Impose compatibility between blocks Note: If NBLOCK is negative value, the output file contains plotting information for block diagram

Card Group	Input Data and Definitions
1	1.3 Six cards starting from right, left, top, bottom, front, back For SMAP-S2/S3/2D/3D ISG, ISX, ISY,ISZ, IFG, IFX, IFY,IFZ, IRG, IRX, IRY,IRZ For SMAP-T2/T3 ITG, IDF, T, CF ISG, IFG, IRG = 0 None = 1 Free boundary = 2 Fixed boundary = 3 Roller boundary = 4 Specified in X, Y, Z directions ITG = 0 None = 1 Heat Flow = 2 Temperature IDF Time function identification number T Initial temperature CF Time function coefficient
General Information	1.4 ELMIN, MAXNEL ELMIN Minimum element length MAXNEL Maximum number of elements Note: ELMIN and MAXNEL are used in PLOT-3D as control parameters to generate automatically finite elements

Card Group	Input Data and Definitions
2 Block Coordinate	2.1 NBNODE Cards: $NODE_1$, X_1, Y_1, Z_1 $NODE_2$, X_2, Y_2, Z_2 - - - - - - - - NODE Node number X X-coordinate Y Y-coordinate Z Z-coordinate

Card Group	Input Data and Definitions
3 Data for Each Block	3.0 IBETYPE IBETYPE = 1 Line block (Beam or Truss Element) = 2 Quad surface block = -2 Triangle surface block Surface block generates plane strain/stress, or axisymmetric element for ISMAP = 1 or 2 and shell/ membrane element for ISMAP = 3 = 3 Hexahedron volume block = -3 Prism volume block. Volume block generates 3-D Continuum element or 3-D Joint element. Note: Card Group 3 requires following cards: At the beginning of each block StartBlock At the end of each block EndBlock At the end of last block EndOfLastBlock

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td>3</td><td>3.1

BLNAME

BLNAME Block name (Max = 60 characters)</td></tr>
<tr><td>Data for Each Line Block [IBETYPE =1]</td><td>3.2

ICOORD, IMODE, ILAG

Interpolation based on
ICOORD = 1 Rectangular coordinate
= 2 Spherical coordinate
= 3 Cylindrical coordinate

Modify generated coordinate
IMODE = 0 Do not modify
= 1 Modify using reference node (M_5)
as origin for ICOORD = 1.
Modify coordinate based on rectangular
grid for ICOORD = 2 or 3.

ILAG = 0 Generate Beam element
= 1 Generate Truss element</td></tr>
</table>

Card Group	Input Data and Definitions
3 Data for Each Line Block [IBETYPE =1]	3.3 I_1, I_2 M_3 M_4 M_5, M_6, M_7 See Figure 7.22 I_1 - I_2 Corner node number of a block M_3 Side node number of a block M_4 Reference node number For ICOORD = 2 M_5 Node number defining origin of spherical coordinate For ICOORD = 3 M_5 Node number defining reference origin of cylindrical coordinate M_6 Node number defining cylinder axis M_5 - M_6 M_7 Node number defining other local axis M_5 - M_7 which is normal to cylinder axis.

Card Group	Input Data and Definitions	
3 Data for Each Line Block [IBETYPE =1]	3.4	3.4.1 NBOUND NBOUND Number of boundaries to be specified If NBOUND = 0, go to Card group 3.5
		3.4.2 NBOUND cards For SMAP-S2/S3/2D/3D IBTYPE, ISX, ISY,ISZ, IFX, IFY,IFZ, IRX, IRY,IRZ For SMAP-T2/T3 IBTYPE, ID, IDF, T, CF IBTYPE = 1 Interior line = 2 Node I_1 = 3 Node I_2 = 4 Node M_4 Skeleton X, Y, Z DOF : ISX, ISY, ISZ Pore fluid X, Y, Z DOF relative to skeleton : IFX, IFY, IFZ Rotational DOF about X, Y, Z axis : IRX, IRY, IRZ ISX, ISY, ISZ, IFX, IFY, IFZ, IRX, IRY, IRZ = 0 Free to move in specified direction = 1 Fixed in specified direction Default boundary conditions ISX=ISY=ISZ=0, IFX=IFY=IFZ=1, IRX=IRY=IRZ=0 For SMAP-T2/T3 ID = 0 Heat flow is specified = 1 Temperature is specified IDF Time function identification number T Initial temperature CF Time function coefficient

Card Group	Input Data and Definitions
3 Data for Each Line Block [IBETYPE =1]	3.5 MATNO, NDX MATNO Material property number NDX Number of elements in x-direction

Card Group	Input Data and Definitions
3 Data for Each Quad Surface Block [IBETYPE =2]	3.1 BLNAME BLNAME Block name (Max = 60 characters)
	3.2 ICOORD, IMODE, ILAG Interpolation based on ICOORD = 1 Rectangular coordinate = 2 Spherical coordinate = 3 Cylindrical coordinate Modify generated coordinate IMODE = 0 Do not modify = 1 Modify using reference node (M_{10}) as origin for ICOORD = 1. Modify coordinate based on rectangular grid for ICOORD = 2 or 3. ILAG = 0 Serendipity interpolation = 1 Lagrangian interpolation = 2 Surface sector generation

Card Group	Input Data and Definitions
3 Data for Each Quad Surface Block [IBETYPE =2]	3.3 I_1, I_2, I_3, I_4 M_5, M_6, M_7, M_8 M_9 M_{10}, M_{11}, M_{12} See Figure 7.22 I_1 - I_4 Corner node number of a block M_5-M_8 Side node number of a block M_9 Center node number of a block, used for ILAG = 1 For ICOORD = 2 M_{10} Node number defining origin of spherical coordinate For ICOORD = 3 M_{10} Node number defining reference origin of cylindrical coordinate M_{11} Node number defining cylinder axis M_{10} - M_{11} M_{12} Node number defining other local axis M_{10} - M_{12} which is normal to cylinder axis

Card Group		Input Data and Definitions
3 Data for Each Quad Surface Block [IBETYPE =2]	3.4	3.4.1 NBOUND NBOUND Number of boundaries to be specified If NBOUND = 0, go to Card group 3.5
		3.4.2 NBOUND cards For SMAP-S2/S3/2D/3D IBTYPE, ISX, ISY,ISZ, IFX, IFY,IFZ, IRX, IRY,IRZ For SMAP-T2/T3 IBTYPE, ID, IDF, T, CF IBTYPE = 1 Interior surface = 2 Line I_1 - I_2 = 3 Line I_2 - I_3 = 4 Line I_3 - I_4 = 5 Line I_4 - I_1 = 6 Node I_1 = 7 Node I_2 = 8 Node I_3 = 9 Node I_4 Skeleton X, Y, Z DOF : ISX, ISY, ISZ Pore fluid X, Y, Z DOF relative to skeleton : IFX, IFY, IFZ Rotational DOF about X, Y, Z axis : IRX, IRY, IRZ ISX, ISY, ISZ, IFX, IFY, IFZ, IRX, IRY, IRZ = 0 Free to move in specified direction = 1 Fixed in specified direction Default boundary conditions ISX=ISY=ISZ=0, IFX=IFY=IFZ=1, IRX=IRY=IRZ=0 For SMAP-T2/T3 ID = 0 Heat flow is specified = 1 Temperature is specified IDF Time function identification number T Initial temperature CF Time function coefficient

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td>3

Data for Each Quad Surface Block [IBETYPE =2]</td><td>3.5

MATNO, NDX, NDY
NT_1, NT_2, NT_3, NT_4
MAT_1, MAT_2, MAT_3, MAT_4
THICK, DENSITY (For ISMAP = 1)
KS, KF (For ISMAP = 2)
IDH (For ISMAP =-2 or -3)

MATNO Material property number
NDX Number of elements in I_2 to I_1 direction
NDY Number of elements in I_2 to I_3 direction

NT For NT i is greater than zero, a triangle at block node i with NT i divisions along the triangle base. NT i ≤ min (NDX, NDY) and
NT i + NT j ≤ min (NDX, NDY) where
i =1, 2, 3, 4
j =2, 3, 4, 1
MAT_i Material property number for the triangle at block node i.
Zero value of MAT will remove the triangle.

THICK Thickness of element.
For plane strain, use THICK = 1.0
DENSITY Unit weight of element

KS = -1 Element has high explosive solid phase
= 0 Element has solid phase
> 0 Element has joint and absolute value of KS represents face designation number.

KF = 0 Element has fluid phase
= 1 Element has no fluid phase

IDH Heat generation history ID number</td></tr>
</table>

Card Group	Input Data and Definitions
3	3.6

Data for Each Quad Surface Block [IBETYPE =2]

Only for ICOORD = 2 and ILAG = 2

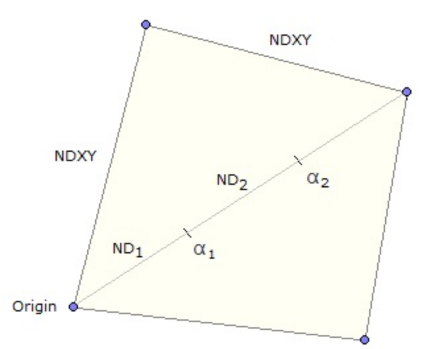
NSEG

NSEG Cards	$ALPA_1$,	$NDIV_1$
	$ALPA_2$,	$NDIV_2$
	-	-

NSEG	Number of segments
ALPA	Percent radial distance from origin
NDIV	Number of divisions between $ALPA_{i-1}$ and $ALPA_i$

Note: This option (ILAG=2) is to generate surface sector and has the following restrictions:

1. ICOORD = 2 (Spherical Coordinate)
2. IMOD = 0 Curved edge
   = 2 Straight edge
3. Midside and center nodes are not used.
4. NDX = NDY = NDXY = Σ $NDIV_i$

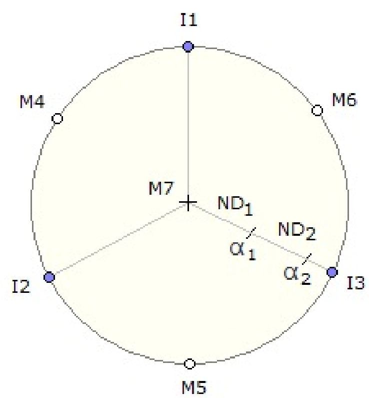


Card Group	Input Data and Definitions
3 Data for Each Triangle Surface Block [IBETYPE =-2]	3.1 BLNAME BLNAME Block name (Max = 60 characters)
	3.2 ICOORD, IMODE, ILAG Interpolation based on ICOORD = 1 Rectangular coordinate = 2 Spherical coordinate = 3 Cylindrical coordinate Modify generated coordinate IMODE = 0 Do not modify = 1 Modify using reference node (M_8) as origin for ICOORD = 1. Modify coordinate based on rectangular grid for ICOORD = 2 or 3. ILAG = 0 Serendipity interpolation = 1 Lagrangian interpolation = 2 Circular surface generation

Card Group	Input Data and Definitions
3 Data for Each Triangle Surface Block [IBETYPE =-2]	3.3 I_1, I_2, I_3 M_4, M_5, M_6 M_7 M_8, M_9, M_{10} See Figure 7.22 I_1 - I_3 Corner node number of a block M_4- M_6 Side node number of a block M_7 Center node number of a block for ILAG = 1 For ICOORD = 2 M_8 Node number defining origin of spherical coordinate For ICOORD = 3 M_8 Node number defining reference origin of cylindrical coordinate. M_9 Node number defining cylinder axis M_8-M_9 M_{10} Node number defining other local axis M_8-M_{10} which is normal to cylinder axis.

Card Group	Input Data and Definitions	
3 Data for Each Triangle Surface Block [IBETYPE=-2]	3.4	3.4.1 NBOUND NBOUND — Number of boundaries to be specified If NBOUND = 0, go to Card group 3.5
		3.4.2 NBOUND cards For SMAP-S2/S3/2D/3D IBTYPE, ISX, ISY,ISZ, IFX, IFY,IFZ, IRX, IRY,IRZ For SMAP-T2/T3 IBTYPE, ID, IDF, T, CF IBTYPE = 1 Interior surface = 2 Line I_1 - I_2 = 3 Line I_2 - I_3 = 4 Line I_3 - I_1 = 5 Node I_1 = 6 Node I_2 = 7 Node I_3 Skeleton X, Y, Z DOF : ISX, ISY, ISZ Pore fluid X, Y, Z DOF relative to skeleton : IFX, IFY, IFZ Rotational DOF about X, Y, Z axis : IRX, IRY, IRZ ISX, ISY, ISZ, IFX, IFY, IFZ, IRX, IRY, IRZ = 0 Free to move in specified direction = 1 Fixed in specified direction Default boundary conditions ISX=ISY=ISZ=0, IFX=IFY=IFZ=1, IRX=IRY=IRZ=0 For SMAP-T2/T3 ID = 0 Heat flow is specified = 1 Temperature is specified IDF — Time function identification number T — Initial temperature CF — Time function coefficient

Card Group	Input Data and Definitions
3	3.5
Data for Each Triangle Surface Block [IBETYPE =-2]	MATNO, NDXY THICK, DENSITY (For ISMAP = 1) KS, KF (For ISMAP = 2) IDH (For ISMAP =-2 or -3) MATNO Material property number NDXY Number of elements along triangle edge For wedge surface block, use negative NDXY Refer to Example problem 11 THICK Thickness of element. For plane strain, use THICK = 1.0 DENSITY Unit weight of element KS = -1 Element has high explosive solid phase = 0 Element has solid phase > 0 Element has joint and absolute value of KS represents face designation number. KF = 0 Element has fluid phase = 1 Element has no fluid phase IDH Heat generation history ID number

Card Group	Input Data and Definitions
3 Data for Each Triangle Surface Block [IBETYPE =-2]	3.6 Only for ICOORD = 2 and ILAG = 2 NSEG NSEG Cards: $ALPA_1$, $NDIV_1$ $ALPA_2$, $NDIV_2$ - - NSEG Number of segments ALPA Percent radial distance from origin NDIV Number of divisions between $ALPA_{i-1}$ and $ALPA_i$ Note: This option (ILAG = 2) is to generate circular surface and has the following restrictions: 1. ICOORD =2 (Spherical Coordinate) 2. IMOD = 0 Curved edge = 2 Straight edge 3. Block center node should be origin (M_7=M_8) 4. Midside nodes (M_4, M_5 and M_6) are interpolated based on spherical coordinate 

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td rowspan="2">3

Data for Each Hexahedron Volume Block [IBETYPE =3]</td><td>3.1

BLNAME

BLNAME Block name (Max = 60 characters)</td></tr>
<tr><td>3.2

ICOORD, IMODE, ILAG

Interpolation based on
ICOORD = 1 Rectangular coordinate
= 2 Spherical coordinate
= 3 Cylindrical coordinate

Modify generated coordinate
IMODE = 0 Do not modify
= 1 Modify using reference node (M_{28})
as origin for ICOORD = 1.
Modify coordinate based on rectangular
grid for ICOORD = 2 or 3.

ILAG = 0 Serendipity interpolation
= 1 Lagrangian interpolation</td></tr>
</table>

Card Group	Input Data and Definitions
3 Data for Each Hexahedron Volume Block [IBETYPE =3]	3.3 I_1, I_2, I_3, I_4, I_5, I_6, I_7, I_8 M_9, M_{10}, M_{11}, M_{12}, M_{13}, M_{14}, M_{15}, M_{16}, M_{17}, M_{18}, M_{19}, M_{20} M_{21}, M_{22}, M_{23}, M_{24}, M_{25}, M_{26}, M_{27} M_{28} M_{28}, M_{29}, M_{30} See Figure 7.22 I_1 - I_8 Corner node number of a block M_9 - M_{20} Side node number of a block M_{21} -M_{27} Side node number of a block required for Lagrangian interpolation For ICOORD = 2 or IMODE = 1 M_{28} Node number defining origin of spherical coordinate for ICOORD = 2, or node number defining reference origin to the whole volume for IMODE = 1 For ICOORD = 3 M_{28} Node number defining reference origin of cylindrical coordinate M_{29} Node number defining cylinder axis M_{28}-M_{29} M_{30} Node number defining other local axis M_{28}-M_{30} which is normal to cylinder axis

Card Group	Input Data and Definitions	
3	3.4	3.4.1 NBOUND NBOUND — Number of boundaries to be specified If NBOUND = 0, go to Card group 3.5
Data for Each Hexahedron Volume Block [IBETYPE =3]		3.4.2 NBOUND cards For SMAP-S2/S3/2D/3D IBTYPE, ISX, ISY,ISZ, IFX, IFY,IFZ, IRX, IRY,IRZ For SMAP-T2/T3 IBTYPE, ID, IDF, T, CF IBTYPE = 1 Interior Volume = 2 Front surface = 3 Back surface = 4 Left surface = 5 Right surface = 6 Top surface = 7 Bottom surface = 8 Line I_1 - I_2 = 9 Line I_2 - I_3 = 10 Line I_3 - I_4 = 11 Line I_4 - I_1 = 12 Line I_5 - I_6 = 13 Line I_6 - I_7 = 14 Line I_7 - I_8 = 15 Line I_8 - I_5 = 16 Line I_1 - I_5 = 17 Line I_2 - I_6 = 18 Line I_3 - I_7 = 19 Line I_4 - I_8 = 20 Node I_1 = 21 Node I_2 = 22 Node I_3 = 23 Node I_4 = 24 Node I_5

Card Group	Input Data and Definitions
3 Data for Each Hexahedron Volume Block [IBETYPE =3]	3.4.2 IBTYPE = 25 Node I_6 = 26 Node I_7 = 27 Node I_8 See Figure 7.23 Skeleton X, Y, Z DOF : ISX, ISY, ISZ Pore fluid X, Y, Z DOF relative to skeleton : IFX, IFY, IFZ Rotational DOF about X, Y, Z axis : IRX, IRY, IRZ ISX, ISY, ISZ, IFX, IFY, IFZ, IRX, IRY, IRZ = 0 Free to move in specified direction = 1 Fixed in specified direction Default boundary conditions ISX=ISY=ISZ=0, IFX=IFY=IFZ=1, IRX=IRY=IRZ=1 For SMAP-T2/T3 ID = 0 Heat flow is specified = 1 Temperature is specified IDF Time function identification number T Initial temperature CF Time function coefficient

Card Group	Input Data and Definitions
3	3.5
Data for Each Hexahedron Volume Block [IBETYPE =3]	MATNO, NDX, NDY, NDZ, KS, KF (For ISMAP = 3) MATNO, NDX, NDY, NDZ, IDH (For ISMAP =-3) NT_1, NT_2, NT_3, NT_4 MAT_1, MAT_2, MAT_3, MAT_4 MATNO Material property number NDX Number of elements in I_2 - I_1 direction NDY Number of elements in I_2 - I_3 direction NDZ Number of elements in I_2 - I_6 direction KS = -1 Element has high explosive solid phase = 0 Element has solid phase > 0 Element has joint and absolute value of KS represents face designation number. KF = 0 Element has fluid phase = 1 Element has no fluid phase IDH Heat generation history ID number NT & MAT See descriptions on page 7-92

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td rowspan="2">3

Data for Each Prism Volume Block [IBETYPE = -3]</td><td>3.1

BLNAME

BLNAME Block name (Max = 60 characters)</td></tr>
<tr><td>3.2

ICOORD, IMODE, ILAG

Interpolation based on
ICOORD = 1 Rectangular coordinate
= 2 Spherical coordinate
= 3 Cylindrical coordinate

Modify generated coordinate
IMODE = 0 Do not modify
= 1 Modify using reference node (M_{22}) as origin for ICOORD = 1
Modify coordinate based on rectangular grid for ICOORD = 2 or 3

ILAG = 0 Serendipity interpolation
= 1 Lagrangian interpolation</td></tr>
</table>

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td>3

Data for Each Prism Volume Block [IBETYPE =-3]</td><td>3.3

I_1, I_2, I_3, I_4, I_5, I_6
M_7, M_8, M_9, M_{10}, M_{11}, M_{12}, M_{13}, M_{14}, M_{15}, M_{16}, M_{17}
M_{18}, M_{19}, M_{20}, M_{21}
M_{22}, M_{23}, M_{24}

See Figure 7.22

I_1 - I_6 Corner node number of a block
M_7- M_{20} Side node number of a block
M_{21} Center node number of a block

For ICOORD = 2 or IMODE = 1
M_{22} Node number defining origin of spherical coordinate for ICOORD = 2, or node number defining reference origin to the whole volume for IMODE = 1

For ICOORD = 3
M_{22} Node number defining reference origin of cylindrical coordinate.
M_{23} Node number defining cylinder axis M_{22}-M_{23}
M_{24} Node number defining other local axis M_{22}-M_{24} which is normal to cylinder axis.</td></tr>
</table>

Card Group	Input Data and Definitions	
3 Data for Each Prism Volume Block [IBETYPE =-3]	3.4	3.4.1 NBOUND NBOUND Number of boundaries to be specified If NBOUND = 0, go to Card group 3.5
		3.4.2 NBOUND cards For SMAP-S2/S3/2D/3D IBTYPE, ISX, ISY,ISZ, IFX, IFY,IFZ, IRX, IRY,IRZ For SMAP-T2/T3 IBTYPE, ID, IDF, T, CF IBTYPE = 1 Interior volume = 2 Front surface = 3 Back surface = 4 Left surface = 5 Right surface = 6 Bottom surface = 7 Line I_1 - I_2 = 8 Line I_2 - I_3 = 9 Line I_3 - I_1 = 10 Line I_4 - I_5 = 11 Line I_5 - I_6 = 12 Line I_6 - I_4 = 13 Line I_1 - I_4 = 14 Line I_2 - I_5 = 15 Line I_3 - I_6 = 16 Node I_1 = 17 Node I_2 = 18 Node I_3 = 19 Node I_4 = 20 Node I_5 = 21 Node I_6 See Figure 7.24

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td rowspan="2">3

Data for Each Prism Volume Block [IBETYPE =-3]</td><td>3.4.2

Skeleton X, Y, Z DOF : ISX, ISY, ISZ
Pore fluid X, Y, Z DOF relative to skeleton : IFX, IFY, IFZ
Rotational DOF about X, Y, Z axis : IRX, IRY, IRZ

ISX, ISY, ISZ, IFX, IFY, IFZ, IRX, IRY, IRZ
= 0 Free to move in specified direction
= 1 Fixed in specified direction

Default boundary conditions
ISX=ISY=ISZ=0, IFX=IFY=IFZ=1, IRX=IRY=IRZ=1

For SMAP-T2/T3
ID = 0 Heat flow is specified
= 1 Temperature is specified
IDF Time function identification number
T Initial temperature
CF Time function coefficient</td></tr>
<tr><td>3.5

MATNO, NDXY, NDZ, KS, KF (For ISMAP = 3)
MATNO, NDXY, NDZ, IDH (For ISMAP =-3)

MATNO Material property number
NDXY Number of elements along triangular edge
For wedge volume block, use negative NDXY
Refer to Example problem 11
NDZ Number of elements in z-direction

KS = -1 Element has high explosive solid phase
= 0 Element has solid phase
> 0 Element has joint and absolute value of KS represents face designation number.

KF = 0 Element has fluid phase
= 1 Element has no fluid phase

IDH Heat generation history ID number</td></tr>
</table>

Note: Mesh Control Data on File DV-GP.DAT

To control mesh generation, users can change the values in file DV-GP.DAT in the directory C:\SMAP\CT\CTDATA.

1. Variables Controlling Coinsident Nodes
   RLIMIT
   When the distance between two adjacent nodes is less than RLIMIT, those two nodes are assumed to be coinsident.

2. Variables Contolling Spherical Coordinate
   SDCLOSE, SDTOL, SDZERO
   When the angle of block corner node reaches SDCLOSE (degree), program will set 360 degrees. The tolerance angle is SDTOL (degree). When the angle of block corner node is greater than (360-SDZERO), program will set zero degree.

3. Variables Contolling Cylindrical Coordinate
   CDCLOSE, CDTOL, CDZERO
   When the angle of block corner node reaches CDCLOSE (degree), program will set 360 degrees. The tolerance angle is CDTOL (degree). When the angle of block corner node is greater than (360-CDZERO), program will set zero degree.

4. For spherical block having the angle of longitude greater than π and for the cylindrical block occupying more than two quadrants, the block node numbers referring to the origin should be prefixed by negative sign.

5. Current Default Values
   RLIMIT = 0.001
   SDCLOSE = 359.1   SDTOL = 0.001   SDZERO = 0.001
   CDCLOSE = 359.1   CDTOL = 0.001   CDZERO = 0.001

Note: Boundary Conditions

Boundary conditions at nodes are generated based on following rules:

1. Default conditions are applied first based on block type

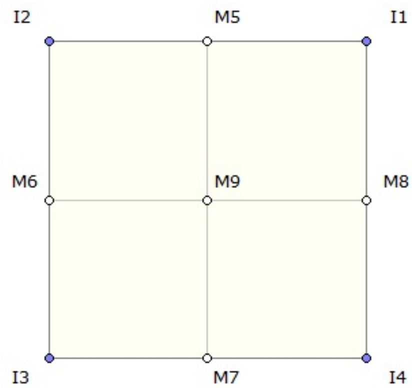
2. Default conditions can be overrided by specifying IBTYPE = 1

3. Higher IBTYPE overrides lower IBTYPE in a given block

4. Each block number defined later governs conditions along the block interface

Line Block

Quad Surface Block



Triangle Surface Block

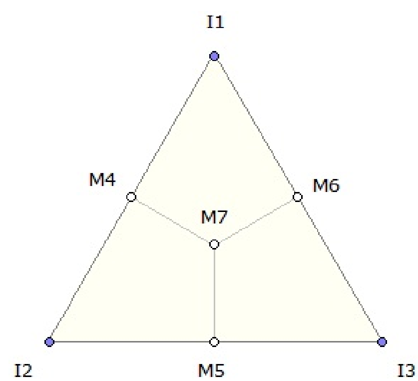


Figure 7.22 Block index for PRESMAP-GP

Hexahedron Volume Block

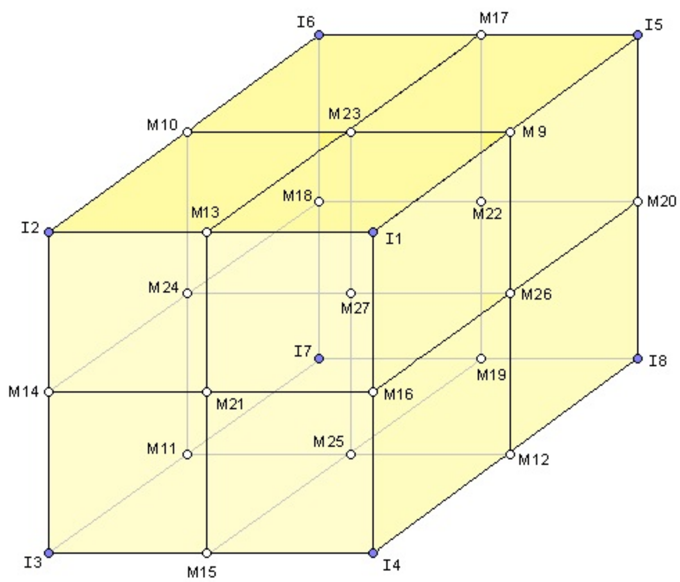


Figure 7.22 Block index for PRESMAP-GP (Continued)

Prism Volume Block

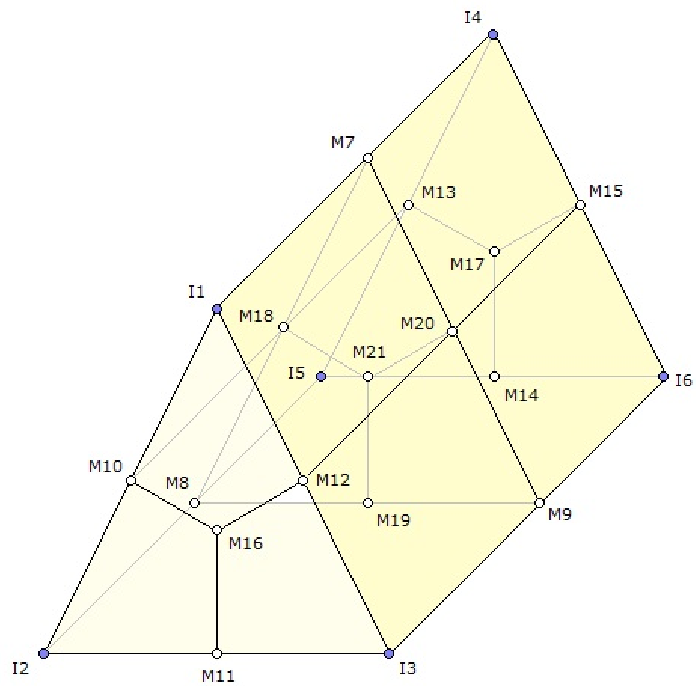


Figure 7.22 Block index for PRESMAP-GP (Continued)

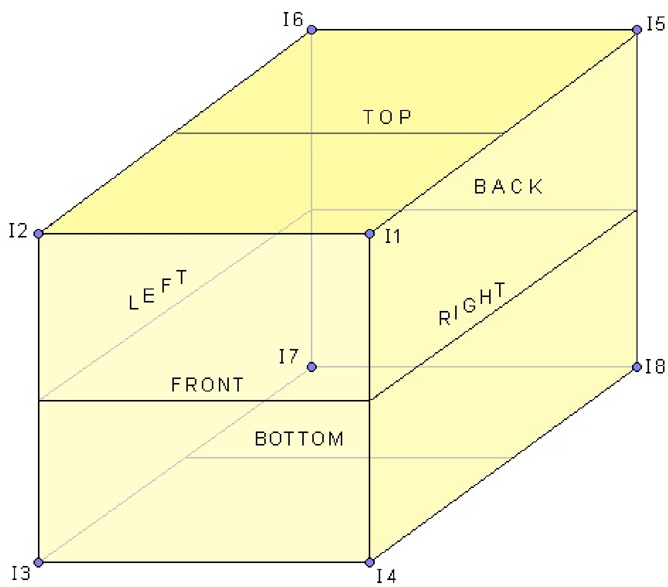


Figure 7.23 Boundary surface designation for Hexahedron Volume Block

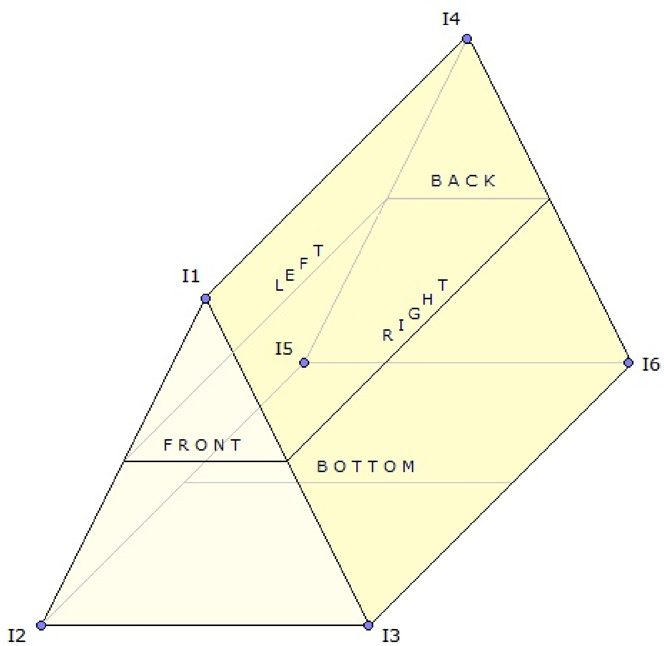


Figure 7.24 Boundary surface designation for Prism Volume Block

JOINT-2D

User's Manual

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td>1

General Information</td><td>1.1

TITLE

TITLE Any title of up to 80 characters</td></tr>
<tr><td></td><td>1.2

AllJoint, ThicAJ

AllJoint = 0 Generates Joint Elements along the All Interfaces between Continuum Elements. Cards 2, 3, and 4 are not used.

= 1 Generates Joint Elements for the Material Numbers of Continuum Elements as specified in Cards 2 and 3. Card 4 is not used.

= 2 Generates Joint Elements for the Element Surface Numbers of Continuum Elements as specified in Card 4. Cards 2 and 3 are ignored.

ThicAJ Thickness used for AllJoint = 0.

To Run JOINT-2D

Method 1

SMAP-2D > Run > Mesh Generator > PreSmap > Joint
Specify input and output file names shown on the screen.

Method 2

1. Select SMAP-2D > Setup > PLOT 3D
Specify Joint Thickness View Factor which is greater than 0.0
Example: Joint Thickness View Factor = 1.0

2. Select SMAP-2D > Mesh > F.E. Mesh > Open
This wil open Mesh File of Continuum Elements.

Input file Joint.inp should exist in the Working Directory.
Output File JointedMesh.Mes is shown in Working Directory.</td></tr>
</table>

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions</th></tr>
<tr><td rowspan="4">AllJoint = 1: Internal / Boundary Joint Generation</td><td rowspan="2">Internal Joint Generation</td><td>2.1
NumIJ, ThicIJ
NumIJ Number of continuum materials for Internal Joint. If NumIJ = 0, go to Card 3
ThicIJ Thickness of Internal Joints</td></tr>
<tr><td>2.2
NumIJ Cards: $MatIJ_1$ $InnerBeam_1$ $OuterBeam_1$; $MatIJ_2$ $InnerBeam_2$ $OuterBeam_2$; - - -
MatIJ Material property number of continuum element for Internal Joints (See Fig. 1)
InnerBeam = 0 Do not include; = 1 Include Inner Beam element
OuterBeam = 0 Do not include; = 1 Include Outer Beam element</td></tr>
<tr><td rowspan="2">Boundary Joint Generation</td><td>3.1
NumBJ, ThicBJ, InterfaceJoint
NumBJ Number of continuum materials for Boundary Joint. If NumBJ = 0, go to Card 4
ThicBJ Thickness of Boundary Joints. If negative, inside continuum elem. contacts joint face
InterfaceJoint = 0 Do not include; = 1 Include Interface Joint Element</td></tr>
<tr><td>3.2
NumBJ Cards: $MatBJ_1$ $InnerBeam_1$ $OuterBeam_1$; $MatBJ_2$ $InnerBeam_2$ $OuterBeam_2$; - - -
MatBJ Material property number of continuum element for Boundary Joints (See Fig. 1)
InnerBeam = 0 Do not include; = 1 Include Inner Beam element
OuterBeam = 0 Do not include; = 1 Include Outer Beam element</td></tr>
</table>

Card Group	Input Data and Definitions
4 AllJoint = 2 : Surface Joint Generation	4.1 NumSJG NumSJG — Number of Groups for Surface Joints If NumSJG = 0, end of data
	4.2 NumSJG Cards: $NumSJ_1$ $ThicSJ_1$ $NumSJ_2$ $ThicSJ_2$ - - - - $NumSJ_i$ — Number of element surfaces in Group i $ThicSJ_i$ — Thickness of Surface Joint in Group i
For Each Surface Joint Group	4.3 $NumSJ_i$ Cards: $ElementNo_1$ $SurfaceNo_1$ $ElementNo_2$ $SurfaceNo_2$ - - - - ElementNo — Continuum Element No SurfaceNo — Continuum Element Surface No where Surface Joint is generated Note: To take new node number for corner contact element, set SurfaceNo = 0 Refer to page 4-67 of SMAP-2D User's Manual for Element Surface designation

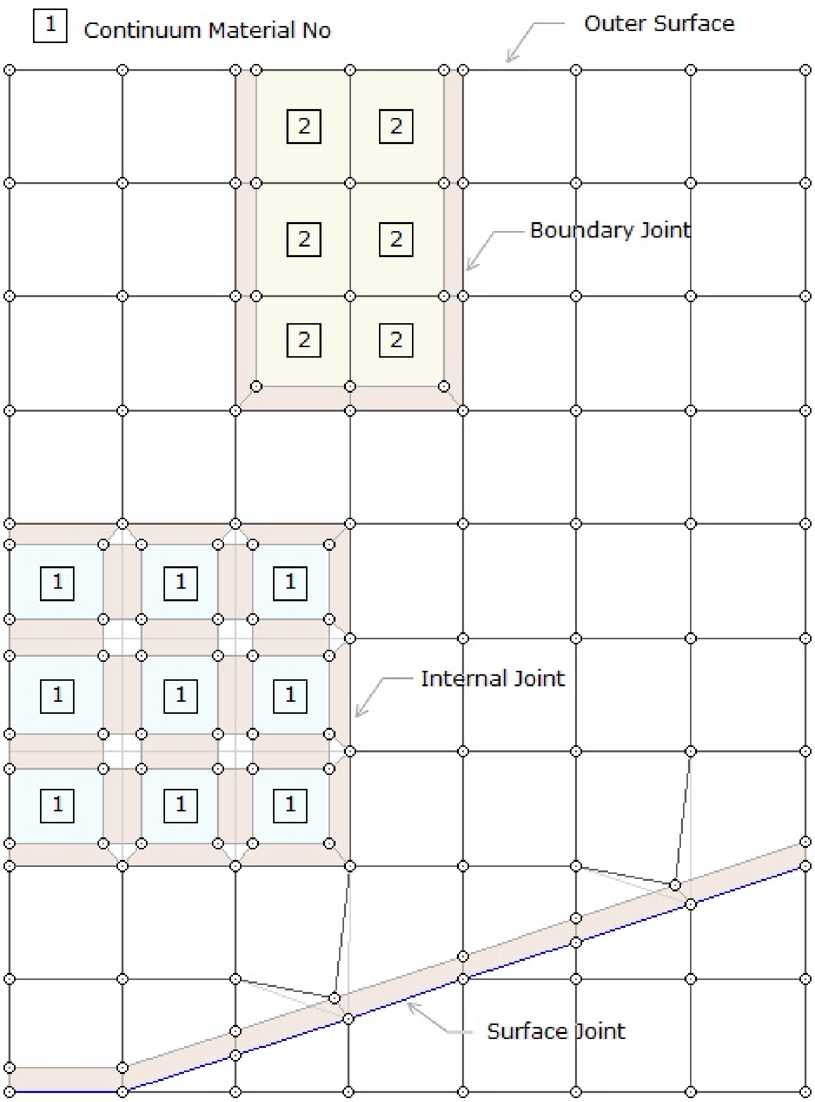


Figure 1 Joint element generation

JOINT-3D

User's Manual

Card Group	Input Data and Definitions
1 General Information	1.1 TITLE TITLE Any title of up to 80 characters
	1.2 AllJoint, ThicAJ AllJoint = 0 Generates Joint Elements along the All Interfaces between Continuum Elements. Cards 2, 3, and 4 are not used. = 1 Generates Joint Elements for the Material Numbers of Continuum Elements as specified in Cards 2 and 3. Card 4 is not used. = 2 Generates Joint Elements for the Element Surface Numbers of Continuum Elements as specified in Card 4. Cards 2 and 3 are ignored. ThicAJ Thickness used for AllJoint = 0. To Run JOINT-3D **Method 1** SMAP-3D > Run > Mesh Generator > PreSmap > Joint Specify input and output file names shown on the screen. **Method 2** 1. Select SMAP-3D > Setup > PLOT 3D Specify Joint Thickness View Factor which is greater than 0.0 Example: Joint Thickness View Factor = 1.0 2. Select SMAP-3D > Mesh > F.E. Mesh > Open This wil open Mesh File of Continuum Elements. Input file Joint.inp should exist in the Working Directory. Output File JointedMesh.Mes is shown in Working Directory.

Card Group		Input Data and Definitions
AllJoint = 1: Internal / Boundary Joint Generation	Internal Joint Generation	2.1 NumIJ, ThicIJ NumIJ — Number of continuum materials for Internal Joint. If NumIJ = 0, go to Card 3 ThicIJ — Thickness of Internal Joints
		2.2 NumIJ Cards: $MatIJ_1$ $InnerShell_1$ $OuterShell_1$ $MatIJ_2$ $InnerShell_2$ $OuterShell_2$ - - - MatIJ — Material property number of continuum element for Internal Joints (See Fig. 1) InnerShell = 0 Do not include = 1 Include Inner Shell element OuterShell = 0 Do not include = 1 Include Outer Shell element
	Boundary Joint Generation	3.1 NumBJ, ThicBJ, InterfaceJoint NumBJ — Number of continuum materials for Boundary Joint. If NumBJ = 0, go to Card 4 ThicBJ — Thickness of Boundary Joints. If negative, inside continuum elem. contacts joint face. InterfaceJoint = 0 Do not include = 1 Include Interface Joint Element
		3.2 NumBJ Cards: $MatBJ_1$ $InnerShell_1$ $OuterShell_1$ $MatBJ_2$ $InnerShell_2$ $OuterShell_2$ - - - MatBJ — Material property number of continuum element for Boundary Joints (See Fig. 1) InnerShell = 0 Do not include = 1 Include Inner Shell element OuterShell = 0 Do not include = 1 Include Outer Shell element

Card Group	Input Data and Definitions
4 AllJoint = 2 : Surface Joint Generation	4.1 NumSJG NumSJG — Number of Groups for Surface Joints If NumSJG = 0, end of data
	4.2 NumSJG Cards: $NumSJ_1$ $ThicSJ_1$; $NumSJ_2$ $ThicSJ_2$; - -; - - $NumSJ_i$ — Number of element surfaces in Group i $ThicSJ_i$ — Thickness of Surface Joint in Group i
For Each Surface Joint Group	4.3 $NumSJ_i$ Cards: $ElementNo_1$ $SurfaceNo_1$; $ElementNo_2$ $SurfaceNo_2$; - -; - - ElementNo — Continuum Element No SurfaceNo — Continuum Element Surface No where Surface Joint is generated Note: To take new node number for corner contact element, set SurfaceNo = 0 Refer to page 4-68 of SMAP-3D User's Manual for Element Surface designation

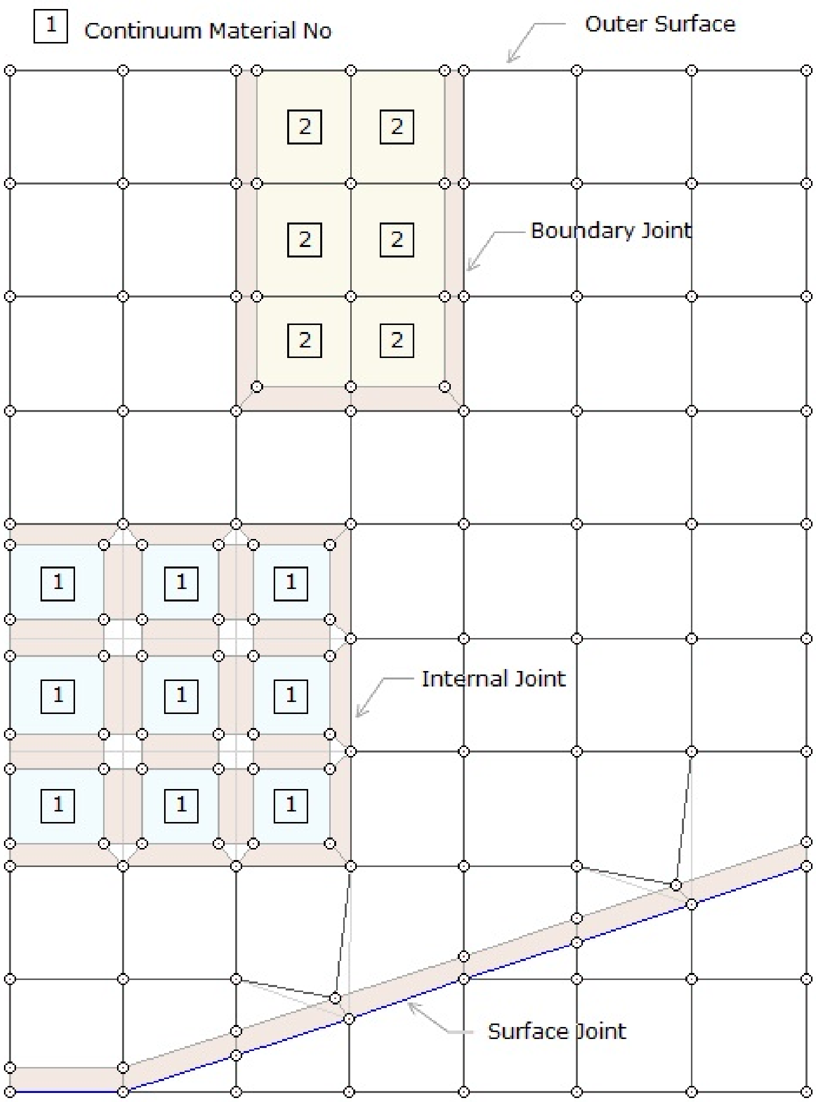


Figure 1 Joint element generation

INTERSECTION
User's Manual

Introduction

INTERSECTION programs are mainly used to compute the locations of the 3D surfaces crossing each other. These surfaces consist of Shell Elements with different materials. The computed coordinates of intersections can be used for the construction of complicated three-dimensional meshes.

There are two INTERSECTION programs provided in this manual; SHELL ELEMENT and TWO TUNNELS.

7.12.1 SHELL ELEMENT

SHELL ELEMENT intersection is the basic program which can be applied to find the line of intersection of three-dimensional surfaces.

First, you need to prepare a SMAP-3D mesh file composed of Shell Elements with different material numbers.

SHELL ELEMENT intersection can be accessed by selecting menu
Run → Mesh Generator → PreSmap → Intersection → Shell Element

or
Setup → PLOT 3D → Compute Intersection → Yes
and then open mesh file Plot → Mesh → F. E. Mesh → Open

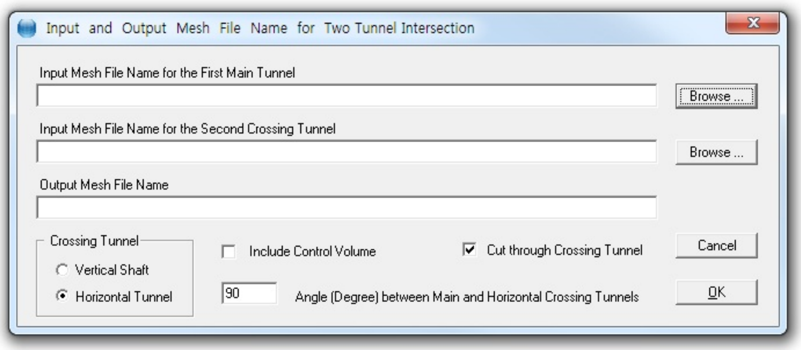
Note that computed coordinates of intersections are represented by Truss Elements.

7.12.2 TWO TUNNELS

TWO TUNNELS intersection is the special program where the second crossing tunnel cuts through the first main tunnel at some angle.

First, you need to prepare two SMAP-2D mesh files representing for cross sections of the first main and the second crossing tunnels. These cross sections are modeled by two-dimensional Beam Elements with different beam section numbers.

TWO TUNNELS intersection can be accessed by selecting menu
Run → Mesh Generator → PreSmap → Intersection → Two Tunnels



Browse the input files for Main and Crossing tunnels.
Select the Vertical Shaft or Horizontal Tunnel at some angles.

Main and Crossing tunnels are extended to three dimensional Shell elements using the default GEN-3D input file; ZI-A.dat and ZI-B.dat, respectively, in the sub directory Temp.

Note that output file Intersection.Mes contains Shell Elements representing both main and crossing tunnels.

For best appearance of generated meshes, you need to copy C:\SMAP\CT\CTDATA\DV-ADRGN.dat into Working Directory and then modify control parameters in Intersection for PLOT-3D.

ADDRGN
User's Manual

8.1 Introduction

ADDRGN is the pre-processing program which has the following two basic functions:

- Combine two different meshes
- Modify existing meshes

A problem geometry can be composed of a number of regions. Parts of the problem geometry can be generated using the PRESMAP programs described in Section 7. Then ADDRGN is used to combine two different regions (Region A and Region B). When Region B is added to Region A to make Combined Region, following restrictions are applied:

- Element numbers for Region A and Region B should be continuous
- Only those node numbers for Region B are modified to be consistent with the Region A, but element numbers for both regions do not change.

Though the program ADDRGN combines only two regions at a time, users can apply ADDRGN many times to assemble all the different regions.

ADDRGN can also be used to modify the existing meshes:

- Change coordinates
- Change boundary codes
- Cut elements
- Change material numbers

ADDRGN-2D deals with two dimensional meshes and ADDRGN-3D deals with three dimensional meshes.

ADDRGN-2D has an additional powerful feature which is very useful to generate meshes for complicated underground structures. This special feature modifies the existing meshes such that new structures can be easily added by simply specifying the geometries and material properties of structures. It can even generate a base mesh and then add new structures (IMOD=2).

ADDRGN-2D

User's Manual

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td>1
Mode Type</td><td>1.1
IMOD, JK
IMOD = 0 Add Region B to Region A
= 1 Modify existing mesh
= 2 Generate base mesh and then modify. Generated base mesh is saved as BMESH.Dat
=-1 Same as IMOD = 0 except it uses DOF of Region B mesh along the interface

JK 1 (T2), 2 (S2), 3 (2D), 9 (W2)</td></tr>
<tr><td>2
Adding Region B to Region A (IMOD = 0)</td><td>2.1
FILEA
FILEB
FILEC
FILEA Input file name containing Region A mesh
FILEB Input file name containing Region B mesh
FILEC Output file name to store Combined Region mesh

When combining Region B mesh to Region A mesh, only Region B node numbers are changed. Element numbers for Region A and Region B should be continuous, otherwise element numbers are automatically reordered by program.</td></tr>
<tr><td></td><td>2.2
INTERFACE
INTERFACE = 0 Interface is found automatically
= 1 Interface is specified by user</td></tr>
<tr><td></td><td>2.3
Required only for INTERFACE = 1
NODE
$NODA_1$, $NODA_2$, ..., $NODA_{NODE}$
$NODB_1$, $NODB_2$, ..., $NODB_{NODE}$

NODE Number of interface nodes.
$NODA_i$ Interface node numbers in Region A
$NODB_i$ Interface node numbers in Region B

Note: $NODB_i$ should be the same location as $NODA_i$</td></tr>
</table>

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td rowspan="3">3

Modifying Existing Mesh (IMOD =1)</td><td>3.1
FILEA
FILEM

FILEA Input file name containing existing mesh
FILEM Output file name to store modified mesh</td></tr>
<tr><td>3.2
NSNEL, NSNODE, NBNEL, NTNEL

NSNEL New starting continuum element number
NSNODE New starting node number
NBNEL New starting beam element number
NTNEL New starting truss element number
Note: NBNEL & NTNEL are used for IEDIT = 0, 1, 6</td></tr>
<tr><td>3.3
IEDIT, MC_1, MC_2, MC_3, MB, MT

IEDIT = 0 Change coordinates
= 1 Change boundary codes
= 2 Cut elements
= 3 Change material numbers
= 4 Build user-defined curves and material zones
= 6 Change element index order

MC Continuum material number to be kept
MB Beam material number to be kept
MT Truss material number to be kept

Note: MC, MB, and MT are applicable
only for IEDIT = 2 and 3</td></tr>
</table>

Card Group		Input Data and Definitions
3 Modifying Existing Mesh (IMOD = 1)	Changing Coordinates (IEDIT = 0)	3.3.1.1 X_o, Y_o, X_{oNew}, Y_{oNew} X_o, Y_o Reference origin X_{oNew}, Y_{oNew} New origin
		3.3.1.2 X_{scale}, Y_{scale} X_{scale}, Y_{scale} Scale factors for X, Y coordinates Note: New coordinates $X_{(new)}$ and $Y_{(new)}$ are computed as follows: $X_{(new)} = X_{oNew} + (X-X_0)\ X_{scale}$ $Y_{(new)} = Y_{oNew} + (Y-Y_0)\ Y_{scale}$

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions</th></tr>
<tr><td rowspan="5">3
Modifying Existing Mesh (IMOD = 1)</td><td rowspan="5">Changing Boundary Codes (IEDIT = 1)</td><td>3.3.2.1
IRANGE
IRANGE = 0 Range specified by coordinates
= 1 Range specified by node numbers
= 2 Range specified by line strip
= 3 Range specified by material numbers</td></tr>
<tr><td>3.3.2.2.1
Required only for IRANGE = 0
X_{start}, Y_{start}, X_{end}, Y_{end}
X_{start}, Y_{start} Coordinates for lower left boundary
X_{end}, Y_{end} Coordiantes for upper right boundary</td></tr>
<tr><td>3.3.2.2.2
Required only for IRANGE = 1, 2, 3
NODE
NOD_1, NOD_2, ..., NOD_{NODE}
NODE Number of nodes/materials to be specified
NOD_i Node/Material number (Note 1 in page 8-7)
Line strip is defined counterclockwise.
For IRANGE = 3, Nodes refer to Material numbers.</td></tr>
<tr><td>3.3.2.3
INSIDE (Not applicable for IRANGE= 3)
INSIDE = 0 Apply inside of range
= 1 Apply outside of range</td></tr>
<tr><td>3.3.2.4
ISX, ISY, IFX, IFY, IRZ (SMAP-2D)
IDX, IDY, IDT (SMAP-S2)
ID, IDF (SMAP-T2)

ISX, ISY X and Y DOF for skeleton motion
IFX, IFY X and Y DOF for relative motion
IRZ Z DOF for beam rotation

IDX, IDY X and Y DOF for skeleton motion
IDT Z DOF for beam rotation

ID Heat flow (0), Temperature (1) specified
IDF Time history identification number</td></tr>
</table>

Card Group		Input Data and Definitions
3 Modifying Existing Mesh (IMOD = 1)	Cutting Elements (IEDIT = 2)	3.3.3.1 IRANGE IRANGE = 0 Range specified by coordinates = 1 Range specified by element numbers
		3.3.3.2.1 Required only for IRANGE = 0 X_{start}, Y_{start}, X_{end}, Y_{end} X_{start}, Y_{start} Coordinates for lower left boundary X_{end}, Y_{end} Coordinates for upper right boundary
		3.3.3.2.2 Required only for IRANGE = 1 NOEL NEL_1, NEL_2, ..., NEL_{NOEL} NOEL Number of elements to be specified NEL_i Element number (See Note 2)
		3.3.3.3 INSIDE INSIDE = 0 Apply inside of range = 1 Apply outside of range Note 1: NOD_1, $-NOD_2$ generates from NOD_1 to NOD_2 Note 2: NEL_1, $-NEL_2$ generates from NEL_1 to NEL_2

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions</th></tr>
<tr><td rowspan="5">3

Modifying Existing Mesh (IMOD = 1)</td><td rowspan="5">Change Material No (IEDIT = 3)</td><td>3.3.4
IRANGE

IRANGE = 0 Range specified by coordinates
= 1 Range specified by element numbers</td></tr>
<tr><td>3.3.4.1
Required only for IRANGE = 0
X_{start}, Y_{start}, X_{end}, Y_{end}

X_{start}, Y_{start} Coordinates for lower left boundary
X_{end}, Y_{end} Coordiantes for upper right boundary</td></tr>
<tr><td>3.3.4.2
Required only for IRANGE = 1
NOEL
NEL_1, NEL_2, ..., NEL_{NODE}

NOEL Number of elements to be specified
NEL_i Element number (See Note 2 in page 8-7)</td></tr>
<tr><td>3.3.4.3
INSIDE
INSIDE = 0 Apply inside of range
= 1 Apply outside of range</td></tr>
<tr><td>3.3.4.4
MATC, MATB, MATT

MATC New continuum material number
MATB New beam material number
MATT New truss material number

Note: When new material number is zero,
keep the old material number</td></tr>
</table>

Card Group		Input Data and Definitions
3 Modifying Existing Mesh (IMOD = 1)	3.3.5 Build User-Defined Curves and Material Zones (IEDIT = 4)	3.3.5.1 NODE NOD_1, NOD_2, ..., NOD_{NODE} NODE — Number of nodes which are not movable NOD_i — Node number
		3.3.5.2 NOEL NEL_1, NEL_2, ..., NEL_{NOEL} NOEL — Number of elements whose nodal coordinates are not movable NEL_i — Element number
		3.3.5.3 IBOUND IBOUND = 0 Do not apply = 1 Nodal coordinates outside of rectangle are not movable Required only for IBOUND = 1 X_{LEFTt}, X_{RIGHT}, Y_{BOTTOM}, Y_{TOP} X_{LEFTt}, X_{RIGHT}, Y_{BOTTOM}, Y_{TOP} — Coordinates of rectangle
		3.3.5.4 NGROUP, IGTITL X_{REF}, Y_{REF} NGROUP — Number of curve groups. X_{REF}, Y_{REF} — Coordinates of reference point IGTITL = 0 Do not specify = 1 Specify group title

Card Group			Input Data and Definitions
Modifying Existing Mesh (IMOD = 1)	Build User-Defined Curves and Material Zones (IEDIT = 4)	For Each Curve Group	3.3.5.4.1 GTITL (For IGTITL= 1) MTYPE, IGPOST, OVERLAY, GCOLOR, GLTYPE, GLTHIC, GHIDE GTITL Group title MTYPE = 1 Generate lines & remove within closed loop = -1 Remove elements outside closed loop = 2 Generate lines = -2 Generate slip lines with joint elements = 3 Assign new material number within the closed loop = -3 Assign new material number within the closed loop and generate slip lines with joint elements along the loop. MTYPE = 4 and -4 are the same as MTYPE=3 and -3, respectively, except that old material zone is not removed for MTYPE = 4 and -4. To make the group null, use MTYPE = 0. IGPOST Generate Post file for element activity (1) OVERLAY Overlaid over existing group mesh (1) GCOLOR Group color index number GLTYPE Group line type index number GLTHIC Group line thickness index number GHIDE Group hide (1)

Card Group			Input Data and Definitions
Modifying Existing Mesh (IMOD = 1)	Build User-Defined Curves and Material Zones (IEDIT = 4)	For Each Curve Group	3.3.5.4.1 For MTYPE = 1 or MTYPE = 2 LTP, LMAT For MTYPE =-2 $MATNO_{JT}$, DD_{JT}, $THIC_{JT}$, LTP_I, $LMAT_I$, LTP_O, $LMAT_O$ For MTYPE = 3 MATNO, DD, LTP, LMAT For MTYPE =-3 MATNO, DD, $MATNO_{JT}$, DD_{JT}, $THIC_{JT}$, LTP_I, $LMAT_I$, LTP_O, $LMAT_O$ For MTYPE = 4 MATNO, DD, LTP, LMAT, MATold For MTYPE =-4 MATNO, DD, $MATNO_{JT}$, DD_{JT}, $THIC_{JT}$, LTP_I, $LMAT_I$, LTP_O, $LMAT_O$, MATold DD = KF (SMAP-2D) = DEN (SMAP-S2) = IDH (SMAP-T2) DD_{JT} = KF_{JT} (SMAP-2D) = DEN_{JT} (SMAP-S2) = IDH_{JT} (SMAP-T2) For MTYPE = 4 or -4 MATold takes initial value if MATNO < 0 MATold takes MATNO + 1 if MATold = 0

Card Group	Input Data and Definitions
Modifying Existing Mesh (IMOD = 1) / Build User-Defined Curves and Material Zones (IEDIT = 4) / For Each Curve Group	3.3.5.4.1 MATNO — Material No for continuum element MATold — Additional MATNO for MTYPE = 4 or -4 KF = 0 Material has fluid phase = 1 Material has no fluid phase DEN — Unit weight IDH — Heat generation ID $MATNO_{JT}$ — Material No for joint element KF_{JT} = 0 Joint has fluid phase = 1 Joint has no fluid phase DEN_{JT} — Unit weight for joint element IDH_{JT} — Heat generation ID for joint element $THIC_{JT}$ — Apparent thickness of joint element LTP = 0 Do not generate = 2 Generate beam element Heat pipe (IDFNP=LFUN), T2 = 3 Generate truss element Convection (IDFNC=LFUN, IDFNT=LFUN+1), T2 = 4 External heat flow (ID=0, IDF=LFUN), T2 = 5 Temperature boun. (ID=1, IDF=LFUN), T2 LMAT — Material No for line element LTP_I, $LMAT_I$ — Subscript i refers to inner face LTP_o, $LMAT_o$ — Subscript o refers to outer face Note: For negative value of LTP, line elements take nodes in opposite face of joint element For negative value of $THIC_{JT}$, joint elements are fully connected to the surrounding continuum elements (MTYPE = -2 or -3)

Card Group	Input Data and Definitions
Modifying Existing Mesh (IMOD = 1) / Build User-Defined Curves and Material Zones (IEDIT = 4) / For Each Curve Group	3.3.5.4.1 Required only for IGPOST= 1 NAC, NDAC (MATold) NAC, NDAC (MATNO) NAC, NDAC ($MATNO_{JT}$) NAC, NDAC (LMAT) NAC, NDAC ($LMAT_I$) NAC, NDAC ($LMAT_o$) NAC Active step number NDAC Deactive step number Required only for IGPOST= 1 CHKBOX (Mesh) CHKBOX (Principal Stress) CHKBOX (Deformed Shape) CHKBOX (Beam) CHKBOX (Truss) CHKBOX (Contour) CHKBOX (Reference Line) CHKBOX = 0 Do not plot = 1 Plot the checked item Note: IGPOST= 1 will generate main file Group.man for element activity and post file Group.pos for PLOT-2D

Card Group			Input Data and Definitions
3 Modifying Existing Mesh (IMOD = 1)	Build User-Defined Curves and Material Zones (IEDIT = 4)	For Each Curve Group	3.3.5.4.2

NPOINT, MOVE, IREF, X_{LO}, Y_{LO}

NPOINT — Number of points defining X and Y coordinates of segments. Point numbering is counter-clockwise

MOVE = 0 — Generated coordinates are movable
= 1 — Generated coordinates are not movable

IREF = 0 — Do not apply
= 1 — Local Origin (X_{LO}, Y_{LO}) is relative to Reference Point in Card 3.3.5.4

X_{LO}, Y_{LO} — Coordinates of Local Origin

NPOINT Cards:

NP_1, X_1, Y_1
NP_2, X_2, Y_2
- - -
- - -

NP — Point number
X — X-coordinate
Y — Y-coordinate

Card Group	Input Data and Definitions			
3 Modifying Existing Mesh (IMOD = 1)	Build User-Defined Curves and Material Zones (IEDIT = 4)	For Each Curve Group	3.3.5.4.3 NSEGMENT, GX, GY NSEGMENT — Number of segments If NSEGMENT is equal to NPOINT, the generated curve is closed loop. If NSEGMENT is less than NPOINT, the generated curve is open. GX, GY — Group No coordinates used in AIG	
			For Each Segment	3.3.5.4.3.1 SEGNO, LTYPE, NDIV, IEND SEGNO — Segment No in sequential order LTYPE = 1 Straight line = 2 Elliptical line NDIV — Number of divisions. Use NIDV=0 for default divisions. Use negative value to consider intermediate points as line path only. IEND = 0 Include beginning and ending points but do not register contact information =-1 Include beginning point = 1 Include ending point = 2 Same as IEND=0 but register and split =-2 Same as IEND=2 but do not split = 3 This segment is only for reference line For LTYPE = 2 X_O, Y_O, R_X, R_Y, θ_b, θ_e X_O, Y_O — Arc Origin relative to (X_{LO},Y_{LO}) R_X, R_Y — Radius in X and Y axis, respectively θ_b, θ_e — Beginning and ending angle (°) See Figure 8.2

Card Group		Input Data and Definitions
3 Modifying Existing Mesh (IMOD = 1)	3.6 Change Element Index Order (IEDIT = 6)	3.6.1 NumMATC MAT, I_1, I_2, I_3, I_4, MATC, KS, KF (SMAP-2D) MAT, I_1, I_2, I_3, I_4, MATC, THIC, DEN (SMAP-S2) MAT, I_1, I_2, I_3, I_4, MATC, IDH (SMAP-T2) NumMATC — Number of continuum materials MAT — Material number I_1, I_2, I_3, I_4 — Element corner index numbers MATC — New material property number KS, KF, THIC, DEN. IDH — Refer to Mesh File user manual
		3.6.2 NumSECB SEC, I, J, MSEC, K NumSECB — Number of beam sections SEC — Section number I, J — Element corner index numbers MSEC — New material section number K — New reference node number
		3.6.3 NumMATT MAT, I, J, MATT, K NumMATT — Number of truss materials MAT — Material number I, J — Element corner index numbers MATT — New material property number K — New reference node number Note: Index numbers are required as input. To keep the existing value, set it to -10.

Card Group	Input Data and Definitions
4 Generate Base Mesh and then Modify (IMOD = 2) See Figure 8.1	4.1 NBX, NBY, IB_LEFT, IB_RIGHT, IB_TOP, IB_BOTTOM NBX Number of blocks in X direction NBY Number of blocks in Y direction IB = 0 Free boundary = 1 Roller boundary
	4.2 X_O, Y_O, Y_{WT} X_O, Y_O Origin of X and Y coordinates Y_{WT} Y coordinate of water table (SMAP-2D) Initial temperature (SMAP-T2)
	4.3 NBX Cards W_1, ΔX_1, α_{X1} W_2, ΔX_2, α_{X2} - - - - - - W_i Horizontal length of block Δx_i Minimum horizontal element length α_{Xi} = 0.5 Element length is constant = 0.3 Element length is growing from left to right =-0.3 Element length is growing from right to left
	4.4 NBY Cards H_1, ΔY_1, α_{Y1} H_2, ΔY_2, α_{Y2} - - - - - - H_i Vertical length of block ΔY_i Minimum vertical element length α_Y = 0.5 Element length is constant = 0.3 Element length is growing from top to bottom =-0.3 Element length is growing from bottom to top
	4.5 IGMOD IGMOD = 0 Do not modify = 1 Modify generated base mesh If IGMOD = 1, go to Card 3.1

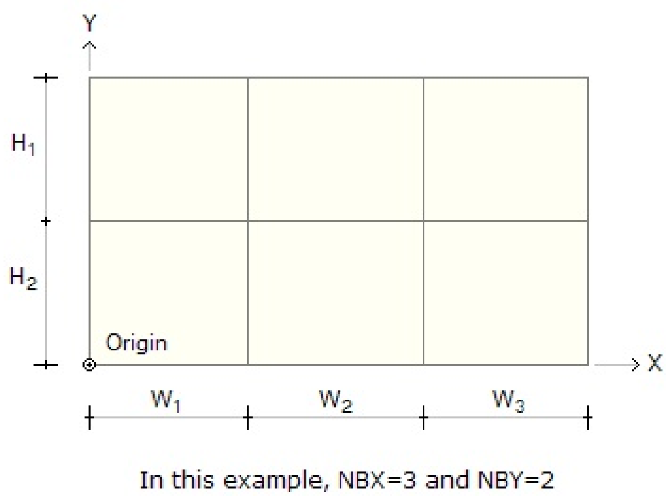


Figure 8.1 Layout of Base Mesh

Case	θ_b	θ_e
1	30 °	310 °
2	310 °	30 °
3	-50 °	30 °
4	30 °	-50 °

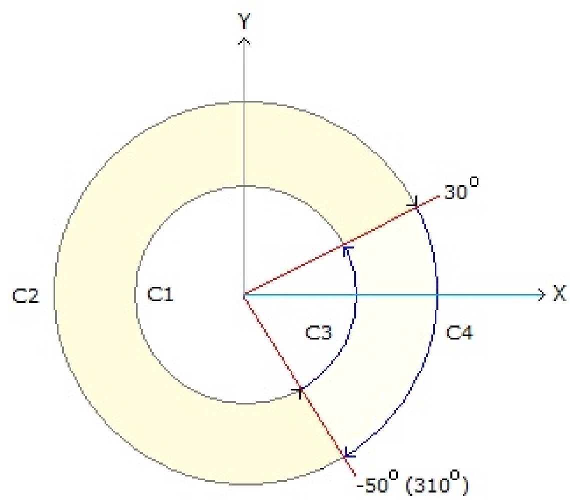


Figure 8.2 Examples of arc specification

ADDRGN-3D

User's Manual

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td>1
IMOD Type</td><td>1.1
IMOD

IMOD = 0 Add Region B to Region A
= 1 Modify existing mesh
=-1 Same as IMOD = 0 except it uses
DOF of Region B mesh along the interface</td></tr>
<tr><td>Adding Region B to Region A (IMOD = 0)</td><td>2.1
FILEA
FILEB
FILEC

FILEA Input file name containing Region A mesh
FILEB Input file name containing Region B mesh
FILEC Output file name to store Combined Region mesh

Note: When combining Region B mesh to Region A mesh, only Region B node numbers are changed. Element numbers for Region A and Region B should be continuous, otherwise element numbers are reordered automatically by program.</td></tr>
</table>

<table>
<tr><th>Card Group</th><th>Input Data and Definitions</th></tr>
<tr><td rowspan="4">3

Modifying Existing Mesh (IMOD = 1)</td><td>3.1

FILEA
FILEM

FILEA Input file name containing existing mesh
FILEM Output file name to store modified mesh</td></tr>
<tr><td>3.2

NSNEL, NSNODE, NBNEL, NTNEL

NSNEL New starting continuum element number
NSNODE New starting node number
NBNEL New starting beam element number
NTNEL New starting truss element number

Note: NBNEL & NTNEL are used for IEDIT = 0, 1, 6</td></tr>
<tr><td>3.3

IEDIT, MC_1, MC_2, MC_3, MB, MT

IEDIT = 0 Change coordinates
= 1 Change boundary codes
= 2 Cut elements
= 3 Change material numbers
= -2 Cut elements in continuum blocks
= -3 Change material numbers so as to match those in continuum blocks
= 5 Add two layers of shell elements with joint elements in-between
= 6 Change element index order

MC Continuum material number to be kept
MB Beam material number to be kept
MT Truss material number to be kept

MC, MB, and MT are applicable for IEDIT = 2, 3, -2, and -3</td></tr>
<tr><td>3.3.1

Required only for IEDIT = -2 or IEDIT = -3
FILEB
FILEB Input file name containing continuum block mesh</td></tr>
</table>

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions</th></tr>
<tr><td rowspan="2">3

Modifying Existing Mesh (IMOD = 1)</td><td rowspan="2">Changing Coordinates (IEDIT = 0)</td><td>3.3.1.1

X_o, Y_o, Z_o, X_{oNew}, Y_{oNew}, Z_{oNew}

X_o, Y_o, Z_o — Reference origin
X_{oNew}, Y_{oNew}, Z_{oNew} — New origin</td></tr>
<tr><td>3.3.1.1

X_{scale}, Y_{scale}, Z_{scale}

X_{scale}, Y_{scale}, Z_{scale} — Scale factors for X,Y, and Z coordinates.

Note: New coordinates $X_{(new)}$, $Y_{(new)}$, $Z_{(new)}$ are computed as follows:
$X_{(new)} = X_{oNew} + (X - X_0)\ X_{scale}$
$Y_{(new)} = Y_{oNew} + (Y - Y_0)\ Y_{scale}$
$Z_{(new)} = Z_{oNew} + (Z - Z_0)\ Z_{scale}$</td></tr>
</table>

Card Group		Input Data and Definitions
3	Modifying Existing Mesh (IMOD = 1) / Changing Boundary Codes (IEDIT = 1)	3.3.2.1 IRANGE Range specified by IRANGE = 0 Coordinates = 1 Node numbers = 2 Polygon = 3 Plane = 4 Line strip = 5 Material numbers
		3.3.2.2.1 Required only for IRANGE = 0 X_{start}, Y_{start}, Z_{start}, X_{end}, Y_{end}, Z_{end} X_{start}, Y_{start}, Z_{start} Coordinates for lower left boundary X_{end}, Y_{end}, Z_{end} Coordinates for upper right boundary
		3.3.2.2.2 Required only for IRANGE = 1, 2, 3, 4, 5 NODE NOD_1, NOD_2, ..., NOD_{NODE} NODE Number of nodes/materials to be specified NOD_i Node/Material number (See Note 1) Polygon (IRANGE = 2) is defined counterclockwise Plane (IRANGE = 3) is defined by 3 nodes For IRANGE = 5, Nodes refer to Material numbers. Note 1: NOD_1, $-NOD_2$ generates from NOD_1 to NOD_2 Note 2: NEL_1, $-NEL_2$ generates from NEL_1 to NEL_2

Card Group		Input Data and Definitions
3 Modifying Existing Mesh (IMOD = 1)	Changing Boundary Codes (IEDIT = 1)	3.3.2.3 INSIDE (Not applicable for IRANGE= 5) INSIDE = 0 Apply inside of range = 1 Apply outside of range
		3.3.2.4 ISX, ISY, ISZ, IFX, IFY, IFZ, IRX, IRY, IRZ (SMAP-3D) ID, IDF (SMAP-T3) ISX, ISY, ISZ X, Y, Z DOF for skeleton motion IFX, IFY, IFZ X, Y, Z DOF for relative fluid motion IRX, IRY, IRZ X, Y, Z DOF for rotation ISX, ISY, ISZ, IFX, IFY, IFZ, IRX, IRY, IRZ = 0 Free to move in specified direction = 1 Fixed in specified direction ID = 0 External heat flow is specified = 1 Temperature is specified IDF Identification number for time dependent function If IDF = 0, external heat flow is zero at all times

Card Group		Input Data and Definitions
3 Modifying Existing Mesh (IMOD = 1)	Cutting Elements (IEDIT = 2)	3.3.3.1 IRANGE Range specified by IRANGE = 0 Coordinates = 1 Eement numbers
		3.3.3.2.1 Required only for IRANGE = 0 X_{start}, Y_{start}, Z_{start}, X_{end}, Y_{end}, Z_{end} X_{start}, Y_{start}, Z_{start} Coordinates for lower left boundary X_{end}, Y_{end}, Z_{end} Coordinates for upper right boundary
		3.3.3.2.2 Required only for IRANGE = 1 NOEL NEL_1, NEL_2, ..., NEL_{NOEL} NOEL Number of elements to be specified NEL_i Element number (See Note 2 in page 8-24)
		3.3.3.3 INSIDE INSIDE = 0 Apply inside of range = 1 Apply outside of range

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions</th></tr>
<tr><td rowspan="5">3

Modifying Existing Mesh (IMOD = 1)</td><td rowspan="5">3.3.4

Change Material Numbers (IEDIT = 3)</td><td>3.3.4
IRANGE

IRANGE = 0 Range specified by coordinates
= 1 Range specified by element numbers</td></tr>
<tr><td>3.3.4.1
Required only for IRANGE = 0
X_{start}, Y_{start}, Z_{start}, X_{end}, Y_{end}, Z_{end}

X_{start}, Y_{start}, Z_{start} Coordinates for lower left boundary
X_{end}, Y_{end}, Z_{end} Coordinates for upper right boundary</td></tr>
<tr><td>3.3.4.2
Required only for IRANGE = 1
NOEL
NEL_1, NEL_2, ..., NEL_{NODE}

NOEL Number of elements to be specified
NEL_i Element number (See Note 2 in page 8-24)</td></tr>
<tr><td>3.3.4.3
INSIDE

INSIDE = 0 Apply inside of range
= 1 Apply outside of range</td></tr>
<tr><td>3.3.4.4
MATC, MATB, MATT

New material number for
MATC Continuum element
MATB Beam element
MATT Truss element

Note: When new material number is zero, keep the old material number.</td></tr>
</table>

<table>
<tr><th>Card Group</th><th colspan="3">Input Data and Definitions</th></tr>
<tr><td rowspan="3">3

Modifying Existing Mesh (IMOD = 1)</td><td rowspan="3">3.5

Add Two Layers of Shell with Joint in-between (IEDIT = 5)</td><td colspan="2">3.5.1

$MATS_1$, MATJ, $MATS_2$, THICJ

$MATS_1$ — 1ST layer shell material number
MATJ — Joint material number
$MATS_2$ — 2nd layer shell material number
THICJ — Apparent thickness of joint element

Note: If the value of THICJ is negative, joint elements are generated inward</td></tr>
<tr><td colspan="2">3.5.2

NSECTION, NUMNODE

NSECTION — Number of sections (Max=200)
NUMNODE — Number of nodes per section (Max=200)</td></tr>
<tr><td>For Each Section</td><td>3.5.3

NOD_1, NOD_2, . . . , $NOD_{NUMNODE}$

NOD_i — Node number

Note: List node numbers in counter clockwise
If NOD_1=$NOD_{NUMNODE}$, the loop is closed</td></tr>
</table>

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions</th></tr>
<tr><td rowspan="3">3

Modifying Existing Mesh (IMOD = 1)</td><td rowspan="3">3.6

Change Element Index Order (IEDIT = 6)</td><td>3.6.1

NumMATC
MAT, I_1, I_2, I_3, I_4, I_5, I_6, I_7, I_8, MATC, KS, KF (SMAP-3D)
MAT, I_1, I_2, I_3, I_4, I_5, I_6, I_7, I_8, MATC, IDH (SMAP-T3)

NumMATC — Number of continuum materials
MAT — Material number
$I_1,I_2,I_3,I_4,I_5,I_6,I_7,I_8$ — Element corner index numbers
MATC — New material property number
KS, KF, IDH — Refer to Mesh File user manual</td></tr>
<tr><td>3.6.2

NumSECB
SEC, I, J, K, MSEC

NumSECB — Number of beam sections
SEC — Section number
I, J — Element corner index numbers
K — New reference node number
MSEC — New material section number</td></tr>
<tr><td>3.6.3

NumMATT
MAT, I, J, MATT, K

NumMATT — Number of truss materials
MAT — Material number
I, J — Element corner index numbers
MATT — New material property number
K — New reference node number

Note: Index numbers are required as input.
To keep the existing value, set it to -10.</td></tr>
</table>

Supplement Program

9.1 Introduction

Supplement programs contain supporting programs which are useful to prepare input data for pre-and main-processing programs and can be accessed through Run → Mesh Generater → Supplement menu.
Currently, there are five programs available:
EDIT, XY, CARDS, SHRINK FILE and CUDSS.

EDIT is used to run text editor.

XY computes coordinates of mid points, cross points, or normal points.

CARDS generates Element Activity data in Card Group 8 in Section 4.4 Main File.

SHRINK FILE removes extra blank spaces before carriage return.
This will reduce the size of the file.

CUDSS simulates cyclic undrained direct simple shear test.

9.2 EDIT

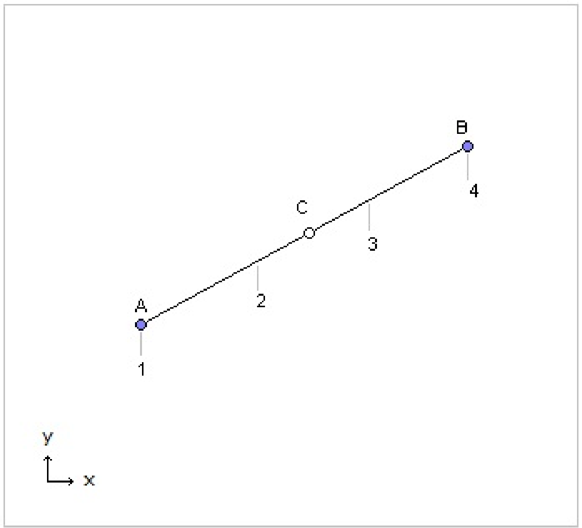
EDIT uses Windows text editor Wordpad to creat, modify, or list file.

9.3 XY

Program XY can be used to compute midpoints, intersection points and normal points of straight line and circular arc. The program is useful to construct the block diagrams of the problem geometry.

To run program XY, simply select XY from SUPPLEMENT Menu and follow instructions shown on the screen.

NF = 1 Compute Midpoint on Straight Line

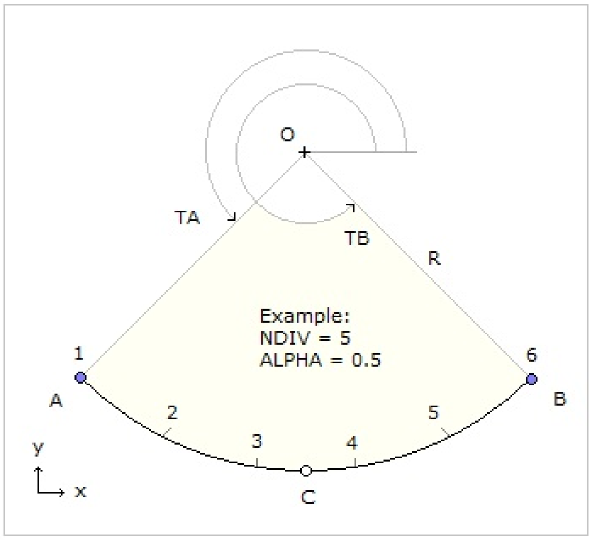


Example: NDIV = 3 and ALPHA = 0.5

INPUT:
XA, YA, XB, YB
NDIV, ALPHA

XA, YA	=	X and Y coordinates of A
XB, YB	=	X and Y coordinates of B
NDIV	=	Number of division
ALPHA	=	Geometric ratio

NF = 2 Compute Midpoint on Circular Arc

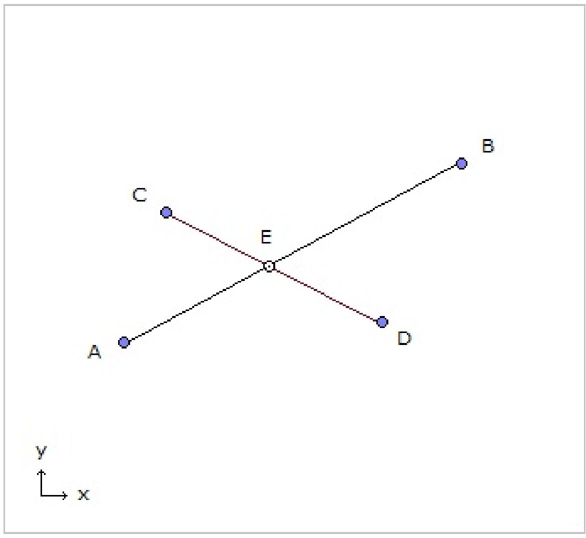


INPUT:
R, X_O, Y_O, TA, TB
NDIV, ALPHA

R	=	Radius
X_O, Y_O	=	X and Y coordinates of origin O
TA,TB	=	Angles (degrees) of A and B
NDIV	=	Number of division
ALPHA	=	Geometric ratio

If ALPHA = 0.5, midpoint C is located in half way between A and B
If ALPHA < 0.5, midpoint is close to A
If ALPHA > 0.5, midpoint is close to B

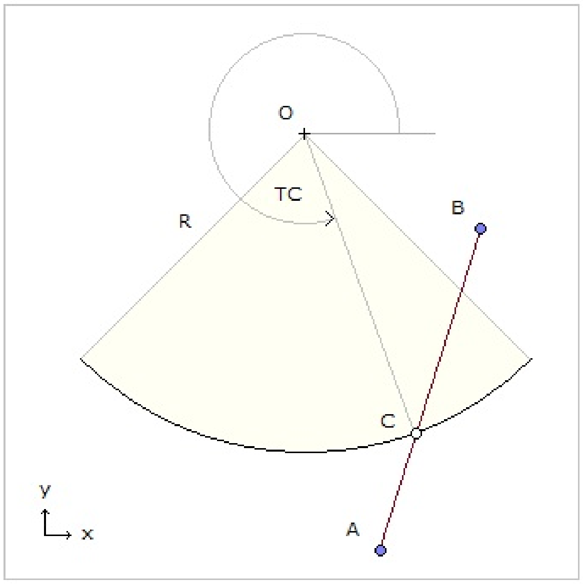
NF = 3 Compute Intersection Point of Two Straight Lines



INPUT:
XA, YA, XB, YB
XC, YC, XD, YD

XA, YA = X and Y coordinates of A
XB, YB = X and Y coordinates of B
XC, YC = X and Y coordinates of C
XD, YD = X and Y coordinates of D

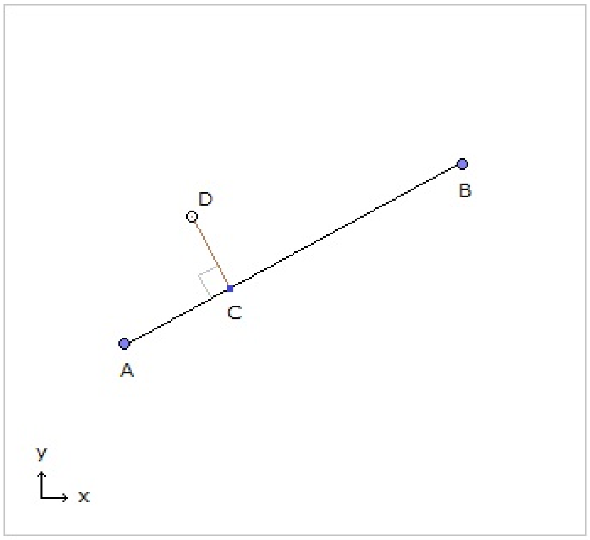
NF = 4 Compute Intersection point of Arc & Straight Line



INPUT:
R, X_o, Y_o
XA, YA, XB, YB

R	=	Radius
X_o, Y_o	=	X and Y coordinates of origin O
XA, YA	=	X and Y coordinates of point A
XB, YB	=	X and Y coordinates of point B

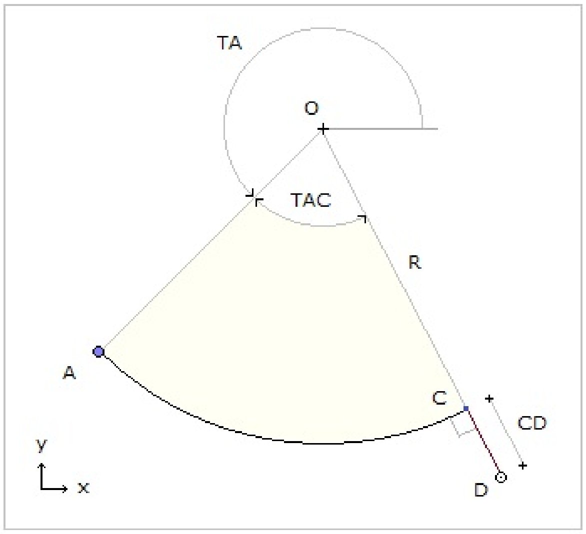
NF = 5 Compute Points Normal to Straight Line



INPUT:
XA, YA, XB, YB
AC, CD

XA, YA	=	X and Y coordinates of A
XB, YB	=	X and Y coordinates of B
AC	=	Distance between A and C
CD	=	Distance between C and D

NF = 6 Compute Points Normal to Circular Arc



INPUT:
R, X_o, Y_o, TA
TAC, CD

R	=	Radius
X_o, Y_o	=	X and Y coordinates of origin O
TA	=	Angle (degree) of A
TAC	=	Angle (degree) between A and C
CD	=	Distance between C and D

9.4 CARDS

Program CARDS is included to aid for users to prepare SMAP-3D input cards. Currently, there is only one routine which generates Element Activity in Card Group 8 in Section 4.4 Main File.

You are asked to type in following input data to generate element activity and deactivity;

NEL (start) NEL(end) NAC NDAC

where

NEL (start)	Starting element number
NEL (end)	Ending element number
NAC	Load step at which elements from NEL(start) to NEL(end) are activated.
NDAC	Load step at which elements from NEL(start) to NEL(end) are deactivated.

Generated element activity data will be written in the output file you specified.

9.5 SHRINK FILE

SHRINK FILE is included to remove extra blank spaces before carriage return. This will reduce the size of the file where blank spaces are existing before the carriage return.

9.6 CUDSS

CUDSS is to simulate cyclic undrained direct simple shear test. PM4Sand material model is used to represent skeleton behavior, developed by Boulanger, R. W. And ziotopoulou, k. (Version 3.1).

Input File CUDSS.inp for PM4Sand Material Model

<table>
<tr><td>Card Group</td><td>Cyclic Undrained Direct Simple Shear Simulation</td></tr>
<tr><td rowspan="4">PM4Sand Material Model</td><td>1.0
Title

Title   Title</td></tr>
<tr><td>2.0
σ_{vo}' K_o α_s

σ_{vo}' Initial effective vertical stress
K_o Coefficient of earth pressure at rest
α_s Initial static shear stress ratio : $\alpha_s = \tau_s \ / \ \sigma_{vo}'$
where τ_s is initial static shear stress</td></tr>
<tr><td>3.0
CSR γ_{max}

CSR   Cyclic stress ratio : $CSR = \tau_p \ / \ \sigma_{vo}'$
where τ_p is cyclic peak shear stress
γ_{max} Maximum cutoff shear strain</td></tr>
<tr><td>4.0
NCYCLE $\Delta\gamma$

NCYCLE   Maximum number of cycles
$\Delta\gamma$ Shear strain increment (Default 1.0e-05)</td></tr>
</table>

Card Group	Cyclic Undrained Direct Simple Shear Simulation
5 PM4Sand Material Model	5.3.2.4.21

For MODELNO = 21 [PM4Sand Model]

D_R G_o h_{po} p_a N_s

Secondary Parameters (Skip these cards for N_s = 1)

h_o e_{max} e_{min} n^b n^d A_{do}
z_{max} c_z C_e φ_{cv} v_o C_{GD}
C_{DR} c_{kaf} Q R m $F_{sed.min}$ p_{sed}

D_R	Apparent relative density (Fraction)
G_o	Shear modulus coefficient
h_{po}	Contraction rate parameter
p_a	Atmospheric pressure (10.33 for stress unit t/m^2)
N_s	Secondary parameter specification: 0 = Yes, 1 = No
h_o	Control parameter for ratio of plastic to elastic modulus
e_{max}	Maximum void ratio (Default 0.8)
e_{min}	Minimum void ratio (Default 0.5)
n^b	Control parameter for dilatancy & peak friction angle
n^d	Control parameter for transition from contr. to dilation
A_{do}	Bolton's dilatancy parameter
z_{max}	Maximum allowable fabric dilatancy tensor z
c_z	Control parameter when fabric effects get important
C_e	Control parameter for adjusting strain accumulation rate
φ_{cv}	Critical state effective friction angle (Default 33°)
v_o	Poisson's ratio (Default 0.3)
C_{GD}	Factor for shear modulus degradation (Default 2.0)
C_{DR}	Control parameter for rotated dilatancy surface
c_{kaf}	Control parameter for effects of sustained shear stress
Q, R	Parameters for Bolton's empirical critical state line
m	Parameter defining size of yield stress (Default 0.01)
$F_{sed.min}$	Parameter for post-shaking elastic modulus reduction
p_{sed}	Mean effective stress for post-shaking reconsolidation

Set -1 for default values of secondary model parameters.

For description, refer to Boulanger, R. W. And ziotopoulou, k. PM4Sand (Version 3.1): A Sand Plasticity Model for Earthquake Engineering Applications, Report No UCD/CGM-17/01, Dept. of Civil & Env. Eng., U. of Cal., Davis, CA, 109 pp.

File Conversion

10.1 Introduction

PRESMAP programs described in Section 7 generate Mesh Files which contain the geometric information of structures to be analyzed. The format of SMAP-3D Mesh File is presented in detail in Section 4.3.

Three-dimensional Mesh Files can also be created by IGES (Initial Graphics Exchange Specification) or FEMAP (Version 4.1 - 4.5, neutral format) program which is developed by EDS.

In this section, we will briefly discuss Mesh File conversion under Mesh Generater → File Conversion menu:

10.2 Conversion to SMAP-3D Mesh File

Following Mesh Files can be converted to SMAP-3D Mesh File format:

- Mesh Files generated for two-dimensional SMAP programs (SMAP-S2, SMAP-2D, and SMAP-T2)
- Mesh Files generated for three-dimensional SMAP program (SMAP-T3)
- IGES (Initial Graphics Exchange Specification)
- FEMAP (Version 4.1 - 4.5, neutral format)

Figure 10.1 shows File Conversion dialog box with Input Mesh File options.

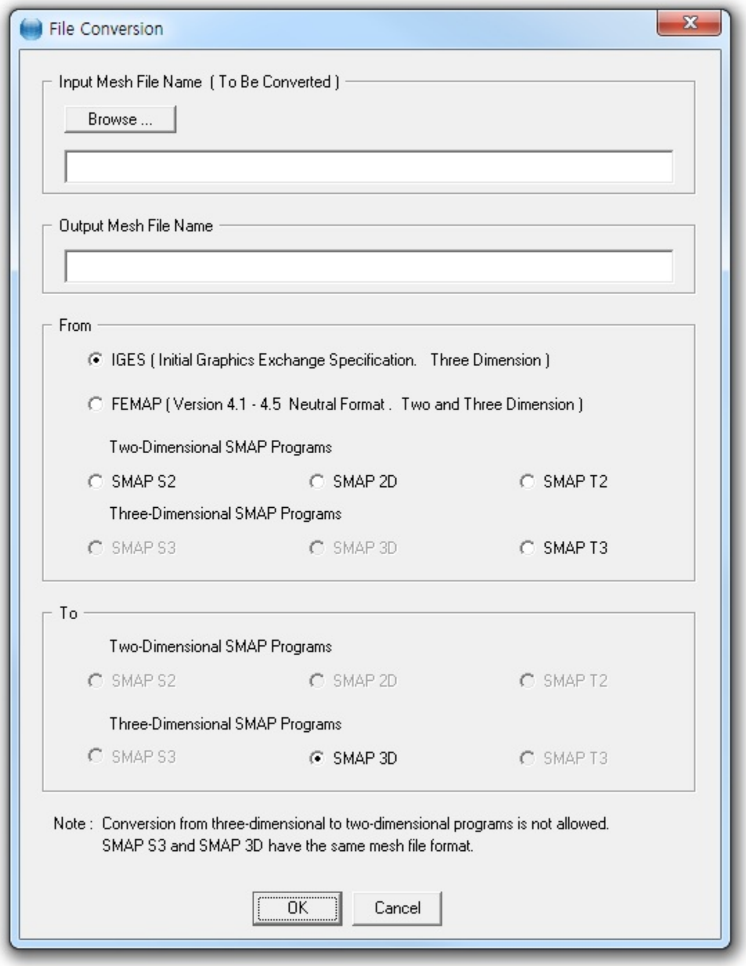


Figure 10.1 File Conversion dialog box

LOAD
User's Manual

11.1 Introduction

LOAD is the pre-processing program which generates nodal values of external forces, specified velocities, initial velocities, accelerations and transmitting boundaries.

Before you prepare LOAD input data in this section, you should have a Mesh File generated from PRESMAP/ADDRGN programs. That is, LOAD input is referred to the geometric surfaces given in the Mesh File.

Generated LOAD output file contains load data which is compatible to the format of Card Group 9 described in Section 4.4 Main File.

LOAD-2D deals with two dimensional meshes and LOAD-3D deals with three dimensional meshes.

LOAD-2D

LDTYPE = 1 [Pressure: SMAP-2D/S2]

<table>
<tr><th>Card Group</th><th colspan="3">Input Data and Definitions (Pressure)</th></tr>
<tr><td rowspan="2">1
Title & Element</td><td colspan="3">1.1
TITLE
TITLE Any title (Max = 60 characters)</td></tr>
<tr><td colspan="3">1.2
NCTYPE
NCTYPE = 0 Axisymmetric element
Y-axis is axis of symmetry
= 1 Plane strain element (Thickness=1.0)
= 2 Plane stress element (Thickness=1.0)
= 3 Spherically symmetric element (SMAP-2D)</td></tr>
<tr><td rowspan="4">2
Loading Surface</td><td colspan="3">2.1
NUMLS
NUMLS Number of loading surfaces where external tractions are specified (Max = 20)</td></tr>
<tr><td rowspan="3">2.2
For Each Loading Surface</td><td colspan="2">2.2.1
LSNO, LSTYPE
LSNO Loading surface number
LSTYPE = 0 All specified nodes
= 1 Line strip = 2 Points
= 3 Node group = 4 Element group</td></tr>
<tr><td rowspan="2">LSTYPE = 0, 1, 2</td><td>2.2.2
NUMNODE
NUMNODE Number of nodes on this loading surface (Max = 9990)</td></tr>
<tr><td>2.2.3
NOD_1, NOD_2, ..., $NOD_{NUMNODE}$
NOD_i Specified node

Line strip (LSTYPE=1) is defined counterclockwise. For LSTYPE=1 and $NOD_{NUMNODE} < 0$, absolute value of $NOD_{NUMNODE}$ is the reference node defining normal to the Line strip.</td></tr>
</table>

Card Group	Input Data and Definitions (Pressure)

2 Loading Surface

2.2 For Each Loading Surface

LSTYPE = 3 (Node Group)

2.2.4

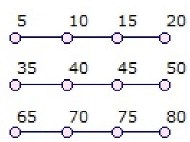
NUMNODG

NUMNODG	Number of node groups on this loading surface (Max = 100)

2.2.5

NSR, JCR, NJR, ICR, NIR For Each Group

NSR	Starting node number of the first row
JCR	Node number increment in a row
NJR	Number of nodes in a row
ICR	Node number increment for next row
NIR	Total number of rows



Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

LSTYPE = 4 (Element Group)

2.2.6

NUMNELG

NUMNELG	Number of element groups on this loading surface (Max = 100)

2.2.7

NSR, JCR, NJR, ICR, NIR, NS For Each Group

NSR	Starting element number of the first row
JCR	Element number increment in a row
NJR	Number of elements in a row
ICR	Element number increment for next row
NIR	Total number of rows
NS	Element surface number (See Mesh File Card 3.2)

5	10	15	20
35	40	45	50
65	70	75	80

Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

Card Group	Input Data and Definitions (Pressure)	
3 Pressure Function	3.1	NUMLP NUMLP Number of pressure functions (Max = 20)
	3.2 For Each Pressure Function	3.2.1 LPNO, LPTYPE LPNO Pressure function number LPTYPE = 0 Use effective surface = 1 Use actual surface Note: Effective surface is normal to force direction (Ex. Wind load)
		3.2.2 a_{xo}, a_{xx}, a_{xy} a_{xi} Coefficients defining surface traction in the x-direction. $P_x = a_{xo} + a_{xx}x + a_{xy}y$
		3.2.3 a_{yo}, a_{yx}, a_{yy} a_{yi} Coefficients defining surface traction in the y-direction. $P_y = a_{yo} + a_{yx}x + a_{yy}y$
		3.2.4 a_{no}, a_{nx}, a_{ny} a_{ni} Coefficients defining surface traction normal to surface. Acting on actual surface $P_n = a_{no} + a_{nx}x + a_{ny}y$

Card Group		Input Data and Definitions (Pressure)
4 Pressure History	4.1	NUMLH NUMLH Number of pressure histories (Max = 20)
	4.2 For Each Pressure History	4.2.1 LHNO LHNO Pressure history number
		4.2.2 NUMTP NUMTP Number of time points (Max = 1000)
		4.2.3 T_1, T_2, …, T_{NUMTP} T_i Specified time
		4.2.4 C_1, C_2, …, C_{NUMTP} C_i Pressure intensity at time T_i

Card Group	Input Data and Definitions (Pressure)
5 Pressure Specification	5.1 LSNO, LPNO, LHNO LSNO Loading surface number LPNO Pressure function number LHNO Pressure history number Repeat Card 5.1 until the last card (LSNO=0) is specified

LOAD-2D

LDTYPE = 2 [Velocity: SMAP-2D]

<table>
<tr><th>Card Group</th><th colspan="3">Input Data and Definitions (Velocity)</th></tr>
<tr><td rowspan="2">1
Title & Element</td><td colspan="3">1.1
TITLE
TITLE Any title (Max = 60 characters)</td></tr>
<tr><td colspan="3">1.2
NCTYPE
NCTYPE = 0 Axisymmetric element
Y-axis is axis of symmetry
= 1 Plane strain element (Thickness=1.0)
= 2 Plane stress element (Thickness=1.0)
= 3 Spherically symmetric element (SMAP-2D)</td></tr>
<tr><td rowspan="4">2
Loading Surface</td><td colspan="3">2.1
NUMLS
NUMLS Number of loading surfaces where velocities are specified (Max = 20)</td></tr>
<tr><td rowspan="3">2.2
For Each Loading Surface</td><td colspan="2">2.2.1
LSNO, LSTYPE
LSNO Loading surface number
LSTYPE = 0 All specified nodes
= 1 Line strip = 2 Points
= 3 Node group = 4 Element group</td></tr>
<tr><td rowspan="2">LSTYPE = 0, 1, 2</td><td>2.2.2
NUMNODE
NUMNODE Number of nodes on this loading surface (Max = 9990)</td></tr>
<tr><td>2.2.3
NOD$_1$, NOD$_2$, ..., NOD$_{NUMNODE}$
NOD$_i$ Specified node

Line strip (LSTYPE=1) is defined counterclockwise. For LSTYPE=1 and NOD$_{NUMNODE}$ < 0, absolute value of NOD$_{NUMNODE}$ is the reference node defining normal to the Line strip.</td></tr>
</table>

Card Group	Input Data and Definitions (Velocity)

2 Loading Surface

2.2 For Each Loading Surface

LSTYPE = 3 (Node Group)

2.2.4

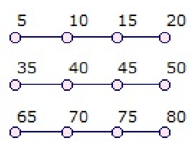
NUMNODG

NUMNODG	Number of node groups on this loading surface (Max = 100)

2.2.5

NSR, JCR, NJR, ICR, NIR   For Each Group

NSR	Starting node number of the first row
JCR	Node number increment in a row
NJR	Number of nodes in a row
ICR	Node number increment for next row
NIR	Total number of rows



Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

LSTYPE = 4 (Element Group)

2.2.6

NUMNELG

NUMNELG	Number of element groups on this loading surface (Max = 100)

2.2.7

NSR, JCR, NJR, ICR, NIR, NS   For Each Group

NSR	Starting element number of the first row
JCR	Element number increment in a row
NJR	Number of elements in a row
ICR	Element number increment for next row
NIR	Total number of rows
NS	Element surface number (See Mesh File Card 3.2)

5	10	15	20
35	40	45	50
65	70	75	80

Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Velocity)</th></tr>
<tr><td rowspan="5">3
Velocity Function</td><td colspan="2">3.1
NUMLV
NUMLV Number of velocity functions (Max = 20)</td></tr>
<tr><td rowspan="4">3.2
For Each Velocity Function</td><td>3.2.1
LVNO
LVNO Velocity function number</td></tr>
<tr><td>3.2.2
a_{xo}, a_{xx}, a_{xy}
a_{xi} Coefficients defining velocity in x-direction
$V_x = a_{xo} + a_{xx}\ x + a_{xy}\ y$</td></tr>
<tr><td>3.2.3
a_{yo}, a_{yx}, a_{yy}
a_{yi} Coefficients defining velocity in y-direction
$V_y = a_{yo} + a_{yx}\ x + a_{yy}\ y$</td></tr>
<tr><td>3.2.4
a_{no}, a_{nx}, a_{ny}
a_{ni} Coefficients defining velocity normal to surface
$V_n = a_{no} + a_{nx}\ x + a_{ny}\ y$</td></tr>
</table>

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Velocity)</th></tr>
<tr><td rowspan="5">4

Velocity History</td><td colspan="2">4.1

NUMLH

NUMLH Number of velocity histories (Max = 20)</td></tr>
<tr><td rowspan="4">4.2

For Each Velocity History</td><td>4.2.1

LHNO

LHNO Velocity history number</td></tr>
<tr><td>4.2.2

NUMTP

NUMTP Number of time points (Max = 1000)</td></tr>
<tr><td>4.2.3

T_1, T_2, …, T_{NUMTP}

T_i Specified time</td></tr>
<tr><td>4.2.4

C_1, C_2, …, C_{NUMTP}

C_i Velocity intensity at time T_i</td></tr>
</table>

Card Group	Input Data and Definitions (Velocity)
5 Velocity Specification	5.1 LSNO, LVNO, LHNO LSNO Loading surface number LVNO Velocity function number LHNO Velocity history number Repeat Card 5.1 until the last card (LSNO=0) is specified

LOAD-2D

LDTYPE = 3 [Initial Velocity: SMAP-2D]

Card Group	Input Data and Definitions (Initial Velocity)
1 Title & Element	1.1 TITLE TITLE Any title (Max = 60 characters)
	1.2 NCTYPE NCTYPE = 0 Axisymmetric element Y-axis is axis of symmetry = 1 Plane strain element (Thickness=1.0) = 2 Plane stress element (Thickness=1.0) = 3 Spherically symmetric element (SMAP-2D)
2 Loading Surface	2.1 NUMLS NUMLS Number of loading surfaces where initial velocities are specified (Max = 20)
	2.2 For Each Loading Surface 2.2.1 LSNO, LSTYPE LSNO Loading surface number LSTYPE = 0 All specified nodes = 1 Line strip = 2 Points = 3 Node group = 4 Element group
	LSTYPE = 0, 1, 2 2.2.2 NUMNODE NUMNODE Number of nodes on this loading surface (Max = 9990)
	2.2.3 NOD_1, NOD_2, ..., $NOD_{NUMNODE}$ NOD_i Specified node Line strip (LSTYPE=1) is defined counterclockwise. For LSTYPE=1 and $NOD_{NUMNODE} < 0$, absolute value of $NOD_{NUMNODE}$ is the reference node defining normal to the Line strip.

Card Group	Input Data and Definitions (Initial Velocity)

2 — Loading Surface

2.2 — For Each Loading Surface

LSTYPE = 3 (Node Group)

2.2.4

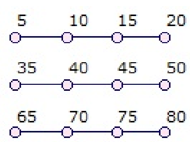
NUMNODG

NUMNODG — Number of node groups on this loading surface (Max = 100)

2.2.5

NSR, JCR, NJR, ICR, NIR   For Each Group

NSR — Starting node number of the first row
JCR — Node number increment in a row
NJR — Number of nodes in a row
ICR — Node number increment for next row
NIR — Total number of rows



Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

LSTYPE = 4 (Element Group)

2.2.6

NUMNELG

NUMNELG — Number of element groups on this loading surface (Max = 100)

2.2.7

NSR, JCR, NJR, ICR, NIR, NS   For Each Group

NSR — Starting element number of the first row
JCR — Element number increment in a row
NJR — Number of elements in a row
ICR — Element number increment for next row
NIR — Total number of rows
NS — Element surface number (See Mesh File Card 3.2)

5	10	15	20
35	40	45	50
65	70	75	80

Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Initial Velocity)</th></tr>
<tr><td rowspan="5">3
Initial Velocity Function</td><td colspan="2">3.1
NUMLIV
NUMLIV Number of initial velocity functions (Max = 20)</td></tr>
<tr><td rowspan="4">3.2
For Each Initial Velocity Function</td><td>3.2.1
LIVNO
LIVNO Initial velocity function number</td></tr>
<tr><td>3.2.2
a_{xo}, a_{xx}, a_{xy}
a_{xi} Coefficients defining initial velocity in the x-direction
$V_{ix} = a_{xo} + a_{xx}x + a_{xy}y$</td></tr>
<tr><td>3.2.3
a_{yo}, a_{yx}, a_{yy}
a_{yi} Coefficients defining initial velocity in the y-direction
$V_{iy} = a_{yo} + a_{yx}x + a_{yy}y$</td></tr>
<tr><td>3.2.4
a_{no}, a_{nx}, a_{ny}
a_{ni} Coefficients defining initial velocity normal to the surface
$V_{in} = a_{no} + a_{nx}x + a_{ny}y$</td></tr>
</table>

Card Group	Input Data and Definitions (Initial Velocity)
4 Initial Velocity Specification	4.1 LSNO, LIVNO LSNO Loading surface number LIVNO Initial velocity function Repeat Card 4.1 until the last card (LSNO=0) is specified

LOAD-2D

LDTYPE = 4 [Acceleration: SMAP-2D]

<table>
<tr><th>Card Group</th><th colspan="3">Input Data and Definitions (Acceleration)</th></tr>
<tr><td rowspan="2">1
Title & Element</td><td colspan="3">1.1
TITLE
TITLE — Any title (Max = 60 characters)</td></tr>
<tr><td colspan="3">1.2
NCTYPE
NCTYPE = 0 Axisymmetric element
Y-axis is axis of symmetry
= 1 Plane strain element (Thickness=1.0)
= 2 Plane stress element (Thickness=1.0)
= 3 Spherically symmetric element (SMAP-2D)</td></tr>
<tr><td rowspan="4">2
Loading Surface</td><td colspan="3">2.1
NUMLS
NUMLS — Number of loading surfaces where accelerations are specified (Max = 20)</td></tr>
<tr><td rowspan="3">2.2
For Each Loading Surface</td><td colspan="2">2.2.1
LSNO, LSTYPE
LSNO — Loading surface number
LSTYPE = 0 All specified nodes
= 1 Line strip = 2 Points
= 3 Node group = 4 Element group</td></tr>
<tr><td rowspan="2">LSTYPE = 0, 1, 2</td><td>2.2.2
NUMNODE
NUMNODE — Number of nodes on this loading surface (Max = 9990)</td></tr>
<tr><td>2.2.3
NOD_1, NOD_2, ..., $NOD_{NUMNODE}$
NOD_i — Specified node

Line strip (LSTYPE=1) is defined counterclockwise. For LSTYPE=1 and $NOD_{NUMNODE} < 0$, absolute value of $NOD_{NUMNODE}$ is the reference node defining normal to the Line strip.</td></tr>
</table>

Card Group			Input Data and Definitions (Acceleration)
2 Loading Surface	2.2 For Each Loading Surface	LSTYPE = 3 (Node Group)	2.2.4

2 — Loading Surface

2.2 — For Each Loading Surface

LSTYPE = 3 (Node Group)

2.2.4

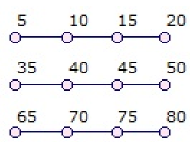
NUMNODG

NUMNODG	Number of node groups on this loading surface (Max = 100)

2.2.5

NSR, JCR, NJR, ICR, NIR   For Each Group

NSR	Starting node number of the first row
JCR	Node number increment in a row
NJR	Number of nodes in a row
ICR	Node number increment for next row
NIR	Total number of rows



Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

LSTYPE = 4 (Element Group)

2.2.6

NUMNELG

NUMNELG	Number of element groups on this loading surface (Max = 100)

2.2.7

NSR, JCR, NJR, ICR, NIR, NS   For Each Group

NSR	Starting element number of the first row
JCR	Element number increment in a row
NJR	Number of elements in a row
ICR	Element number increment for next row
NIR	Total number of rows
NS	Element surface number (See Mesh File Card 3.2)

5	10	15	20
35	40	45	50
65	70	75	80

Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Acceleration)</th></tr>
<tr><td rowspan="5">3
Acceleration Function</td><td colspan="2">3.1
NUMLA

NUMLA Number of acceleration functions (Max = 20)</td></tr>
<tr><td rowspan="4">3.2
For Each Acceleration</td><td>3.2.1
LANO
LANO Acceleration function number</td></tr>
<tr><td>3.2.2
a_{xo}, a_{xx}, a_{xy}

a_{xi} Coefficients defining acceleration in the x-direction
$A_x = a_{xo} + a_{xx}x + a_{xy}y$</td></tr>
<tr><td>3.2.3
a_{yo}, a_{yx}, a_{yy}

a_{yi} Coefficients defining acceleration in the y-direction
$A_y = a_{yo} + a_{yx}x + a_{yy}y$</td></tr>
<tr><td>3.2.4
a_{no}, a_{nx}, a_{ny}

a_{ni} Coefficients defining acceleration normal to the surface
$A_n = a_{no} + a_{nx}x + a_{ny}y$</td></tr>
</table>

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Acceleration)</th></tr>
<tr><td rowspan="5">4

Acceleration History</td><td colspan="2">4.1

NUMLH

NUMLH Number of acceleration histories (Max = 20)</td></tr>
<tr><td rowspan="4">4.1

For Each Acceleration History</td><td>4.2.1

LHNO

LHNO Acceleration history number</td></tr>
<tr><td>4.2.2

NUMTP

NUMTP Number of time points (Max = 1000)</td></tr>
<tr><td>4.2.3

T_1, T_2, …, T_{NUMTP}

T_i Specified time</td></tr>
<tr><td>4.2.4

C_1, C_2, …, C_{NUMTP}

C_i Acceleration intensity at time T_i</td></tr>
</table>

Card Group	Input Data and Definitions (Acceleration)
5 Acceleration Specification	5.1 LSNO, LANO, LHNO LSNO Loading surface number LANO Acceleration function number LHNO Acceleration history number Repeat Card 5.1 until the last card (LSNO=0) is specified

LOAD-2D

LDTYPE = 5 [Transmitting Boundary: SMAP-2D]

<table>
<tr><th>Card Group</th><th colspan="3">Input Data and Definitions (Transmitting Boundary)</th></tr>
<tr><td rowspan="2">1
Title & Element</td><td colspan="3">1.1
TITLE
TITLE Any title (Max = 60 characters)</td></tr>
<tr><td colspan="3">1.2
NCTYPE
NCTYPE = 0 Axisymmetric element
Y-axis is axis of symmetry
= 1 Plane strain element (Thickness=1.0)
= 2 Plane stress element (Thickness=1.0)
= 3 Spherically symmetric element (SMAP-2D)</td></tr>
<tr><td rowspan="4">2
Loading Surface</td><td colspan="3">2.1
NUMLS
NUMLS Number of loading surfaces where transmitting boundaries are specified (Max = 20)</td></tr>
<tr><td rowspan="3">2.2
For Each Loading Surface</td><td colspan="2">2.2.1
LSNO, LSTYPE
LSNO Loading surface number
LSTYPE = 0 All specified nodes
= 1 Line strip = 2 Points
= 3 Node group = 4 Element group</td></tr>
<tr><td rowspan="2">LSTYPE = 0, 1, 2</td><td>2.2.2
NUMNODE
NUMNODE Number of nodes on this loading surface (Max = 9990)</td></tr>
<tr><td>2.2.3
NOD_1, NOD_2, ..., $NOD_{NUMNODE}$
NOD_i Specified node

Line strip (LSTYPE=1) is defined counterclockwise. For LSTYPE=1 and $NOD_{NUMNODE} < 0$, absolute value of $NOD_{NUMNODE}$ is the reference node defining normal to the Line strip.</td></tr>
</table>

Card Group	Input Data and Definitions (Transmitting Boundary)

2 — Loading Surface

2.2 — For Each Loading Surface

LSTYPE = 3 (Node Group)

2.2.4

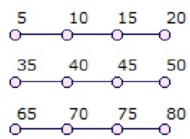
NUMNODG

NUMNODG — Number of node groups on this loading surface (Max = 100)

2.2.5

NSR, JCR, NJR, ICR, NIR — For Each Group

- NSR — Starting node number of the first row
- JCR — Node number increment in a row
- NJR — Number of nodes in a row
- ICR — Node number increment for next row
- NIR — Total number of rows



Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

2.2.6

NUMNELG

NUMNELG — Number of element groups on this loading surface (Max = 100)

2.2.7

NSR, JCR, NJR, ICR, NIR, NS — For Each Group

- NSR — Starting element number of the first row
- JCR — Element number increment in a row
- NJR — Number of elements in a row
- ICR — Element number increment for next row
- NIR — Total number of rows
- NS — Element surface number (See Mesh File Card 3.2)

5	10	15	20
35	40	45	50
65	70	75	80

Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Transmitting Boundary)</th></tr>
<tr><td rowspan="3">3

Material Property</td><td colspan="2">3.1

NUMMP

NUMMP Number of different material property (Max=5)</td></tr>
<tr><td rowspan="2">3.2

For Each Material Property</td><td>3.2.1

MATNO

MATNO Material property number</td></tr>
<tr><td>3.2.2

RO, E, V

RO Mass density
E Young's modulus
V Poisson's ratio</td></tr>
</table>

Card Group	Input Data and Definitions (Transmitting Boundary)
4 Transmitting Boundary Specification	4.1 LSNO, MATNO LSNO — Loading surface number MATNO — Material property number For MATNO = 0, loading surface is related to continuum element surface Refer to Card 9.6.3 in SMAP-2D User's Manual Repeat Card 4.1 until the last card (LSNO=0) is specified

LOAD-3D

LDTYPE = 1 [Pressure: SMAP-3D]

<table>
<tr><th>Card Group</th><th colspan="3">Input Data and Definitions (Pressure)</th></tr>
<tr><td>1
Title</td><td colspan="3">1.1
TITLE
TITLE Any title (Max = 60 characters)</td></tr>
<tr><td rowspan="4">2
Loading Surface</td><td colspan="3">2.1
NUMLS
NUMLS Number of loading surfaces where external tractions are specified (Max = 20)</td></tr>
<tr><td rowspan="3">2.2
For Each Loading Surface</td><td colspan="2">2.2.1
LSNO, LSTYPE
LSNO Loading surface number

LSTYPE = 0 All specified nodes
= 1 Polygon = 2 Plane
= 3 Line strip = 4 Points
= 5 Node group
= 6 Element group</td></tr>
<tr><td rowspan="2">LSTYPE = 0, 1, 2, 3, 4</td><td>2.2.2
NUMNODE
NUMNODE Number of nodes on this loading surface (Max = 9990)</td></tr>
<tr><td>2.2.3
NOD_1, NOD_2, ..., $NOD_{NUMNODE}$
NOD_i Specified node

Polygon (LSTYPE=1) is defined counterclockwise.
Plane (LSTYPE=2) is defined by 3 nodes.
For LSTYPE=3 and $NOD_{NUMNODE}$ < 0, absolute value of $NOD_{NUMNODE}$ is the reference node defining normal to the Line strip.</td></tr>
</table>

Card Group	Input Data and Definitions (Pressure)

2 — Loading Surface

2.2 — For Each Loading Surface

LSTYPE = 5 (Node Group)

2.2.4

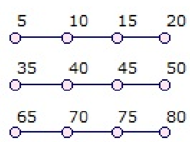
NUMNODG

NUMNODG	Number of node groups on this loading surface (Max = 100)

2.2.5

NSR, JCR, NJR, ICR, NIR — For Each Group

NSR	Starting node number of the first row
JCR	Node number increment in a row
NJR	Number of nodes in a row
ICR	Node number increment for next row
NIR	Total number of rows



Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

LSTYPE = 6 (Element Group)

2.2.6

NUMNELG

NUMNELG	Number of element groups on this loading surface (Max = 100)

2.2.7

NSR, JCR, NJR, ICR, NIR, NS — For Each Group

NSR	Starting element number of the first row
JCR	Element number increment in a row
NJR	Number of elements in a row
ICR	Element number increment for next row
NIR	Total number of rows
NS	Element surface number (See Mesh File Card 3.2)

5	10	15	20
35	40	45	50
65	70	75	80

Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Pressure)</th></tr>
<tr><td rowspan="6">3

Pressure Function</td><td colspan="2">3.1

NUMLP
NUMLP Number of pressure functions (Max = 20)</td></tr>
<tr><td rowspan="5">3.2

For Each Pressure Function</td><td>3.2.1

LPNO, LPTYPE
LPNO Pressure function number
LPTYPE = 0 Use effective surface
= 1 Use actual surface

Note: Effective surface is normal to force direction (Ex. Wind load)</td></tr>
<tr><td>3.2.2

a_{xo}, a_{xx}, a_{xy}, a_{xz}
a_{xi} Coefficients defining surface traction in the x-direction.
$P_x = a_{xo} + a_{xx}x + a_{xy}y + a_{xz}z$</td></tr>
<tr><td>3.2.3

a_{yo}, a_{yx}, a_{yy}, a_{yz}
a_{yi} Coefficients defining surface traction in the y-direction.
$P_y = a_{yo} + a_{yx}x + a_{yy}y + a_{yz}z$</td></tr>
<tr><td>3.2.4

a_{zo}, a_{zx}, a_{zy}, a_{zz}
a_{zi} Coefficients defining surface traction in the z-direction.
$P_z = a_{zo} + a_{zx}x + a_{zy}y + a_{zz}z$</td></tr>
<tr><td>3.2.5

a_{no}, a_{nx}, a_{ny}, a_{nz}
a_{ni} Coefficients defining surface traction normal to surface. Acting on actual surface
$P_n = a_{no} + a_{nx}x + a_{ny}y + a_{nz}z$</td></tr>
</table>

Card Group	Input Data and Definitions (Pressure)		
4 Pressure History	4.1		
	NUMLH		
		NUMLH	Number of pressure histories (Max = 20)
	4.2 For Each Pressure History	4.2.1	
		LHNO	
		LHNO	Pressure history number
		4.2.2	
		NUMTP	
		NUMTP	Number of time points (Max = 1000)
		4.2.3	
		$T_1, T_2, \ldots, T_{NUMTP}$	
		T_i	Specified time
		4.2.4	
		$C_1, C_2, \ldots, C_{NUMTP}$	
		C_i	Pressure intensity at time T

Card Group	Input Data and Definitions (Pressure)
5 Pressure Specification	5.1 LSNO, LPNO, LHNO LSNO Loading surface number LPNO Pressure function number LHNO Pressure history number Repeat Card 5.1 until the last card (LSNO=0) is specified

LOAD-3D

LDTYPE = 2 [Velocity: SMAP-3D]

<table>
<tr><th>Card Group</th><th colspan="3">Input Data and Definitions (Velocity)</th></tr>
<tr><td>1
Title</td><td colspan="3">1.1
TITLE
TITLE Any title (Max = 60 characters)</td></tr>
<tr><td rowspan="4">2
Loading Surface</td><td colspan="3">2.1
NUMLS
NUMLS Number of loading surfaces where velocities are specified (Max = 20)</td></tr>
<tr><td rowspan="3">2.2
For Each Loading Surface</td><td colspan="2">2.2.1
LSNO, LSTYPE
LSNO Loading surface number
LSTYPE = 0 All specified nodes
= 1 Polygon = 2 Plane
= 3 Line strip = 4 Points
= 5 Node group
= 6 Element group</td></tr>
<tr><td rowspan="2">LSTYPE = 0, 1, 2, 3, 4</td><td>2.2.2
NUMNODE
NUMNODE Number of nodes on this loading surface (Max = 9990)</td></tr>
<tr><td>2.2.3
NOD_1, NOD_2, ..., $NOD_{NUMNODE}$
NOD_i Specified node

Polygon (LSTYPE=1) is defined counterclockwise.
Plane (LSTYPE=2) is defined by 3 nodes.
For LSTYPE=3 and $NOD_{NUMNODE} < 0$, absolute value of $NOD_{NUMNODE}$ is the reference node defining normal to the Line strip.</td></tr>
</table>

Card Group	Input Data and Definitions (Velocity)

2 Loading Surface

2.2 For Each Loading Surface

LSTYPE = 5 (Node Group)

2.2.4

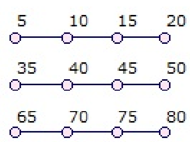
NUMNODG

NUMNODG	Number of node groups on this loading surface (Max = 100)

2.2.5

NSR, JCR, NJR, ICR, NIR   For Each Group

NSR	Starting node number of the first row
JCR	Node number increment in a row
NJR	Number of nodes in a row
ICR	Node number increment for next row
NIR	Total number of rows



Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

LSTYPE = 6 (Element Group)

2.2.6

NUMNELG

NUMNELG	Number of element groups on this loading surface (Max = 100)

2.2.7

NSR, JCR, NJR, ICR, NIR, NS   For Each Group

NSR	Starting element number of the first row
JCR	Element number increment in a row
NJR	Number of elements in a row
ICR	Element number increment for next row
NIR	Total number of rows
NS	Element surface no. (See Mesh File Card 3.2)

5	10	15	20
35	40	45	50
65	70	75	80

Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

Card Group		Input Data and Definitions (Velocity)
3 Velocity Function	3.1	NUMLV NUMLV Number of velocity functions (Max = 20)
	3.2 For Each Velocity Function	3.2.1 LVNO, LVTYPE LVNO Velocity function number LVTYPE = 0 Apply individual components Cards 3.2.2 - 3.2.4 = 1 Apply normal components Cards 3.2.5
		3.2.2 a_{xo}, a_{xx}, a_{xy}, a_{xz} a_{xi} Coefficients defining velocity in x-direction $V_x = a_{xo} + a_{xx}x + a_{xy}y + a_{xz}z$
		3.2.3 a_{yo}, a_{yx}, a_{yy}, a_{yz} a_{yi} Coefficients defining velocity in y-direction $V_y = a_{yo} + a_{yx}x + a_{yy}y + a_{yz}z$
		3.2.4 a_{zo}, a_{zx}, a_{zy}, a_{zz} a_{zi} Coefficients defining velocity in z-direction $V_z = a_{zo} + a_{zx}x + a_{zy}y + a_{zz}z$
		3.2.5 a_{no}, a_{nx}, a_{ny}, a_{nz} a_{ni} Coefficients defining velocity normal to surface $V_n = a_{no} + a_{nx}x + a_{ny}y + a_{nz}z$

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Velocity)</th></tr>
<tr><td rowspan="5">4

Velocity History</td><td colspan="2">4.1

NUMLH

NUMLH Number of velocity histories (Max = 20)</td></tr>
<tr><td rowspan="4">4.2

For Each Velocity History</td><td>4.2.1

LHNO

LHNO Velocity history number</td></tr>
<tr><td>4.2.2

NUMTP

NUMTP Number of time points (Max = 1000)</td></tr>
<tr><td>4.2.3

T_1, T_2, ..., T_{NUMTP}

T_i Specified time</td></tr>
<tr><td>4.2.4

C_1, C_2, ..., C_{NUMTP}

C_i Velocity intensity at time T_i</td></tr>
</table>

<table>
<tr><th>Card Group</th><th>Input Data and Definitions (Velocity)</th></tr>
<tr><td>5

Velocity Specification</td><td>5.1

LSNO, LVNO, LHNO

LSNO Loading surface number
LVNO Velocity function number
LHNO Velocity history number

Repeat Card 5.1 until the last card (LSNO=0) is specified</td></tr>
</table>

LOAD-3D

LDTYPE = 3 [Initial Velocity: SMAP-3D]

<table>
<tr><th>Card Group</th><th colspan="3">Input Data and Definitions (Initial Velocity)</th></tr>
<tr><td>1
Title</td><td colspan="3">1.1
TITLE
TITLE Any title (Max = 60 characters)</td></tr>
<tr><td rowspan="4">2
Loading Surface</td><td colspan="3">2.1
NUMLS
NUMLS Number of loading surfaces where initial velocities are specified (Max = 20)</td></tr>
<tr><td rowspan="3">2.2
For Each Loading Surface</td><td colspan="2">2.2.1
LSNO, LSTYPE
LSNO Loading surface number
LSTYPE = 0 All specified nodes
= 1 Polygon = 2 Plane
= 3 Line strip = 4 Points
= 5 Node group
= 6 Element group</td></tr>
<tr><td rowspan="2">LSTYPE = 0, 1, 2, 3, 4</td><td>2.2.2
NUMNODE
NUMNODE Number of nodes on this loading surface (Max = 9990)</td></tr>
<tr><td>2.2.3
NOD_1, NOD_2, ..., $NOD_{NUMNODE}$
NOD_i Specified node

Polygon (LSTYPE=1) is defined counterclockwise.
Plane (LSTYPE=2) is defined by 3 nodes.
For LSTYPE=3 and $NOD_{NUMNODE}$ < 0, absolute value of $NOD_{NUMNODE}$ is the reference node defining normal to the Line strip.</td></tr>
</table>

Card Group	Input Data and Definitions (Initial Velocity)
2 Loading Surface	2.2 For Each Loading Surface

LSTYPE = 5 (Node Group)

2.2.4

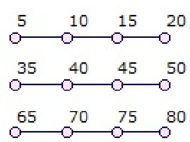
NUMNODG

NUMNODG — Number of node groups on this loading surface (Max = 100)

2.2.5

NSR, JCR, NJR, ICR, NIR — For Each Group

NSR	Starting node number of first row
JCR	Node number increment in a row
NJR	Number of nodes in a row
ICR	Node number increment for next row
NIR	Total number of rows



Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

LSTYPE = 6 (Element Group)

2.2.6

NUMNELG

NUMNELG — Number of element groups on this loading surface (Max = 100)

2.2.7

NSR, JCR, NJR, ICR, NIR, NS — For Each Group

NSR	Starting element number of first row
JCR	Element number increment in a row
NJR	Number of elements in a row
ICR	Element number increment for next row
NIR	Total number of rows
NS	Element surface no. (See Mesh File Card 3.2)

5	10	15	20
35	40	45	50
65	70	75	80

Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Initial Velocity)</th></tr>
<tr><td rowspan="6">3
Initial Velocity Function</td><td colspan="2">3.1
NUMLIV
NUMLIV Number of initial velocity functions (Max = 20)</td></tr>
<tr><td rowspan="5">3.2
For Each Initial Velocity Function</td><td>3.2.1
LIVNO, LIVTYPE
LIVNO Initial velocity function number
LIVTYPE = 0 Apply individual components
Cards 3.2.2 - 3.2.4
= 1 Apply normal components
Cards 3.2.5</td></tr>
<tr><td>3.2.2
a_{xo}, a_{xx}, a_{xy}, a_{xz}
a_{xi} Coefficients defining initial velocity in the x-direction.
$V_{ix} = a_{xo} + a_{xx}x + a_{xy}y + a_{xz}z$</td></tr>
<tr><td>3.2.3
a_{yo}, a_{yx}, a_{yy}, a_{yz}
a_{yi} Coefficients defining initial velocity in the y-direction.
$V_{iy} = a_{yo} + a_{yx}x + a_{yy}y + a_{yz}z$</td></tr>
<tr><td>3.2.4
a_{zo}, a_{zx}, a_{zy}, a_{zz}
a_{zi} Coefficients defining initial velocity in the z-direction.
$V_{iz} = a_{zo} + a_{zx}x + a_{zy}y + a_{zz}z$</td></tr>
<tr><td>3.2.5
a_{no}, a_{nx}, a_{ny}, a_{nz}
a_{ni} Coefficients defining initial velocity normal to the surface.
$V_{in} = a_{no} + a_{nx}x + a_{ny}y + a_{nz}z$</td></tr>
</table>

Card Group	Input Data and Definitions (Initial Velocity)
4 Initial Velocity Specification	4.1 LSNO, LIVNO LSNO Loading surface number LIVNO Initial velocity function Repeat Card 4.1 until the last card (LSNO=0) is specified

LOAD-3D

LDTYPE = 4 [Acceleration: SMAP-3D]

<table>
<tr><th>Card Group</th><th colspan="3">Input Data and Definitions (Acceleration)</th></tr>
<tr><td>1
Title</td><td colspan="3">1.1
TITLE
TITLE Any title (Max = 60 characters)</td></tr>
<tr><td rowspan="4">2
Loading Surface</td><td colspan="3">2.1
NUMLS
NUMLS Number of loading surfaces where accelerations are specified (Max = 20)</td></tr>
<tr><td rowspan="3">2.2
For Each Loading Surface</td><td colspan="2">2.2.1
LSNO, LSTYPE
LSNO Loading surface number
LSTYPE = 0 All specified nodes
= 1 Polygon = 2 Plane
= 3 Line strip = 4 Points
= 5 Node group
= 6 Element group</td></tr>
<tr><td rowspan="2">LSTYPE = 0, 1, 2, 3, 4</td><td>2.2.2
NUMNODE
NUMNODE Number of nodes on this loading surface (Max = 9990)</td></tr>
<tr><td>2.2.3
NOD_1, NOD_2, ..., $NOD_{NUMNODE}$
NOD_i Specified node

Polygon (LSTYPE=1) is defined counterclockwise.
Plane (LSTYPE=2) is defined by 3 nodes.
For LSTYPE=3 and $NOD_{NUMNODE}$ < 0, absolute value of $NOD_{NUMNODE}$ is the reference node defining normal to the Line strip.</td></tr>
</table>

Card Group	Input Data and Definitions (Acceleration)

2 Loading Surface

2.2 For Each Loading Surface

LSTYPE = 5 (Node Group)

2.2.4

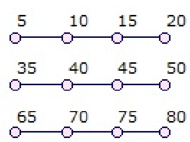
NUMNODG

NUMNODG	Number of node groups on this loading surface (Max = 100)

2.2.5

NSR, JCR, NJR, ICR, NIR For Each Group

NSR	Starting node number of the first row
JCR	Node number increment in a row
NJR	Number of nodes in a row
ICR	Node number increment for next row
NIR	Total number of rows



Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

LSTYPE = 6 (Element Group)

2.2.6

NUMNELG

NUMNELG	Number of element groups on this loading surface (Max = 100)

2.2.7

NSR, JCR, NJR, ICR, NIR, NS For Each Group

NSR	Starting element number of the first row
JCR	Element number increment in a row
NJR	Number of elements in a row
ICR	Element number increment for next row
NIR	Total number of rows
NS	Element surface no. (See Mesh File Card 3.2)

5	10	15	20
35	40	45	50
65	70	75	80

Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

<table>
<tr><th>Card Group</th><th colspan="3">Input Data and Definitions (Acceleration)</th></tr>
<tr><td rowspan="6">3

Acceleration Function</td><td colspan="3">3.1

NUMLA
NUMLA   Number of acceleration functions (Max = 20)</td></tr>
<tr><td rowspan="5">3.2

For Each Acceleration Function</td><td colspan="2">3.2.1

LANO, LATYPE
LANO   Acceleration function number
LATYPE   = 0   Apply individual components (Cards 3.2.2 - 3.2.4)
= 1   Apply normal components (Cards 3.2.5)</td></tr>
<tr><td colspan="2">3.2.2

a_{xo}, a_{xx}, a_{xy}, a_{xz}
a_{xi} Coefficients defining acceleration in the x-direction
$A_x = a_{xo} + a_{xx}x + a_{xy}y + a_{xz}z$</td></tr>
<tr><td colspan="2">3.2.3

a_{yo}, a_{yx}, a_{yy}, a_{yz}
a_{yi} Coefficients defining acceleration in the y-direction.
$A_y = a_{yo} + a_{yx}x + a_{yy}y + a_{yz}z$</td></tr>
<tr><td colspan="2">3.2.4

a_{zo}, a_{zx}, a_{zy}, a_{zz}
a_{zi} Coefficients defining acceleration In the z-direction.
$A_z = a_{zo} + a_{zx}x + a_{zy}y + a_{zz}z$</td></tr>
<tr><td colspan="2">3.2.5

a_{no}, a_{nx}, a_{ny}, a_{nz}
a_{ni} Coefficients defining acceleration normal to the surface.
$A_n = a_{no} + a_{nx}x + a_{ny}y + a_{nz}z$</td></tr>
</table>

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Acceleration)</th></tr>
<tr><td rowspan="5">4
Acceleration History</td><td colspan="2">4.1
NUMLH
NUMLH Number of acceleration histories (Max = 20)</td></tr>
<tr><td rowspan="4">4.1
For Each Acceleration History</td><td>4.2.1
LHNO
LHNO Acceleration history number</td></tr>
<tr><td>4.2.2
NUMTP
NUMTP Number of time points (Max = 1000)</td></tr>
<tr><td>4.2.3
$T_1, T_2, \ldots, T_{NUMTP}$
T_i Specified time</td></tr>
<tr><td>4.2.4
$C_1, C_2, \ldots, C_{NUMTP}$
C_i Acceleration intensity at time T_i</td></tr>
</table>

Card Group	Input Data and Definitions (Acceleration)
5 Acceleration Specification	5.1 LSNO, LANO, LHNO LSNO Loading surface number LANO Acceleration function number LHNO Acceleration history number Repeat Card 5.1 until the last card (LSNO=0) is specified

LOAD-3D

LDTYPE = 5 [Transmitting Boundary: SMAP-3D]

Card Group	Input Data and Definitions (Transmitting Boundary)		
1 Title	1.1 TITLE TITLE Any title (Max = 60 characters)		
2 Loading Surface	2.1 NUMLS NUMLS Number of loading surfaces where transmitting boundaries are specified (Max = 20)		
	2.2 For Each Loading Surface	2.2.1 LSNO, LSTYPE LSNO Loading surface number LSTYPE = 0 All specified nodes = 1 Polygon = 2 Plane = 3 Line strip = 4 Points = 5 Node group = 6 Element group	
		LSTYPE = 0, 1, 2, 3, 4	2.2.2 NUMNODE NUMNODE Number of nodes on this loading surface (Max = 9990)
			2.2.3 NOD_1, NOD_2, ..., $NOD_{NUMNODE}$ NOD_i Specified node Polygon (LSTYPE=1) is defined counterclockwise. Plane (LSTYPE=2) is defined by 3 nodes. For LSTYPE=3 and $NOD_{NUMNODE}$ < 0, absolute value of $NOD_{NUMNODE}$ is the reference node defining normal to the Line strip.

Card Group	Input Data and Definitions (Transmitting Boundary)

2 Loading Surface

2.2 For Each Loading Surface

LSTYPE = 5 (Node Group)

2.2.4

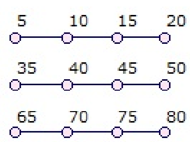
NUMNODG

NUMNODG — Number of node groups on this loading surface (Max = 100)

2.2.5

NSR, JCR, NJR, ICR, NIR — For Each Group

NSR	Starting node number of the first row
JCR	Node number increment in a row
NJR	Number of nodes in a row
ICR	Node number increment for next row
NIR	Total number of rows



Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

LSTYPE = 6 (Element Group)

2.2.6

NUMNELG

NUMNELG — Number of element groups on this loading surface (Max = 100)

2.2.7

NSR, JCR, NJR, ICR, NIR, NS — For Each Group

NSR	Starting element number of the first row
JCR	Element number increment in a row
NJR	Number of elements in a row
ICR	Element number increment for next row
NIR	Total number of rows
NS	Element surface no. (See Mesh File Card 3.2)

5	10	15	20
35	40	45	50
65	70	75	80

Example
NSR = 5
JCR = 5
NJR = 4
ICR = 30
NIR = 3

<table>
<tr><th>Card Group</th><th colspan="2">Input Data and Definitions (Transmitting Boundary)</th></tr>
<tr><td rowspan="3">3

Material Property</td><td colspan="2">3.1

NUMMP

NUMMP Number of different material property (Max = 20)</td></tr>
<tr><td rowspan="2">3.2

For Each Material Property</td><td>3.2.1

MATNO

MATNO Material property number</td></tr>
<tr><td>3.2.2

RO, E, V

RO Mass density
E Young's modulus
V Poisson's ratio</td></tr>
</table>

Card Group	Input Data and Definitions (Transmitting Boundary)
4 Transmitting Boundary Specification	4.1 LSNO, MATNO LSNO Loading surface number MATNO Material property number Repeat Card 4.1 until the last card (LSNO=0) is specified

XY Graph
User's Manual

12.1 Introduction

XY Graph is a two-dimensional graph consisting of lines connecting each pair of data points, which can be plotted by PLOT XY or EXCEL. Figure 12.1 shows schematic flow diagram of plotting simple form of Draft XY data in Table 12.1.

This Draft XY is changed into Standard XY by Converter DS. Then Standard XY can be plotted by directly PLOT XY or by EXCEL with the aid of Converter SE.

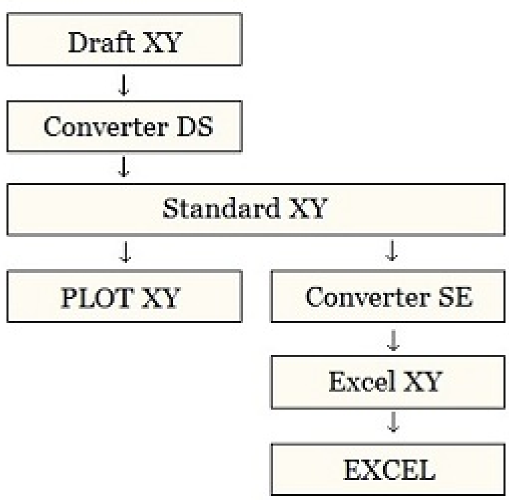


Figure 12.1 Flow diagram of plotting XY graph

Table 12.1 Draft XY Data Format

Card Group		Input Data and Definitions		
First Plot	Title	Title		(Max 50 Characters)
		Sub Title		(Max 50 Characters)
		X-Label		(Max 50 Characters)
		Y-Label		(Max 50 Characters)
	First Curve	X_1	Y_1	
		X_2	Y_2	
		-	-	
		X_n	Y_n	
		0.0	123456	(End of Curve)
		Legend 1		(Max 20 Characters)
		Legend 2		(Max 20 Characters))
	Second Curve	X_1	Y_1	
		X_2	Y_2	
		-	-	
		X_n	Y_n	
		0.0	123456	(End of Curve)
		Legend 1		(Max 20 Characters)
		Legend 2		(Max 20 Characters)
	Last Curve	X_1	Y_1	
		X_2	Y_2	
		-	-	
		X_n	Y_n	
		0.0	123456	(End of Curve)
		Legend 1		(Max 20 Characters)
		Legend 2		(Max 20 Characters)
		0.0	987654	(End of Plot)
Next Plot		Next Plot can be added using the same format as the First Plot		

12.2 New Graph

XY Graph can be created by performing the following steps:

Step 1:

Select the following menu items in SMAP:

Plot → XY → PLOT XY → New

Step 2:

Once selected, initial default file XY.dat will be opened by Notepad as listed in Table 12.2.

Edit this default file according to the format of Draft XY Data in Table 12.1. And then save and exit.

Step 3:

Draft XY.dat is automatically changed into Standard Form by Converter DS as listed in Table 12.3.
Modified graph will be displayed on PLOT XY drawing board.

Step 4:

XY Graph can be further modified by Edit Dialog explained in detail in the next Section 12.3.

Table 12.2 Draft XY Data (Initial Default File XY.dat)

```
Plot No. 1
Sub Title 1
XLabel-1
YLabel-1
 0        10
 100      20
 .000000E+00   .123456E+06
Curve 1
Legend
10,     20
90,     30
 .000000E+00   .123456E+06
Curve 2
Legend
 .000000E+00   .987654E+06
Plot No. 2
Sub Title 2
XLabel-2
YLabel-2
0       100
1000    200
 .000000E+00   .123456E+06
Curve 1
Legend
100     200
900     300
.000000E+00   .123456E+06
Curve 2
Legend
.000000E+00   .987654E+06
Plot No. 3
Sub Title 3
XLabel-3
YLabel-3
0       100
1000    200
.000000E+00   .123456E+06
Curve 1
Legend
200,    200
900,    300
.000000E+00   .123456E+06
Curve 2
Legend
.000000E+00   .987654E+06
```

Table 12.3 Standard XY Data (Initial Default File XY.dat)

```
*************************************************************
*                            PLOT NO:    1                  *
*************************************************************
C Following data can be modified for plotting configuration
 TITLE     (50 CHAR) = Plot No. 1
 SUB-TITLE (50 CHAR) = Sub Title 1
 XLABLE    (50 CHAR) = XLabel-1
 YLABLE    (50 CHAR) = YLabel-1
C
 MAN.-SCALE :     IXY = 1
 LEGEND-OPT. :    ILG = 1
 TOTAL CURVE :    NLG = 2
 LEGEND-LEN : DXLEGN = 0.0
C
C IELEM= 0: no list data, list X-label & X-tick number
C        1:    list data, list X-label & X-tick number
C       -2:    node data, list node numbers only
C        2: element data, list element numbers only
C       -3:    node data, list node no, X-tick no. & X-label
C        3: element data, list elem no, X-tick no. & X-label
 EL-LIST-OPT : IELEM = 0
C
 FRAMING :        IFM = 1
 CENTERING :    ICENL = 1
 GRIDDING :     IGRID = 1
C X-coordinate data
                 XMAX = 5.0
                 NODX = 6
                   XS =  .000000E+00
                   XE =  .120000E+03
                NXDEC =-1
               XSCALE = 1.0
C
                IGENX = 0
               XDELTA = 0.0
C
                 LOGX = 0
                  NXD = 0
C Y-coordinate data
                 YMAX = 5.0
                 NODY = 6
                   YS =  .800000E+01
                   YE =  .320000E+02
                NYDEC = 2
               YSCALE = 1.0
C
                 LOGY = 0
                  NYD = 0
C Individual Curve
C                  NO :   1   2   3   4   5   6   7   8   9  10
                 HIDE =   0   0   0   0   0   0   0   0   0   0
                 LINE =   1   1   1   1   1   1   1   1   1   1
                 DASH =   1   2   3   4   5   6   7   8   9  10
                 MARK =   1   2   3   4   5   6   7   8   9  10
                 COLR =   1   2   3   4   5   6   7   8   9  10
C ***********************************************************
  .000000E+00   .100000E+02
  .100000E+03   .200000E+02
  .000000E+00   .123456E+06
Curve 1
Legend
  .100000E+02   .200000E+02
  .900000E+02   .300000E+02
  .000000E+00   .123456E+06
Curve 2
Legend
  .000000E+00   .987654E+06
```

12.3 Edit Dialog

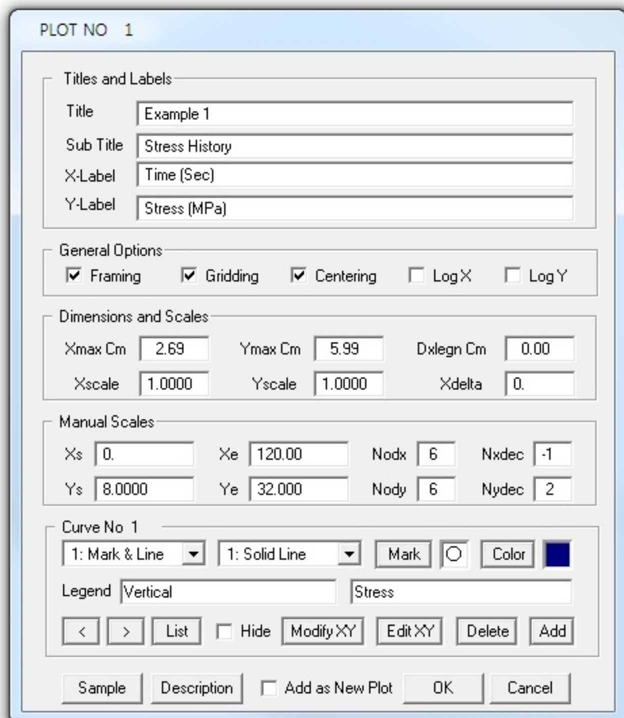
Edit Dialog in Figure 12.2 can be accessed by selecting the Edit menu in PLOT-XY.

Edit Dialog consists of following six parts:

- Titles and Labels
- General Options
- Dimensions and Scales
- Manual Scales
- Curve Data
- Command Buttons & Check Box

Refer to description in Sample Graph in Figure 12.3.

Figure 12.2
Edit dialog



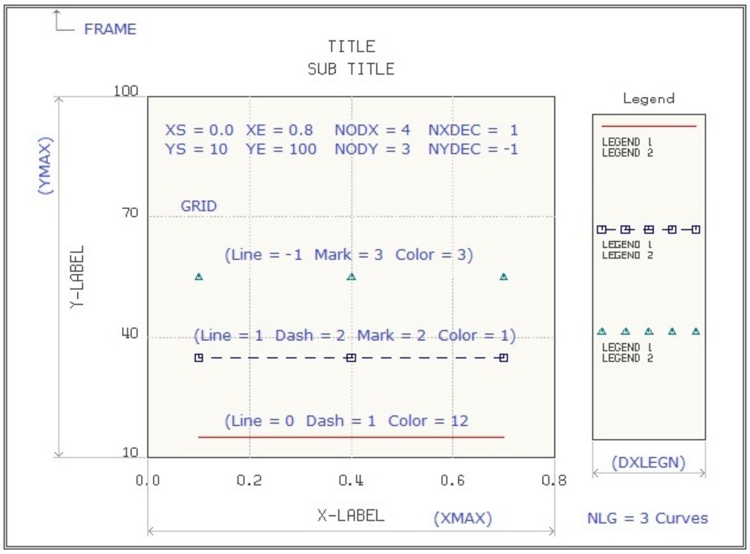


Figure 12.3 Sample graph

12.3.1 Titles and Labels

Here, you type:
Title, Sub Title, X-Label, and Y-Label.

12.3.2 General Options

Check the box for the option item to be active:

Framing	Draw Frame
Gridding	Draw Grid lines
Centering	Center Titles and X & Y Labels
Log X	Log scale in X axis
Log Y	Log scale in Y axis

12.3.3 Dimensions and Scales

Refer to description in Sample Graph in Figure 12.3.

12.3.4 Manual Scales

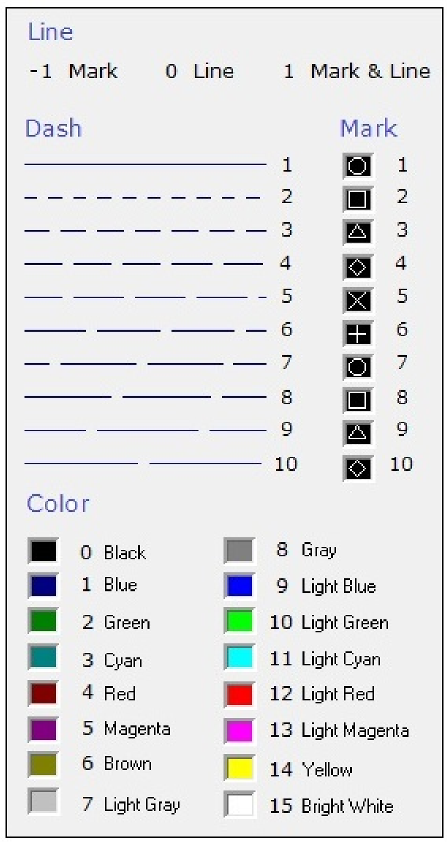
Refer to description in Sample Graph in Figure 12.3.

12.3.5 Curve Data

For each curve, you can select Line type, Dash type, Mark type, Color as in Figure 12.4, and type in Legends.

Check Hide Curve to hide the current curve.

Figure 12.4
Curve options



Curve Data has the following seven command buttons:

Back	Open previous curve
Next	Open next curve
List	List all curves as in Figure 12.5a
Modify XY	Modify current curve XY data as in Figure 12.5b
Edit XY	Edit current curve XY data
Delete	Delete current curve
Add	Add new curve to current plot

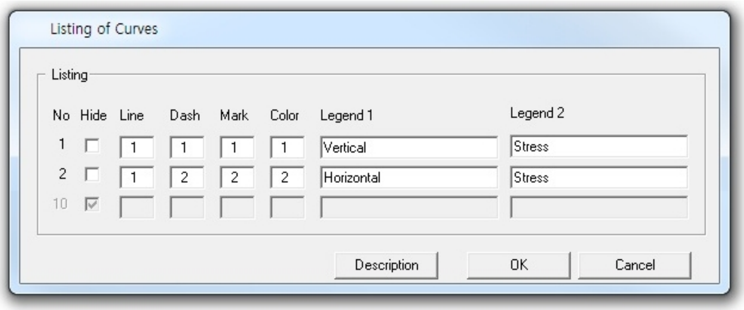


Figure 12.5a Listing of curves

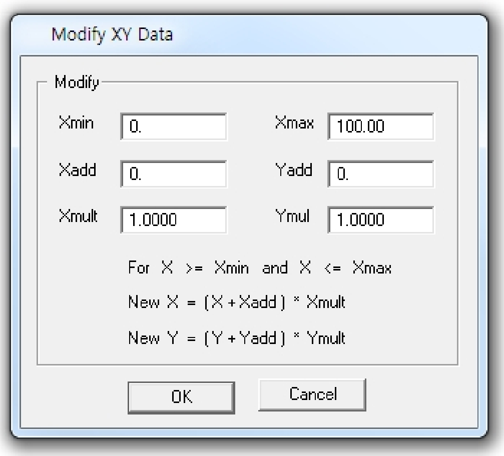


Figure 12.5b Modify current curve XY data

12.3.6 Command Buttons & Check Box

Sample	Show Sample graph in Figure 12.3
Description	Show Curve options in Figure 12.4
Add as New Plot	Copy Current plot and Add as New plot
OK	Save and exit Edit dialog
Cancel	Cancel and exit Edit dialog

12.4 Existing Graph

XY Graph can be opened by performing the following steps:

Step 1:

Select the following menu items in SMAP:
Plot → XY → PLOT XY → Open

Step 2:

If input file is Draft Form, then it will be automatically changed into Standard Form by Converter DS as listed in Table 12.3.

XY Graph will be displayed on PLOT XY drawing board.

Step 3:

XY Graph can be modified by Edit Dialog as explained in detail in the previous Section 12.3.

Refer to samples in the following directory:

C:\Smap \Smap3D \Example \XY_Graph \PLOT XY Graph Sample.docx

12.5 Excel XY Graph

Excel XY Graph can be made by performing the following steps:

Step 1:

Select the following menu items in SMAP:
Plot → XY → EXCEL → Open

Step 2:

If input file is Draft Form, then it will be automatically changed into Standard Form by Converter DS as listed in Table 12.3.

Then this Standard XY Graph will be changed into Excel Form by Converter SE and displayed on EXCEL Spreadsheet as shown in Figure 12.6.

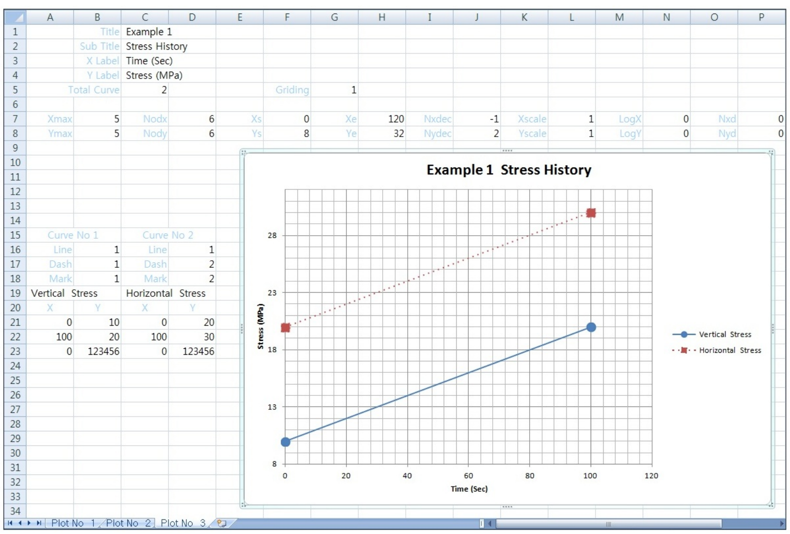


Figure 12.6 XY graph on Excel spread sheet

Notes on Excel XY Graph

Excel XY Graph can be influenced by the following input parameters in Standard Form:

Note 1: Input Parameters Not Considered

Following parameters are not considered:
Plot dimensions: XMAX , YMAX
Number of digits after decimal point: NXDEC, NYDEC

Note 2: Automatic Scaling (Xscale = 0, Yscale = 0)

For XSCALE = 0
X axis is automatically scaled and XS, XE and NODX are not used.

For YSCALE = 0
Y axis is automatically scaled and YS, YE and NODY are not used.

Note 3: Logarithmic Scaling (Logx = 1, Logy = 1)

For LOGX = 1
NODX and NXD are not used.

If XSCALE ≠ 0 and XS < 1 and XE > 1, XS is automatically scaled.

For LOGY = 1
NODY and NYD are not used.

If YSCALE ≠ 0 and YS < 1 and YE > 1, YS is automatically scaled.

Refer to samples in the following directory:
C:\Smap \Smap3D \Example \XY_Graph \Excel XY Graph Sample.pdf

12.6 SMAP Results

Figure 12.7 shows schematic flow diagram of processing SMAP Results corresponding to Card Group 12 in SMAP Post File.

This Standard Form of PlotXy.dat can be opened by either PLOT XY or EXCEL spreadsheet.

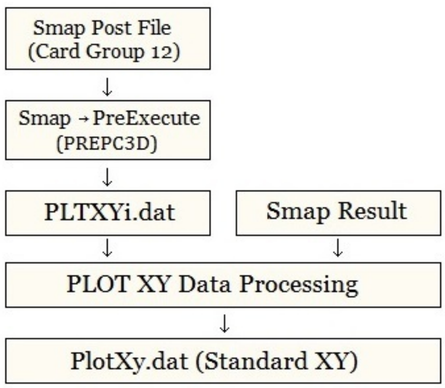


Figure 12.7 Processing SMAP results

SMAP Results can be plotted by performing following steps:

Step 1:

Select the following menu items in SMAP:
Plot → Result

Step 2:

Select PLOT XY
in Plot Menu dialog
in Figure 12.8.

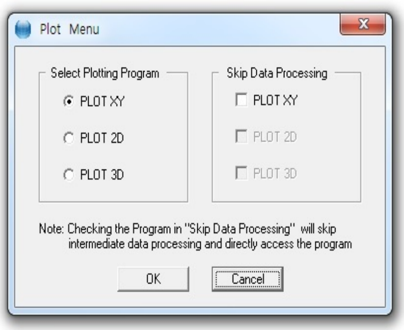


Figure 12.8
Plot menu dialog

12.6.1 PLOT XY Setup

PLOT XY Setup in Figure 12.9 can be accessed by selecting the following item in SMAP main menu.
Setup → PLOT XY

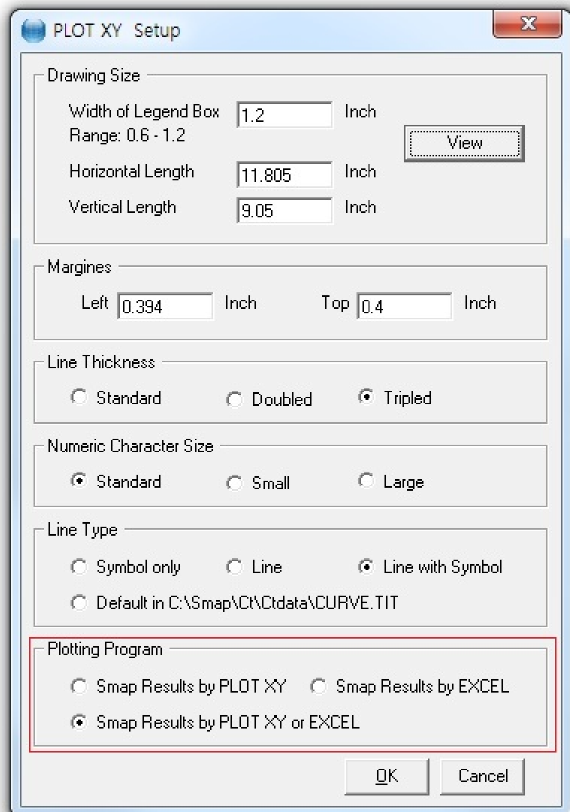


Figure 12.9 PLOT XY setup dialog

Refer to description in Sample Graph in Figure 12.4.

12.7 PlotXY Generator

PlotXY Generator is the graphical user interface which is mainly used to generate or edit Simplified Time History and Simplified Snapshot of Card Group 12 in SMAP Post File.

All different cases will be discussed in the following sections.

12.7.1 Accessing PlotXY Generator

PlotXY Generator can be accessed by selecting the following item in SMAP main menu as in Figure 12.10.
Run → PlotXY Generator → New / Open

New is used to generate new Post File.

You can edit sample input with all different cases.

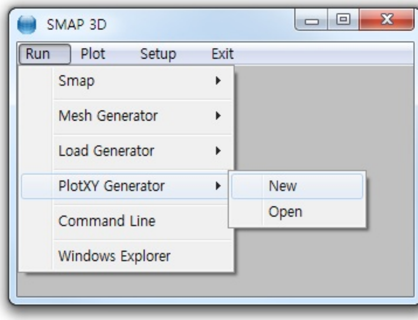


Figure 12.10
Menu for PlotXY Generator

Open is used to edit existing Post File. You can specify different output Post File name as shown in Figure 12.11.

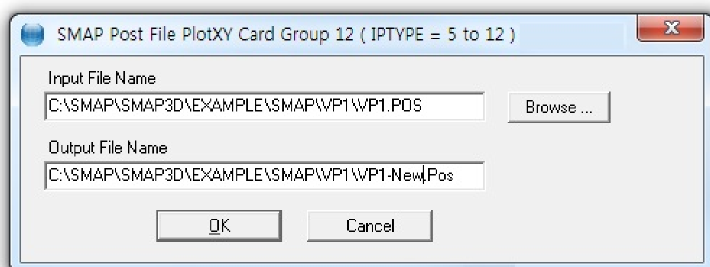


Figure 12.11 PlotXY input and output file dialog

12.7.2 Time History for a Given Element

Main Dialog for Time History of Stresses / Strains for a Given Element (IPTYPE = 5) is shown in Figure 12.12.

Element should be listed in Card 10.2.2 in SMAP Main File.
Table shows available data as in Figure 12.13.

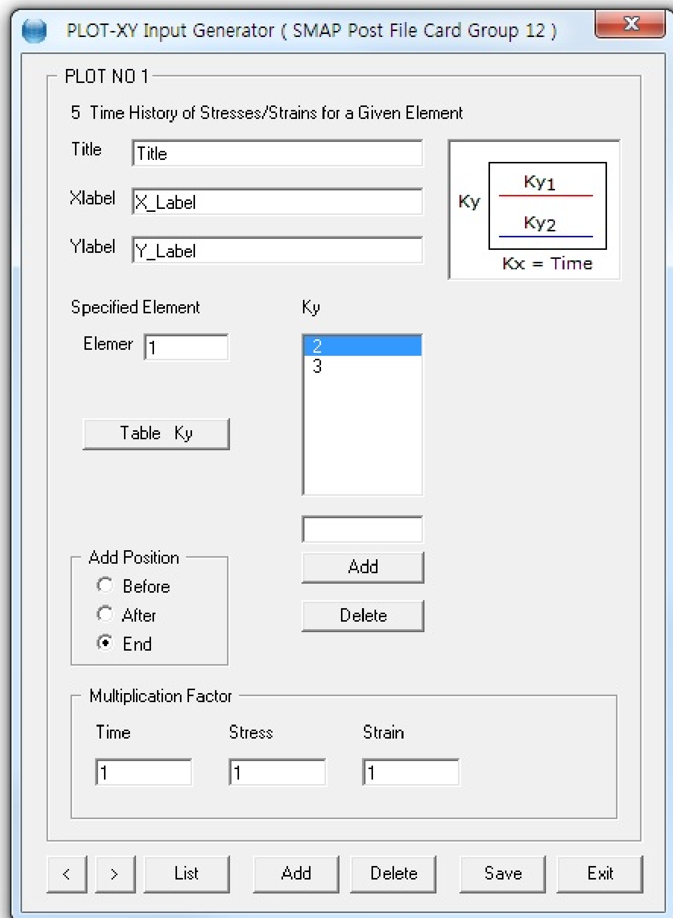


Figure 12.12 Time history for a given element

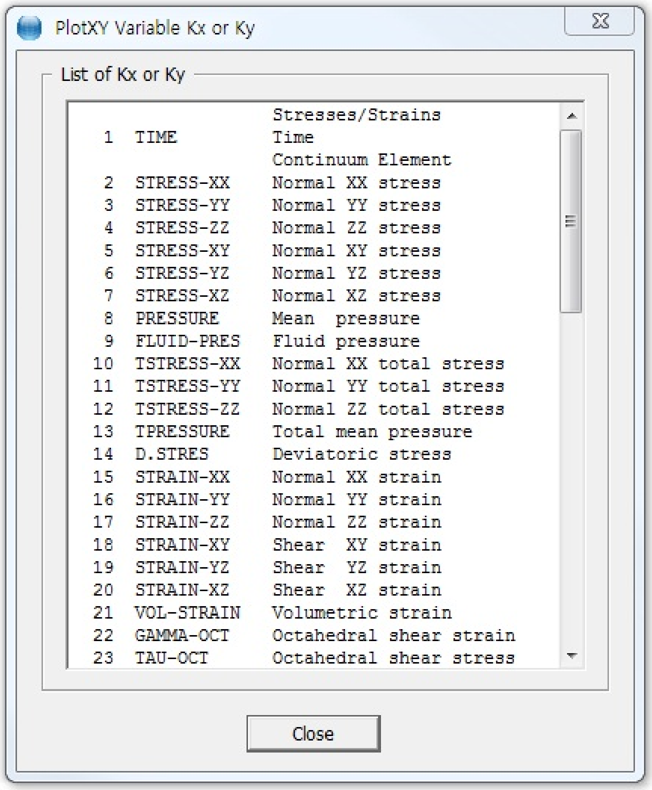


Figure 12.13 Available data for stresses / strains

Buttons at Main Dialog Bottom

Back	Show previous plot
Next	Show next plot
List	Show listing of all plots
Add	Add new plot at the end
Delete	Delete the current plot
Save	Save all updates
Exit	Save and exit

List shows summary of all plots as shown in Figure 12.14.

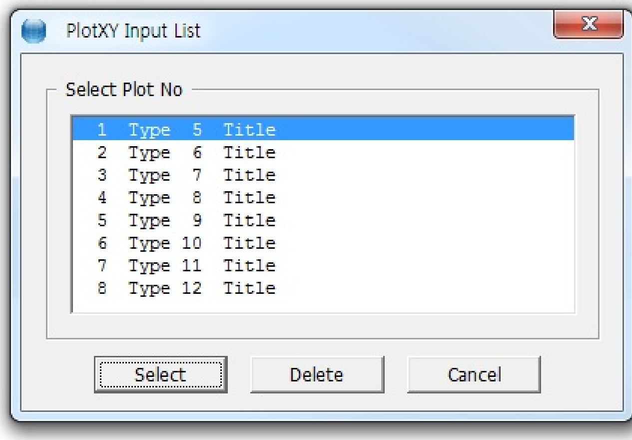


Figure 12.14 Listing of plots

Add shows new plot type to be added as in Figure 12.15.

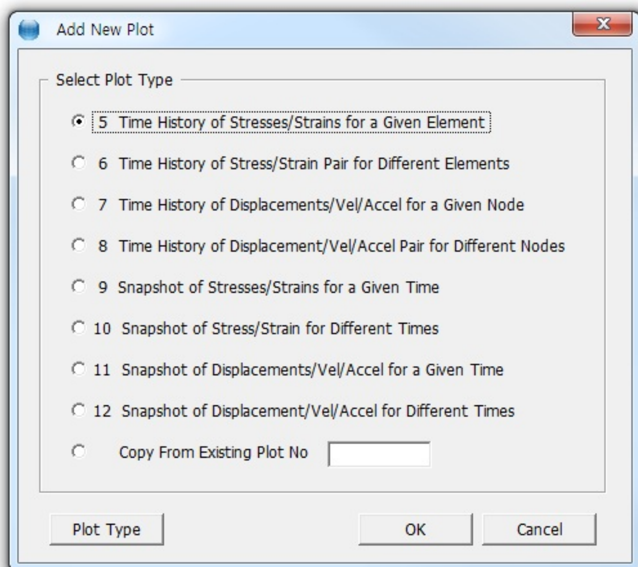


Figure 12.15 Add options for new plot

Plot Type in Add dialog illustrates graphically available plot types as shown schematically in Figure 12.16.

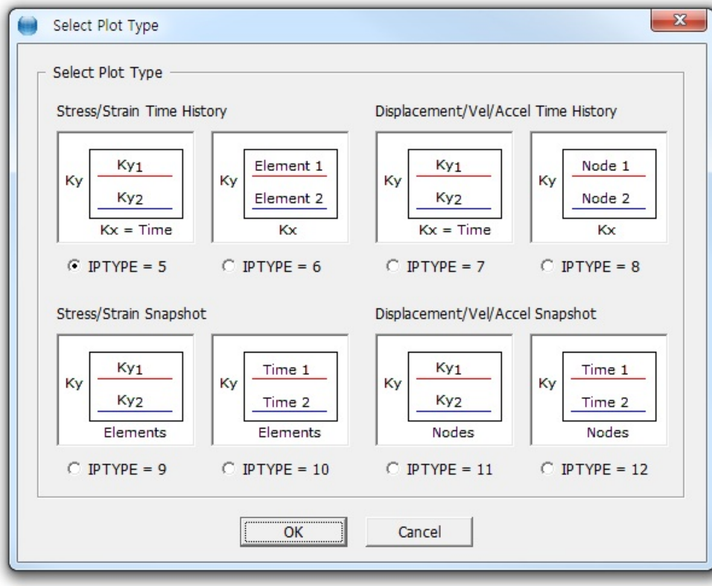


Figure 12.16 Available plot types

12.7.3 Time History for Different Elements

Main Dialog for Time History of Stresses / Strains for Different Elements (IPTYPE = 6) is shown in Figure 12.17.

Elements should be listed in Card 10.2.2 in SMAP Main File.
Table shows available data as in Figure 12.13.

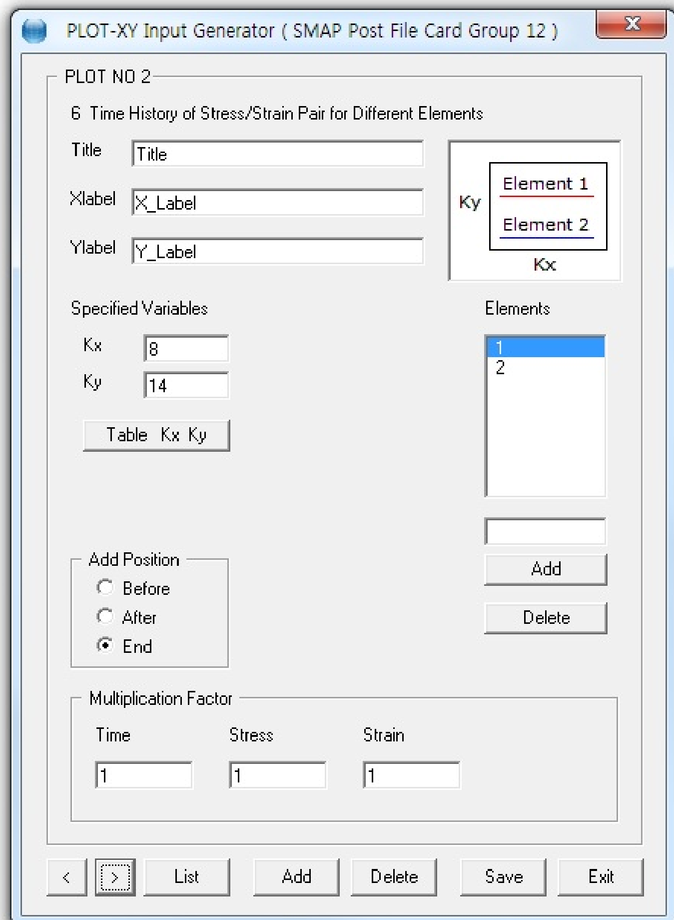


Figure 12.17 Time history for different elements

12.7.4 Time History for a Given Node

Main Dialog for Time History of Displacement / Vel / Accel for a Given Node (IPTYPE = 7) is shown in Figure 12.18.

Node should be listed in Card 10.3.2 in SMAP Main File.
Table shows available data as shown in Figure 12.19.

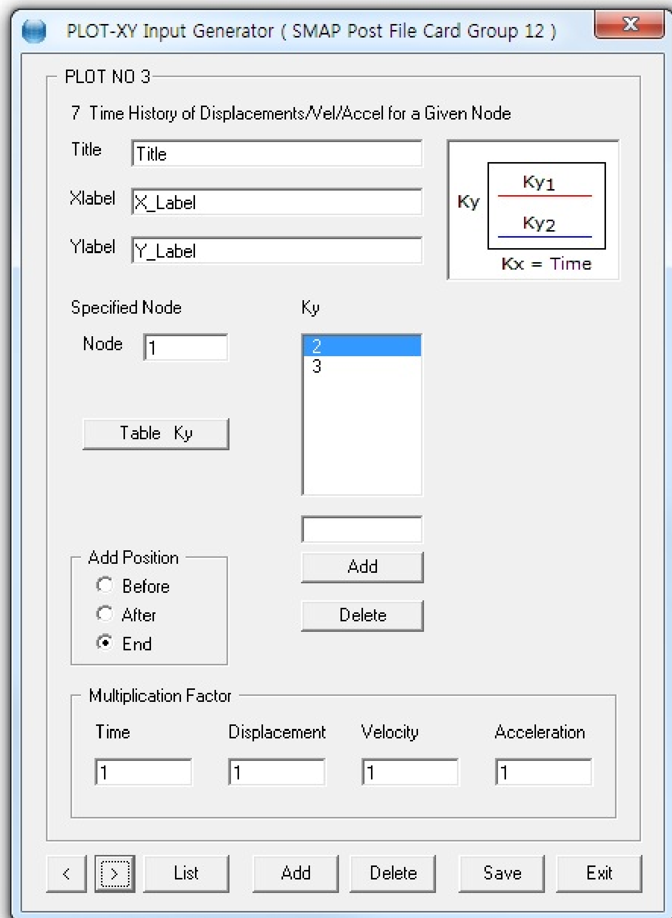


Figure 12.18 Time history for a given node

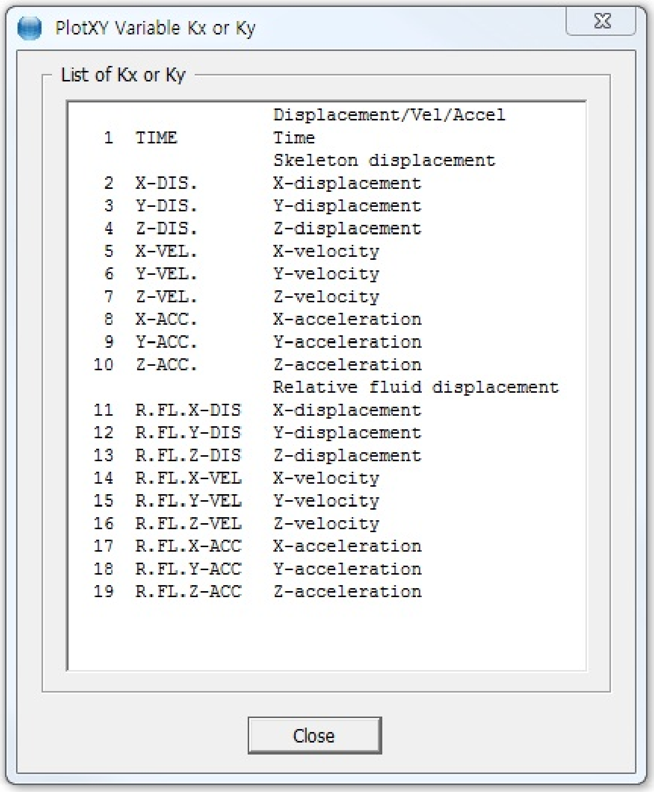


Figure 12.19 Available data for displacement/vel/accel

12.7.5 Time History for Different Nodes

Main Dialog for Time History of Displacement / Vel / Accel for Different Nodes (IPTYPE = 8) is shown in Figure 12.20.

Nodes should be listed in Card 10.3.2 in SMAP Main File.
Table shows available data as in Figure 12.19.

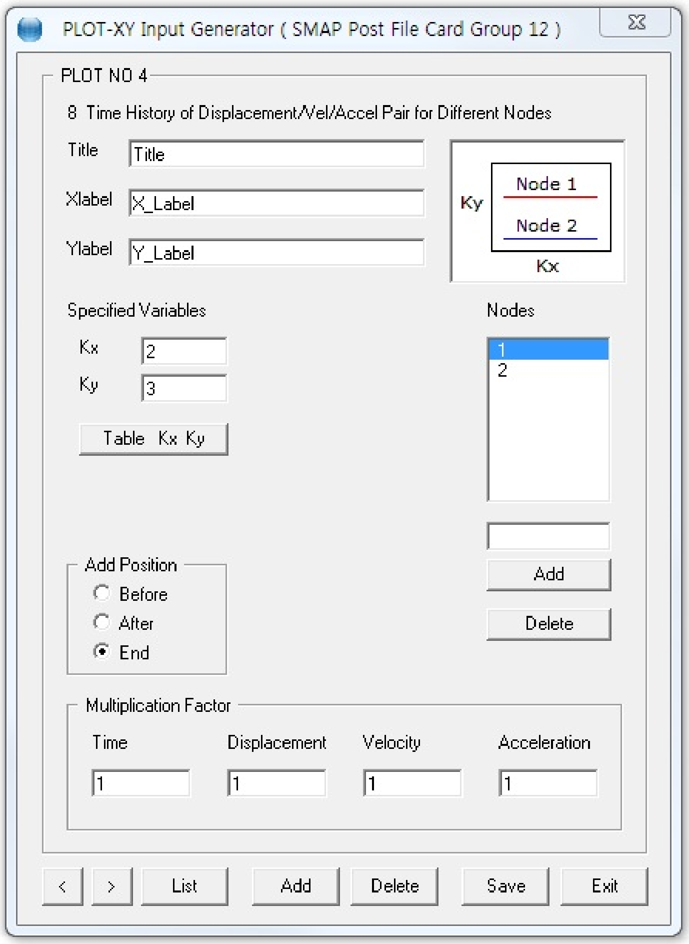


Figure 12.20 Time history for different nodes

12.7.6 Stress/Strain Snapshot for a Given Time

Main Dialog for Snapshot of Stresses / Strains for a Given Time (IPTYPE = 9) is shown in Figure 12.21.

Time should be listed in Card 10.4.2 in SMAP Main File.
Table shows available data as in Figure 12.13.
Elements represent a series of data points in SMAP Mesh.

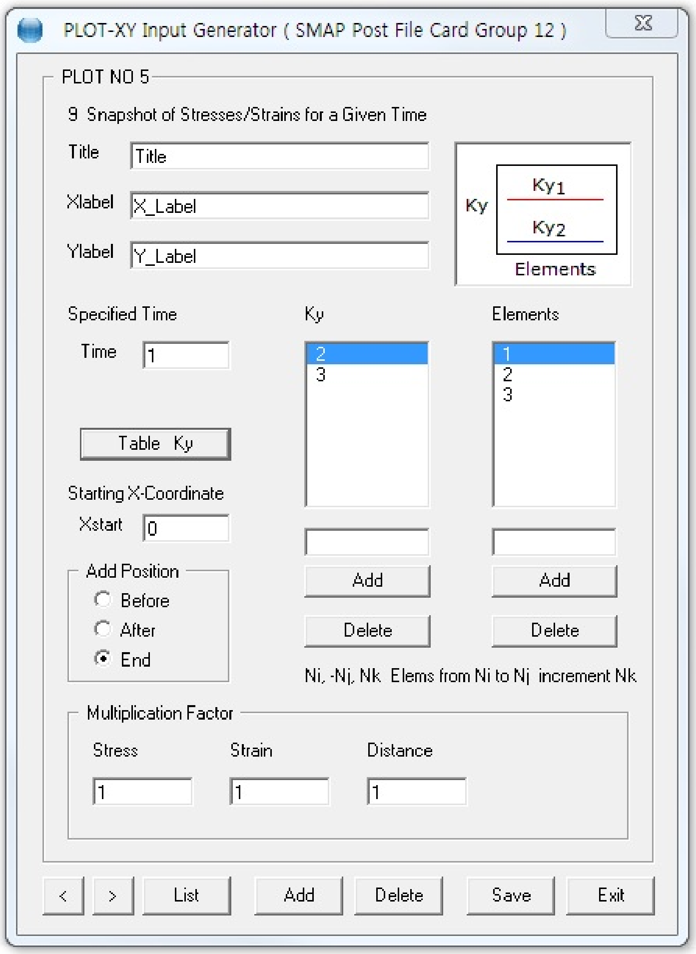


Figure 12.21 Stress/strain snapshot for a given time

12.7.7 Stress/Strain Snapshot for Different Times

Main Dialog for Snapshot of Stresses / Strains for Different Times (IPTYPE = 10) is shown in Figure 12.22.

Times should be listed in Card 10.4.2 in SMAP Main File.
Table shows available data as in Figure 12.13.
Elements represent a series of data points in SMAP Mesh.
This example will select a series of Elements (1,2,3,4,5,6,7,8,9,10).

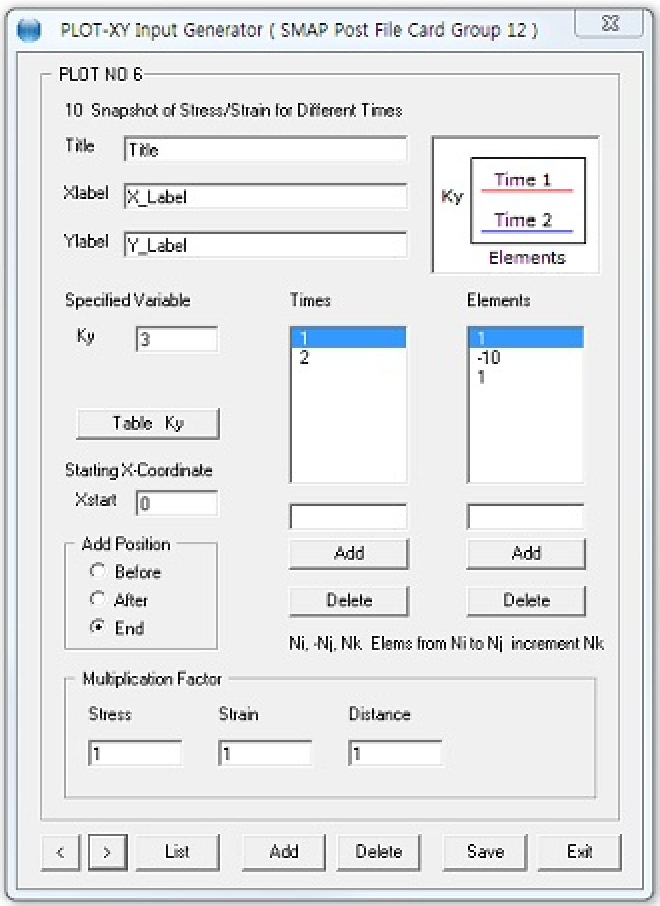


Figure 12.22 Stress/strain snapshot for different times

12.7.8 Displ/Vel/Acc Snapshot for a Given Time

Main Dialog for Snapshot of Displacement / Vel / Accel for a Given Time (IPTYPE = 11) is shown in Figure 12.23.

Time should be listed in Card 10.4.2 in SMAP Main File.
Table shows available data as in Figure 12.19.
Nodes represent a series of data points in SMAP Mesh.

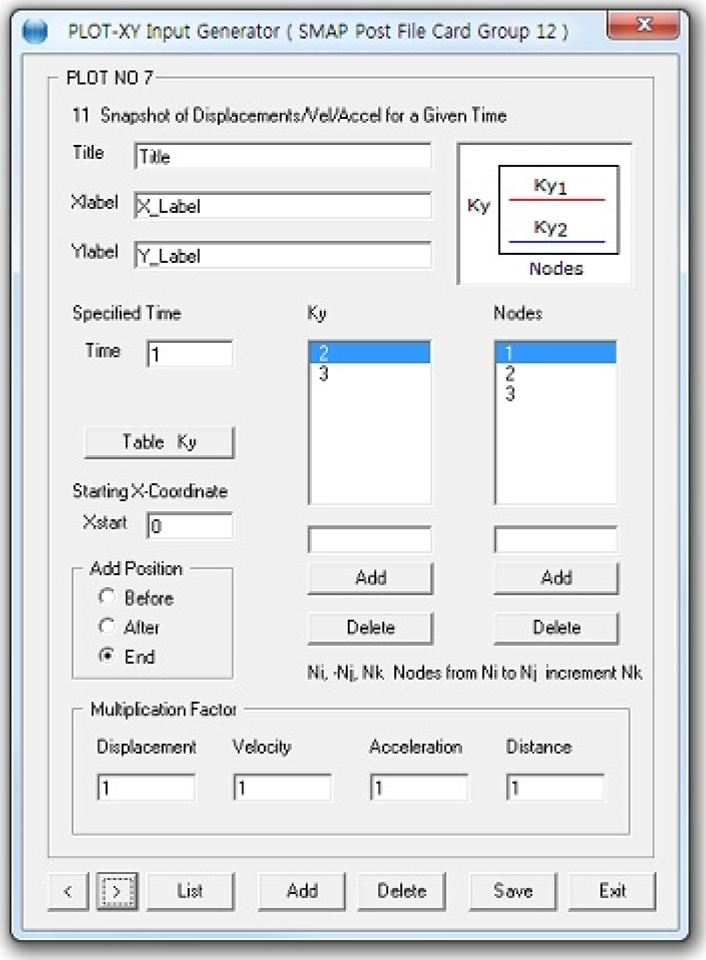


Figure 12.23 Displ/vel/accel snapshot for a given time

12.7.9 Displ/Vel/Acc Snapshot for Different Times

Main Dialog for Snapshot of Displacement / Vel / Accel for Different Times (IPTYPE = 12) is shown in Figure 12.24.

Times should be listed in Card 10.4.2 in SMAP Main File.
Table shows available data as in Figure 12.19.
Nodes represent a series of data points in SMAP Mesh.
This example will select a series of Nodes (1,2,3,11,13,15,17,19,21).

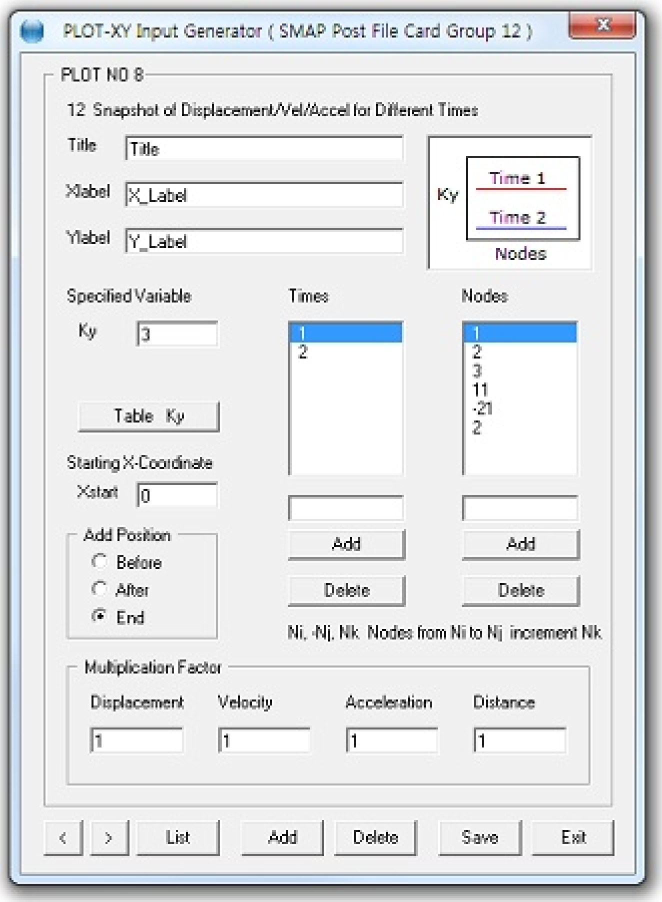


Figure 12.24 Displ/vel/accel snapshot for different times

PLOT-XY
User's Manual

13.1 Introduction

PLOT-XY is a two-dimensional graphical program specially designed to perform scatter plotting and post processing for SMAP programs. The key features of PLOT-XY are:

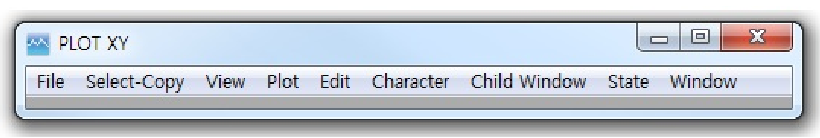
- Plot scatterplot data
  It reads the scatterplot data in text file and plots lines connecting each pair of data points.

- Plot results of analyses
  It reads Card 12 of Post File and SMAP Output and plots time histories of stress/strain/displacement/temperature and snap shots of stress/strain/displacement/temperature vs. distance.

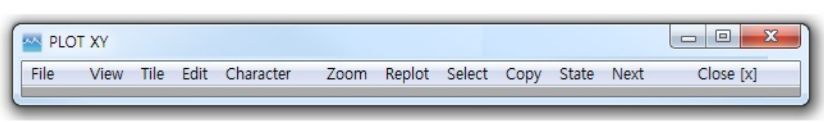
- Edit XY graph
  It reads XY data, edits titles and scales, adds user-defined additional curves.

PLOT-XY has two menu styles, General and Express.

General Style includes 9 menus consisting of all menu items available.
For General Style, specify 1 in C:\Smap\Ct\Ctdata\MenuStyle_XY.dat

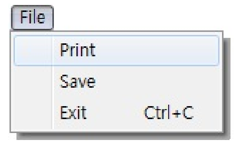


Express Style includes 12 menus which are rearranged so as to quickly access most frequently used menu items in practice.
For Express Style, specify 0 in C:\Smap\Ct\Ctdata\MenuStyle_XY.dat



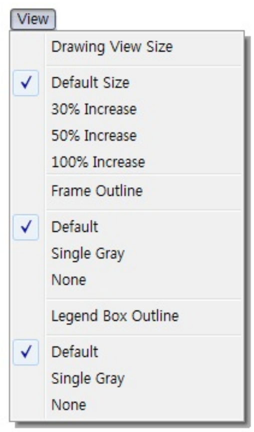
13.2 Menus

File has three sub menus.
Print is to get the hard copy of the current view.
Save is to save the current view.
Exit is to exit PLOT-XY.



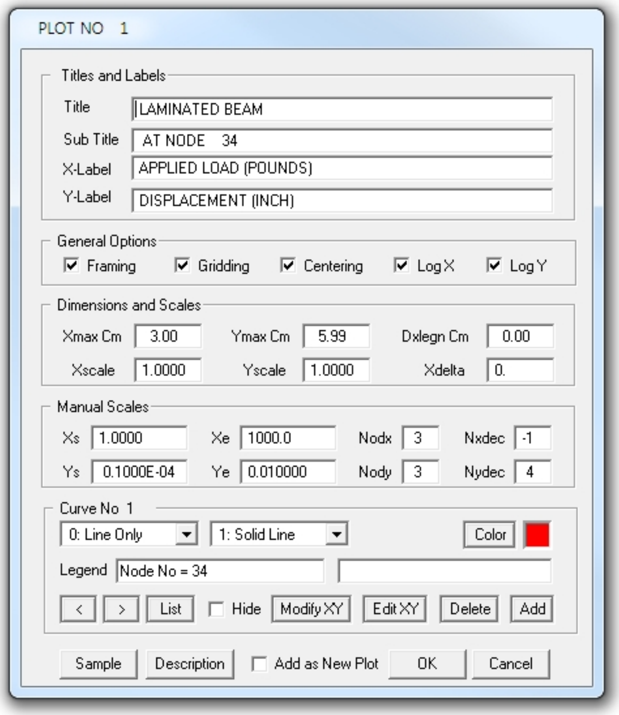
Select-Copy is mainly used to select and then copy the current view.

View is mainly used to select
Drawing View Size:
Default Size, 30%, 50%, or 100%
Increase Frame Outline:
Default, Single Gray, or None
Legend Box Outline:
Default, Single Gray, or None

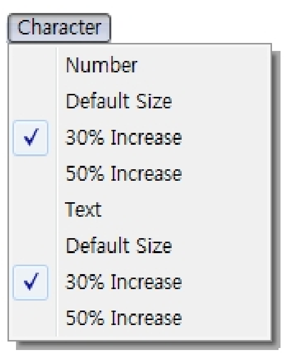


Plot has the following five sub menus.
Replot is to replot the currently focused child window. Zoom is to zoom the currently focused child window. Once this sub menu is selected, you can specify the rectangular zoom area by left mouse button down at the left top corner and then left mouse button up at the right bottom corner.
Hardcopy is to print the currently focused window.
Next is to plot the next graph.
Stop is to stop plotting.

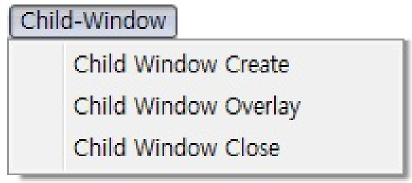
Edit opens following dialog to edit XY graph data.
It is described in detail in Section 12.3 in XY graph User's Manual.



Character is used to change sizes of number and text fonts. Default sizes are specified in PLOT-XY setup menu.



Child-Window is used to create, overlay, or close child window. A maximum of 40 child windows can be opened.



PLOT-2D
User's Manual

14.1 Introduction

PLOT-2D is a two-dimensional graphical program specially designed to perform pre and post processing for SMAP programs.
The key features of PLOT-2D are:

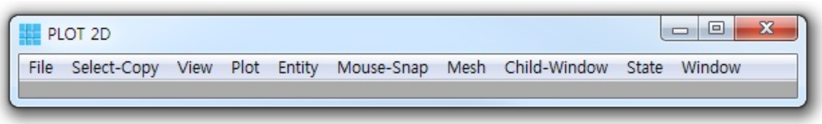
- **Plot finite element meshes**
  It reads the Mesh File and plots meshes along with node, element, boundary code, and material numbers.

- **Plot results of analyses**
  It reads Mesh File, Card 11 of Post File, SMAP Output Files and plots contours of continuum stress/strain/temperature, beam section forces, truss axial force/stress/strain, principal stress vectors, and deformed shapes.

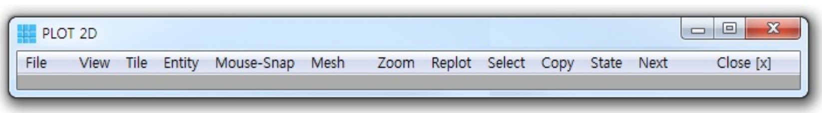
- **Edit finite element or group meshes**
  It reads finite element or group mesh files and edit these meshes.

PLOT-2D has two menu styles, General and Express.

General Style includes 11 menus consisting of all menu items available.
For General Style, specify 1 in C:\Smap\Ct\Ctdata\MenuStyle_2D.dat

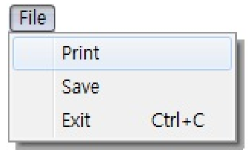


Express Style includes 13 menus which are rearranged so as to quickly access most frequently used menu items in practice.
For Express Style, specify 0 in C:\Smap\Ct\Ctdata\MenuStyle_2D.dat



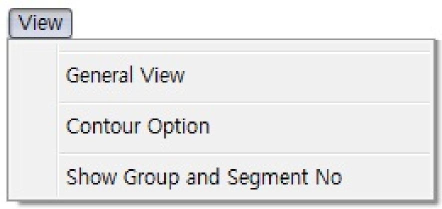
14.2 Menus

File has three sub menus.
Print is to get the hard copy of the current view.
Save is to save the current mesh file. Exit is to exit PLOT-2D.

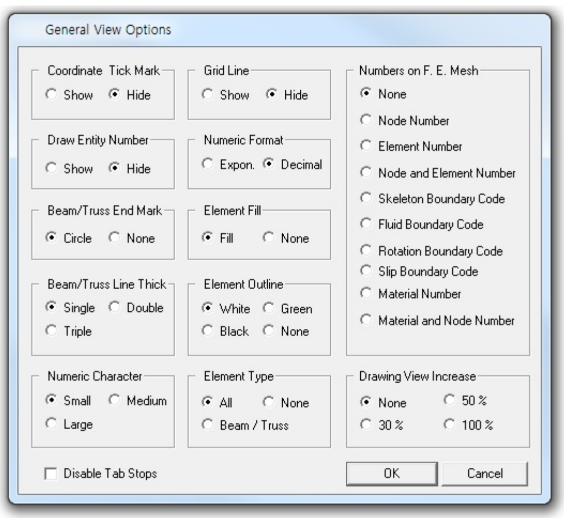


Select-Copy is mainly used to select and then copy the current view.

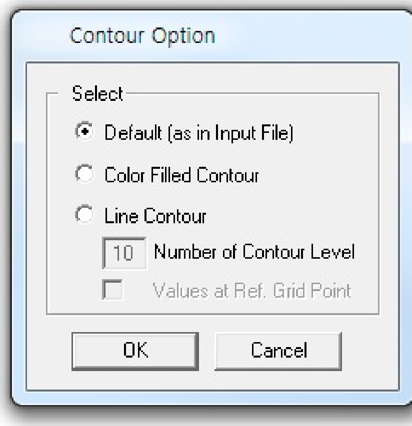
View has three sub menus; General View, Contour Option, and Show Group and Segment No.



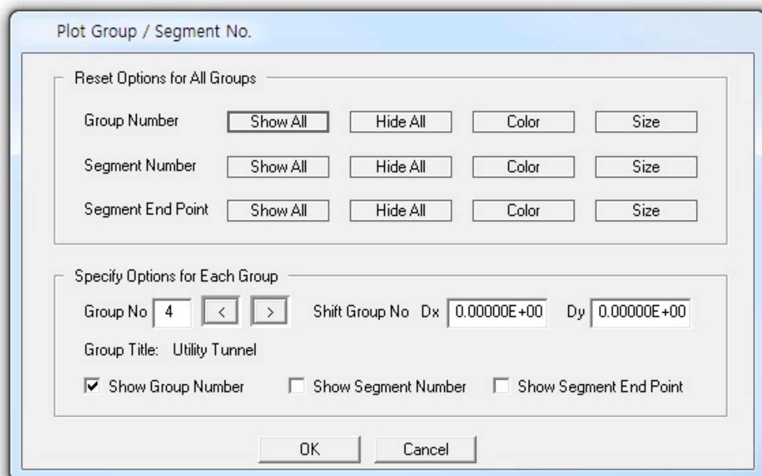
General View options affect all types of plots.



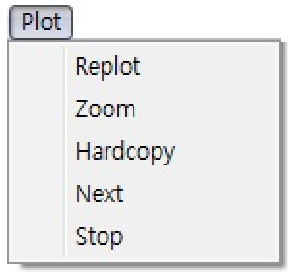
Contour Options affect contour plots of continuum element data for analysis results.



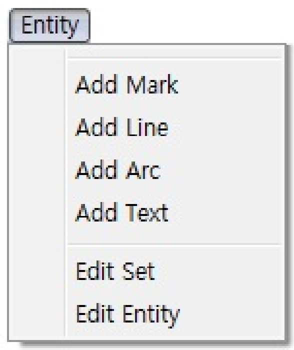
Show Group and Segment No is to show group and segment numbers when editing group meshes. It is described in detail in Section 5.3 in Group Mesh User's Manual.



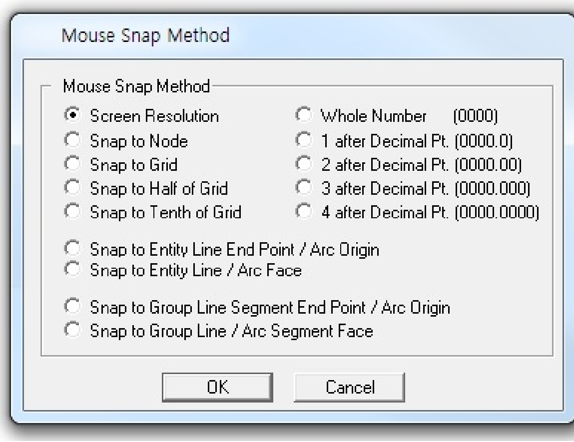
Plot has the following five sub menus.
Replot is to replot the currently focused child window. Zoom is to zoom the currently focused child window.
It zooms only mesh. Once this sub menu is selected, you can specify the rectangular zoom area by left mouse button down at the left top corner and then left mouse button up at the right bottom corner.
Hardcopy is to print the currently focused window.
Next is to plot the next graph.
Stop is to stop plotting.



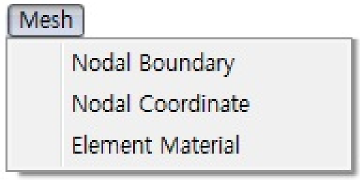
Entity is the graphical object which is mainly used to assist editing geometry of groups and elements. It has following six sub menus; Add Mark, Add Line, Add Arc, Add Text, Edit Set, and Edit Entity. It is described in detail in Section 5.7 in Group Mesh User's Manual.



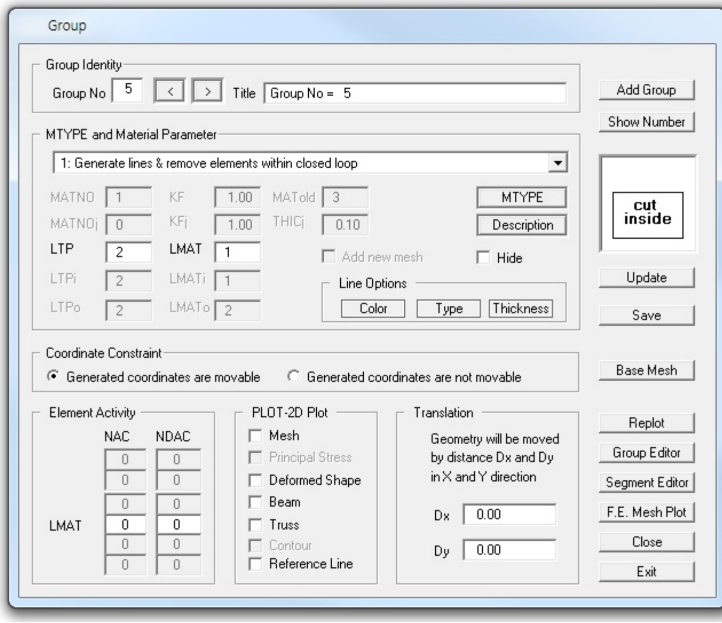
Mouse-Snap is to control the position of mouse cursor when you work for finite element mesh, group mesh, or entities. Mouse Snap Method helps you place the mouse cursor more accurately.



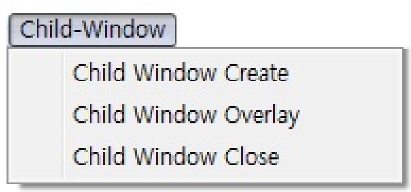
Mesh is used to directly modify finite element meshes. It has three sub menus; Nodal Boundary, Nodal Coordinate, and Element Material.
It is described in detail in Section 5.6 in Group Mesh User's Manual.



Group is used to build or edit group mesh.
It is described in detail in Section 5.3 in Group Mesh User's Manual.



Child-Window is used to create, overlay, or close child window.
A maximum of 40 child windows can be opened.



PLOT-3D
User's Manual

15.1 Introduction

PLOT-3D is a three-dimensional graphical program specially designed to perform pre and post processing for SMAP programs. The key features of PLOT-3D are:

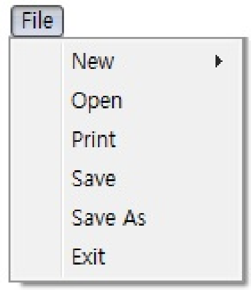
- **Plot finite element meshes**
  It reads the Mesh File and plots meshes along with node, element, boundary code, and material numbers.

- **Plot results of analyses automatically**
  It reads Mesh File and SMAP Output Files and with no input for Post File, plots contours of stress/strain/displacement, iso surface, principal stress vectors, load vectors and deformed shapes.

- **Compute intersections of surfaces**
  It reads the Mesh File containing shell elements for 3D surfaces and shows the locations of the computed intersections.
  The computed coordinates of intersections are saved in a file "Intersection.dat" which can be used for the construction of complicated 3D meshes.

PLOT-3D has 5 menus; File, Model, Plot, View and Help along with 25 toolbars.

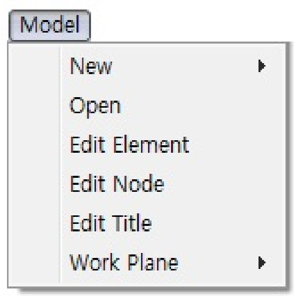
15.2 Menus



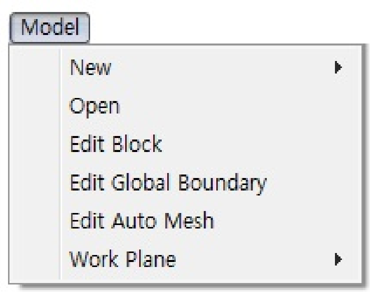
File has six sub menus.
New is used to build Finite Element Mesh or Block Mesh.
Open is used to open existing mesh file.
Print is to get the hard copy of the current view.
Save is to save the current mesh file or current view.

Save As is to save the current mesh file as another name.
Exit is to exit PLOT-3D.

Model is mainly used to edit Finite Element or Block Mesh file.
For detailed description, refer to Block Mesh User's Manual in Section 6.

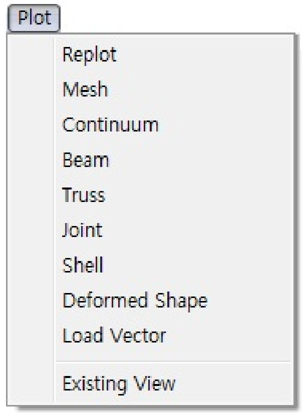


For editing Finite Element Mesh, 6 menus are shown.
New is to build new mesh file.
Open is to open existing mesh file.
Edit Element is to edit parameters related to element.
Edit Node is to edit parameters related to node.
Edit Title is to edit title.
Work Plane is to show prebuilt work planes.



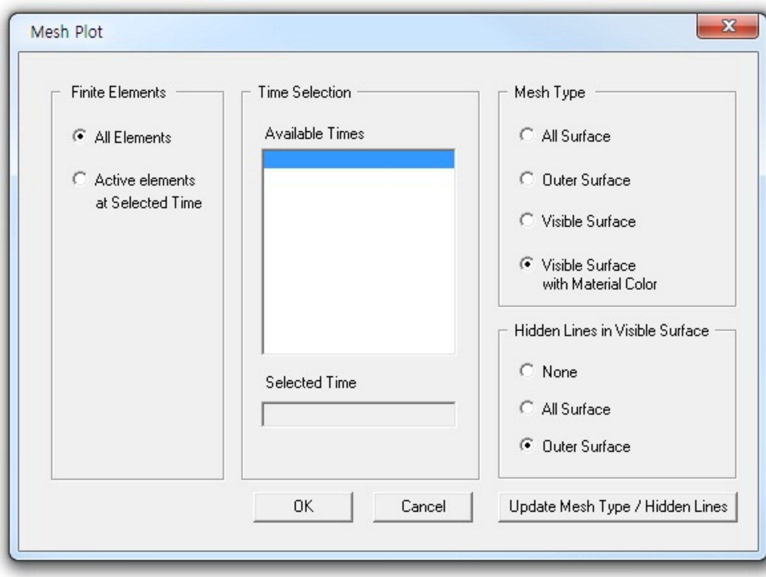
For editing Block Mesh, 6 menus are shown.
New is to build new mesh file.
Open is to open existing mesh file.
Edit Block is to edit parameters related to block.
Edit Global Boundary is to edit parameters related to boundary.
Edit Auto Mesh is to edit parameters related to auto mesh.
Work Plane is to show prebuilt work planes.

Plot is mainly used to plot Finite Element mesh and analysis results.
It has 10 sub menus; Replot, Mesh, Continuum, Beam, Truss, Joint, Shell, Deformed Shape, Load Vector, Existing View.
Joint plot is not available.

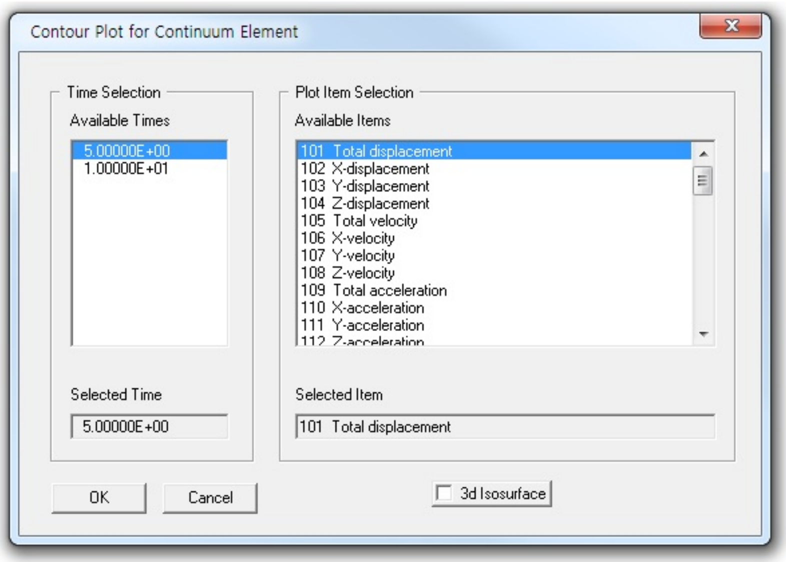


Replot is mainly used to refresh the current view.

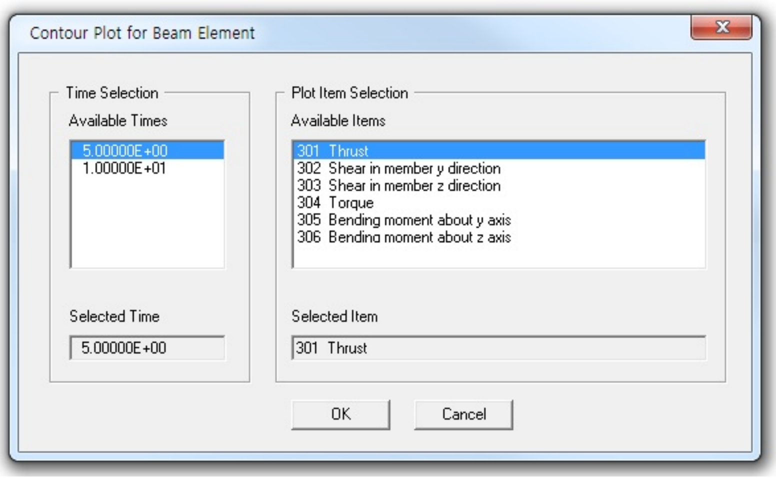
Mesh is to plot Finite Element meshes (Default plot type).
Mesh plot requires only Mesh File.



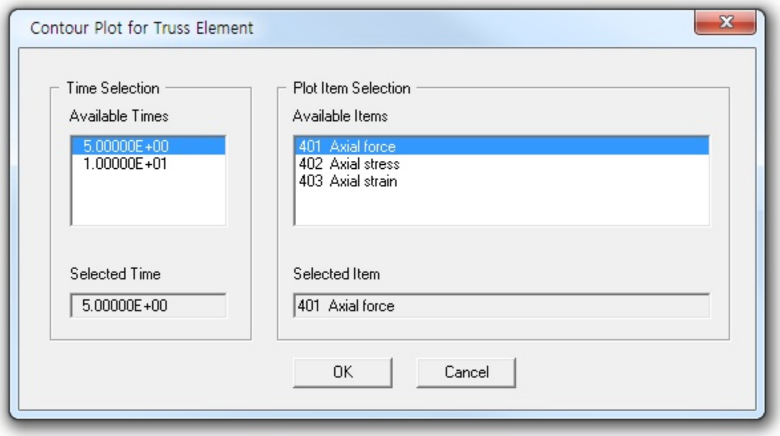
Continuum is to plot contours or principal stress vectors for continuum elements. By checking "3d Isosurface", iso surface will be shown.



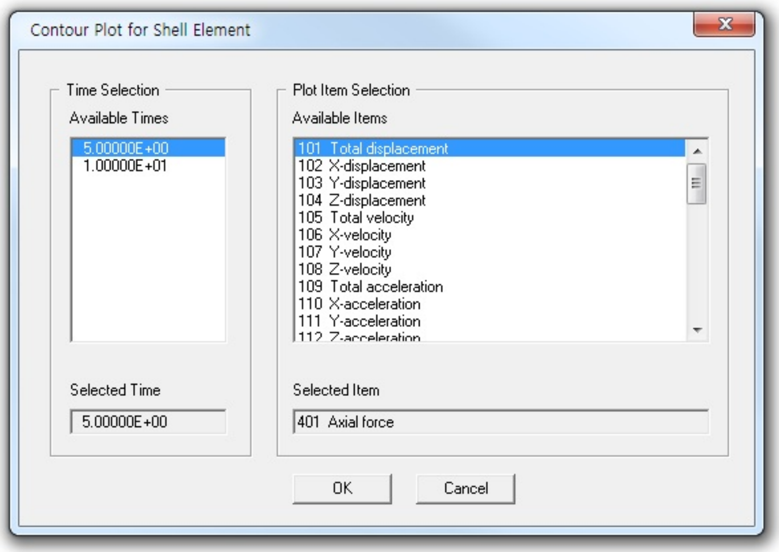
Beam is to plot section forces of beam elements.



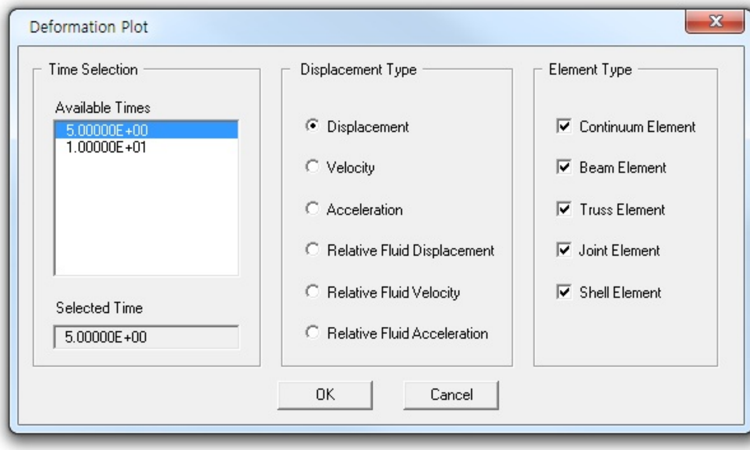
Truss is to plot axial force/stress/strain of truss elements.



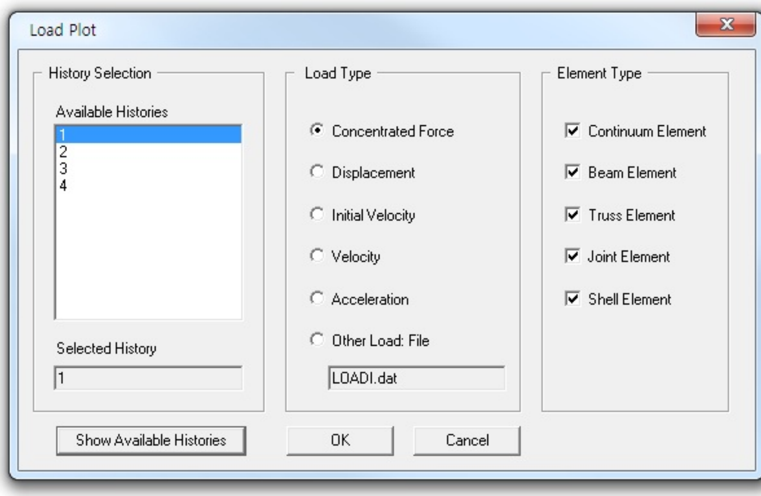
Shell is to plot contours or principal stress vectors for shell elements.



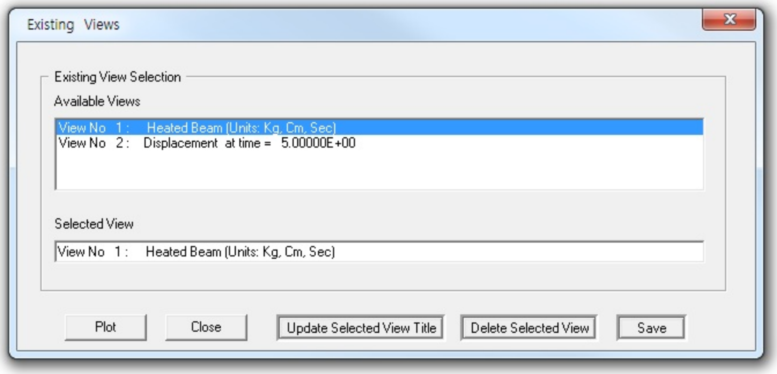
Deformed Shape is to plot the snap shot of all kinds of displacement/velocity/accelerations.
Note that deformed meshes can be combined with other plot types as discussed in "Displacement" option in view menu.



Load Vector is to plot the external loads of concentrated forces/ displacements/velocities/accelerations along with load intensity.
Note that load vectors can be plotted on deformed meshes as discussed in "Load Vector" option in view menu.



Existing View is to replot the saved views.

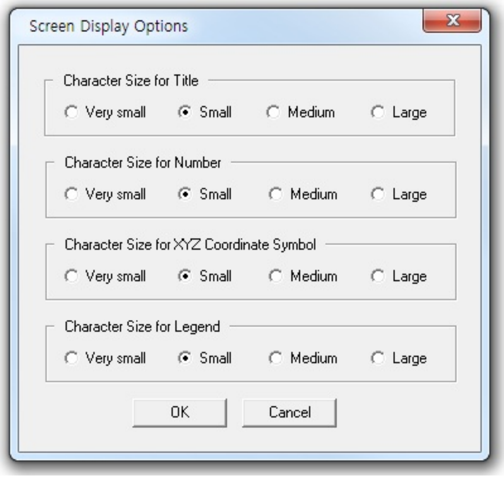


View is used to change the appearance of a selected plot.

It has eleven sub menus; General, Screen, Printer, Contour, Clip Plane, Mesh, Beam, Truss, Principal Stress, Displacement, and Load Vector.

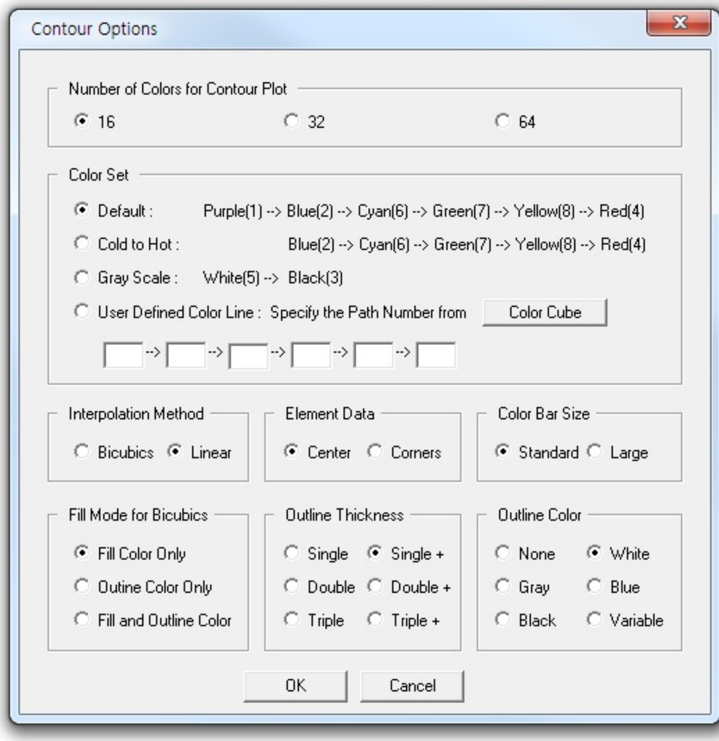
General view options affect most plot types.

Screen display options affect character sizes shown on the monitor.

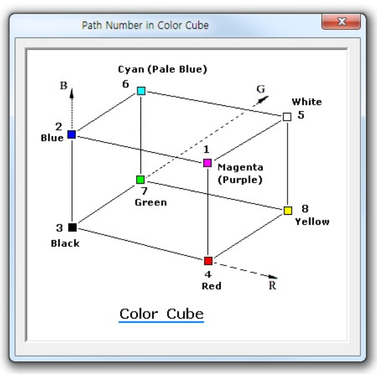


Printer display options affect character sizes and plot dimensions shown on the hard copy.

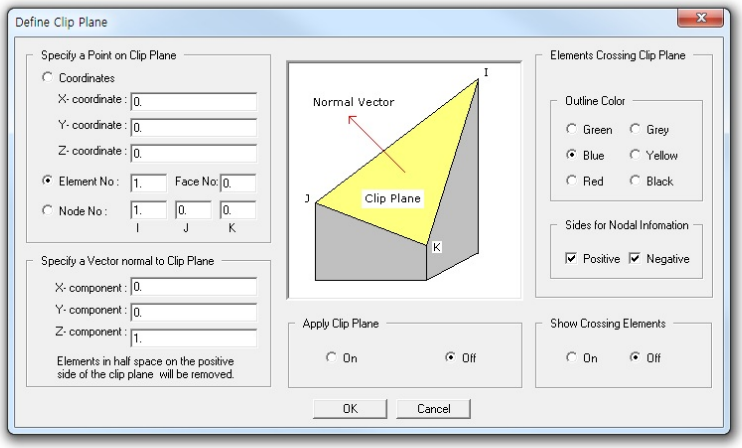
Contour options affect all types of plots involving contours.



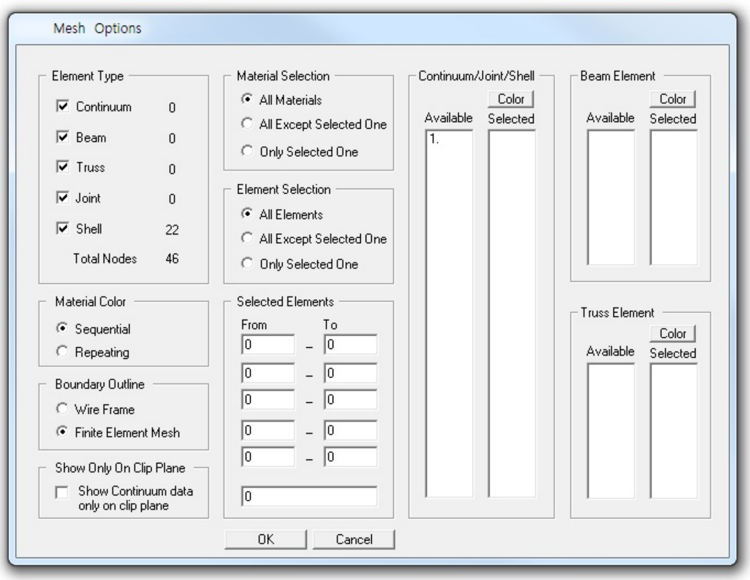
Color cube is to use for user defined color line.



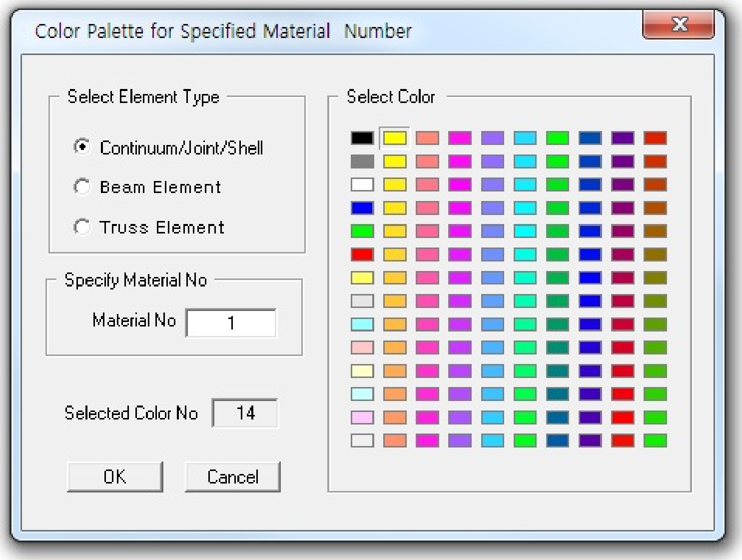
Clip plane defines parameters associated with the clip plane which cuts through the internal part of the 3D domain. When "Apply Clip Plane" is on, contours or deformed shapes are shown on such user defined plane.



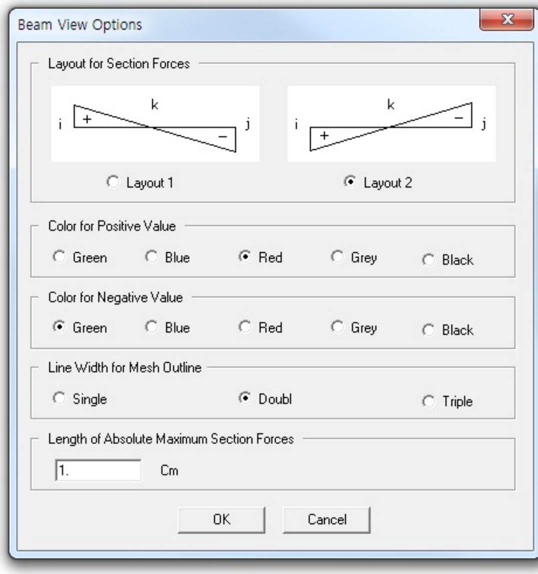
Mesh options affect all plot types. As one of useful features, it can select particular types of elements and materials.



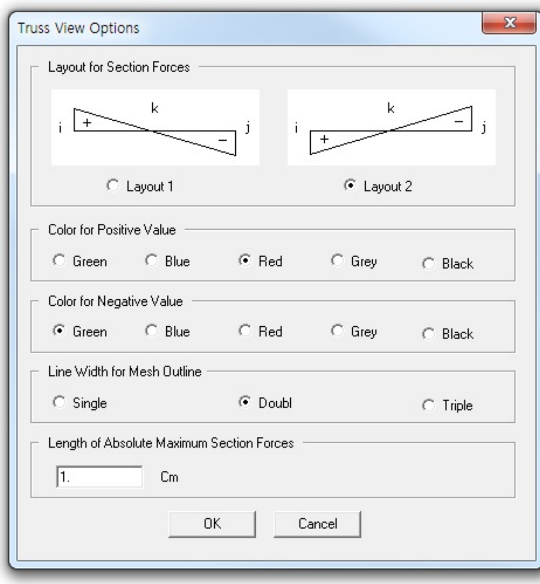
Color is to use for user defined mesh color.



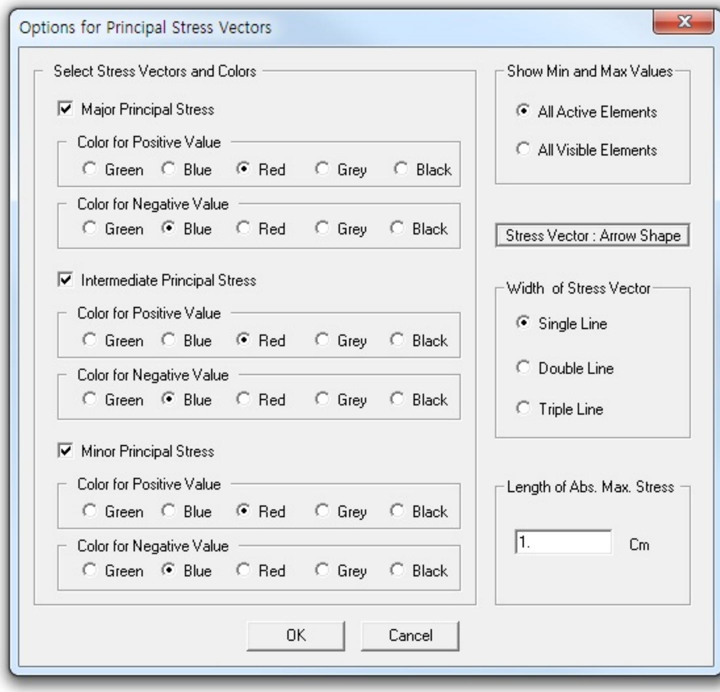
Beam view options affect only beam plot.



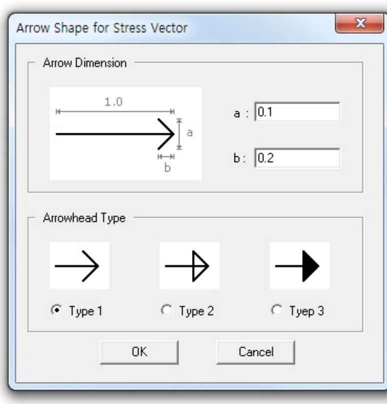
Truss view options affect only truss plot.



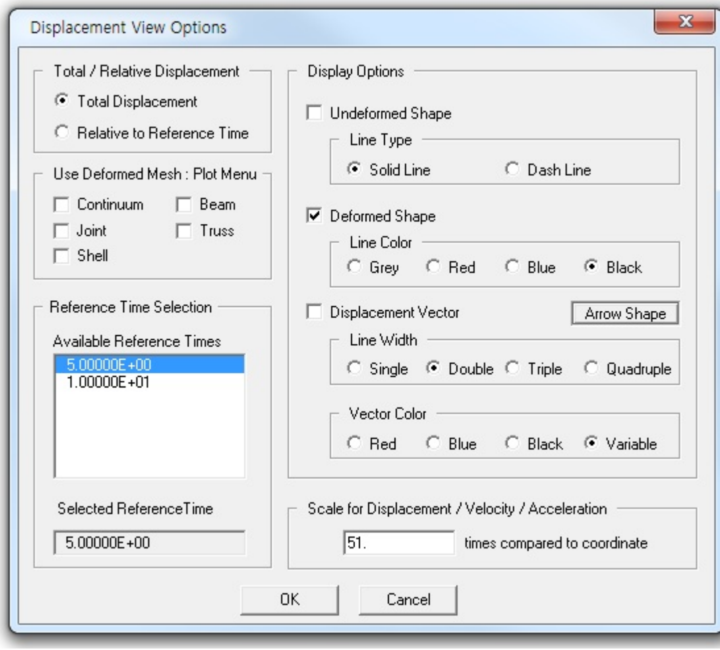
Principal Stress options affect only plots of principal stress vectors in continuum or shell elements.



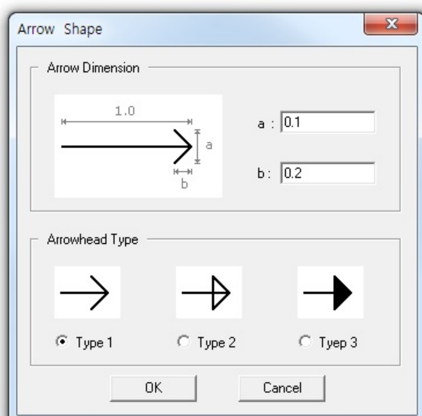
Users can specify the arrow shape for stress vector.



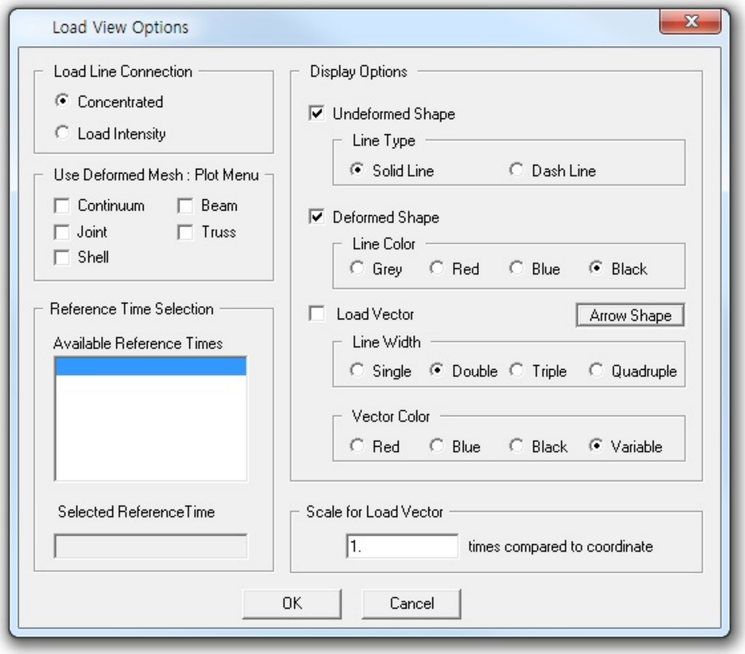
Displacement view options affect only deformed shape plot.
Continuum, Beam, Truss, and Shell plots can be displayed
over deformed mesh by checking types in "Use Deformed Mesh".



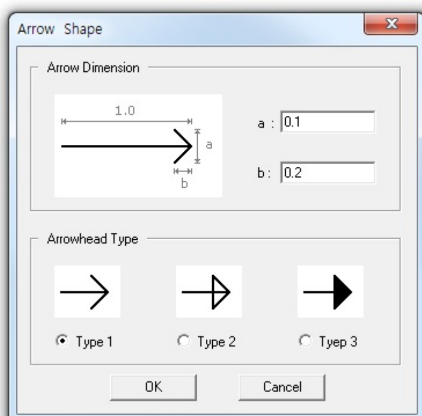
Users can specify the arrow shape
for displacement vector.



Load Vector view options affect only load vector plot.
Load vectors can be displayed over deformed mesh by checking "Deformed Shape" in Display Options



Users can specify the arrow shape for load vector.



15.3 Toolbars

Open Toolbar

This button activates the file open dialog box to open mesh file.

Print Toolbar

This button is used to get the hard copy of current view.

Save Toolbar

This button is used to save current view or working file.

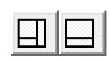
Model Toolbar

This button is used to edit finite element or block mesh.

Work Plane Toolbar

This button is to set work plane used for Model.

Layout Toolbar

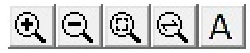


These buttons are used to show different layouts.
The first button divides the plot area into three parts; mesh, title, and legend. The second button divides the plot area into two parts; mesh and title.

XYZ Toolbar

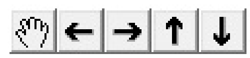
This button is used to locate position of XYZ coordinate symbol in the two part layout mode. Each time you click this button, the XYZ symbol moves counterclockwise along the corners of rectangle. XYZ button is also used to control the amount of movement, rotation, and zoom.

Zoom Toolbar



The first button is used to magnify the mesh.
And the second button is used to reduce the mesh.
The third button is used to activate the selection of zoom area.
Once this button is on, you can specify the rectangular zoom area by left mouse button down at the left top corner and left mouse button up at the right bottom corner. To deactivate, click the button again.
The fourth button is used to switch from the currently zoomed view to the previously zoomed view or vice versa. The last button with "A" is to go back to the initial default configuration.

Translation Toolbar



The first button is to activate drag mode. Once this button is on, you can move the mesh by dragging the mouse. To deactivate, click the button again.
The other buttons move the mesh to the left, right, up, and down, respectively.

Rotation Toolbar

The first button changes direction of rotation.
The other three buttons rotate the mesh about X, Y, and Z axes, respectively.

Number Toolbar

The first button is to activate number mode.
Once this button is on, the selected data will be shown.
Clicking the button again will hide the selected data.
The other two buttons are used to select next and previous number, respectively. The description of selected number is listed at the bottom of PLOT-3D window.

SMAP ® - 3D

Structure Medium Analysis Program

3-D Static, Consolidation and Dynamic
Analysis for Dry, Saturated and
Partially Saturated Soils
and Rock Mass

Example Problems

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Contents

7. PRESMAP Example

Introduction

Example Problems are mainly provided:

- To give you some guide in preparing input data.
- To demonstrate the validity of SMAP programs.

Section 2 describes methods of preparing Mesh Files which represent the geometry of structures to be analyzed.

Section 3 describes two different methods of running main- and post-processing programs.

Section 4 illustrates SMAP-3D main example problems as summarized in Table 1.1. First 9 problems are presented to demonstrate the accuracy and validity of SMAP-3D main- processing program.

Section 5 illustrates Group Mesh examples. Group Mesh Generator is a two dimensional CAD program specially designed to build group mesh which can be used to generate finite element mesh with the aid of program ADDRGN-2D.

Section 6 illustrates Block Mesh examples. Block Mesh Generator is a three dimensional CAD program specially designed to build block mesh which can be used to generate finite element mesh with the aid of program PRESMAP-GP.

Section 7 illustrates PRESMAP examples which are used to generate two and three dimensional Mesh Files.

Section 8 illustrates ADDRGN examples which are used to combine or modify existing Mesh Files. ADDGRN-2D has a powerful mesh generation feature as demonstrated in sub section 8.1.3.

Section 9 illustrates SUPPLEMENT examples which are useful to prepare input data for pre- and main-processing programs.

Section 10 illustrates LOAD examples which are used to generate external nodal loads in two and three dimensional coordinate systems.

Section 11 illustrates XY Graph examples. XY Graph is a two dimensional graph consisting of lines connecting each pair of data points, which can be plotted by PLOT-XY or Excel.

Table 1.1 List of SMAP-3D example problem

Problem Number	Project File Name	Run Time Pent. III 850	Description
1	VP1.dat	0.01 min.	Undrained uniaxial strain compression. Check: • Static • Fully coupled two-phase medium
2	VP2.dat	0.03	Terzaghi's linear consolidation Check: • Consolidation • Gravity load
	VP2-1.dat	0.10	Using linear wedge element
3	VP3.dat	0.37	Planar compression wave propagation Check: • Dynamic two-phase response
	VP3-1.dat	0.13	Using transmitting boundary
4	VP4.dat	0.35	Circular tunnel in Drucker-Prager medium Check: • 3-D elasto-plastic matrix of Generalized Hoek and Brown Model
	VP4-1.dat		Using element surface load
	VP4-2.dat		Using linear wedge element
5	VP5.dat	0.15	Laminated beam with slip interface Check: • Joint element • Joint model
	VP5-1.dat	0.98	Thin layer joint element, NM=4 Joint thickness by CARD 5.3.2.4.11

Table 1.1 List of SMAP-3D example problem, continued

Problem Number	Project File Name	Run Time Pent. III 850	Description
6	VP6.dat	0.02 min.	Gibson's construction pore pressure Check: • Consolidation • Variable time step • Moving boundary
	VP6-1.dat		Using linear wedge element
7	VP7.dat	0.01	Drained triaxial compression test Check: • Modified Cam Clay Model • Drained triaxial compression path
8	VP8.dat	0.01	Undrained plane strain comp. test. Check: • Modified Cam Clay Model • Undrained plane compression path
9	VP9.dat	0.01	Volumetric creep in isotropic undrained test. Check: • Modified Cam Clay Model • Volumetric creep
10	VP10.dat	0.01	Space truss analysis
11	VP11.dat	0.01	Fixed end beam analysis
12	VP12.dat	0.01	Beam dynamic analysis
13	VP13.dat	0.85	William's toggled beam analysis
14	VP14.dat	0.02	Plane strain tunnel analysis
15	VP15.dat	0.01	Hemispherical shell
	VP15-1.dat		Using triangular shell element
16	VP16.dat	0.02	Simply supported plate analysis

Table 1.1 List of SMAP-3D example problem, continued

Problem Number	Project File Name	Run Time Pent. III 850	Description
17	VP17.dat	0.01 min.	Heated beam modeled by shell
	VP17-1.dat		Heated beam modeled by beam
	VP17-2.dat		Heated beam modeled by continuum
18	VP18.dat	0.01	Thin pipe subjected to internal pressure
	VP18-1.dat		Single precision with FACBD = $1x10^6$
19	VP19.dat	24.12	Preload consolidation & excavation
20	VP20.dat	16.93	Seismic tunnel analysis
21	VP21.dat	0.01	Frames with hinge connection Modeled by beam element
	VP21-1.dat		Modeled by shell element
22	VP22.dat		Embedded rebars with slip
23	VP23.dat		Pseudo dynamic embankment fill
24	VP24.dat		Plane strain tunnel in jointed continuum
25	VP25.dat		Spring analysis
26	VP26.dat		Nonlinear truss analysis
27	VP27.dat		SDOF System To Ground Acceleration
28	VP28.dat		Frames with Rotational Spring Connection
29	VP29.dat		Reinforced Concrete Beam
30	VP30.dat		Reinforced Concrete Cylinder
31	VP31.dat		Plate Modal Analysis
32	VP32.dat		Seismic Response Analysis
33	VP33.dat		Silo Lining Analysis
34	VP34.dat		Liquefaction Analysis with PM4Sand

Pre-Processing Programs

Pre-Processing programs are mainly used to generate Mesh File described in Section 4.3 of SMAP-3D User's Manual. The Mesh File represents the geometry of the structure to be analyzed. This file contains information about nodal coordinates, element indexes, material property numbers, and boundary codes. In SMAP-3D, you may generate such Mesh Files using the following methods:

Method 1

First, generate 2D Mesh File representing a typical two dimensional section using Group Mesh Generator, Block Mesh Generator, or 2D PRESMAP. Modify this 2D Mesh File using ADDRGN-2D if you need to do it. And then extend the 2D mesh into 3D mesh using GEN-3D.

1. Generate 2D Mesh File

GROUP MESH GENERATOR	
BLOCK MESH GENERATOR	
PRESMAP-2D	NATM-2D
CIRCLE-2D	PRESMAP-GP

2. Modify 2D Mesh File

ADDRGN-2D

3. Extend into 3D Mesh File

GEN-3D

Method 2

Generate 3D Mesh Files using Block Mesh Generator or 3D PRESMAP. Then combine or modify these 3D Mesh Files using ADDRGN-3D if you need to do it.

1. Generate 3D Mesh File

BLOCK MESH GENERATOR	
PRESMAP-3D	CROSS-3D
PRESMAP-GP	

2. Combine or modify 3D Mesh File

ADDRGN-3D

Above two methods can be combined to make a final 3D Mesh File representing the structure to be analyzed.

To view the Mesh Files, you can use PLOT-3D by selecting following order:
Plot → Mesh → F. E. Mesh → Open

Boundary codes can affect analysis result significantly so that it is strongly recommended for you to double check those codes to avoid solving wrong problems.

Main- and Post-Processing Programs

Main-Processing program reads Mesh and Main Files as input and performs static, consolidation, or dynamic analysis. Post-Processing programs read Post File along with analysis results from Main-Processing program and then produce graphical output.

Mesh Files can be generated using Pre-Processing programs as outlined in the previous Section 2. Main and Post Files can be created according to Section 4.4 and 4.5, respectively, in SMAP-3D User's Manual. Normally, they can copy existing Main or Post Files which are similar to the problem to be analyzed and modify those files using Text Editor.

Main- and Post-Processing programs can be executed using the following methods:

Method 1

Prepare Mesh, Main, and Post Files. Run EXECUTE menu to get analysis results. And run PLOT menu to view graphical output of analysis results.

1. Prepare All Input Files

   Mesh, Main and Post Files

2. Get Analysis Results

   RUN → SMAP → EXECUTE

3. View Graphical Output

   PLOT → RESULT → PLOT-XY, PLOT-2D, PLOT-3D

Method 2

Prepare Mesh, Main, and Blank Post Files. Run EXECUTE menu to get analysis results. Now, prepare Post File according to Section 4.5 in SMAP-3D User's Manual. Run PRE EXECUTE menu to obtain intermediate plotting information files. And then run PLOT menu to view graphical output of analysis results. Note that Blank Post File consists of following 3 lines:

```
┌  0,  1,  2
│  0
└  0,  4.5
```

1. Prepare Mesh and Main Files

Mesh, Main and Blank Post Files

2. Get Analysis Results

RUN → SMAP → EXECUTE Menu

3. Prepare Post File

Post File in Section 4.5 of User's Manual

4. Get Plotting Information Files

RUN → SMAP → PreEXECUTE

5. View Graphical Output

PLOT → RESULT → PLOT-XY, PLOT-2D, PLOT-3D

Method 2 is particularly useful when you are running large problems which take long execution time. You have to care in preparing Card Group 10 in Main File since Post File can only address those data requested in Card Group 10. You can repeat Steps 3 and 4 as long as your Post File addresses the output data within the range specified in Card Group 10 in Main File.

Post-Processing programs are mainly used to show graphical output of the analysis results.

PLOT-XY reads Card Group 12 in Post File and plots time histories of stresses, strains, and displacements. Once you run PLOT-XY , you will obtain intermediate plotting information file (PLOTXY.Lin). PLOTXY.Lin file can be modified as it will be described in Section 11 of SMAP Examples.

PLOT-2D reads Card Group 11 in Post File and plots two dimensional snap shots. Once you run PLOT-2D in PLOT menu, you will obtain intermediate plotting information file (PLOT2D.DAT).

PLOT-3D does not need any Post File.
This program plots following three dimensional snapshots:

- Finite element mesh
- Deformed shape
- Principal stress distribution
- Section forces in beam element
- Extreme fiber stresses/strains in beam elements (2D)
- Axial force/stress/strain in truss element
- Contours of stresses, strains and factor of safety
- 3D iso surface of stresses and strains

SMAP-3D
Example Problem

SMAP-3D is the main-processing program which computes static, consolidation and dynamic response of three-dimensional problems. Input parameters of SMAP-3D are described in detail in Section 4 of SMAP-3D User's Manual.

Running SMAP-3D is described in Section 3.2.1 of User's Manual and can be selected in the following order:

RUN → SMAP → EXECUTE

Manual procedure to run SMAP-3D is outlined in Section 3.5 of User's Manual. Once you finished execution of SMAP-3D, you can obtain graphical outputs by selecting:

PLOT → RESULT → PLOT-XY, PLOT-2D, or PLOT-3D

PLOT Menu is described in Section 3.3 of SMAP-3D User's Manual.

Table 1.1 in Section 1 shows the summary of SMAP-3D example problems. First nine example problems are the verification problems. The main objective of these verification problems is to demonstrate the accuracy and validity of SMAP-3D.

You can access all input files of example problems in the directory:

C:\Smap\Smap3D\Example\Smap

For each example problem, brief problem descriptions and partial graphical outputs will be presented in this section.

4.1 Undrained Uniaxial Strain Compression

The problem concerns fully coupled undrained uniaxial strain response of saturated porous linear elastic medium as shown in Figure 4.1.

Finite element mesh in Figure 4.2 is generated by Block Mesh Generator as explained in detail in Section 6.1 in SMAP-3D Example Problem.

The exact solution for the undrained stress response is given by Blouin and Kim, 1984.

$$\pi_o = \sigma_v \frac{1}{1 + \beta_m} \tag{4.1}$$

$$\beta_m = \frac{K_g^2 M_s + K_m K_s^2 - M_s K_m K_s - K_g K_m K_s}{K_m K_g (K_g - K_s)} \tag{4.2}$$

Where

σ_v	Applied total vertical stress
π_o	Pore water pressure
K_s	Bulk modulus of skeleton
G_s	Shear modulus of skeleton
M_s	Constrained modulus of skeleton ($M_s = K_s + 4G_s /3$)
n	Porosity
K_g	Bulk modulus of grain
K_w	Bulk modulus of water
K_m	Mixture modulus $K_m = K_g \; K_w / \{K_w + n [K_g - K_w]\}$

The following material properties are used for computing undrained uniaxial strain response:

K_g = 3.5210×10^6 t/m^2
K_w = 0.2042×10^6 t/m^2
E = 0.7042×10^6 t/m^2
ν = 0
n = 0.3
G_s = 2.674

K_s = 0.2347×10^6 t/m^2
G_s = 0.3521×10^6 t/m^2

The exact ratio of pore water pressure (π_o) to applied total vertical stress (σ_v) is obtained from equations 4.1 and 4.2

$$\pi_o / \sigma_v = 0.4592$$

and the exact ratio of effective vertical stress (σ_v') to applied total vertical stress (σ_v) is given by

$$\sigma'_v / \sigma_v = 0.5408$$

Figure 4.3 shows predicted undrained uniaxial stress response compared with an exact solution. As shown in Figure 4.3, the predicted response by program SMAP-3D is identical to the exact solution.

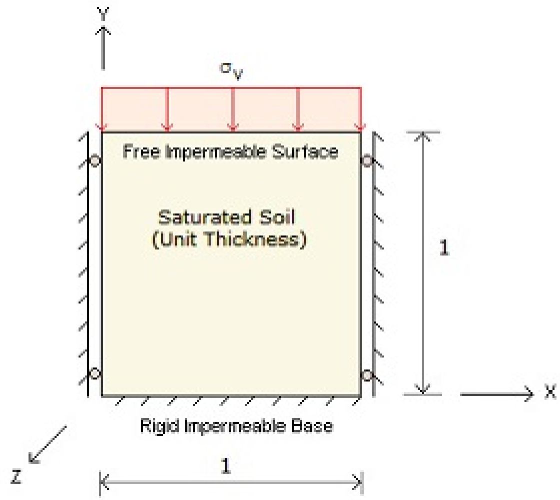


Figure 4.1 A cubic element subjected to undrained uniaxial strain loading

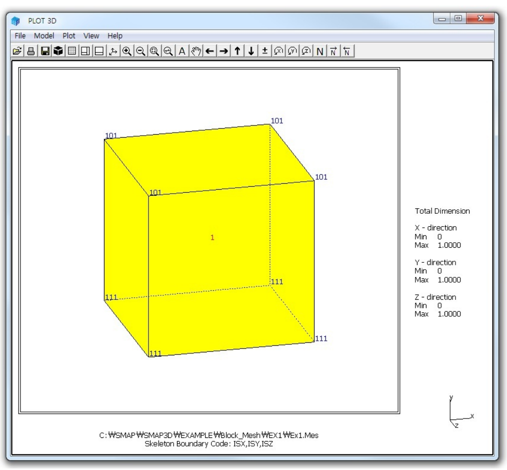


Figure 4.2 Finite element mesh

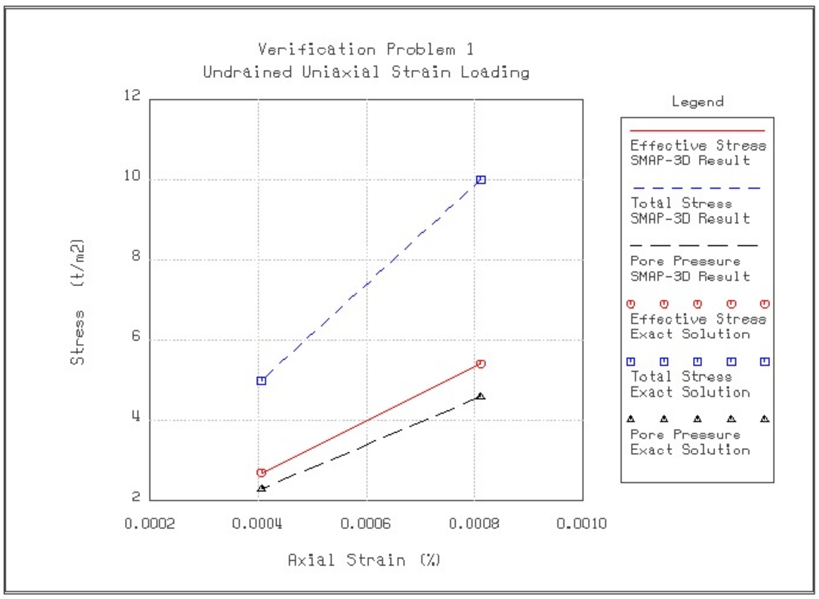


Figure 4.3 Computed undrained stress response compared with exact solution

4.2 Terzaghi's Linear Consolidation

The problem concerns Terzaghi's linear consolidation with initial triangular distribution of excess pore water pressures. As initial conditions, it is assumed that soil is liquefied and pore water takes all the weight. The exact solution for the excess pore water pressure (π_e) is given by

$$\pi_e = \sum_{m=1,3}^{\infty} \left(\frac{8\ \gamma'\ H}{m^2\ \pi^2} \right) \left(\mathrm{Sin}\ \frac{m\ \pi}{2} \right) \left(\mathrm{Sin}\ \frac{m\ \pi}{2\ H}\ y \right) e^{-\frac{m^2\ \pi^2}{4}\ T} \tag{4.3}$$

where

- H Thickness of soil deposit. Top is free surface, bottom is rigid impermeable base.
- y Distance from the free surface.
- $\gamma' = \gamma - \gamma_w$
  - γ is the total unit weight and
  - γ_w is the unit weight of pore water.

And the time factor (T) is given by

$$T = \frac{k\ M\ t}{\gamma_w\ H^2}$$

where

- t Time
- k Coefficient of permeability
- M Constrained modulus

To simulate numerically, following material parameters are assumed:

n = 0.3 Porosity
G_s = 2.7 Specific gravity of grain
γ_w = 1.0 t/m^3
γ = $\gamma_w\ (G_s\ (1-n) + n)$ = 2.19 t/m^3
γ' = 1.19 t/m^3

$E = 1,000\ t/m^2$
$\nu = 0.3$
$M = (1-\nu)\ E\ /\ ((1+\nu)(1-2\nu)) = 1,346\ t/m^2$
$k = 0.001$ m/day
$H = 10$ m

Figure 4.4 shows finite element mesh consisting of 20 elements used for this example problem.

Figure 4.5 shows profiles of pore water pressures at T = 0.05 and 0.5. And Figure 4.6 shows profiles of effective vertical stresses at T = 0.05 and 0.5. SMAP-3D calculations are very close to the exact solution.

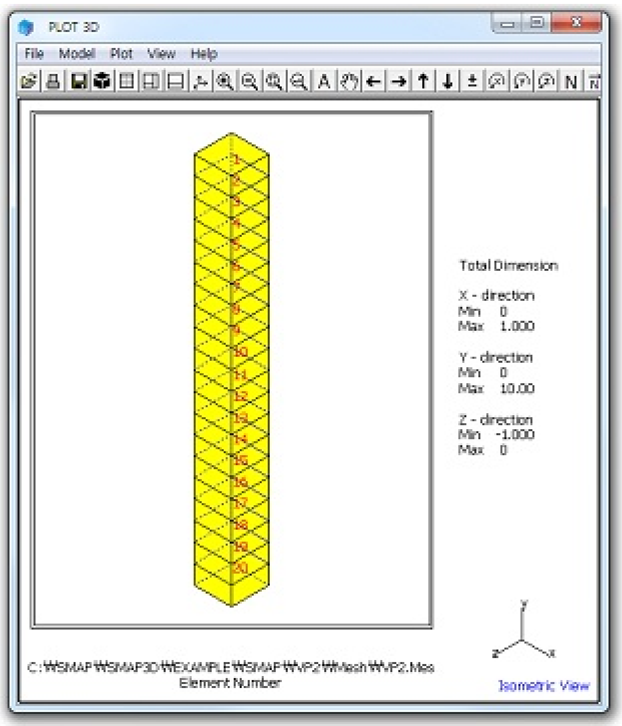


Figure 4.4 Finite element mesh

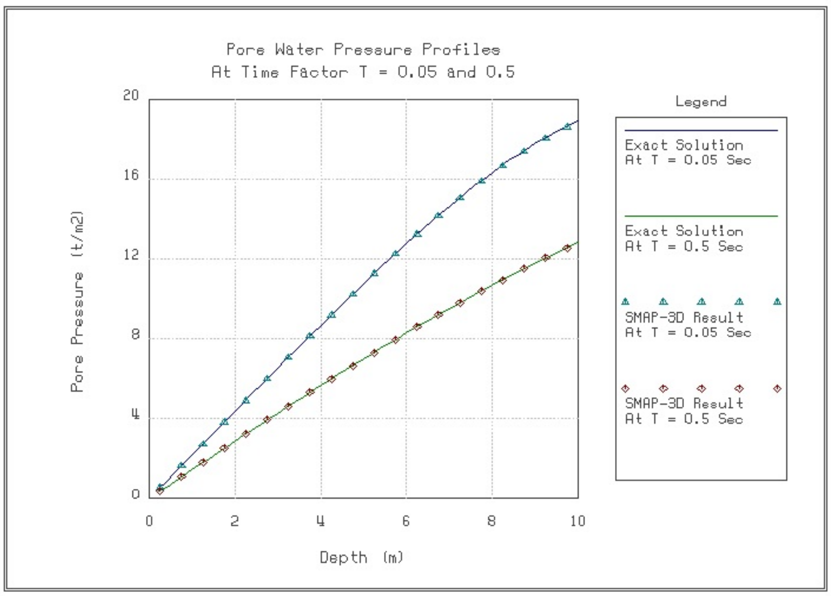


Figure 4.5 Pore water pressure profiles at T = 0.05 and 0.5

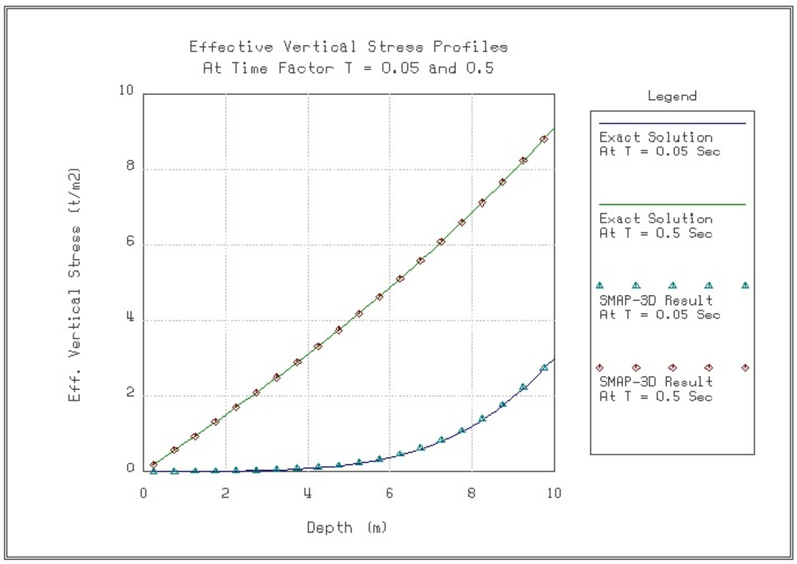


Figure 4.6 Effective vertical stress profiles at T = 0.05 and 0.5

4.3 Planar Compression Wave Propagation

The problem is to check overall two-phase dynamic equations implemented in the program SMAP-3D. A vertically propagating planar compression wave through idealized saturated soil is considered. The input loading, as shown in Figure 4.8, is a short rise time triangular pulse with a peak stress of 3,521 t/m^2 and a positive phase duration of 10 msec. The loading pulse is applied to the saturated sand having the properties listed in Figure 4.8. The load is applied to an impermeable boundary at the ground surface.

Figure 4.7 shows finite element mesh consisting of 200 elements.

Computed profiles of pore water pressure and effective vertical stress at 20 msec are shown in Figures 4.9 and 4.10, respectively. The closed-form solution for this problem is not available. So, the same problem has been solved by the existing two-dimensional version of TPDAP-II for direct comparison. These TPDAP-II results are not shown in Figures 4.9 and 4.10, but they are identical to the SMAP-3D results.

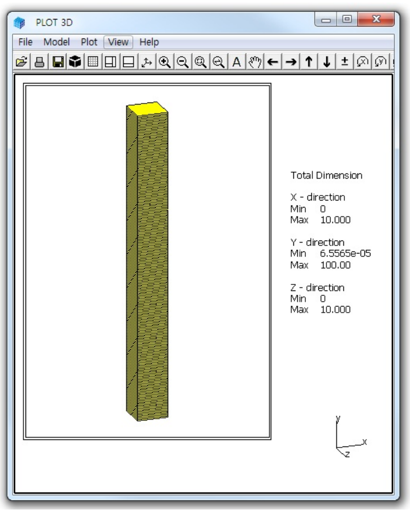
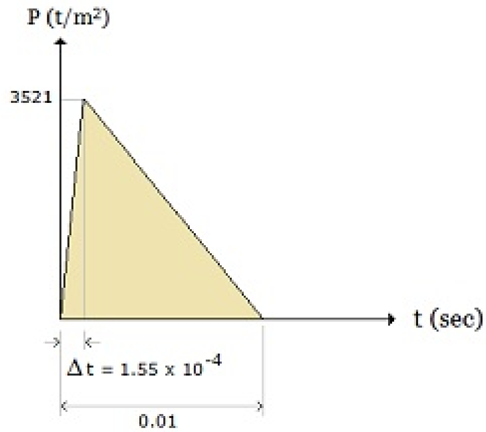


Figure 4.7 Finite element mesh



Assumed Material Properties

Pore Water	
Bulk Modulus	0.2042×10^{6} t/m^{2}
Solid Grains	
Bulk Modulus	3.521×10^{6} t/m^{2}
Specific Gravity	2.67
Drained Skeleton Properties	
Bulk Modulus	2113 t/m^{2}
Constrained Modulus	4225 t/m^{2}
Poisson's Ratio	0.20
Porosity	0.35
Permeability	2.54×10^{-5} m/s

Figure 4.8 Loading time history and material properties used in planar compression wave propagation through saturated soil

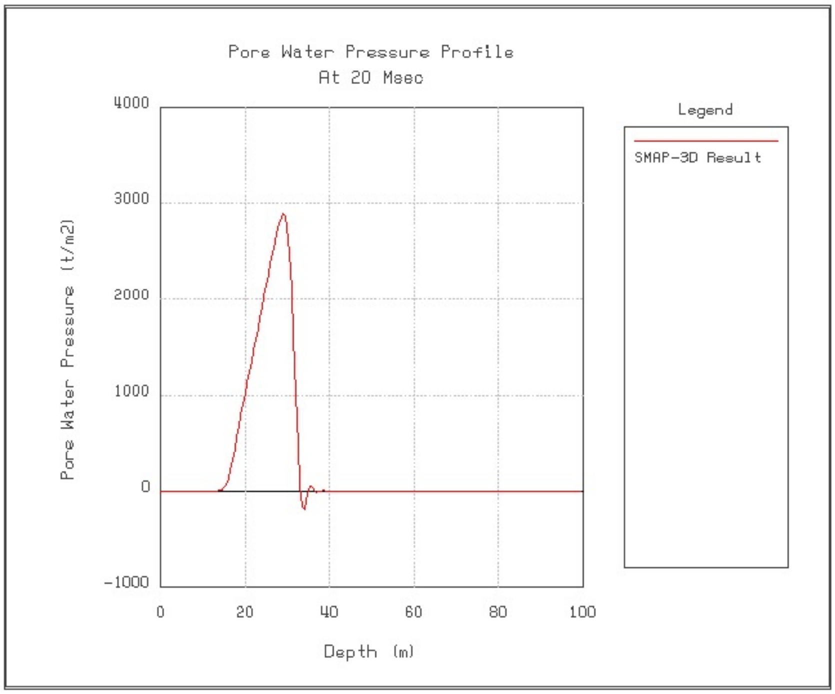


Figure 4.9 Profiles of pore water pressure at 20 msec

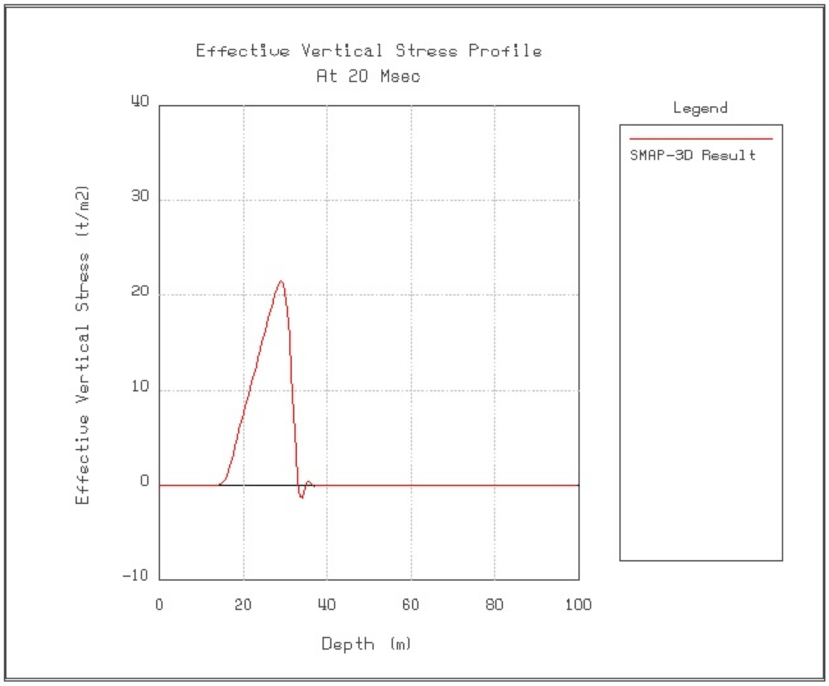


Figure 4.10 Profiles of effective vertical stress at 20 msec

4.4 Circular Tunnel in Drucker-Prager Medium

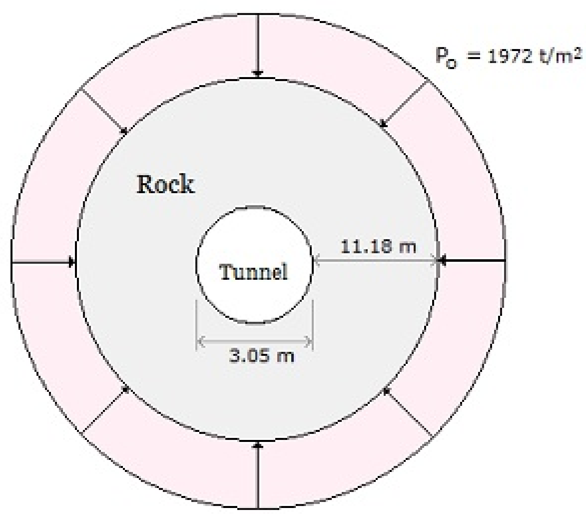
The problem is to check the implementation of the 3-dimensional formulation of elasto-plastic matrix derived for the Generalized Hoek and Brown Model. In this problem, the plane strain response of a tunnel subjected to axisymmetric loading as calculated using SMAP-3D is compared to a semi-analytical solution developed by Piepenburg, Kim and Davister (1986).

Figure 4.11 shows a schematic section view of 3.05m (10 feet) diameter circular tunnel subjected to a hydrostatic loading of 1972 t/m^2 (2800 psi). The surrounding rock is assumed to be linear elastic beneath the failure surface and to follow the Drucker-Prager plasticity model upon reaching the failure surface. The elastic and strength properties of the rock are listed in Figure 4.11.

By symmetry, only a quadrant of tunnel cross section is modeled as shown in Figure 4.12. Along the axis of tunnel (in z-direction), three elements (sections) are used so that the internal section can have unconstrained full 3 degrees of freedom per each node.
This is to check the uniform response of the integrated three dimensional grids though problem is essentially one dimensional axisymmetric.

Figure 4.13 shows tunnel displacement contour. Figure 4.14 shows stresses along the 4.5° from the X-axis in Section 2. And Figure 4.15 shows stresses along the 85.5° from the X-axis in Section 2. As we see, both deformations and stresses are uniform along the tunnel tangential direction. The computed tunnel radial displacement (0.896 Cm) is very close to the semi-analytical solution (0.89 Cm). The computed stress profiles agree well with the semi-analytical solution in both the plastic and elastic zones of deformation surrounding tunnel.

It should be noted that the stresses plotted in Figures 4.14 and 4.15 are in X, Y and Z coordinates so that for exact comparison, these stresses should have transformed to radial and tangential coordinate system.



Material Model: Drucker-Prager Model

Rock Properties:

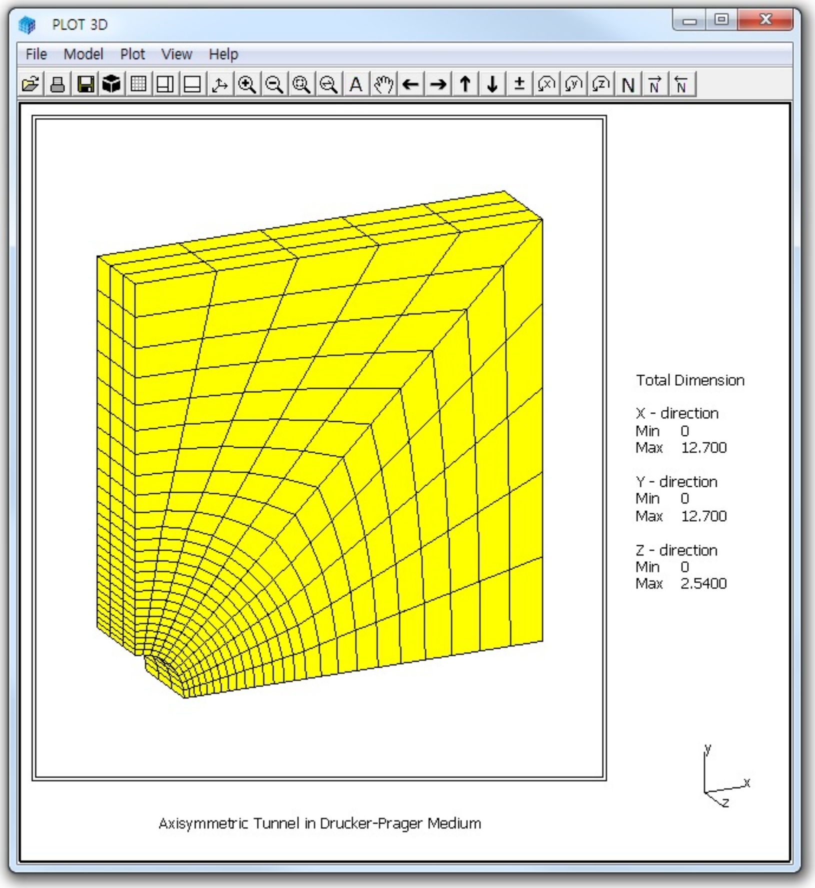
E = 810,000 t/m^2 Young's Modulus

ν = 0.33 Poisson's Ratio

σ_c = 1,268 t/m^2 (1800 psi) Unconfined Strength

ϕ = 18° Friction Angle

Figure 4.11 Circular tunnel subjected to axisymmetric loading



4.12 Finite element mesh

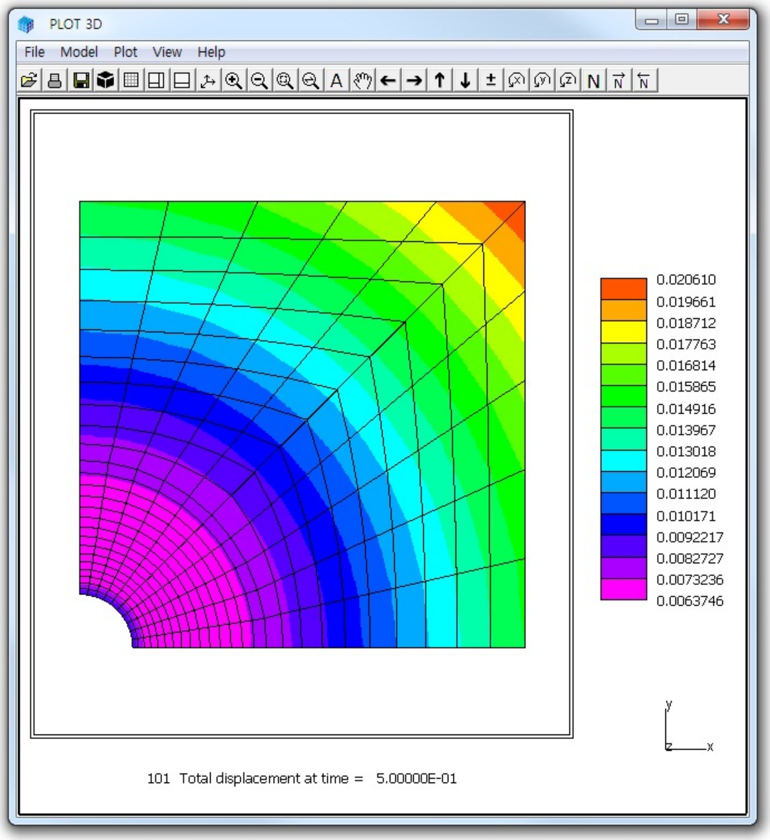


Figure 4.13 Tunnel displacement contour

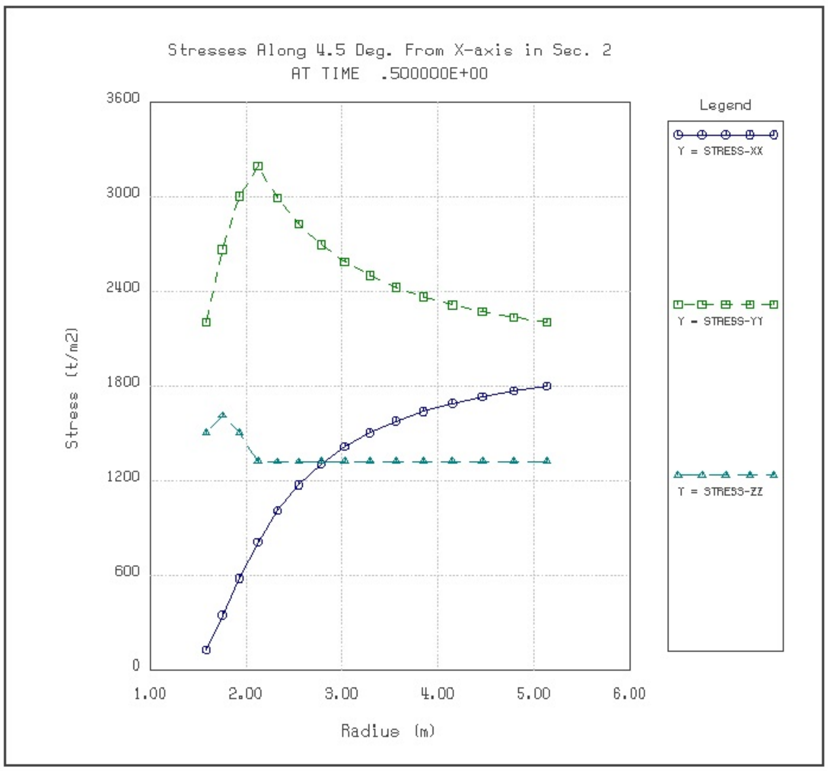


Figure 4.14 Stresses along 4 degree from X-axis in Section 2

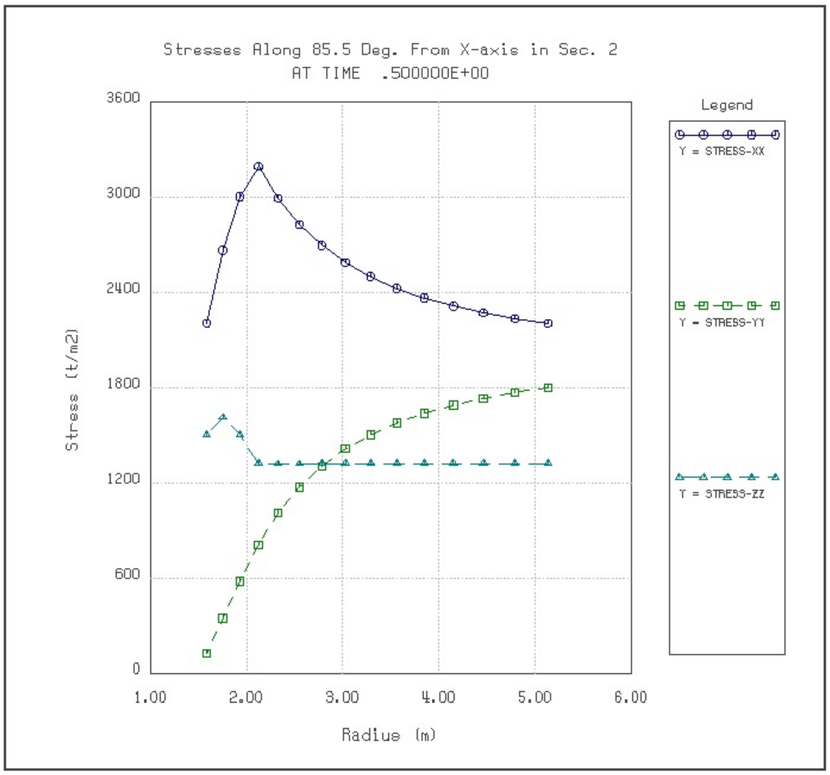


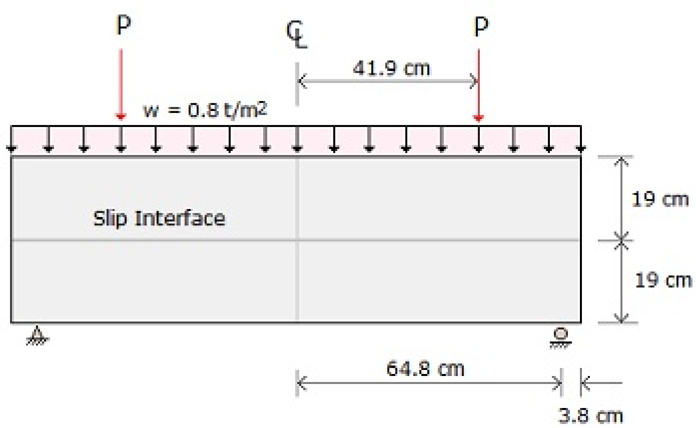
Figure 4.15 Stresses along 85.5 degree from X-axis in Section 2

4.5 Laminated Beam with Slip Interface

The problem is to check the joint element and the nonlinear joint model described in Section 3.6 in theory. Figure 4.16 shows the schematic view of a laminated simply supported beam subjected to uniform and concentrated transverse loads along with the material properties of the beam and the interface.

By symmetry, only the right half of the beam is modeled by 60 continuum elements and 10 joint elements as shown in Figures 4.17 and 18. Element numbers from 61 to 70 are joint elements which represent the slip interface. Joint face is designated along the line from nodes 4 to 144. Thus, nodal coordinates along the other side of joint face are used mainly for visual presentation of joint elements. That is, program SMAP-3D resets internally the nodal coordinates of nodes from 157 to 176 equal to the nodal coordinates of the joint face (nodes from 4 to 144). Then joint thickness (t=0.00254 cm) is specified through the material properties of the joint model.

In Figure 4.19, the midspan deflections by SMAP-3D are compared to the closed-form solution derived from beam theory (Agbabian Associates, 1981). Overall, SMAP-3D results show good agreement with the closed-form solution, especially when the sliding occurs along the interface. It should be noted that there are some differences between the beam and continuum theories, to which slight overestimation by SMAP-3D may be attributed.



Beam Properties	E	=	2.635×10^6 t/m^2
	ν	=	0.1
Interface Properties			
	C	=	4.93 t/m^2
	ϕ	=	0
	t	=	0.00254 cm
Transverse Loads	P	=	0.03 to 17 ton
	w	=	0.8 t/m^2

Figure 4.16 Laminated beam subjected to uniform and concentrated transverse loads

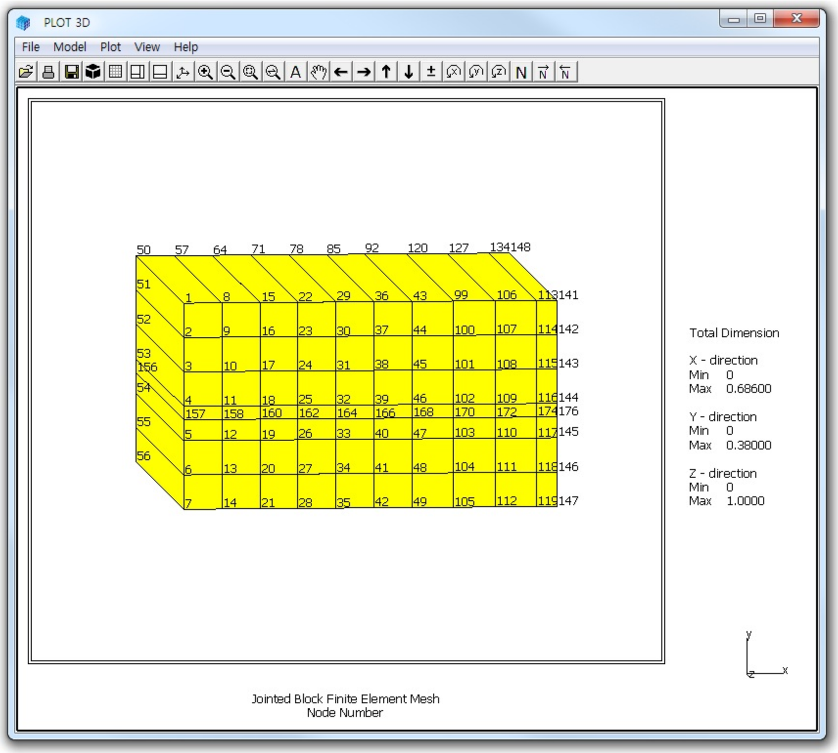


Figure 4.17 Node numbers of laminated beam

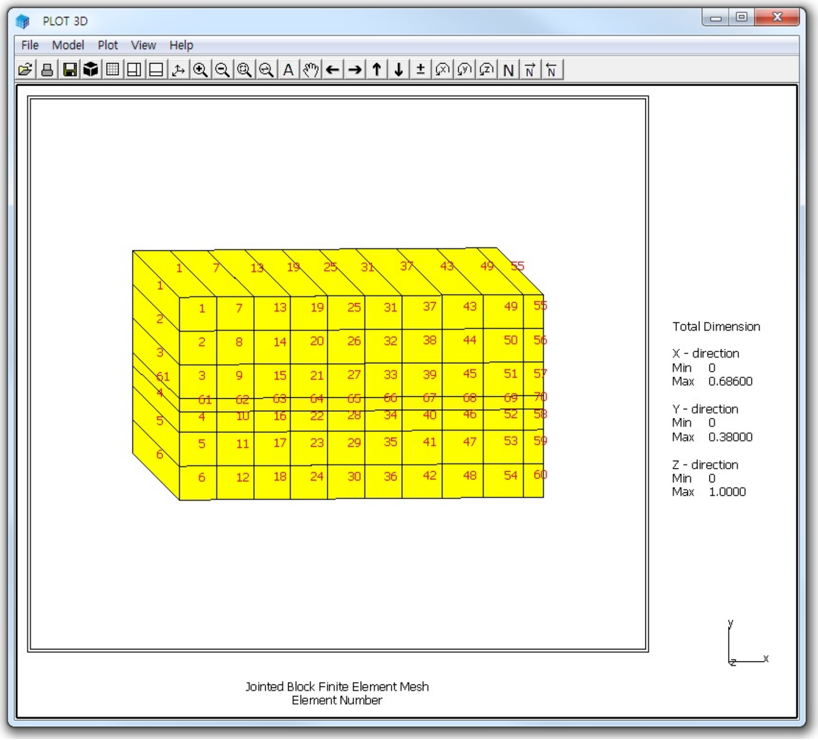


Figure 4.18 Element numbers of laminated beam

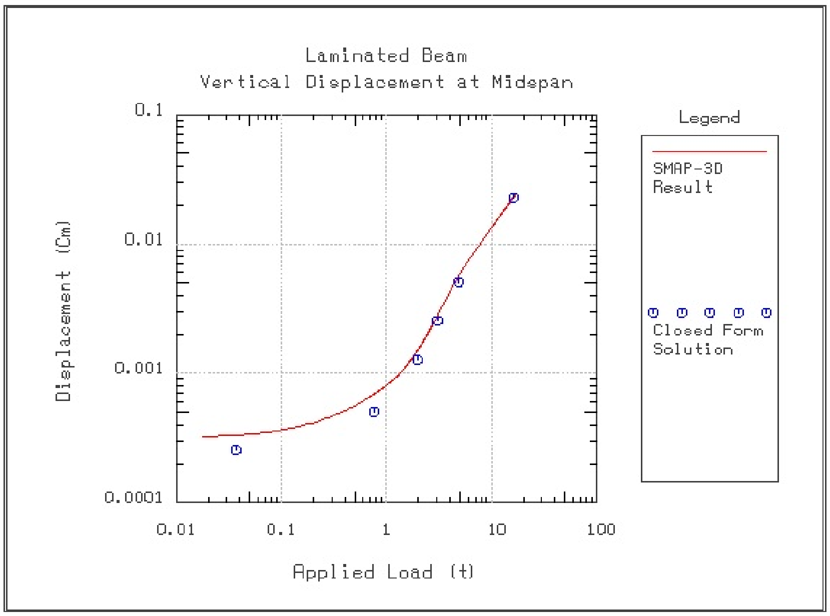


Figure 4.19 Vertical displacement at midspan

4.6 Gibson's Construction Pore Pressure

The problem is to check variable time steps and moving boundary during the construction of the fully saturated fill at constant rate. This problem, as schematically outlined in Figure 4.21, has been analytically solved by Gibson (1958).

$$\pi_e = \gamma' \, m t - \gamma' \, (\pi \, C_v \, t)^{-1/2} \cdot$$

$$\exp\frac{-x^2}{4C_v t} \int_0^\infty \left(\xi \tanh\frac{m\xi}{2C_v} \cosh\frac{x\xi}{2C_v t} \exp - \frac{\xi^2}{4C_v t} \right) d\xi \qquad (4.4)$$

π_e	Excess pore pressure
C_v	Coefficient of consolidation
t	Time

All other parameters in Equation 4.4 are described in Figure 4.21.

The saturated fill has been modeled using 36 equally spaced laterally confined 3D continuum elements as shown in Figure 4.20.

Computation is performed until the height of fill reaches to 18 meters at time t = 60 days. Each time when new element is placed, dissipation of fill is followed according to the variable time steps listed in Table 4.1.

Table 4.1 Variable time steps applied for each lift

Sequence	$\Delta t/(\Delta h/m)$
Beginning	0.001
	0.106
	0.106
Intermediate	0.160
	0.160
	0.234
End	0.234

where Δt is time step and Δh thickness of current top layer.

Following input parameters are used to compute profiles of pore pressure.

$E = 1000\ t/m^2$
$\nu = 0.3$
$G_s = 2.7$
$\gamma_w = 1.0\ t/m^3$
$n = 0.6$
$k = 0.001\ m/day$
$h = 18\ m$
$t = 60.03\ days$

$T = 4$
$m = 0.3\ m/day$
$M_s = 1346.15\ t/m^2$
$C_v = 1.3462\ m^2/day$
$\gamma' = 0.68\ t/m^3$

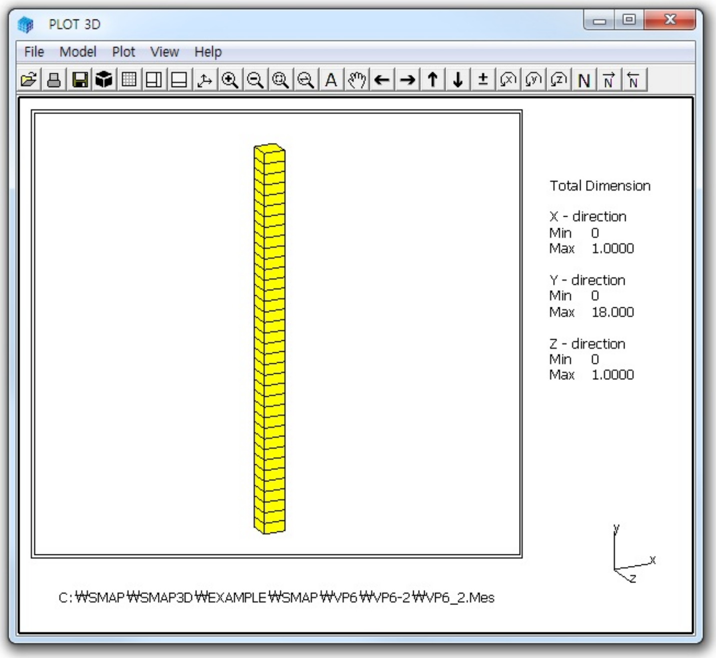
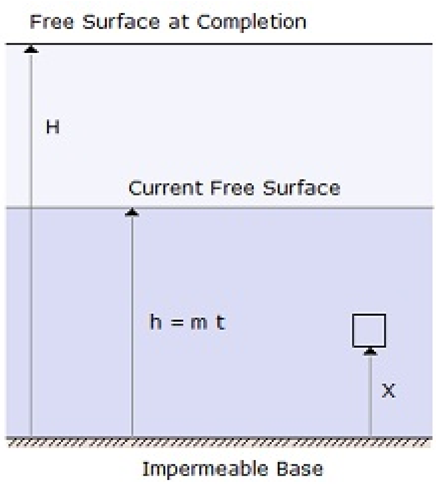


Figure 4.20 Finite element mesh



H	=	Height of complete fill
h	=	Current height of fill
T	=	m^2 t / C_v Time factor
m	=	Constant rate of increase of fill height
t	=	Current time
C_v	=	M_s k / γ_w Coefficient of consolidation
k	=	Coefficient of permeability
γ_w	=	Unit weight of water
M_s	=	$(1-\nu) E / ((1+\nu)(1-2\nu))$ Constrained modulus
E	=	Young's modulus
ν	=	Poisson's ratio
γ'	=	$\gamma - \gamma_w$ Effective unit weight
γ	=	$G_s \gamma_w (1-n) + \gamma_w n$ Wet unit weight
G_s	=	Specific gravity of solid grain
n	=	Porosity
ξ	=	X / h Normalized depth
π_{en}	=	$\pi_e / (\gamma_w X)$ Normalized excess pressure
π_e	=	$\pi_w - \gamma_w X$ Excess pore pressure
π_w	=	Pore water pressure

Figure 4.21 Fully saturated fill constructed at constant rate

Figure 4.22 shows the normalized excess pore pressure profiles at time factor T = 4. It has been normalized by the height of current fill.

As you see, the computed results of SMAP-3D are very close to Gibson's exact solution.

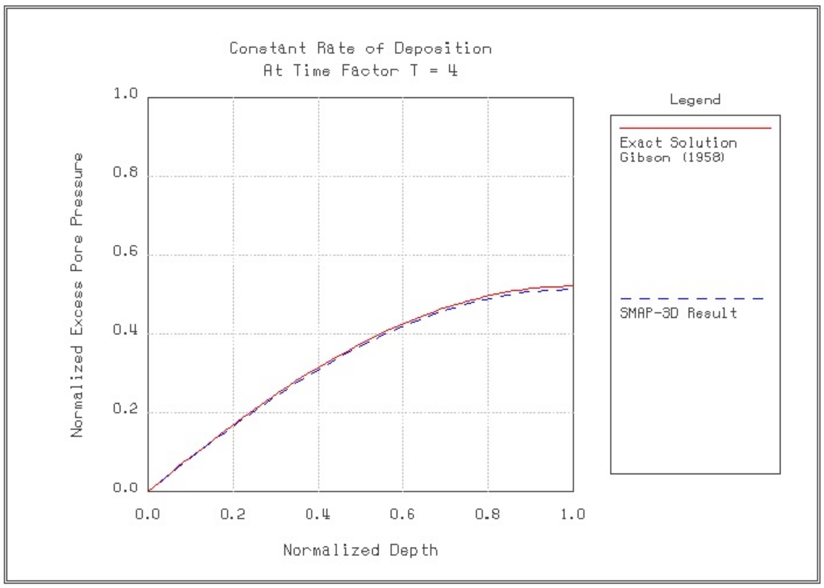


Figure 4.22 Normalized excess pressure profiles at T = 4

4.7 Drained Triaxial Compression Test

The problem is to check the implemented algorithm of the Modified Cam Clay Model in drained triaxial compression mode. The problem is to model the experimental test used by Karshenas and Ghaboussi.

The sample is modeled by a single cubic element with unit length as shown in Figure 4.23. The sample is artificial soil which is composed of 90% $C0_3C_a$ and 10% kaolinite. The material parameters tabulated in Figure 4.24 are those determined by Karshenas and Ghaboussi.

Both computed and measured values are plotted as a function of axial strain in Figure 4.25 for deviatoric stresses and in Figure 4.26 for volumetric strains. As you see, the SMAP-3D results reflect well the overall behavior of test results for the normally consolidated clay.

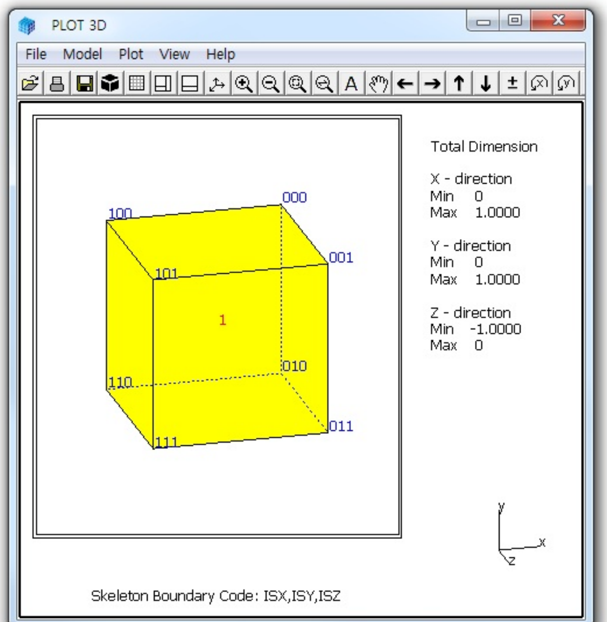
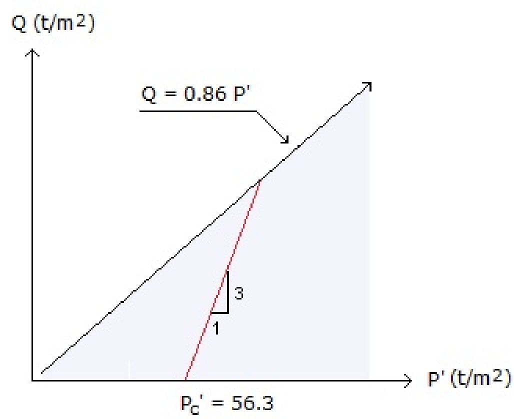


Figure 4.23 Finite element mesh



Material Parameters for Modified Cam Clay Model

Pre-consolidated Pressure	P_c'	=	56.3 t/m^2
Initial Elastic moduli	B_o	=	2540 t/m^2
	G_o	=	1530 t/m^2
Failure Parameter	M	=	0.86
Deformation Parameter	e_o	=	1.0
	ν	=	0.249
	C_c	=	0.2892
	C_r	=	0.1022

Figure 4.24 Drained triaxial compression test

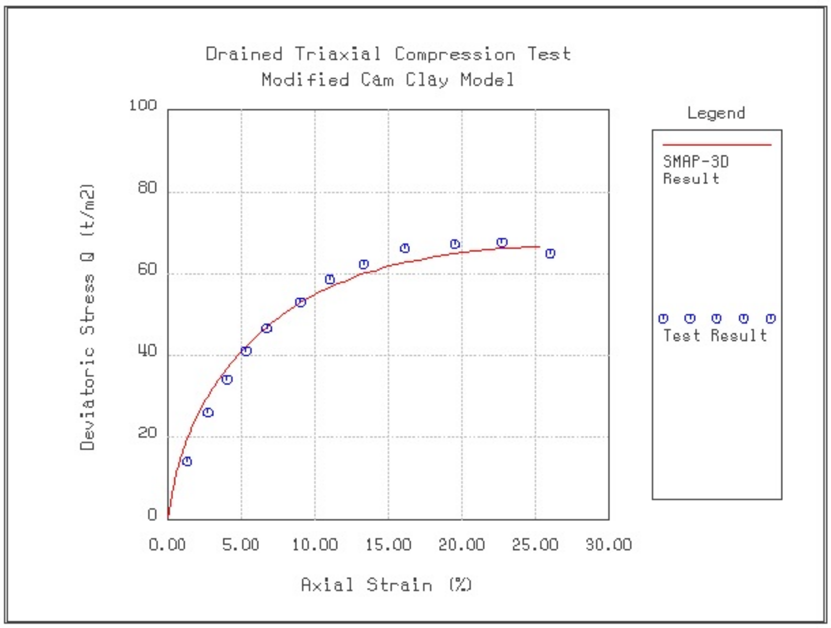


Figure 4.25 Deviatoric stress vs. axial strain for drained triaxial compression test

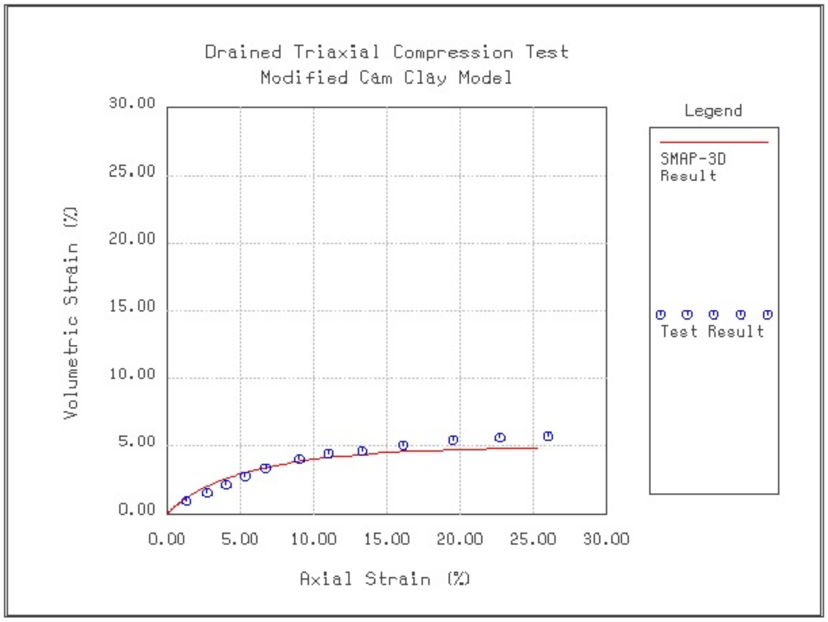


Figure 4.26 Volumetric strain vs. axial strain for drained triaxial compression test

4.8 Undrained Plane Strain Compression Test

The problem is to check the implemented algorithms of Modified Cam Clay Model in undrained plane strain compression stress path. The following analytical solution for this problem has been presented by Kim (1982).

Three components of the effective principal stresses are directly obtained from the specified value of axial strain increment.

$$d\sigma_x' = g_x \, d\varepsilon_x \qquad d\sigma_y' = g_y \, d\varepsilon_y \qquad d\sigma_z' = g_z \, d\varepsilon_z \tag{4.5}$$

$$\sigma_x' = \int d\sigma_x' \qquad \sigma_y' = \int d\sigma_y' \qquad \sigma_z' = \int d\sigma_z' \tag{4.6}$$

where

$$g_x = (b-a) - f\,[3a_o b + (a-b)\,a_x]$$

$$g_y = (a-b) - f\,[3a_o b + (a-b)\,a_y]$$

$$g_z = \quad - f\,[3a_o b + (a-b)\,a_z]$$

$$f = \frac{(a-b)\,(a_y - a_x)}{(a-b)\,(a_x^2 + a_y^2 + a_z^2) + q\,a_0^2\,b + \beta\,M^2 P'\,P_o'\,(2P' - P_o')}$$

$$a = \frac{6.9\,(1+e_o)\,(1-\nu)}{C_r\,(1+\nu)}\,P' \qquad b = \frac{6.9\,(1+e_o)\,\nu}{C_r\,(1+\nu)}\,P'$$

$$a_x' = a_o + 3(\sigma_x' - P') \qquad a_y' = a_o + 3(\sigma_y' - P') \qquad a_z' = a_o + 3(\sigma_z' - P')$$

$$\beta = \frac{2.3\,(1+e_o)}{(C_c - C_r)} \qquad a_o = \frac{2}{3}\,M^2\,(P' - \frac{1}{2}P_o')$$

$$P_o' = P_c' \exp\,(\,\beta\,\varepsilon_v^p\,)$$

Note that the initial stress conditions in Equation 4.6 should be imposed on the basis of the stress-strain state at the end of K_o-consolidated condition.

To perform numerical and analytical solutions, following K_o initial stresses and material parameters are assumed:

Initial stresses:

$\sigma_x' = 0.764\ t/m^2$ $\sigma_y' = 1.472\ t/m^2$ $\sigma_z' = 0.764\ t/m^2$

Material Parameters:

$e_o = 1.339$ $C_c = 0.508$ $C_r = 0.254$ $M = 1.1137$ $\nu = 0.4$

The sample is modeled by a single cubic element with unit length as shown in Figure 4.27.

Figure 4.28 shows effective stresses normalized by preconsolidation pressure and plotted as a function of axial strain. It seems that the SMAP-3D results are very close to the analytical solution. It is interesting to note that the effective stress (σ_x') in x direction where total stress remains constant is decreasing while other effective stresses (σ_y' and σ_z') change very little.

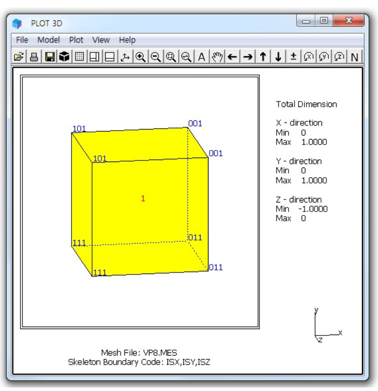


Figure 4.27 Finite element mesh

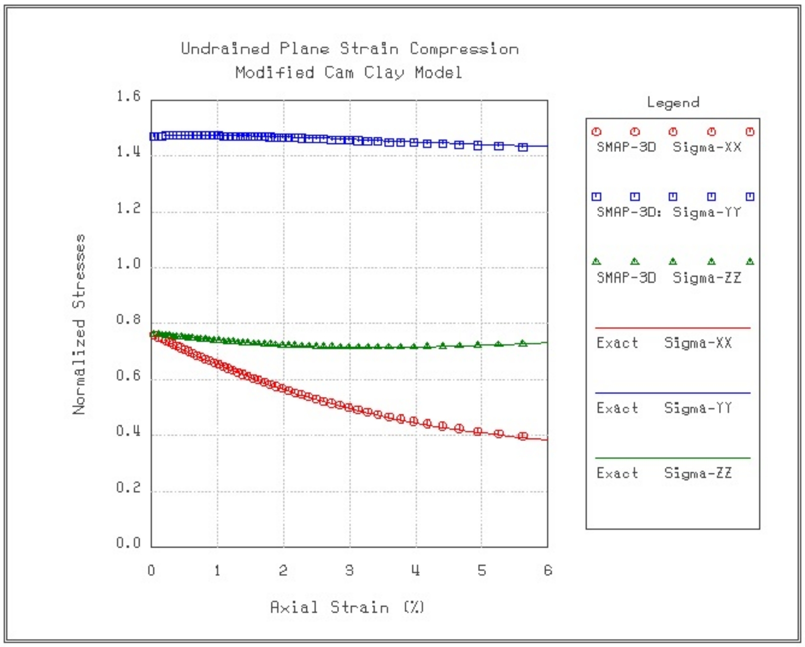


Figure 4.28 Effective stresses as a function of axial strain in K_o consolidated undrained plane strain compression test

4.9 Volumetric Creep in Isotropically Undrained Test

The problem is to check volumetric creep behavior in isotropically undrained test. The closed-form solution for this problem has been presented by Borja (1992).

$$P' = P_o \left[1 + \frac{C_c}{C_r} \left(\frac{t}{t_o} - 1 \right) \right]^{-\frac{C_r}{C_c}} \qquad \pi = P_o - P' \qquad (4.7)$$

Note that effective mean pressure (P') was P_o at initial time (t_o) but decreases with time (t) while total mean pressure (P_o) remains constant during the volumetric creep. Consequently, the excess pore pressure (π) increases with time.

The sample is modeled by a single cubic element with unit length as shown in Figure 4.29.

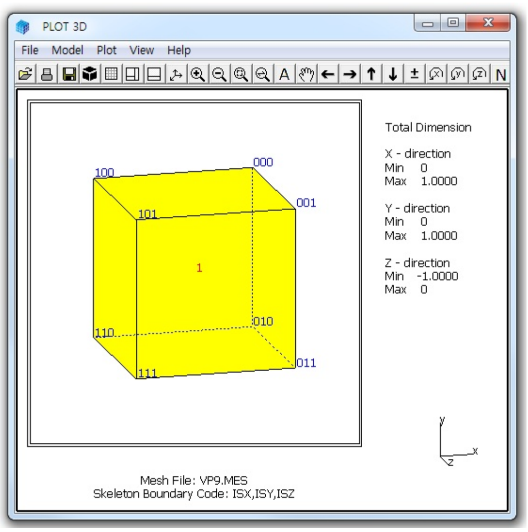


Figure 4.29 Finite element mesh

To conduct numerical calculation, the following initial conditions and material parameters are assumed:

e_o = 1.339 t_o = 1 day P_o = 1 t/m^2
C_c = 0.508 C_r = 0.254 C_α = 0.0374

Figure 4.30 shows variation of effective mean pressure and excess pore pressure as a function of time while total mean pressure remains constant. SMAP-3D results are almost identical to the closed-form solution.

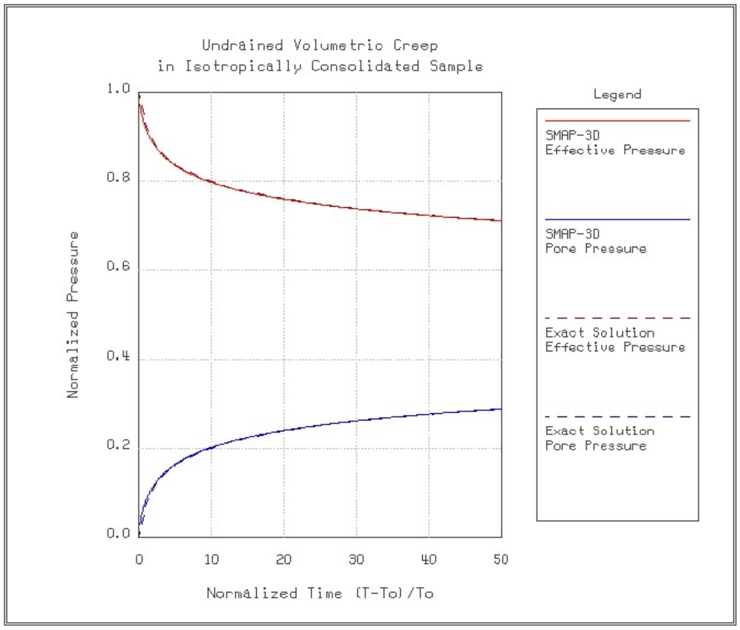


Figure 4.30 Volumetric creep in isotropically undrained test

4.10 Space Truss Analysis

This example problem is to solve the static response of space truss as shown in Figure 4.31. Block mesh example 5 illustrates how to generate this mesh. This space structure is subjected to a horizontal load along the negative z direction.

Graphical outputs are shown in Figure 4.32 for member axial forces and in Figure 4.33 for deformed shape of the structure. Note that the computed member forces are exact compared to the closed form solution.

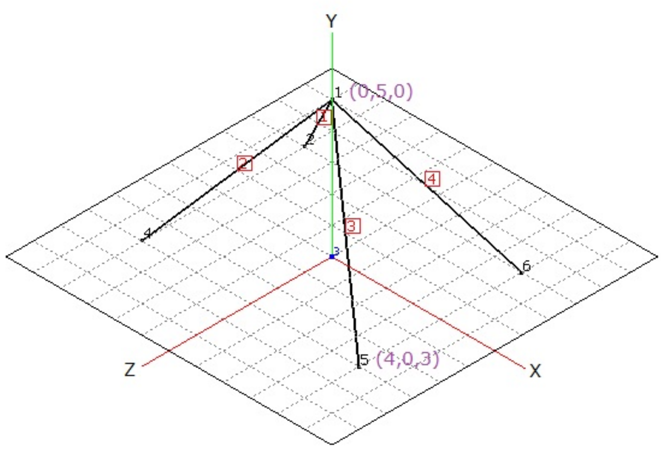


Figure 4.31 Schematic section view of space truss

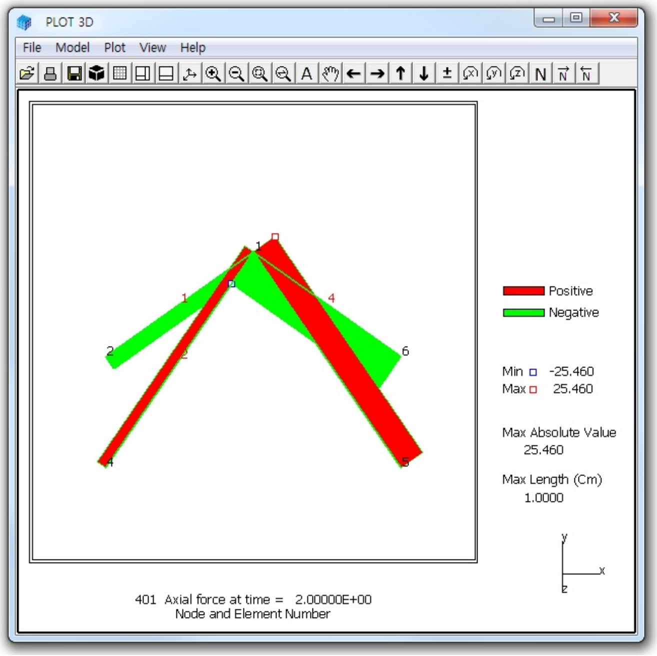


Figure 4.32 Member axial forces

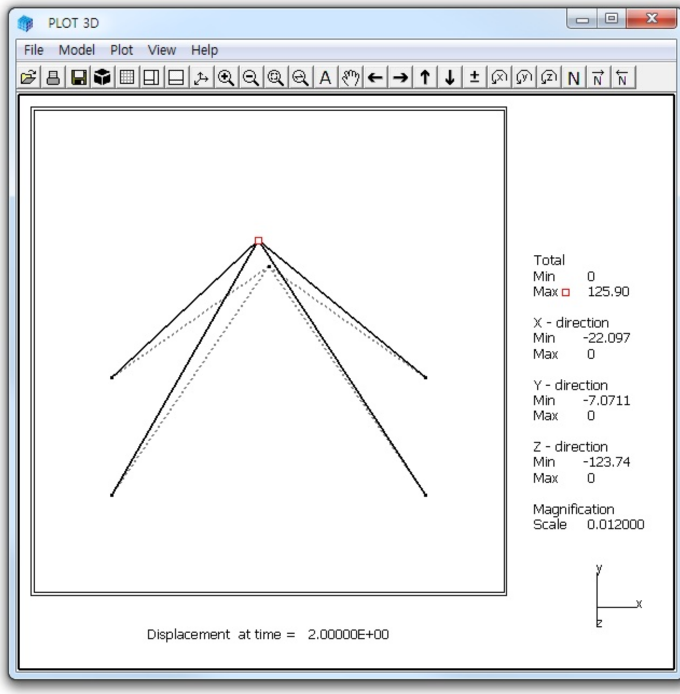


Figure 4.33 Deformed shape of space truss

4.11 Fixed End Beam Analysis

This example problem is to solve fixed end beam subjected to a concentrated load at mid span as schematically shown in Figure 4.34.

The exact solution for this beam is given below

$$\delta_{max} = \frac{PL^3}{192\,EI} = 0.01046 \text{ m} \qquad M_{max} = \frac{PL}{8} = 12.5 \text{ t-m}$$

$E = 21 \times 10^6 \text{ t/m}^2$ $\quad \nu = 0.3$ $\quad L = 10 \text{ m}$

$A = 0.008412 \text{ m}^2$ $\quad I = 2.37 \times 10^{-4} \text{ m}^4$

δ_{max} = Maximum deflection at mid span

M_{max} = Maximum bending moment at mid span

The problem has been modeled by 20 beam elements as shown in Figure 4.35. Graphical outputs are plotted in Figures 4.36 and 4.37 for deformed shape and bending moment diagram, respectively. Both computed mid span deflection and maximum bending moment are the same as those of the exact solution.

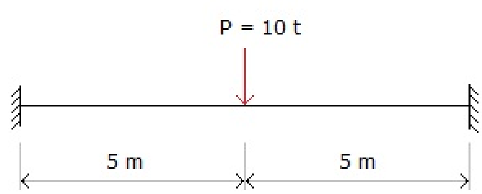


Figure 4.34 Fixed end beam subjected to concentrated load

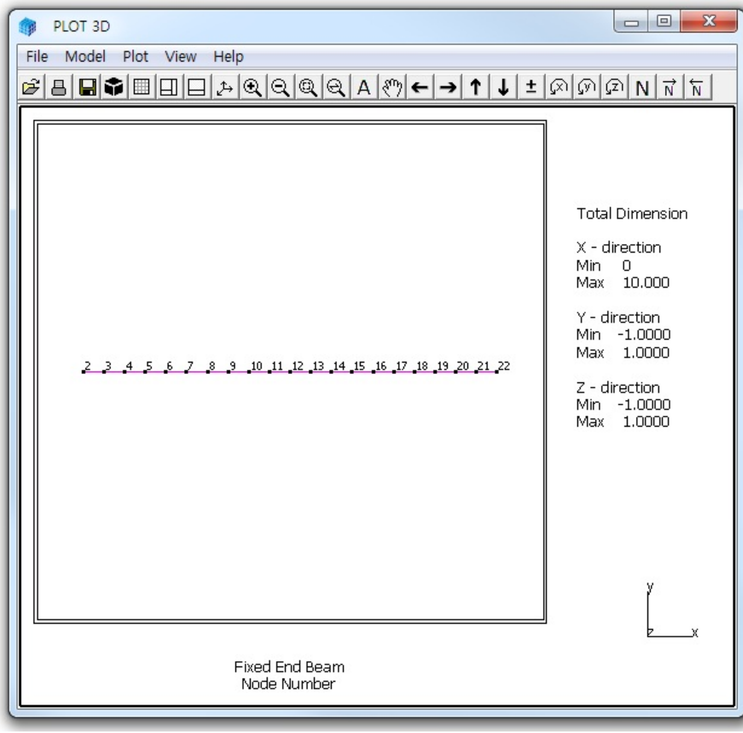


Figure 4.35 Beam node numbers

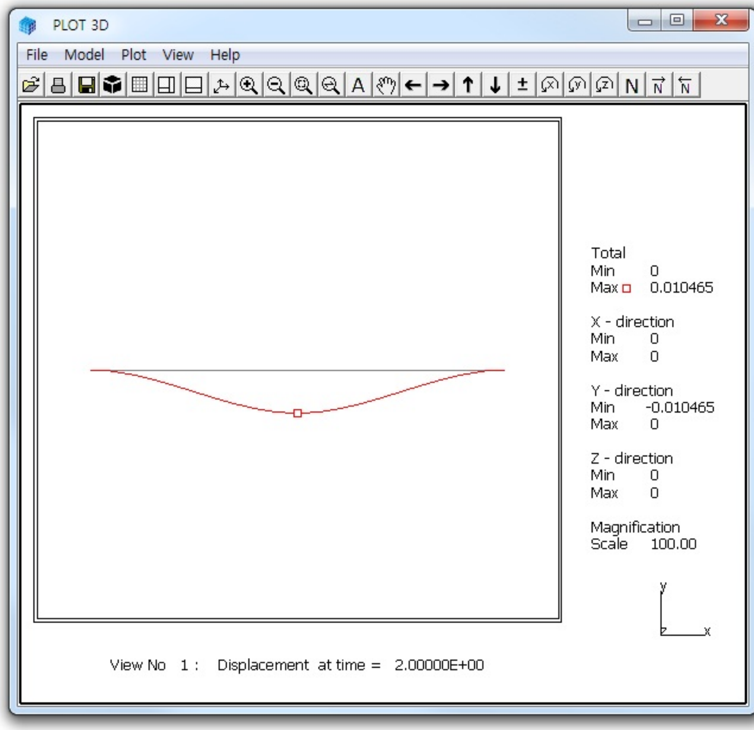


Figure 4.36 Beam deformed shape

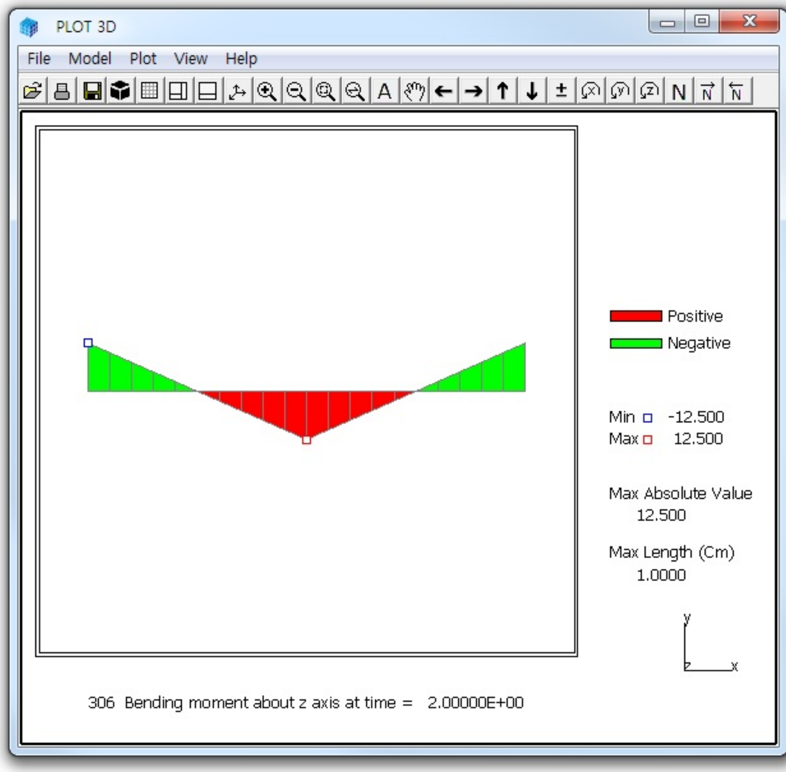


Figure 4.37 Bending moment diagram

4.12 Beam Dynamic Analysis

This example problem is to solve dynamic response of a simply supported beam subjected to a concentrated impact load at mid span. The exact solution for the deflection is given by:

$$\delta = \frac{2\, I_o\, L}{\pi^2 \sqrt{m\, E\, I}} \sum_{n=1,2}^{\infty} \frac{1}{n^2} \, \mathrm{Sin}\, \frac{n\, \pi\, x}{L} \, \mathrm{Sin}\, \frac{n\, \pi}{2} \, \mathrm{Sin}\, \omega_n\, t$$

$$\omega_n = n^2\, \omega_1 \qquad \omega_1 = \frac{\pi^2}{L^2} \sqrt{\frac{E\, I}{m}} \qquad m = \rho\, A$$

ρ	Mass density
A	Cross section area
L	Length of beam
I	Moment of inertia
I_o	Impulse
E	Young's modulus
x	Distance from beam support
t	Time

Numerical analysis for this simply supported beam shown in Figure 4.38 has been performed using the following parameters:

I_o = 0.1 t-sec

ρ = 0.786 $t\text{-}s^2/m^4$ L = 10 m

A = 0.008412 m^2 I = 2.37×10^{-4} m^4

E = 21×10^6 t/m^2 ν = 0.3

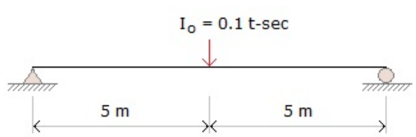


Figure 4.38 Simply supported beam subjected to impact load

The problem is modeled by 20 beam elements as shown in Fig. 4.39.
And impact load is simulated by the initial velocity applied at mid span.

Figure 4.40 shows the deformed shape at time t = 0.1 second.
Figure 4.41 shows time history plot of deflection at mid span.
SMAP-3D results agree well with the exact solution.

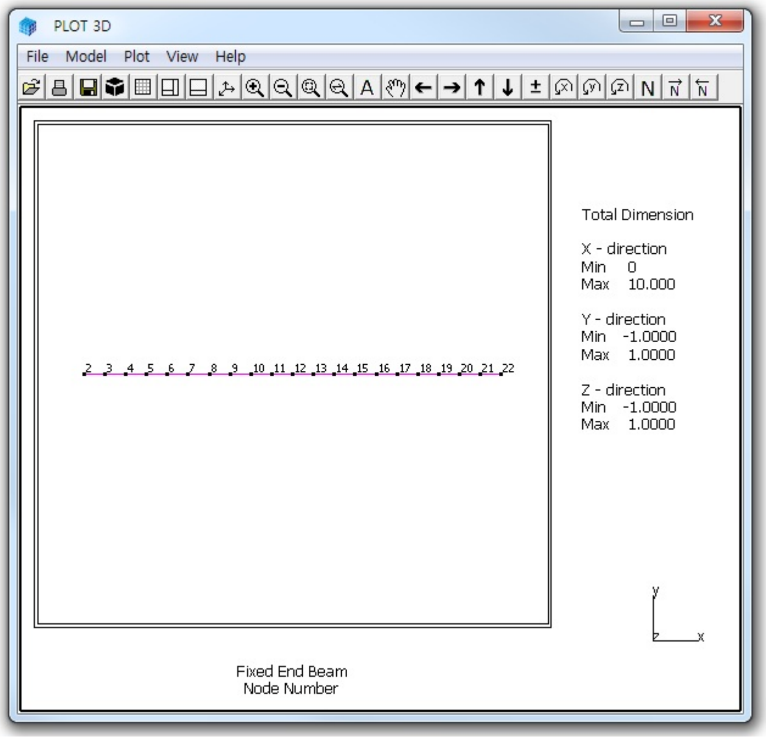


Figure 4.39 Beam node numbers

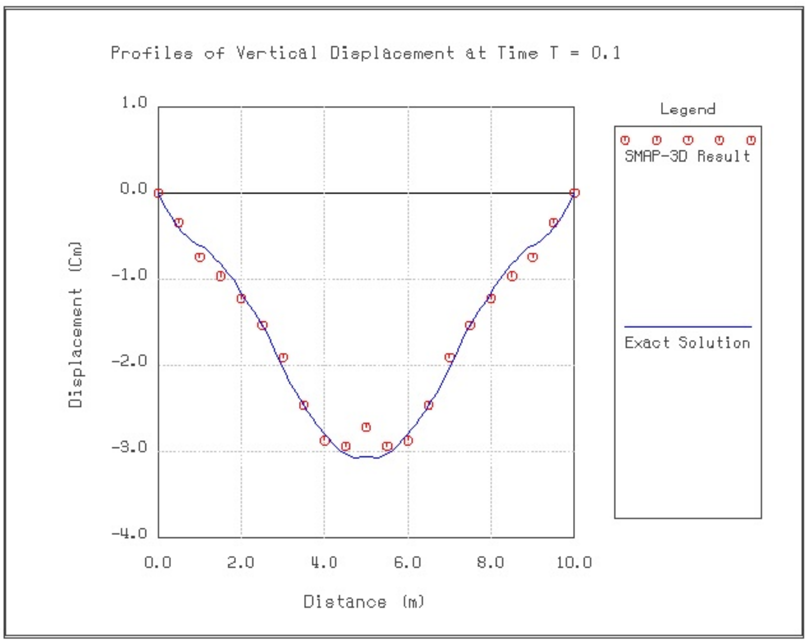


Figure 4.40 Beam deformed shape at 100 msec

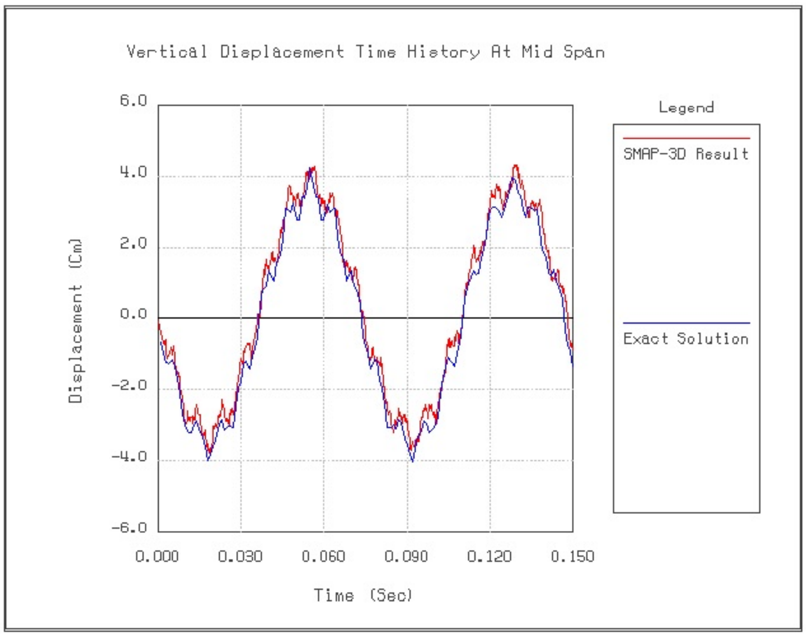


Figure 4.41 Time history of deflection at mid span

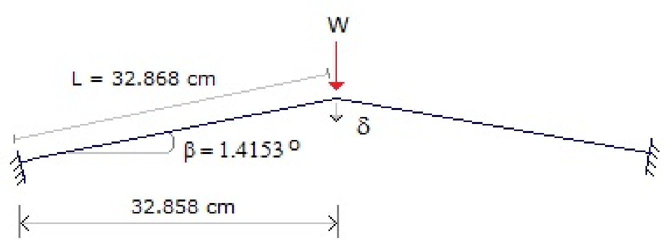
4.13 William's Toggled Beam Analysis

This classic problem of a rigidly jointed toggle is selected to verify the geometric nonlinear behavior of the continuum element.

For the toggle shown in Figure 4.42 the closed form solution as well as experimental results was obtained by Williams (Williams, F.W., An Approach to the Nonlinear Behavior of the Members of a Rigidly Jointed Plane Framework with Finite Deflections, Quarterly Journal of Mechanics and Applied Mathematics, Vol. 17, London, UK, 1964, pp. 451-469)

This toggled structure is modeled by 400 continuum finite elements: 100 elements along the beam axis, 4 elements across the depth, and only 1 element through the thickness.

Figures 4.43 and 4.44 show the load-deflection response at mid span and deformed shape at applied load of 16 kg, respectively.
SMAP-3D results are very close to the Williams' closed form solution.



$E = 7.238 \times 10^{6}\ t/m^{2}$

$I = 3.746 \times 10^{-2}\ cm^{4}$ $A = 1.18\ cm^{2}$

Figure 4.42 William's toggled beam (Not Scaled)

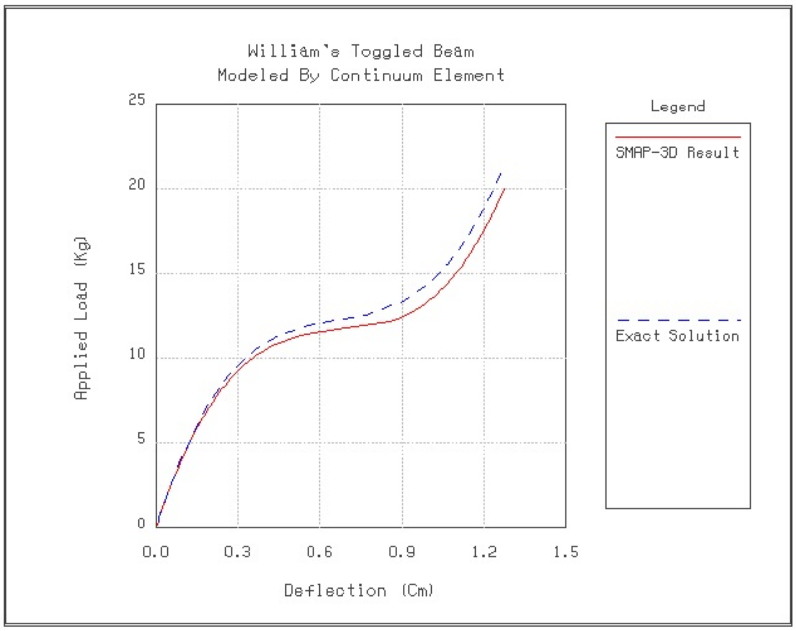


Figure 4.43 Load-deflection curve at mid span

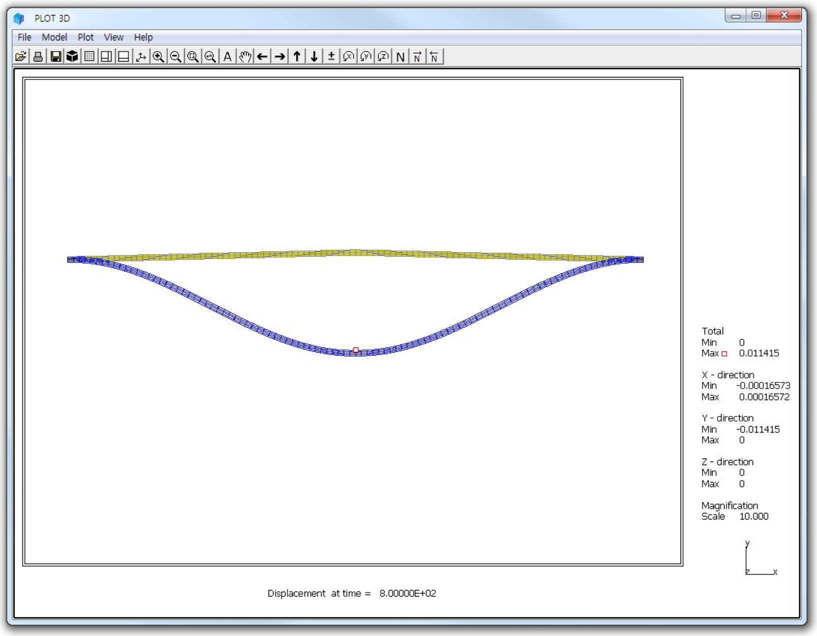


Figure 4.44 Deformed shape at applied load of 16 kg

4.14 Plane Strain Tunnel Analysis

The objective of this problem is to verify generation of in situ stresses and interaction of a tunnel liner with the surrounding soils. This example problem has been presented in SMAP-S2. Figure 4.45 shows schematic tunnel section view and material properties of soil and steel liner.

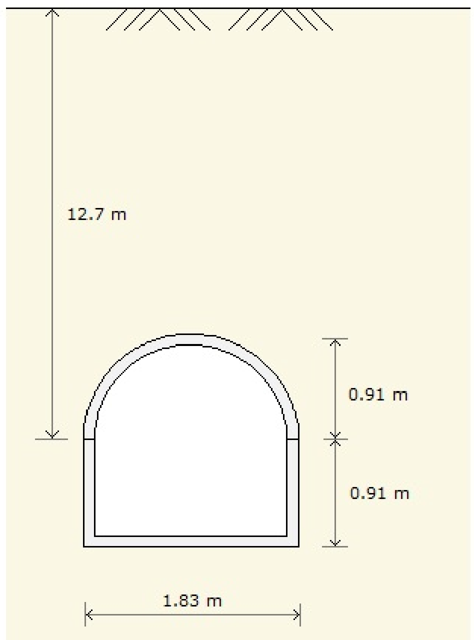
Figure 4.46 shows Finite element mesh. By symmetry, only the right half of the tunnel is modeled. Tunnel liner is modeled by shell elements as shown in Figure 4.47. Block mesh example 4 illustrates how to generate this mesh.

The first two load steps were used to generate in situ stresses. Tunnel excavation and liner installation were simulated by deactivating soil elements within the tunnel and activating liner elements at the third load step.

Graphical results are presented in the following order:

- Figure 4.48 Tunnel deformed shape
- Figure 4.49 Tunnel liner bending moment
- Figure 4.50 Tunnel liner axial stress
- Figure 4.51 Principal stress vector
- Figure 4.52 Major principal stress distribution
- Figure 4.53 Minor principal stress distribution

SMAP-3D results are almost identical to SMAP-S2 results



Soil		Steel Liner	
E	= 7,030 t/m^2	E	= 20.4 x10^6 t/m^2
ν	= 0.23	ν	= 0.3
K_o	= 0.3	t	= 2.54 cm (Thickness)

Figure 4.45 Schematic tunnel section view (Not Scaled)

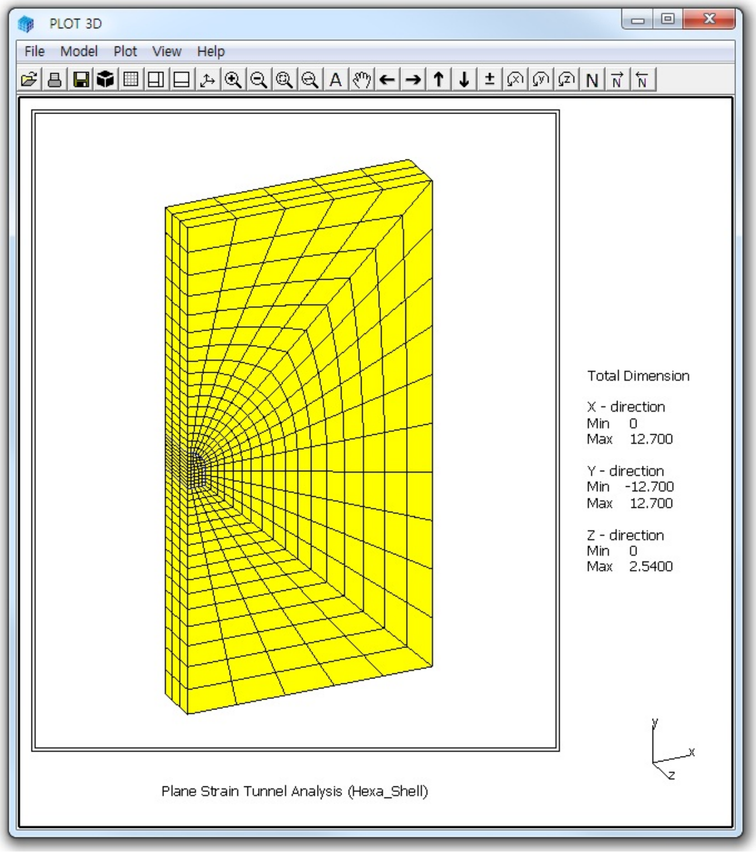


Figure 4.46 Finite element mesh

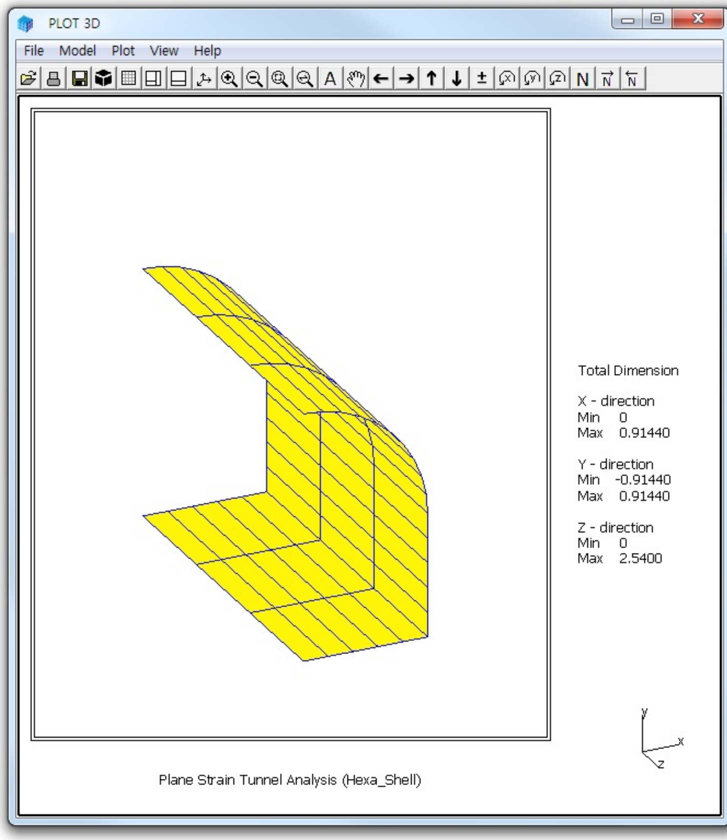


Figure 4.47 Finite element mesh for liner

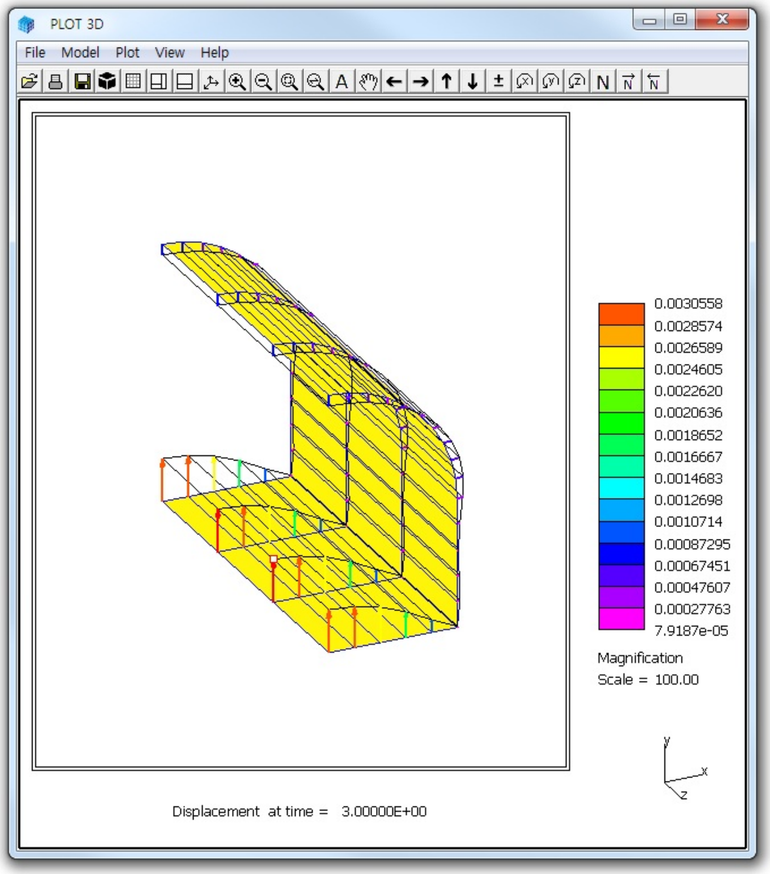


Figure 4.48 Tunnel deformed shape

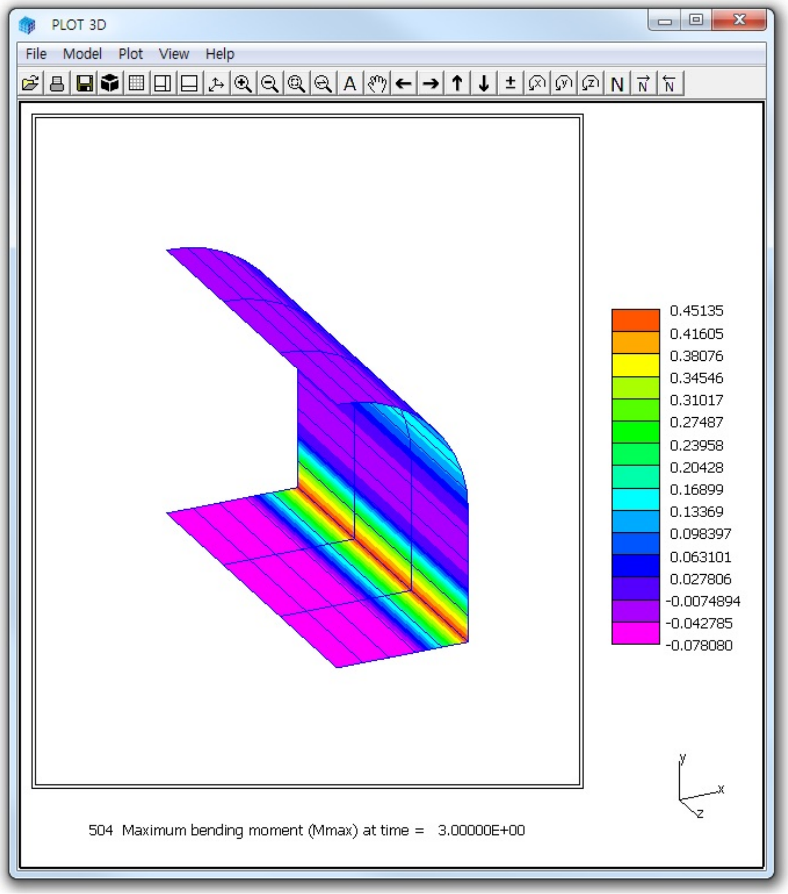


Figure 4.49 Tunnel liner bending moment

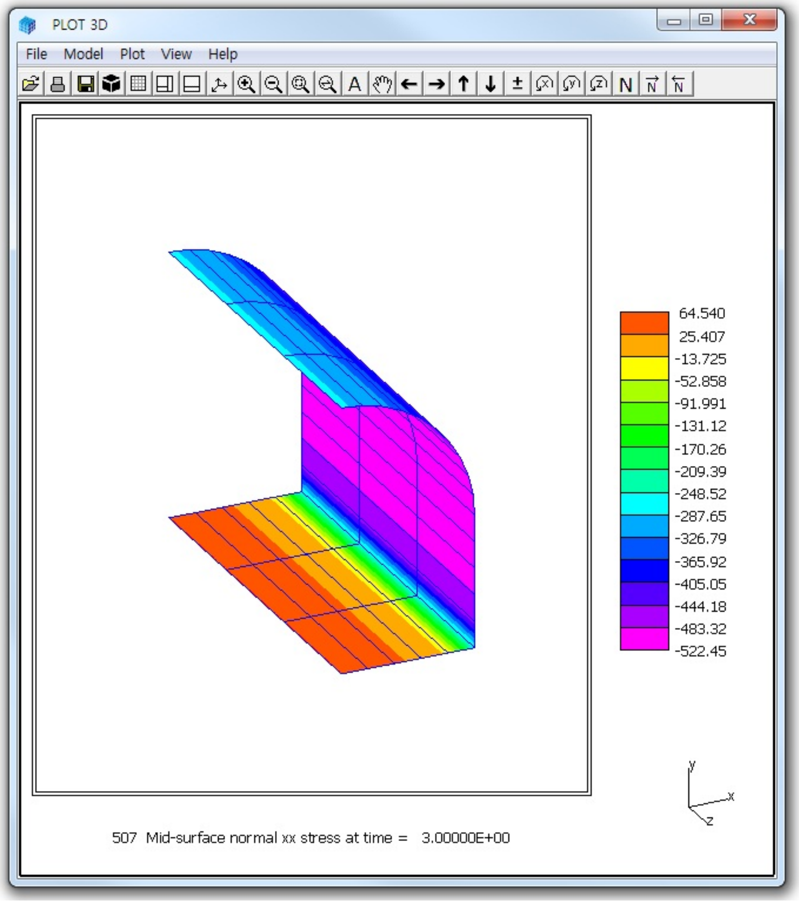


Figure 4.50 Tunnel liner axial stress

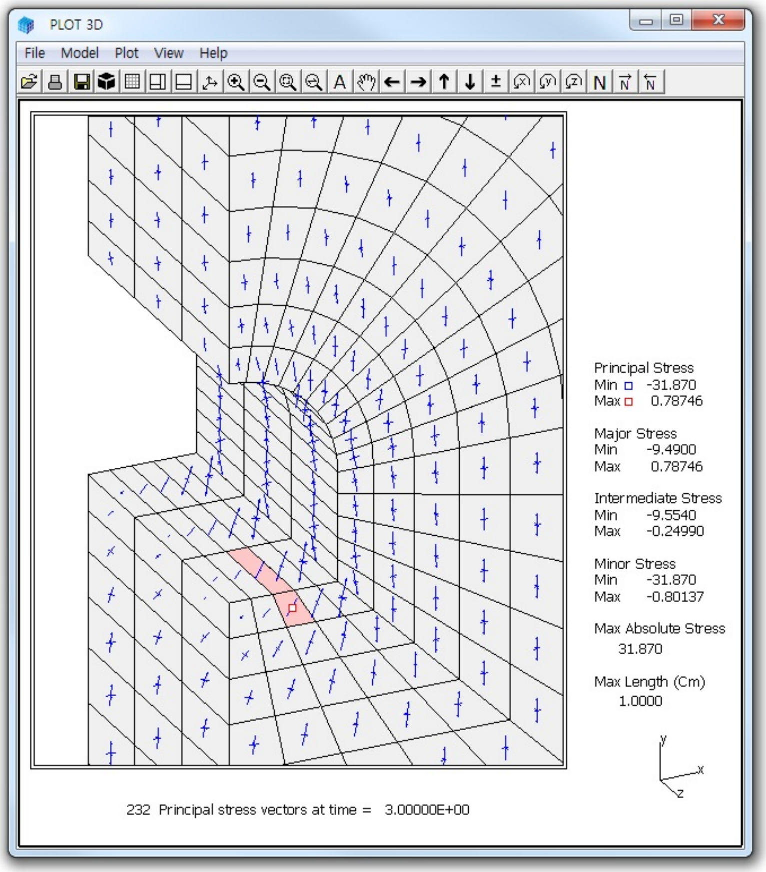


Figure 4.51 Principal stress vector

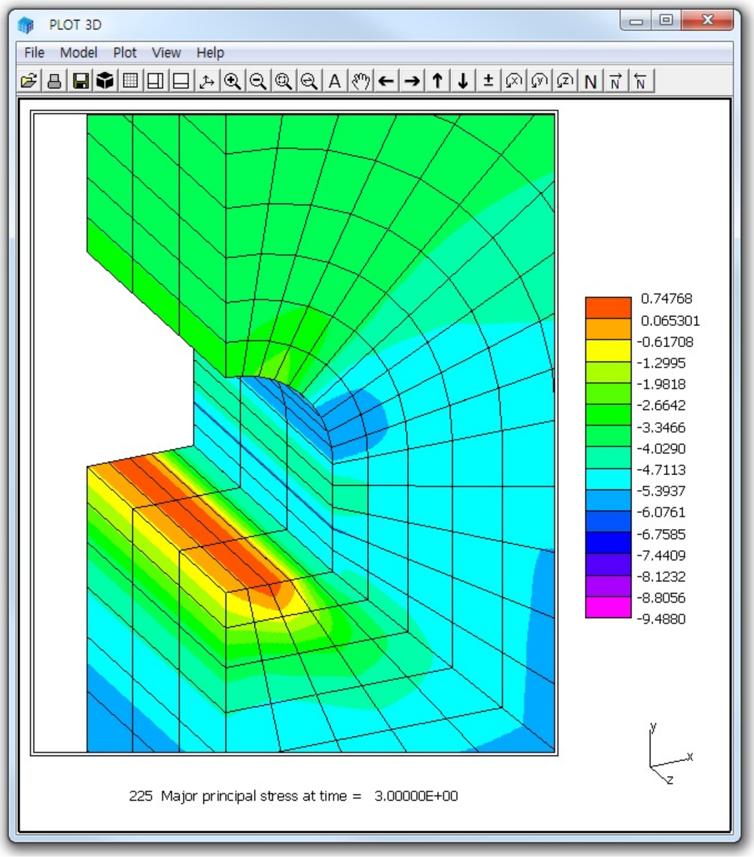


Figure 4.52 Major principal stress distribution

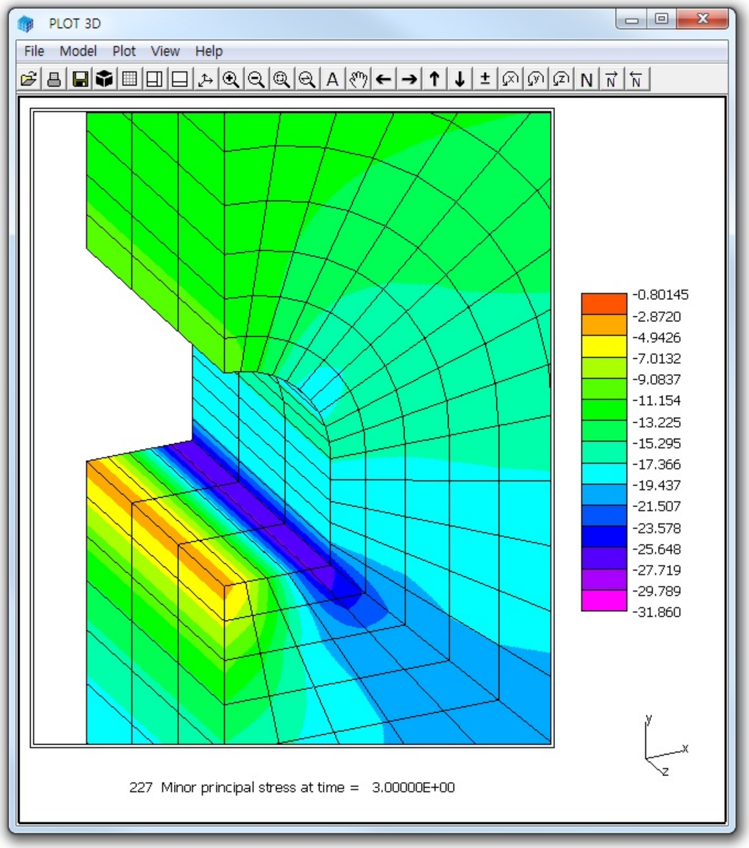


Figure 4.53 Minor principal stress distribution

4.15 Hemispherical Shell

This classic problem of a hemispherical shell with 18° hole is selected to verify accuracy of the membrane and bending performance of shell element.

The theoretical solution for this problem was presented by R. H. MacNeal and R. L. Harder (A proposed standard set of problems to test finite element accuracy, Finite Element Anal. Des., 1, 3-20, 1985).

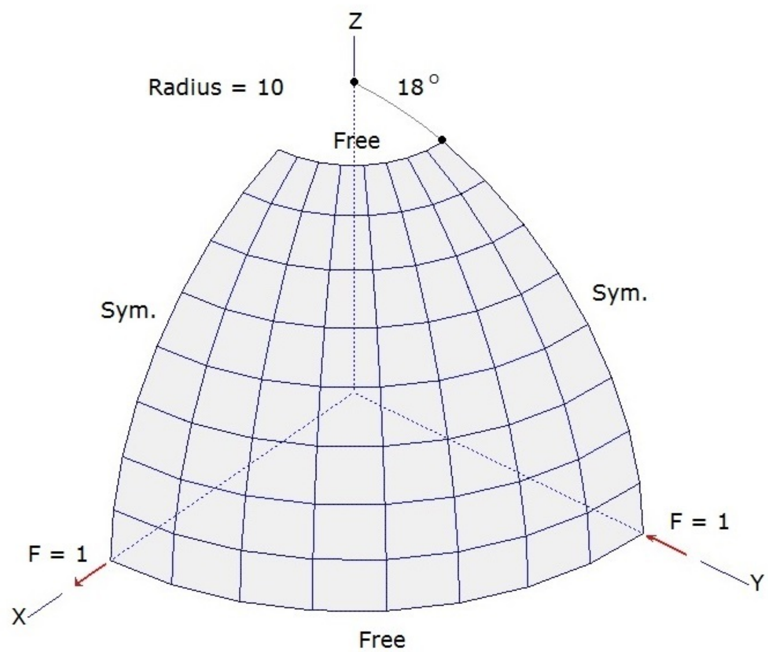
Figure 4.54 shows finite element mesh, material properties, loading and boundary conditions. By symmetry, only a quadrant of the shell is modeled. Block mesh example 3 illustrates how to generate this mesh.

Graphical results are presented in the following order:

- Figure 4.55 Deformed shape
- Figure 4.56 Maximum bending moment

SMAP-3D result gives excellent results for the displacement at the point of load in the direction of load as compared below:

Theoretical solution = 0.094
SMAP-3D result = 0.0944



$E = 6.825 \times 10^7$ $\nu = 0.3$ Thickness = 0.04

Figure 4.54 Hemispherical shell subjected to pinched point loads

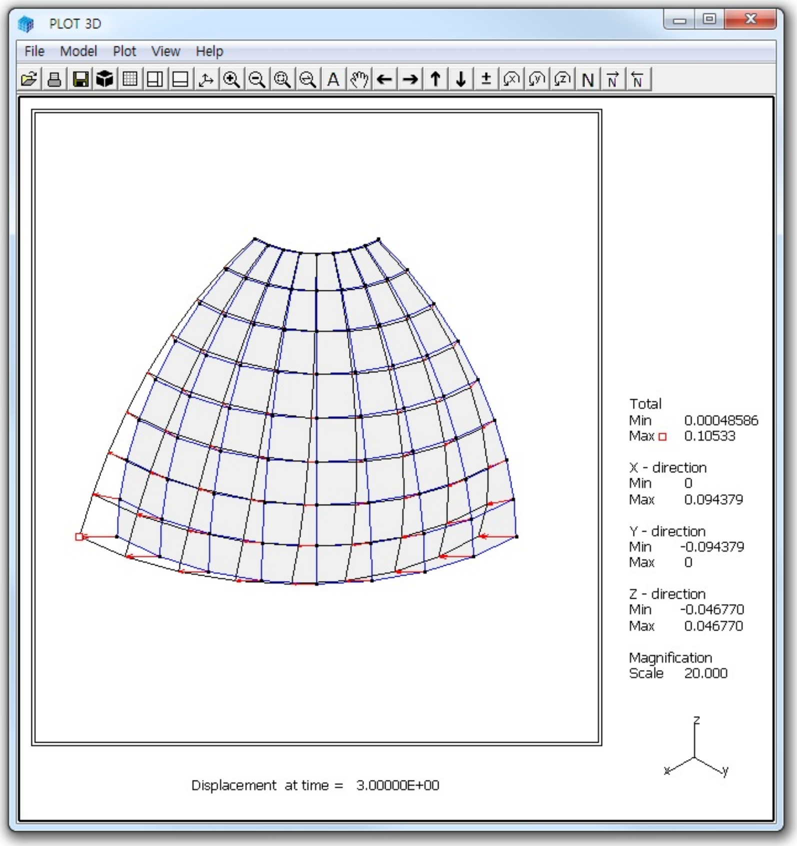


Figure 4.55 Deformed shape

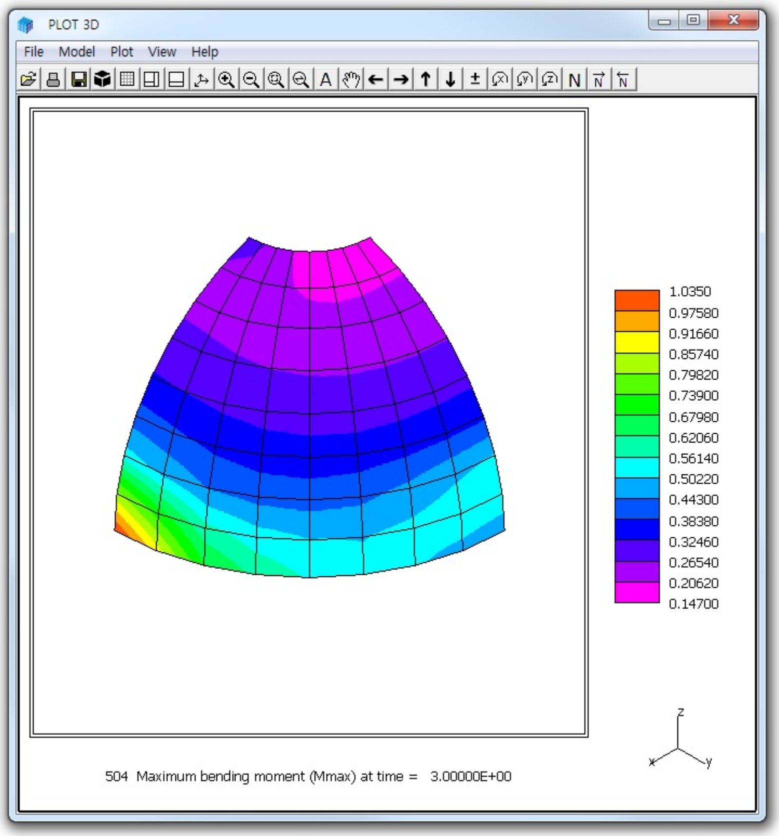


Figure 4.56 Maximum bending moment

4.16 Simply Supported Plate Analysis

A simply supported rectangular plate, shown in Figure 4.57, is selected to verify the dynamic response of shell element. By symmetry, only a quarter of the plate is modeled. The plate is subjected to a concentrated step load at center.

The computed displacement time history at plate center is shown in Figure 4.58 along with static results. SMAP-3D solution shows good results with such a relatively coarse mesh:

Static vertical displacement at plate center

Kirhhoff theory = 0.925 cm
SMAP-3D result = 0.942 cm

Period of the first mode

Kirhhoff theory = 0.2366 sec
SMAP-3D result = 0.237 sec
(Estimated from Figure 4.58)

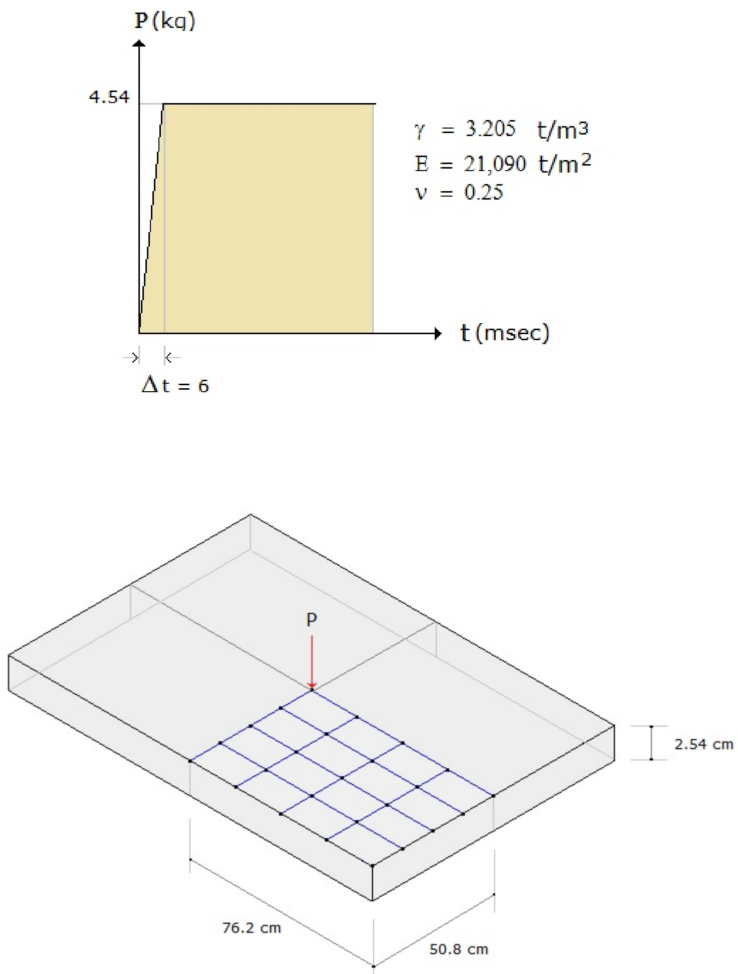


Figure 4.57 Simply supported plate (Not Scaled)

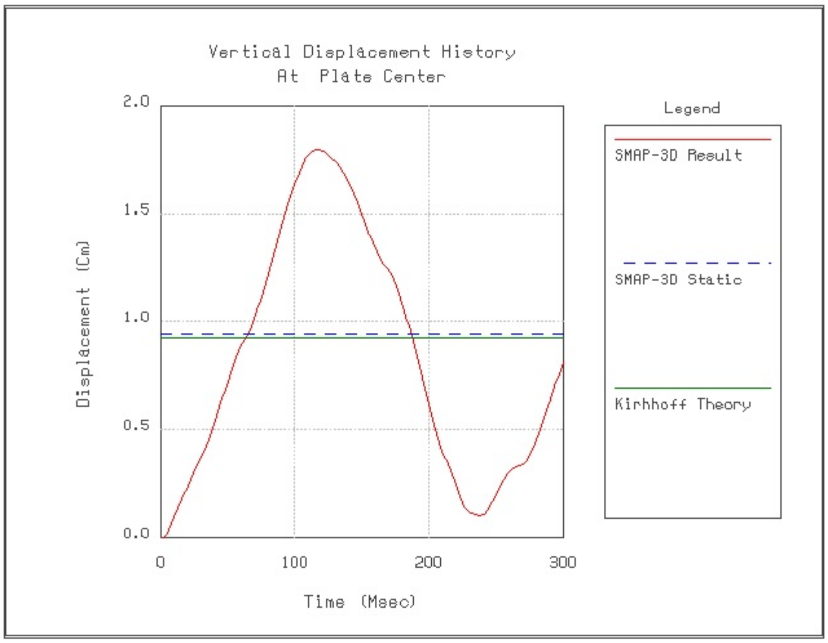


Figure 4.58 Vertical displacement time history at plate center

4.17 Heated Beam Analysis

A Simply supported plain concrete beam, shown schematically in Figure 4.59, is subjected to linear temperature increase through depth.

The temperature of the top surface of the beam is increased from -30° C to 50° C while temperature of the bottom surface remains constant at -30° C. Consequently, it is expected that the top surface expands relative to the bottom surface and the beam deflects upwards.

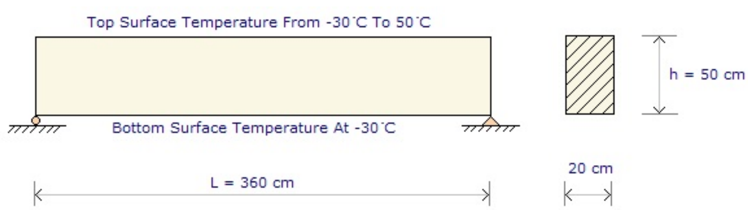


Figure 4.59 Heated beam subjected to temperature difference

By symmetry, only right half of the beam is modeled using a total of 22 shell elements as shown in Figure 4.60.

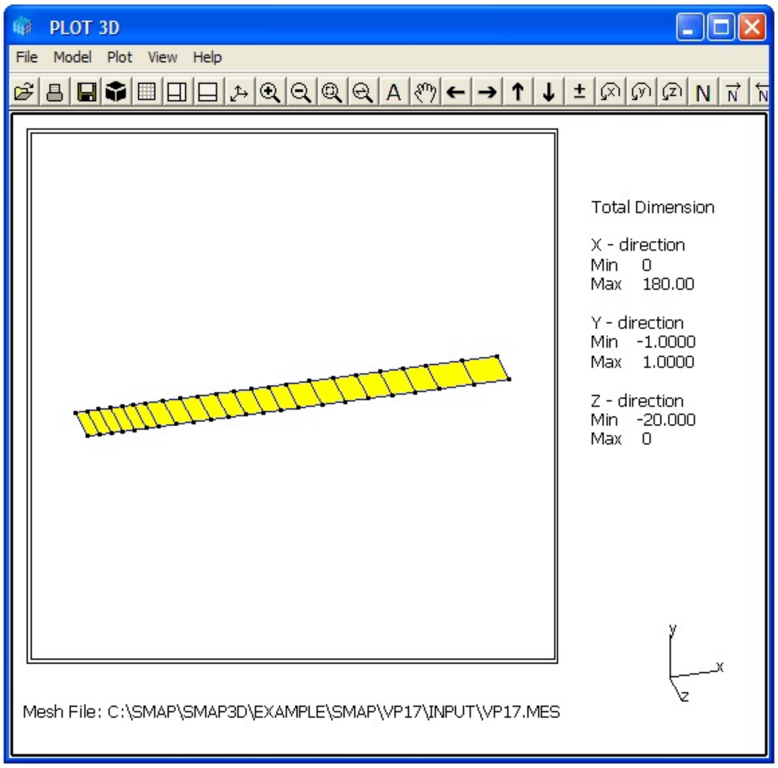


Figure 4.60 Finite element mesh

Material Properties are assumed as:

$\alpha = 3.2 \times 10^{-5}\,^{\circ}C^{-1}$ $E = 2.7 \times 10^{5}$ kg/cm^2 $\nu = 0.15$

Theoretical Maximum Deflection is given as:

$\delta_{max} = \alpha L^2 (T_{top} - T_{bottom})/(8\,h) = 0.8294$ cm

Figure 4.61 shows beam deflections. SMAP-3D result gives excellent results for vertical displacement at center of the beam.

Theoretical solution = 0.8294 cm

SMAP-3D solution = 0.8276 cm

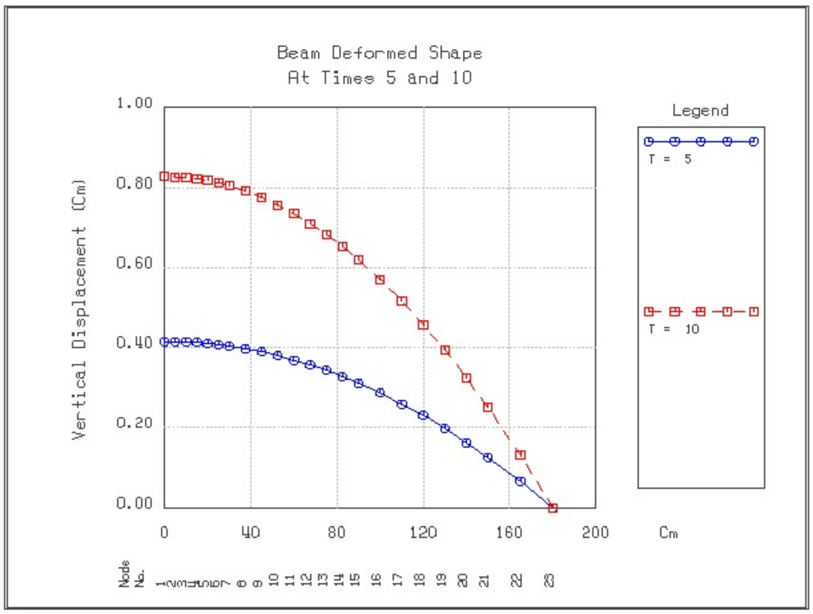


Figure 4.61 Beam deformed shapes

4.18 Thin Pipe Subjected To Internal Pressure

A very thin steel pipe, with radius of 20 cm and thickness of 0.003 cm, is subjected to the internal pressure of 0.2 kg/cm^2. The pipe is assumed to be in plain strain condition in the axial direction. Theoretically, the pipe is radially expanding due to the in-plane (membrane) deformations.

A total of 32 Shell elements is used to model the circular pipe as shown in Figure 4.62. A constant internal pressure is regarded as the hydrostatic pressure acting on the inner surface of Shell element.

Since the bending stiffness of the pipe is proportional to the third power of the pipe thickness while the in-plane stiffness is linearly proportional to the pipe thickness, the bending stiffness in such a very thin pipe would be much smaller than in-plane stiffness.

Thus, even a very small force associated with the bending degrees of freedoms may induce unrealistically large displacement. To improve the accuracy of displacement result, bending stiffness is multiplied by a factor of 100000.

The theoretical elastic solution gives the following radial displacement (u_r) and the hoop stress (σ_θ):

$$u_r = \frac{P \cdot r^2}{E \cdot t}(1 - v^2) \qquad \sigma_\theta = \frac{P \cdot r}{t}$$

where

E	Young's modulus	v	Poisson's ratio
t	Thickness of pipe	r	Radius of pipe
p	Internal pressure		

Numerical parameters are assumed as:

E	= 2.0×10^6 kg/cm^2	v	= 0.3
t	= 0.003 cm	r	= 20 cm
p	= 0.2 kg/cm^2		

Figure 4.63 shows deformed shape of pipe and Figure 4.64 shows hoop stress developed in the pipe.

SMAP-3D results are very close to theoretical solutions as listed below:

Theoretical solution: u_r = 0.01213 cm σ_θ = 1,333 kg/cm^2

SMAP-3D result: u_r = 0.01207 cm σ_θ = 1,327 kg/cm^2

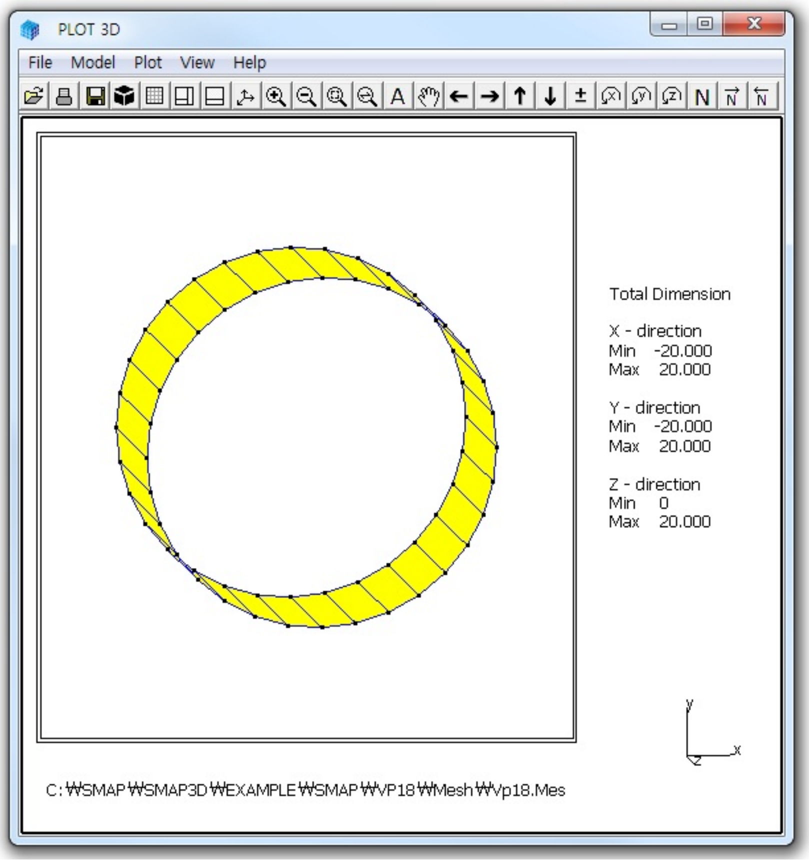


Figure 4.62 Finite element mesh

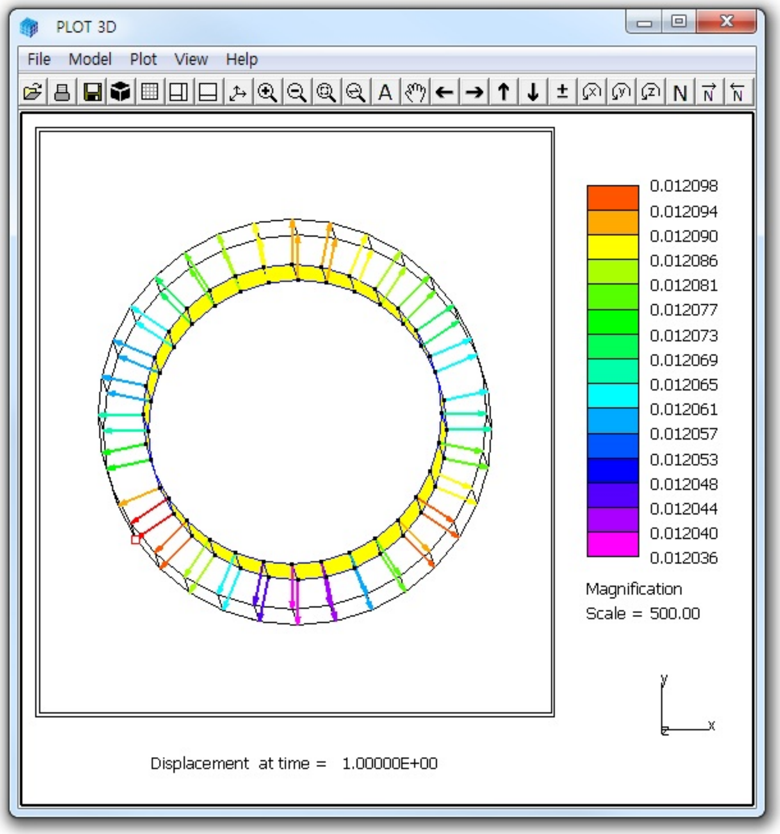


Figure 4.63 Pipe deformed shape

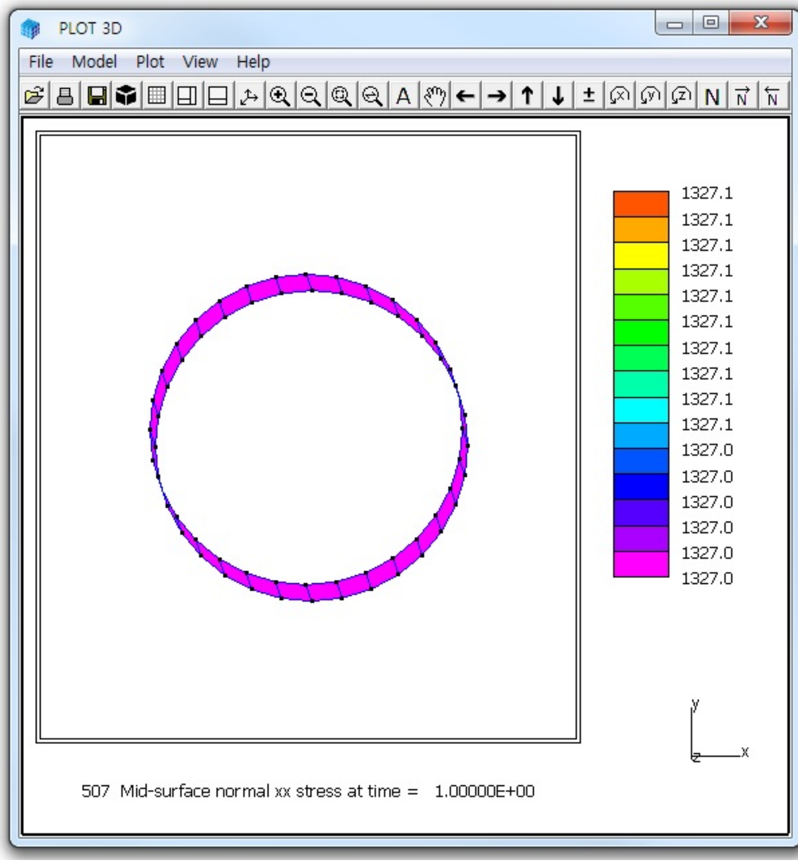


Figure 4.64 Pipe hoop stress

4.19 Preload Consolidation and Excavation

This example problem is to illustrate the analysis of the slope to be constructed under sea water. The in situ soil consists of about 40 meters of soft clay layer overlying hard soil layers.

Figure 4.65 shows schematically four stages of preloading embankment construction followed by excavation up to 17.6 meters below sea level.
Before preloading embankment, material zones 4, 5, 7, 8, 12 and 13 shown in Figure 4.66 are to be improved by drain methods (sand drain and PDB). In situ and improved soil properties are listed in Table 4.1.

The rate of embankment construction and excavation is shown schematically in Figure 4.67 along with computational steps used for SMAP-3D analysis.

Finite element meshes used for the analysis are shown:
- Figure 4.68 Finite element mesh
- Figure 4.69 Finite element mesh around preload
- Figure 4.70 Finite element mesh at completion
- Figure 4.71 Finite element mesh around slope

A total of 2330 elements is used to model a sequence of embankment construction and excavation.

Computed results at 152 days after completion of excavation are plotted by PLOT-3D in the following order:
- Figure 4.72 Deformed shape around slope
- Figure 4.73 Horizontal displacement distribution
- Figure 4.74 Pore pressure distribution
- Figure 4.75 Effective mean pressure distribution
- Figure 4.76 Deviatoric stress distribution

The horizontal contour lines of the hydrostatic water pressure in Figure 4.74 indicates that there will be no further consolidation settlement at 152 days after completion of excavation. Figure 4.76 shows that deviatoric stresses are concentrated around the base of the slope. Looking at both effective mean pressure (p') and deviatoric stress (q), the value of stress ratio (q/p′) is less than one at locations approximately 3 meters away from the surface of slope.

Figure 4.77 shows the location of selected elements where time histories of stresses and stress path are plotted. These selected elements are located within 10 meters from the surface of slope.

Computed results of time history of stresses are plotted by PLOT-XY in the following order:

Figure 4.78 Stress time history at element 120
Figure 4.79 Stress path at element 120

It should be noted that first 2000 days are used to generate in situ k_0 stresses. During embankment construction, excess pore water pressures develop mostly immediately after placement and then dissipate with time while effective stresses develop gradually. During excavation, effective stresses undergo unloading stress paths which will end up with higher horizontal stresses in over consolidated soil condition and pore water pressures drop rapidly and then get gradually back to the hydrostatic water pressure level as the dissipation length is shorter.

It is worth noting that the effective mean pressures decrease slightly while deviatoric stresses increase during the short period of placement of preloading fills. This is due to the fact that the compressive plastic volumetric strains develop while the total volumetric strains remain nearly constant since very little excess pore pressure dissipations are expected in such a short period.

Examining all the stress path plots, elements 120, 299, 477, 655 and 833 lie on the failure surface and elements 300 and 478 are slightly below the failure surface. Noting that elements 120, 299, 477, 655 and 833 are located within 2 meters from the surface of slope and elements 300 and 478 are located within 4 meters from the surface of slope, it is expected that soil failure would occur around the slope base within approximately 3 meters from the surface of slope. It may require redesign of the slope or accompany engineered structures for the slope to stay in safe.

Table 4.1 Material model parameters

Elastic Model Parameters

Material Number	Porosity (%)	Specific Gravity	k (m/day)	E (t/m^2)	v	Remark
1	42	2.7	0.0864	600	0.33	Dry
2	42	2.7	0.0864	600	0.33	Dry
3	42	2.7	0.0864	600	0.33	Saturated
6	44	2.7	0.0864	1400	0.33	Saturated
14	99.9	2.7	10.0	10.0	0.2	Water

Modified Cam-Clay Model Parameters

Material Number	Porosity (%)	Specific Gravity	k (m/day)	e_o	C_c	C_r	M
4	59.1	2.72	* 0.0274	1.49	0.55	0.077	1.2
5	61.0	2.72	* 0.0274	1.57	0.70	0.098	1.2
7	59.1	2.72	* 0.0274	1.49	0.55	0.077	1.2
8	61.0	2.72	* 0.0274	1.57	0.70	0.098	1.2
9	59.1	2.72	4.32×10^{-5}	1.49	0.55	0.077	1.2
10	61.0	2.72	4.32×10^{-5}	1.57	0.70	0.098	1.2
11	61.0	2.72	4.32×10^{-5}	1.62	0.80	0.112	1.2
12	61.0	2.72	* 0.0274	1.62	0.80	0.112	1.2
13	61.0	2.72	* 0.0274	1.62	0.80	0.112	1.2

(*) Soil permeability improved by sand drain or PDB

<table>
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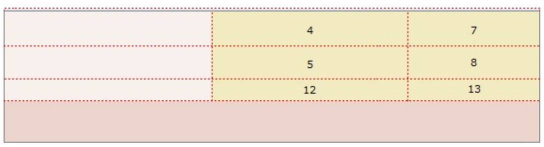
<tr><td>Step 101 (2000 days)
</td><td>In Situ State

Sand Drain:
Material 4, 5, 12

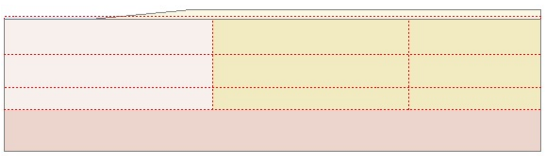
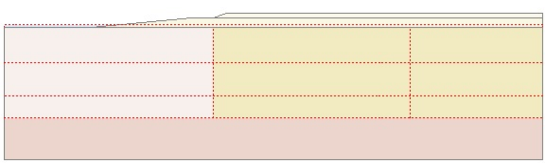
PDB:
Material 7, 8, 13</td></tr>
<tr><td>Step 104 (+ 15 days)
</td><td>Completion of Stage 1 Embankment</td></tr>
<tr><td>Step 165 (+ 321 days)
</td><td>Completion of Stage 2 Embankment</td></tr>
</table>

Figure 4.65 Construction sequence

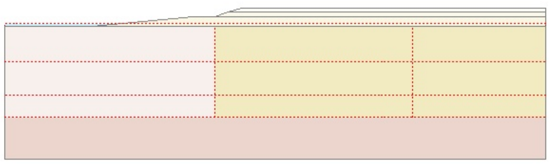
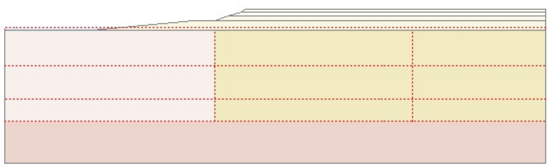
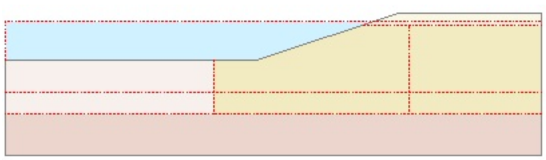
Construction State	Description
Step 216 (+ 627 days) 	Completion of Stage 3 Embankment
Step 267 (+ 933 days) 	Completion of Stage 4 Embankment
Step 333 (+ 1265 days) 	Completion of Final Excavation (Dredging)

Figure 4.65 Construction sequence (Continued)

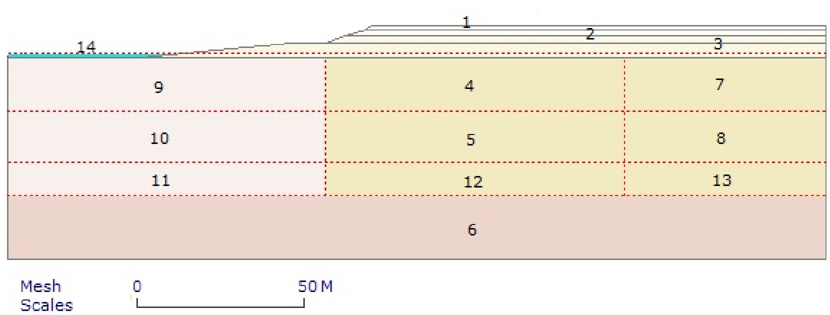


Figure 4.66 Material number

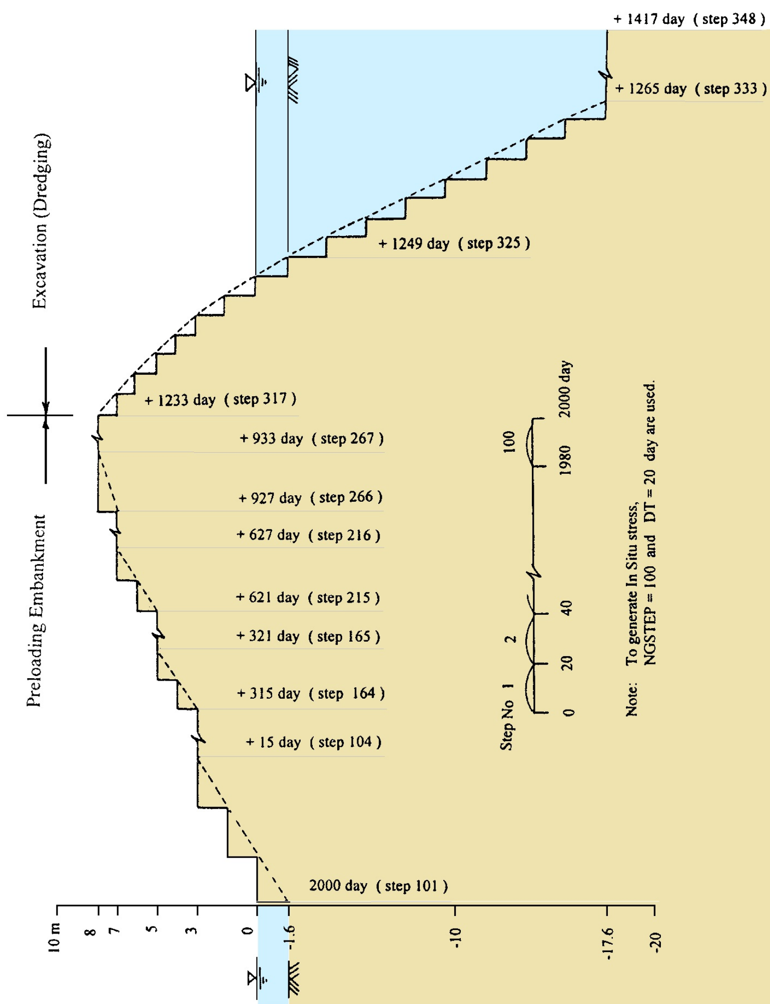


Figure 4.67 Construction rate

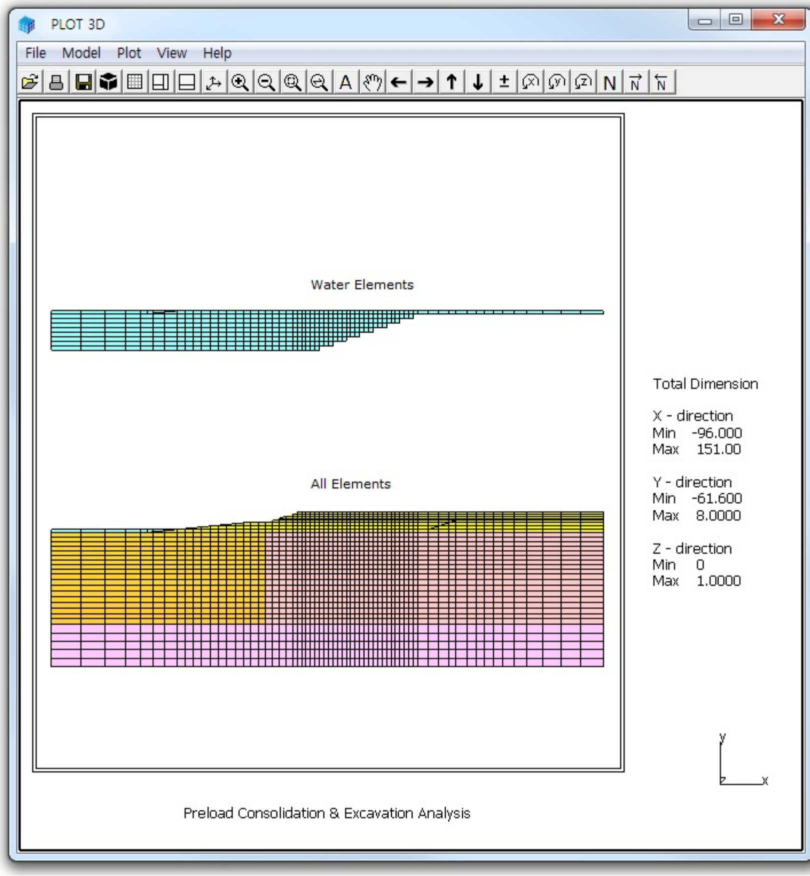


Figure 4.68 Finite element mesh

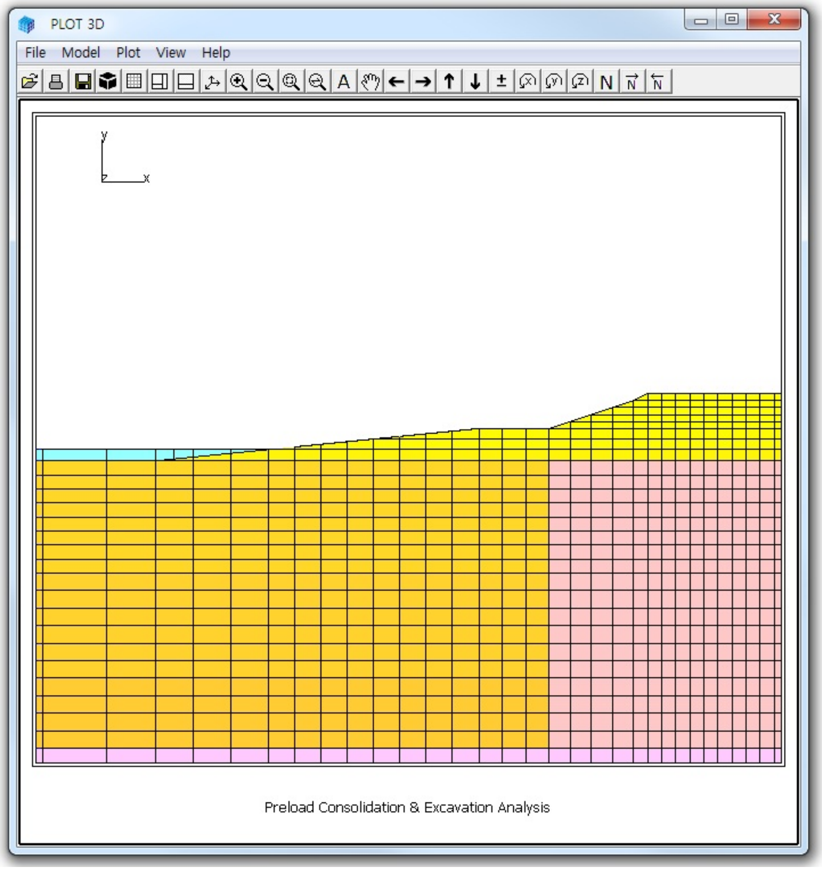


Figure 4.69 Finite element mesh around preload

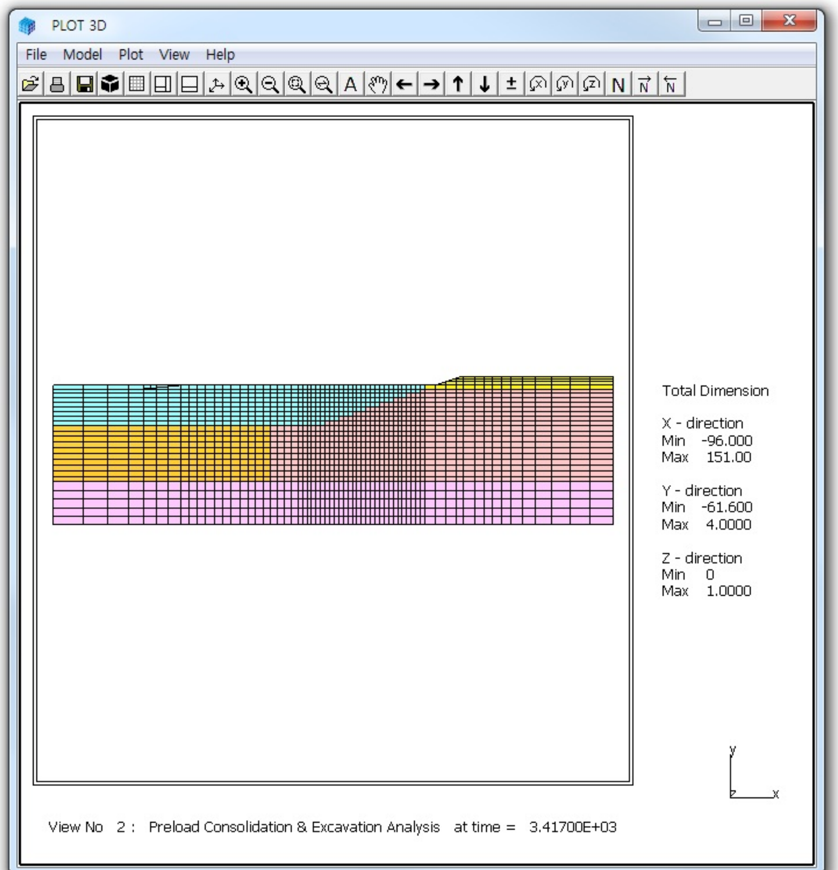


Figure 4.70 Finite element mesh at completion

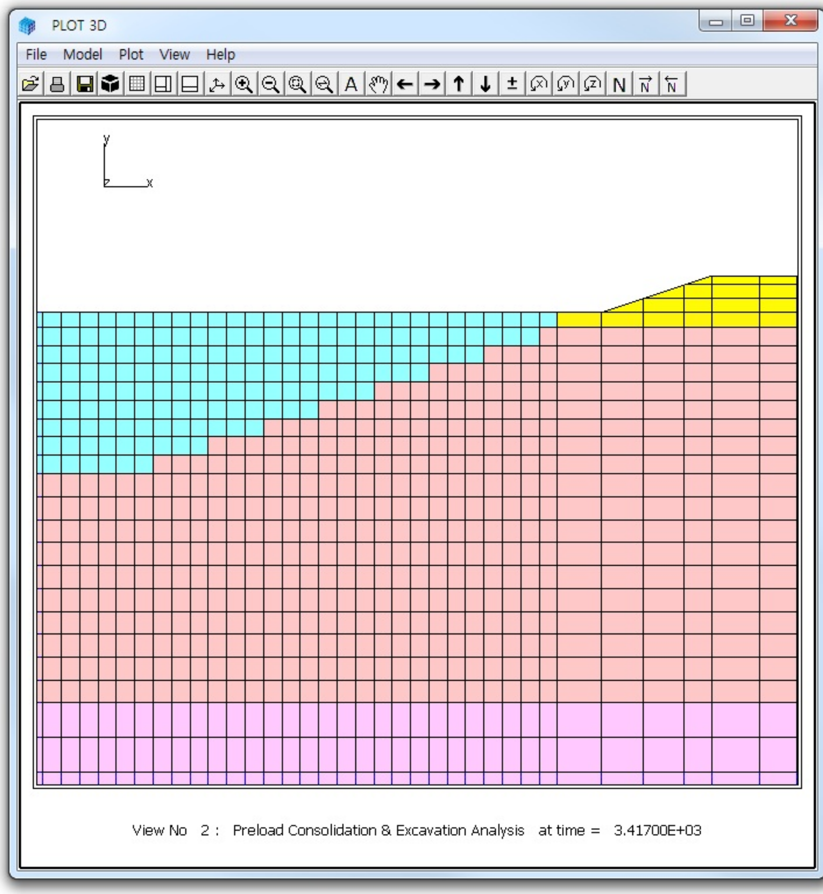


Figure 4.71 Finite element mesh around slope

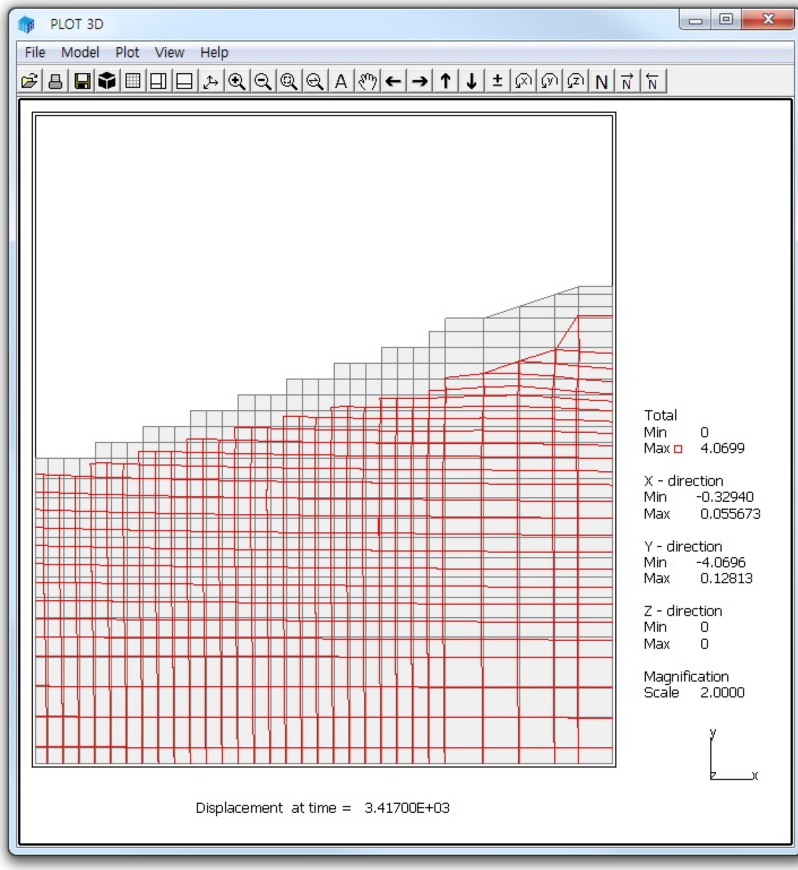


Figure 4.72 Deformed shape around slope

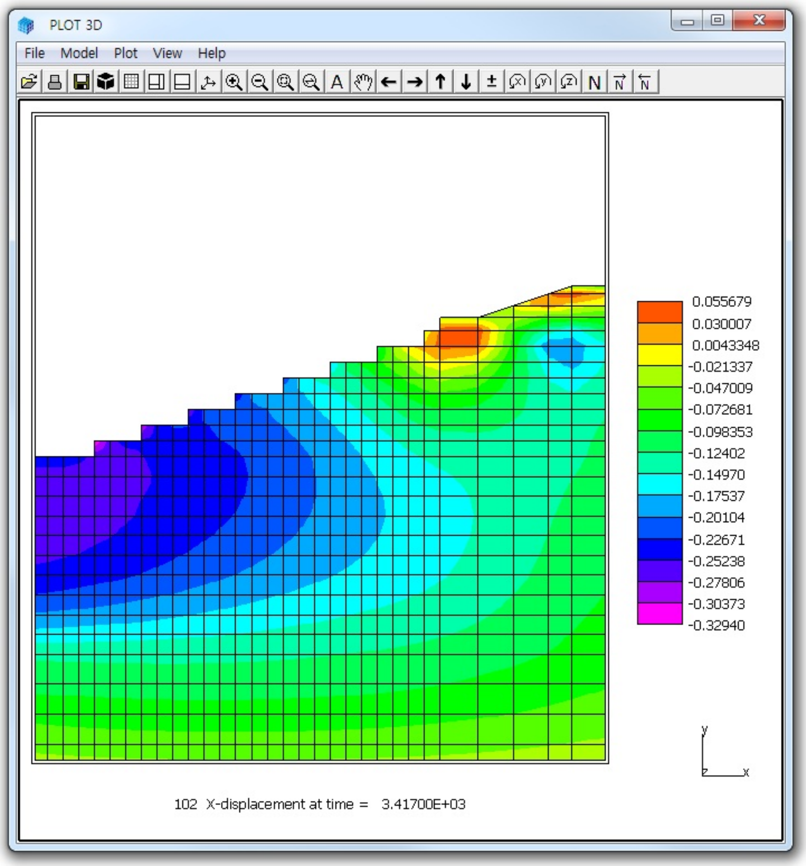


Figure 4.73 Horizontal displacement distribution

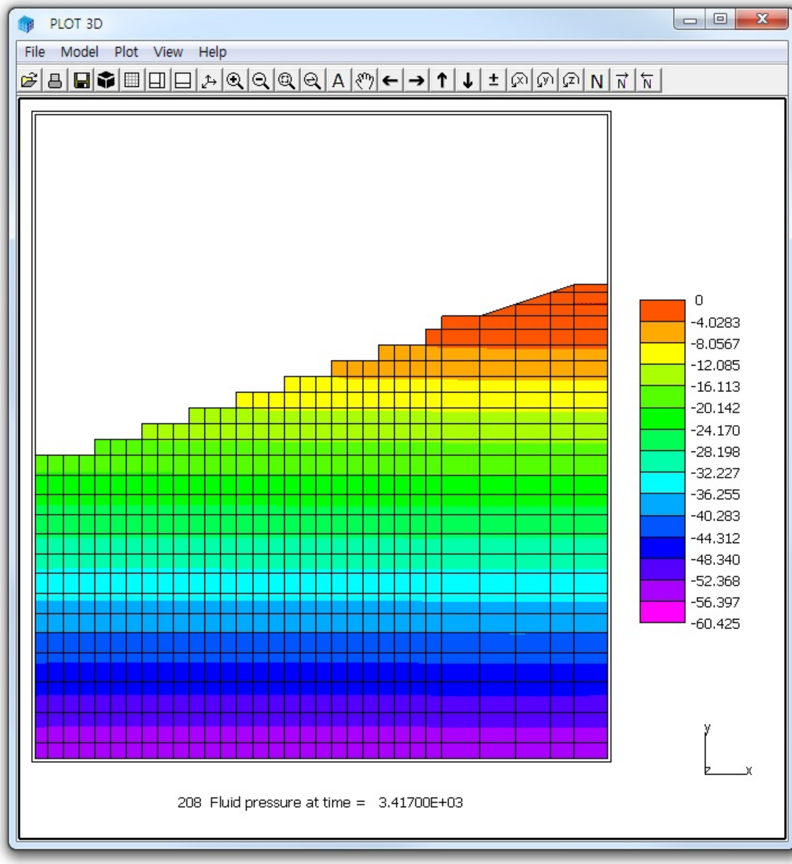


Figure 4.74 Pore pressure distribution

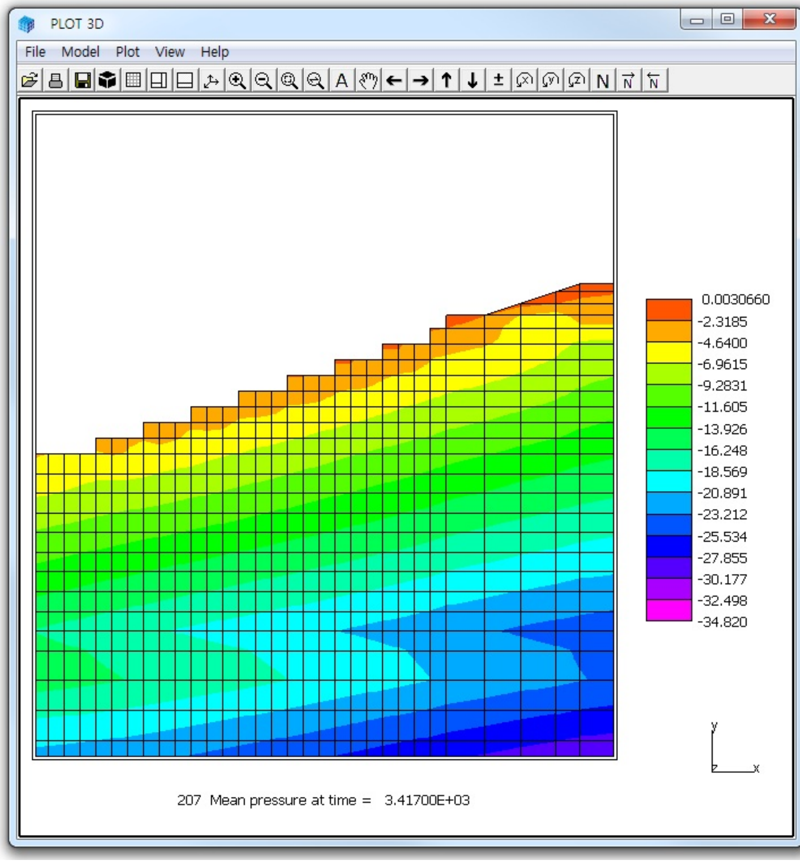


Figure 4.75 Effective mean pressure distribution

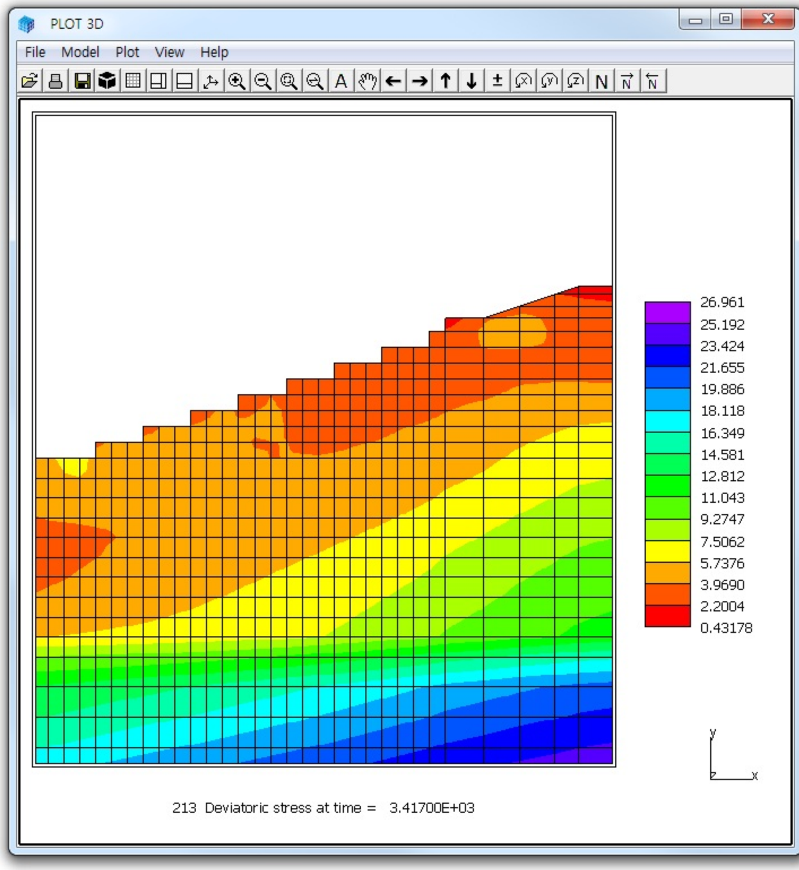


Figure 4.76 Deviatoric stress distribution

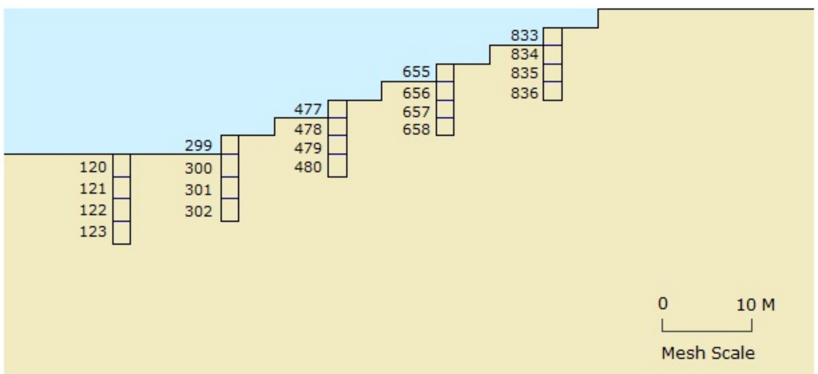


Figure 4.77 Element locations for time history plot

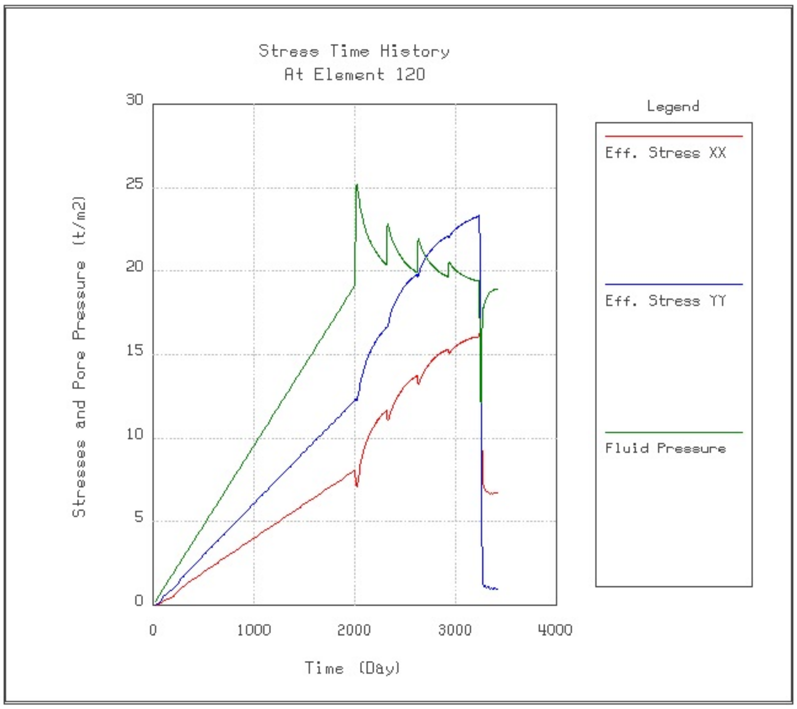


Figure 4.78 Stress time history at element 120

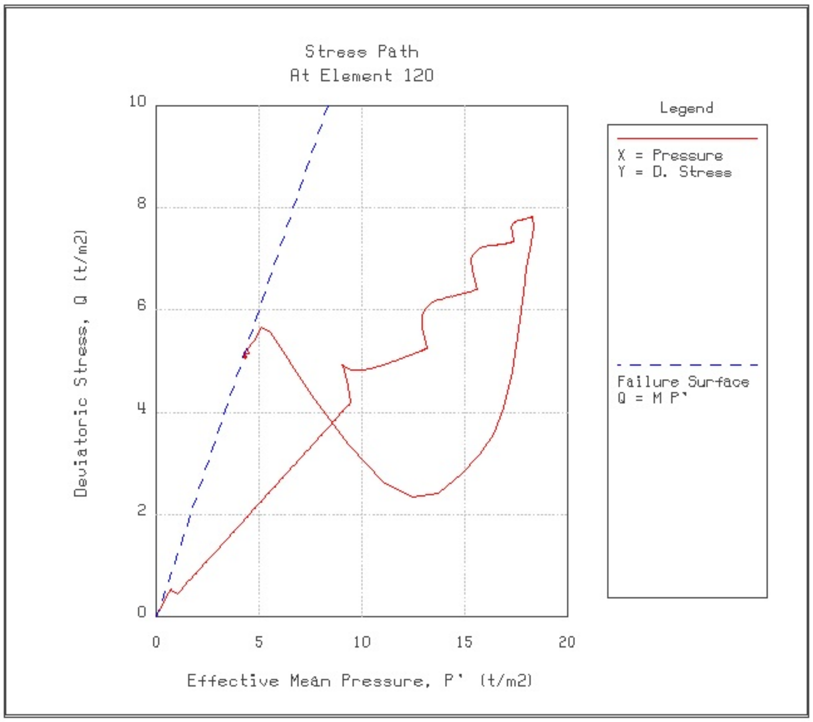


Figure 4.79 Stress path at element 120

4.20 Seismic Tunnel Analysis

This example problem is to analyze a typical NATM tunnel subjected to earthquake loading. The tunnel is located about 22 meters below ground surface as shown in Figure 4.80. Figure 4.81 shows detailed tunnel cross section. Material properties are listed in Table 4.2.

This example problem consists of static and dynamic analyses for the typical horseshoe tunnel constructed by NATM method.

The static part (Steps 1 thru 9) of the analyses as shown in Figure 4.82 is the same as the example problem 2 in TUNA Plus User's Manual except the followings:

- Top core excavation followed by lower core excavation.
- Lining modeled by Shell element with plain concrete.

The dynamic part starting from Step 10 as in Figure 4.83 is performed by applying following boundary conditions and base acceleration:

- Left and right sides of boundary are horizontal roller and bottom of mesh is fixed.
- As horizontal base acceleration, N-S component of the El Centro earthquake is applied with scaled maximum acceleration of 0.2g.

Figure 4.84 shows key location selected for displacement time history plot. Numbers shown in the figure represent node numbers. Figure 4.85 thru 4.87 show finite element meshes used for the analysis.

Figure 4.88 shows tunnel deformed shape at 5 seconds after the onset of earthquake loading. Figures 4.89 and 4.90 show top and bottom surface extreme fiber stresses at 5 seconds after onset of earthquake loading.

The graphical outputs of inner (bottom) and outer (top) extreme fiber stresses of the lining show the maximum compressive stress of 119.9 t/m^2 and the maximum tensile stress of 31.88 t/m^2 at 5 seconds after onset of earthquake loading. Such maximum extreme fiber stresses are far below the strength of the typical plain concrete.

Figure 4.91 shows ground surface horizontal displacement time histories at selected locations: Nodes 609, 837, and 2020. As it can be seen, horizontal ground surface displacements are influenced very little due to the presence of the tunnel.

Figures 4.92 and 4.93 show springline horizontal displacement time histories at the right and left sides of the tunnel, respectively. Each figure shows two adjacent nodes: inner and outer nodes which are separated by interface element as shown in Figures 4.84 and 4.87.

Compared with ground surface, displacements at tunnel springlines are much less amplified. Overall, tunnel lining is moving with the surrounding rock mass but the outgoing lining displacements are limited to the adjacent rock mass displacements. In other words, at those locations where lining is in contact with the adjacent rock mass, the outgoing lining displacements do not exceed the rock mass displacements.

Table 4.2 Material property

Material Type	γ (t/m³)	K_o	E (t/m²)	ν	φ deg.	C (t/m²)	T (t/m²)
Weathered Soil	1.90	0.50	2.00×10^3	0.33	30	3	20
Weathered Rock	1.90	0.43	5.000×10^3	0.30	35	30	30
Soft Rock	2.40	0.33	2.00×10^4	0.25	40	70	40
Hard Rock	2.55	0.25	2.00×10^5	0.20	45	100	50
Shotcrete (Soft)	2.40		0.50×10^6	0.20	30	500	100
Shotcrete (Hard)	2.40		1.50×10^6	0.20	30	500	100
Rock Bolt			2.10×10^7				
Reinforced Concrete Lining	2.50		2.10×10^6	0.20	30	500	300
Reinforcing Bar			2.10×10^7	0.20			
Interface Joint			2.00×10^5		5	0.001	0.02

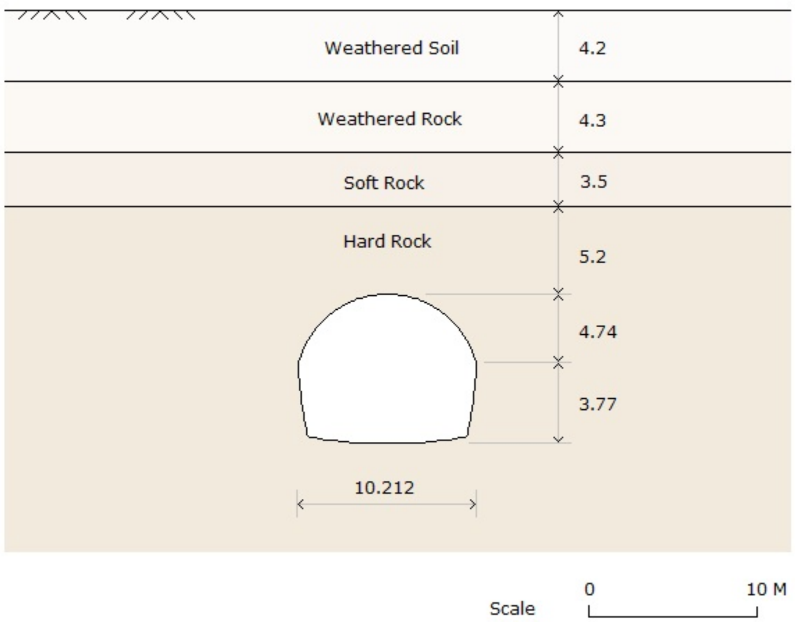
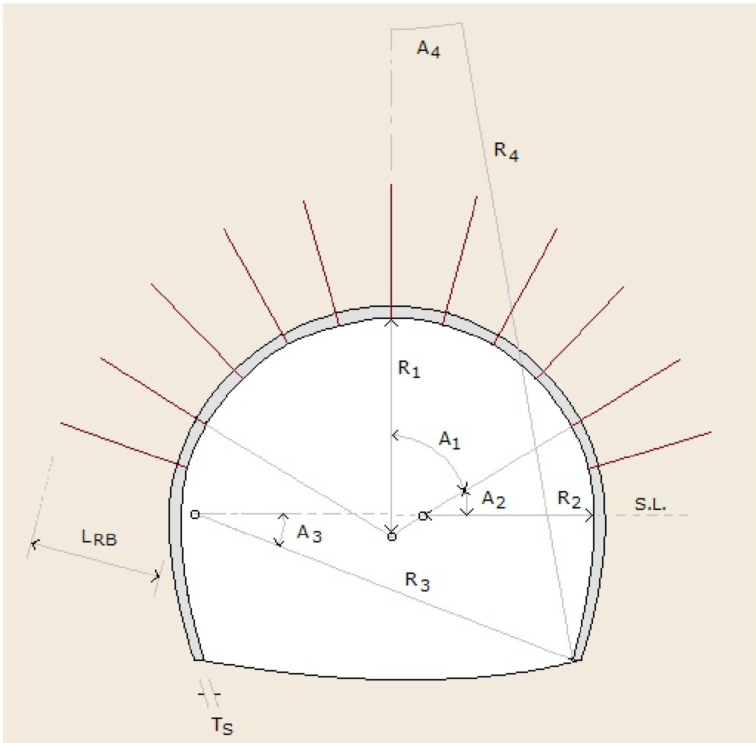


Figure 4.80 Geological profile



R_1	=	7.24 m	A_1	=	60 deg.
R_2	=	6.24 m	A_2	=	30 deg.
R_3	=	11.86 m	A_3	=	21.781 deg.
R_4	=	25.86 m			

Number of Rock Bolts (NUMRB)	=	11
Length of Rock Bolts (LRB)	=	3.0 m
Spacing of Rock Bolts (TSPACING)	=	1.2 m
Thickness of Shotcrete (TS)	=	20 cm
Thickness of Liner (TL)	=	40 cm

Figure 4.81 Tunnel cross section

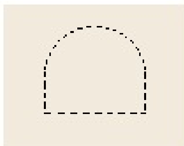
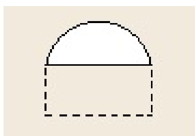
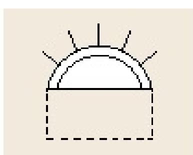
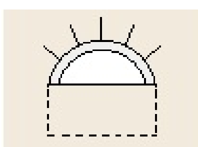
<table>
<tr><th>Step</th><th>Construction State</th><th colspan="2">Descriptions</th></tr>
<tr><td>1,2</td><td></td><td colspan="2">In Situ K_o State</td></tr>
<tr><td>3</td><td></td><td>50 % Stress Relief</td><td rowspan="3">Upper Core
Excavation</td></tr>
<tr><td>4</td><td></td><td>75 % Stress Relief
Soft Shotcrete
Rock Bolt</td></tr>
<tr><td>5</td><td></td><td>100 % Stress Relief
Hard Shotcrete
Rock Bolt</td></tr>
</table>

Figure 4.82 Construction sequence, static part

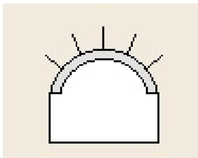
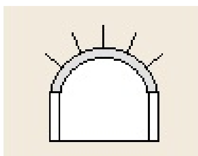
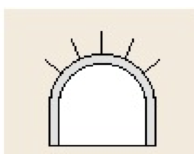
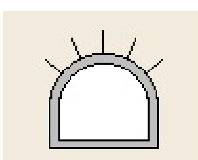
<table>
<tr><th>Step</th><th>Construction State</th><th colspan="2">Descriptions</th></tr>
<tr><td>6</td><td></td><td>50% Stress Relief</td><td rowspan="3">Lower Core Excavation</td></tr>
<tr><td>7</td><td></td><td>75% Stress Relief
Soft Shotcrete</td></tr>
<tr><td>8</td><td></td><td>100% Stress Relief
Hard Shotcrete</td></tr>
<tr><td>9</td><td></td><td colspan="2">Lining Subjected to: Weight</td></tr>
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Figure 4.82 Construction sequence, static part (Continued)

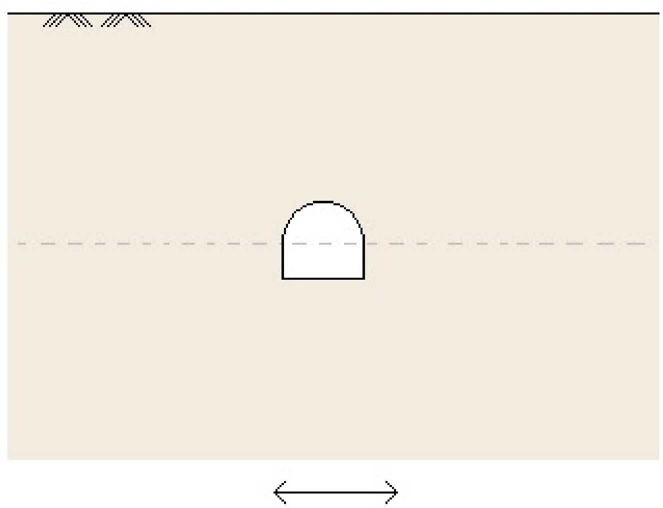


Figure 4.83 Seismic load, dynamic part

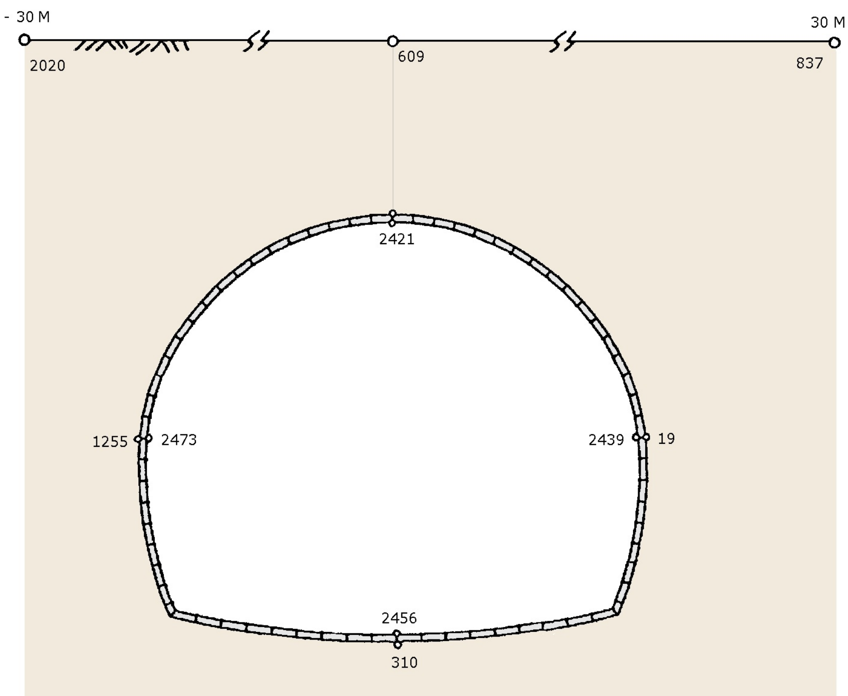


Figure 4.84 Key locations for displacement time history plot

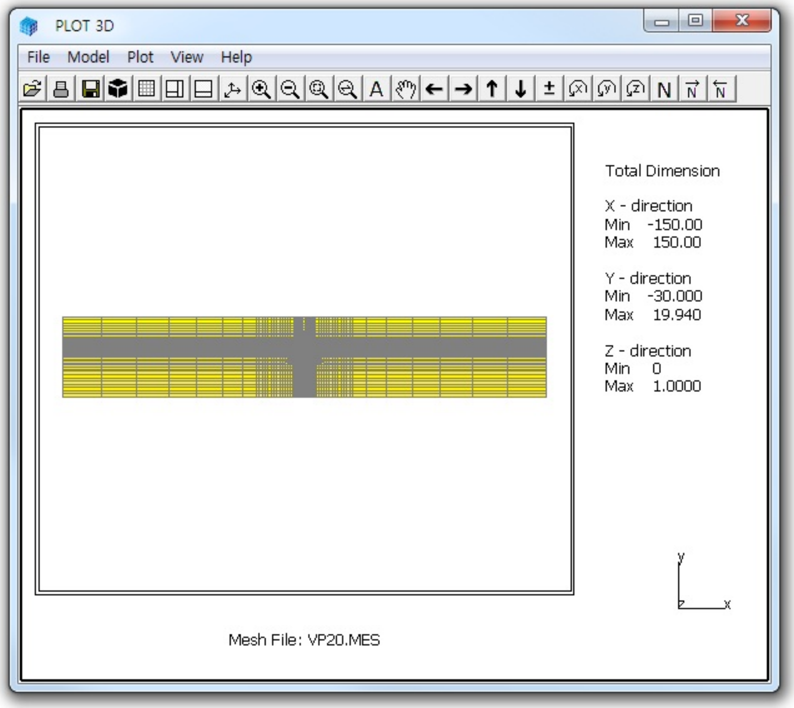


Figure 4.85 Finite element mesh

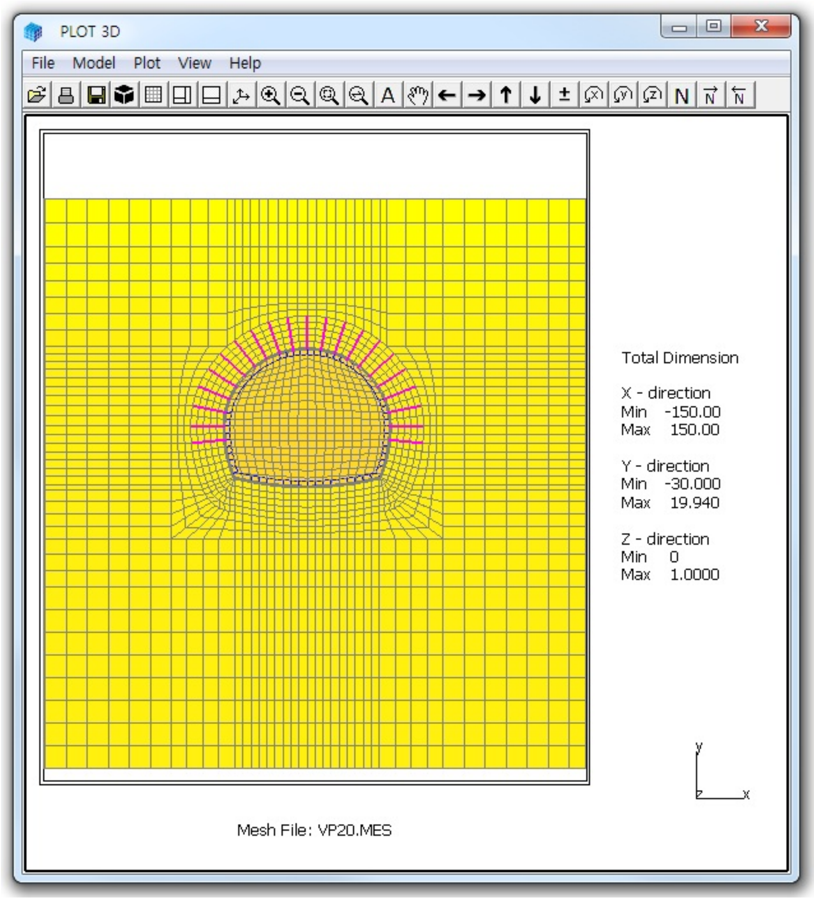


Figure 4.86 Finite element mesh around tunnel

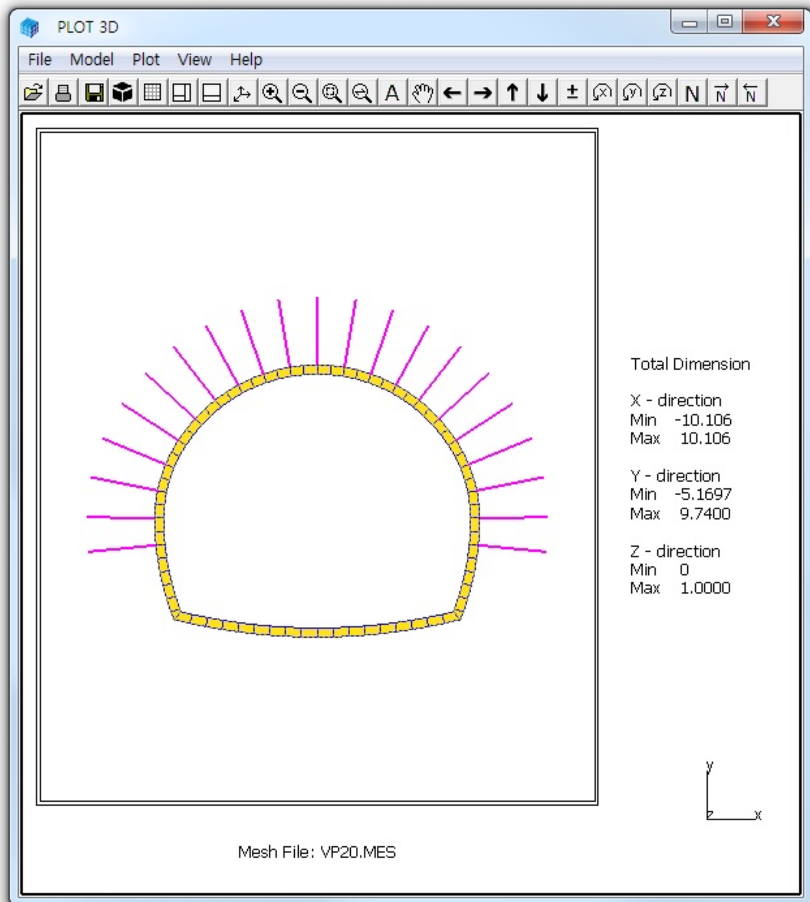


Figure 4.87 Interface between shotcrete and lining

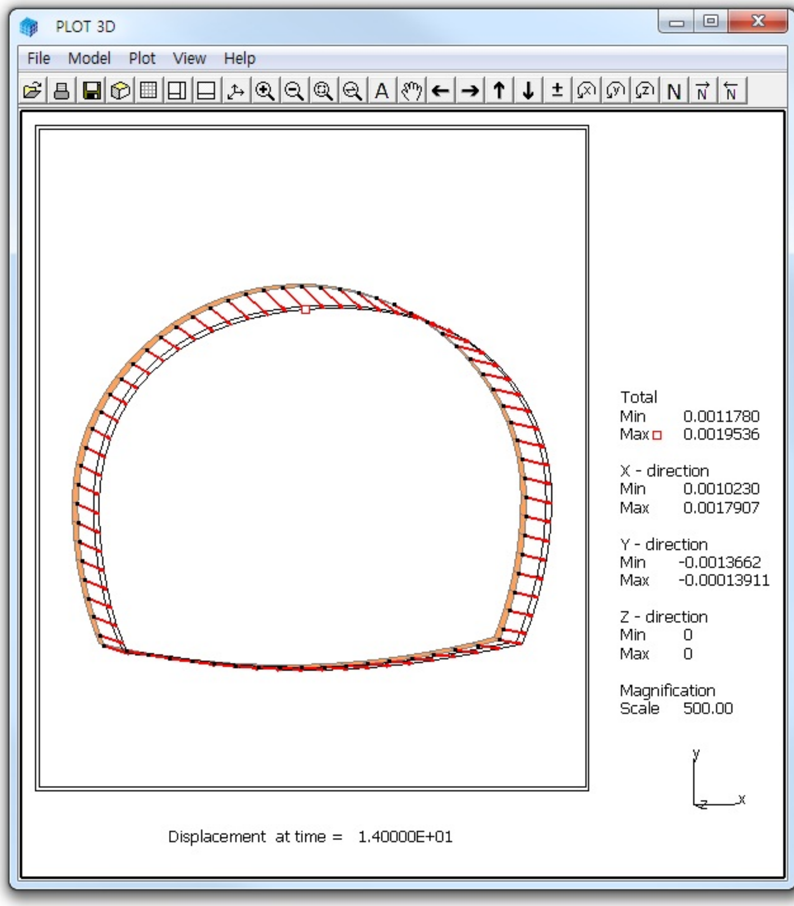


Figure 4.88 Tunnel deformed shape at t=14 sec

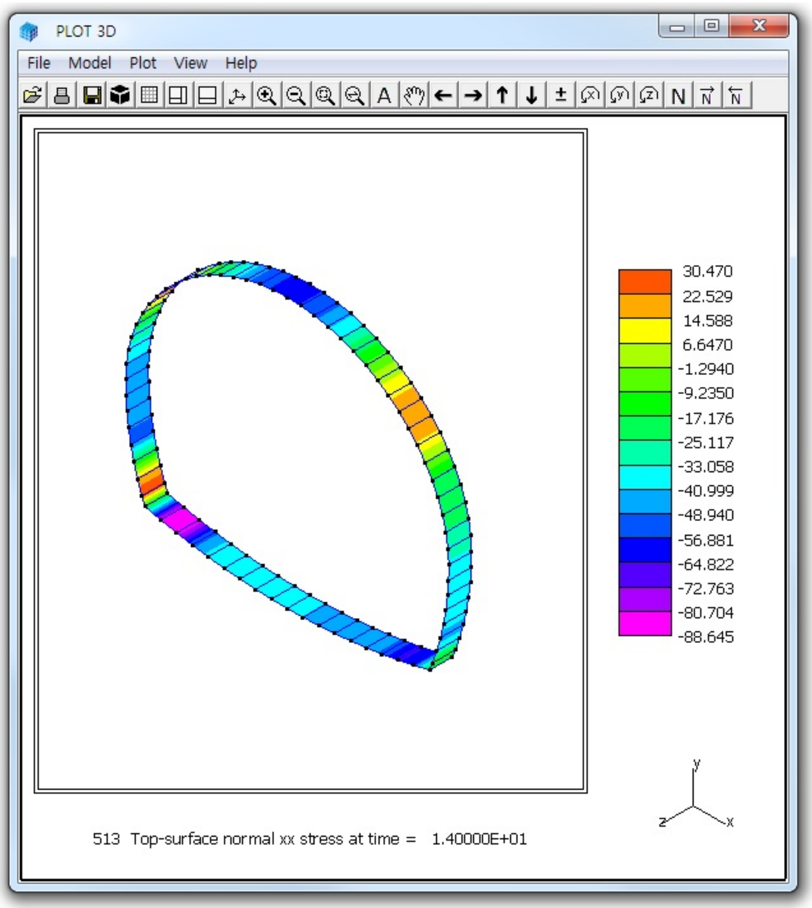


Figure 4.89 Top surface extreme fiber stress at t=14 sec

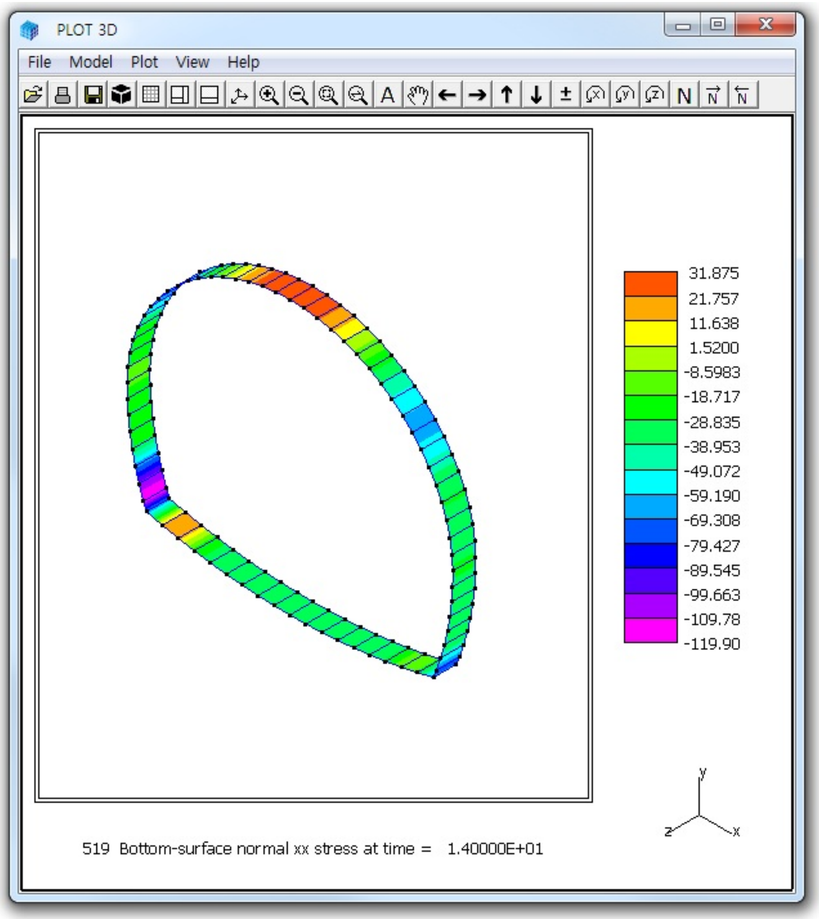


Figure 4.90 Bottom surface extreme fiber stress at t=14 sec

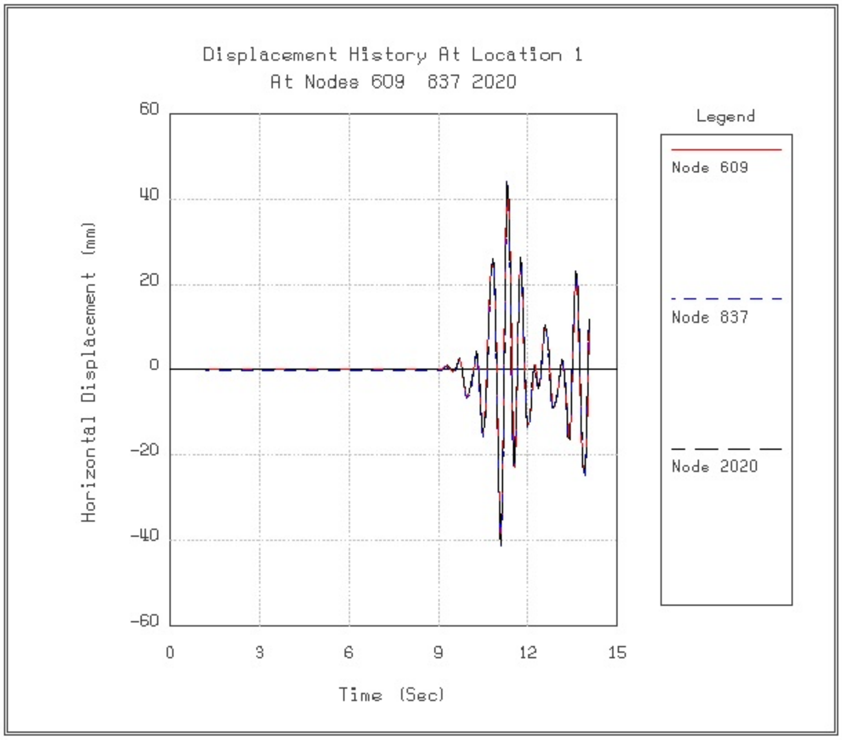


Figure 4.91 Horizontal displacement at nodes 609, 837, and 2020

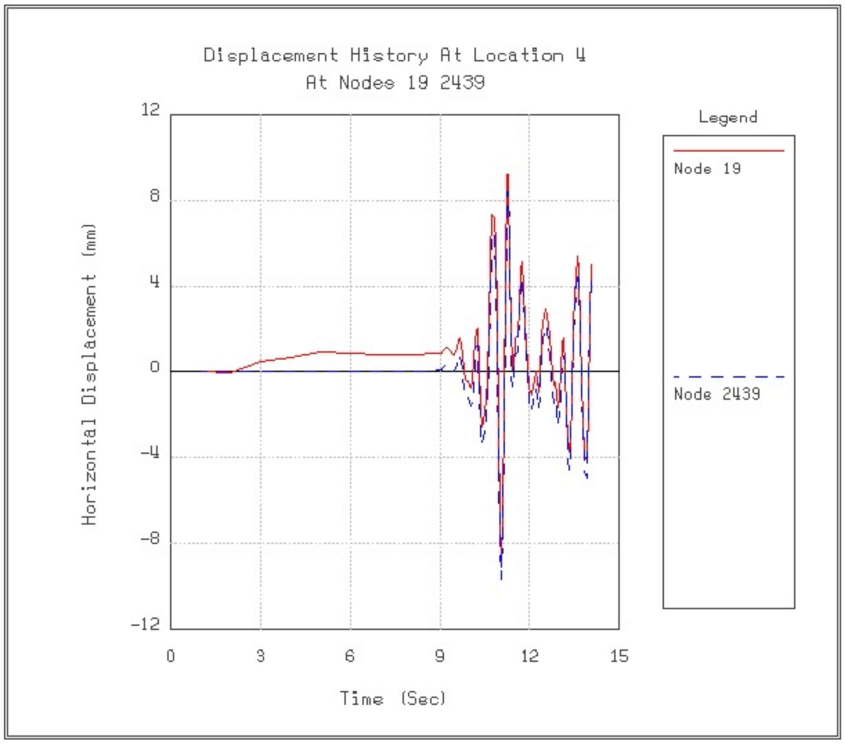


Figure 4.92 Horizontal displacement at nodes 19 and 2439

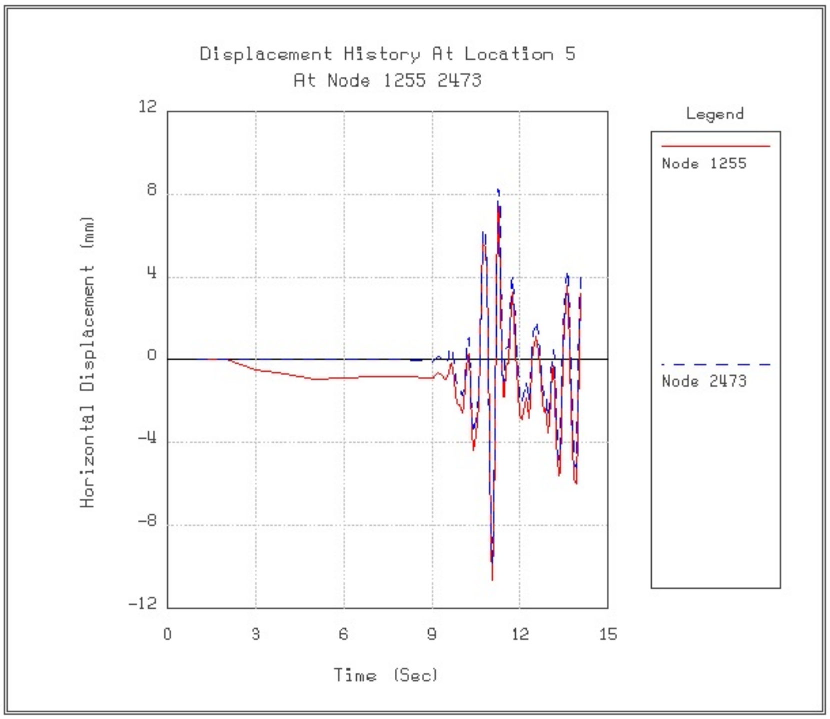


Figure 4.93 Horizontal displacement at nodes 1255 and 2473

4.21 Frames with Hinge Connection

This example problem is to solve symmetric plane frame members subjected to a vertical concentrated load at the hinge connecting both frames as shown is Figure 4.94.

The exact solutions for this frame structures without shear deformation are given below:

$$\delta = \frac{P}{EA/L + 3EI/L^3} \qquad M_{max} = \frac{PL/\sqrt{2}}{1 + AL^2/3I}$$

where

- δ Maximum deflection at the center
- M_{max} Maximum moment at fixed end

Two SMAP-3D calculations are performed using the geometrical and material parameters listed in Figure 4.94.

Frames modeled by 10 beam elements:

- Figure 4.95 Beam element with material number
- Figure 4.96 Beam deformed shape
- Figure 4.97 Beam bending moment diagram

Frames modeled by 40 shell elements:

- Figure 4.98 Shell element with material number
- Figure 4.99 Shell deformed shape
- Figure 4.100 Shell bending moment diagram

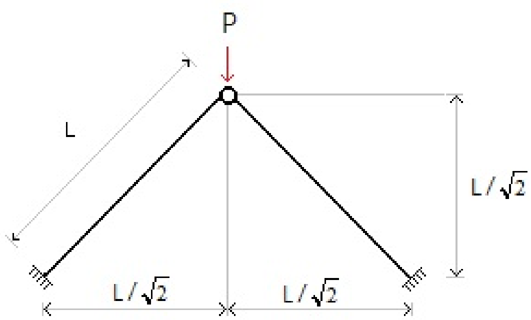
SMAP-3D results show good agreement with the exact solutions.

Maximum deflection at the center (δ)

- Exact solution = 0.01768 cm
- SMAP-3D (Beam) = 0.01767 cm
- SMAP-3D (Shell) = 0.01767 cm

Maximum moment at fixed end (M_{max})

- Exact solution = 0.1000 t-m
- SMAP-3D (Beam) = 0.1000 t-m
- SMAP-3D (Shell) = 0.1003 t-m



P	= 100 t	L	= 7.071 m
E	= 20×10^{6} t/m^2	v	= 0.0
A	= 0.2 m^2	I	= 0.000667 m^4

Figure 4.94 Frames with hinge connection

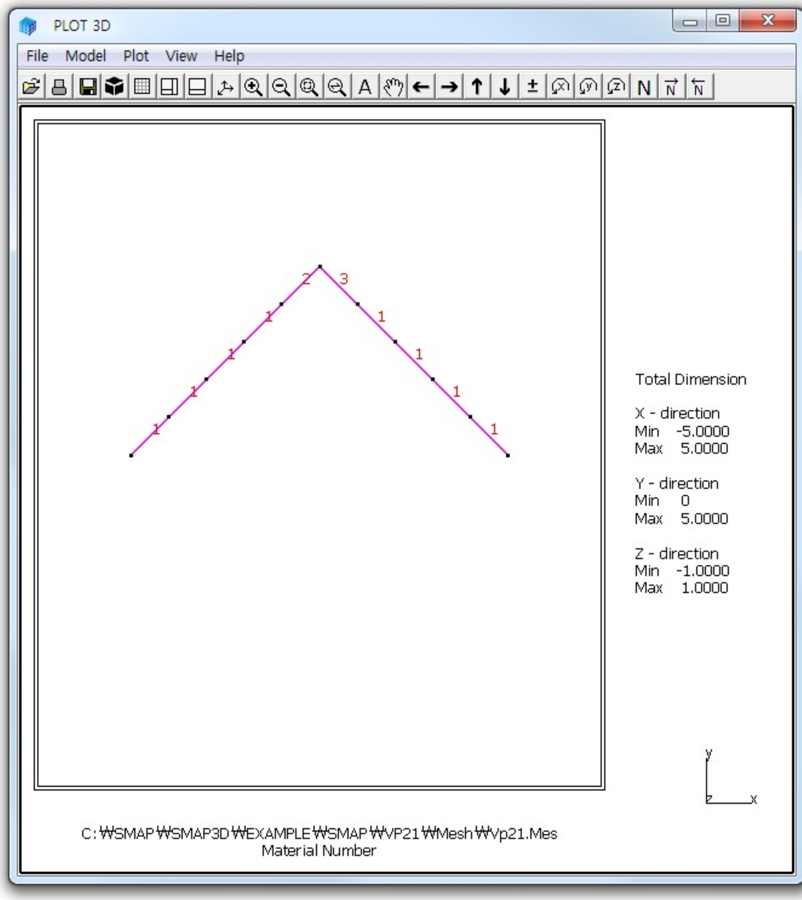


Figure 4.95 Beam element with material number

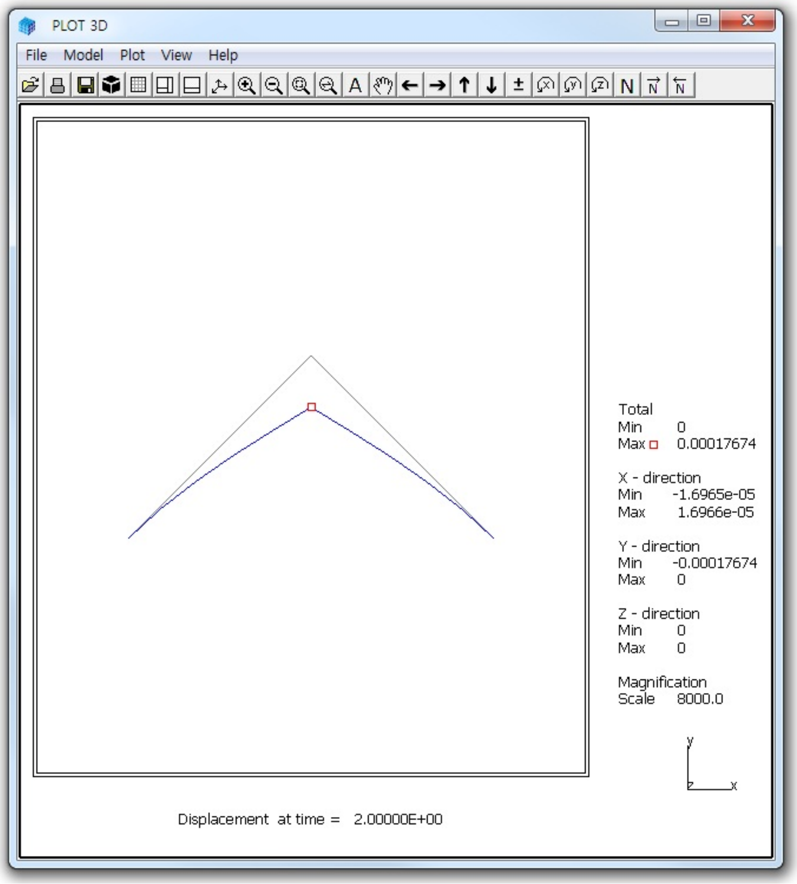


Figure 4.96 Beam deformed shape

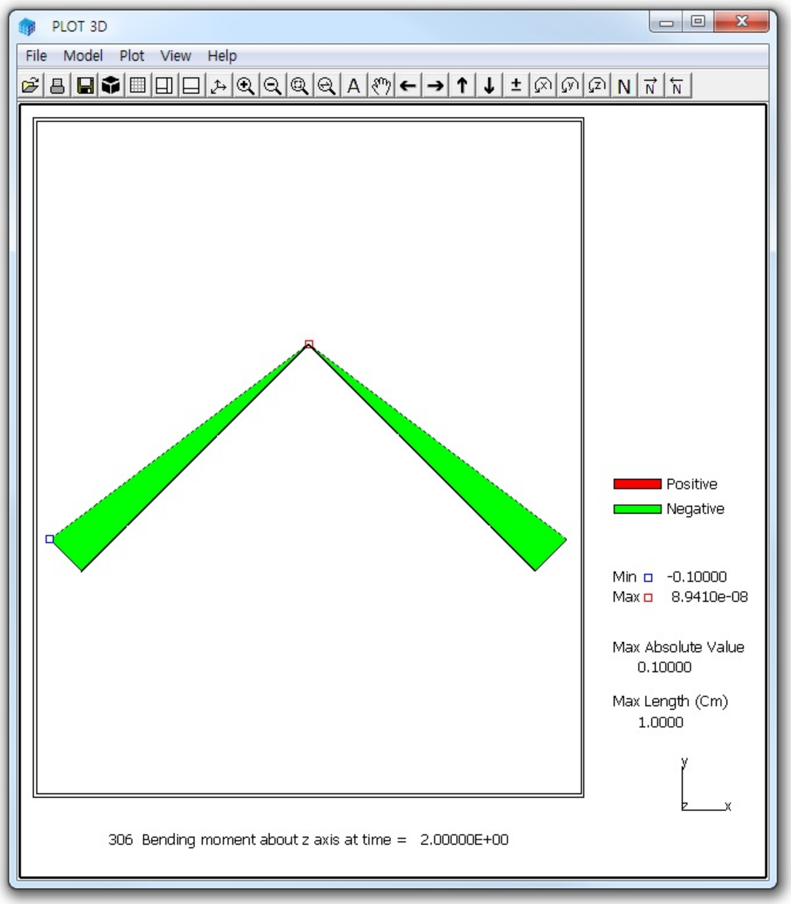


Figure 4.97 Beam bending moment diagram

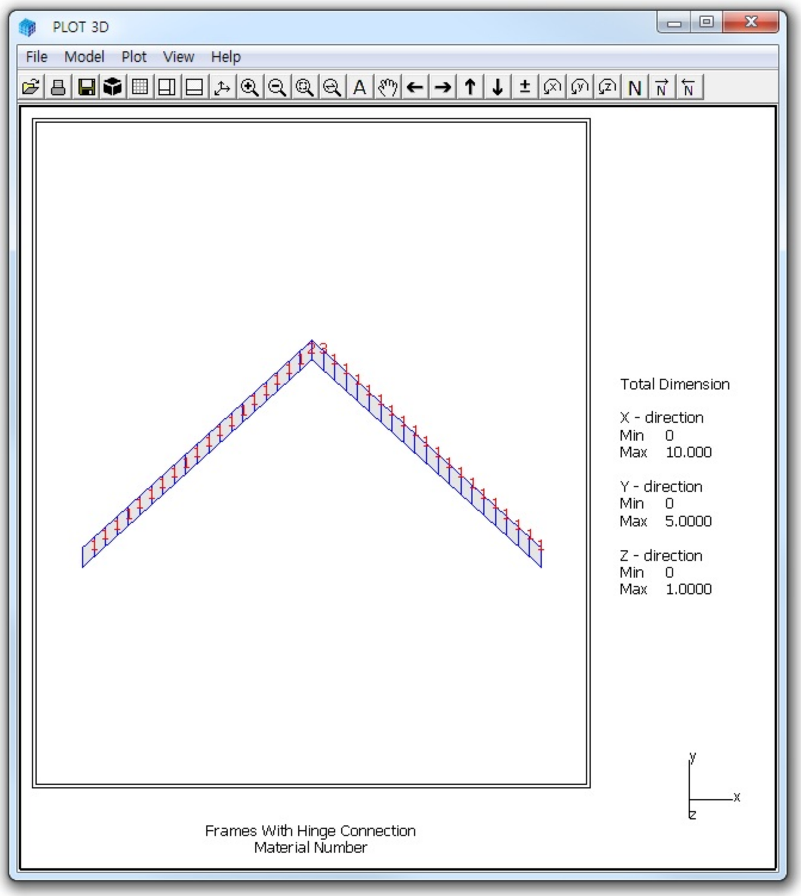


Figure 4.98 Shell element with material number

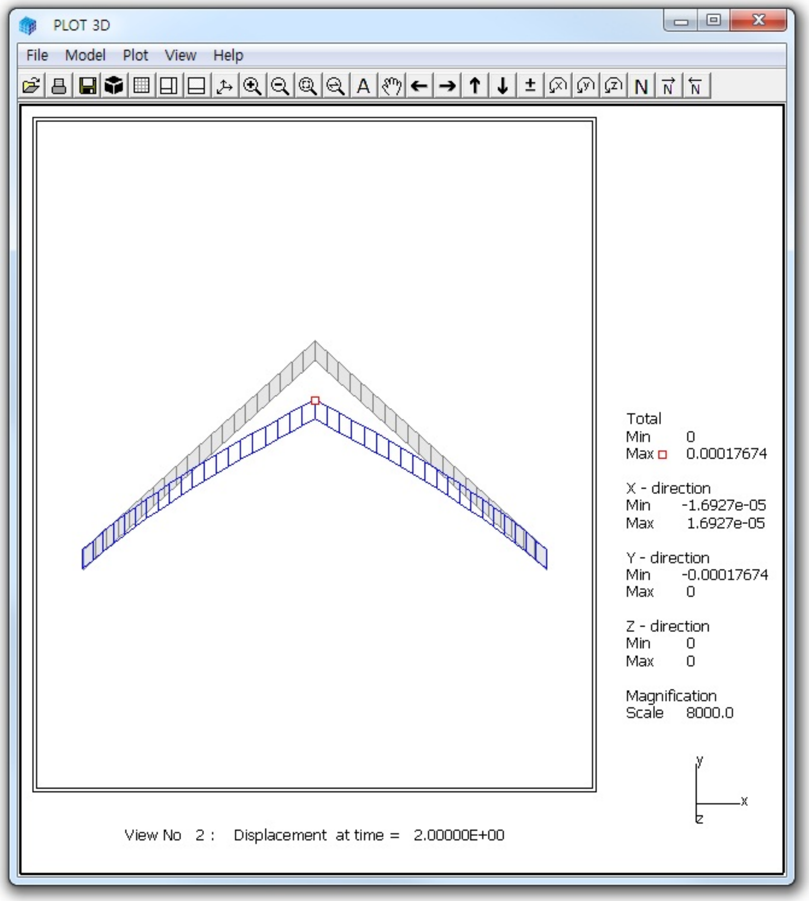


Figure 4.99 Shell deformed shape

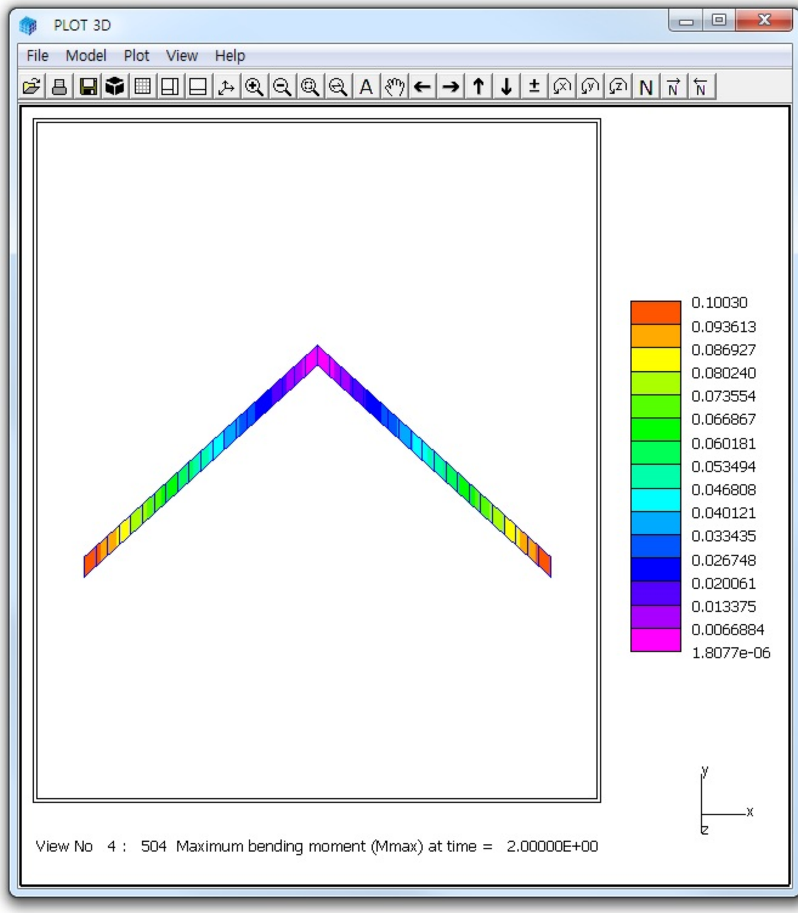


Figure 4.100 Shell bending moment diagram

4.22 Embedded Rebars with Slip

This example problem is to verify the implementation of the embedded reinforcing bars (rebars) with interface shear (slip) between rebars and surrounding concrete. Figure 4.101 shows a simply supported reinforced concrete beam subjected to a concentrated load at midspan. To simplify the problem, it was assumed that both reinforcing bars and concrete are linearly elastic while the interface shear is elastic - perfectly plastic with a limiting constant cohesion.

The exact beam solution without shear deformation is given below:

Maximum deflection at the center without rebars,

$$\delta = \frac{P \cdot L^3}{48 \ E_c \cdot I_c} = 1.190 \text{ Cm}$$

Maximum deflection at the center with rebars,

$$\delta = \frac{P \cdot L^3}{48 \ E_c \cdot I_t} = 1.040 \text{ Cm}$$

By symmetry, only left half of the beam is modeled using 60 continuum elements for concrete and 2 embedded truss elements for reinforcing bars as shown in Figure 4.102. It should be noted that the end points of embedded truss elements do not belong to the corner nodes of continuum elements.

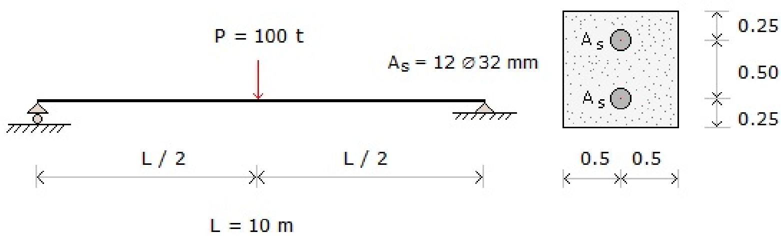
The computed center deflections are compared with the exact beam solution as shown in Table 4.3. SMAP-3D results approach to the upper bound beam solution at lower cohesion and the lower bound beam solution at higher cohesion. At the intermediate cohesion, however, the computed deflection is in between upper and lower bound beam solutions, indicating some resistance from slip strength.

Figures 4.103 and 4.104 show the deformed shape and the axial stress distribution, respectively, from SMAP-3D result at the intermediate cohesion of 5 t/m^2.

Table 4.3 Computed center deflections

Cmax (t/m^2)	SMAP-3D Result	Exact Beam Solution
0.1	1.1746 Cm	1.190 Cm (without rebar)
5.0	1.0990 Cm	
280	1.0379 Cm	1.040 Cm (with rebar)

Cmax : Interface Cohesion



E_c = 2.1×10^6 t/m^2
v_c = 0.2
E_s = 2.1×10^7 t/m^2
I_c = 0.0833 m^4
I_s = $2\,(E_s / E_c)\,A_s\,(0.25)^2 =$ 0.012063 m^4
I_t = $I_c + I_s =$ 0.0954 m^4

<u>Property of interface between rebar and concrete</u>

G = 0.875×10^6 t/m^2 (shear modulus)
r_b = 0.016 m (radius of rebar)
t_g = 0.002 m (thickness of interface)
K_{se} = $G / (\,(r_b + t_g)\ln(1 + t_g / r_b)\,) = K_s$(each)
K_s = 12 K_s(each) = 4.956×10^9 t/m^3

Figure 4.101 Embedded rebars with slip

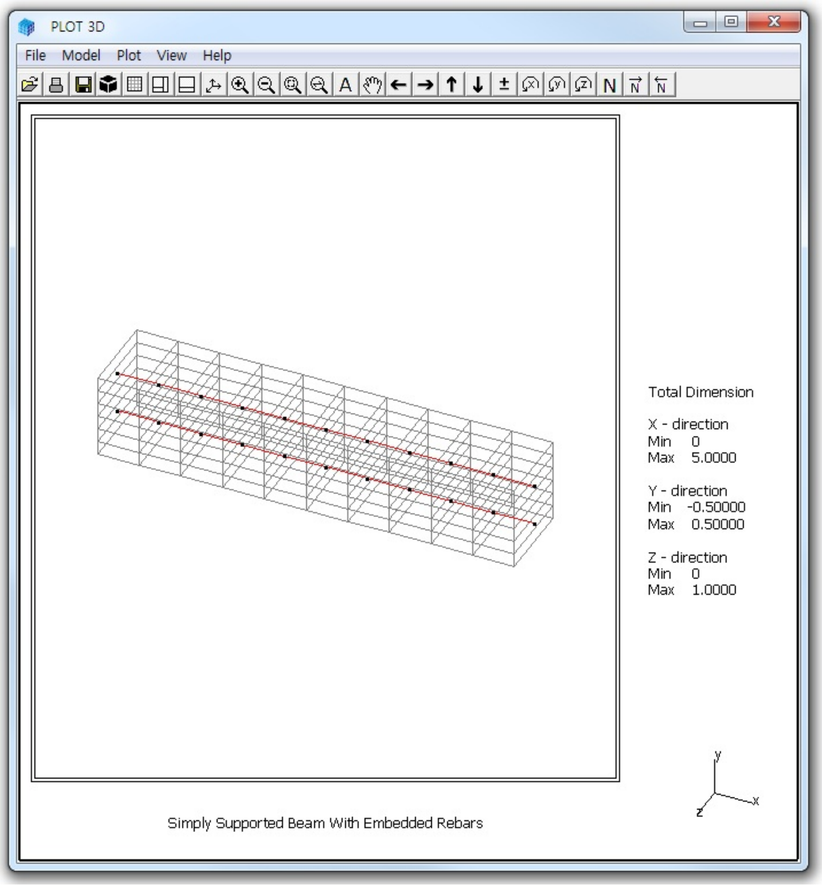


Figure 4.102 Finite element mesh

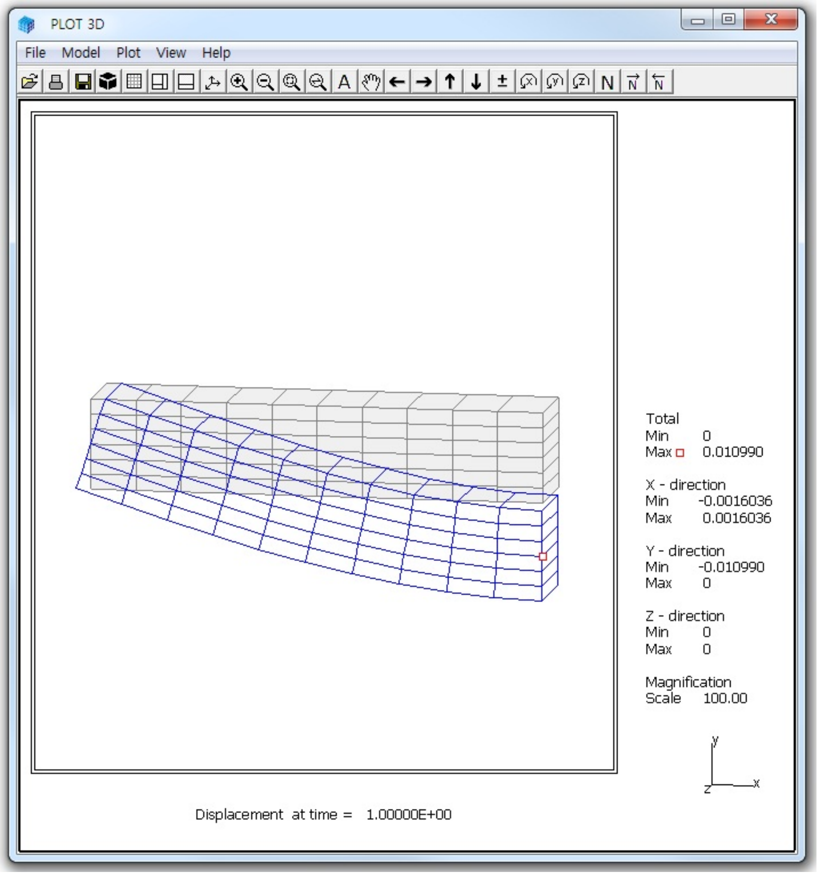


Figure 4.103 Deformed shape

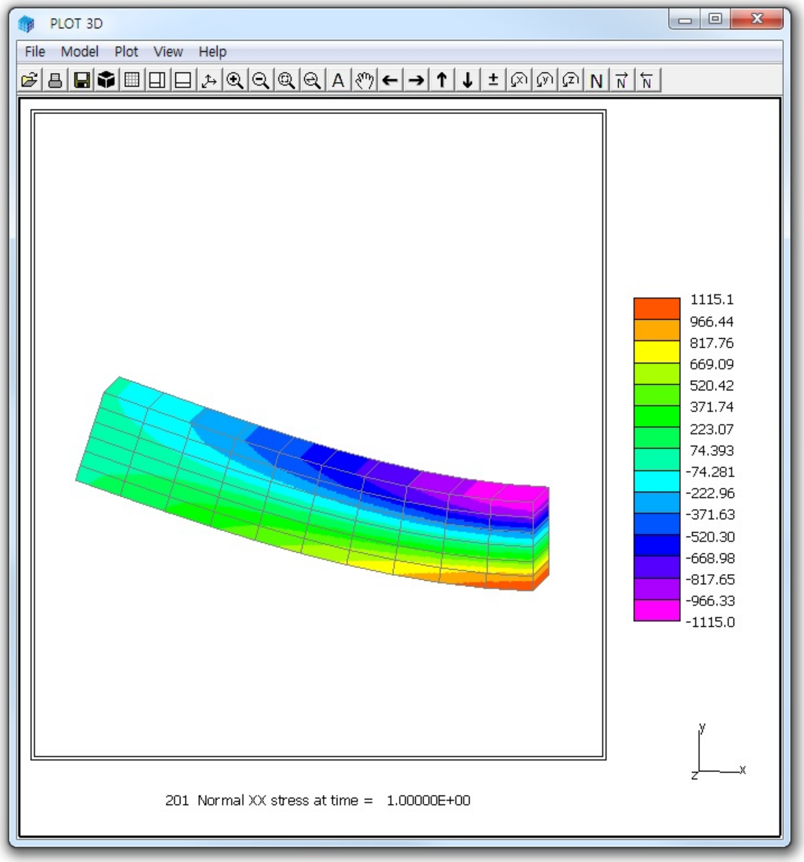


Figure 4.104 Axial stress distribution

4.23 Pseudo-Dynamic Embankment Fill Analysis

This example problem is to solve the response of an embankment fill subjected to pseudo-dynamic earthquake load as schematically shown in Figure 4.105.

As listed in Table 4.4, the sequence of construction consists of 5 steps. The first two steps are used to compute in situ Ko state with water table at GL-25. At step 3, water table is raised up to GL-5. At step 4, embankment fill is completed. At final step 5, pseudo-dynamic earthquake load is applied to the embankment fill.

Material properties are listed in Table 4.5.

The change of water table is modeled by adding Intensity times Distribution Factor to the Y component of unit gravity load (FRY). Intensity history number and distribution factor are specified in Card Group 9.1.2.

The pseudo-dynamic earthquake load is modeled by adding Intensity times Distribution Factor to the X component of unit gravity load (FRX).

Figure 4.106 shows the finite element mesh used for the analysis. Figures 4.107 and 108 show deformed shape and vertical stress distribution, respectively, at final step 5 where pseudo-dynamic earthquake load is applied to the embankment fill.

Computed vertical stress at GL-23 is reduced by 18 t/m^2 due to the water table at GL-5. The reduction of vertical stress is associated with the water head of 18 m at GL-23.

Horizontal displacement of 1.16 Cm is obtained at the top surface of embankment fill due to the pseudo dynamic load. Exact solution for this problem is not available. However, SMAP-S2 and SMAP-2D analyses show the same results.

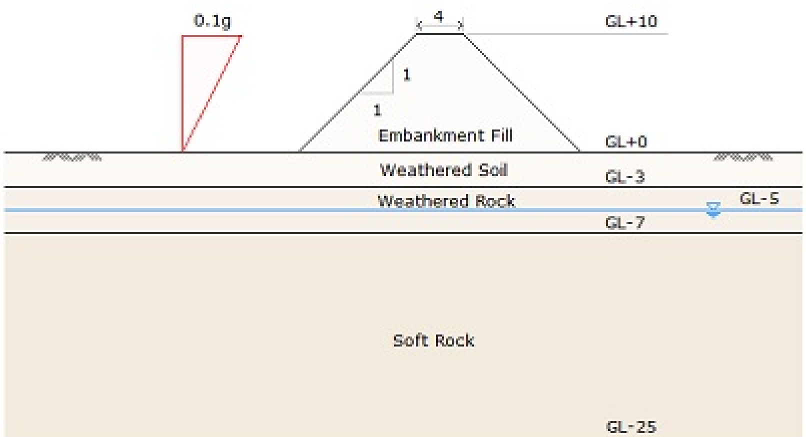


Figure 4.105 Schematic section view of embankment

Table 4.4 Construction sequence

Step	Description
1, 2	In Situ Ko state with water table at GL-25
3	In Situ Ko state with water table at GL-5
4	Completion of embankment fill
5	Embankment fill subjected to pseudo-dynamic load

Table 4.5 Material property

Material Type	γ (t/m^3)	K_o	E (t/m^2)	v	φ deg.	C (t/m^2)	T (t/m^2)
Weathered Soil	1.90	0.50	2.0×10^3	0.33	30	3	20
Weathered Rock	1.90	0.43	5.0×10^3	0.30	35	30	30
Soft Rock	2.40	0.33	2.0×10^4	0.25	40	70	40
Embankment Fill	2.00	0.50	3.0×10^3	0.33	30	3	20

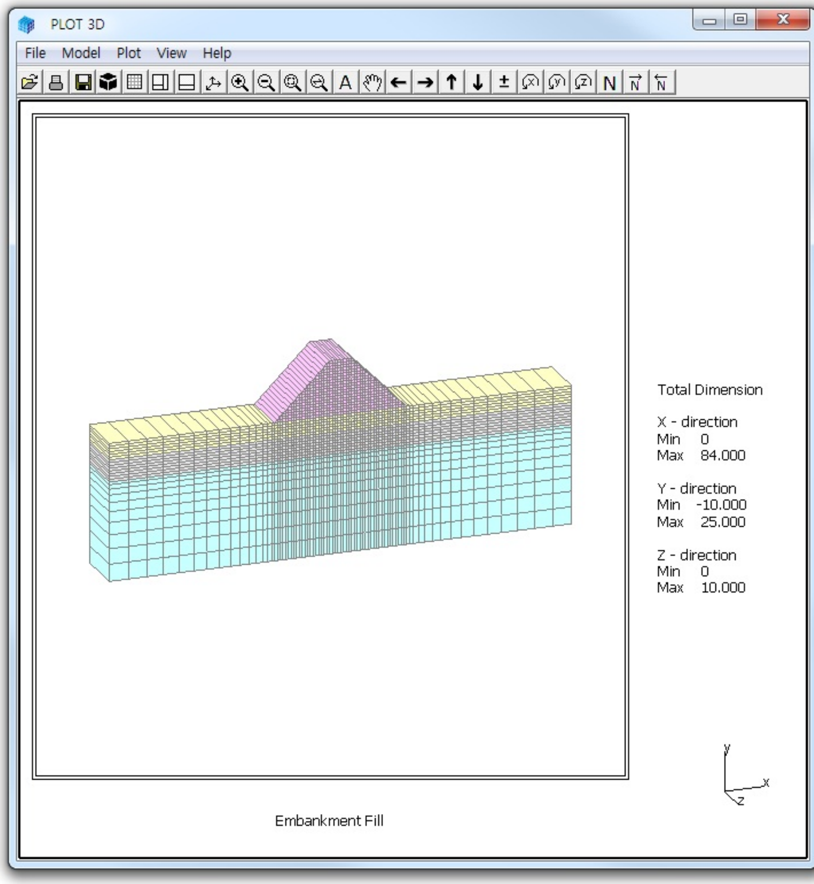


Figure 4.106 Finite element mesh

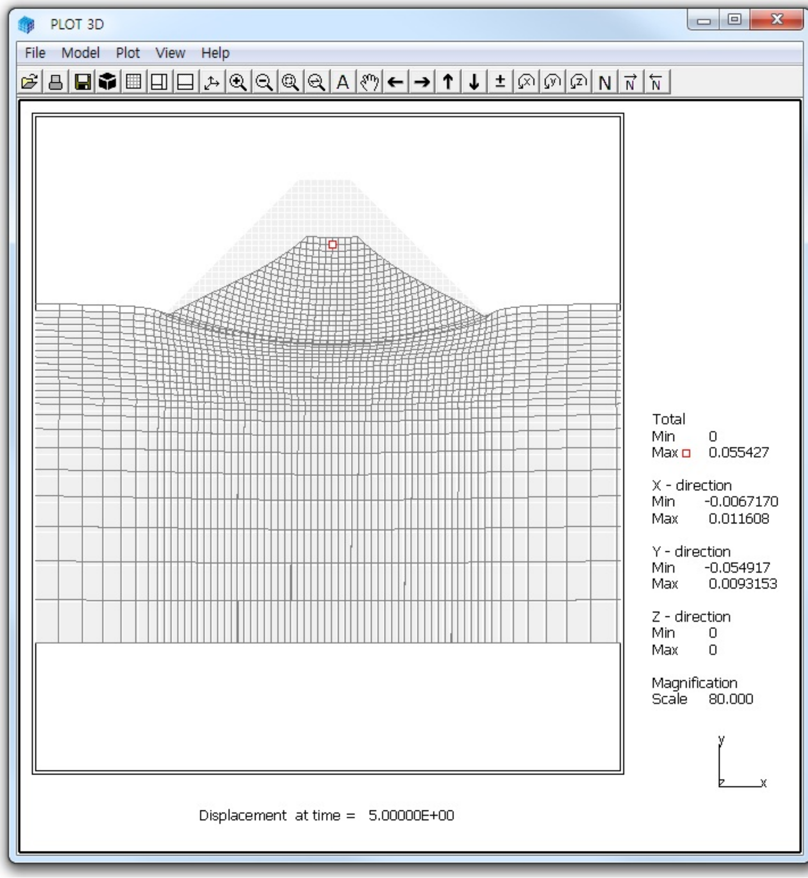


Figure 4.107 Deformed shape

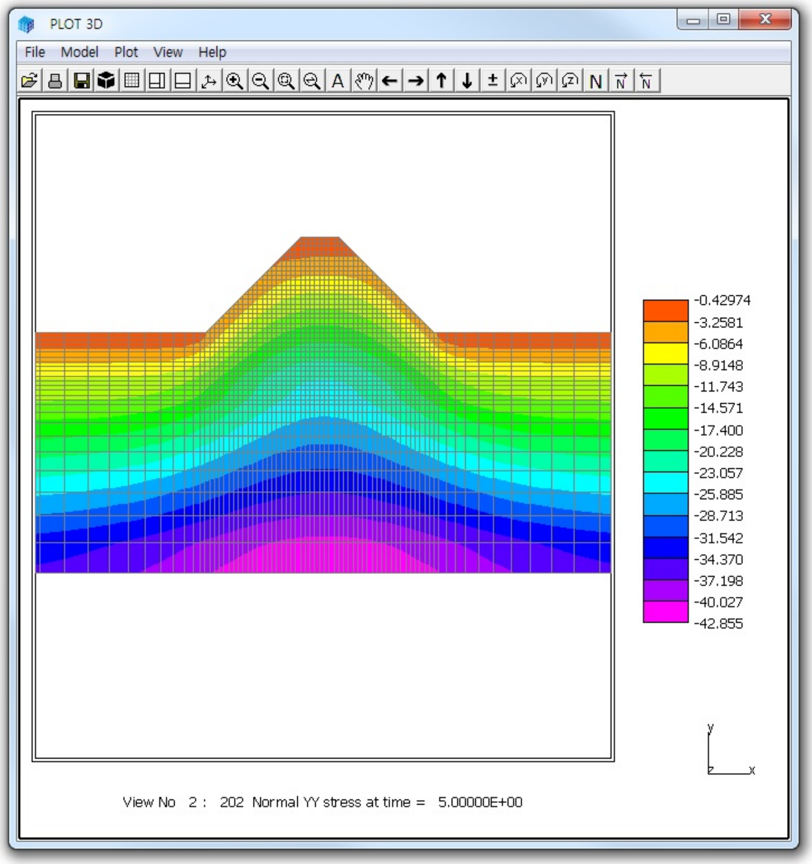


Figure 4.108 Vertical stress distribution

4.24 Plane Strain Tunnel in Jointed Continuum

This example problem is to verify the jointed continuum mesh generated by JOINT-3D pre-processing program. Jointed continuum analysis is similar to the discrete element analysis. For the jointed continuum analysis, each continuum finite element is surrounded by joint elements.

The main advantages of using such joint elements are to allow slippage along the joint when reaching shear strength and debonding normal to joint face when exceeding tensile strength.

This example is identical to the Example Problem 14 except that the tunnel is located in the jointed continuum. The jointed continuum mesh is generated by JOINT-3D program with the input file Joint.inp. Refer to JOINT-3D User's Manual.

Figure 4.109 shows the finite element mesh consisting of the jointed continuum around tunnel.

To compare with continuum model (Example Problem 14), two analyses are performed with Elastic and Plastic Joint Models.
The Elastic Joint Model assumes strong joint properties so that it essentially represents continuum model. The Plastic Joint Model assumes lower shear and tensile strengths so that it allows slippage and debonding along the joints.

Results are listed in the following order:

- Figure 4.110 Deformed shape for Elastic Joint
- Figure 4.111 Principal stress vector for Elastic Joint
- Figure 4.112 Bending moment for Elastic Joint
- Figure 4.113 Deformed shape for Plastic Joint
- Figure 4.114 Principal stress vector for Plastic Joint
- Figure 4.115 Bending moment for Plastic Joint

In general, rersults of the Elastic Joint Model are close to those of conventional continuum analysis in Example Problem 14.

On the other hand, Plastic Joint Model shows considerable amount of slippage below bottom corner of tunnel as in Figures 4.113 and 4.114. Stress distributions are quite different from Elastic Joint Model.

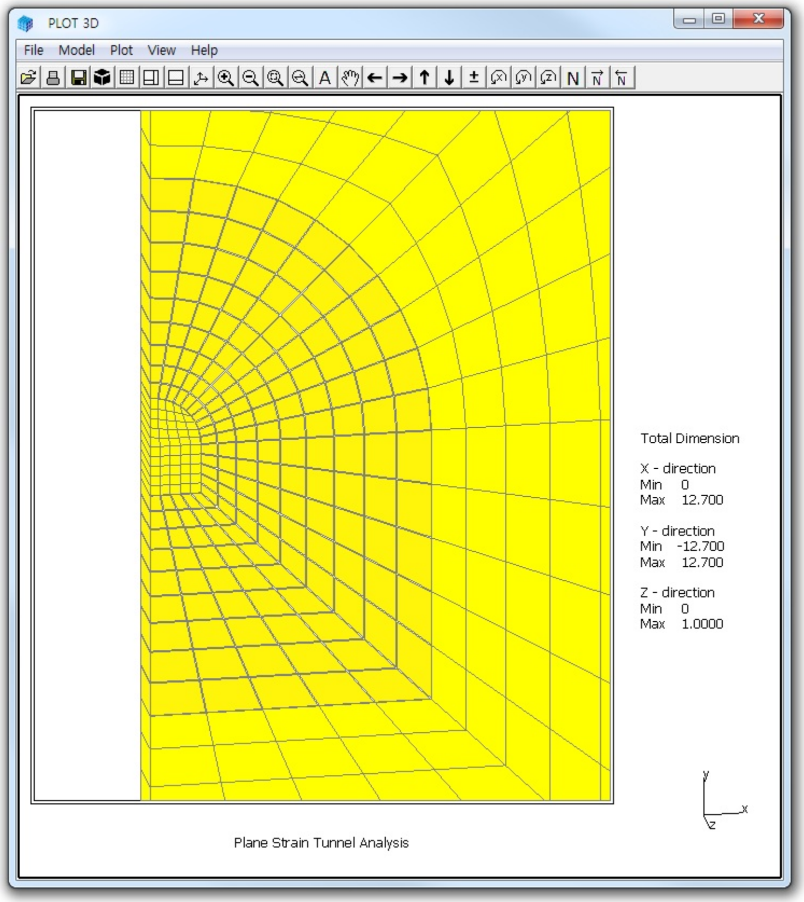


Figure 4.109 Finite element mesh around tunnel

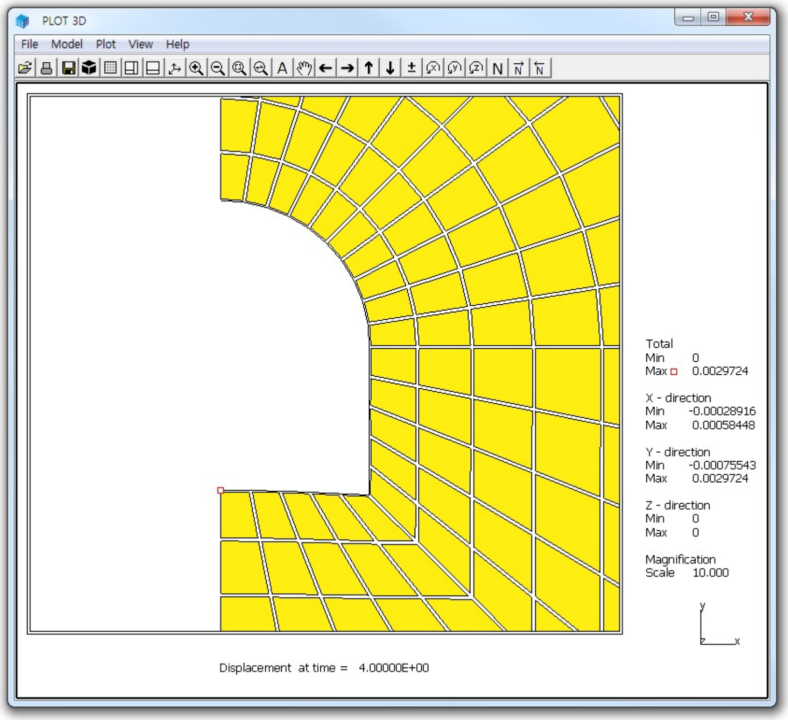


Figure 4.110 Deformed shape for Elastic Joint

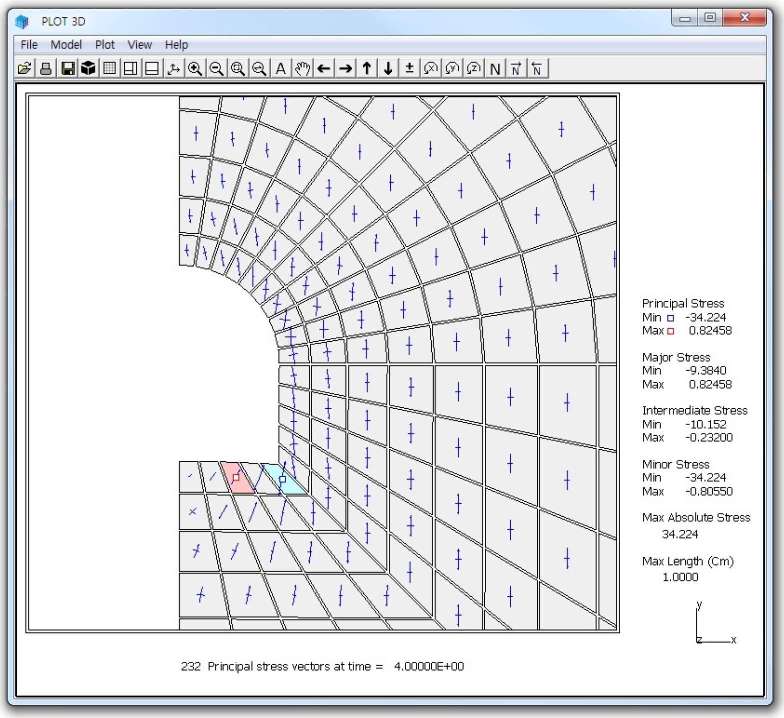


Figure 4.111 Principal stress vector for Elastic Joint

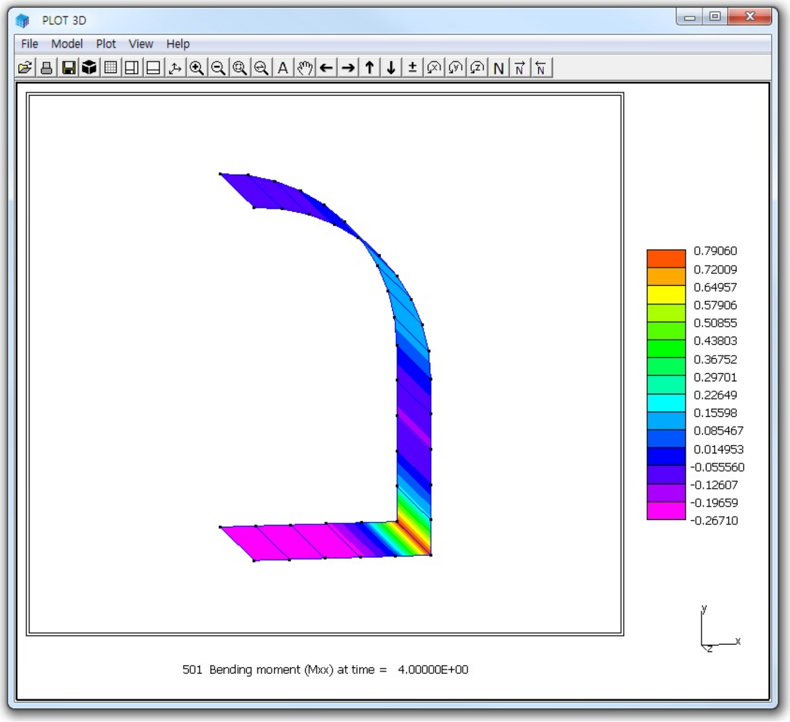


Figure 4.112 Bending moment for Elastic Joint

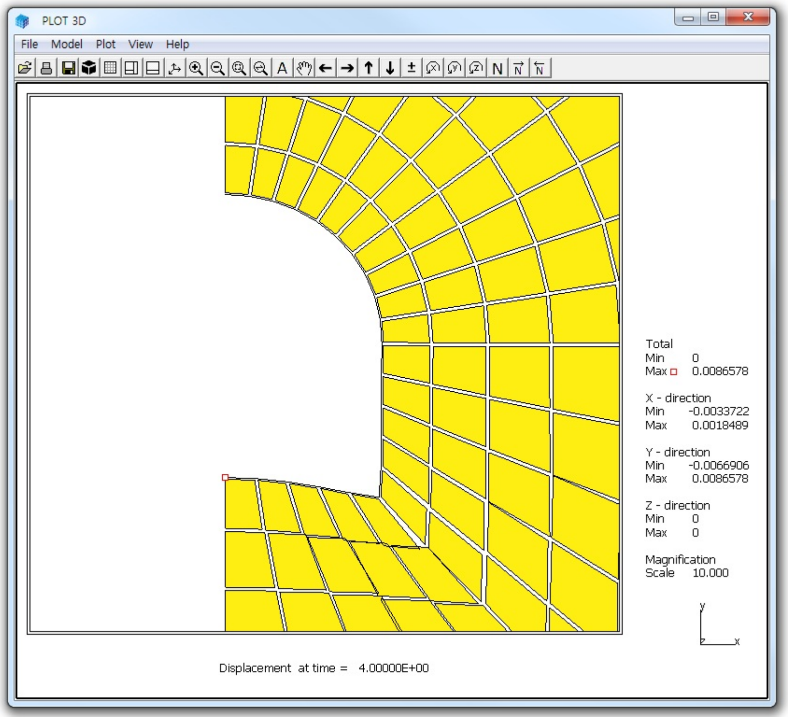


Figure 4.113 Deformed shape for Plastic Joint

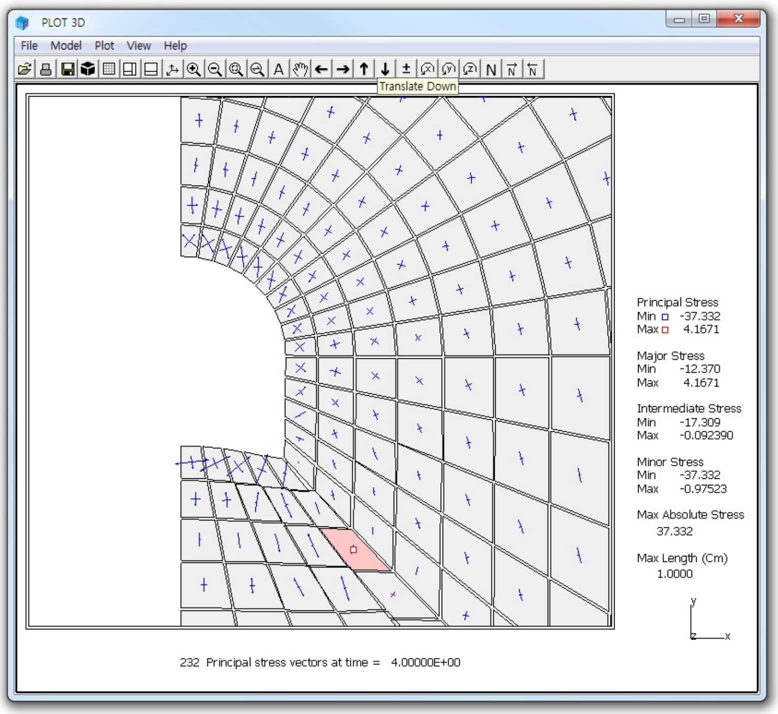


Figure 4.114 Principal stress vector for Plastic Joint

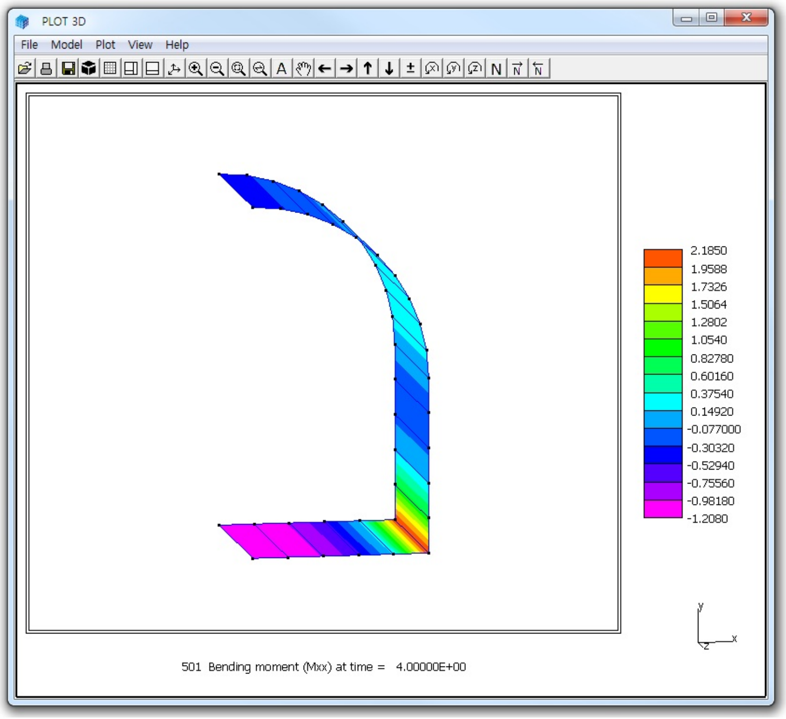


Figure 4.115 Bending moment for Plastic Joint

4.25 Spring Analysis

This example problem is to show how to model springs using special features in beam element in Card 6.4.1 of SMAP-3D User's Manual.

The example is composed of two truss members connected by horizontal and vertical springs as shown in Figure 4.116. The structure is subjected to external horizontal and vertical nodal forces.

Figure 4.117 shows the finite element mesh consisting of two beam elements and two truss elements. Beam element 1 and 2 are used to model vertical and horizontal spring, respectively. When you specify MR = 11 or -11 in Card 6.4.1, beam axial stiffness (E A/L) represents axial spring constant (Ks).

For the material properties, dimensions and loads in Figure 4.116, the exact solution gives following displacements and truss axial forces:

Horizontal	Displacement = 0.04	
Vertical	Displacement = 0.02	
Horizontal	Truss Axial Force = 40 (Compression)	
Vertical	Truss Axial Force = 20 (Tension)	

SMAP-3D results show exact as shown in Figures 4.118, and 4.119 for displacements and truss axial forces, respectively.

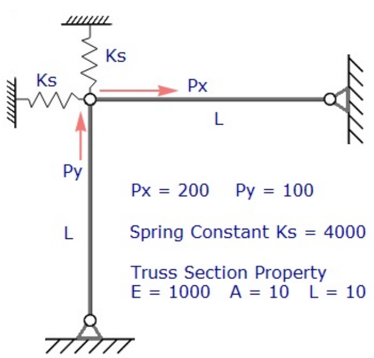


Figure 4.116 Truss members connected by springs

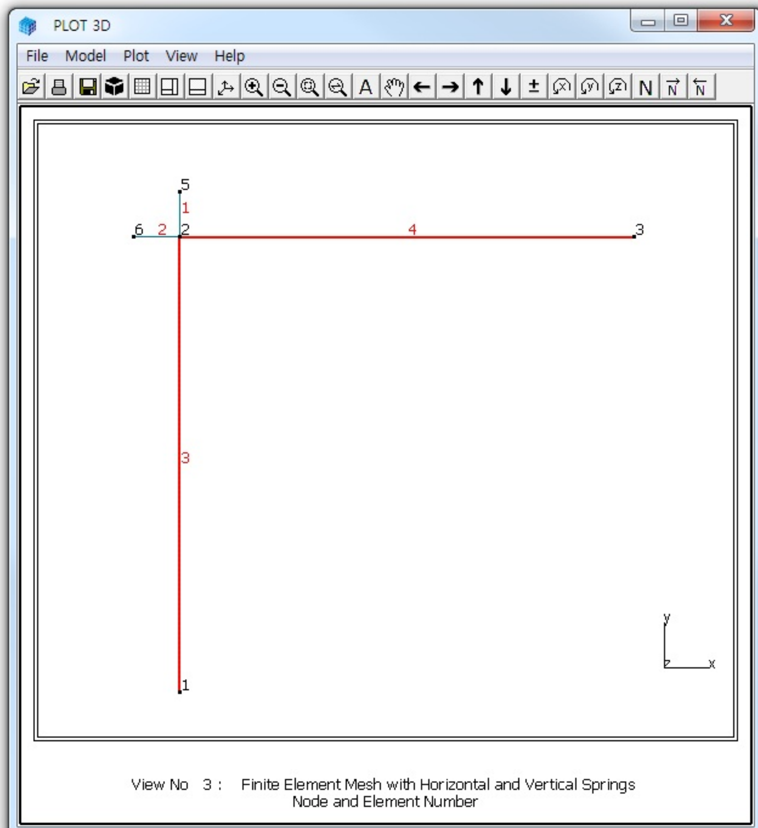


Figure 4.117 Finite element mesh for Example Problem 25

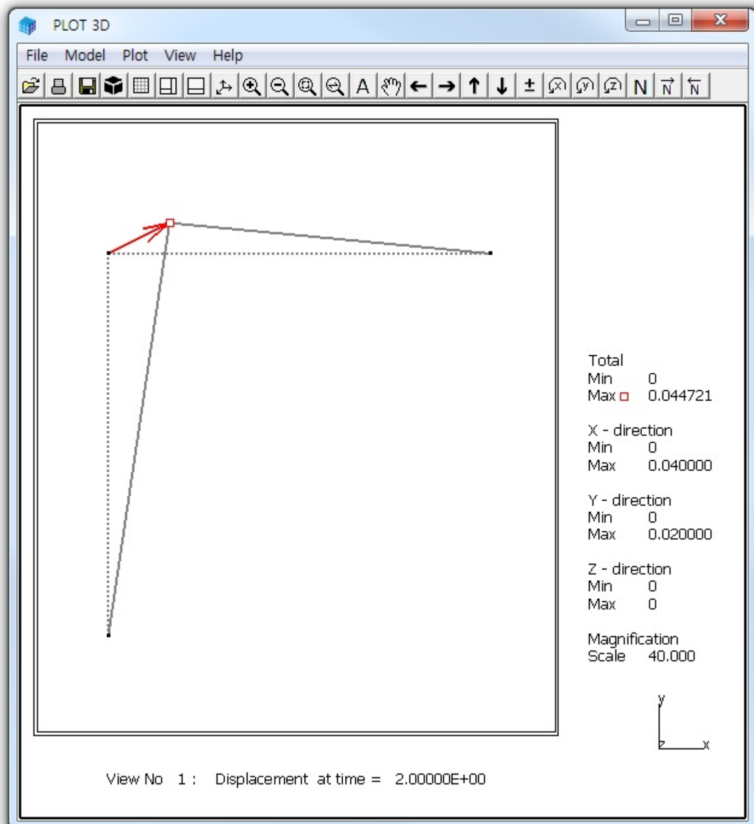


Figure 4.118 Deformed shape for Example Problem 25

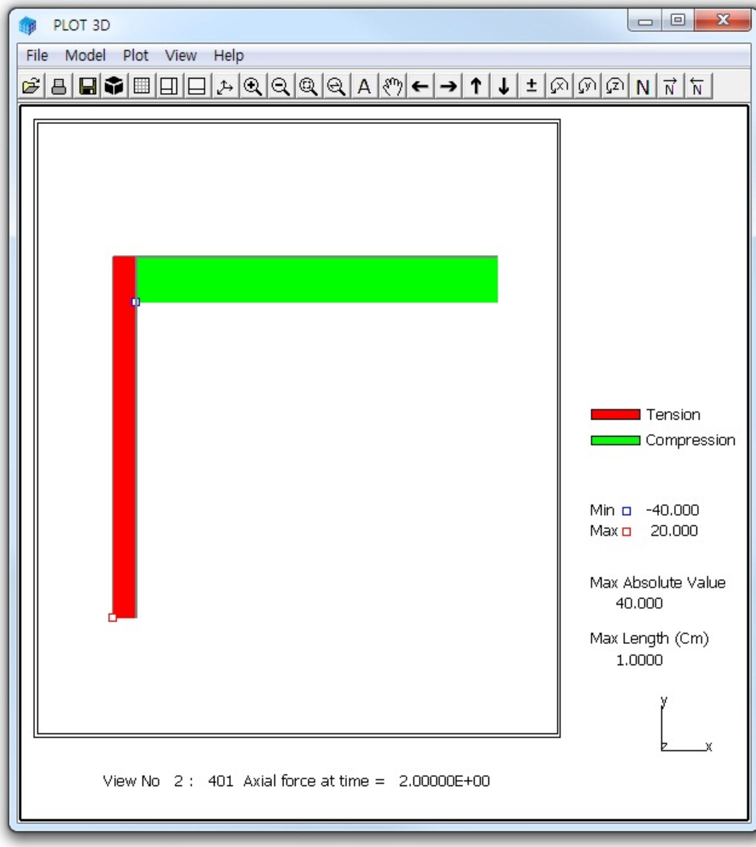


Figure 4.119 Truss axial force for Example Problem 25

4.26 Nonlinear Truss Analysis

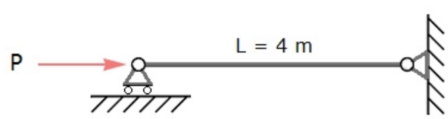
Truss elements in SMAP can consider nonlinear behavior such as yielding and post buckling as schematically illustrated in Figure 4.121. Following examples are to show how to use such material parameters in truss element in Card 7.4.3 of SMAP-3D User's Manual.

Figure 4.120 shows a horizontal truss element subjected to axial force. A typical I-section (400x150@720kN/m) is assumed for truss member with material and cross section properties as listed in the figure.

Six different cases are performed:

1. Buckling and Tension Yielding (Figure 4.122)
2. Compression and Tension Yielding (Figure 4.123)
3. Tension Yielding for No Compression Member (Figure 4.124)
4. Compression Yielding for No Tension Member (Figure 4.125)
5. Buckling for No Tension Member (Figure 4.126)
6. Initial Stress (See Case 6 at the end of example)

Compression resistance is not allowed for No Compression Member such as cable and tension resistance is not allowed for No Tension Member such as strut. A linear elastic truss element is added to prevent the structure from being unstable when plastic yielding. Both compression and tension yield strengths are increased more than 12 times in order to make an exaggerated graphical presentation associated with load and unload.



I-Section (400x150@720 kN/m)

I_x = 2400 Cm^4 I_y = 887 Cm^4
E = 21000 kN/Cm^2 A = 91.73 Cm^2
σ_y = 32.5 kN/Cm^2 σ_{cr} = 12.52 kN/Cm^2

Figure 4.120 Truss member subjected to axial force

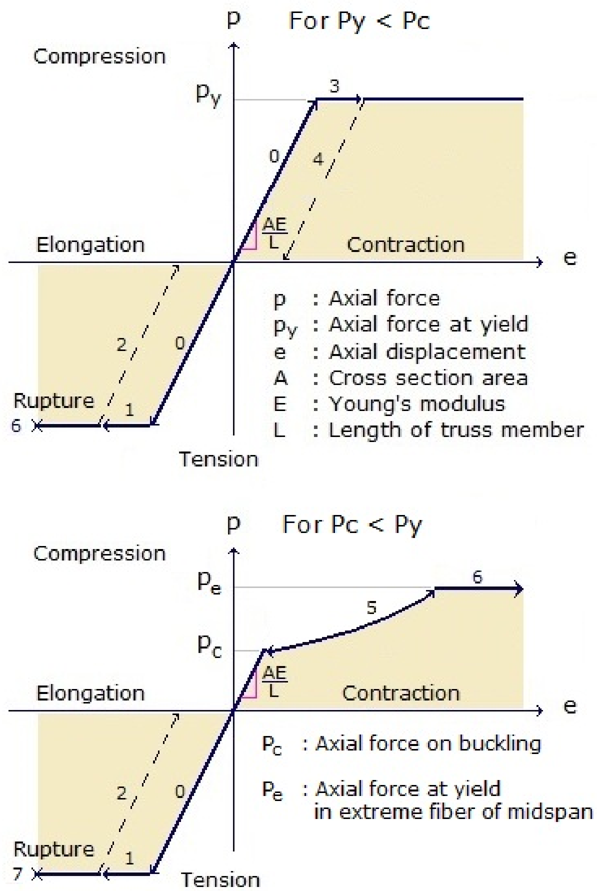


Figure 4.121 Nonlinear force displacement curve

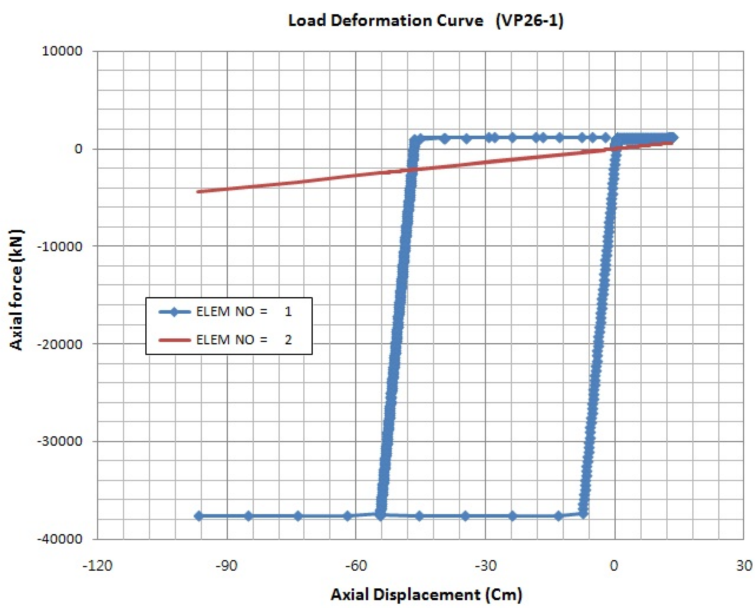


Figure 4.122 Buckling and Tension Yielding (VP26-1)

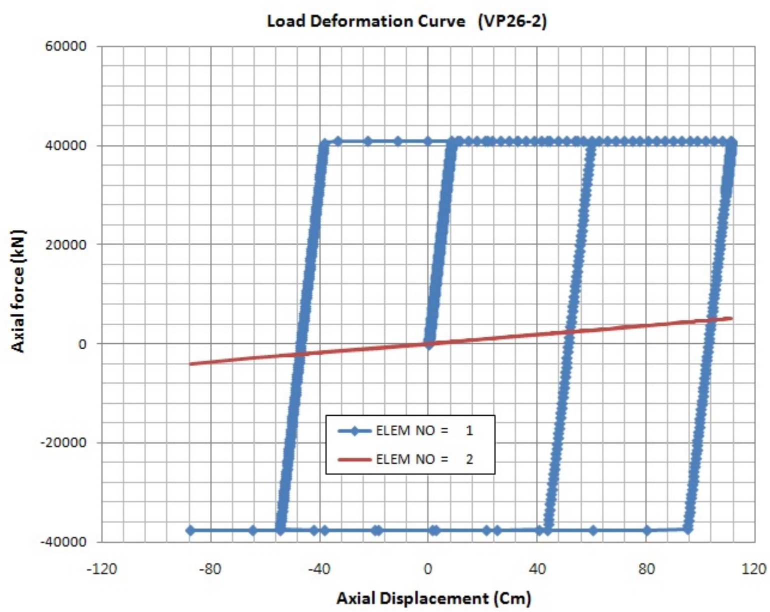


Figure 4.123 Compression and Tension Yielding (VP26-2)

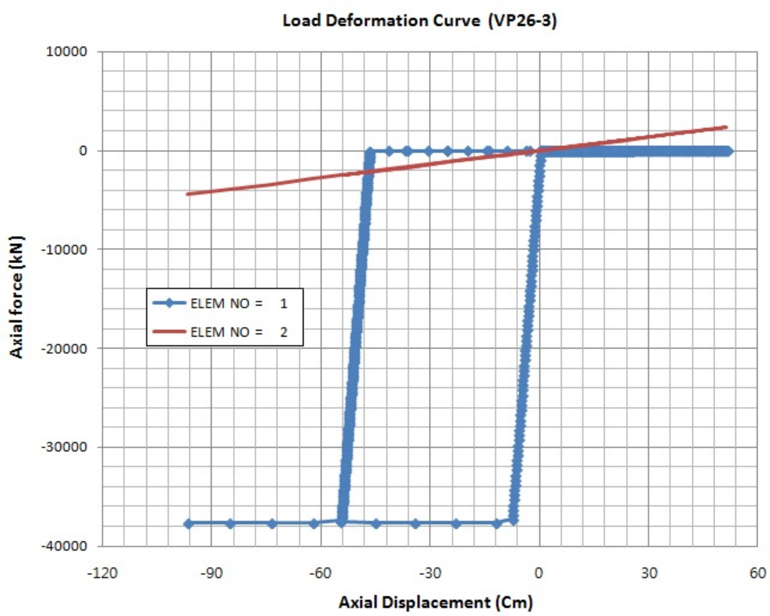


Fig 4.124 Tension Yielding for No Compression Member (VP26-3)

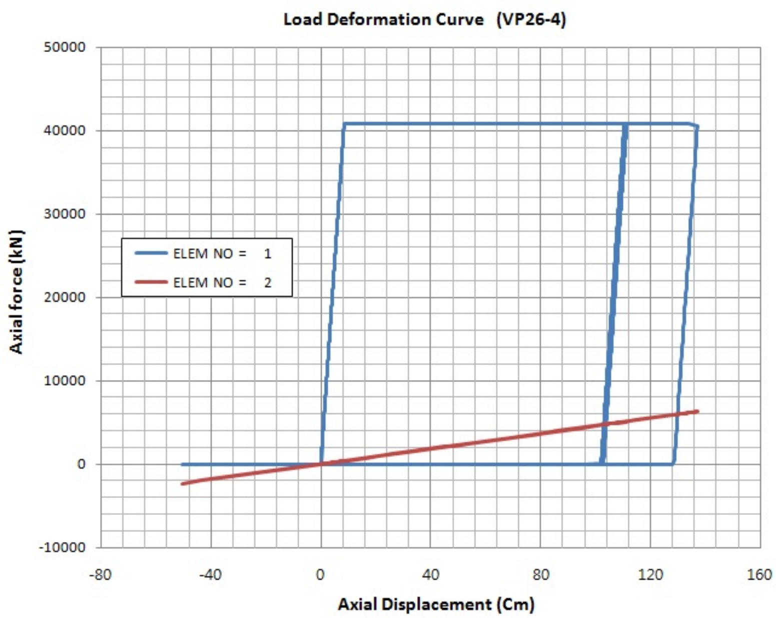


Fig 4.125 Compression Yielding for No Tension Member (VP26-4)

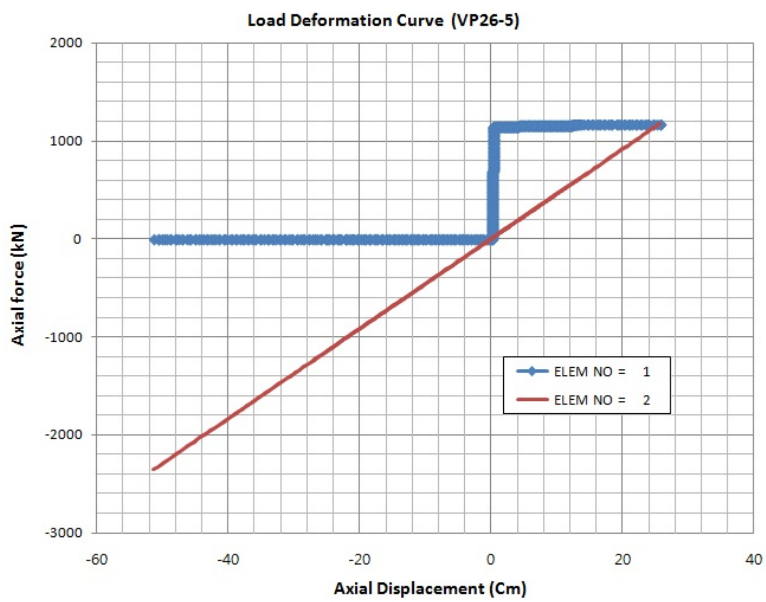


Figure 4.126 Buckling for No Tension Member (VP26-5)

Case 6 Initial Stress

For this example, following parameters are used:

$L = 400$ Cm $E_1 = 21000$ kN/Cm^2 $E_2 = 1000$ kN/Cm^2

To check Initial Stress, Member 1 is assumed to have initial compressive stress ($\sigma_i = -10$ kN/Cm^2) with the corresponding initial strain ($\varepsilon_i = \sigma_i / E_1 = -0.00047619$).

Thus the original length of Member 1 at stress free

$Lo = L / (1 + \varepsilon_i) = 400 / (1 - 0.00047619) = 400.19057$ Cm

Now, when Members 1 and 2 are connected,

$$\sigma_1 \cdot A + \sigma_2 \cdot A = P = 0 \quad \text{i.e.} \quad \sigma_2 = -\sigma_1 \tag{1}$$

$$\sigma_2 = E_2 \cdot \varepsilon_2 \tag{2}$$

$$\begin{aligned} \varepsilon_1 &= ((L + \Delta L \quad) - Lo) / Lo \\ &= ((L + \varepsilon_2 \cdot L) - Lo) / Lo \\ &= (L / Lo) \cdot (1 + \varepsilon_2) - 1 \end{aligned} \tag{3}$$

$$\begin{aligned} \sigma_1 &= E_1 \cdot \varepsilon_1 \\ &= (E_1 \cdot L / Lo) \cdot (1 + \varepsilon_2) - E_1 \end{aligned} \tag{4}$$

Substituting (2) and (4) into (1),

$$\begin{aligned} \varepsilon_2 &= E_1 (1 - L / Lo) / (E_2 + E_1 \cdot L / Lo) \\ &= 0.00045475 \end{aligned} \tag{5}$$

From (3)

$\varepsilon_1 = -0.000021654$

And from (2) and (1)

$\sigma_1 = -0.45475$ kN/Cm^2 (Compression)

$\sigma_2 = 0.45475$ kN/Cm^2 (Tension)

SMAP results show exact solution.

4.27 SDOF System To Ground Acceleration

A single Truss element is used to model axial spring subjected to sinusoidal ground acceleration as schematically shown in Figure 4.127. Mass is lumped at the node in the right side of truss member.

Following parameters are assumed:

$L = 120$ inch $A = 1\ in^2$ $E = 30\text{x}10^6$ psi

$\rho = (1/1.2)\ lb\text{-}s^2/in^4$ $a = 200\ in/s^2$ $\omega = 40$ rad/s

$c = 500$ lb-s/in

Lumped mass at right node:

$m = \rho\ A\ L = (1/1.2)\ (1)\ (120) = 100\ lb\text{-}s^2/in$

Equivalent spring constant:

$k = E\ A\ /\ L = (30\text{x}10^6)\ (1)\ /\ (120) = 250{,}000$ lb/in

Natural frequency:

$\omega_n = (k\ /\ m)^{1/2} = (250{,}000\ /\ 100)^{1/2} = 50$ rad/s

Critical damping ratio: $\xi = c\ /\ (2\ m\ \omega_n) = 0.05$

Damped natural frequency : $\omega_d = \omega_n \sqrt{1-\xi^2}$

Frequency ratio: $\beta = \omega\ /\ \omega_n = 40\ /\ 50 = 0.8$

For systems with viscously damped single degree of freedom, the relative displacement is given by

$$\bar{x}(t) = e^{-\xi \omega_n t}(A \text{ Cos } \omega_d\ t + B \text{ Sin } \omega_d\ t) + C \sin \omega\ t + D \text{ Cos } \omega\ t$$

The constants C and D are given by

$$C = \frac{m\,a}{k}\ \frac{1-\beta^2}{(1-\beta^2)^2 + (2\xi\beta)^2} \qquad D = \frac{m\,a}{k}\ \frac{-2\xi\beta}{(1-\beta^2)^2 + (2\xi\beta)^2}$$

Assuming initial conditions at rest, constants A and B are given by

$$A = -D \qquad B = -\left(\frac{\omega}{\omega_d}\right) C - \xi \left(\frac{\omega_n}{\omega_d}\right) D$$

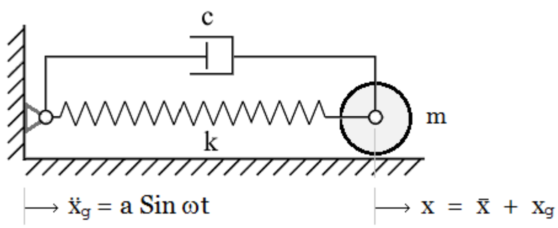


Figure 4.127 SDOF system subjected to ground acceleration

Figure 4.128 shows time history of computed relative displacements. SMAP results are almost identical to the exact solution.

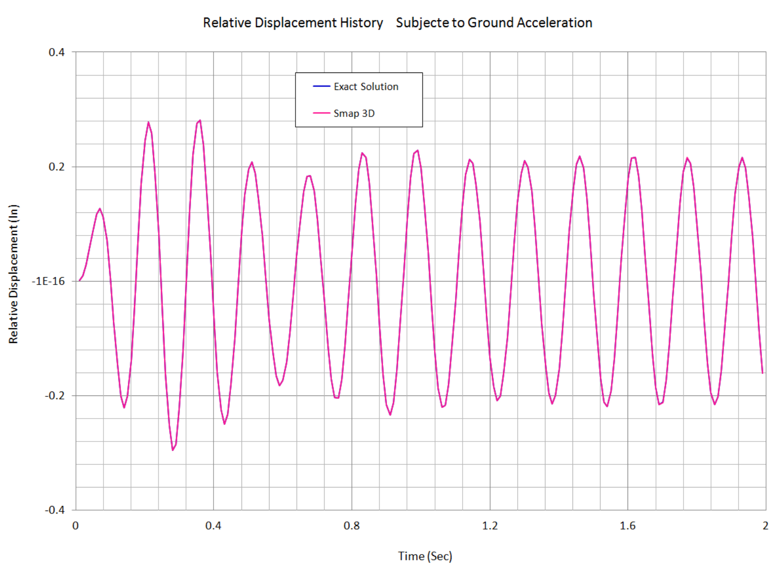


Fig 4.128 Time history of relative displacements

4.28 Frames with Rotational Spring Connection

This example is the same as Example problem 21 except that it is connected by rotational spring and subjected to both moment and horizontal force at the connection as shown in Figure 4.129.

The rotational spring is modeled by the simple Joint Spring Element which can consider axial, shear, torsional and flexural resistances. For this example, the Joint Spring properties are assumed very rigid in all deformation modes except the rotation about z-axis.

Five analyses are performed to see the influence of connection:

1. Rigid connection
2. Hinge connection
3. Rotational spring connection, rigid Kr = 1×10^{6} t-m/rad
4. Rotational spring connection, very flexible Kr = 1×10^{-3} t-m/rad
5. Rotational spring connection, somewhat rigid Kr = 1×10^{4} t-m/rad

Computed results are summarized in detail in Joint_Spring_3D.pdf. It approaches to rigid connection when the rotational spring is rigid and hinge connection when the spring constant is very flexible.

Figures 4.130 to 4.134 show finite element mesh, deformed shape, thrust, shear and bending moment distributions, respectively, for the rotational spring connection with Kr = 1×10^{4} t-m/rad.

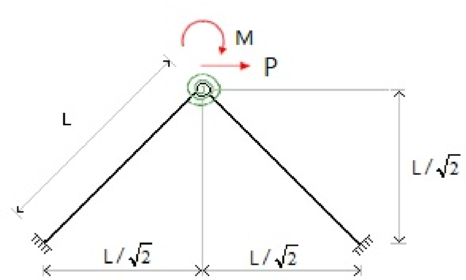


Fig 4.129 Frames with rotational spring connection

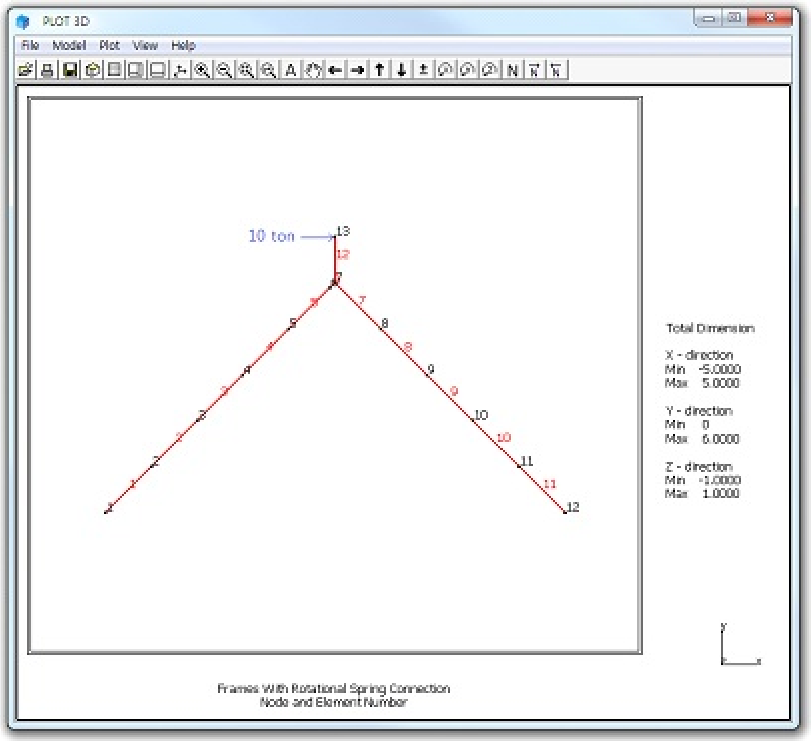


Figure 4.130 Finite element mesh

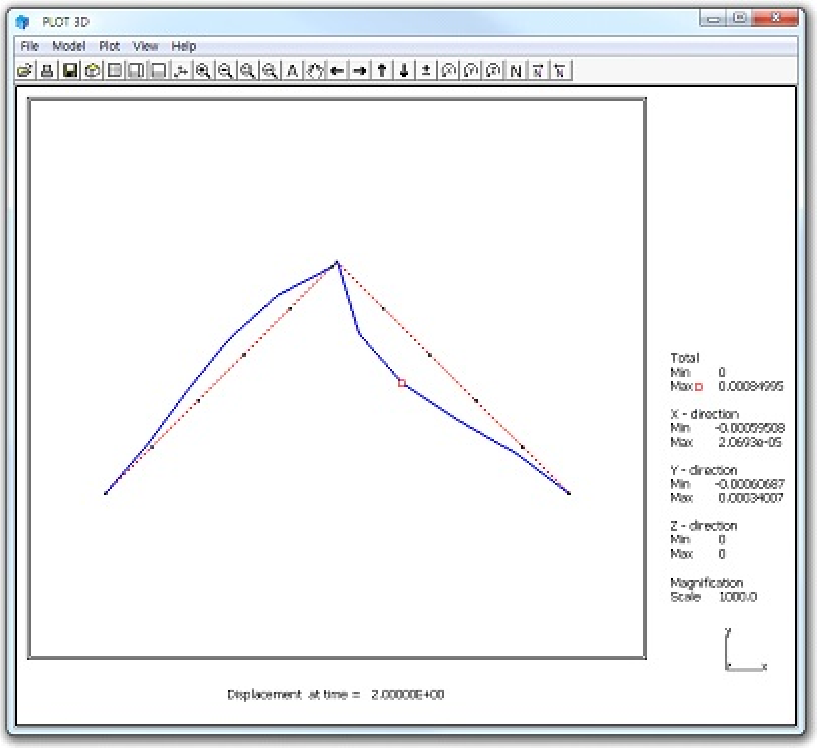


Figure 4.131 Deformed shape

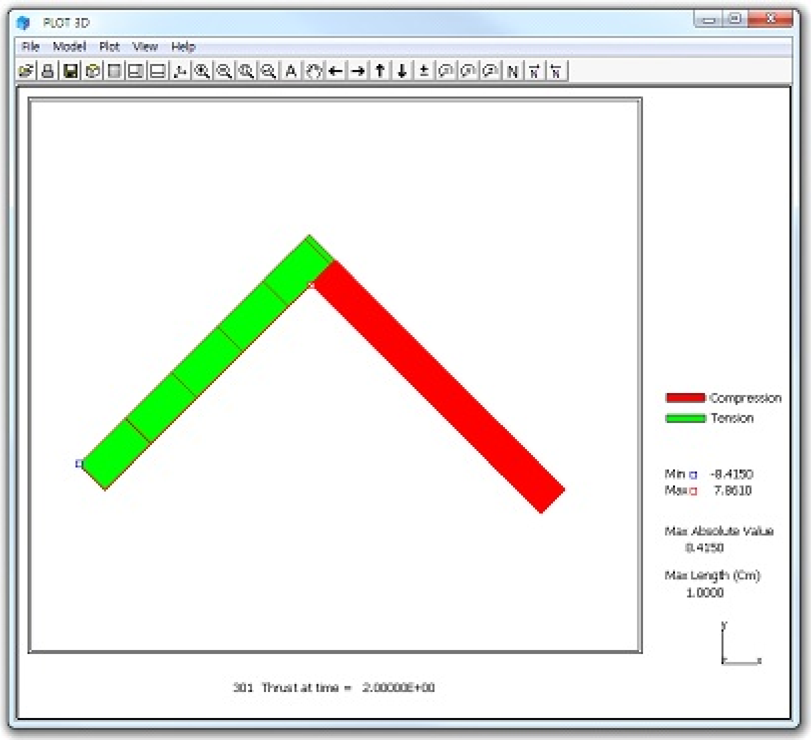


Figure 4.132 Thrust distribution

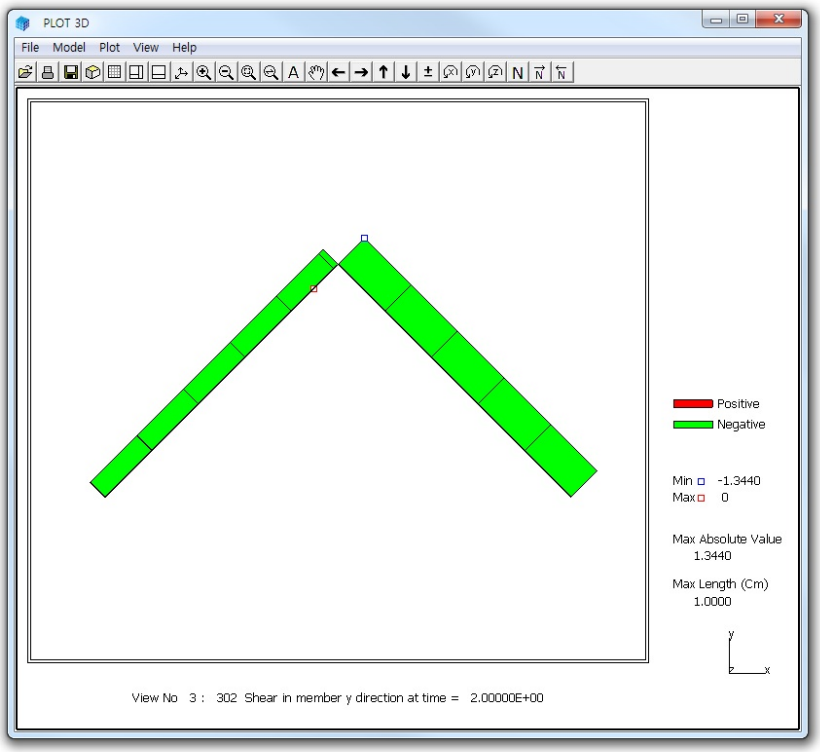


Figure 4.133 Shear distribution

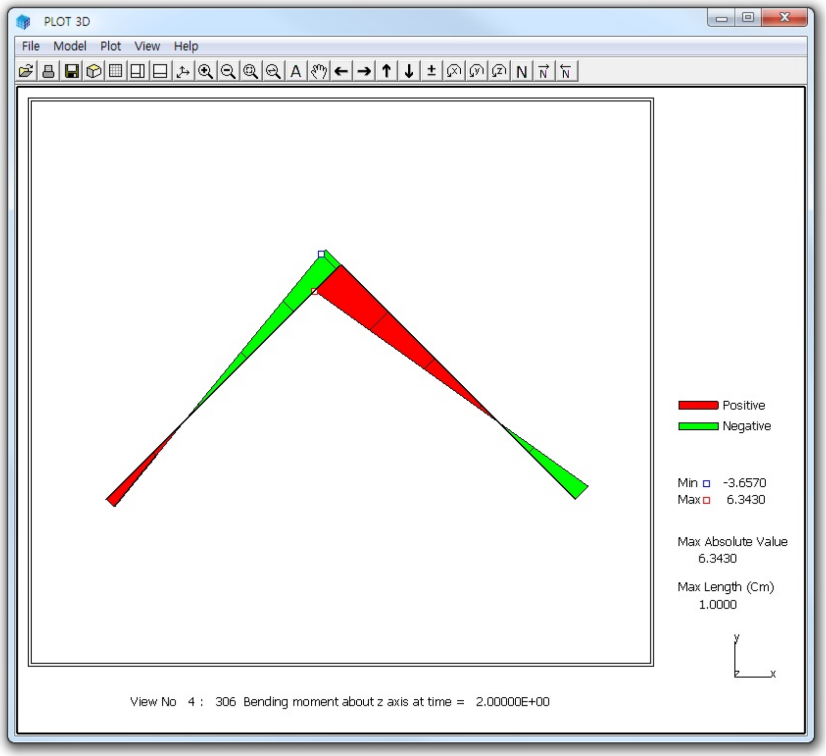


Figure 4.134 Bending moment distribution

4.29 Reinforced Concrete Beam

This example problem is to verify the implementation of reinforcing bars (rebars) into quadrilateral shell element. This example is the same as Example problem 22 except that it is modeled by reinforced shell element. Figure 4.135 shows a simply supported reinforced concrete beam subjected to a concentrated load at midspan.
To simplify the problem, it was assumed that both reinforcing bars and concrete are linearly elastic.

The exact beam solution without shear deformation is given below:

Maximum deflection at the center without rebars,

$$\delta = \frac{P \cdot L^3}{48 \;\; E_c \cdot I_c} = 1.190 \text{ Cm}$$

Maximum deflection at the center with rebars,

$$\delta = \frac{P \cdot L^3}{48 \;\; E_c \cdot I_t} = 1.040 \text{ Cm}$$

By symmetry, only left half of the beam is modeled using 10 reinforced shell elements.

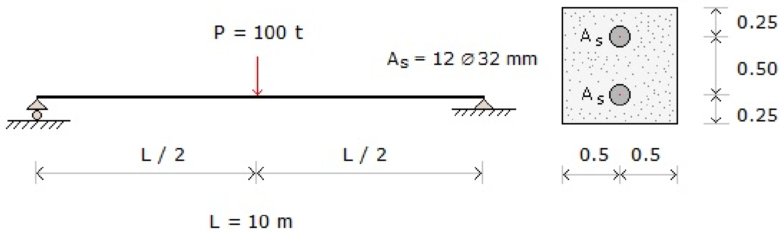
The computed center deflections are compared with the exact beam solution as shown in Table 4.6. SMAP-3D results are very close to the exact beam solutions.

Computed results are shown in the following order:

Figure 4.136	Deformed shape
Figure 4.137	Bending moment
Figure 4.138, 4.139	Top and bottom surface axial stress
Figure 4.140, 4.141	Top and bottom reinforing bar axial stress

Table 4.6 Computed center deflections

Reinforcement	SMAP-3D Shell Element	Exact Beam Solution
Plain Concrete	1.1812 Cm	1.190 Cm (without rebar)
Reinforced Concrete	1.0329 Cm	1.040 Cm (with rebar)



$E_c = 2.1 \times 10^6\ t/m^2$

$v_c = 0.2$

$E_s = 2.1 \times 10^7\ t/m^2$

$I_c = 0.0833\ m^4$

$I_s = 2\ (E_s / E_c)\ A_s\ (0.025)^2 = 0.012063\ m^4$

$I_t = I_c + I_s = 0.0954\ m^4$

<u>Shell Element Input Parameter for Reinforcing Bars</u>

NRBX = 1, NRBY = 0, d1 = d2 = 0.25 m

As1 = As2 = 0.00965 m^2

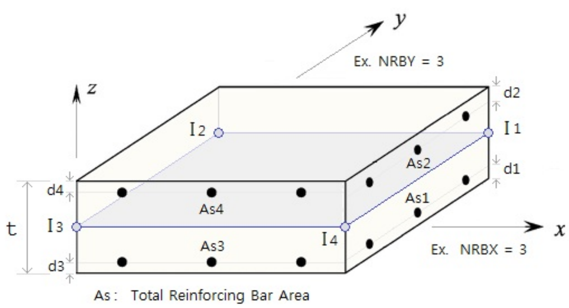


Figure 4.135 Reinforced concrete beam

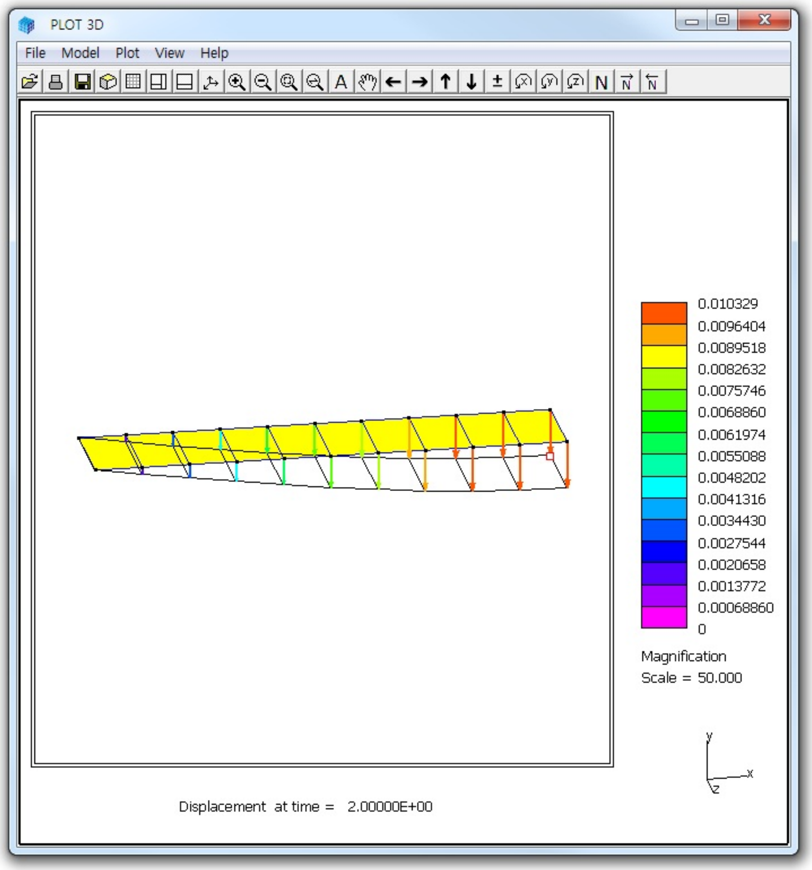


Figure 4.136 Deformed shape

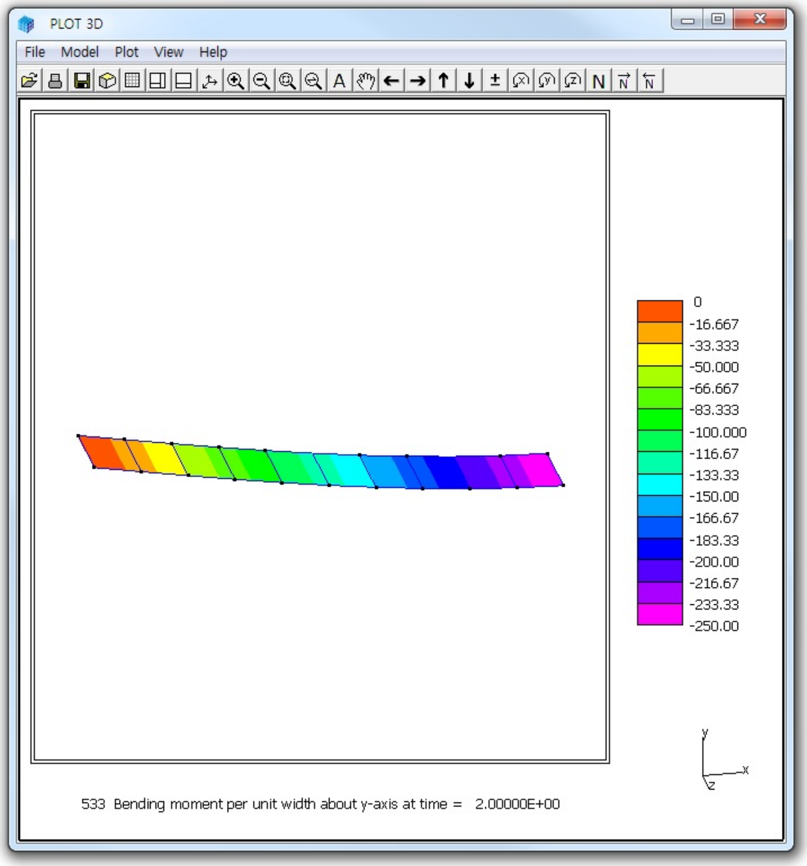


Figure 4.137 Bending moment about y-axis

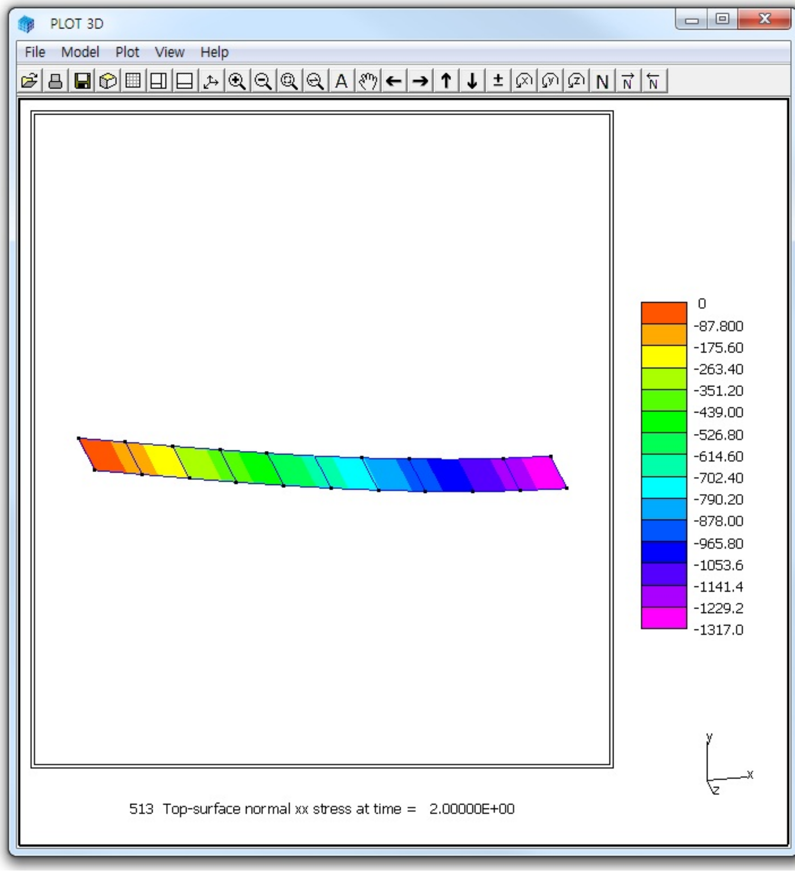


Figure 4.138 Top surface axial stress

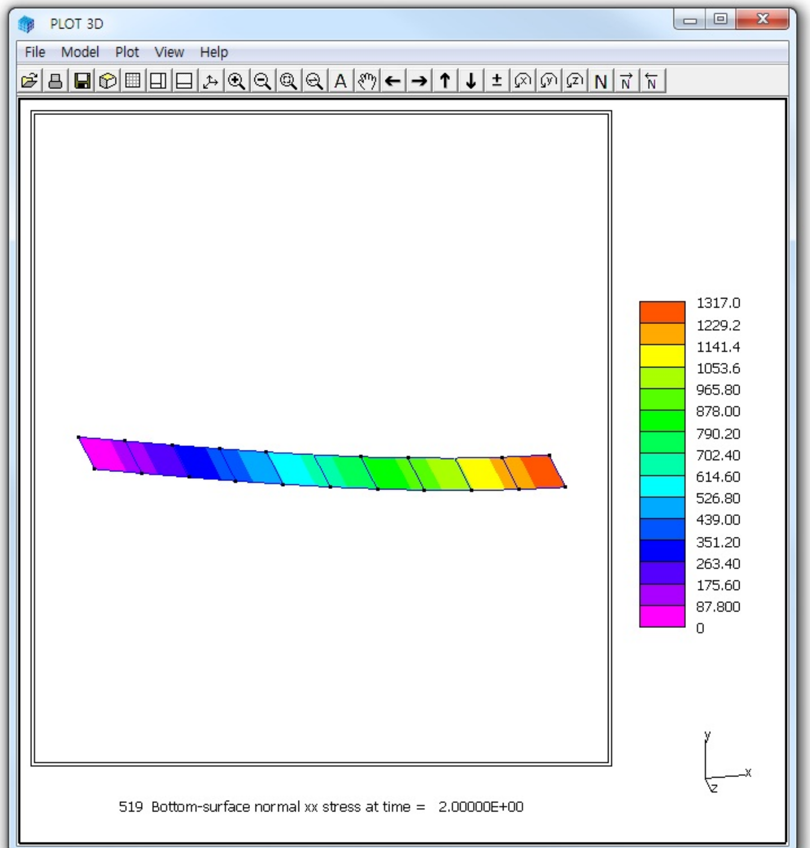


Figure 4.139 Bottom surface axial stress

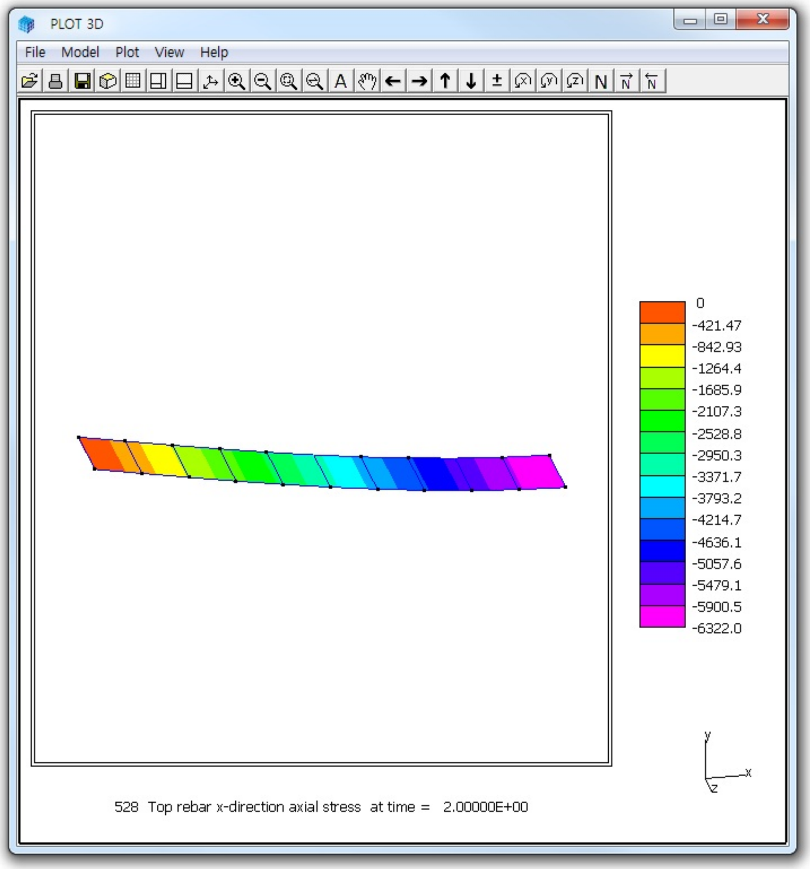


Figure 4.140 Top reinforcing bar axial stress

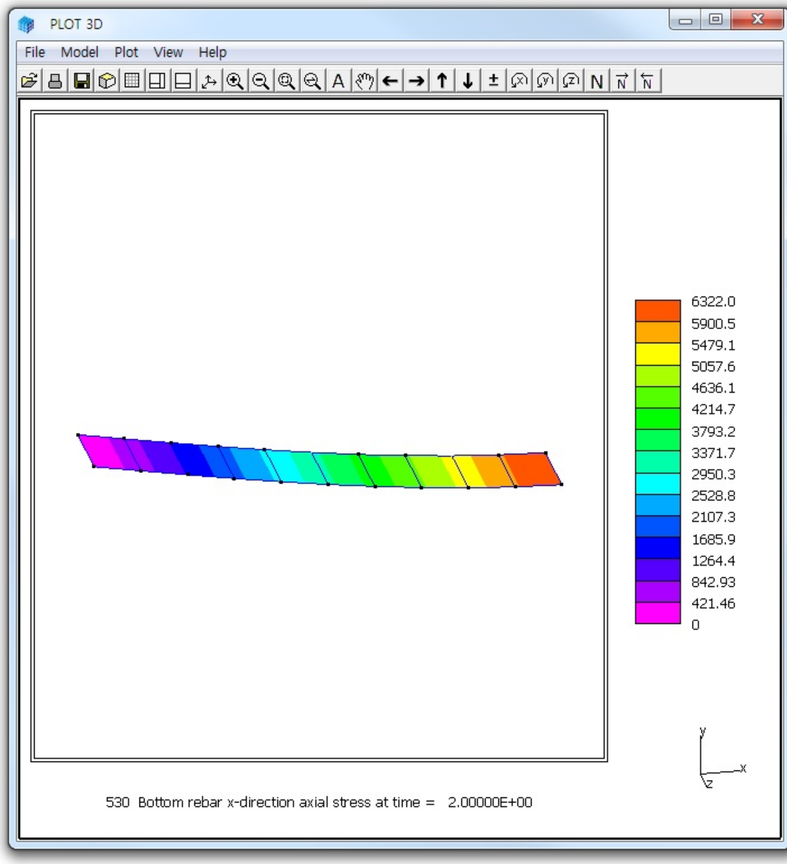


Figure 4.141 Bottom reinforcing bar axial stress

4.30 Reinforced Concrete Cylinder

This example is to check the reinforced concrete cylinder subjected to uniformly distributed radial line loads as shown in Figure 4.142.
This example is an axially symmetric problem since both the structure and the external load are axially symmetric.

The exact solution for unreinforced cylinder can be obtained from the reference: Timoshenko and Woinowsky-Krieger, Theory of Plates and Shells, 2nd Edition, McGraw-Hill International Series, 28th Printing 1989.

This exact solution is further modified here such that it includes both axial (meridian) and hoop (circumferential) reinforcements as listed in the file Reinforced_Cylinder_3D.pdf.

Four cases are performed with different reinforcements:

1. Concrete without reinforcements
2. Concrete with hoop reinforcements
3. Concrete with axial & hoop reinforcements, ν_c = 0.15
4. Concrete with axial & hoop reinforcements, ν_c = 0.0

Note that the analytical solutions represent exact solutions except the case 3 where it is an approximate closed-form solution.

As in Figure 4.143, the structure is modeled by quadrilateral shell elements which have capability of modeling two way reinforcements.

Overall, SMAP-3D results are very close to the exact solutions.
Refer to the following two files for detailed graphical outputs:
Reinforced_Cylinder_3D.pdf and Smap-3D_Vp30.pdf.

SMAP-3D results for case 3 are compared with closed-form solutions:
  Figure 4.144 Radial displacement profile
  Figure 4.145 Meridian bending moment profile

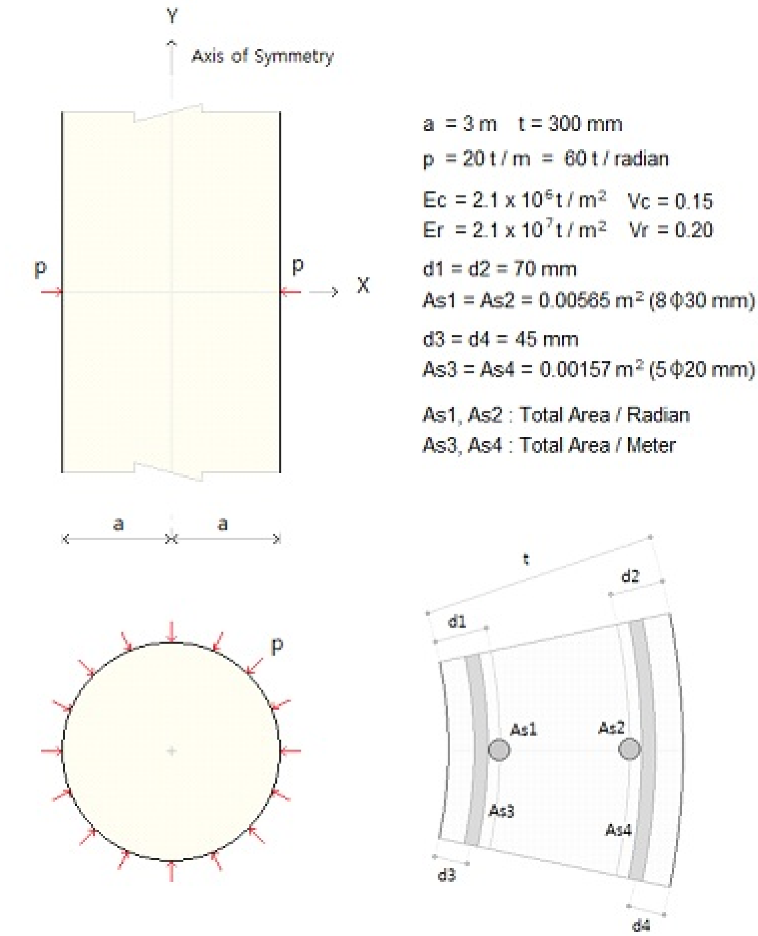


Figure 4.142 Reinforced cylinder section view

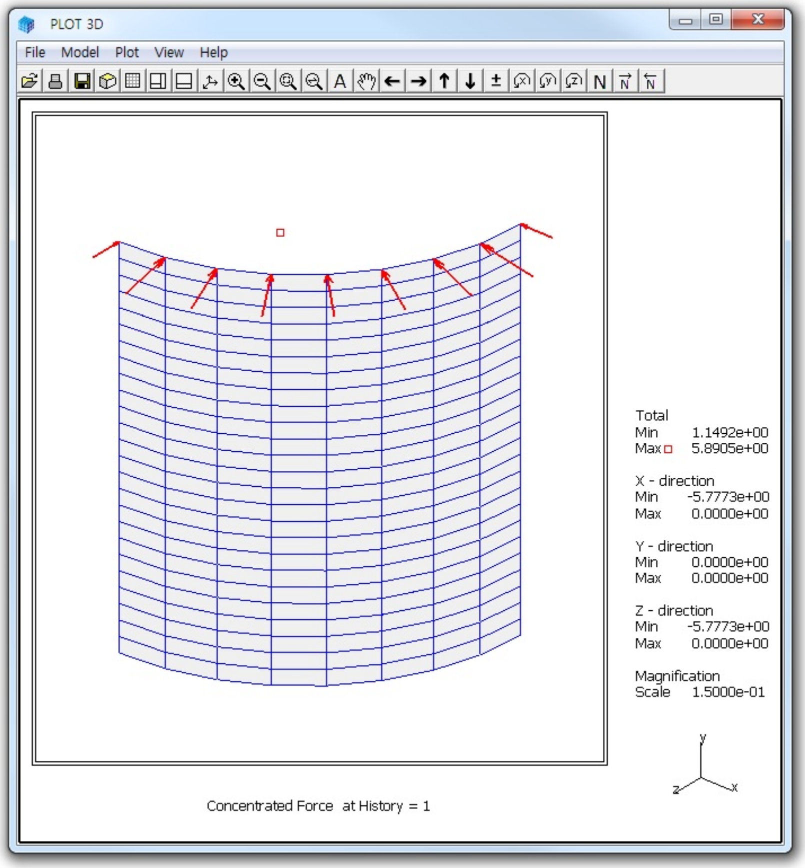


Figure 4.143 Finite element mesh with applied load

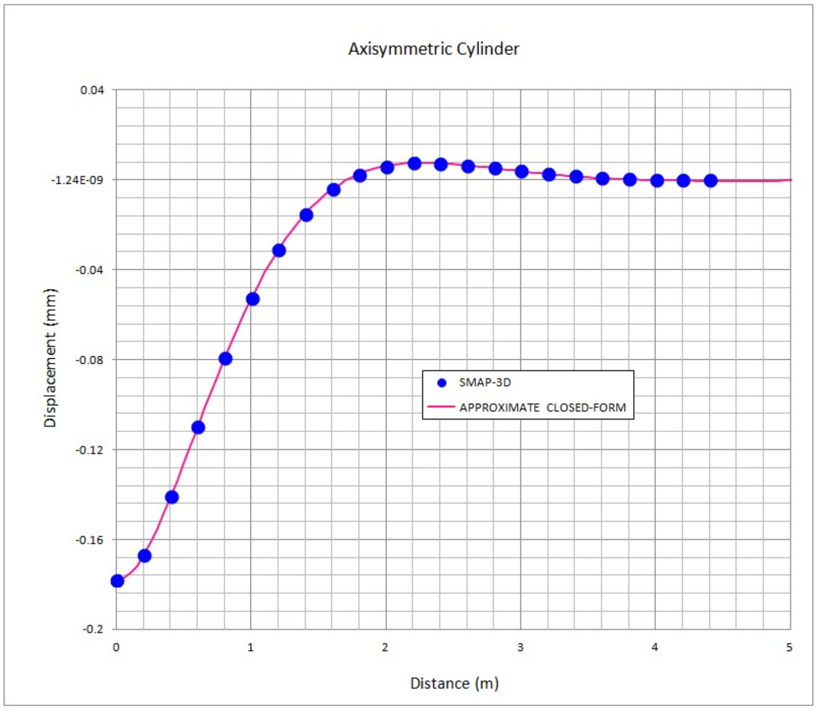


Figure 4.144 Radial displacement profile

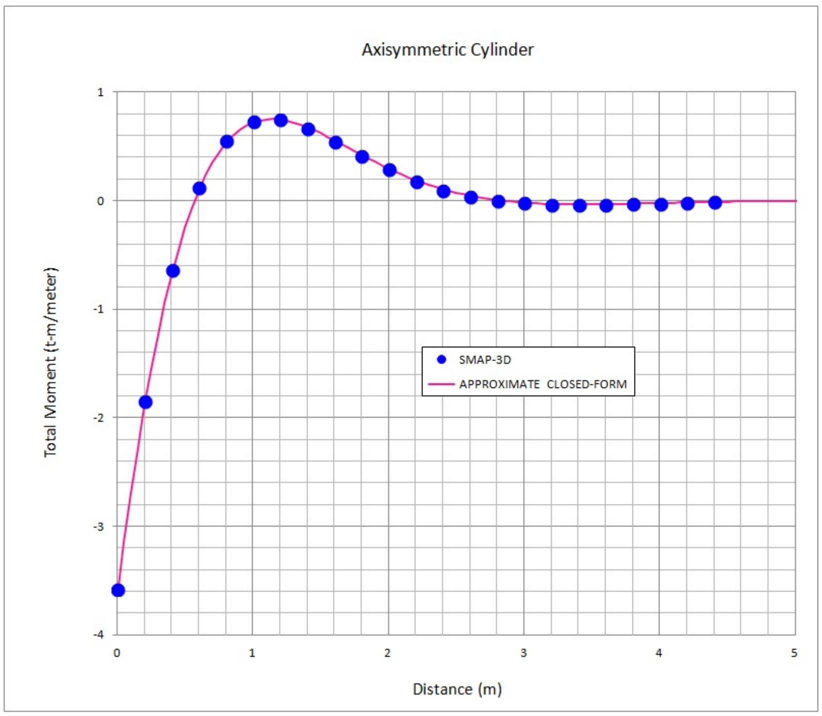


Figure 4.145 Meridian bending moment profile

4.31 Plate Modal Analysis

A simply supported rectangular plate, shown in Figure 4.146, is selected to verify the Modal Superposition method for the dynamic response. By symmetry, only a quarter of the plate is modeled. The plate is subjected to a concentrated step load at center. This problem is identical to the Verification Problem 4.16 which was solved by Direct Integration method.

The closed form solution of natural frequencies of simply supported rectangular plate is given by Kirchhoff plate theory:

$$\omega_{mn} = \sqrt{\frac{D}{\rho\ h}} \left[\left(\frac{m\,\pi}{a}\right)^2 + \left(\frac{n\,\pi}{b}\right)^2 \right] \qquad D = \frac{E\ h^3}{12\,(1 - v^2)}$$

$= 0.0003$ lb-s^2 / in^4	$= 0.25$	h = 1 in
E = 3x10^4 lb / in^2	a = 60 in	b = 40 in

Table 4.7 summarizes the computed natural frequencies along with closed form solution. Both shell and continuum modal analyses predict pretty well natural frequencies of the simply supported rectangular plate.

Figure 4.147 shows the contours of the first three modes solved by shell modal analysis.

Figure 4.148 shows deflection time history at plate center as predicted by modal superposition method using only first 6 mode shapes.
To verify the computed response of the modal superposition method, step-by-step solution by direction integration with the same shell element mesh which was used in shell modal superposition is included. SMAP-3D modal superposition solutions predict very closely the direct integration solution.

Table 4.7 Computed natural frequencies (rad/s)

Mode No	Kirchhoff Plate Theory	Shell 4 Node Quad 16x24 Mesh	Continuum 8 Node Hexa* 8x12 Mesh
1	26.565	26.544	26.412
2	91.955	91.729	91.356
3	173.693	172.992	173.411

Notes:

1. Computed frequencies represent natural frequencies associated with symmetric boundary conditions.

   $\omega_1 = \omega_{11}$ $\omega_2 = \omega_{31}$ $\omega_3 = \omega_{13}$

2. All modal analyses used Subspace Iteration method with lumped mass to compute natural frequencies.

3. Shell modal analysis used 16x24 mesh consisting of 4 node quadrilateral shell elements.

4. Continuum modal analysis used 8x12 mesh consisting of 8 node hexahedral continuum elements with 3 incompatible extra degrees of freedom* (IEDOF =1).

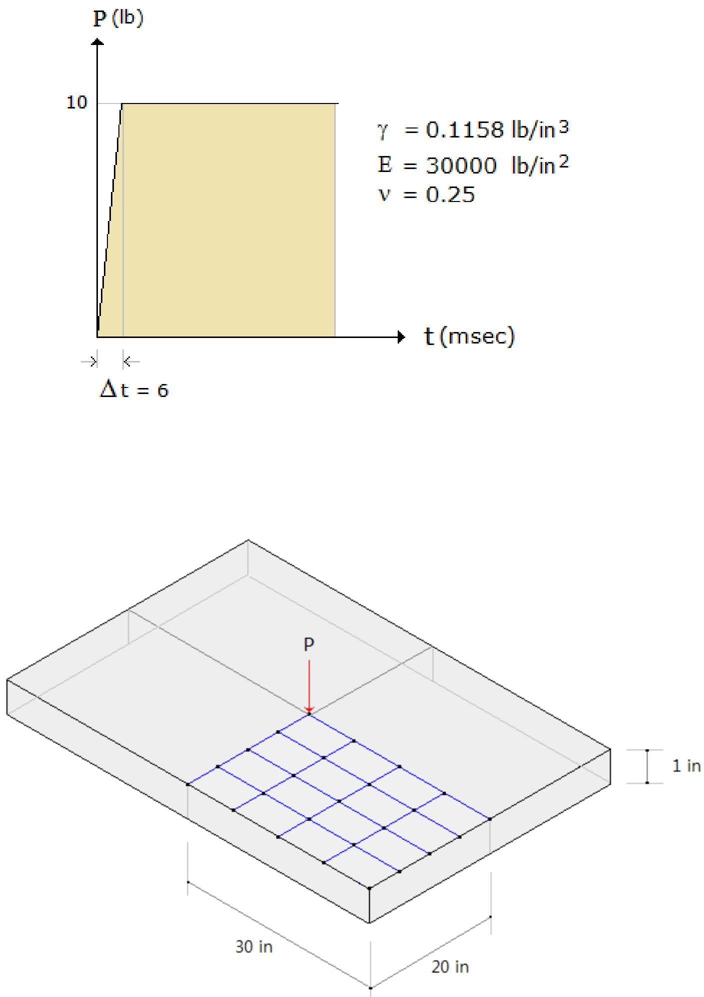


Figure 4.146 Simply supported plate (Not Scaled)

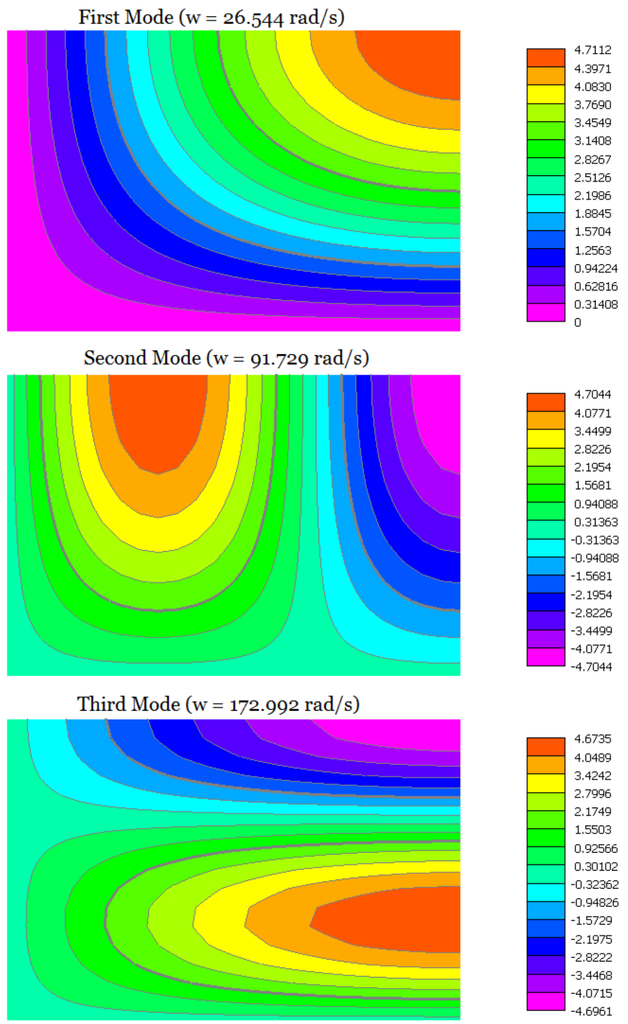


Figure 4.147 First three modes solved by Shell (16x24)

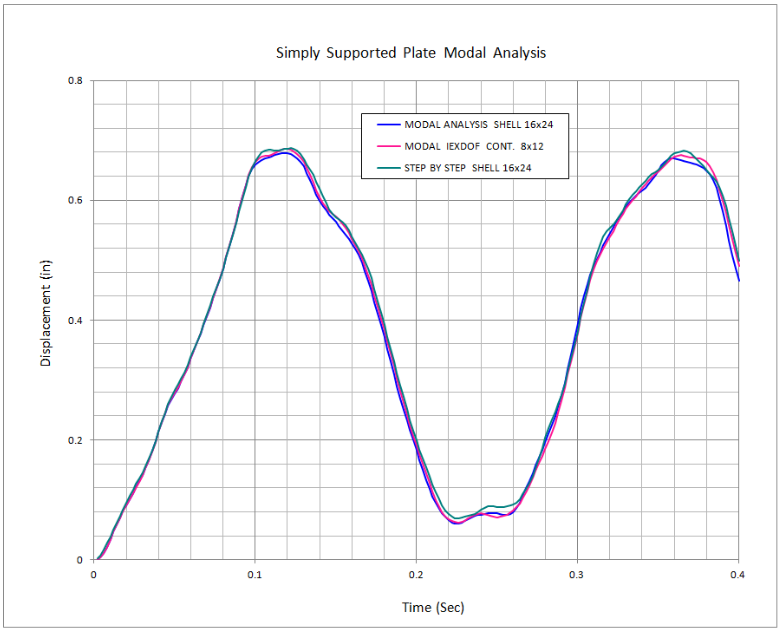


Figure 4.148 Deflection time history at plate center

4.32 Seismic Response Analysis

This example is to solve the free-field seismic response of the linearly viscous elastic soil profile, shown in Figure 4.149 along with material properties, subjected to earthquake excitations from the bedrock.

This problem is the same as the sample problem in SHAKE91 (Idriss and Sun, 1992). A 45.72 m (150 ft) soil profile is subjected to Diamond Heights earthquake in 1989 as outcrop to the elastic half space. The earthquake is scaled to peak acceleration of 0.1g. Scaled earthquake time history and its spectral acceleration are shown in Figures 4.150 and 4.151, respectively. The predominant period of the earthquake is about 0.4 second as shown in the response spectrum.

To mitigate frequency dependency, Rayleigh mass and stiffness proportional damping constants (a, b) are computed in the equation:

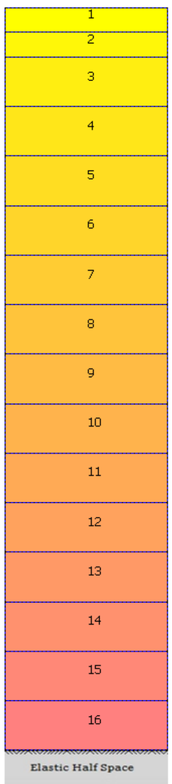
$$\mathbf{a} = 2\ \beta\ \omega_1\ \omega_i\ /\ (\ \omega_1 + \omega_i\ ) \qquad \mathbf{b} = 2\ \beta\ /\ (\ \omega_1 + \omega_i\ )$$

where ω_1 represents for fundamental natural circular frequency of soil profile, ω_i for predominant circular frequency of the input earthquake motion and β for critical damping ratio in an element.

Figure 4.152 shows computed acceleration time histories on the ground surface and Figure 4.153 shows the same accelerations between 10 and 12 seconds where strong motions occur. SMAP-3D solutions predict very closely the closed-form frequency domain SHAKE91 solution.

Figure 4.154 shows spectral accelerations with 5% structural damping on the ground surface and Figure 4.155 shows the same accelerations between 0.1 and 1 seconds. SMAP-3D solutions are very close to SHAKE91 solution.

It should be noted that both base shear and base acceleration options for earthquake load produce exactly the same results as presented in the reference (S. H. Kim and K. J. Kim, 2024).



γ (kN/m³)	G (MPa)	β (%)
19.63	185.963	5
19.63	150.608	5
19.63	150.608	5
19.63	167.878	5
19.63	185.963	5
19.63	185.963	5
19.63	225.042	5
19.63	225.042	5
20.42	326.859	5
20.42	326.859	5
20.42	379.120	5
20.42	379.120	5
20.42	435.255	5
20.42	435.255	5
20.42	495.264	5
20.42	626.675	5
21.99	3333.116	1

Total Dimension

X - direction
Min 0
Max 10.000

Y - direction
Min 0
Max 45.720

Figure 4.149 Finite element meshes and material properties

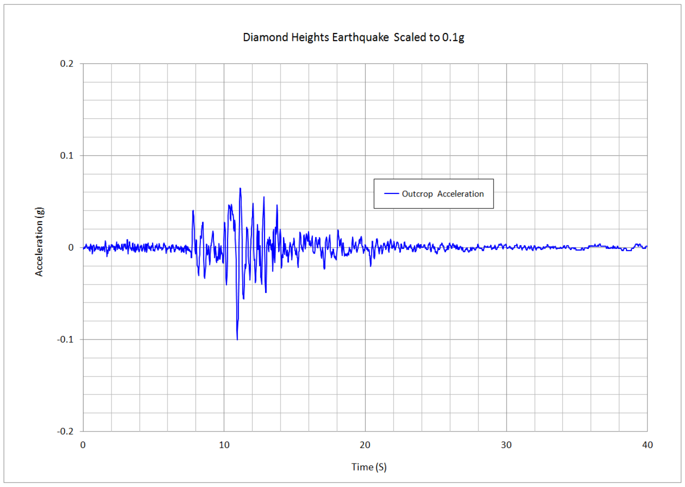


Figure 4.150 Diamond Heights acceleration time history

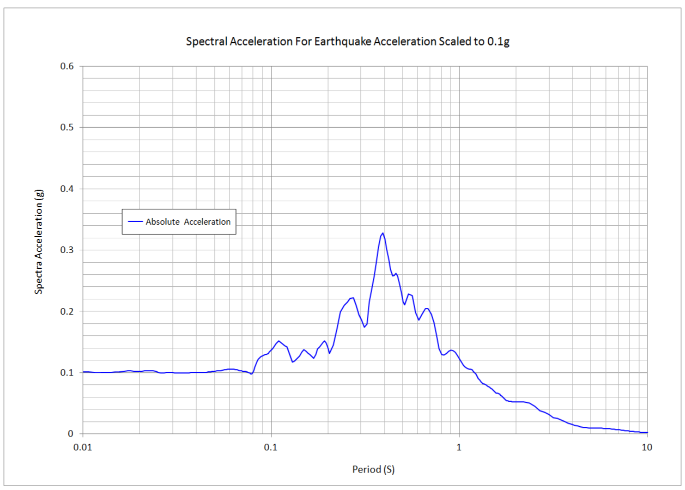


Figure 4.151 Spectral acceleration for input earthquake

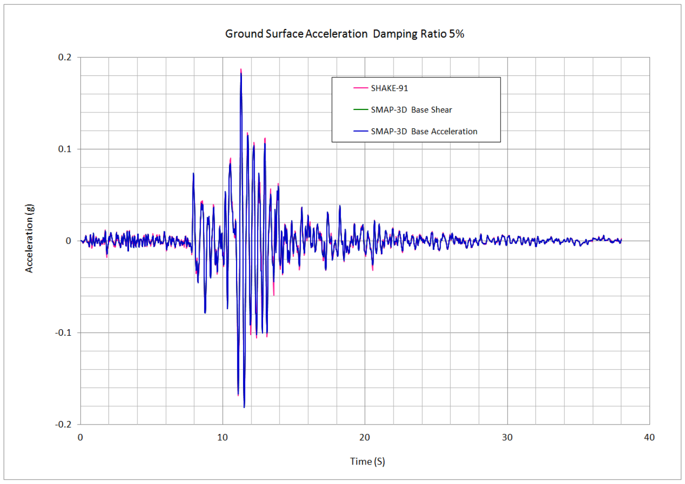


Figure 4.152 Ground surface accelerations

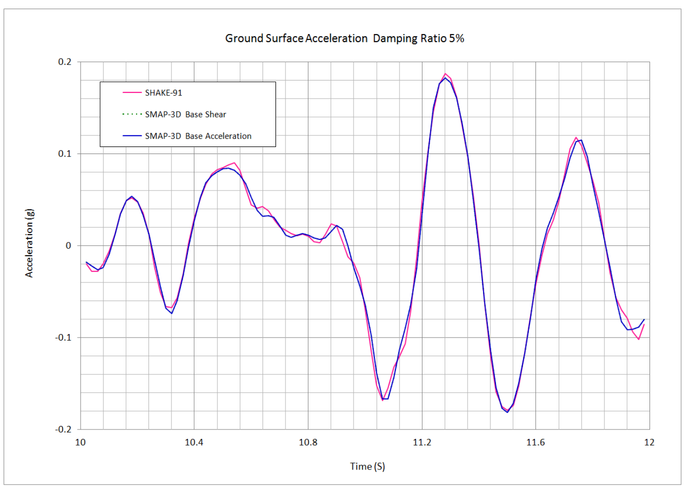


Figure 4.153 Ground surface accelerations between 10 and 12 sec.

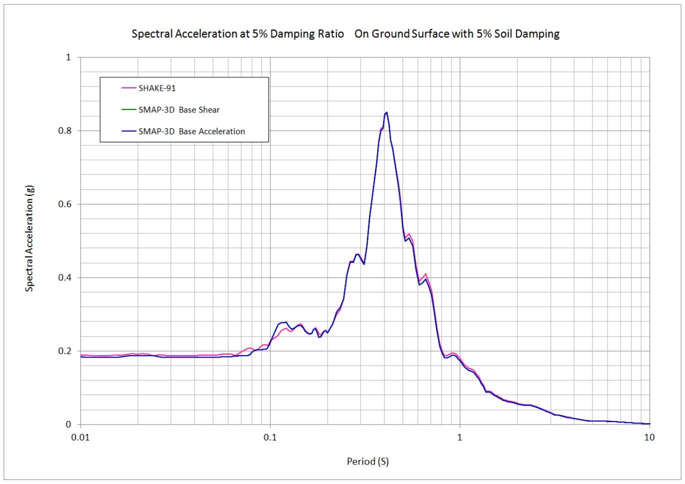


Figure 4.154 Spectral accelerations on ground surface

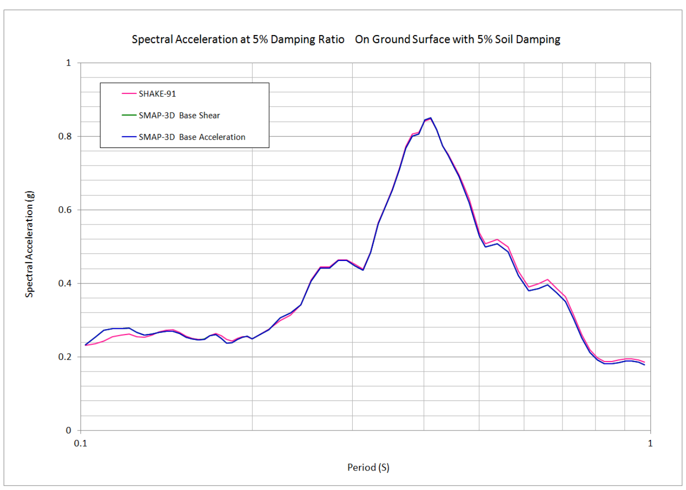


Figure 4.155 Spectral accelerations between 0.1 and 1 sec.

4.33 Silo Lining Analysis

This example is to solve the lining stresses developed in underground silo subjected to residual water pressure. This silo structure in Gyeongju, South Korea, was constructed to store the low-and-intermediate-level radioactive waste.

Figures 4.156 and 4.157 show finite element meshes and close-up view around silo, respectively. This 3 dimensional model consists of 65,598 continuum, 792 joint, 1,584 shell elements and 71,867 nodes. Program used thin shell elements to model reinforced concrete lining.

Table 4.8 lists material properties and Figure 4.158 shows schematic view of detailed silo lining structure. Table 4.9 lists lining thickness and reinforcement. Figure 4.159 shows silo lining material numbers. Table 4.10 shows schematically the sequence of silo construction including residual water pressure applied at step 5. Figure 4.160 shows key locations along the silo lining.

The following is a partial listing of graphical outputs at load step 5 when lining is subjected to residual water pressure head of 17.47m:

Figure 4.161	Deformed shape of silo lining
Figure 4.162	Dome deflection along A-B
Figure 4.163	Storage wall radial displacement along C-D
Figure 4.164	Dome lining inner hoop stress along A-B
Figure 4.165	Dome outer rebar meridian stress along A-B
Figure 4.166	Storage wall lining inner hoop stress along C-D
Figure 4.167	Storage wall outer rebar meridian stress along C-D

SMAP-3D results are compared with SMAP-2D results to verify the validity of the solution. As shown, SMAP-3D results are very close to SMAP-2D results. It seems that the reinforced concrete lining is in safe condition under the applied residual water pressure head of 17.47m.

Note: It takes about 5 hours of run time in the following computer:
64 Bit Windows 11, 8 Core i7-11700F CPU, 16 GB of DDR4 Ram.

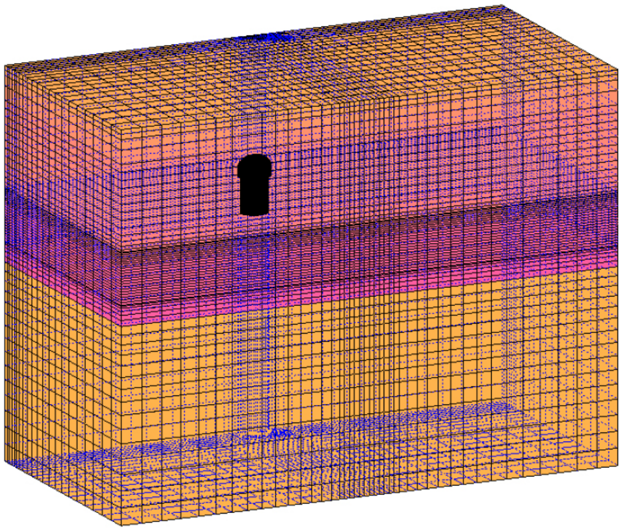


Figure 4.156 Finite element meshes

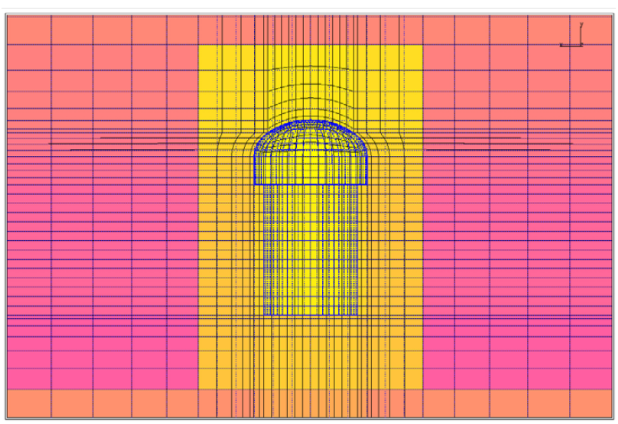


Figure 4.157 Finite element meshes around silo

Table 4.8 Material properties

Ground Layer	Unit weight (KN/m^3)	Young's modulus (MPa)	Poisson's ratio	Internal Friction Angle
Soil Layer	18.56	0.124×10^4	0.33	30°
Weathering Rock	20.52	0.342×10^4	0.30	38°
Rock	26.28	8.260×10^4	0.27	43°
Shotcrete	23.0	24,500	0.167	-
Concrete	23.5	29,500	0.167	-
Rebar	-	210,000	0.25	-

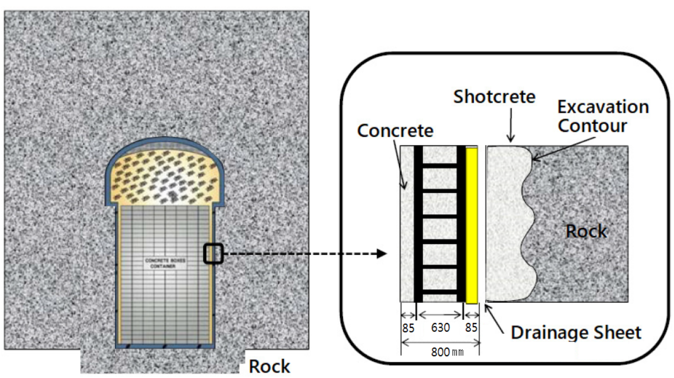


Figure 4.158 Schematic view of detailed silo lining structure

Table 4.9 Silo lining thickness and reinforcement

Material Number	Thickness (Meter)	Steel Ratio (%)		Location
		Hoop	Meridian	
1	1.211	0.85	0.85	Dome Crown
4	1.246	0.83	0.83	Dome Crown
5	1.279	0.81	0.81	Dome Crown
6	1.328	0.78	0.78	Dome Crown
7	1.398	0.74	0.74	Dome Crown
8	1.475	0.70	0.70	Dome Crown
9	1.547	0.67	0.67	Dome Crown
10	1.594	0.65	0.65	Dome Crown
11	1.600	0.65	0.65	Dome Wall
12	1.200	0.86	0.86	Dome Bottom
13	0.800	1.29	1.29	Storage Wall
14	1.200	0.86	0.86	Storage Bottom
15	1.200	0.86	0.86	Storage Bottom

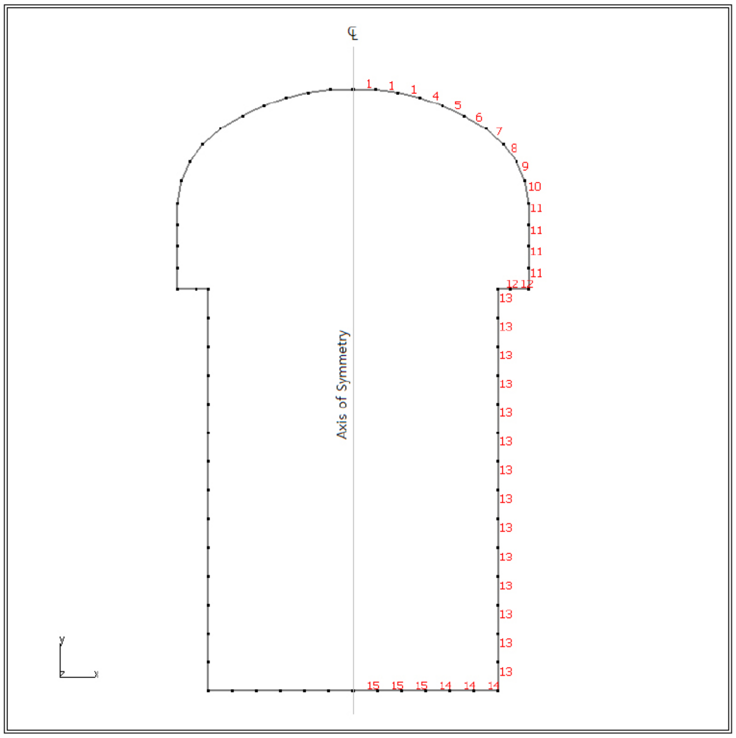
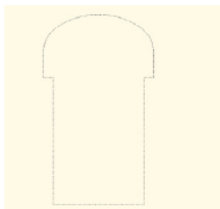
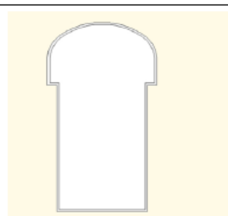
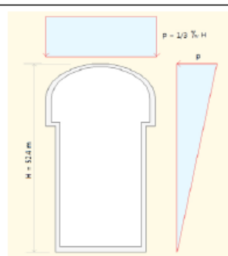


Figure 4.159 Silo lining material numbers

Table 4.10 Construction sequence

Step	Construction State	Descriptions
1,2		In Situ Ko State
3		Excavate Silo and install Shotcrete of 50cm Thickness
4		Install Reinforced Concrete Lining with its Own Self Weight
5		Lining is Subjected to Residual Water Pressure Head of 17.47m

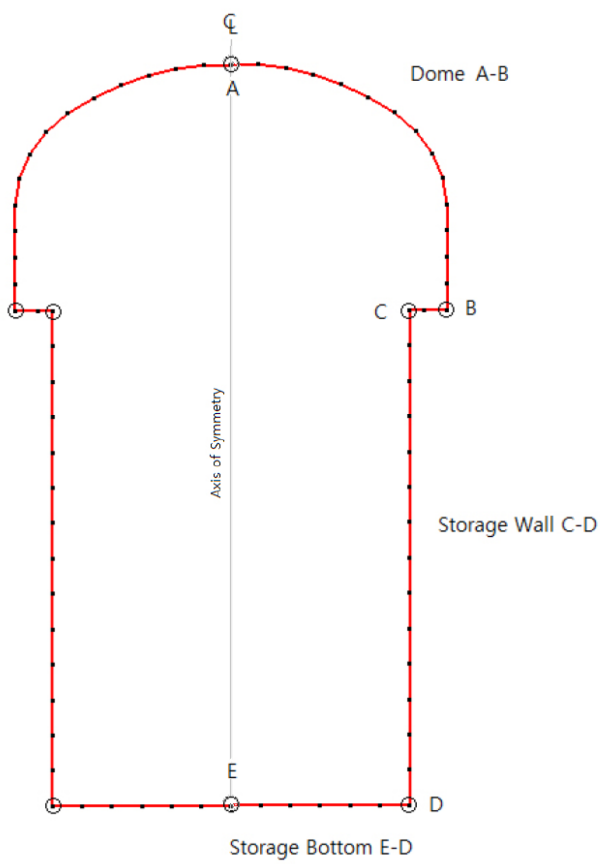


Figure 4.160 Key locations along silo lining

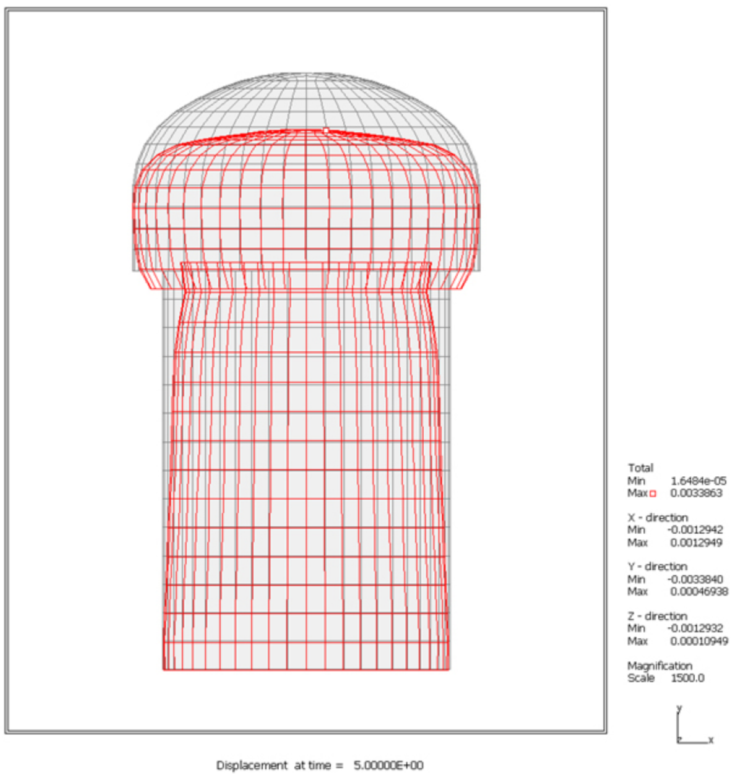


Figure 4.161 Deformed shape of silo lining

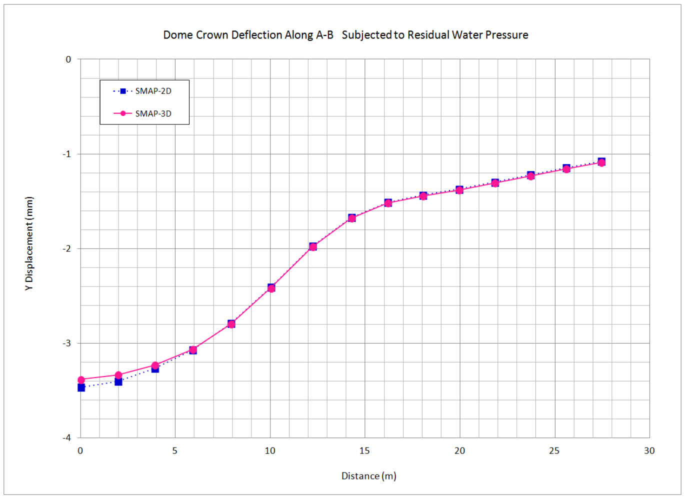


Figure 4.162 Dome deflection along A-B

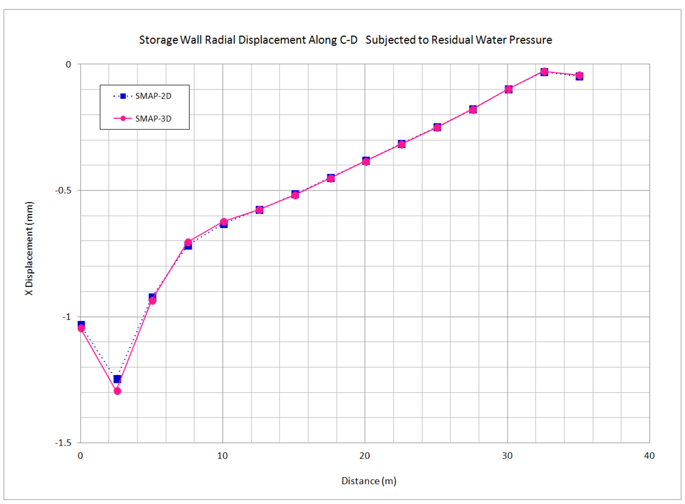


Figure 4.163 Storage wall radial displacement along C-D

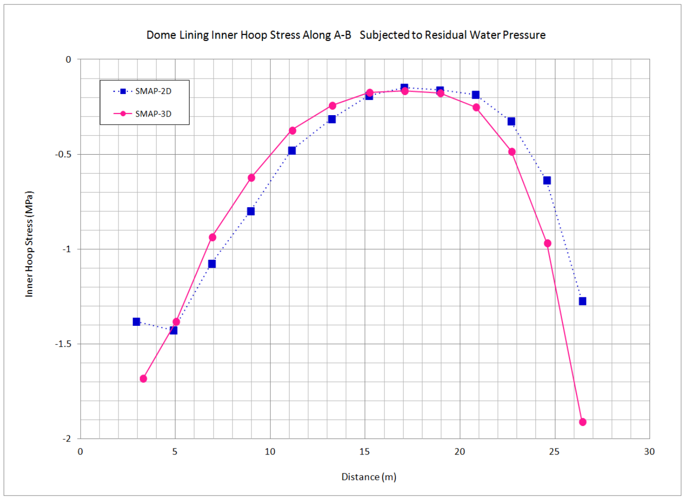


Figure 4.164 Dome lining inner hoop stress along A-B

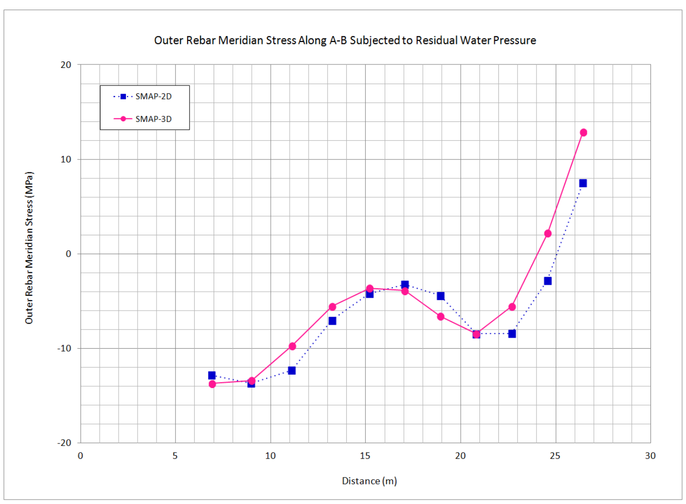


Figure 4.165 Dome outer rebar meridian stress along A-B

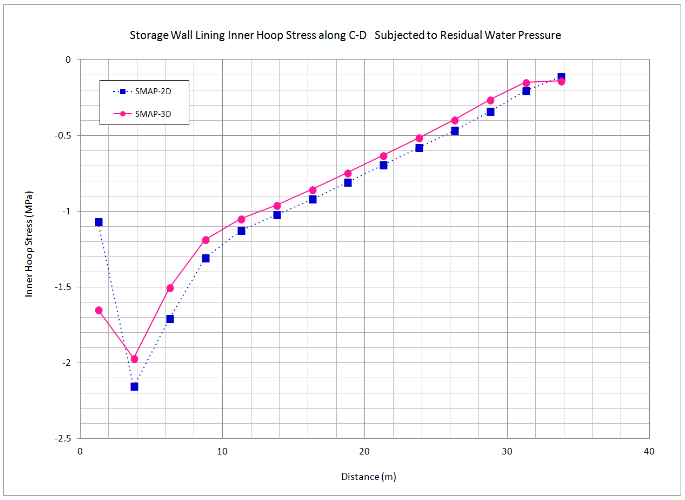


Figure 4.166 Storage wall lining inner hoop stress along C-D

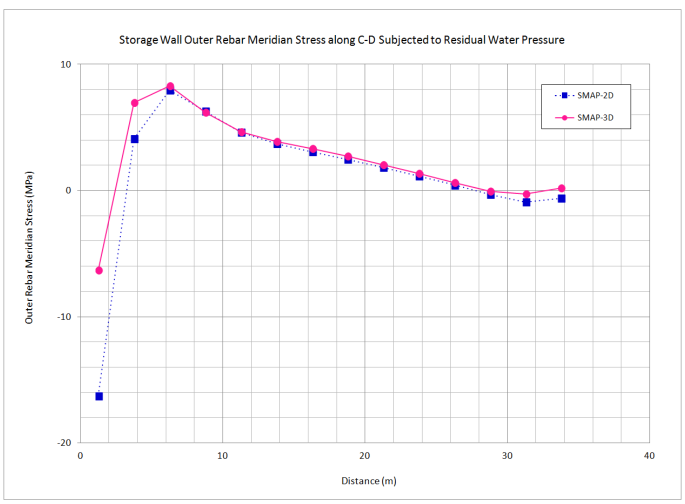


Figure 4.167 Storage wall outer rebar meridian stress along C-D

4.34 Liquefaction Analysis with PM4Sand

It should be noted that PM4Sand in SMAP-3D works only for plane strain condition. It does not work for general 3 dimensional condition.

The main objective of this example is to verify PM4Sand model implemented in SMAP-3D finite element program. The PM4Sand model (Boulanger and Ziotopoulou, 2017) is the effective stress material model which is calibrated in the finite difference program FLAC 8.0 (Itasca 2016) for the plane strain condition.

As first step, several different stress paths for a single element are considered to verify implementation; including drained and undrained conditions, monotonic and cyclic loadings, and isotropic and K_O initial conditions. Figure 4.168 shows isotropic consolidated drained cyclic direct simple shear test. All other results are summarized in the file; Single Element Stress-Strain Response of PM4Sand Model.pdf

This analysis is to solve the free-field seismic response of the soil profile, shown in Figure 4.169 along with material properties, subjected to earthquake excitation from the bedrock.

This problem is the same as the problem in the report (Chen and Arduino, 2021). A 6 m soil profile is subjected to Loma Prieta earthquake in 1989 (RSN766) as outcrop to the elastic half space. Earthquake time history with peak acceleration 0.37g and its spectral acceleration are shown in Figures 4.170 and 4.171, respectively.

Figures 4.172 and 4.173 show computed profiles of peak ground accelerations and maximum shear strains, respectively, compared with SHAKE 91 and DEEP SOIL. Note that this linear elastic analysis is performed to check the initial stresses and boundary conditions prior to liquefaction analysis by scaling down peak acceleration to 0.02g.

Results of liquefaction analysis are presented in the following:

Figure 4.174	Maximum acceleration profile (PGA)
Figure 4.175	Maximum displacement profile
Figure 4.176	Maximum shear strain profile
Figure 4.177	Maximum r_u profile
	r_u = Excess Pore Pressure / Initial Effective Ver. Stress

Overall, PM4Sand in SMAP-3D is performing very well in predicting the stress-strain responses compared to the calibrated FLAC results.

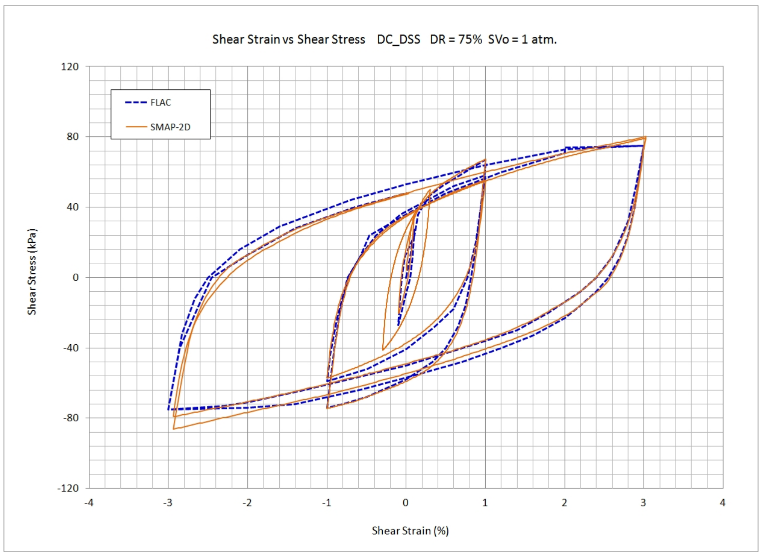
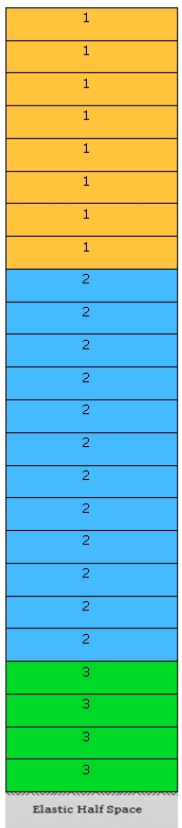


Figure 4.168 Isotropic consolidated drained cyclic direct simple shear test



No	γ (kN/m^3)	G (MPa)	β (%)
1	15.8	Dry n = 39.66% PM4Sand Dr = 0.466 Go = 468.3 hpo = 0.463	3.55
2	19.7	Saturated n = 39.59% PM4Sand Dr = 0.466 Go = 584.1 hpo = 0.45	3.55
3	19.7	Saturated n = 39.59% Elastic G = 66.54	3.55
	19.7	Elastic G = 66.54	

Total Dimension

Y - direction
Min -6.0000
Max 0

Figure 4.169 Finite element meshes and material properties

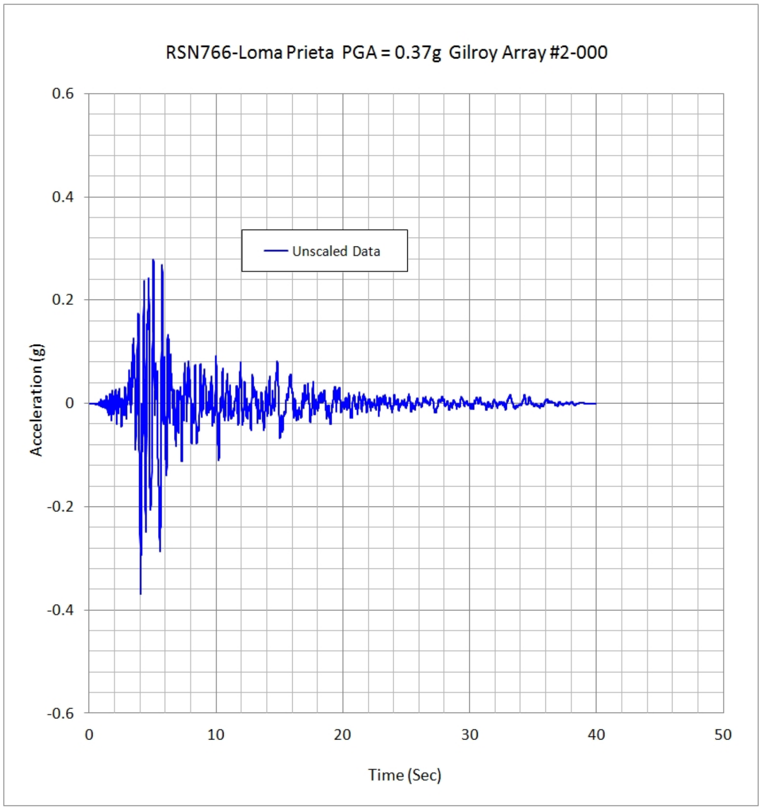


Figure 4.170 Loma Prieta (RSN766) acceleration time history

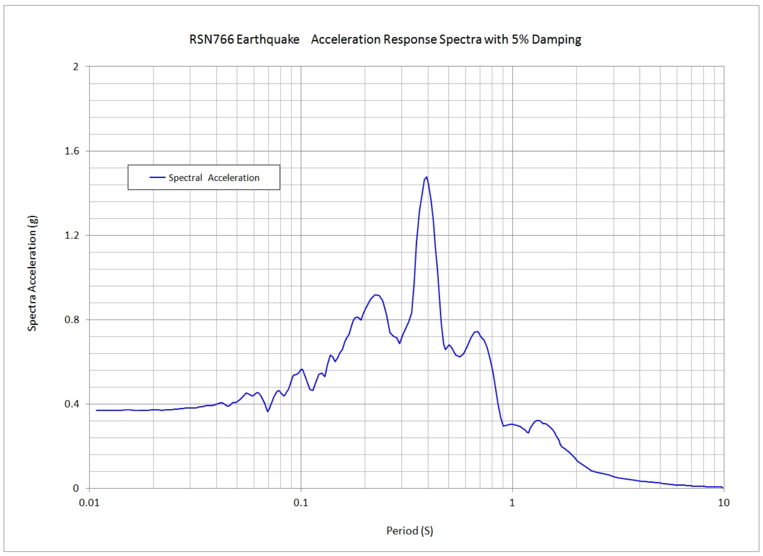


Figure 4.171 Spectral acceleration for input earthquake (RSN766)

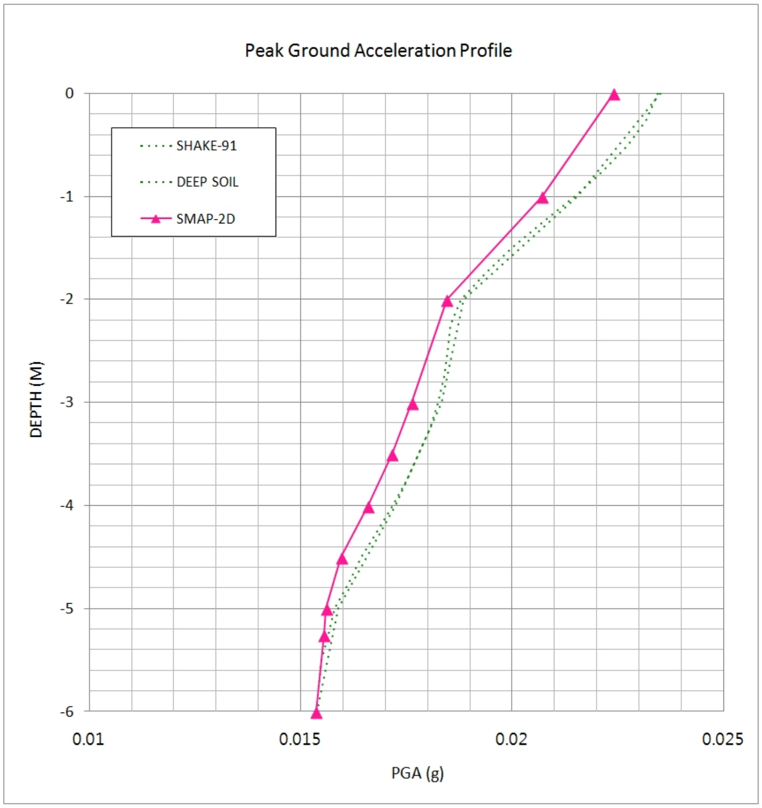


Figure 4.172 Peak ground acceleration profile, Elastic analysis

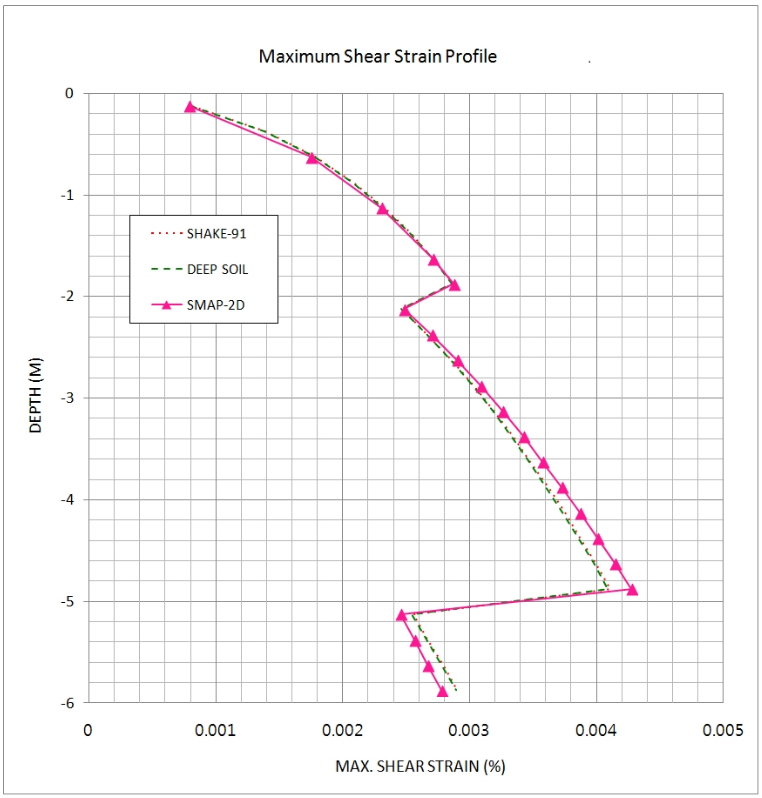


Figure 4.173 Maximum shear strain profile, Elastic analysis

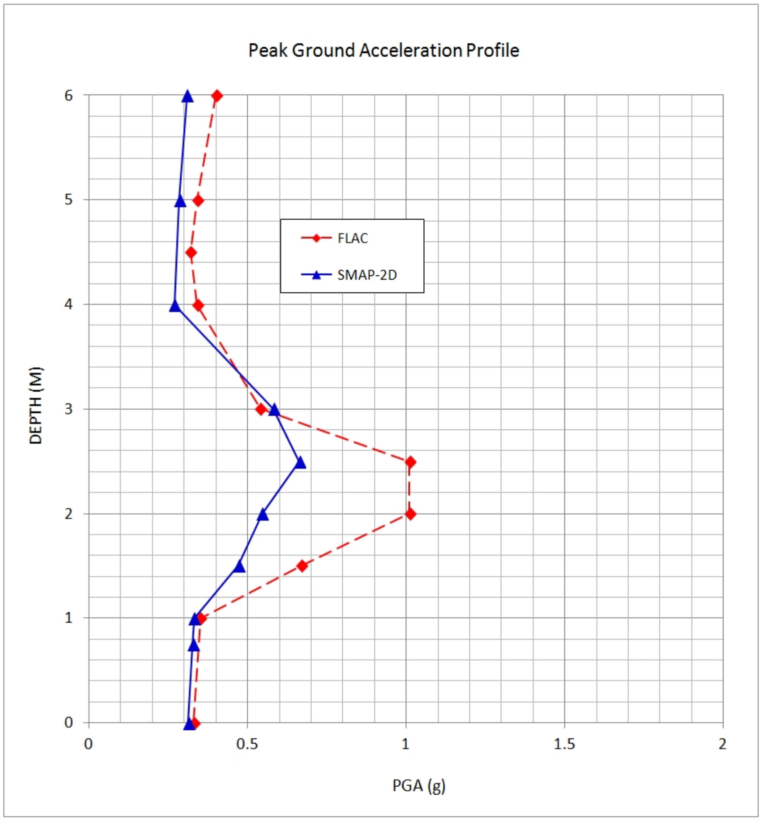


Figure 4.174 Maximum acceleration profile, Liquefaction analysis

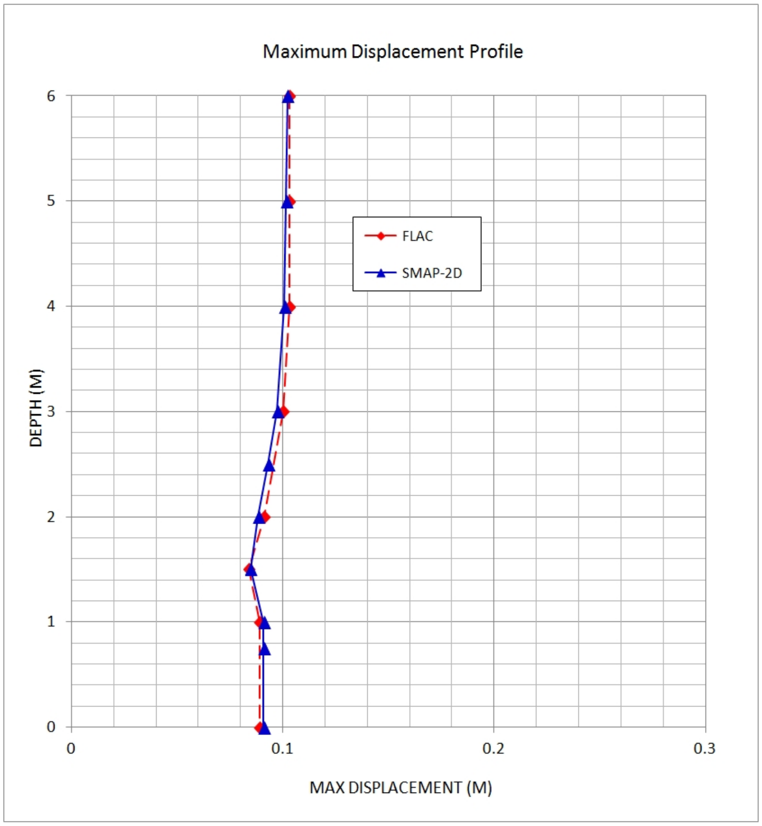


Figure 4.175 Maximum displacement profile, Liquefaction analysis

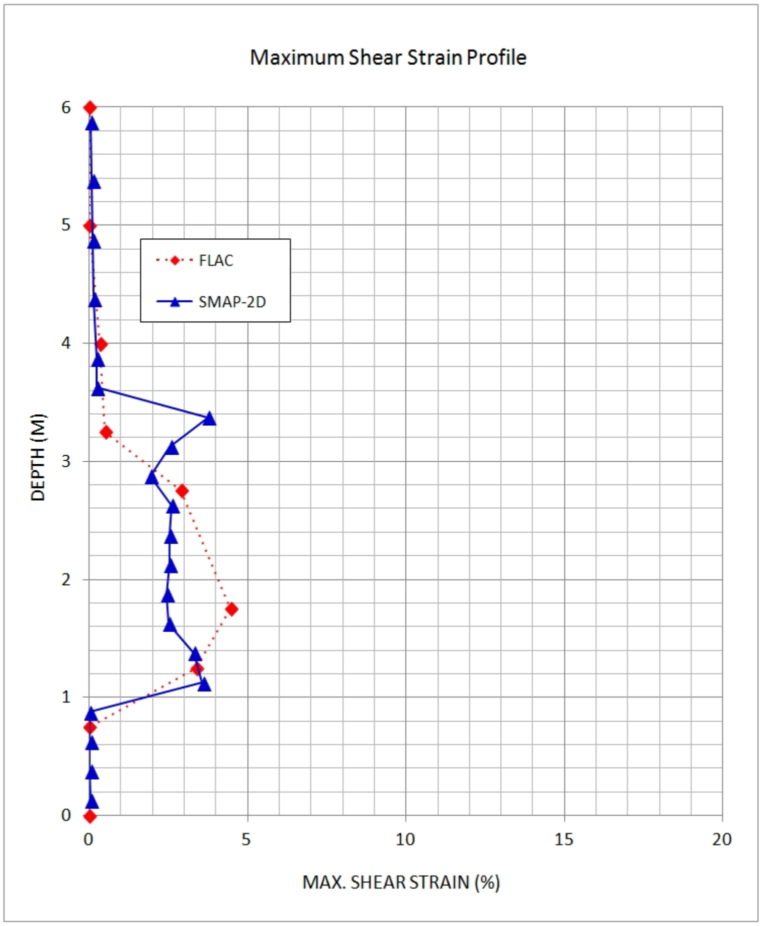


Figure 4.176 Maximum shear strain profile. Liquefaction analysis

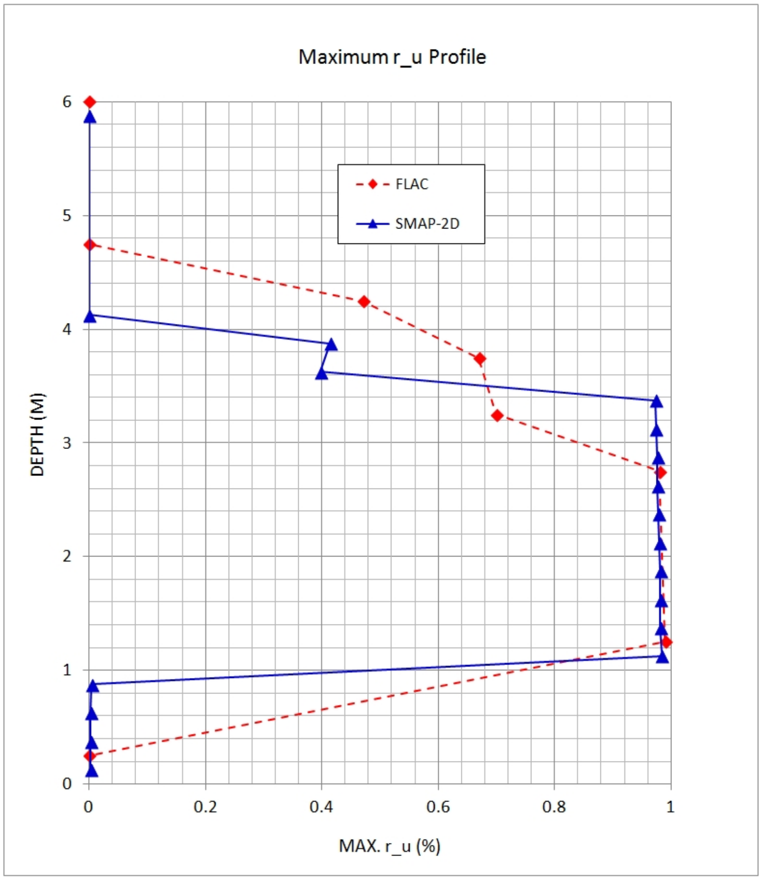


Figure 4.177 Maximum r_u profile, Liquefaction analysis

Group Mesh
Example Problem

Group Mesh Generator is a two-dimensional CAD program specially designed to build group mesh which can be used to generate finite element mesh with the aid of program ADDRGN-2D. Group Mesh User's Manual describes all the basic functions associated with group mesh generation and modifications.

Six example problems are presented:

1. Arch Tunnel
   Shows step by step procedure to create and modify group meshes.
2. NATM Tunnel
   Builds group mesh for typical NATM tunnel.
3. Excavation
   Builds group mesh for typical multi-step excavations performed near the existing structure.
4. Buried Pipe
   Builds group mesh for typical pipe buried in the trench followed by multi-step embankment lifts.
5. Arch Warehouse
   Builds group mesh for typical arch warehouse structure.
6. Finite Element Mesh Modification
   Illustrates how to modify existing finite element meshes using Mesh Generator.

5.1 Arch Tunnel

The main objective of this first example is to show the step by step procedure to create and modify group meshes.

This example has the following three parts:

Part 1 : Creating Arch Tunnel (Figure 5.1)

- Create group mesh
- Set built-in base mesh
- Draw arch tunnel
- Plot finite element mesh

Part 2 : Adding Rock Bolts (Figure 5.2)

- Open the group mesh file in part 1
- Add three rock bolts
- Plot finite element mesh

Part 3 : Adding Utility Tunnel (Figure 5.3)

- Open the group mesh file in part 2
- Remove the first rock bolt
- Change the second rock bolt length
- Replace the third rock bolt by utility tunnel
- Plot finite element mesh

Table 5.1 shows the construction sequence.

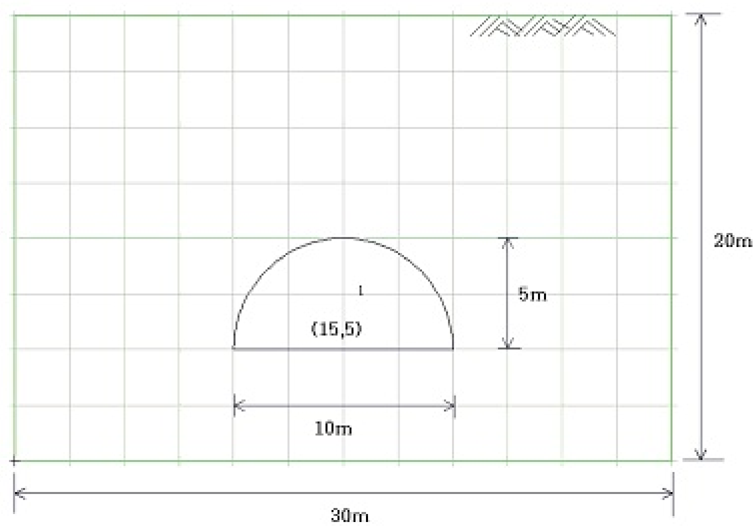


Figure 5.1 Arch tunnel (Part 1)

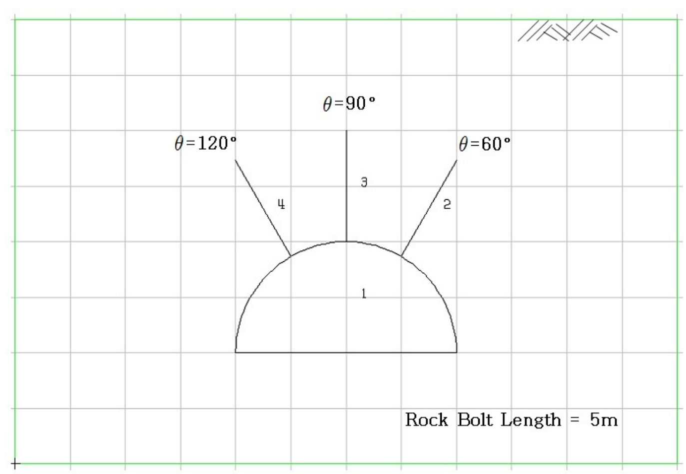


Figure 5.2 Arch tunnel with rock bolts (Part 2)

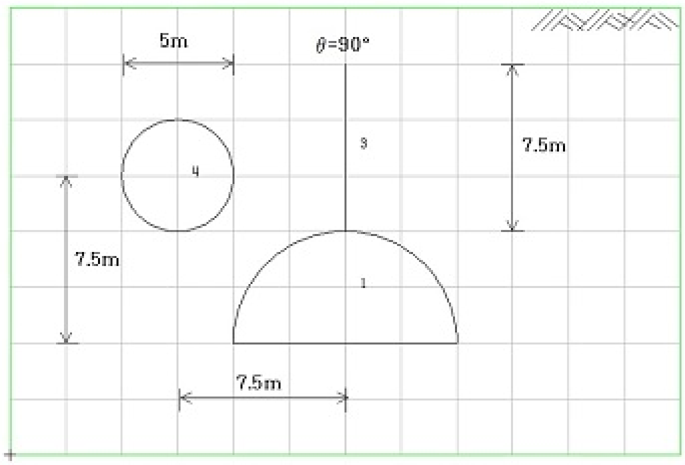


Figure 5.3 Arch tunnel with utility tunnel (Part 3)

Step No	Description
1, 2	In-Situ Stress
3	Arch Tunnel Excavation & Lining Installation
4	Rock Bolt Installation
5	Utility Tunnel Construction

Table 5.1 Construction sequence

5.1.1 Part 1: Creating Arch Tunnel

Part 1 consists of the following main actions:

- Create group mesh
- Set built-in base mesh
- Draw arch tunnel
- Plot finite element mesh

Step 1: Group Mesh Generator (New)

Access Group Mesh Generator by selecting the following menu items in SMAP (Figure 5.4):
Run → Mesh Generator → Group Mesh → New

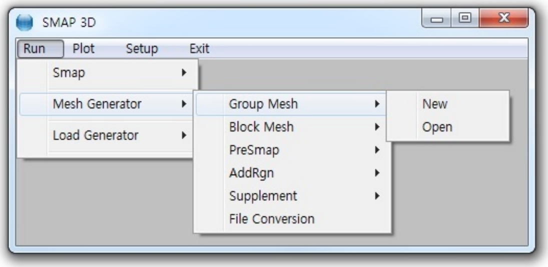


Figure 5.4 Accessing group mesh generator (New)

Step 2: Group Input (New)

Select Built-in Base Mesh in Figure 5.5.
Click OK.

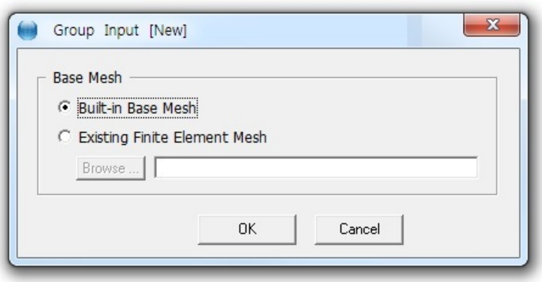


Figure 5.5 Group input (New)

Step 3: Group Menu and Dialog

Click Group menu in PLOT-2D as shown in Figure 5.6.

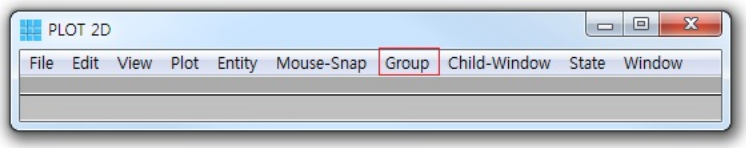


Figure 5.6 Group menu

Group dialog in Figure 5.7 is displayed with initial default values.

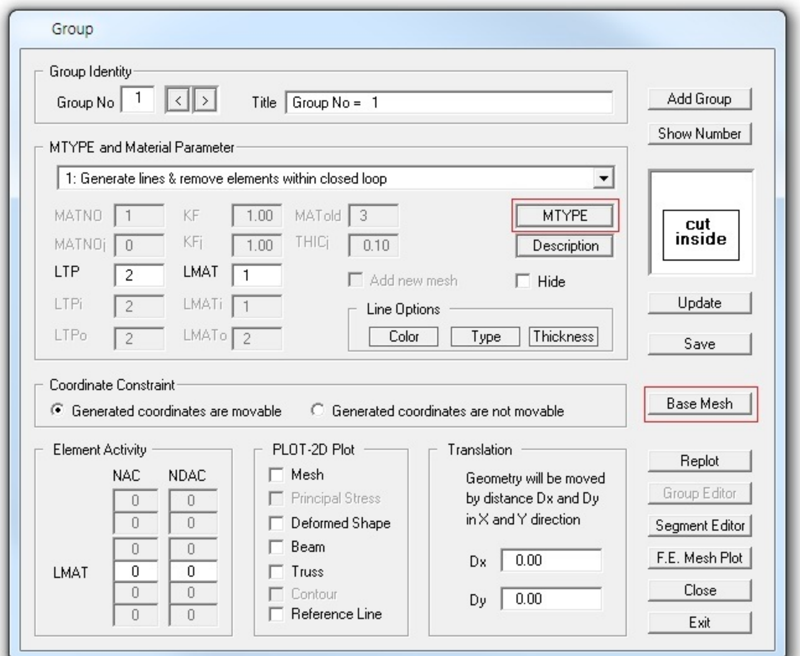


Figure 5.7 Group dialog with initial default values

Step 4: Built-in Base Mesh

Click Base Mesh button in Group dialog.

Fill in input fields for Built-in Base Mesh as shown in Figure 5.8.

Click OK.

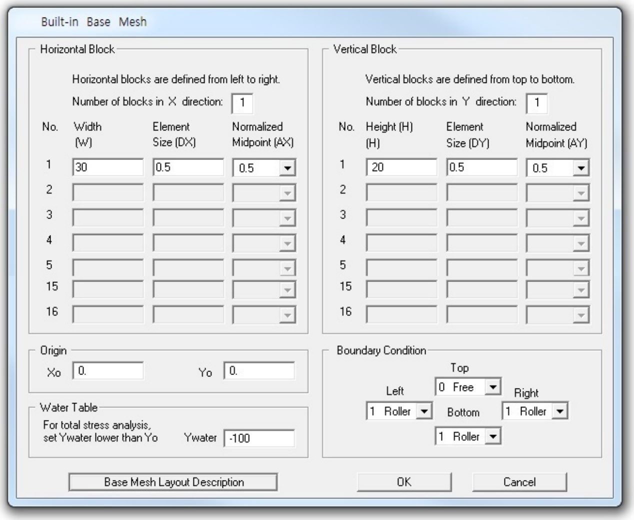


Figure 5.8 Built-in base mesh dialog

Figure 5.9 shows Base Mesh with dimensions of 30m x 20m on drawing board in PLOT-2D.

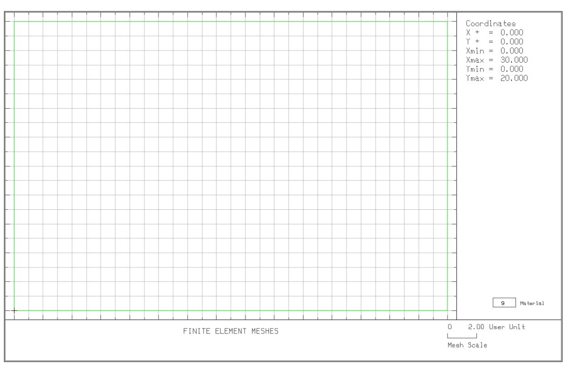


Figure 5.9 Base mesh on drawing board

Step 5: MTYPE

Click MTYPE button in Group dialog.
Select MTYPE=3 in MTYPE dialog in Figure 5.10.
Click OK.

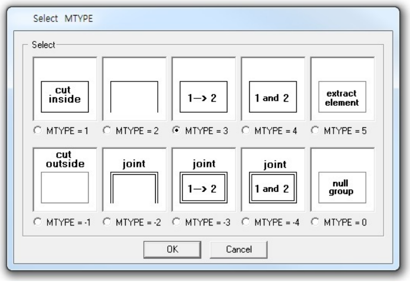


Figure 5.10 MTYPE dialog

Fill in input fields for Group dialog as shown in Figure 5.11.

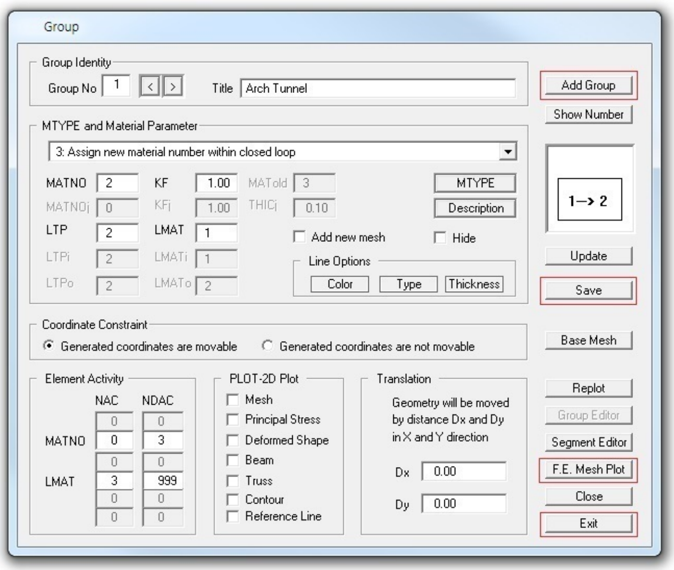


Figure 5.11 Group dialog with MTYPE = 3

Step 6: Mouse Snap

Click Mouse-Snap menu in PLOT-2D.
Select Snap to Grid in Figure 5.12. Click OK.

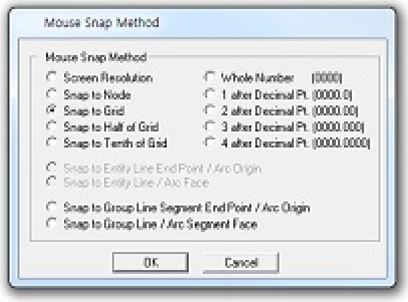


Figure 5.12
Mouse snap dialog

Step 7: Add Group

Click Add Group button in Group dialog.

Table 5.2 summarizes group parameters used for arch tunnel.

Group No	MTYPE	Description	Element Type	Mat. Np.	Element Activity	
					NAC	NDAC
1	3	Core	Cont.	MATNO=2	0	3
		Lining	Beam (LPT=2)	LMAT=1	3	999

Group No	Seg. No	Line Segment				Arc Segment						IEND
		Beginning Point		Ending Point		Origin		Radius and Angle				
		X	Y	X	Y	X_o	Y_o	R_X	R_Y	Θ_b	Θ_e	
1	1	10	5	20	5							2
	2					15	5	5	5	0	180	2

Table 5.2 Group parameters for arch tunnel

Step 8: Line Segment

Click Draw button in Line Segment dialog in Figure 5.13.

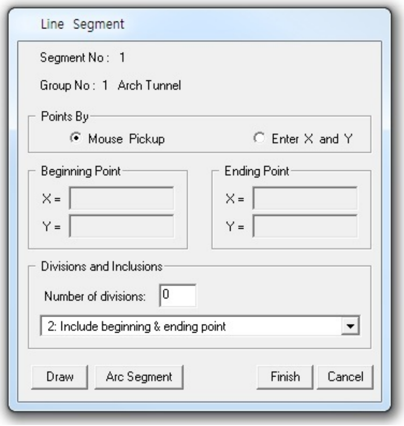


Figure 5.13 Line segment dialog

Click the mouse where the line begins and then click the mouse where the line ends as shown in Figure 5.14.

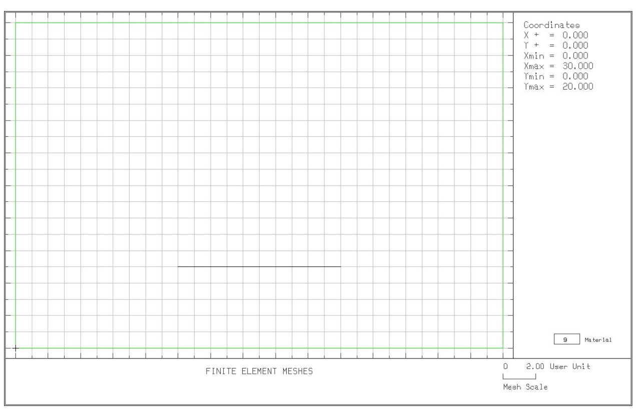


Figure 5.14 Line segment on drawing board

Step 9: Arc Segment

Click Arc Segment button in Line Segment dialog.
Fill in input fields for Arc Segment as shown in Figure 5.15.
Click Draw.

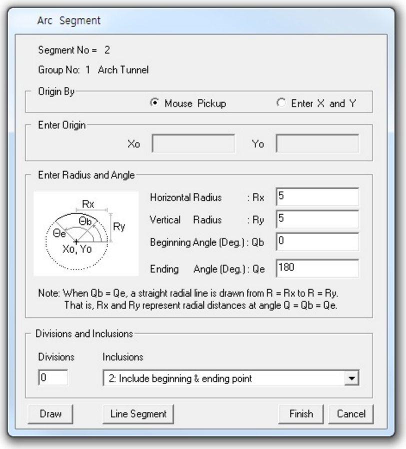


Figure 5.15 Arc segment dialog

Press down and hold mouse button on the drawing board.
Drag the mouse to the location of arc origin and then release the mouse button as shown in Figure 5.16.
Click Finish.

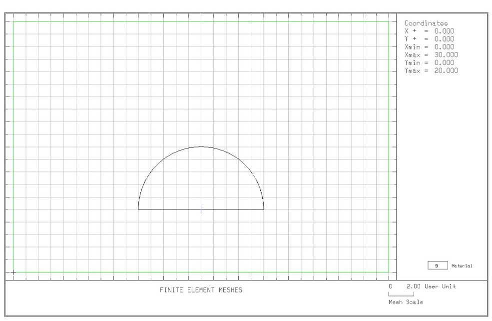


Figure 5.16 Arc segment on drawing board

Step 10: Save

Click Save button in Group dialog.
Group.Meg is saved as shown in Figure 5.17.

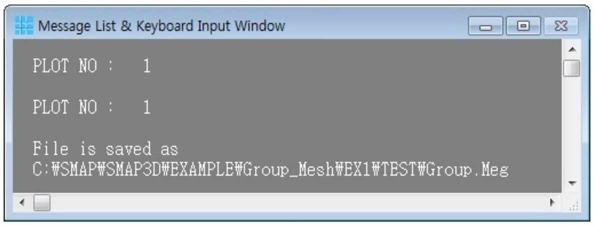


Figure 5.17 Message for file save

Step 11: Finite Element Mesh

Click F.E. Mesh Plot button in Group dialog.
Click Yes in Figure 5.18.

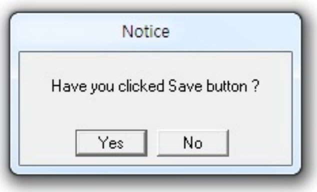


Figure 5.18 Notice for finite element mesh plot

Please Wait... message in Figure 5.19 is shown on the screen while generating finite element mesh plot.

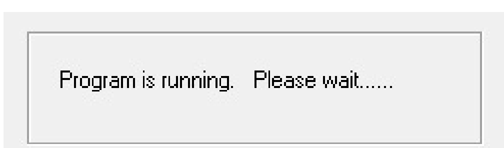


Figure 5.19 Notice while generating finite element mesh plot

Once finished, finite element mesh file is generated as Group.Mes in the directory Plot_Mesh as shown in Figure 5.20 along with finite element mesh plot in Figure 5.21.

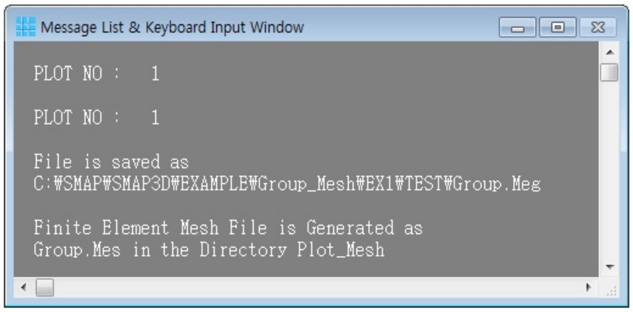


Figure 5.20 Message for finite element mesh file

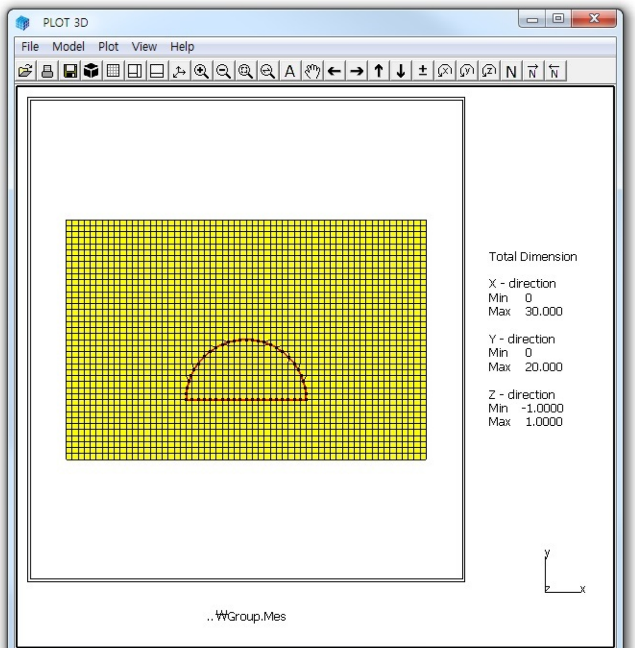


Figure 5.21 Finite element mesh plot

Step 12: Exit

Click Exit button in Group dialog.
Click OK in Exit dialog as shown in Figure 5.22.

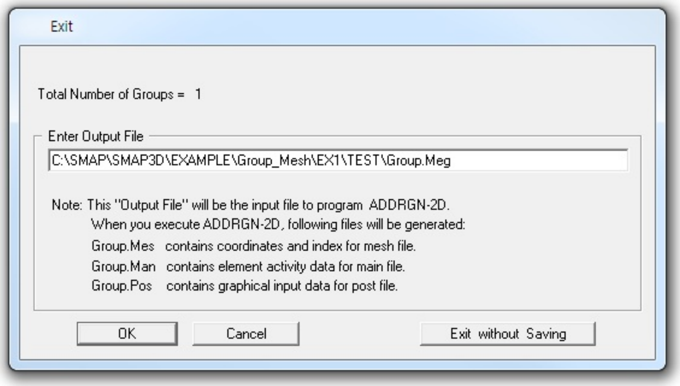


Figure 5.22 Exit dialog

5.1.2 Part 2: Adding Rock Bolts

Part 2 consists of the following main actions:

- Open the group mesh file in part 1
- Add three rock bolts
- Plot finite element mesh

Step 13: Group Mesh Generator (Open)

Access Group Mesh Generator by selecting the following menu items in SMAP (Figure 5.4):
Run → Mesh Generator → Group Mesh → Open

Step 14: Group Input (Open)

File open dialog will be displayed as in Figure 5.23.
Select group mesh file Group.Meg in Part 1 and click Open.

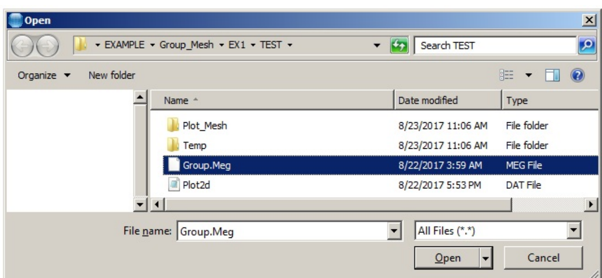


Figure 5.23 File open dialog

Step 15: Group Menu and Dialog

Click Group menu in PLOT-2D as shown in Figure 5.6.
Group dialog for Group No 2 is displayed with initial default values.

Step 16: MTYPE

Click MTYPE button in Group dialog.
Select MTYPE=2 in MTYPE dialog in Figure 5.10.
Click OK.

Step 17: Group No 2 for Rock Bolt 1

Table 5.3 summarizes group parameters for rock bolts.
Rock bolt is modeled by a straight radial line in Arc Segment.

Group No	Bolt No	MTYPE	Elem. Type (LTP)	Mat. No (LMAT)	Element Activity		Radius and Angle				IEND
					NAC	NDAC	R_X	R_Y	Θ_b	Θ_e	
2	Bolt-1	2	Truss (3)	1	4	999	5	10	60	60	-2
3	Bolt-2	2	Truss (3)	1	4	999	5	10	90	90	-2
4	Bolt-3	2	Truss (3)	1	4	999	5	10	120	120	-2

Table 5.3 Group parameters for rock bolts

Group No 2 represents Rock Bolt 1 with a length of 5m at 60 degrees.
Fill in input fields for Group dialog as shown in Figure 5.24.

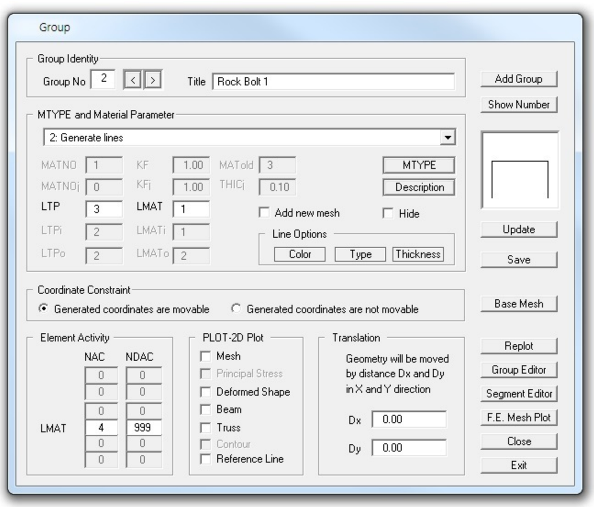


Figure 5.24 Group dialog for Rock Bolt 1

Step 18: Mouse Snap

Click Mouse-Snap menu in PLOT-2D.
Select Snap to Grid in Figure 5.12.
Click OK.

Step 19: Add Group

Click Add Group button in Group dialog.

Step 20: Arc Segment

Click Arc Segment button in Line Segment dialog.
Fill in input fields for Arc Segment as shown in Figure 5.25.
Click Draw.

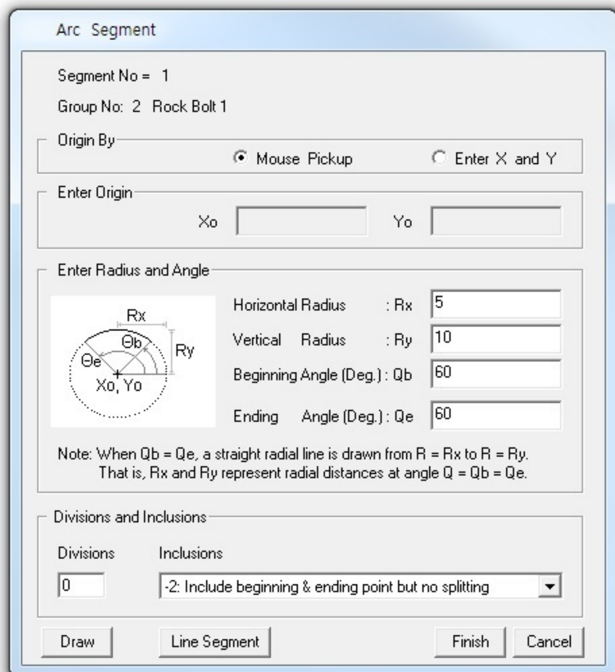


Figure 5.25 Arc segment dialog for Rock Bolt 1

Press down and hold mouse button on the drawing board. Drag the mouse to the location of arc origin and then release the mouse button as shown in Figure 5.26. Click Finish.

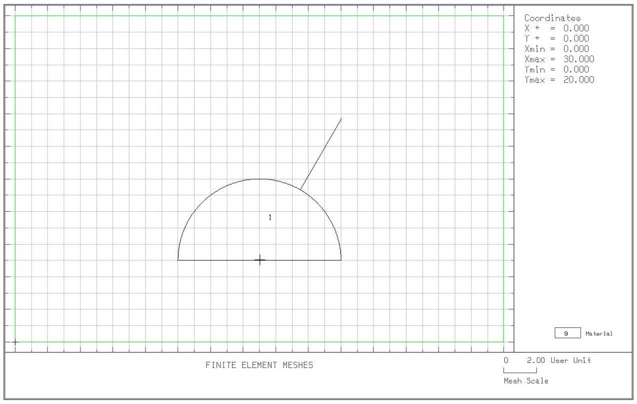


Figure 5.26 Rock Bolt 1 on drawing board

Step 21: Group No 3 & 4 for Rock Bolt 2 & 3

Repeat Steps 16 through 20 to add rock bolts at 90 and 120 degrees. All three rock bolts are shown on drawing board in Figure 5.27. Click Save button in Group dialog.

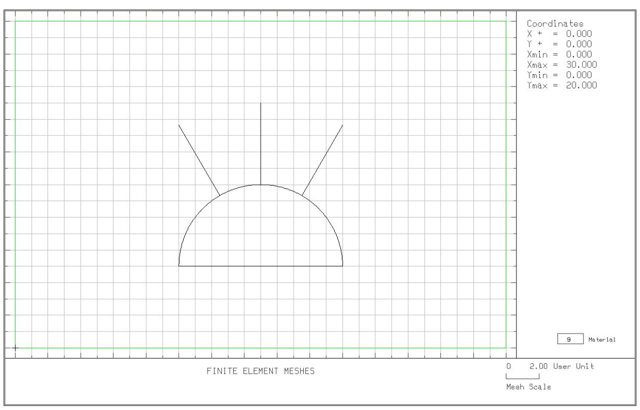


Figure 5.27 All three rock bolts on drawing board

Step 22: Finite Element Mesh

Click F.E. Mesh Plot button in Group dialog.
Follow the same procedure as in Steps 10 and 11.
Finite element meshes are shown in Figure 5.28

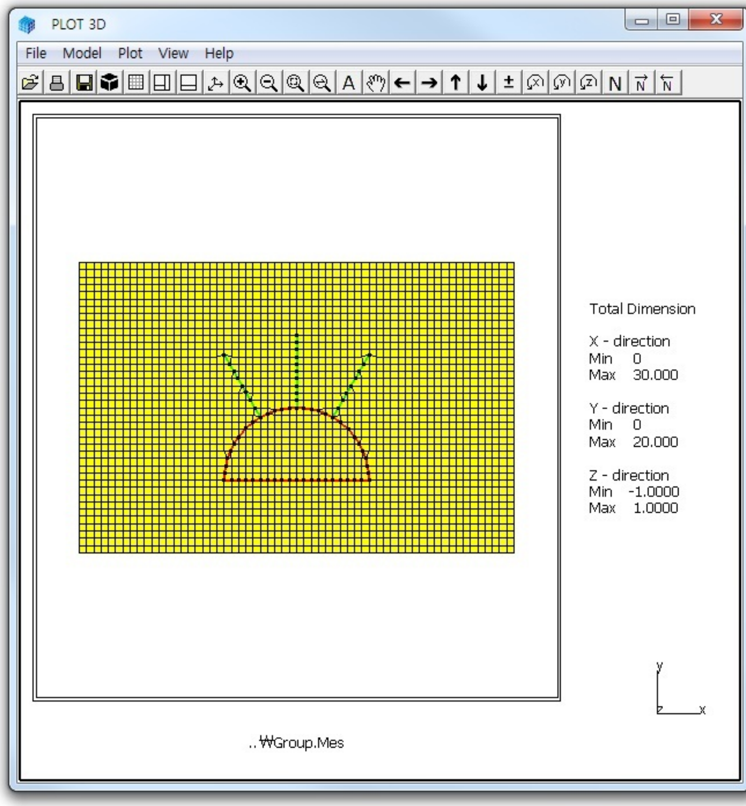


Figure 5.28 Finite element mesh plot

Step 23: Exit

Click Exit button in Group dialog.
Click OK in Exit dialog as in Figure 5.22.

5.1.3 Part 3: Adding Utility Tunnel

Part 3 consists of the following main actions:

- Open the group mesh file in part 2
- Remove the first rock bolt
- Change the second rock bolt length
- Replace the third rock bolt by utility tunnel
- Plot finite element mesh

Step 24: Open Group Mesh File in Part 2

Follow Steps 13 through 15 to open Group dialog for Group No 2.

Step 25: Remove Rock Bolt 1

Select Group No 2 in Group dialog.
Click MTYPE button in Group dialog.
Select MTYPE=0 in MTYPE dialog in Figure 5.10.
Click OK.

Click Update and then Replot buttons in Group dialog.
A new plot with the Group No 2 missing is displayed in Figure 5.29

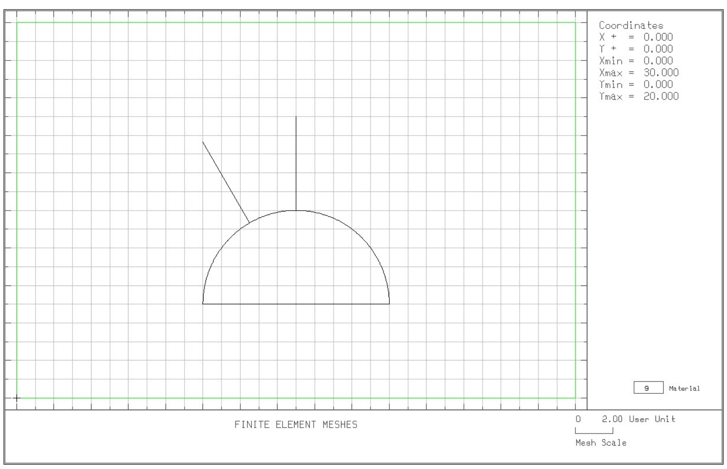


Figure 5.29 Rock Bolt 1 removed on drawing board

Step 26: Change Length of Rock Bolt 2

Select Group No 3 in Group dialog.
Click Edit Group button in Group dialog.

Click Edit button in Edit Segment dialog in Figure 5.30.

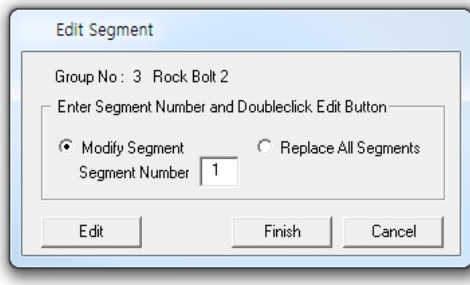


Figure 5.30
Edit segment dialog
for Group No 3

Fill in input fields for Arc Segment dialog as shown in Figure 5.31.
Click Draw and then Finish in Arc Segment dialog.
Click Finish in Edit Segment dialog.

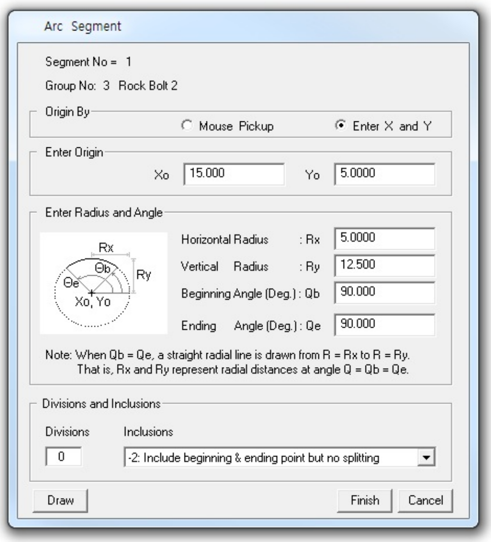


Figure 5.31
Arc segment dialog with
rock bolt length modified

Figure 5.32 shows a new plot with longer Rock Bolt 2.

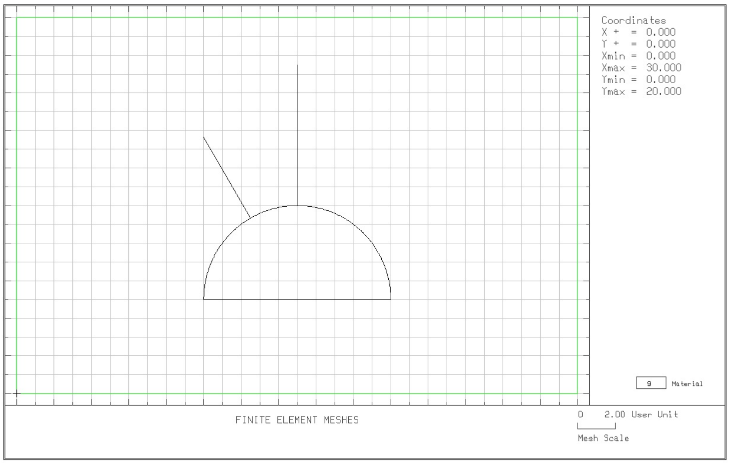


Figure 5.32 Longer Rock Bolt 2 on drawing board

Step 27: Replace Rock Bolt 3 by Utility Tunnel

Select Group No 4 in Group dialog.

Click MTYPE button in Group dialog.
Select MTYPE=1 in MTYPE dialog in Figure 5.10.
Click OK.

Fill in input fields for Group dialog as shown in Figure 5.33.
Click Edit Group.

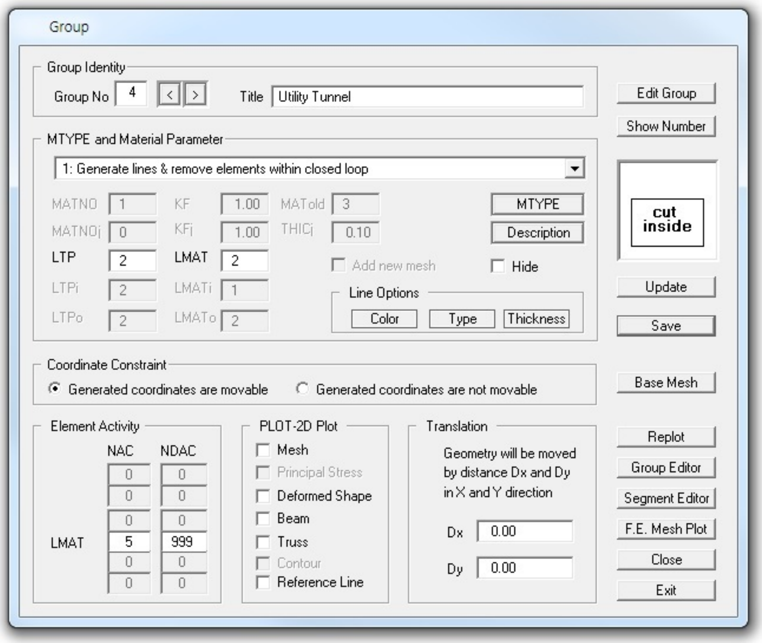


Figure 5.33 Group dialog for Utility Tunnel

Select Replace All Segments in Edit Segment dialog in Figure 5.34
Click Edit.

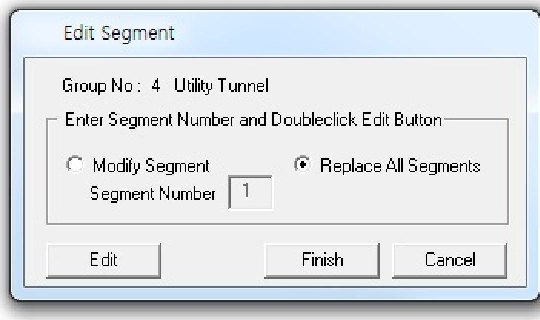


Figure 5.34 Edit segment dialog for Group No 4

Warning message is displayed as shown in Figure 5.35.
Click OK.

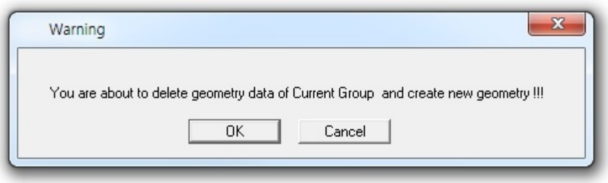


Figure 5.35 Warning message

Fill in input fields for Arc Segment dialog as shown in Figure 5.36.
Click Draw and then Finish in Arc Segment dialog.
Click Finish in Edit Segment dialog in Figure 5.34.

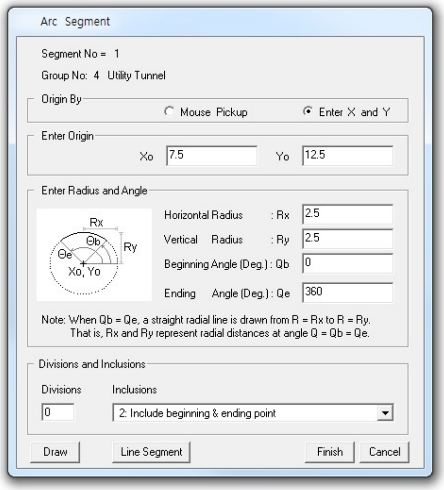


Figure 5.36 Arc segment dialog for Utility Tunnel

Click Update and then Replot buttons in Group dialog.
Figure 5.37 shows a new plot with Utility Tunnel on drawing board.

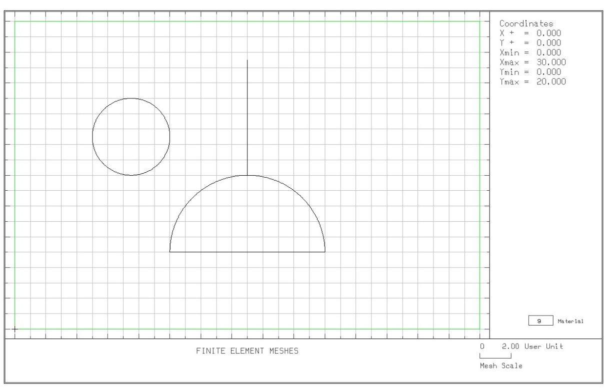


Figure 5.37 Arch and Utility Tunnels on drawing board

Step 28: Finite Element Mesh

Click Save and F.E. Mesh Plot button in Group dialog.
Follow the same procedure as in Steps 10 and 11.
Finite element meshes are shown in Figure 5.38

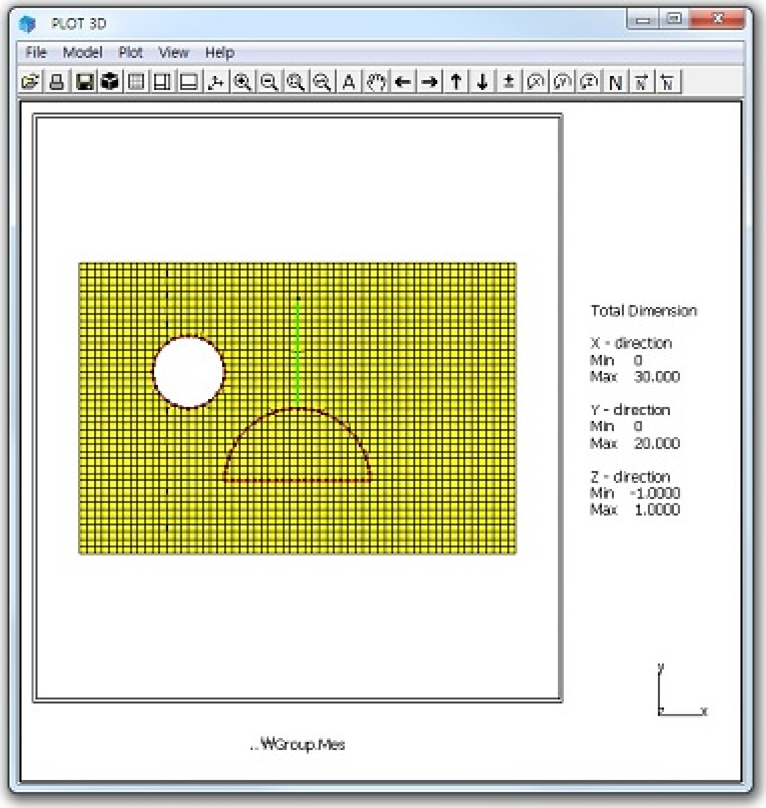


Figure 5.38 Finite element mesh plot

Step 29: Exit

Click Exit button in Group dialog.
Click OK in Exit dialog as in Figure 5.22.

5.2 NATM Tunnel

This example illustrates how to build group meshes for typical NATM (New Austrian Tunneling Method) tunnel.

5.2.1 Overview

The cross section of NATM tunnel consists of rock bolts, shotcrete, reinforced concrete liner, and core as schematically shown in Figure 5.39.

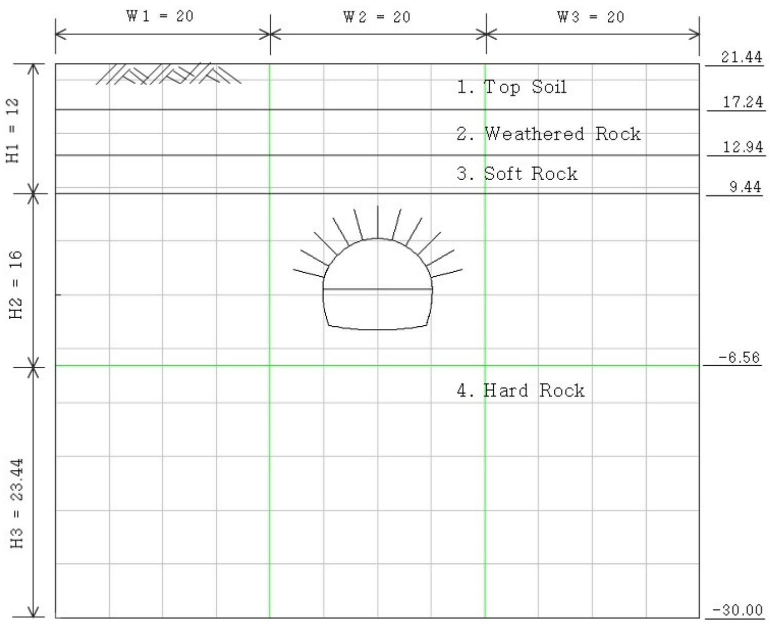


Figure 5.39 Tunnel cross section

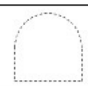
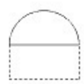
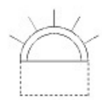
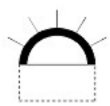
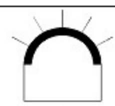
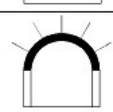
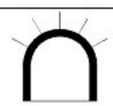
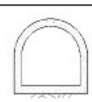
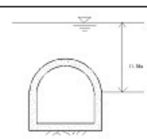
Step	Construction State	Description	
1, 2		In Situ Ko State	
3		50 % Stress Relief	Upper Core Excavation
4		75 % Stress Relief Soft Shotcrete, Rock Bolt	
5		100 % Stress Relief Hard Shotcrete, Rock Bolt	
6		50 % Stress Relief	Lower Core Excavation
7		75 % Stress Relief, Soft Shotcrete	
8		100 % Stress Relief, Hard Shotcrete	
9		Lining Subjected to : Weight	
12		Lining Subjected to : Weight + Water Pressure	

Table 5.4 Construction sequence

A total of 21 groups are used to model NATM tunnel as schematically shown in Figure 5.40: 4 for in situ geological profile, 11 for rock bolts, 1 for lining, 3 for shotcrete, and 2 for core.

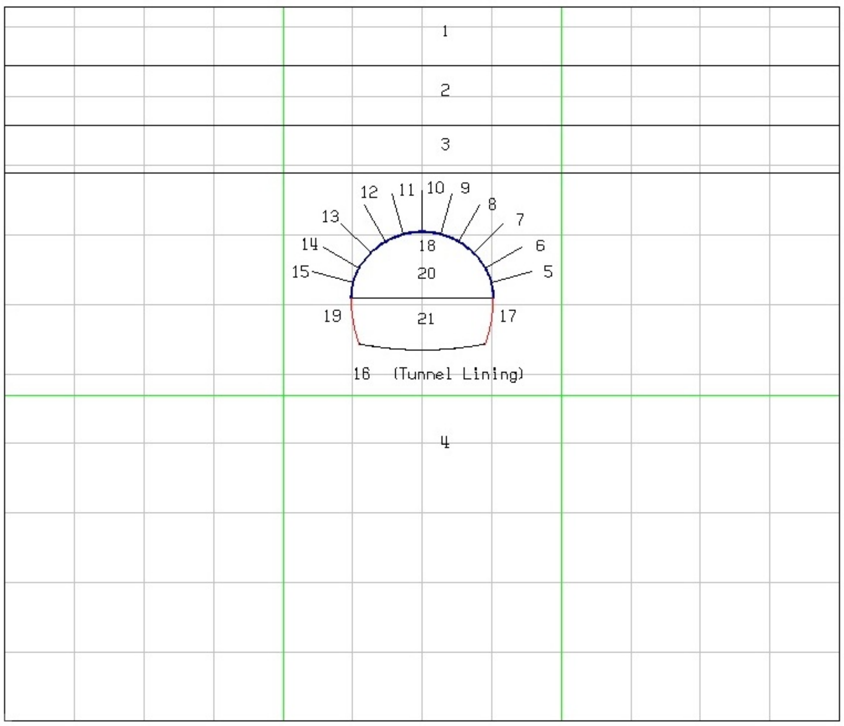


Figure 5.40 Group section view

Table 5.5 summarizes key parameters of groups.

Group	Name	MTYPE	NAC	NDAC	MATNO / LTP / LMAT / IEND
1	Top Soil	3			1 / 0 / 0 / 2
2	Weathered Rock	3			2 / 0 / 0 / 2
3	Soft Rock	3			3 / 0 / 0 / 2
4	Hard Rock	3			4 / 0 / 0 / 2
5	Rock Bolt-1	2	4	999	0 / 3 / 1 / -2
6	Rock Bolt-2	2	4	999	0 / 3 / 1 / -2
7	Rock Bolt-3	2	4	999	0 / 3 / 1 / -2
8	Rock Bolt-4	2	4	999	0 / 3 / 1 / -2
9	Rock Bolt-5	2	4	999	0 / 3 / 1 / -2
10	Rock Bolt-6	2	4	999	0 / 3 / 1 / -2
11	Rock Bolt-7	2	4	999	0 / 3 / 1 / -2
12	Rock Bolt-8	2	4	999	0 / 3 / 1 / -2
13	Rock Bolt-9	2	4	999	0 / 3 / 1 / -2
14	Rock Bolt-10	2	4	999	0 / 3 / 1 / -2
15	Rock Bolt-11	2	4	999	0 / 3 / 1 / -2
16	Tunneling Lining	-2	9	999	MATNOj = 7, LTPi = 0, LTPo = 2 LMATo = 2, IEND = 2
17	Shotcrete Right Lower	2	7	999	0 / 2 / 1 / 3
18	Shotcrete Upper	2	4	999	0 / 2 / 1 / 3
19	Shotcrete Left Lower	2	7	999	0 / 2 / 1 / 3
20	Upper Core	3	0	5	5 / 0 / 0 / 3
21	Lower Core	3	0	8	6 / 0 / 0 / 3

Table 5.5 Group key parameters

5.2.2 Base Mesh

Built-in Base Mesh dialog is shown in Figure 5.41 with input data for blocks and boundary condition. Element size is more refined at the center block considering relatively high stress change here due to tunnel construction.

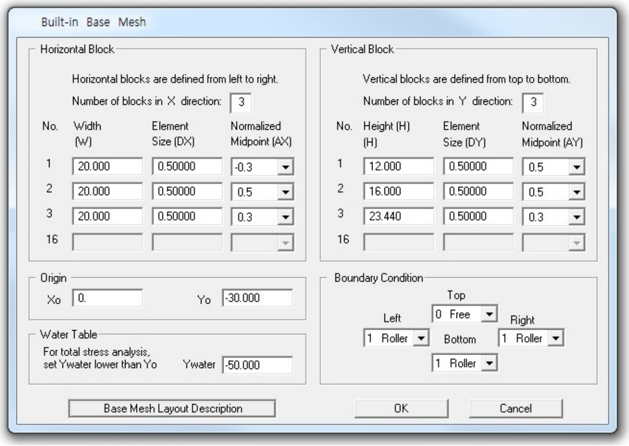


Figure 5.41 Built-in base mesh dialog

Figure 5.42 shows base mesh plot on drawing board.

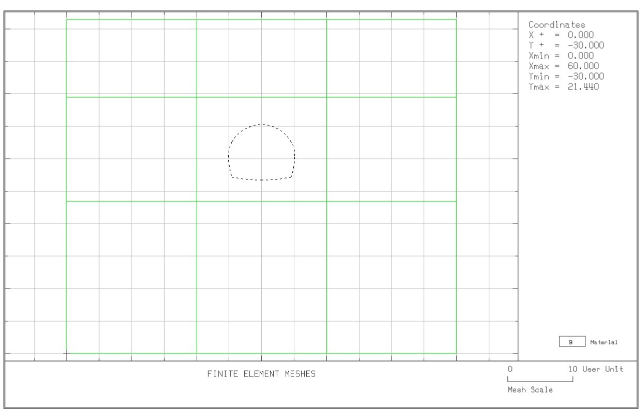


Figure 5.42 Base mesh plot on drawing board

5.2.3 Groups

Group meshes are divided into five parts:

- Geological profile
- Rock bolt
- Lining
- Shotcrete
- Core

Final finite element meshes are most influenced by group order and IEND.

5.2.3.1 Geological Profile

In situ geological profile consists of four layers: top soil, weathered rock, soft rock, and hard rock. Table 5.6 lists key parameters of these groups.

Group	Profile	MTYPE	Elem.	MATNO	Seg.	Beginning Point		Ending Point		IEND
						X	Y	X	Y	
1	Top Soil	3	Cont	1	1	0	17.24	60	17.24	2
					2	60	17.24	60	21.44	2
					3	60	21.44	0	21.44	2
					4	0	21.44	0	17.24	2
2	Weathered Rock	3	Cont	2	1	0	12.94	60	12.94	2
					2	60	12.94	60	17.24	2
					3	60	17.24	0	17.24	2
					4	0	17.24	0	12.94	2
3	Soft Rock	3	Cont	3	1	0	9.44	60	9.44	2
					2	60	9.44	60	12.94	2
					3	60	12.94	0	12.94	2
					4	0	12.94	0	9.44	2
4	Hard Rock	3	Cont	4	1	0	-30	60	-30	2
					2	60	-30	60	9.44	2
					3	60	9.44	0	9.44	2
					4	0	9.44	0	-30	2

Table 5.6 Key parameters for geological profile

Figure 5.43 shows Group dialog for top soil layer.
Group dialogs for the other layers are very similar to this group 1.
It is a good idea to click Save button occasionally in case of system down.

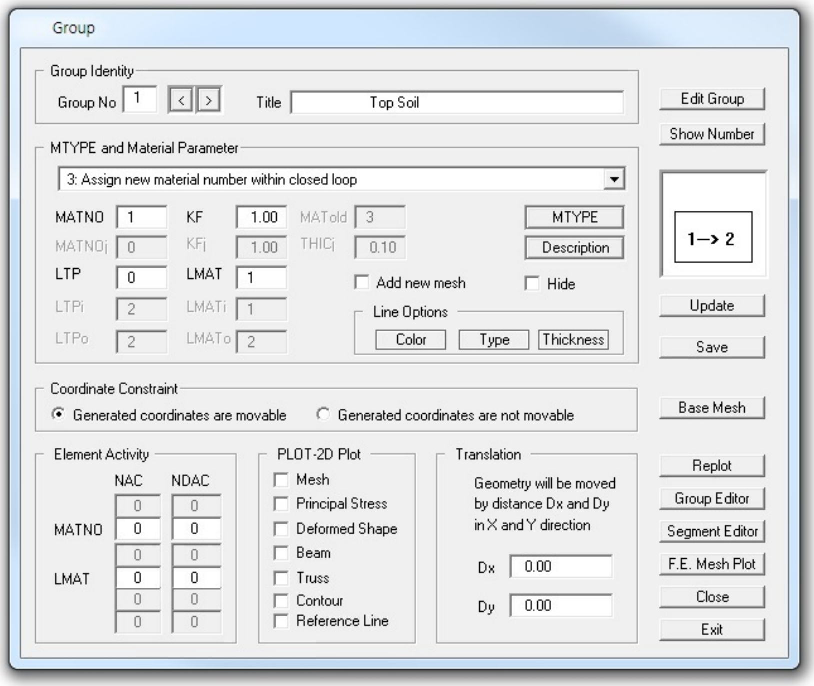


Figure 5.43 Group dialog for top soil layer

5.2.3.2 Rock Bolt

There are eleven rock bolts above the tunnel crown as schematically shown in Figure 5.44. Table 5.7 lists key parameters of these groups.

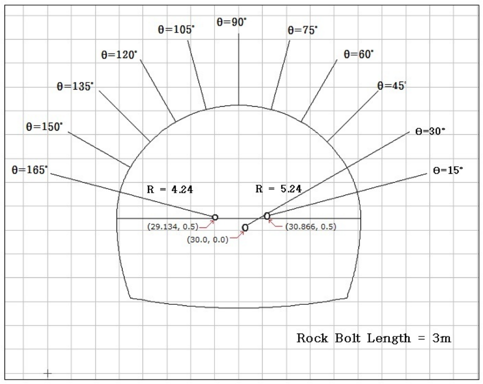


Figure 5.44 Rock bolt layout

Group	Name	NAC/NDAC	Origin		Radius & Angle				MTYPE/LTP/LMAT/IEND
			X_o	Y_o	R_X	R_Y	Θ_b	Θ_e	
5	Bolt-1	4 / 999	30.866	0.5	4.24	7.24	15	15	2 / 3 / 1 / -2
6	Bolt-2	4 / 999	30	0	5.24	8.24	30	30	2 / 3 / 1 / -2
7	Bolt-3	4 / 999	30	0	5.24	8.24	45	45	2 / 3 / 1 / -2
8	Bolt-4	4 / 999	30	0	5.24	8.24	60	60	2 / 3 / 1 / -2
9	Bolt-5	4 / 999	30	0	5.24	8.24	75	75	2 / 3 / 1 / -2
10	Bolt-6	4 / 999	30	0	5.24	8.24	90	90	2 / 3 / 1 / -2
11	Bolt-7	4 / 999	30	0	5.24	8.24	105	105	2 / 3 / 1 / -2
12	Bolt-8	4 / 999	30	0	5.24	8.24	120	120	2 / 3 / 1 / -2
13	Bolt-9	4 / 999	30	0	5.24	8.24	135	135	2 / 3 / 1 / -2
14	Bolt-10	4 / 999	30	0	5.24	8.24	150	150	2 / 3 / 1 / -2
15	Bolt-11	4 / 999	29.134	0.5	4.24	7.24	165	165	2 / 3 / 1 / -2

Table 5.7 Key parameters for rock bolt

Figure 5.45 shows Group dialog for the first rock bolt at 15 degrees. Group dialogs for other rock bolts are very similar to this group 5.

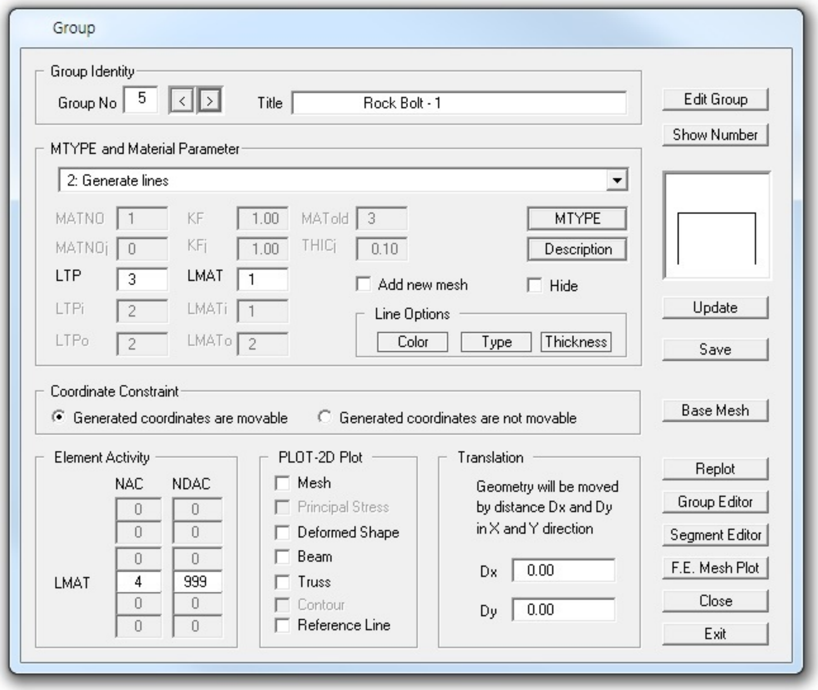


Figure 5.45 Group dialog for rock bolt at 15 degrees

5.2.3.3 Lining

Lining is the reinforced concrete liner which is modeled by beam elements. Seven segments are used to model lining as shown in Figure 5.46.
The interface between lining and shotcrete is modeled by joint element as shown in Figure 5.47. It should be noted that MTYPE = -2 in this group includes both lining and joint elements.

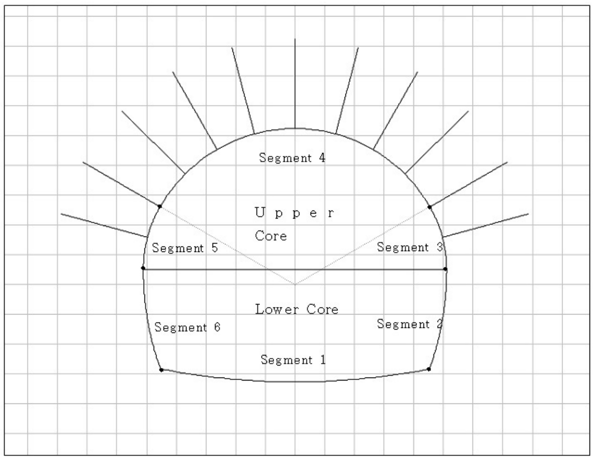


Figure 5.46 Lining segments

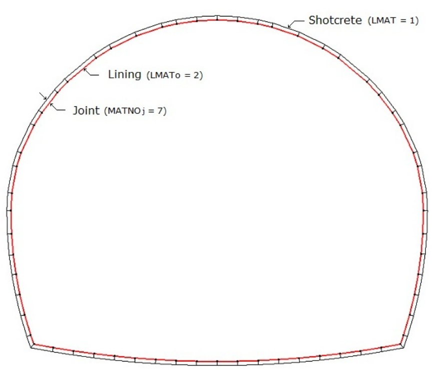


Figure 5.47 Interface joint element

Table 5.8 lists key parameters of this group.

	Element Type	Material No	Element Activity	
			NAC	NDAC
Interface	Joint	MATNOj = 7	9	999
Lining	Beam (LTPo = 2)	LMATo = 2	9	999

Group	Name	MTYPE	Seg.	Origin		Radius & Angle				IEND
				X_o	Y_o	R_X	R_Y	Θ_b	Θ_e	
16	Tunnel Lining	-2	1	30	20.59	23.86	23.86	270	280.94	2
			2	25.25	0.5	9.86	9.86	-19.78	0	2
			3	30.866	0.5	4.24	4.24	0	30	2
			4	30	0	5.24	5.24	30	150	2
			5	29.134	0.5	4.24	4.24	150	180	2
			6	34.75	0.5	9.86	9.86	-180	-160.22	2
			7	30	20.59	23.86	23.86	259.06	270	2

Table 5.8 Key parameters for lining and joint elements

Figure 5.48 shows Group dialog for tunnel lining.

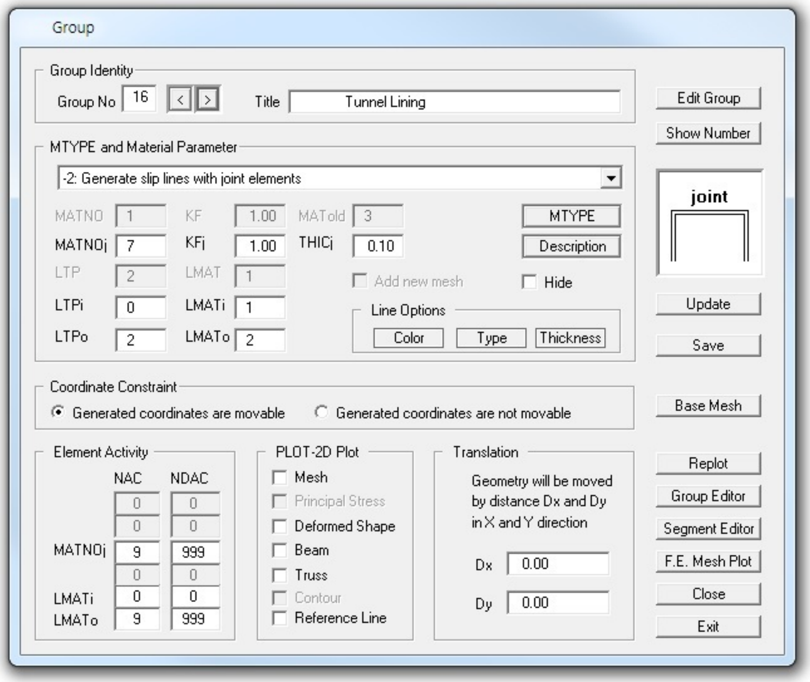
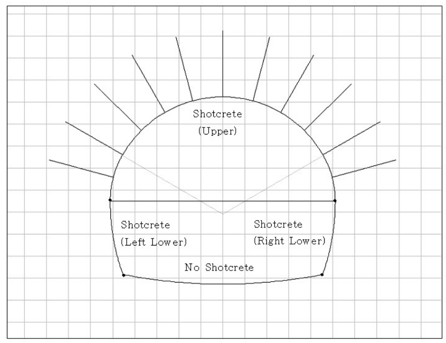


Figure 5.48 Group dialog for tunnel lining

5.2.3.4 Shotcrete

Shotcrete is applied to upper tunnel wall right after excavation of upper core and lower tunnel walls right after excavation of lower core as shown in Figure 5.49. But shotcrete is not applied at tunnel invert.
Table 5.9 lists key parameters of these groups.

Figure 5.49
Shotcrete cross section



Group	Name	MTYPE	LTP	LMAT	Element Activity	
					NAC	NDAC
17	Shotcrete: Right Lower	2	2	1	7	999
18	Shotcrete: Upper	2	2	1	4	999
19	Shotcrete: Left Lower	2	2	1	7	999

Group	Name	MTYPE	Seg	Origin		Radius & Angle				IEND
				X_o	Y_o	R_X	R_Y	Θ_b	Θ_e	
17	Shotcrete Right Lower	2	1	25.25	0.5	9.86	9.86	-19.78	0	3
18	Shotcrete Upper	2	1	30.866	0.5	4.24	4.24	0	30	3
			2	30	0	5.24	5.24	30	150	3
			3	29.134	0.5	4.24	4.24	150	180	3
19	Shotcrete Left Lower	2	1	34.75	0.5	9.86	9.86	-180	-160.22	3

Table 5.9 Key parameters for shotcrete elements

Figure 5.50 shows Group dialog for the upper shotcrete.
Group dialogs for other lower shotcrete are very similar to this group 18.

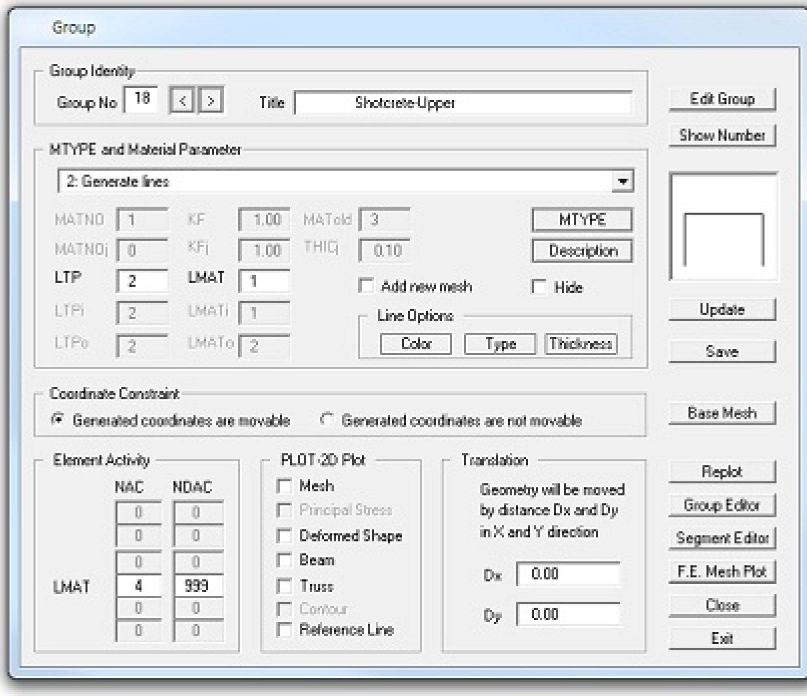


Figure 5.50 Group dialog for upper shotcrete

5.2.3.5 Core

Core is divided into upper and lower parts as in Figure 5.46 considering the order of excavation. Table 5.10 lists key parameters of these groups.

Group	Name	MTYPE	Element	MATNO	Element Activity	
					NAC	NDAC
20	Upper Core	3	Cont.	5	0	5
21	Lower Core	3	Cont.	6	0	8

Group	Seg	Line Segment				Arc Segment						IEND
		Beginning Pt.		Ending Pt.		Origin		Radius & Angle				
		X	Y	X	Y	X_o	Y_o	R_X	R_Y	Θ_b	Θ_e	
20	1	24.894	0.5	35.106	0.5							3
	2					30.866	0.5	4.24	4.24	0	30	3
	3					30	0	5.24	5.24	30	150	3
	4					29.134	0.5	4.24	4.24	150	180	3
21	1					30	20.59	23.86	23.86	259.06	280.94	3
	2					25.25	0.5	9.86	9.86	-19.78	0	3
	3	35.106	0.5	24.894	0.5							3
	4					34.75	0.5	9.86	9.86	-180	-160.22	3

Table 5.10 Key parameters for core elements

Figure 5.51 shows Group dialog for the upper core.
Group dialog for the other lower core is very similar to this group 20.

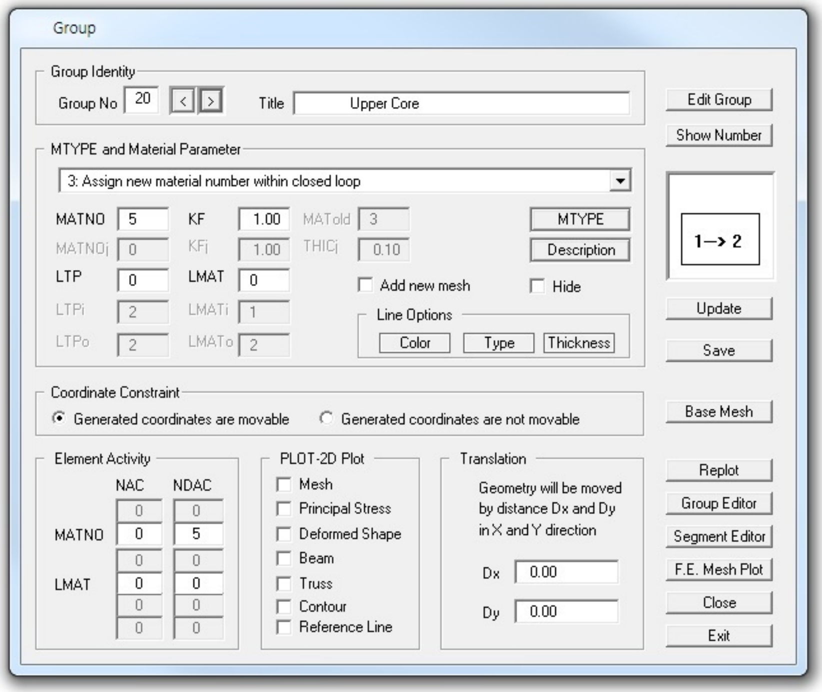


Figure 5.51 Group dialog for upper core

5.2.4 Finite Element Mesh Plot

Figure 5.52 shows finite element meshes generated from group meshes. Finite element meshes around tunnel are shown in Figure 5.53.

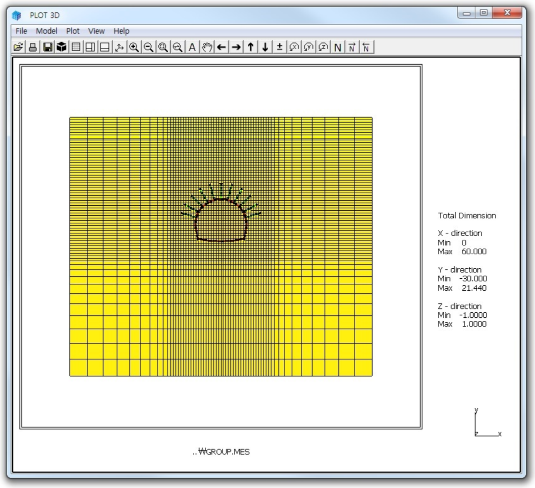


Figure 5.52 Finite element meshes for NATM tunnel

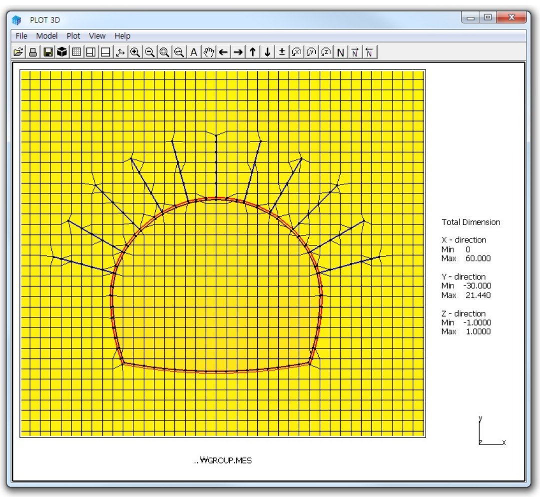


Figure 5.53 Finite element meshes around tunnel

5.3 Excavation

This example illustrates how to build group meshes for typical multi-step excavations performed near the existing box structure.

5.3.1 Overview

The cross section of this excavation problem consists of box structure, SCE-wall, anchors, and excavation zones as shown in Figure 5.54.

Cross section near the box structure is shown in detail in Figure 5.55.

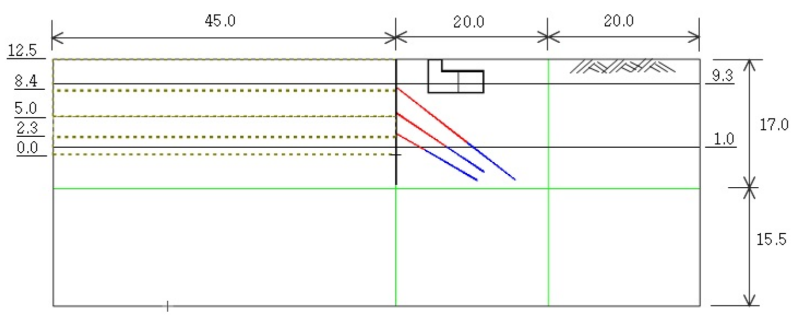


Figure 5.54 Schematic section of excavation problem

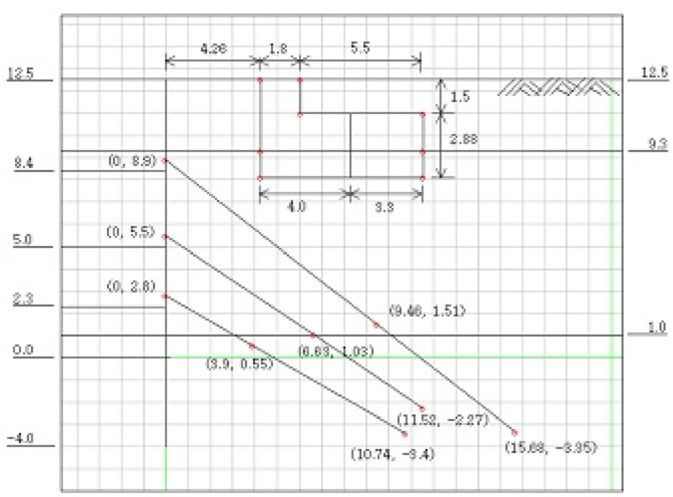


Figure 5.55 Cross section near box structure

Table 5.11 shows the construction sequence associated with multi-step excavations.

Step	Description
1,2	In situ stress
3	Box Excavation and Frame Construction
4	First Excavation (Y = 8.4 m)
5	First Anchor Installation
6	Second Excavation (Y = 5.0 m)
7	Second Anchor Installation
8	Third Excavation (Y = 2.3 m)
9	Third Anchor Installation
10	Fourth Excavation (Y = 0.0 m)

Table 5.11 Construction sequence

A total of 17 groups are used to model this excavation problem as schematically shown in Figure 5.56: 3 for in situ geological profile, 3 for box structure, 1 for SCE-wall, 4 for excavations, and 6 for anchors.

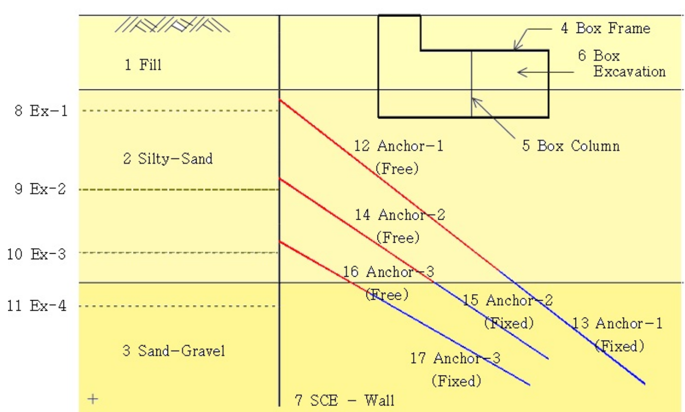


Figure 5.56 Group section view

Table 5.12 summarizes key parameters of groups.

Group	Name	MTYPE	NAC	NDAC	MATNO / LTP / LMAT / IEND
1	Fill	3	0	0	1 / 0 / 0 / 2
2	Silty-Sand	3	0	0	2 / 0 / 0 / 2
3	Sand-Gravel	3	0	0	3 / 0 / 0 / 2
4	Box Frame	2	3	999	0 / 2 / 2 / 2
5	Box Column	2	3	999	0 / 2 / 3 / 2
6	Box Excavation	3	0	3	0 / 0 / 0 / 3
7	SCE-Wall	2	4	999	0 / 2 / 1 / 2
8	Excavation-1	3	0	4	0 / 0 / 0 / 2
9	Excavation-2	3	0	6	0 / 0 / 0 / 2
10	Excavation-3	3	0	8	0 / 0 / 0 / 2
11	Excavation-4	3	0	10	0 / 0 / 0 / 2
12	Anchor-1 Free	2	5	999	0 / 3 / 1 / 0
13	Anchor-1 Fixed	2	5	999	0 / 3 / 2 / -2
14	Anchor-2 Free	2	7	999	0 / 3 / 3 / 0
15	Anchor-2 Fixed	2	7	999	0 / 3 / 4 / -2
16	Anchor-3 Free	2	9	999	0 / 3 / 5 / 0
17	Anchor-3 Fixed	2	9	999	0 / 3 / 6 / -2

Table 5.12 Group key parameters

5.3.2 Base Mesh

Built-in Base Mesh dialog is shown in Figure 5.57 with input data for blocks and boundary condition. Element size is more refined at the top center block considering relatively high stress change here due to excavation.

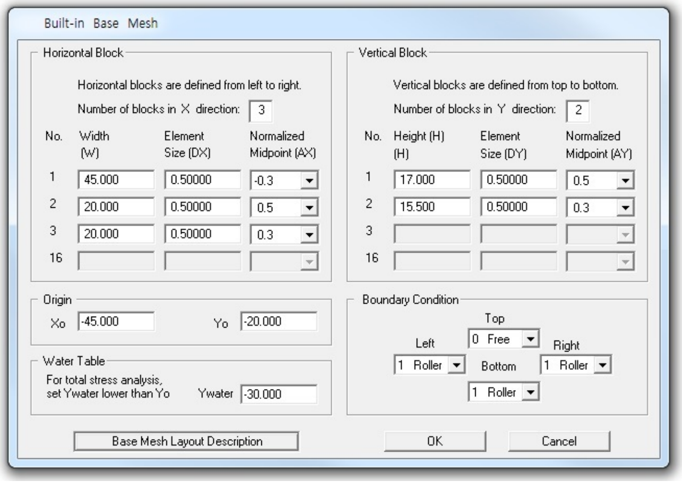


Figure 5.57 Built-in base mesh dialog

Figure 5.58 shows base mesh plot on drawing board.

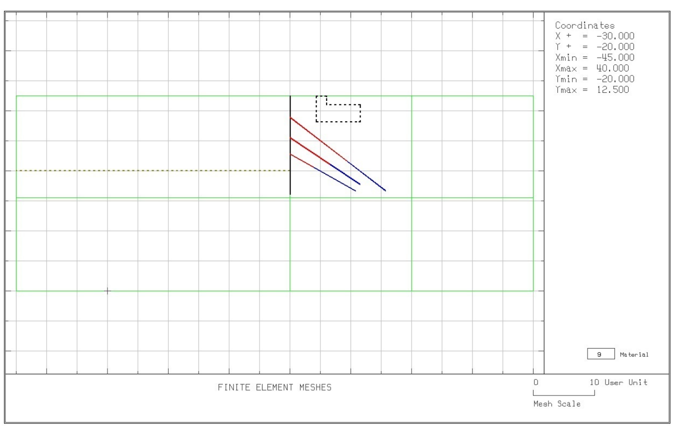


Figure 5.58 Base mesh plot on drawing board

5.3.3 Groups

Group meshes are divided into five parts:

- Geological profile
- Box structure
- SCE-Wall
- Excavation
- Anchor

It should be noted that the final finite element meshes are most influenced by group order and IEND.

5.3.3.1 Geological Profile

In situ geological profile consists of three layers: fill, silty-sand, and sand-gravel. Table 5.13 lists key parameters of these groups

Group	Profile	MTYPE	Elem.	MATNO	Seg.	Beginning Point		Ending Point		IEND
						X	Y	X	Y	
1	Fill	3	Cont	1	1	-45	9.3	40	9.3	2
					2	40	9.3	40	12.5	2
					3	40	12.5	-45	12.5	2
					4	-40	12.5	-45	9.3	2
2	Silty-Sand	3	Cont	2	1	-45	1	40	1	2
					2	40	1	40	9.3	2
					3	40	9.3	-45	9.3	2
					4	-45	9.3	-45	1	2
3	Sand-Gravel	3	Cont	3	1	-45	-20	40	-20	2
					2	40	-20	40	1	2
					3	40	1	-45	1	2
					4	-45	1	-45	-20	2

Table 5.13 Key parameters for geological profile

Figure 5.59 shows Group dialog for top fill.
Group dialogs for the other layers are very similar to this group 1.

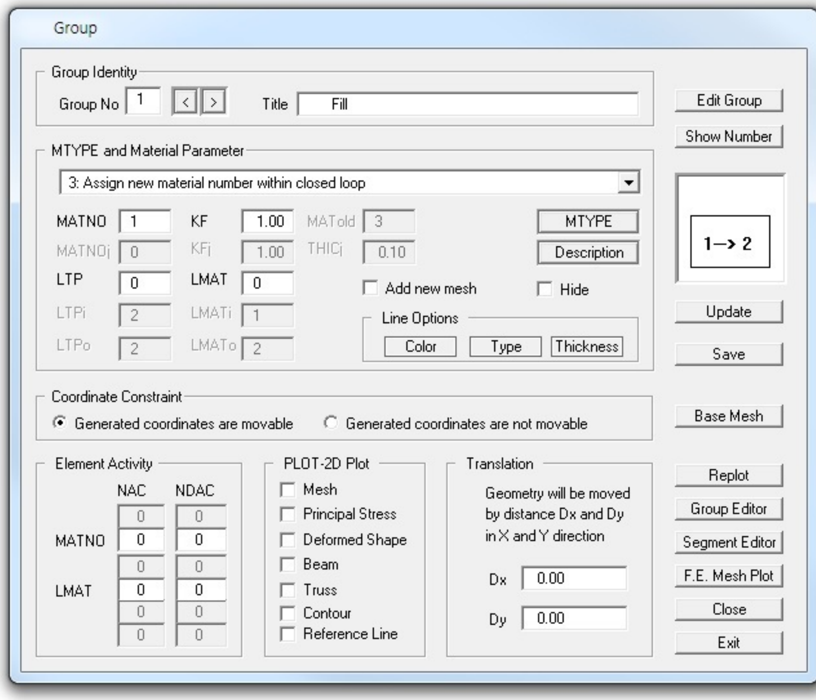


Figure 5.59 Group dialog for top fill

5.3.3.2 Box Structure

Box structure consists of frame, column, and excavation as schematically shown in Figure 5.56. Table 5.14 lists key parameters of these groups.

Group	Name	MTYPE	LTP	LMAT	Element Activity		Seg	Beginning Point		Ending Point		IEND
					NAC	NDAC		X	Y	X	Y	
4	Frame	2	2	2	3	999	1	4.26	8.12	11.56	8.12	2
							2	11.56	8.12	11.56	11	2
							3	11.56	11	6.06	11	2
							4	6.06	11	6.06	12.5	2
							5	6.06	12.5	4.26	12.5	2
							6	4.26	12.5	4.26	8.12	2
5	Column	2	2	3	3	999	1	8.26	11	8.26	8.12	2

Group	Name	MTYPE	Elem	MATNO	Element Activity		Seg	Beginning Point		Ending Point		IEND
					NAC	NDAC		X	Y	X	Y	
6	Excavation	3	Cont	0	0	3	1	4.26	8.12	11.56	8.12	2
							2	11.56	8.12	11.56	11	2
							3	11.56	11	6.06	11	2
							4	6.06	11	6.06	12.5	2
							5	6.06	12.5	4.26	12.5	2
							6	4.26	12.5	4.26	8.12	2

Table 5.14 Key parameters for box structure

Figure 5.60 shows Group dialog for the box frame.
Group dialog for box column is very similar to this group 4.

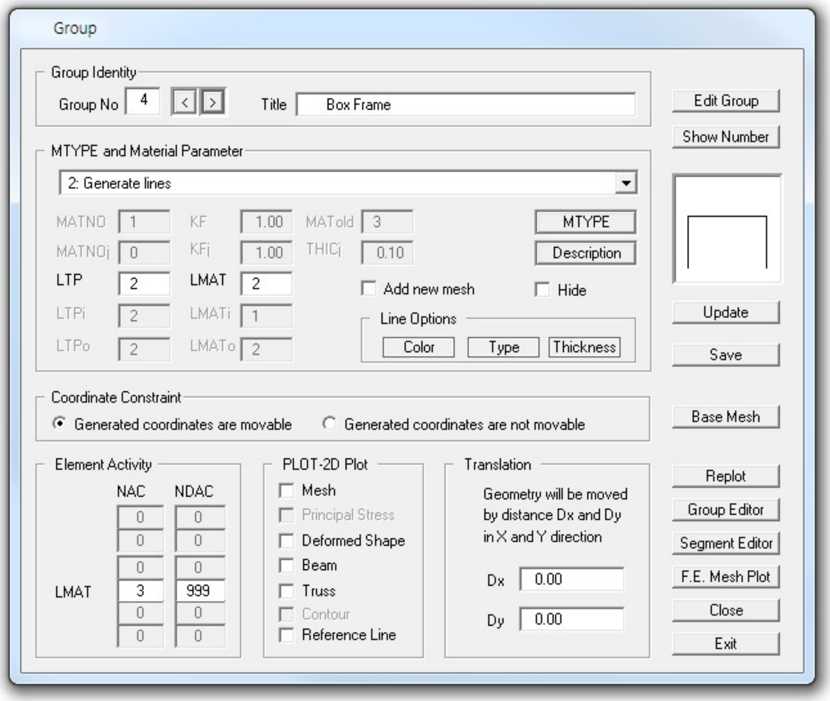


Figure 5.60 Group dialog for box frame

Figure 5.61 shows Group dialog for the box excavation.

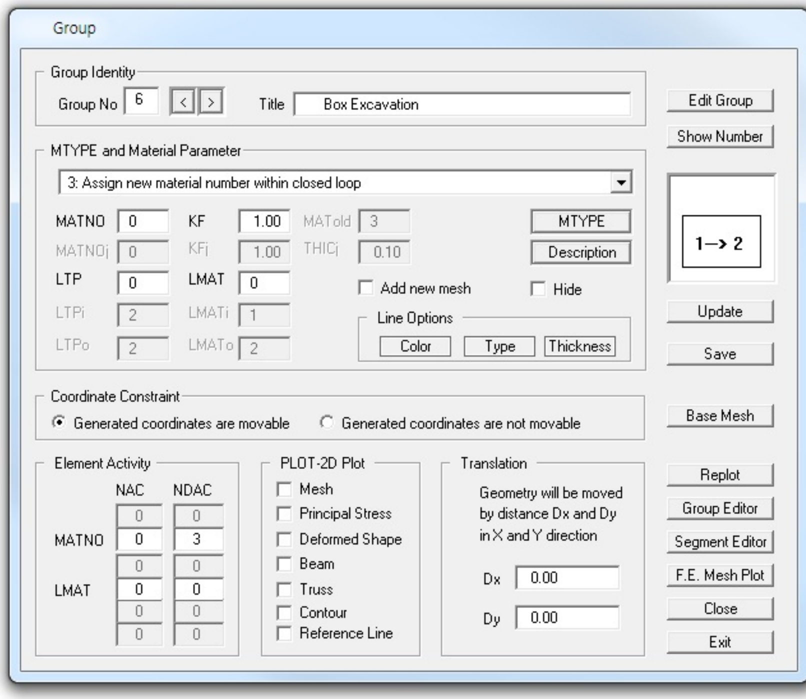


Figure 5.61 Group dialog for box excavation

5.3.3.3 SCE-Wall

SCE-Wall is the structure to prevent ground movement due to excavations and is supported by anchors as schematically shown in Figure 5.56. Table 5.15 lists key parameters of this group.

Group	Name	MTYPE	LTP	LMAT	Element Activity		Seg	Beginning Point		Ending Point		IEND
					NAC	NDAC		X	Y	X	Y	
7	SCE-Wall	2	2	1	4	999	1	0	12.5	0	-4	2

Table 5.15 Key parameters for SCE-wall

Figure 5.62 shows Group dialog for SCE-wall.

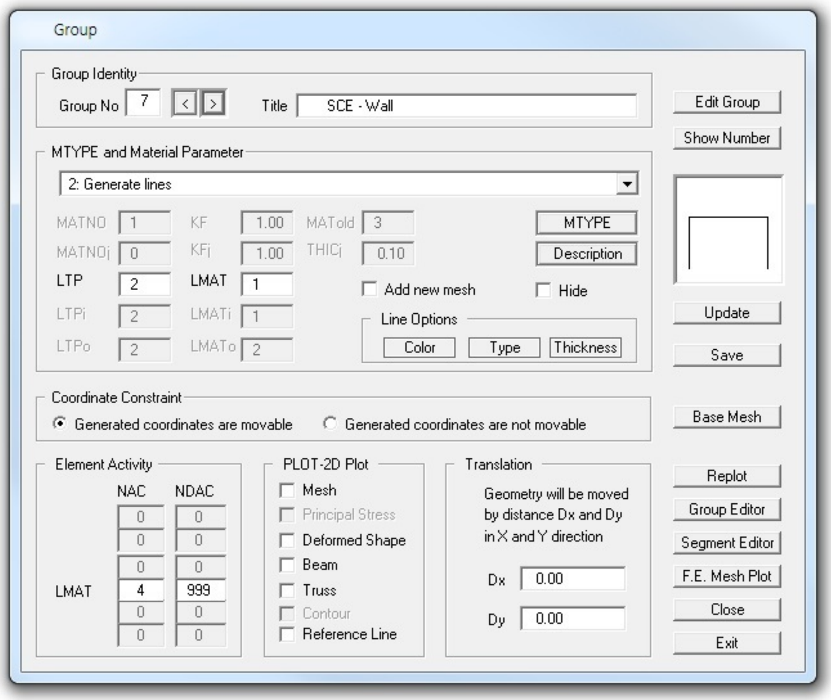


Figure 5.62 Group dialog for SCE-wall

5.3.3.4 Excavation

Excavations are conducted through four stages as schematically shown in Figure 5.56. Table 5.16 lists key parameters of these groups.

Group	Name	MTYPE	Elem	MATNO / NAC / NDAC	Seg.	Beginning Point		Ending Point		IEND
						X	Y	X	Y	
8	Excavation-1	3	Cont	0 / 0 / 4	1	-45	8.4	0.0	8.4	2
					2	0	8.4	0	12.5	2
					3	0	12.5	-45	12.5	2
					4	-45	12.5	-45	8.4	2
9	Excavation-2	3	Cont	0 / 0 / 6	1	-45	5	0	5	2
					2	0	5	0	8.4	2
					3	0	8.4	-45	8.4	2
					4	-45	8.4	-45	5	2
10	Excavation-3	3	Cont	0 / 0 / 8	1	-45	2.3	0	2.3	2
					2	0	2.3	0	5	2
					3	0	5	-45	5	2
					4	-45	5	-45	2.3	2
11	Excavation-4	3	Cont	0 / 0 / 10	1	-45	0	0	0	2
					2	0	0	0	2.3	2
					3	0	2.3	-45	2.3	2
					4	-45	2.3	-45	0	2

Table 5.16 Key parameters for excavation

Figure 5.63 shows Group dialog for the first excavation.
Group dialogs for the other excavations are very similar to this group 8.

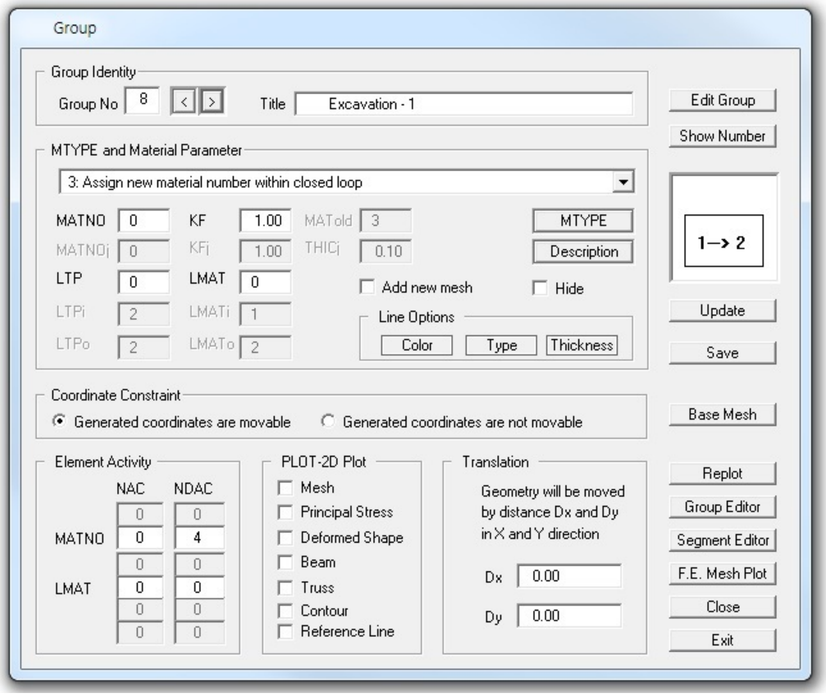


Figure 5.63 Group dialog for the first excavation

5.3.3.5 Anchor

Three anchors are used to support SCE-wall as schematically shown in Figure 5.56. Each anchor consists of two parts: free and fixed length. Table 5.17 lists key parameters of these groups.

Group	Name	MTYPE / LTP / LMAT / NAC / NDAC	Seg.	Beginning Point		Ending Point		NDIV	IEND
				X	Y	X	Y		
12	Anchor-1 Free	2 / 3 / 1 / 5 / 999	1	0	8.9	9.46	1.51	1	0
13	Anchor-1 Fixed	2 / 3 / 2 / 5 / 999	1	9.46	1.51	15.68	-3.35	0	-2
14	Anchor-2 Free	2 / 3 / 3 / 7 / 999	1	0	5.5	6.63	1.03	1	0
15	Anchor-2 Fixed	2 / 3 / 4 / 7 / 999	1	6.63	1.03	11.52	-2.27	0	-2
16	Anchor-3 Free	2 / 3 / 5 / 9 / 999	1	0	2.8	3.9	0.55	1	0
17	Anchor-3 Fixed	2 / 3 / 6 / 9 / 999	1	3.9	0.55	10.74	-3.4	0	-2

Table 5.17 Key parameters for anchor

Figure 5.64 shows Group dialog for the first anchor (free part). Group dialogs for other anchors are very similar to this group 12.

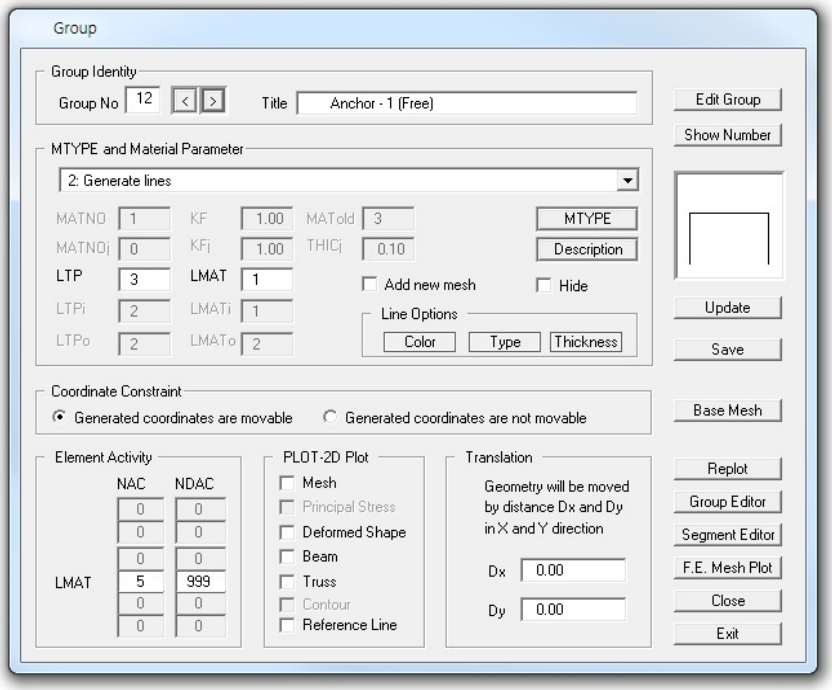


Figure 5.64 Group dialog for the first anchor (free part)

5.3.4 Finite Element Mesh Plot

Figure 5.65 shows finite element meshes generated from group meshes. Finite element meshes near box structure are shown in Figure 5.66.

Figure 5.65
Finite element meshes

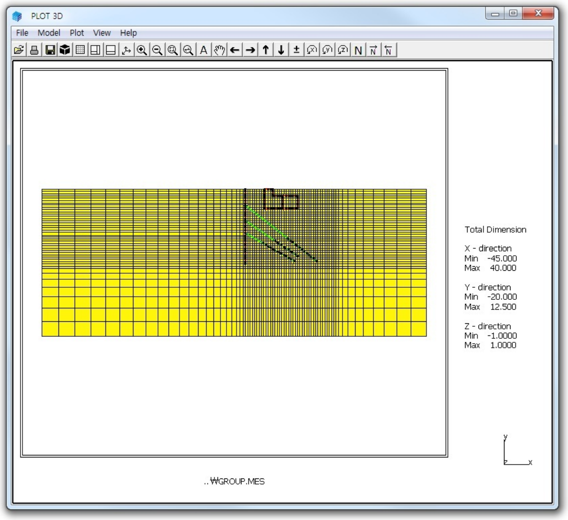
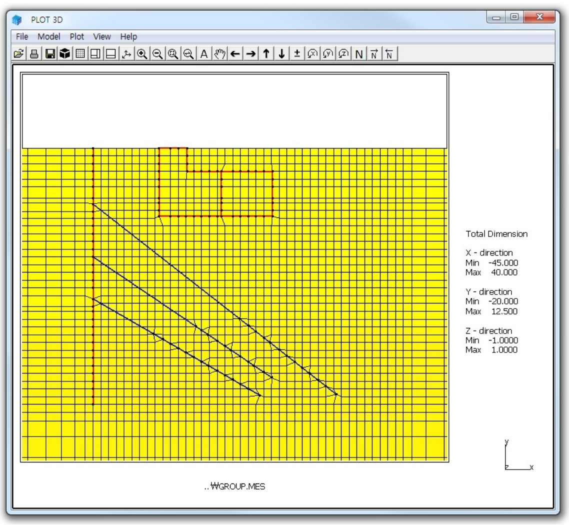


Figure 5.66
Finite element meshes
near box structure



5.4 Buried Pipe

This example illustrates how to build group meshes for typical pipe buried in the trench followed by multi-step embankment lifts.

5.4.1 Overview

The cross section of this buried pipe consists of natural soil, bedding, steel pipe, backfill, and lifts as shown in Figure 5.67.

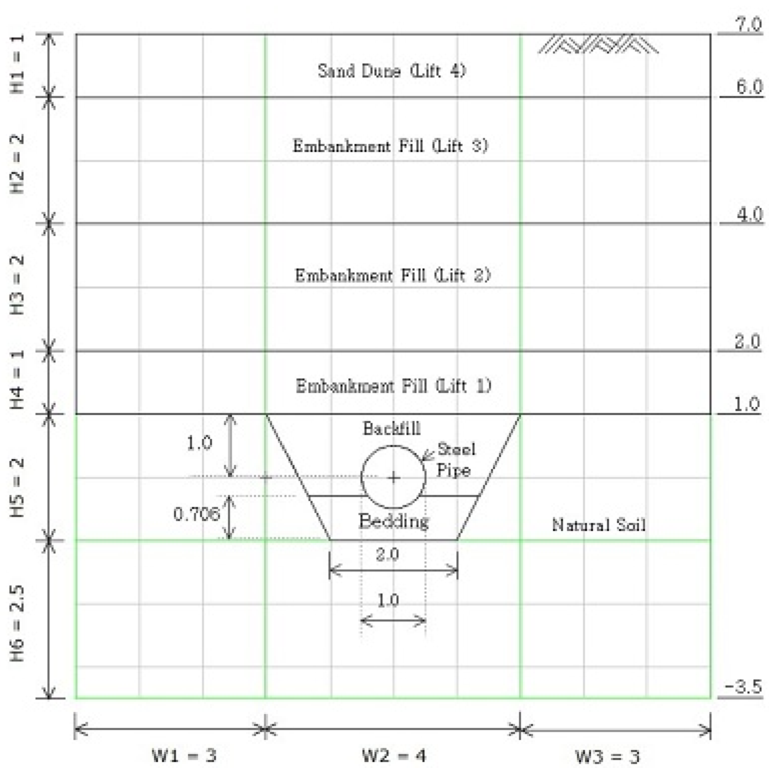


Figure 5.67 Schematic section of buried pipe

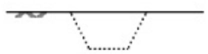
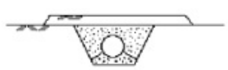
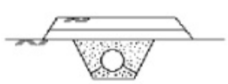
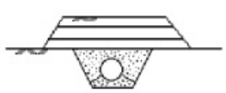
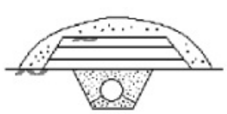
Step	Construction Sequence	Description	Element Activity
1,2		In situ K_o state	Active elements: Natural soil within trench
3		Excavate trench	Deactive elements: Natural soil within trench
4		Place bedding	Active elements: Compacted sand for bedding
5		Place steel pipe Fill the backfill	Active elements: Steel pipe Compacted sand for backfill
6		Place first lift of embankment fill	Active elements: First lift of embankment fill
7		Place second lift of embankment fill	Active elements: Second lift of embankment fill
8		Place third lift of embankment fill	Active elements: Third lift of embankment fill
9		Place fourth lift of sand done	Active elements: Fourth lift of sand done

Table 5.18 Construction sequence

A total of 9 groups are used to model this buried pipe as schematically shown in Figure 5.68: 1 for natural soil, 1 for excavation, 2 for compacted sands, 1 for steel pipe, and 4 for lifts.

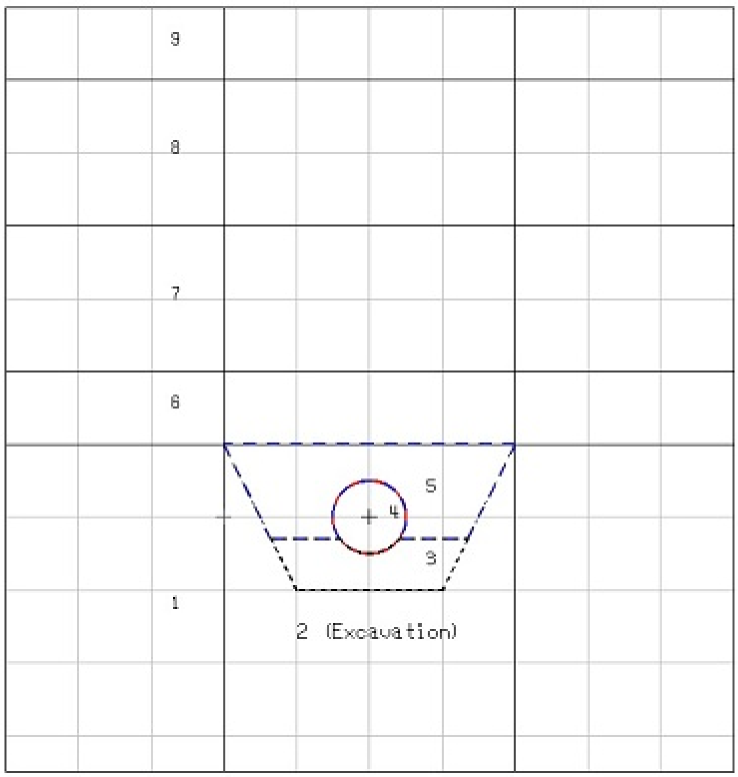


Figure 5.68 Group section view

Table 5.19 summarizes key parameters of groups.

Group	Name	MTYPE	NAC	NDAC	MATNO / LTP / LMAT / IEND
1	Natural Soil	3	0	0	1 / 0 / 0 / 2
2	Excavation	3	0	3	1 / 0 / 0 / 2
3	Bedding	3	4	999	2 / 0 / 0 / 2
4	Steel Pipe	2	5	999	0 / 2 / 1 / 2
5	Backfill	3	5	999	3 / 0 / 0 / 2
6	Lift-1	3	6	999	4 / 0 / 0 / 2
7	Lift-2	3	7	999	5 / 0 / 0 / 2
8	Lift-3	3	8	999	6 / 0 / 0 / 2
9	Lift-4	3	9	999	7 / 0 / 0 / 2

Table 5.19 Group key parameters

5.4.2 Base Mesh

Built-in Base Mesh dialog is shown in Figure 5.69 with input data for blocks and boundary condition. Element size is more refined at the block in trench considering relatively high stress change here due to pipe construction. Figure 5.70 shows base mesh plot on drawing board.

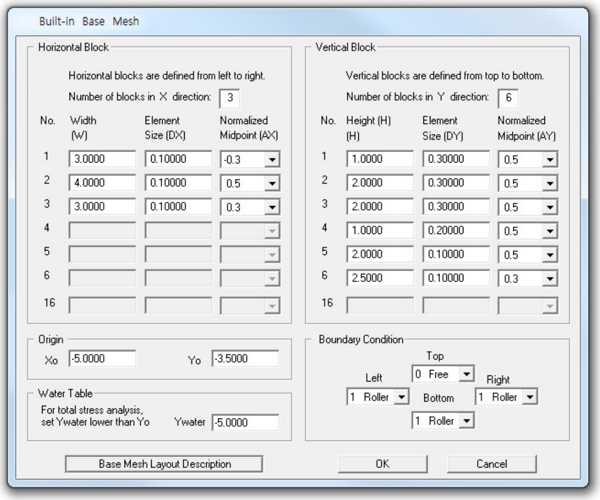


Figure 5.69 Built-in base mesh dialog

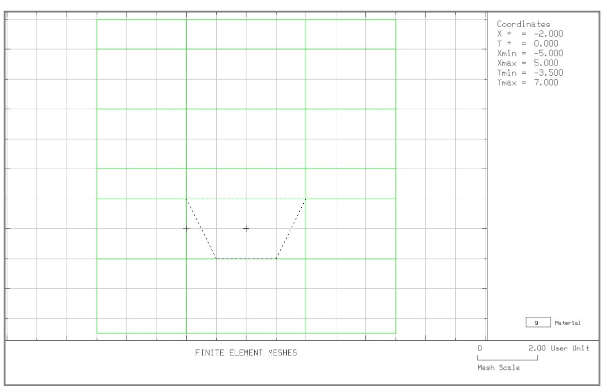


Figure 5.70 Base mesh plot on drawing board

5.4.3 Groups

Group meshes are divided into three parts:

- Natural soil and excavation
- Pipe construction
- Lift

It should be noted that the final finite element meshes are most influenced by group order and IEND.

5.4.3.1 Natural Soil and Excavation

Excavation is performed in natural soil to make trench.
Table 5.20 lists key parameters of these groups

Group	Name	MTYPE	Elem	MATNO / NAC / NDAC	Seg.	Beginning Point		Ending Point		IEND
						X	Y	X	Y	
1	Natural Soil	3	Cont	1 / 0 / 0	1	-5	-3.5	5	-3.5	2
					2	5	-3.5	5	1	2
					3	5	1	-5	1	2
					4	-5	1	-5	-3.5	2
2	Excavation	3	Cont	1 / 0 / 3	1	-1	-1	1	-1	2
					2	1	-1	2	1	2
					3	2	1	-2	1	2
					4	-2	1	-1	-1	2

Table 5.20 Key parameters for natural soil and excavation

Figure 5.71 shows Group dialog for natural soil.

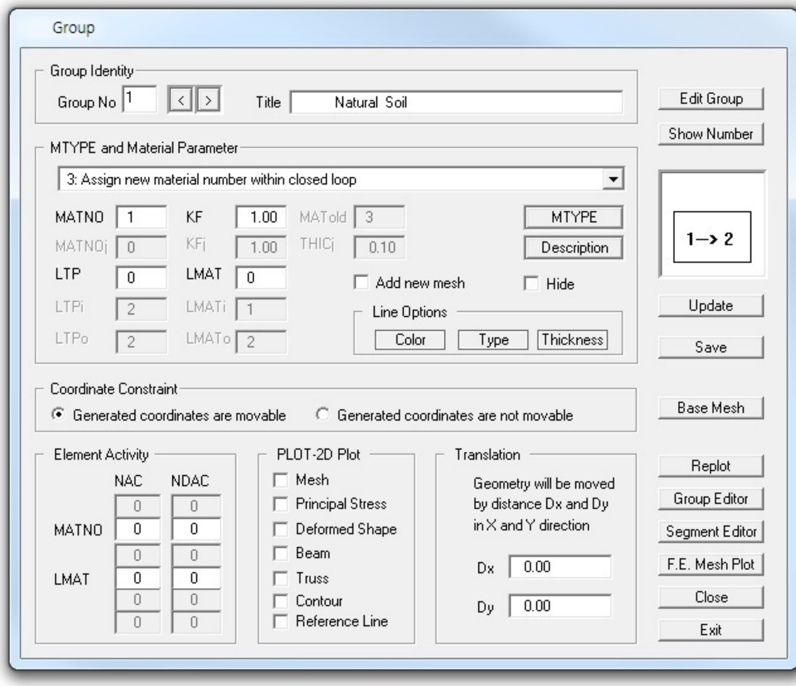


Figure 5.71 Group dialog for natural soil

Figure 5.72 shows Group dialog for excavation.

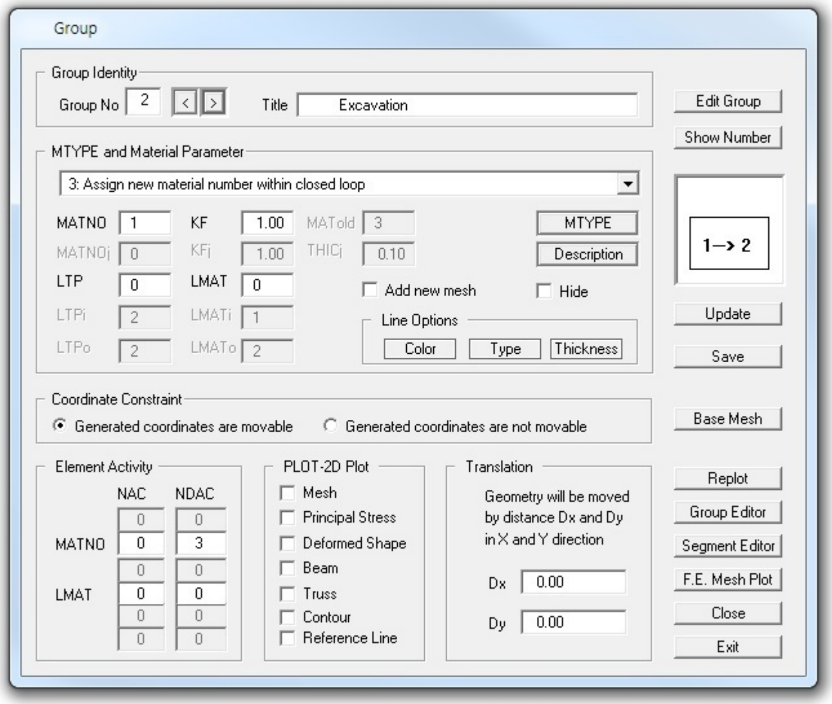


Figure 5.72 Group dialog for excavation

5.4.3.2 Pipe Construction

Pipe construction consists of bedding, steel pipe, and backfill as shown in Figure 5.67. Table 5.21 lists key parameters of these groups

Group	Name	MTYPE	Add New Mesh	Element	MATNO / LMAT	Element Activity	
						NAC	NDAC
3	Bedding	3	Checked	Cont.	2 / 0	4	999
4	Steel Pipe	2		Beam	0 / 1	5	999
5	Backfill	3	Checked	Cont.	3 / 0	5	999

Group	Seg	Line Segment				Arc Segment						IEND
		Beginning Point		Ending Point		Origin		Radius & Angle				
		X	Y	X	Y	X_o	Y_o	R_X	R_Y	Θ_b	Θ_e	
3	1	-1	-1	1	-1							2
	2	1	-1	1.353	-0.294							2
	3	1.353	-0.294	0.4045	-0.294							2
	4					0	0	0.5	0.5	-36	-144	2
	5	-0.4045	-0.294	-1.353	-0.294							2
	6	-1.353	-0.294	-1	-1							2
4	1					0	0	0.5	0.5	0	360	2
5	1	2	1	-2	1							2
	2	-2	1	-1.353	-0.294							2
	3	-1.353	-0.294	-0.4045	-0.294							2
	4					0	0	0.5	0.5	216	-36	2
	5	0.4045	-0.294	1.353	-0.294							2
	6	1.353	-0.294	2	1							2

Table 5.21 Key parameters for pipe construction

Figure 5.73 shows Group dialog for bedding.
Group dialog for backfill is very similar to this group 3.

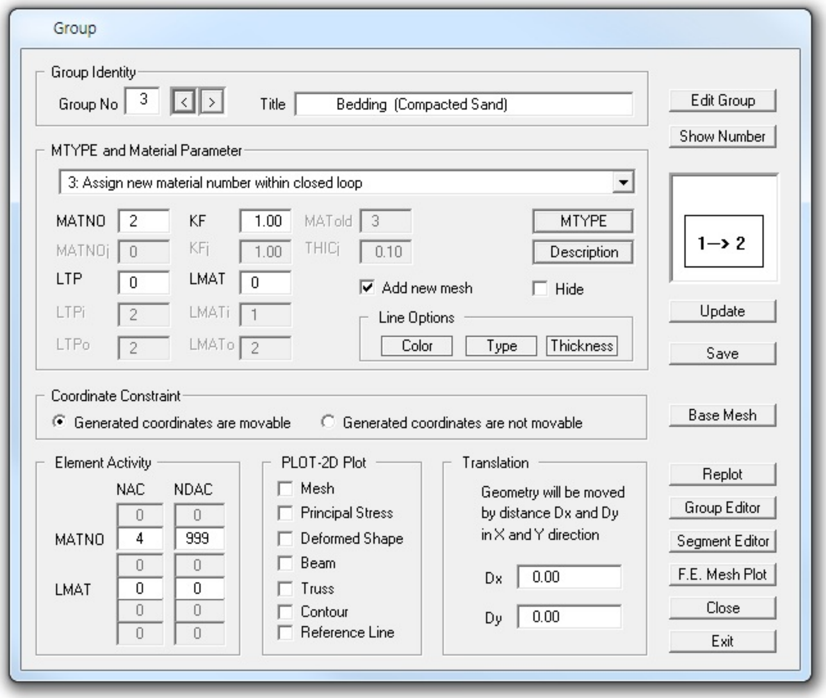


Figure 5.73 Group dialog for bedding

Figure 5.74 shows Group dialog for steel pipe.

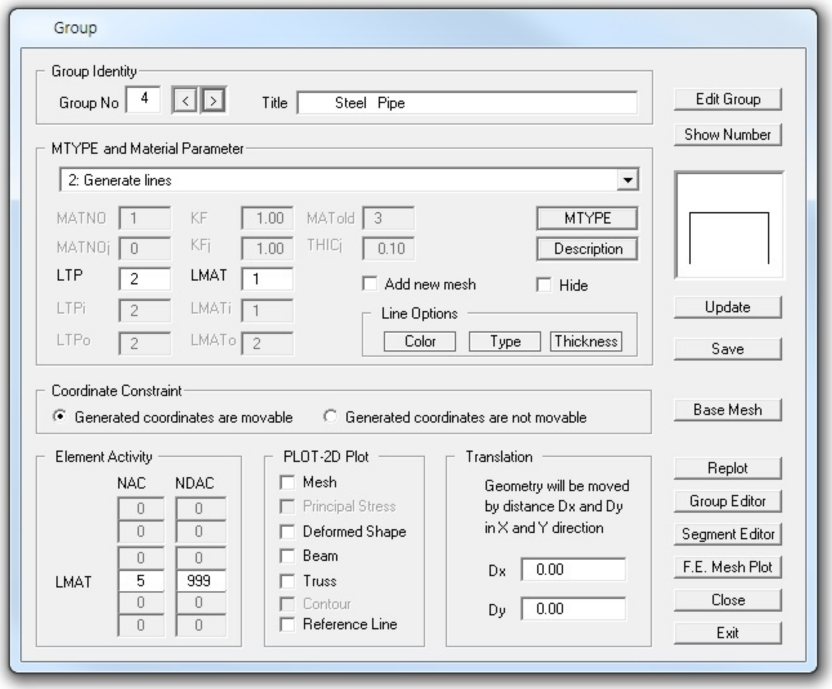


Figure 5.74 Group dialog for steel pipe

5.4.3.3 Lift

Embankment lifts are placed through four steps as shown in Figure 5.67. Table 5.22 lists key parameters of these groups

Group	Name	MTYPE	Element	MATNO / NAC / NDAC	Seg.	Beginning Point		Ending Point		IEND
						X	Y	X	Y	
6	Lift-1	3	Cont	4 / 6 / 999	1	-5	1	5	1	2
					2	5	1	5	2	2
					3	5	2	-5	2	2
					4	-5	2	-5	1	2
7	Lift-2	3	Cont	5 / 7 / 999	1	-5	2	5	2	2
					2	5	2	5	4	2
					3	5	4	-5	4	2
					4	-5	4	-5	2	2
8	Lift-3	3	Cont	6 / 8 / 999	1	-5	4	5	4	2
					2	5	4	5	6	2
					3	5	6	-5	6	2
					4	-5	6	-5	4	2
9	Lift-4	3	Cont	7 / 9 / 999	1	-5	6	5	6	2
					2	5	6	5	7	2
					3	5	7	-5	7	2
					4	-5	7	-5	6	2

Table 5.22 Key parameters for lift

Figure 5.75 shows Group dialog for the first lift.
Group dialogs for other lifts are very similar to this group 6.

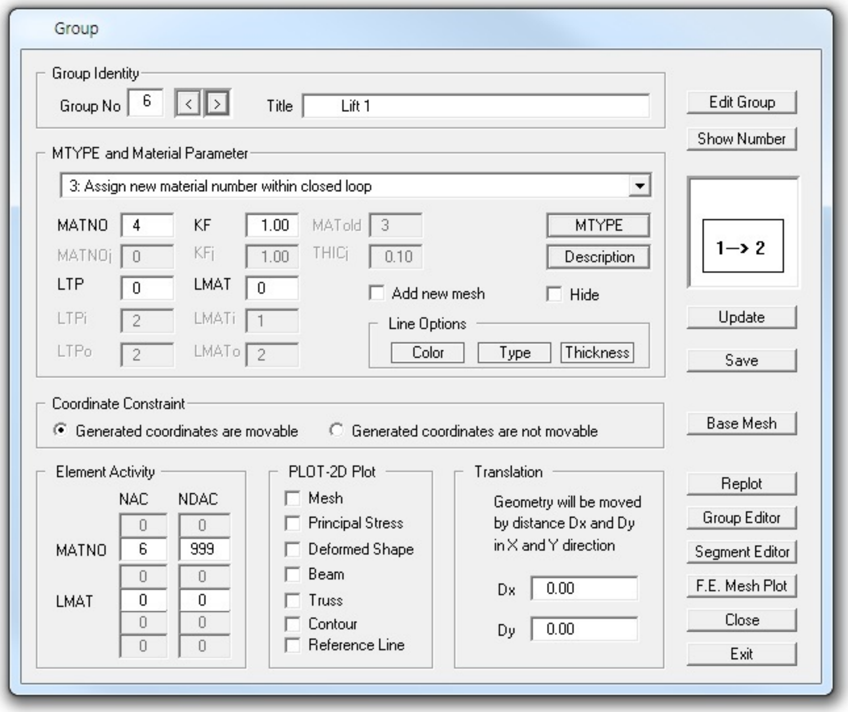


Figure 5.75 Group dialog for first lift

5.4.4 Finite Element Mesh Plot

Figure 5.76 shows finite element meshes generated from group meshes. Finite element meshes near buried pipe are shown in Figure 5.77.

Figure 5.76
Finite element meshes

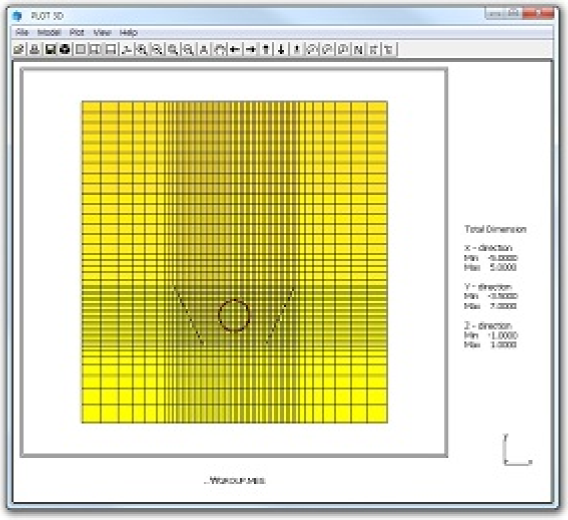
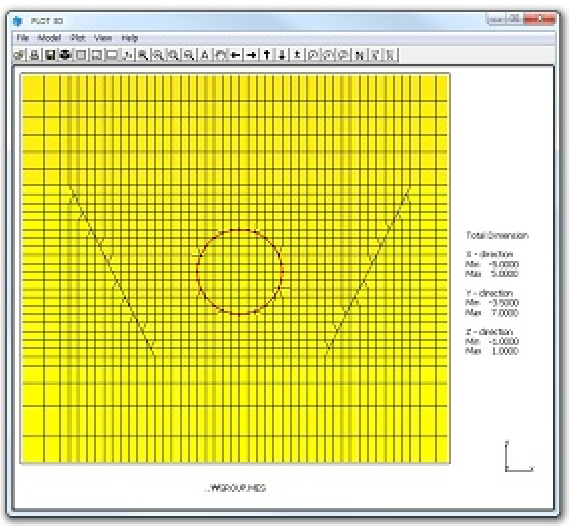


Figure 5.77
Finite element meshes near buried pipe



5.5 Arch Warehouse

This example illustrates how to build group meshes for typical arch warehouse structure.

5.5.1 Overview

The cross section of this arch warehouse consists of soil layer, foundations, and arch frame as shown in Figure 5.78.
Construction sequence is listed in Table 5.23.

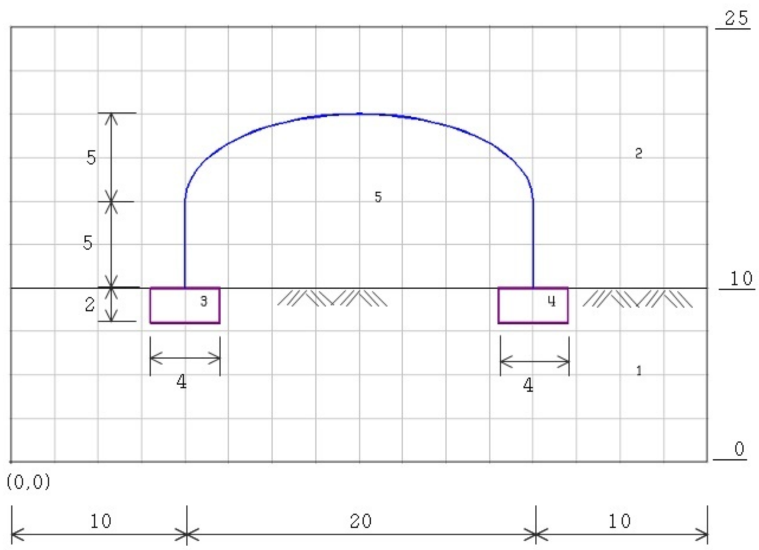


Figure 5.78 Schematic section of arch warehouse

Step	Description
1,2	In situ stress
3	Excavate trench & place foundation
4	Construct steel arch frame

Table 5.23 Construction sequence

A total of 5 groups are used to model this arch warehouse as schematically shown in Figure 5.79: 1 for soil layer, 1 for above ground, 2 for foundations, and 1 for arch frame. Table 5.24 summarizes key parameters of groups.

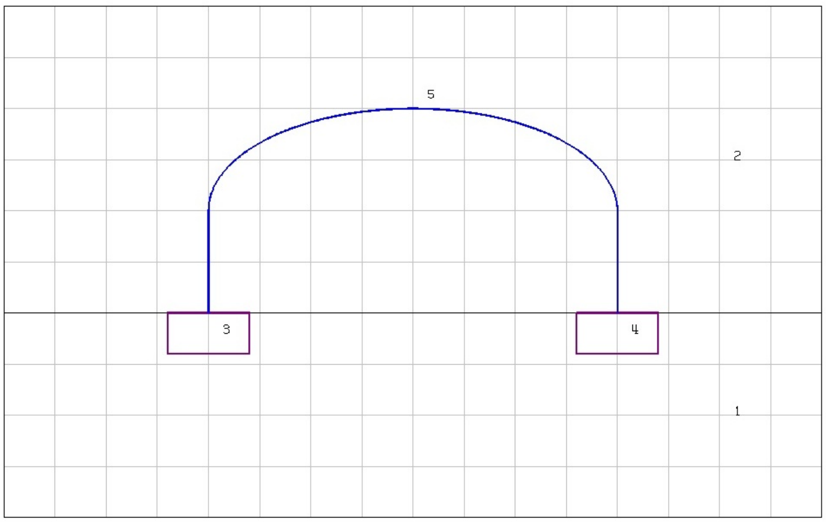


Figure 5.79 Group section view

<table>
<tr><th>Group</th><th colspan="2">Name</th><th>MTYPE</th><th>NAC / NDAC</th><th>MAT_{OLD} / MATNO / LTP / LMAT / IEND</th></tr>
<tr><td>1</td><td colspan="2">Soil Layer</td><td>3</td><td>0 / 0</td><td>1 / 0 / 0 / 0 / 2</td></tr>
<tr><td>2</td><td colspan="2">Above Ground</td><td>1</td><td>0 / 0</td><td>0 / 0 / 0 / 0 / 0</td></tr>
<tr><td rowspan="2">3</td><td rowspan="2">Left Foundation</td><td>MAT_{OLD}</td><td rowspan="2">4</td><td>0 / 3</td><td rowspan="2">2 / 3 / 0 / 0 / 2</td></tr>
<tr><td>MATNO</td><td>3 / 999</td></tr>
<tr><td rowspan="2">4</td><td rowspan="2">Right Foundation</td><td>MAT_{OLD}</td><td rowspan="2">4</td><td>0 / 3</td><td rowspan="2">2 / 3 / 0 / 0 / 2</td></tr>
<tr><td>MATNO</td><td>3 / 999</td></tr>
<tr><td>5</td><td colspan="2">Arch Frame</td><td>2</td><td>4 / 999</td><td>0 / 0 / 2 / 1 / 2
(Checked Add new mesh)</td></tr>
</table>

Table 5.24 Group key parameters

5.5.2 Base Mesh

Built-in Base Mesh dialog is shown in Figure 5.80 with input data for blocks and boundary condition. Figure 5.81 shows base mesh plot on drawing board.

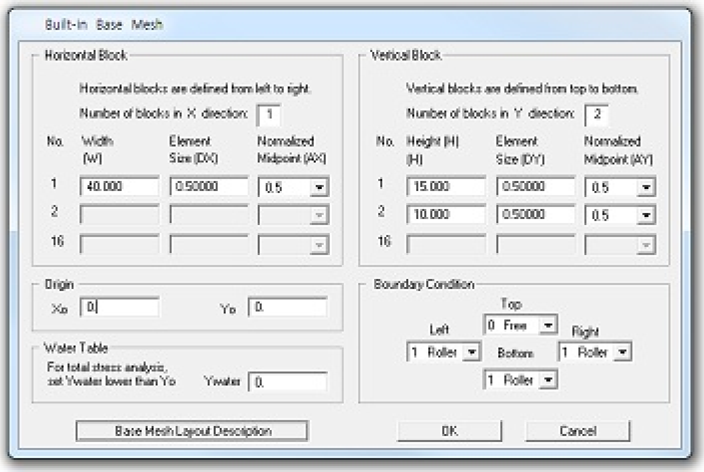


Figure 5.80 Built-in base mesh dialog

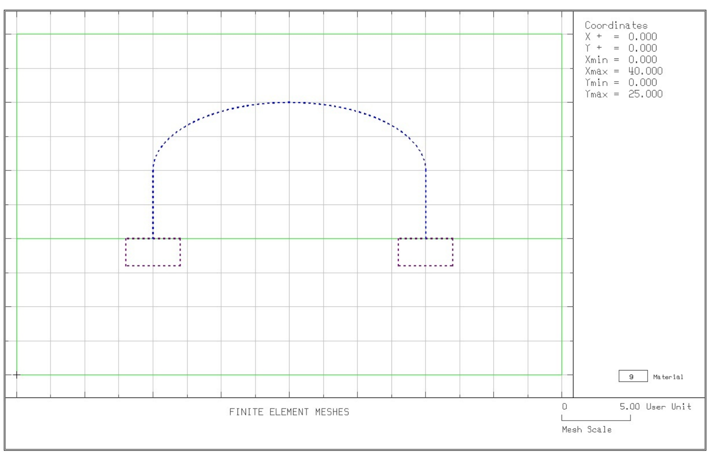


Figure 5.81 Base mesh plot on drawing board

5.5.3 Groups

Group meshes are divided into three parts:

- Soil layer and above ground
- Foundation
- Arch frame

It should be noted that the final finite element meshes are most influenced by group order and IEND.

5.5.3.1 Soil Layer and Above Ground

Above Ground represents upper block of base mesh which will vanish. Table 5.25 lists key parameters of these groups

Group	Name	MTYPE	Elem	MATNO / NAC / NDAC	Seg.	Beginning Point		Ending Point		IEND
						X	Y	X	Y	
1	Soil Layer	3	Cont	1 / 0 / 0	1	0	0	40	0	2
					2	40	0	40	10	2
					3	40	10	0	10	2
					4	0	10	0	0	2
2	Above Ground	1	Cont	0 / 0 / 0	1	0	10	40	10	2
					2	40	10	40	25	2
					3	40	25	0	25	2
					4	0	25	0	10	2

Table 5.25 Key parameters for soil layer and above ground

Figure 5.82 shows Group dialog for soil layer.

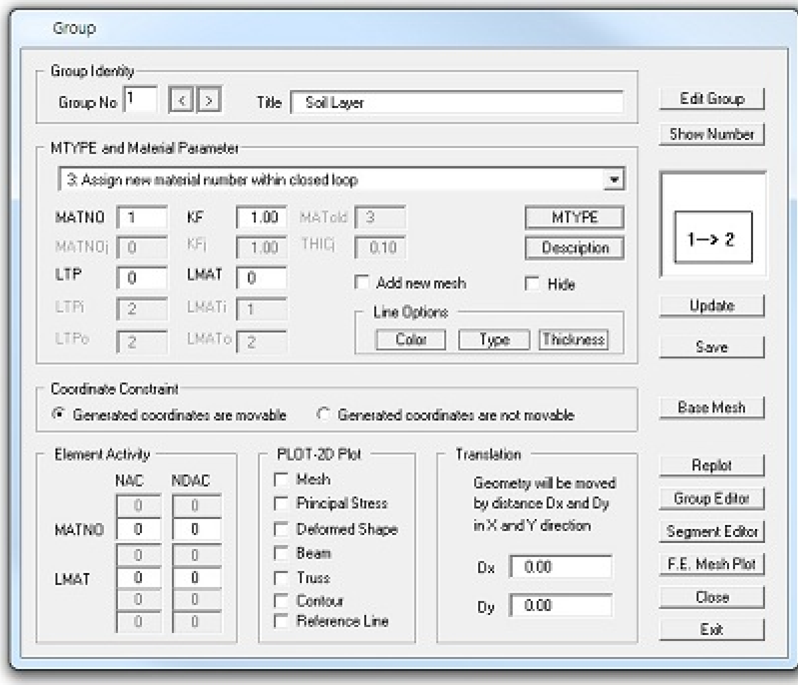


Figure 5.82 Group dialog for soil layer

Figure 5.83 shows Group dialog for above ground.

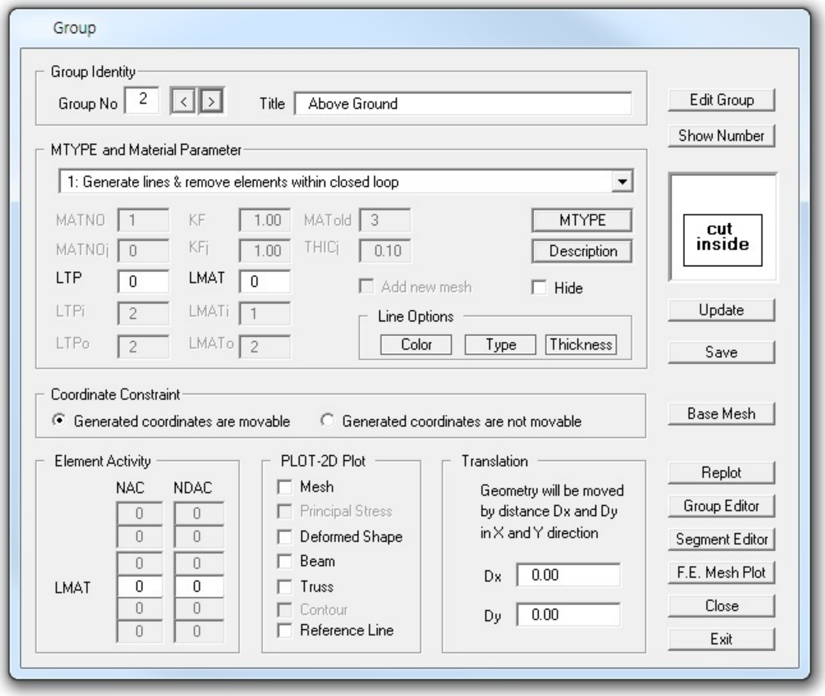


Figure 5.83 Group dialog for above ground

5.5.3.2 Foundation

Each foundation group includes both in situ soils and concrete block such that in situ soils are replaced by concrete block when foundation is built. Table 5.26 lists key parameters of these groups.

Group	Name		NAC / NDAC	MTYPE Elem	Seg.	Beginning Point		Ending Point		IEND
						X	Y	X	Y	
3	Left Foundation	$MAT_{OLD}=2$	0 / 3	4 Cont	1	8	8	12	8	2
					2	12	8	12	10	2
		MATNO=3	3 / 999		3	12	10	8	10	2
					4	8	10	8	8	2
4	Right Foundation	$MAT_{OLD}=2$	0 / 3	4 Cont	1	28	8	32	8	2
					2	32	8	32	10	2
		MATNO=3	3 / 999		3	32	10	28	10	2
					4	28	10	28	8	2

Table 5.26 Key parameters for foundation

Figure 5.84 shows Group dialog for left foundation.
Group dialog for right foundation is very similar to this group 3.

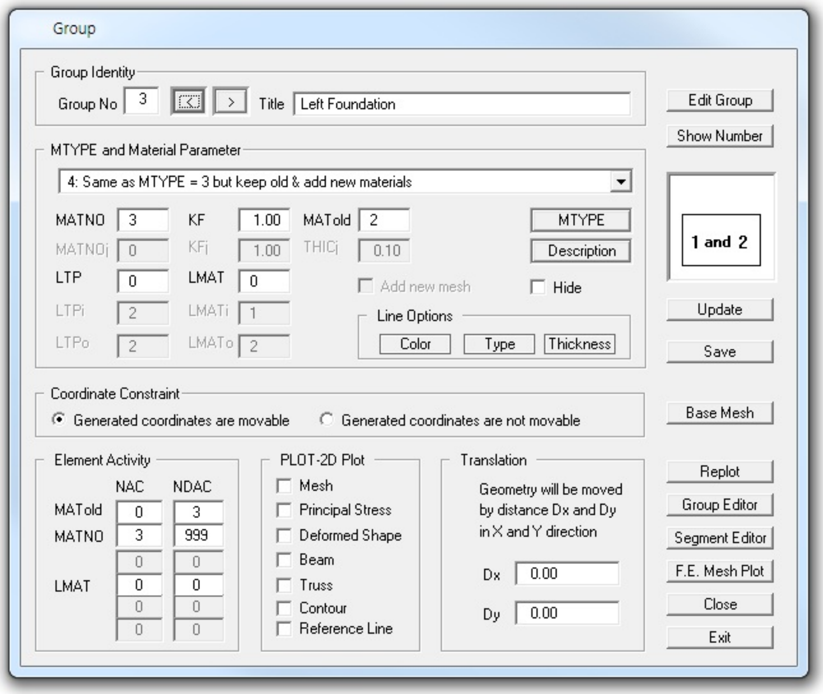


Figure 5.84 Group dialog for left foundation

5.5.3.3 Arch Frame

Arch Frame is the only structure in the upper block of base mesh since the Above Ground group generates void space. Table 5.27 lists key parameters of this group.

Group	Name	MTYPE	Element	LTP / LMAT	Element Activity	
					NAC	NDAC
5	Arch Frame	2	Beam	2 / 1	4	999

Group	Seg	Line Segment				Arc Segment						NDIV	IEND
		Begin. Pt.		Ending Pt.		Origin		Radius & Angle					
		X	Y	X	Y	X_o	Y_o	R_X	R_Y	Θ_b	Θ_e		
5	1	30	10	30	15							5	2
	2					20	15	10	5	0	180	20	2
	3	10	15	10	10							5	2

Table 5.27 Key parameters for arch frame

Figure 5.85 shows Group dialog for the arch frame. It should be noted that Add new mesh be checked.

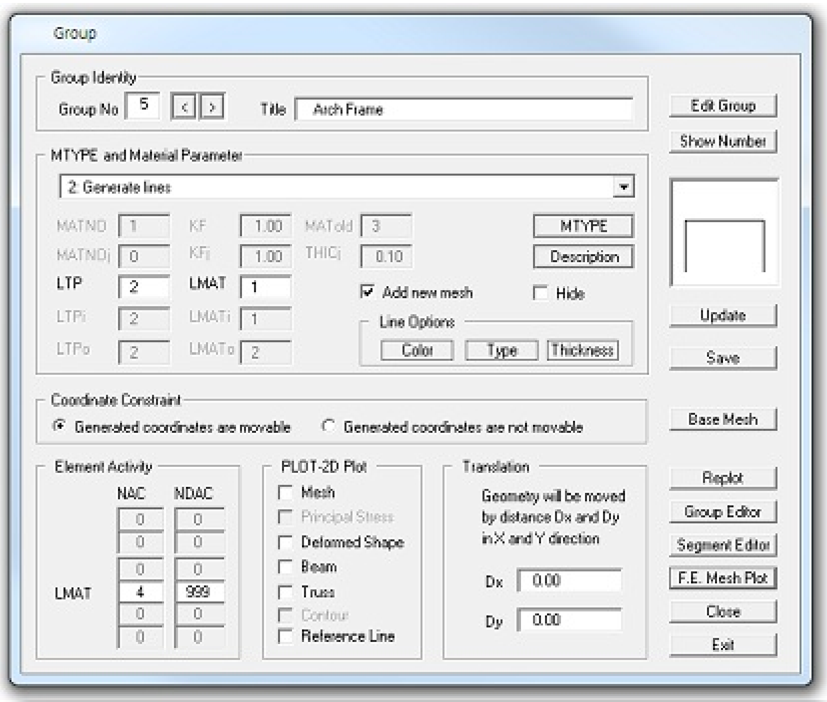


Figure 5.85 Group dialog for arch frame

5.5.4 Finite Element Mesh Plot

Figure 5.86 shows finite element meshes generated from group meshes.

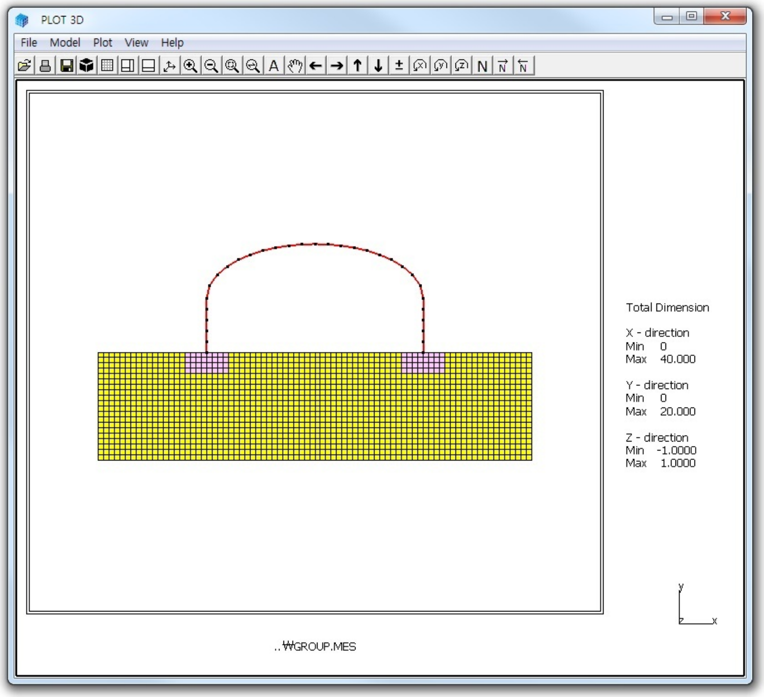


Figure 5.86 Finite element meshes

5.6 Finite Element Mesh Modification

This example illustrates how to modify existing finite element meshes using Mesh Generator.

5.6.1 Overview

When you open input file, Mesh Generator reads the extension of the input file name and it assumes that the input file is the finite element mesh file if the extension is .Mes.

Editing finite element meshes has three parts: Nodal Boundary, Nodal Coordinate and Element Material. These editing modes can be accessed from Mesh menu in PLOT-2D as shown in Figure 5.87.

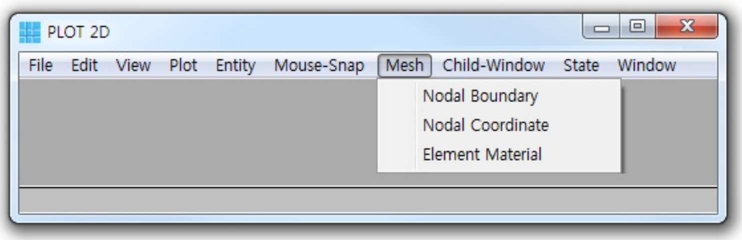


Figure 5.87 Menu for editing finite element mesh

It should be noted that once you edited the finite element meshes, modified finite element mesh is saved as MeshFile.Mes in the current working directory. The original input mesh file is not changed.

Figure 5.88 shows existing finite element mesh with six layers of natural soils. The top layer of this existing mesh is to be replaced by sand embankment with reduced width as schematically shown in Figure 5.89.

This modification involves following three works:

- Change top surface nodal coordinates
- Change top surface nodal boundaries
- Change top layer element materials

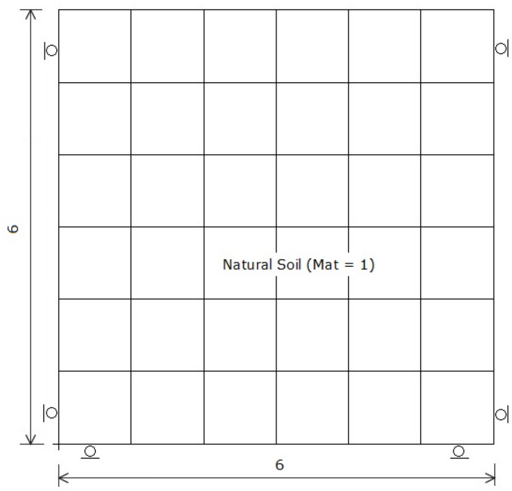


Figure 5.88 Existing finite element mesh

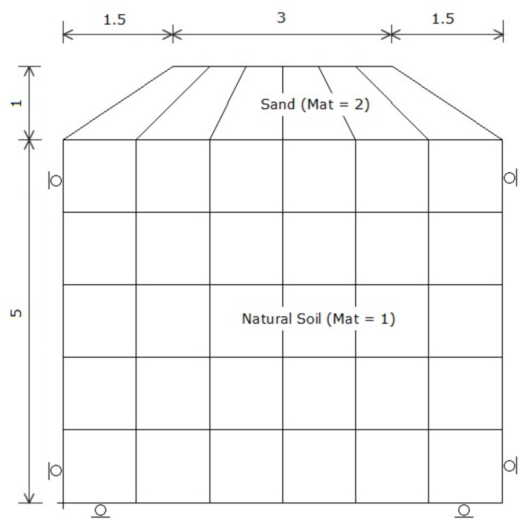


Figure 5.89 Modified finite element mesh

5.6.2 Change Top Surface Nodal Coordinates

Click Nodal Coordinate from the Mesh menu, then Edit Coordinate dialog in Figure 5.90 is displayed.

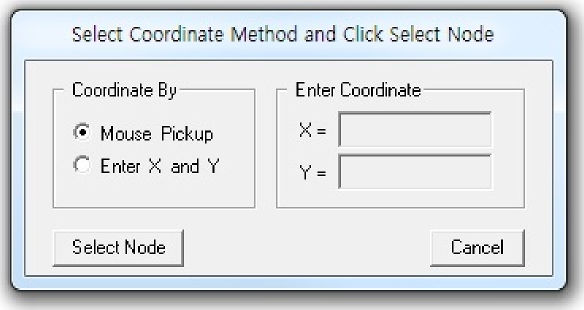


Figure 5.90 Edit coordinate dialog

For this example, Snap to Half of Grid in Figure 5.91 is the most convenient method for Mouse Pickup.

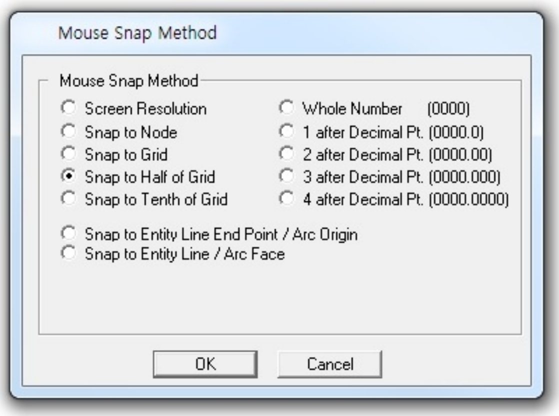


Figure 5.91 Mouse snap method

Click Select Node in Figure 5.90.
When you select the node by Mouse Right Click, the selected node is marked as an open circle on the drawing board as in Figure 5.92.

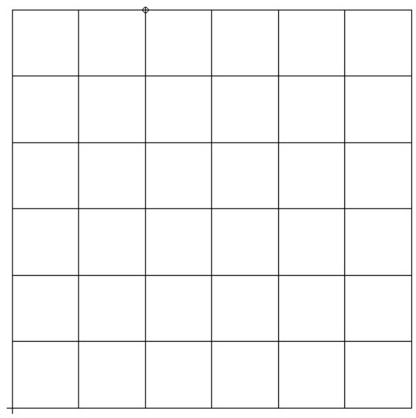


Figure 5.92 First selected node on drawing board

Now, move the first selected node by using drag-and-drop of Mouse Left Button as shown in Figure 5.93.

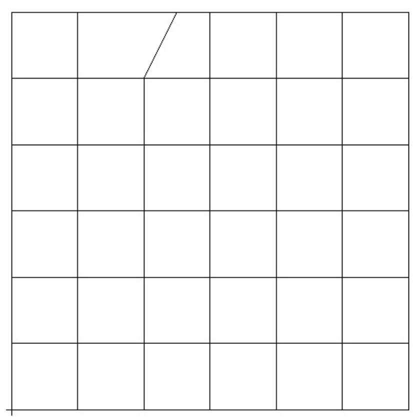


Figure 5.93 New position of first selected node

Select the next node by Mouse Right Click as shown in Figure 5.94.

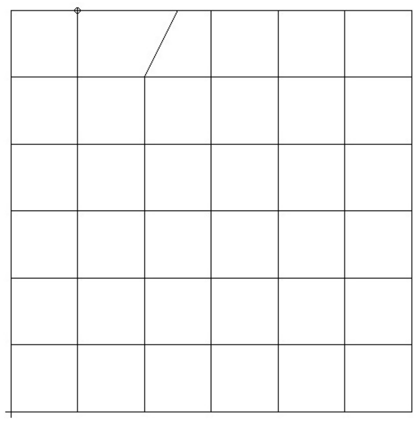


Figure 5.94 Second selected node on drawing board

Now, move the second selected node by using drag-and-drop of Mouse Left Button as shown in Figure 5.95.

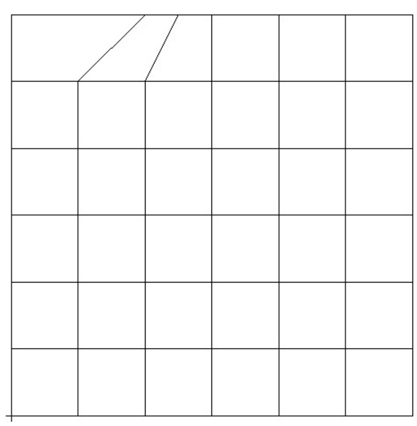


Figure 5.95 New position of second selected node

Repeat the same procedure for all other nodes on the top surface. Once finished, click Finish button in Figure 5.96.

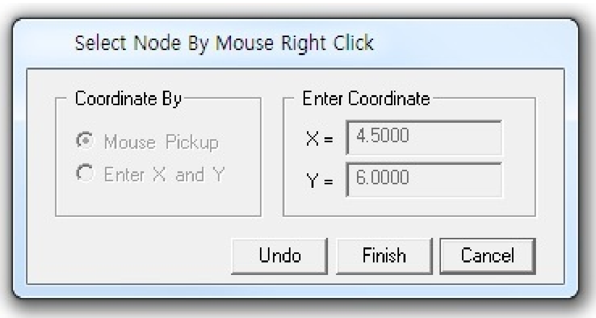


Figure 5.96 Edit coordinate dialog

Figure 5.97 shows final finite element mesh on the drawing board.

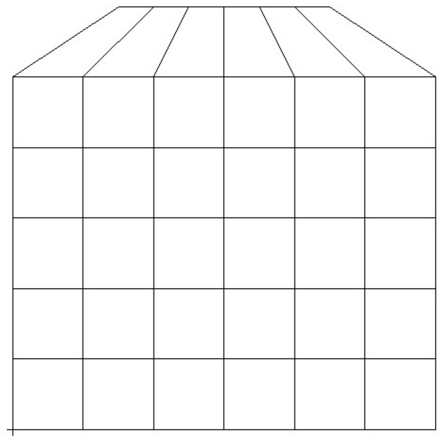


Figure 5.97 Final finite element mesh

5.6.3 Change Top Surface Nodal Boundaries

Click Nodal Boundary from the Mesh menu, then Edit Boundary Code dialog in Figure 5.98 is displayed.

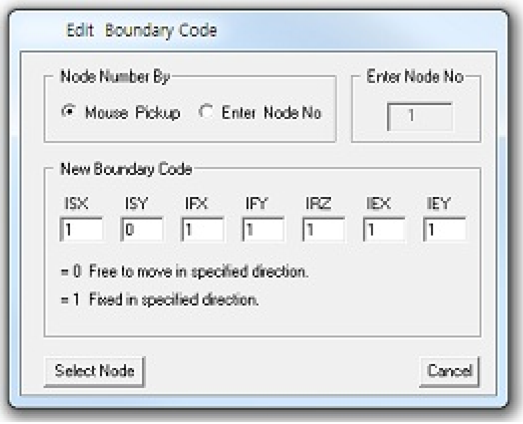


Figure 5.98
Edit boundary dialog

Click Select Node in Figure 5.98.
When you select the node by Mouse Right Click, the selected node is marked as an open circle on the drawing board as in Figure 5.99.

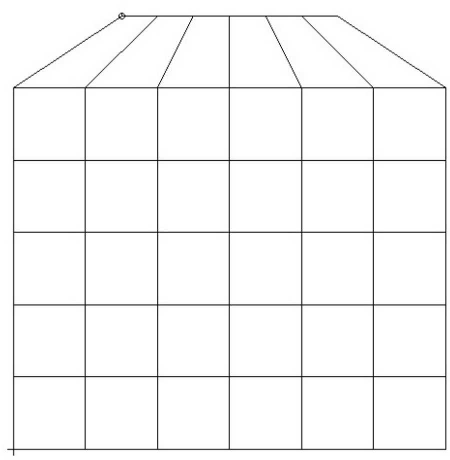
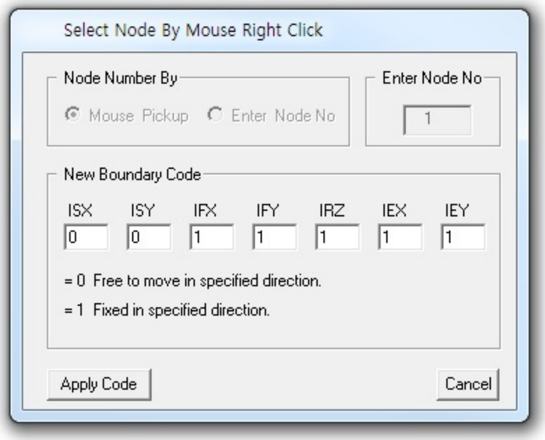


Figure 5.99
Selected node on drawing board

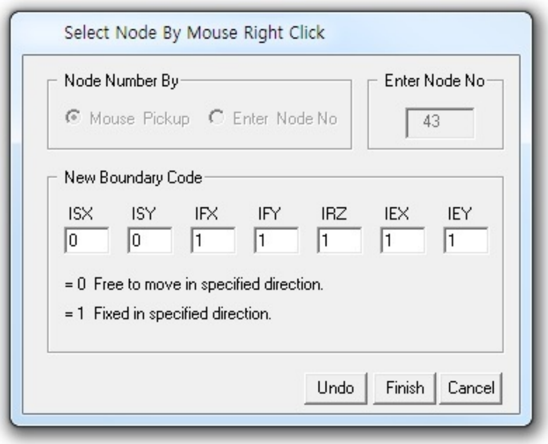
Change the boundary codes as in Figure 5.100 so that the top left node can be free to move in both horizontal and vertical directions and then click Apply Code button.

Figure 5.100
Modified boundary code for top left node



In the same way, select the top right node, modify boundary codes, and click Apply Code. Since all boundary codes are modified, click Finish button in Figure 5.101.

Figure 5.101
Modified boundary code for top right node



Click General View from the View menu. Select Skeleton Boundary Code in General View Options dialog as shown in Figure 5.102 and then click OK button. Modified skeleton boundary codes are shown in Figure 5.103.

Figure 5.102
General view
for skeleton boundary code

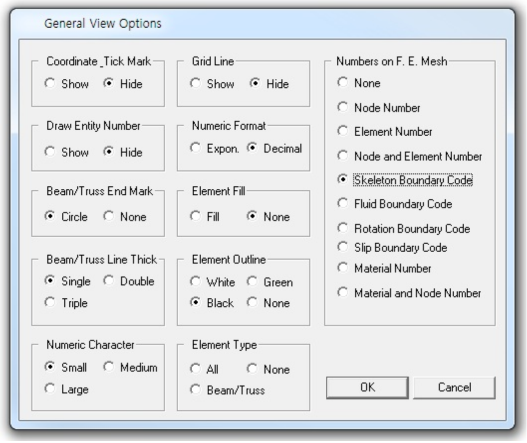
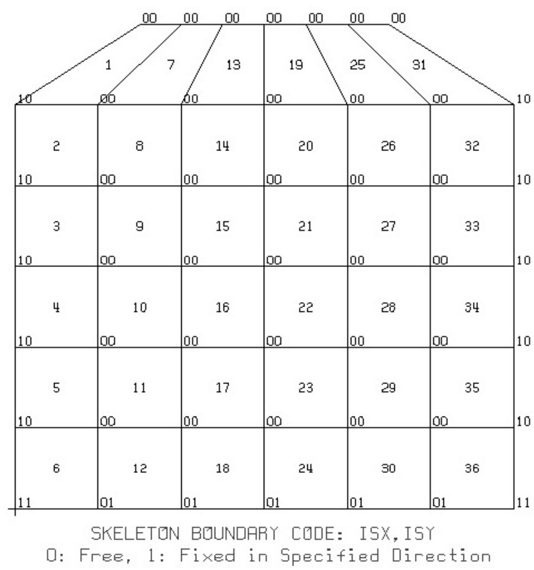


Figure 5.103
Modified skeleton
boundary code plot

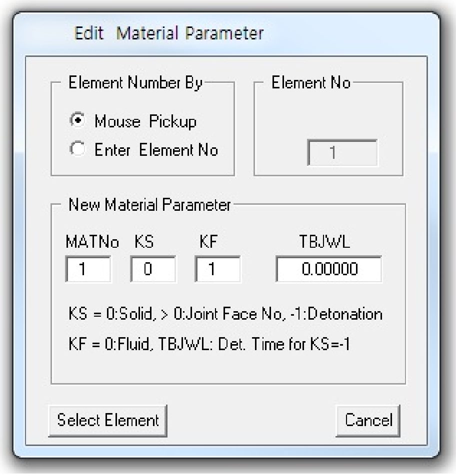


5.6.4 Change Top Layer Element Materials

Click Element Material from the Mesh menu, then Edit Material Parameter dialog in Figure 5.104 is displayed.

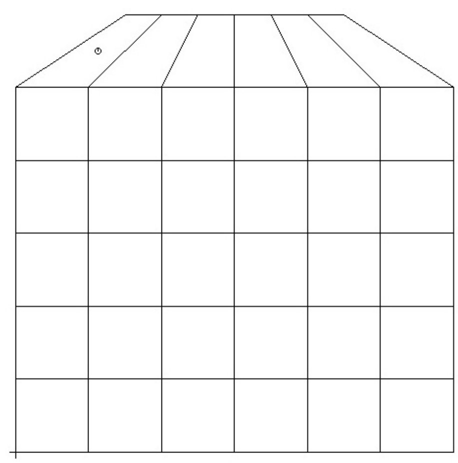
Figure 5.104

Edit element material dialog



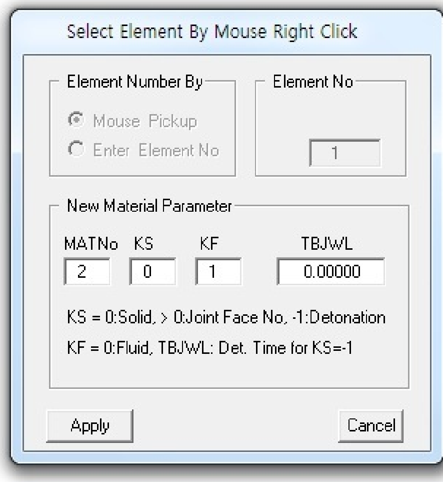
Click Select Element button.
Click the element on the top layer by Mouse Right Click.
Selected element is marked as an open circle as shown in Figure 5.105.

Figure 5.105
Selected element on drawing board



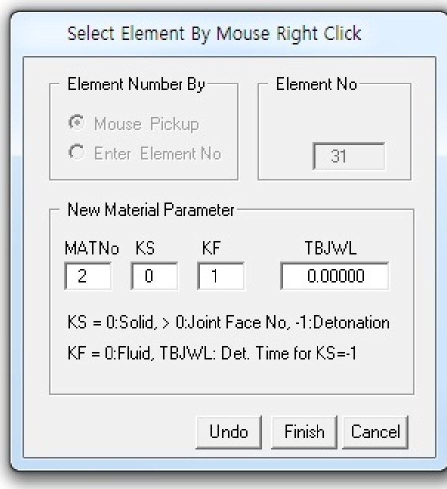
Change the material number as shown in Figure 5.106 and then click Apply button.

Figure 5.106
Modified material number for element 1



Repeat the same procedure for the other elements on the top layer. Once finished, click Finish button in Figure 5.107.

Figure 5.107
Modified material number for element 31



Click General View from the View menu. Select Material Number in General View Options dialog as shown in Figure 5.108 and then click OK button. Modified material number is shown in Figure 5.109.

Figure 5.108
General view
for material number

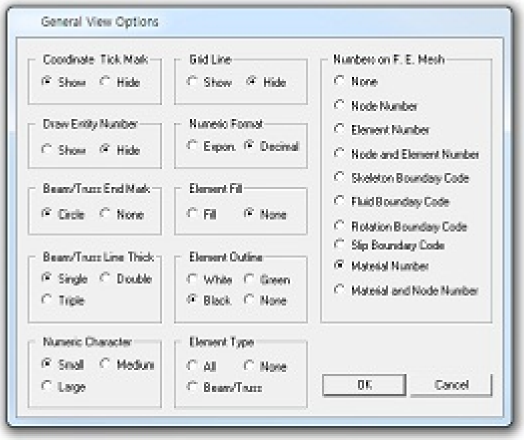
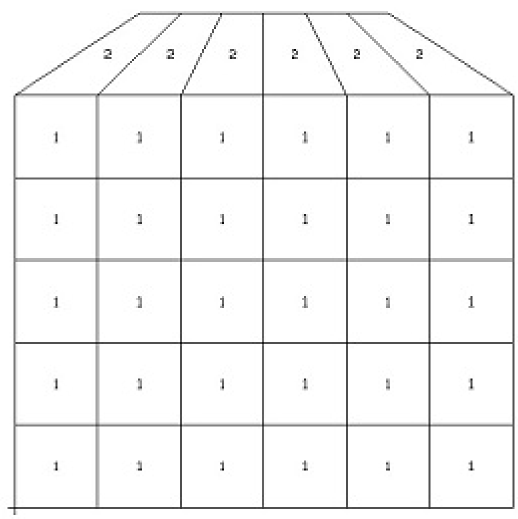


Figure 5.109
Modified material number
plot



Block Mesh
Example Problem

Block Mesh Generator is a three-dimensional CAD program specially designed to build block mesh which can be used to generate finite element mesh with the aid of program PRESMAP-GP. Block Mesh User's Manual describes all the basic functions associated with block mesh generation and modifications.

Five example problems are presented:

1. Single Element
   Shows step by step procedure to create block mesh.
2. Cube Foundation
   Builds block mesh for cube foundation.
3. Hemispherical Shell
   Builds block mesh for hemispherical shell subjected to concentrated loads.
4. Horseshoe Tunnel
   Builds block mesh for typical horseshoe tunnel with reinforced concrete lining.
5. Space Truss
   Builds directly finite element mesh for space truss.

6.1 Single Element

The main objective of this first example is to show the step by step procedure to create block mesh.

This example is to build single cube element in Figure 6.1 by using block mesh generator. This single element is subjected to undrained uniaxial strain loading.

This example involves following seven main steps:
1. Access block mesh generator
2. Set work plane
3. Build cube entity
4. Build hexahedron block
5. Edit block boundary code
6. View skeleton boundary code
7. Plot finite element mesh

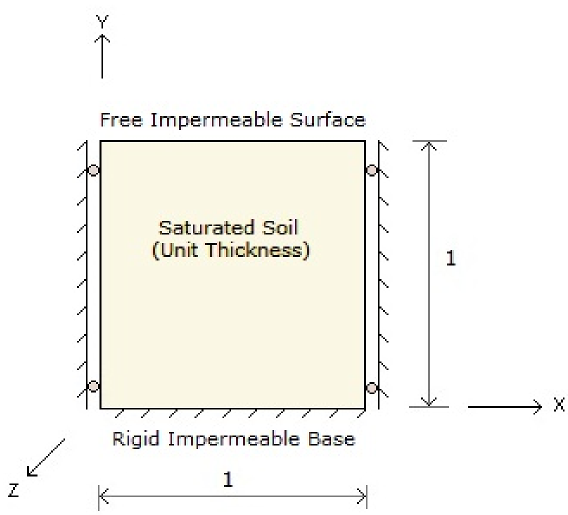


Figure 6.1 Single element in uniaxial strain condition

Step 1: Access Block Mesh Generator (New)

Access Block Mesh Generator by following menu items in SMAP
Run → Mesh Generator → Block Mesh → New

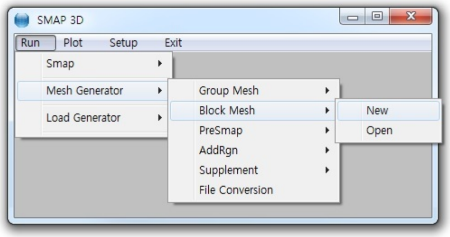


Figure 6.2 Accessing block mesh generator

Step 2: Set Work Plane

Prebuilt Work Plane is displayed on drawing board along with Work Plane Editor dialog. Modify NDx and Wx in Figure 6.3 and click Update.

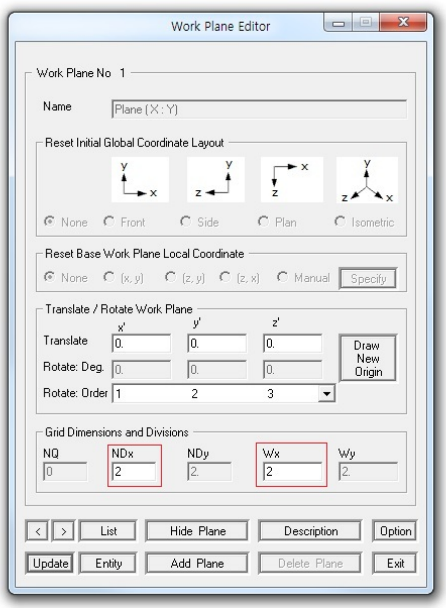


Figure 6.3 Work plane editor

Step 3: Build Cube Entity

1. Click Entity button in Figure 6.3.
2. Entity Editor dialog is displayed as in Figure 6.4.

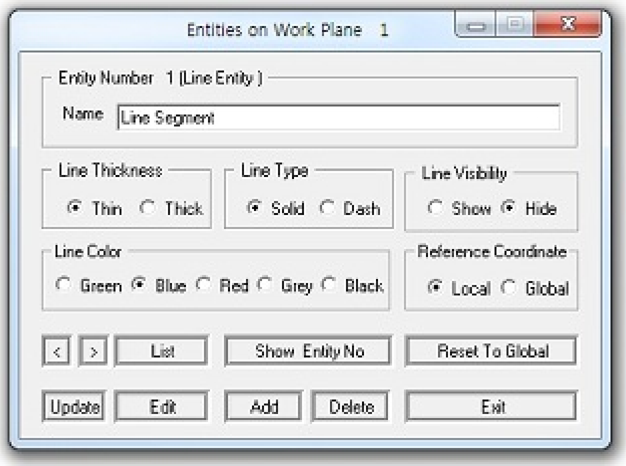


Figure 6.4 Entity editor

3. Click Add button in Figure 6.4.
4. Select Cube entity and click OK button in Figure 6.5.

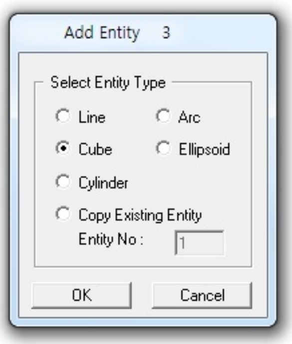


Figure 6.5 Entity type selection

5. Modify input fields of Lx, Ly, and Lz as shown in Figure 6.6.
6. Click Draw Cube Entity button.

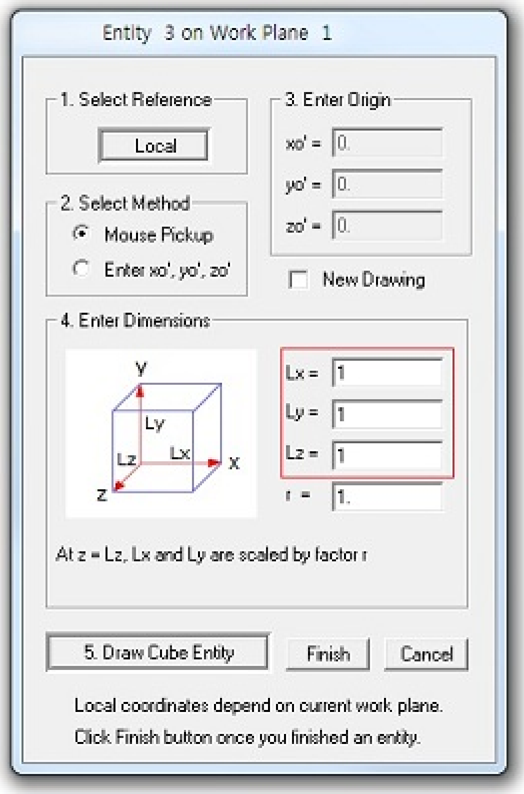


Figure 6.6 Cube entity

7. Coordinates on Work Plane dialog is displayed as in Figure 6.7.
8. Click Info button to see the notes on Mouse Actions on Work Plane as shown in Figure 6.8.

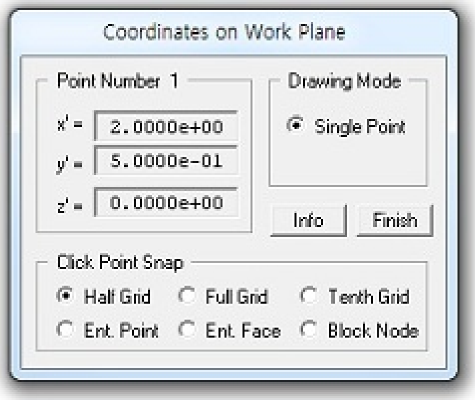


Figure 6.7 Coordinates on work plane

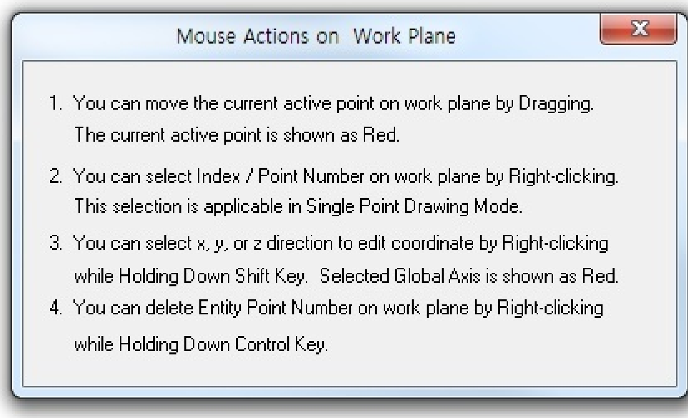


Figure 6.8 Mouse actions on work plane

9. Click Axis toolbar as shown in Figure 6.9.

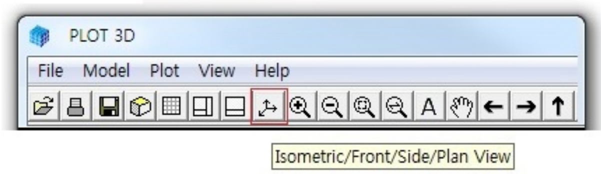


Figure 6.9 Axis toolbar

10. Click Mouse at the origin of coordinates.
11. Cube entity is shown on isometric work plane in Figure 6.10.

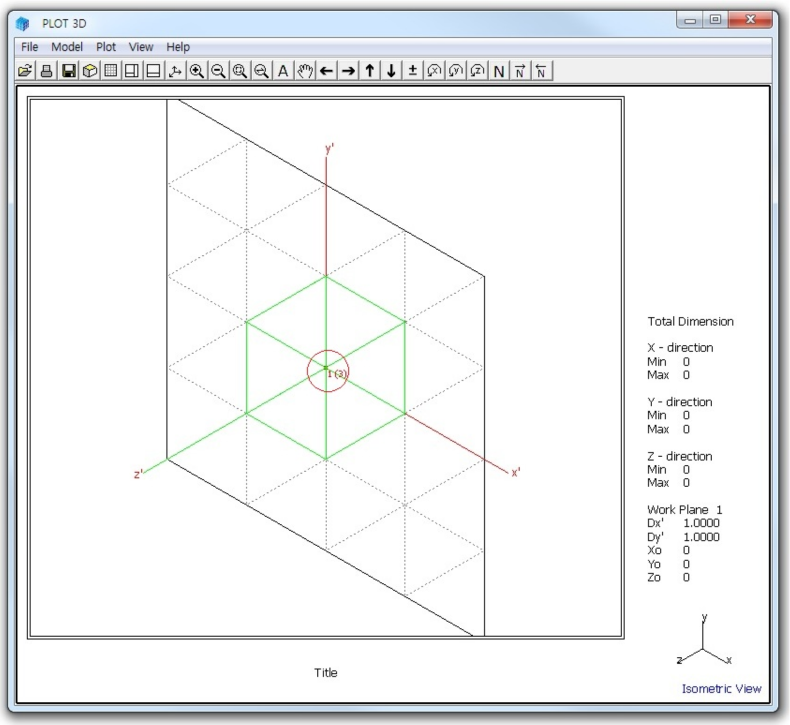


Figure 6.10 Cube entity on isometric work plane

12. Click Finish in Figure 6.7.
13. Click Finish in Figure 6.6.
14. Select Global for Reference Coordinate in Figure 6.11.
15. Click Reset To Global and then Exit buttons in Figure 6.11.

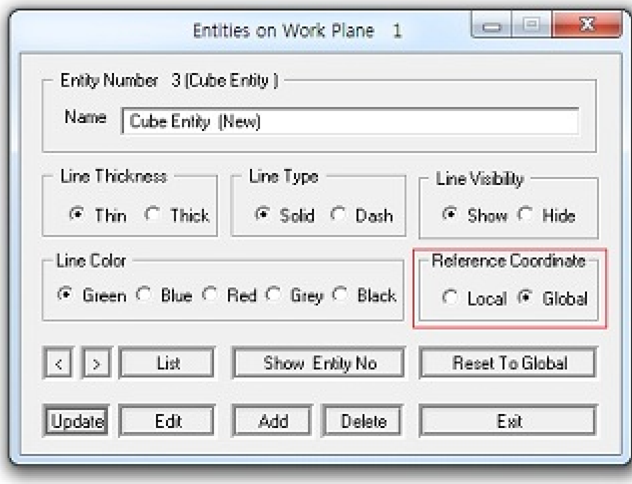


Figure 6.11 Entity editor

Step 4: Build Hexahedron Block

1. Click Block Editor toolbar in Figure 6.12.

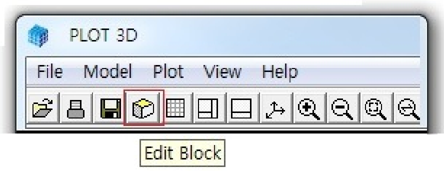


Figure 6.12 Block editor toolbar

2. Select Hexa for block type and click OK in Figure 6.13.

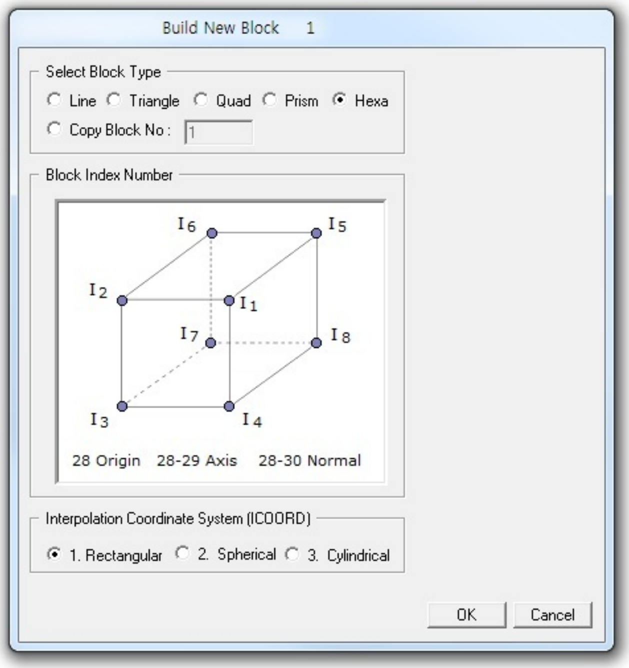


Figure 6.13 Block type selection

3. Click Draw Index Number in Figure 6.14.
4. Coordinates on Work Plane dialog is displayed in Figure 6.15.

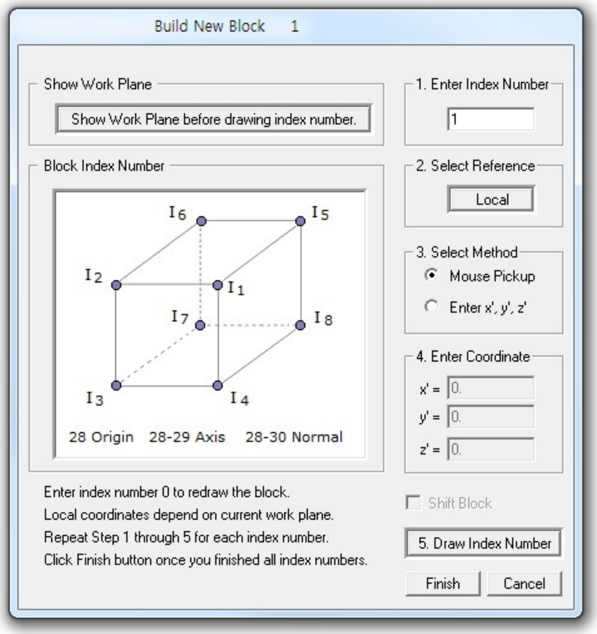


Figure 6.14 Hexa block

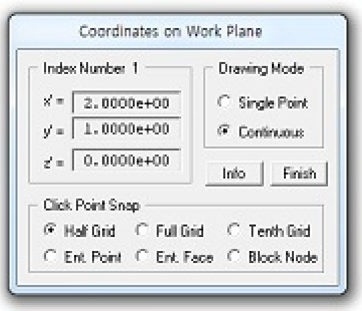


Figure 6.15 Coordinates on work plane

5. Translate work plane as in Figure 6.16 and click Update button.

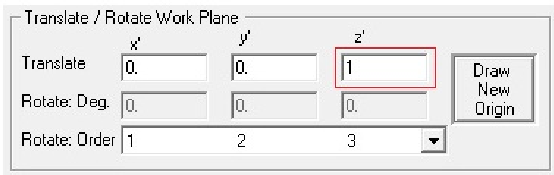


Figure 6.16 Work plane translation (z' = 1)

6. Click the points for index numbers on front surface as in Fig. 6.17.

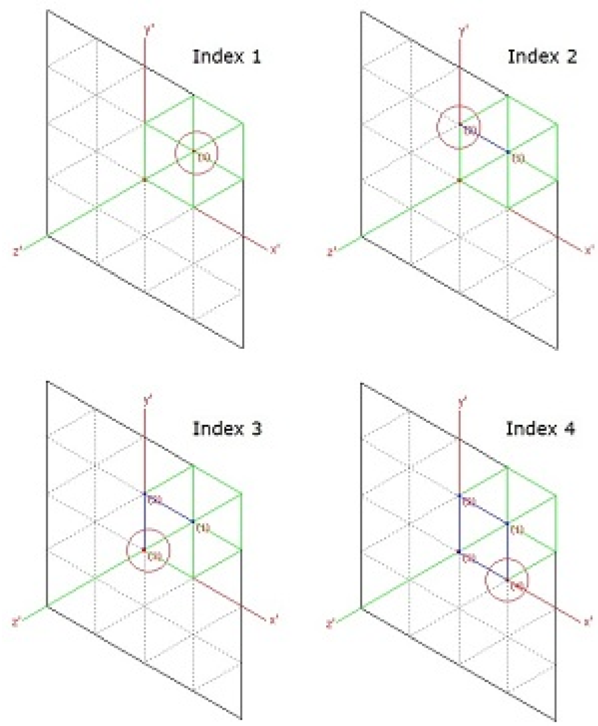


Figure 6.17 Index numbers on front surface

7. Translate work plane as in Figure 6.18 and click Update button.

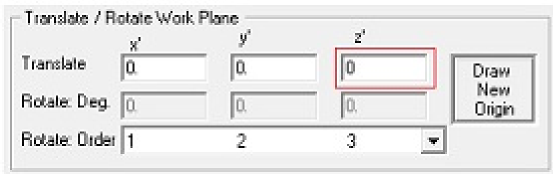


Figure 6.18 Work plane translation (z' = 0)

8. Click the points for index numbers on back surface as in Figure 6.19.

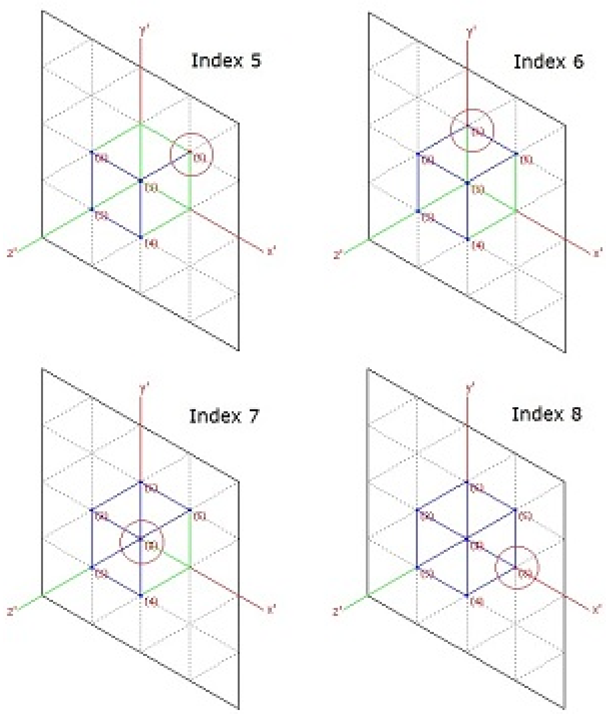


Figure 6.19 Index numbers on back surface

Now, the geometry of hexahedron block is completed.
9. Click Finish in Figure 6.20 and then click Finish in Figure 6.14.

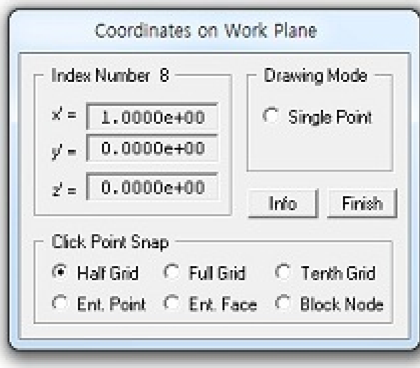


Figure 6.20 Coordinates on work plane

10. Get back to Work Plane Editor dialog and click Entity.
11. Select Entity Number 3, Hide for line visibility, click Update, and click Exit in Figure 6.21.

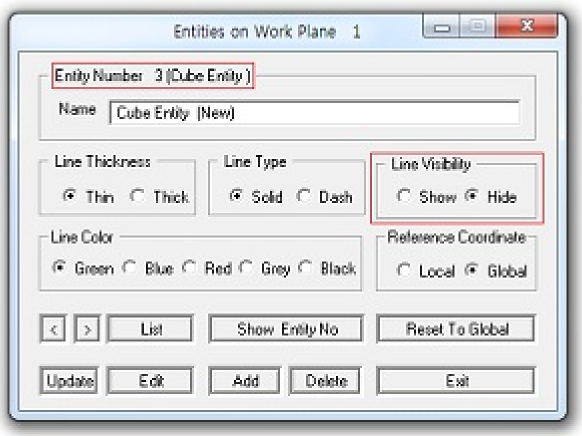


Figure 6.21 Entity editor

12. Modify Title and Material & Element Generation Parameters in Block Editor as shown in Figure 6.22.
13. Click Save and type in file name as EX1.

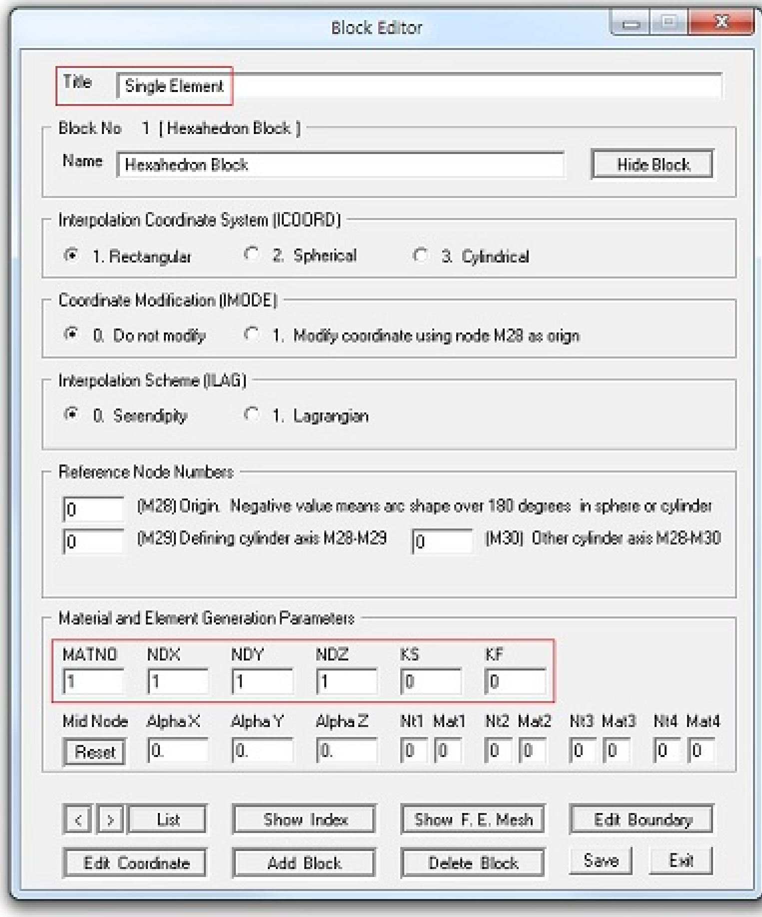


Figure 6.22 Block editor

Step 5: Edit Block Boundary Code

1. Click Edit Boundary in Figure 6.22.
2. Set the boundary codes as shown in Figure 6.23.
3. Click IBTYPE button to see description of boundary type in Fig. 6.24.
4. Click Update and then OK buttons.

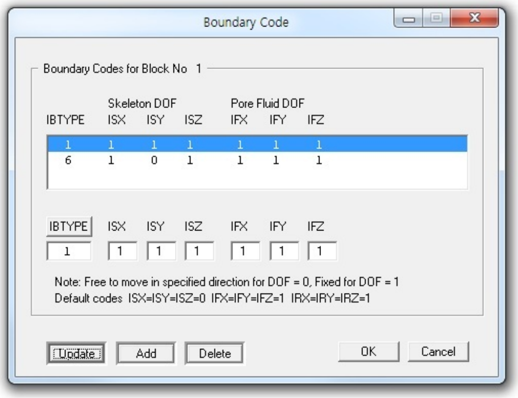


Figure 6.23 Boundary code editor

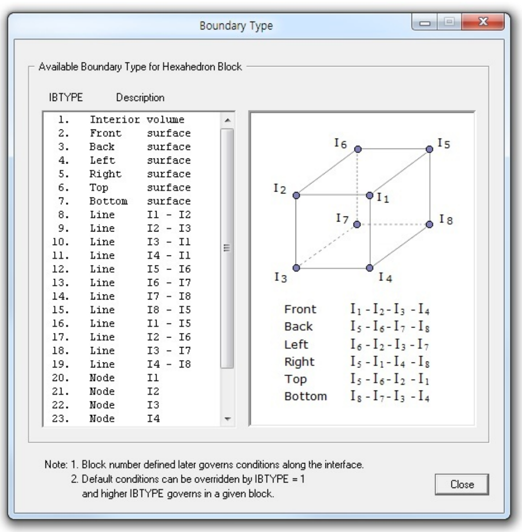


Figure 6.24 Boundary type for hexa block

Step 6: View Skeleton Boundary Code

1. Select View → General in PLOT-3D menu.
2. Select Skeleton Boundary Code and click OK in Figure 6.25.
3. Click Save in Figure 6.22.

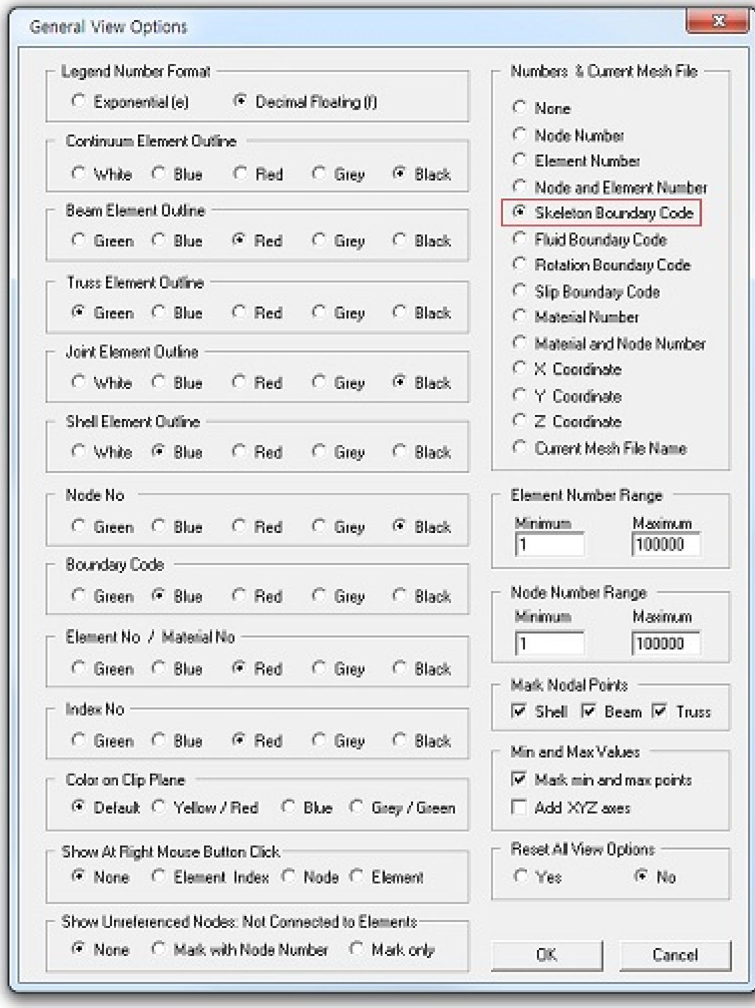


Figure 6.25 General view options

4. Click Show Numbers toolbar as shown in Figure 6.26.

Figure 6.26 Show numbers toolbar

5. Skeleton boundary codes are shown in Figure 6.27.

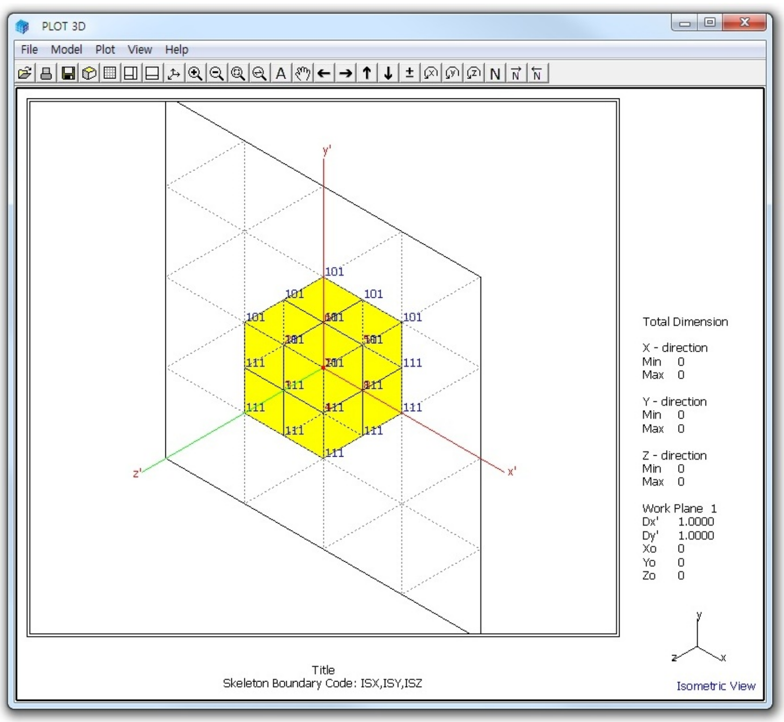


Figure 6.27 Skeleton boundary codes on drawing board

Step 7: Plot Finite Element Mesh

1. Click Show F. E. Mesh in Figure 6.22.
2. Rotate the finite element mesh as shown in Figure 6.28.

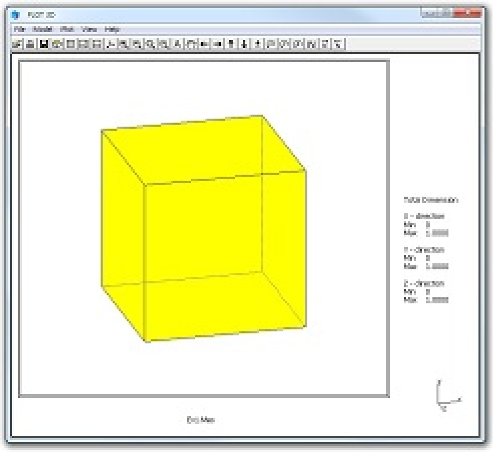


Figure 6.28 Finite element mesh

3. Follow same procedure to plot skeleton boundary codes in Step 6.
4. Figure 6.29 shows skeleton boundary codes for finite element mesh.

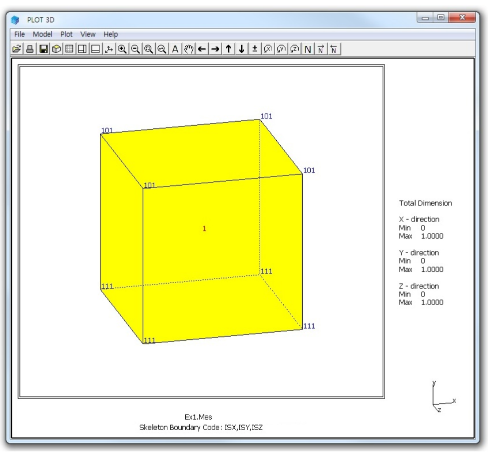


Figure 6.29 Skeleton boundary codes

6.2 Cube Foundation

This example illustrates how to build block mesh for cube foundation. Cube foundation has the dimensions of 100 x 100 x 100 units with all roller boundaries except free on top surface.

This example has the following two parts:

Part 1: Creating Cube Foundation (Figure 6.30)

- Access block mesh generator (New)
- Set work plane
- Build hexahedron block
- Edit block boundary
- Set global boundary
- View skeleton boundary code
- Plot finite element mesh

Part 2: Modifying Cube Foundation (Figure 6.31)

- Access block mesh generator (Open)
- Modify element generation parameters
- Plot finite element mesh

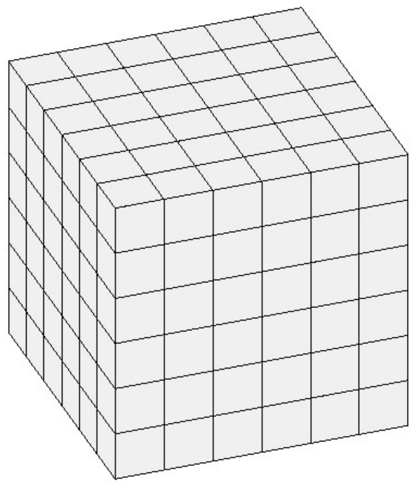


Figure 6.30 Cube foundation with constant element size

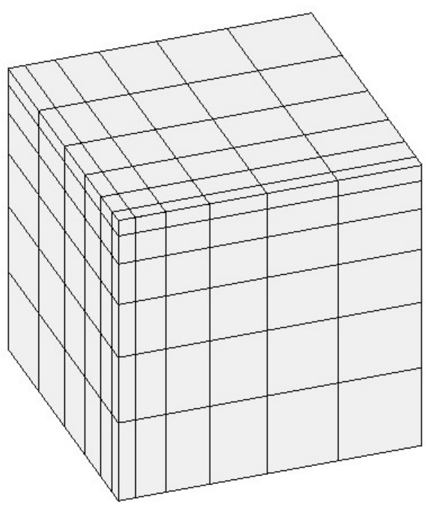


Figure 6.31 Cube foundation with variable element size

6.2.1 Part 1: Creating Cube Foundation

Part 1 consists of the following seven main steps:

1. Access block mesh generator (New)
2. Set work plane
3. Build hexahedron block
4. Edit block boundary
5. Set global boundary
6. View skeleton boundary code
7. Plot finite element mesh

Step 1: Access Block Mesh Generator (New)

Access Block Mesh Generator by selecting the following menu items in SMAP (Figure 6.2):

Run → Mesh Generator → Block Mesh → New

Step 2: Set Work Plane

Prebuilt Work Plane is displayed on drawing board along with Work Plane Editor dialog. Modify NDx and Wx in Figure 6.32 and click Update button.

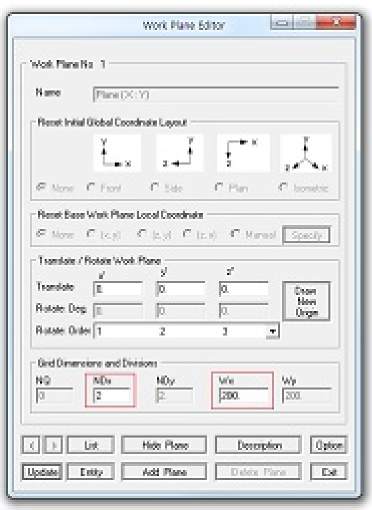


Figure 6.32 Work plane editor

Step 3: Build Hexahedron Block

Follow the same procedure as in Step 4 in the first example.
1. Click Axis toolbar as shown in Figure 6.9.
2. Click Block Editor toolbar in Figure 6.12.
3. Select Hexa for block type and click OK in Figure 6.13.
4. Click Draw Index Number in Figure 6.14.
5. Coordinates on Work Plane dialog is displayed as in Figure 6.15.

Index Numbers on Front Surface

6. Translate work plane as in Figure 6.33 and click Update button.
7. Click the points for index numbers on front surface as in Fig. 6.34.

Index Numbers on Back Surface

8. Translate work plane as in Figure 6.35 and click Update button.
9. Click the points for index numbers on back surface as in Figure 6.36.

Now, the geometry of hexahedron block is completed.
10. Click Finish in Figure 6.20.
11. Click Finish in Figure 6.14.

12. Modify Title and Material & Element Generation Parameters in Block Editor dialog as shown in Figure 6.37.

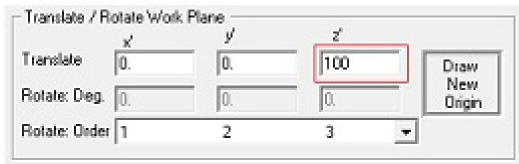


Figure 6.33 Work plane translation (z' = 100)

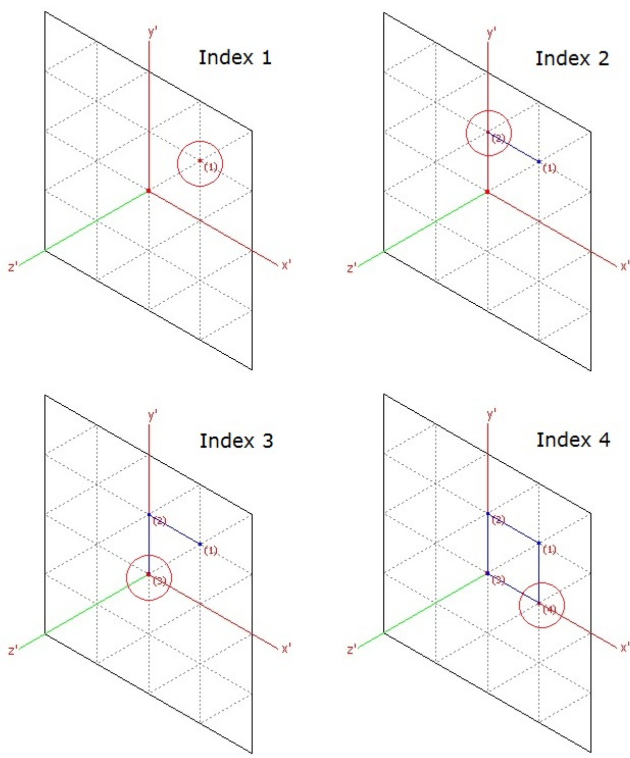


Figure 6.34 Index numbers on front surface

Figure 6.35 Work plane translation (z' = 0)

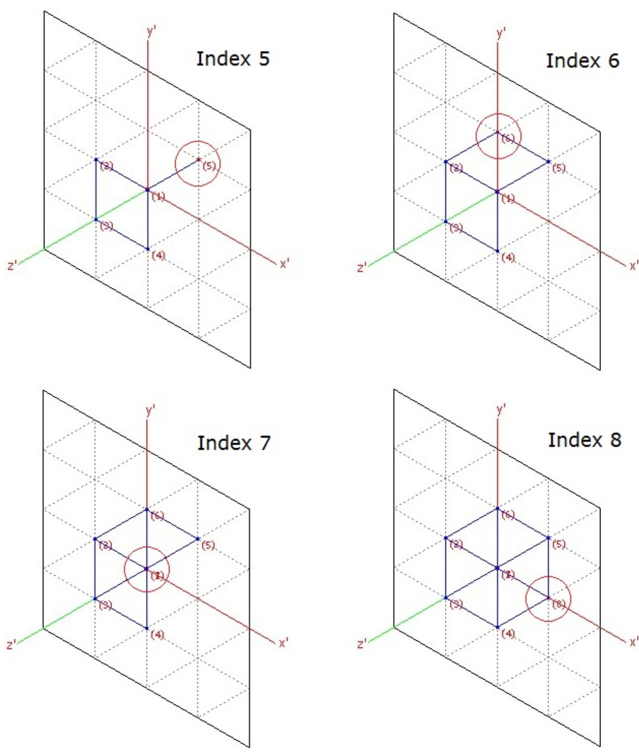


Figure 6.36 Index numbers on back surface

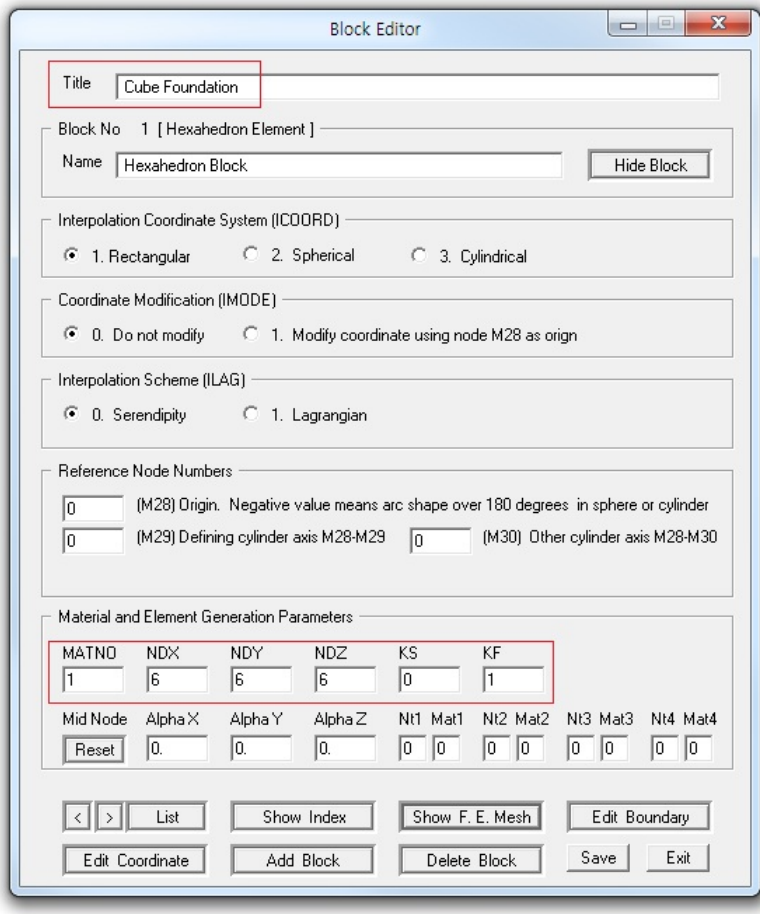


Figure 6.37 Block editor

Step 4: Edit Block Boundary Code

1. Click Edit Boundary in Figure 6.37.
2. Set the boundary codes as shown in Figure 6.38.
3. Click IBTYPE button to see description of boundary type in Fig. 6.39.
4. Click Update and then OK buttons in Figure 6.38.

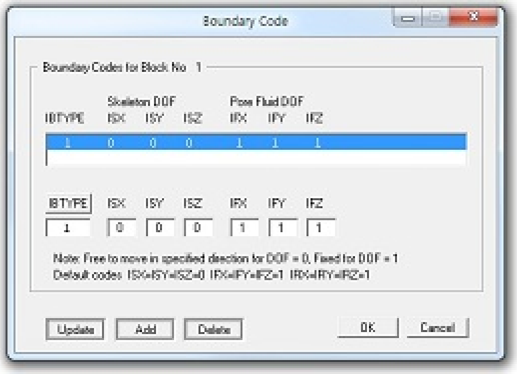


Figure 6.38 Boundary code editor

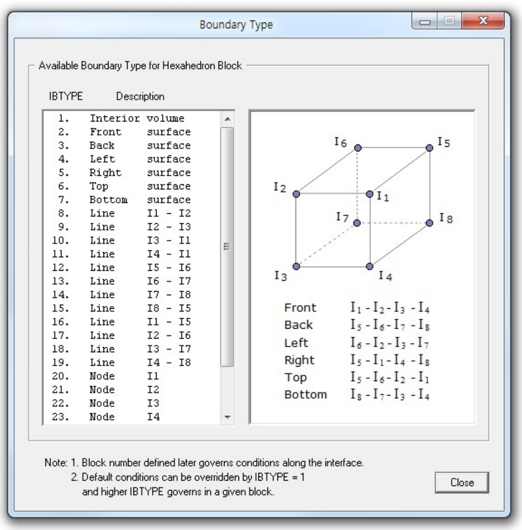


Figure 6.39 Boundary type for hexa block

Step 5: Set Global Boundary Code

1. Select Model → Edit Global Boundary in Figure 6.40.

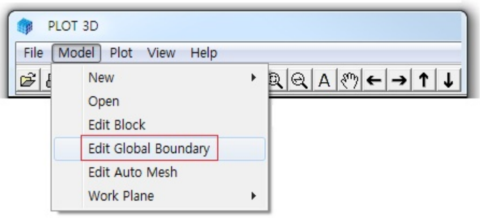


Figure 6.40 Edit global boundary menu

2. Set the boundary codes as shown in Figure 6.41.
3. Select Yes override block boundary.
4. Click Save and type in file name as EX2.

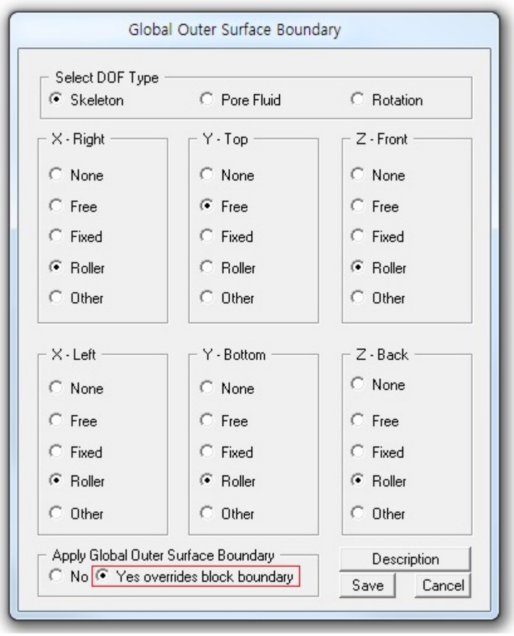


Figure 6.41 Global outer surface boundary

Step 6: View Skeleton Boundary Code

1. Select View → General in PLOT-3D menu.
2. Select Skeleton Boundary Code and click OK in Figure 6.25.
3. Click Show Numbers toolbar as shown in Figure 6.26.
4. Skeleton boundary codes are shown in Figure 6.42.

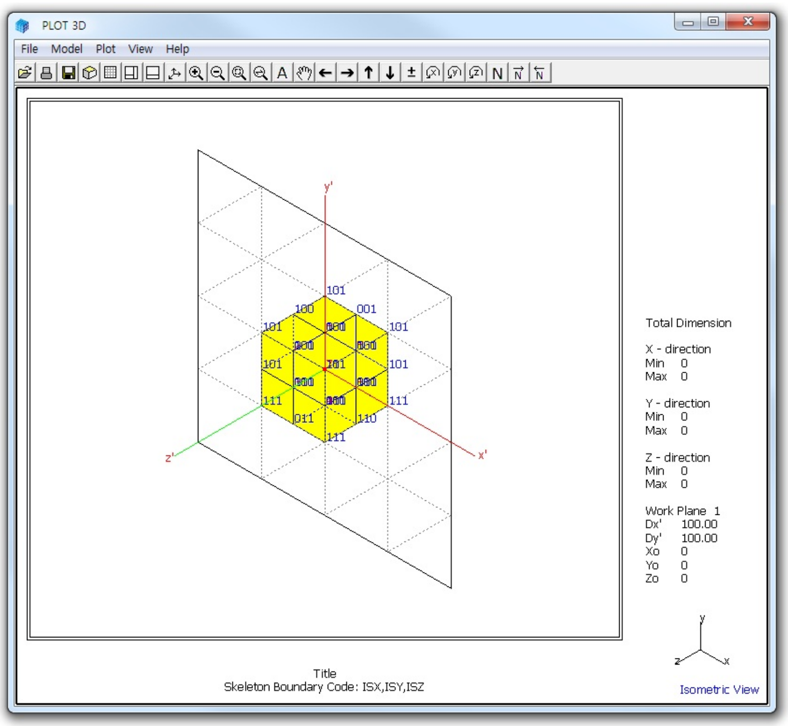


Figure 6.42 Skeleton boundary codes on drawing board

Step 7: Plot Finite Element Mesh

1. Click Show F. E. Mesh in Figure 6.37.
2. Rotate the finite element mesh as shown in Figure 6.43.

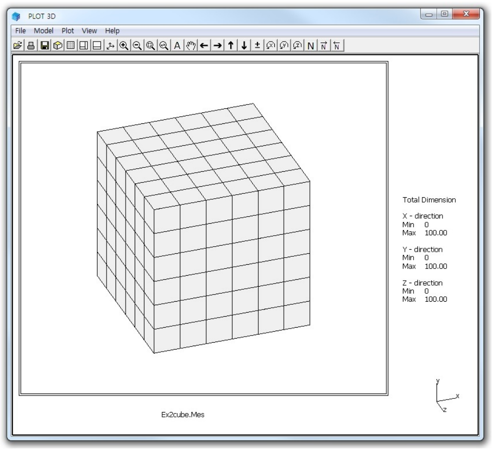


Figure 6.43 Finite element mesh

3. Follow same procedure to plot skeleton boundary codes in Step 6.
4. Figure 6.44 shows skeleton boundary codes for finite element mesh.

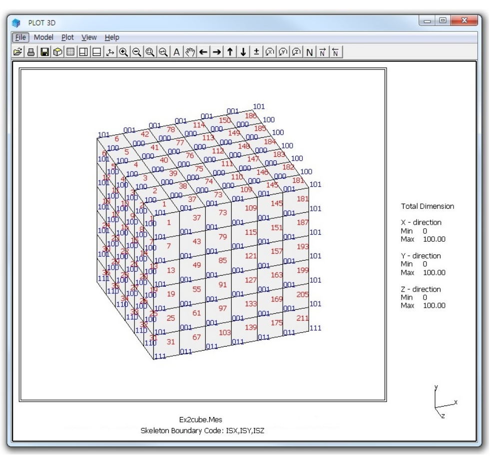


Figure 6.44 Skeleton boundary codes

6.2.2 Part 2: Modifying Cube Foundation

Part 2 consists of the following three main steps:

1. Access block mesh generator (Open)
2. Modify element generation parameters
3. Plot finite element mesh

Step 8: Access Block Mesh Generator (Open)

1. Access Block Mesh Generator by selecting the following menu items in SMAP (Figure 6.2):
   Run → Mesh Generator → Block Mesh → Open

2. Click Browse button in Open Input File dialog in Figure 6.45.
3. Select the input file EX2.Meb generated in Part 1 .

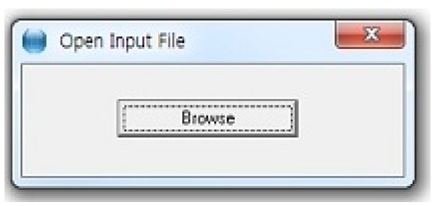


Figure 6.45 Open input file

Step 9: Modify Element Generation Parameters

1. Click Block Editor toolbar in Figure 6.12.
2. Modify Alpha X, Alpha Y, Alpha Z as in Figure 6.46.
3. Click Reset.
4. Click Save.

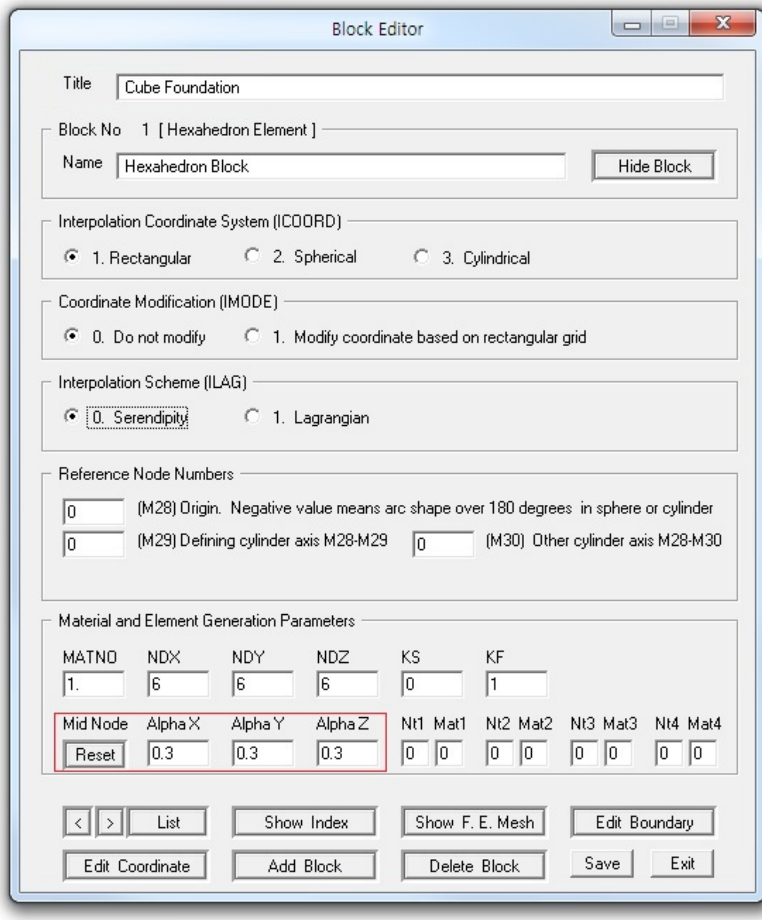


Figure 6.46 Block editor

5. Click Work Plane toolbar and then click Show Plane button.
6. Click Axis toolbar in Figure 6.9.
7. Block mesh is shown in Figure 6.47.

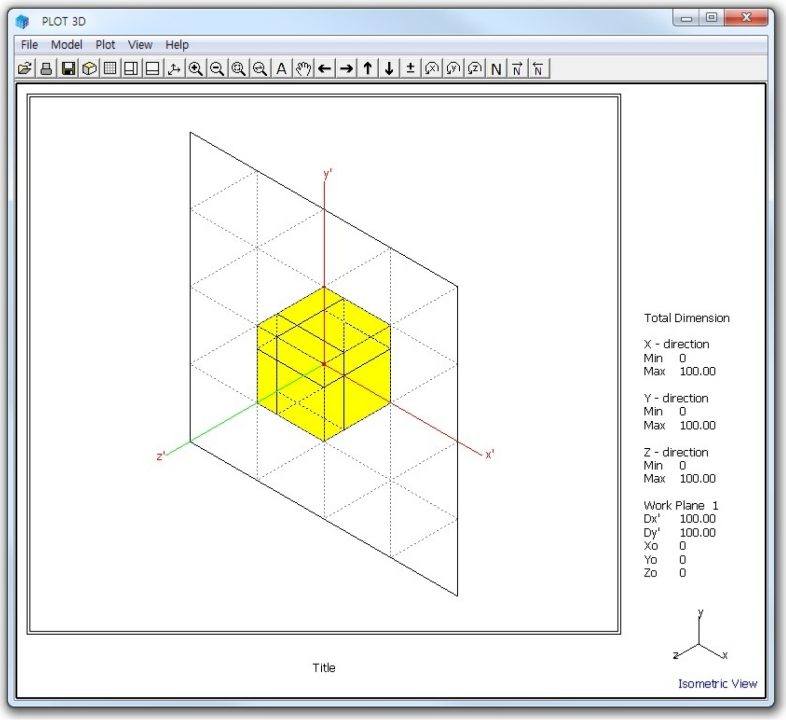


Figure 6.47 Block mesh on drawing board

Step 7: Plot Finite Element Mesh

1. Click Show F. E. Mesh in Figure 6.46.
2. Rotate the finite element mesh as shown in Figure 6.48.

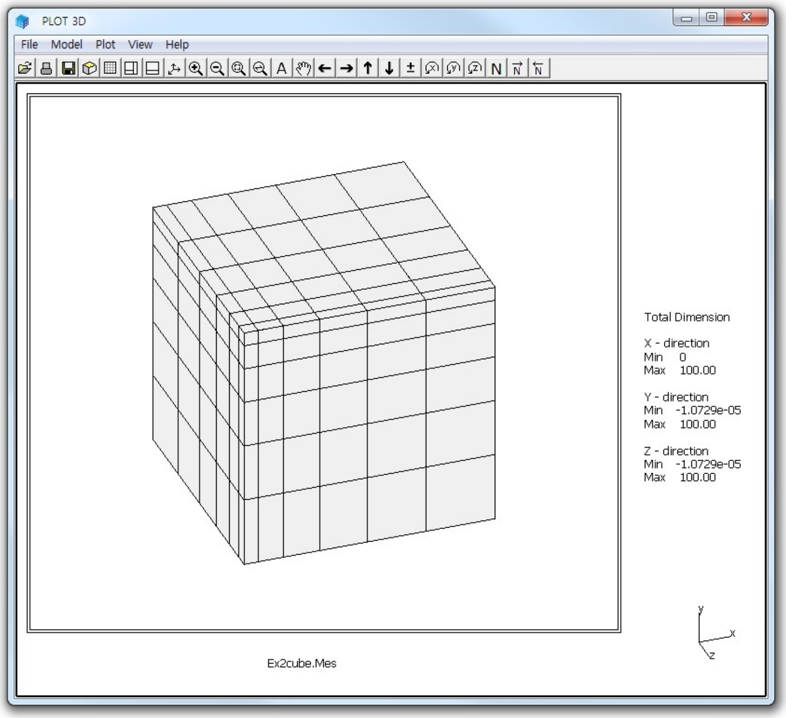


Figure 6.48 Finite element mesh

6.3 Hemispherical Shell

This example illustrates how to build block mesh for hemispherical shell subjected to concentrated loads as schematically shown in Figure 6.49.

This example involves following seven main steps:

1. Access block mesh generator
2. Set work plane
3. Build arc entities
4. Build quad block
5. Edit block boundary code
6. View skeleton and rotation boundary codes
7. Plot finite element mesh

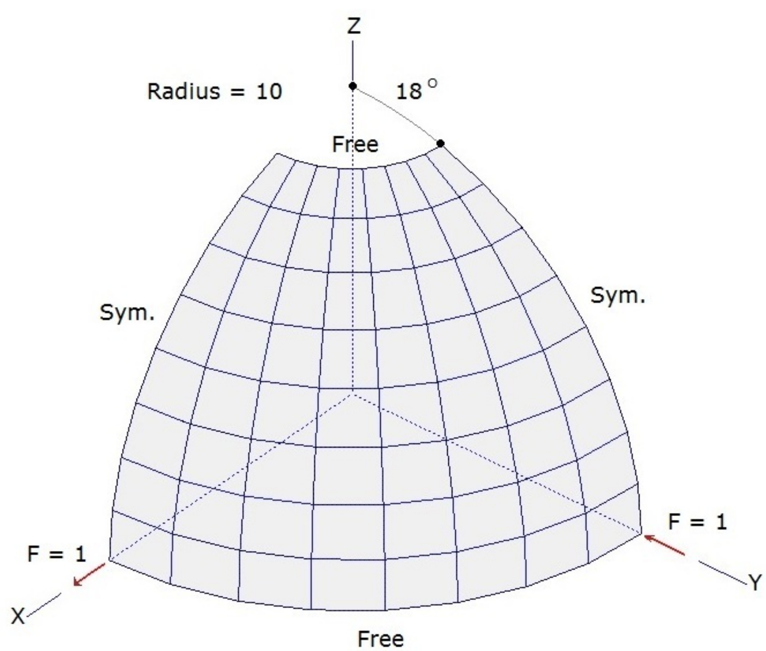


Figure 6.49 Hemispherical shell subjected to concentrated loads

Step 1: Access Block Mesh Generator (New)

Access Block Mesh Generator by selecting the following menu items in SMAP (Figure 6.2):
Run → Mesh Generator → Block Mesh → New

Step 2: Set Work Plane

1. Select Work Plane No 4 and set parameters for Grid Dimension and Division as shown in Figure 6.50.

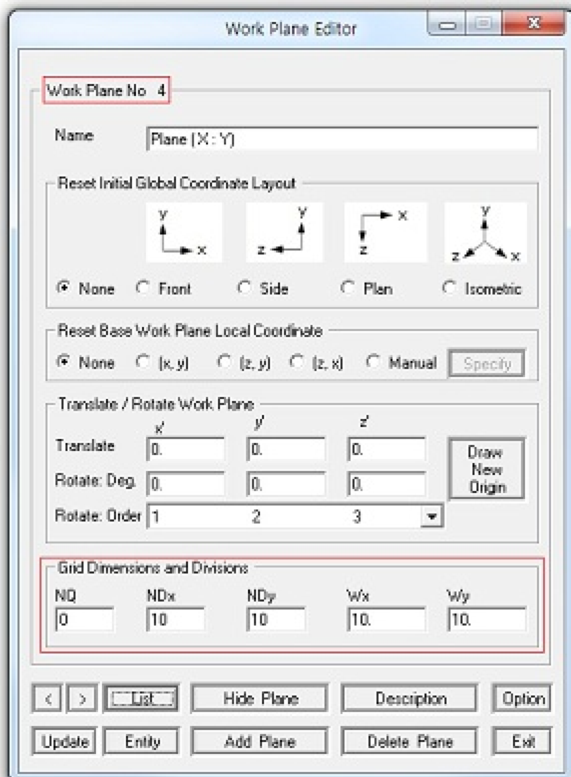


Figure 6.50 Work plane editor

2. Select Model → Work Plane → Isometric Z-axis in Figure 6.51.
3. Figure 6.52 shows work plane with Isometric Z-axis.

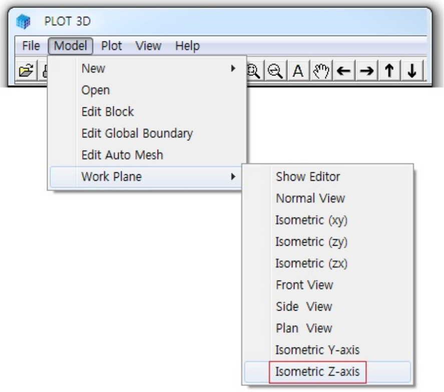


Figure 6.51 Isometric Z-axis menu

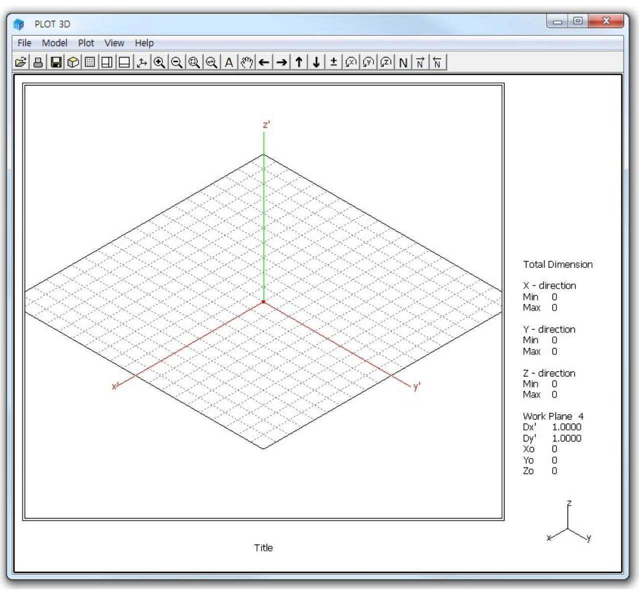


Figure 6.52 Work plane with Isometric Z-axis

Step 3: Build Arc Entities

Click Entity in Figure 6.50.

Arc Entity on YZ plane

1. Click Add in Entity Editor dialog in Figure 6.53.

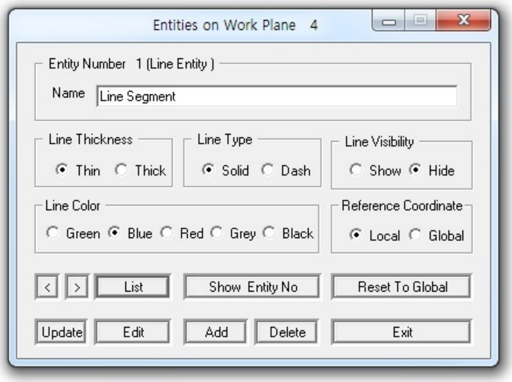


Figure 6.53 Entity editor

2. Select Arc in Entity Type Selection dialog in Figure 6.54.
3. Click OK.

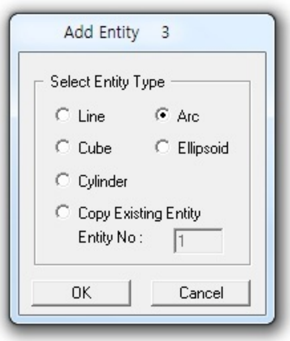


Figure 6.54 Entity type selection

4. Type in dimensions of arc entity as shown in Figure 6.55.
5. Click Draw Arc Entity.

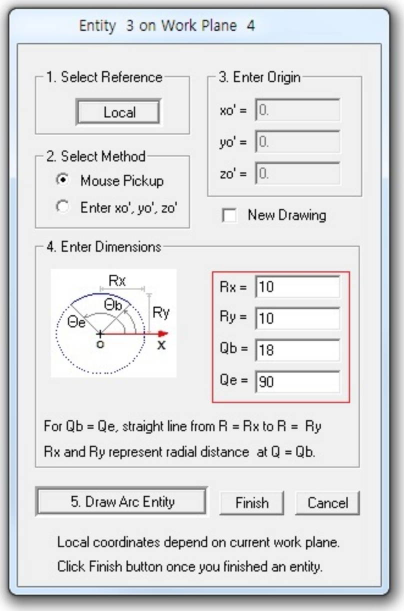


Figure 6.55 Arc entity

6. Figure 6.56 shows Coordinates on Work Plane dialog.

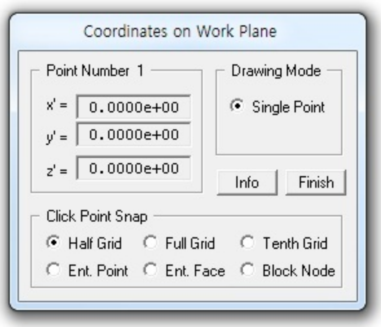


Figure 6.56 Coordinates on work plane

7. Rotate work plane as in Figure 6.57 and click Update button.

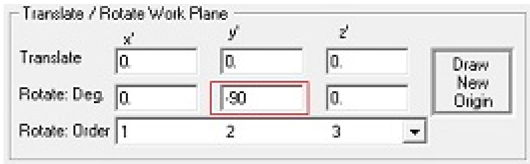


Figure 6.57 Work plane rotation

8. Click Mouse at the origin of coordinates as shown in Figure 6.58.

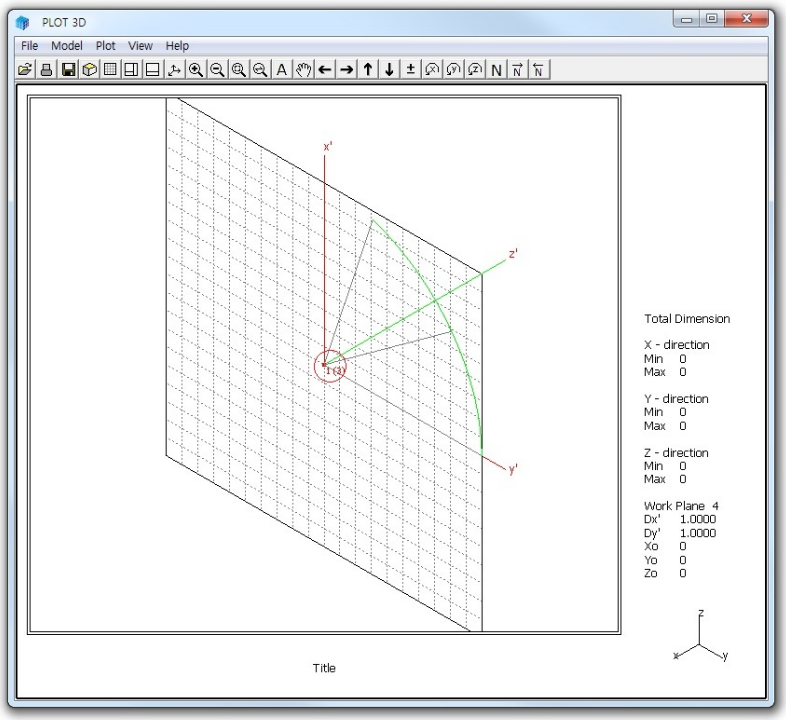


Figure 6.58 Arc entity on YZ plane

9. Click Finish in Figure 6.56.
10. Click Finish in Figure 6.55.
11. Click Global for Reference Coordinate in Figure 6.59.
12. Click Reset To Global.

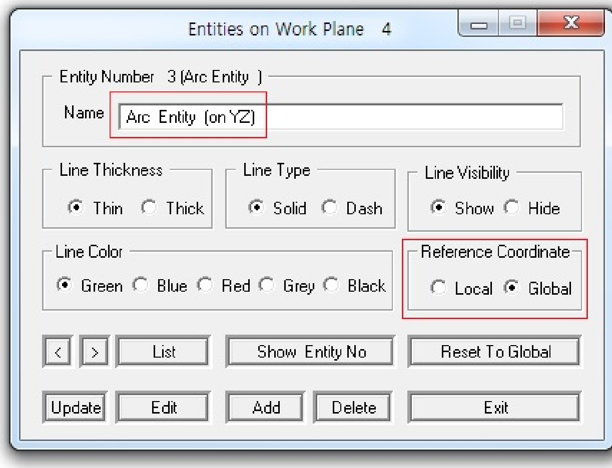


Figure 6.59 Entity editor

Arc Entity on XZ plane

Follow the same procedure as for Arc Entity on YZ plane.

1. Click Add in Entity Editor dialog in Figure 6.59.
2. Select Arc in Entity Type Selection dialog in Figure 6.54.
3. Click OK.

4. Type in dimensions of arc entity as shown in Figure 6.60.
5. Click Draw Arc Entity.
6. Coordinates on Work Plane dialog in Figure 6.56 is shown.

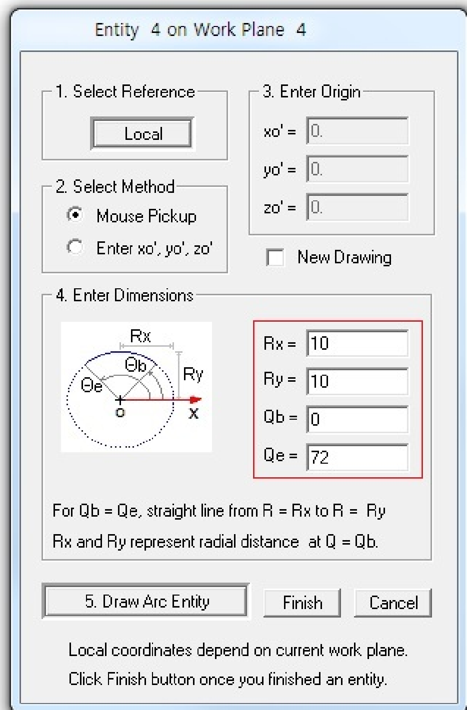


Figure 6.60 Arc entity

7. Rotate work plane as in Figure 6.61 and click Update button.

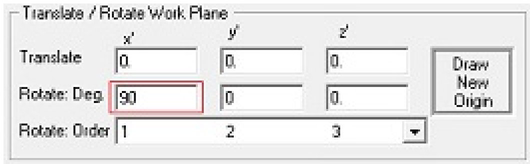


Figure 6.61 Work plane rotation

8. Click Mouse at the origin of coordinates as shown in Figure 6.62.

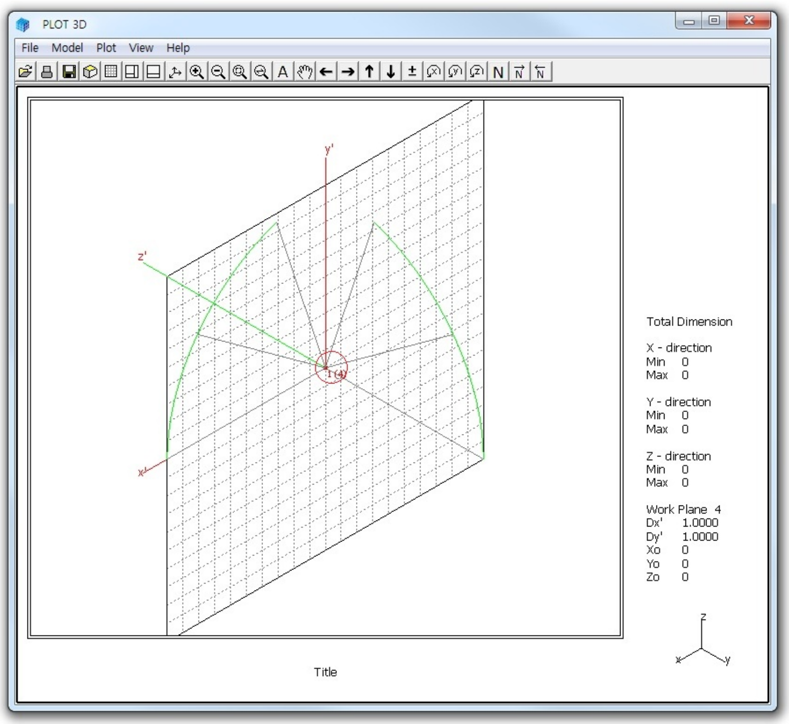


Figure 6.62 Arc entity on XZ plane

9. Click Finish in Figure 6.56.
10. Click Finish in Figure 6.60.
11. Click Global for Reference Coordinate in Figure 6.63.
12. Click Reset To Global and then Exit buttons in Figure 6.63.

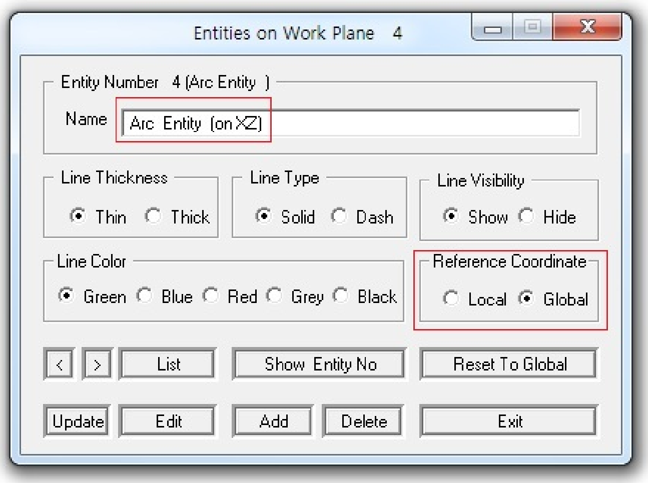


Figure 6.63 Entity editor

Step 4: Build Quad Block

1. Click Block Editor toolbar in Figure 6.64.

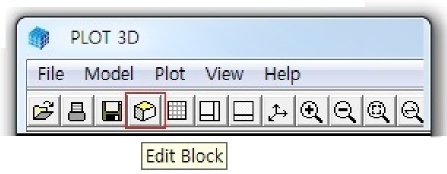


Figure 6.64 Block editor toolbar

2. Select Quad for block type in Figure 6.65.
3. Select Spherical for Interpolation Coordinate System.
4. Select Origin for Reference Point.
5. Click OK.

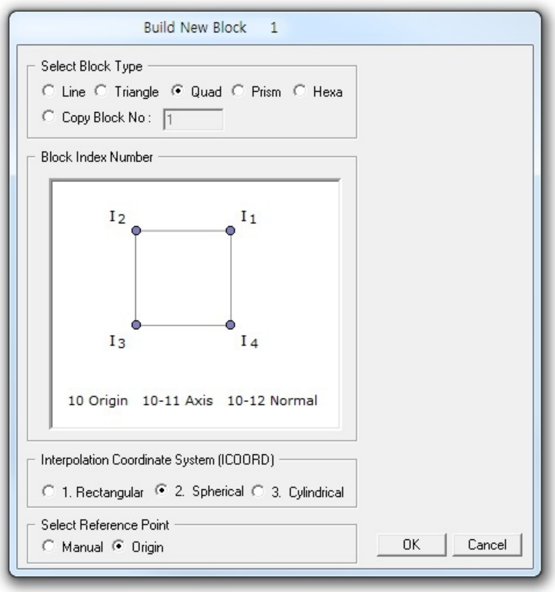


Figure 6.65 Block type selection

6. Click Draw Index Number in Figure 6.66.
7. Select Ent. Point for Click Point Snap as in Figure 6.67.

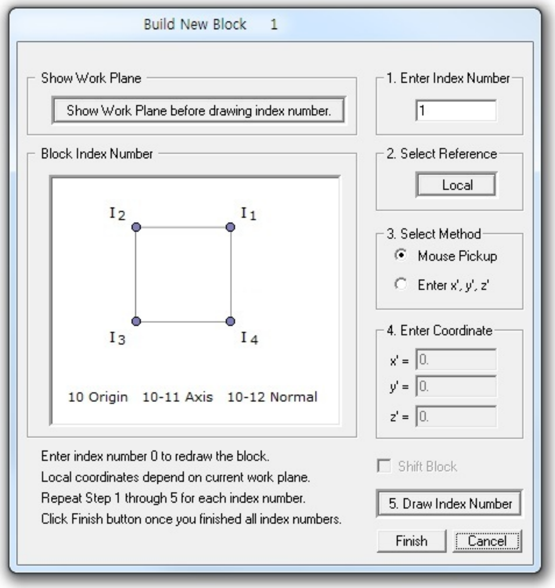


Figure 6.66 Quad block

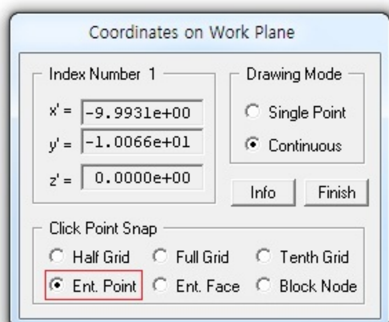


Figure 6.67 Coordinates on work plane

8. Rotate work plane as in Figure 6.68 and click Update button.

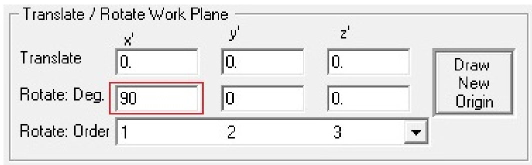


Figure 6.68 Work plane rotation

9. Click the points for index numbers 1 and 2 as in Figure 6.69.

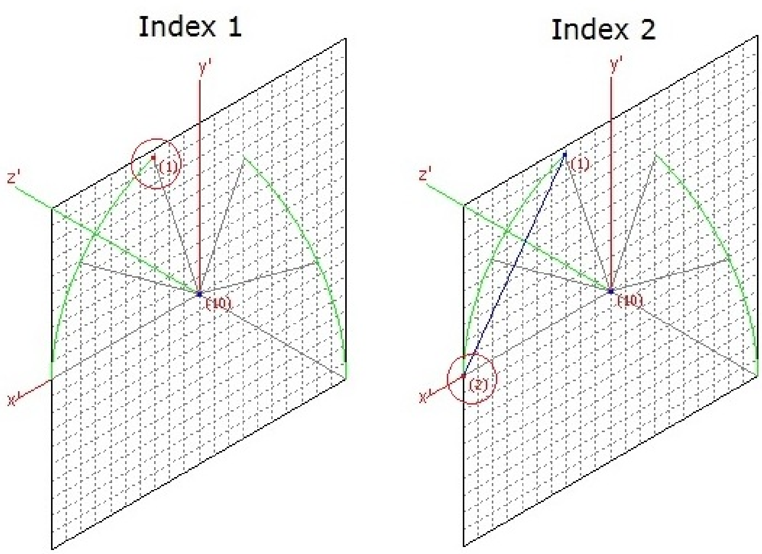


Figure 6.69 Index numbers 1 and 2 on XZ plane

10. Rotate work plane as in Figure 6.70 and click Update button.

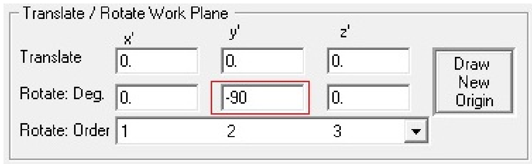


Figure 6.70 Work plane rotation

11. Click the points for index numbers 3 and 4 as in Figure 6.71.

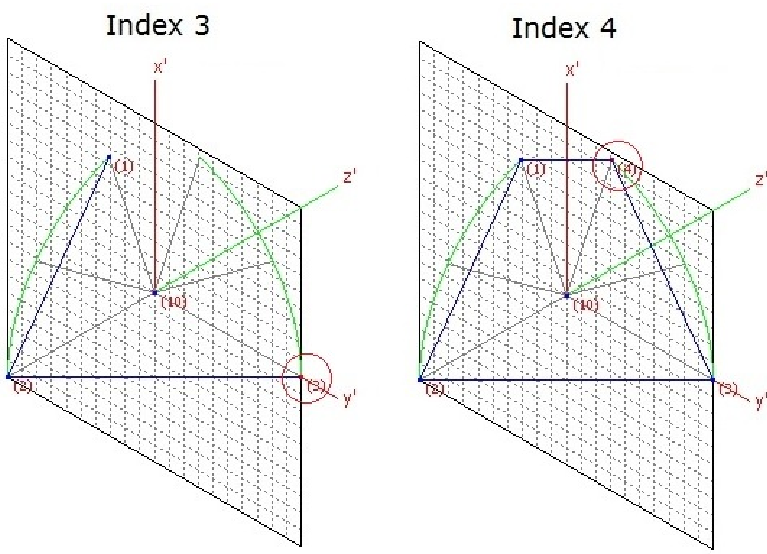


Figure 6.71 Index numbers 3 and 4 on YZ plane

Now, the geometry of quad block is completed.

12. Click Finish in Figure 6.67 and then click Finish in Figure 6.66.
13. Figure 6.72 shows completed quad block with index numbers.

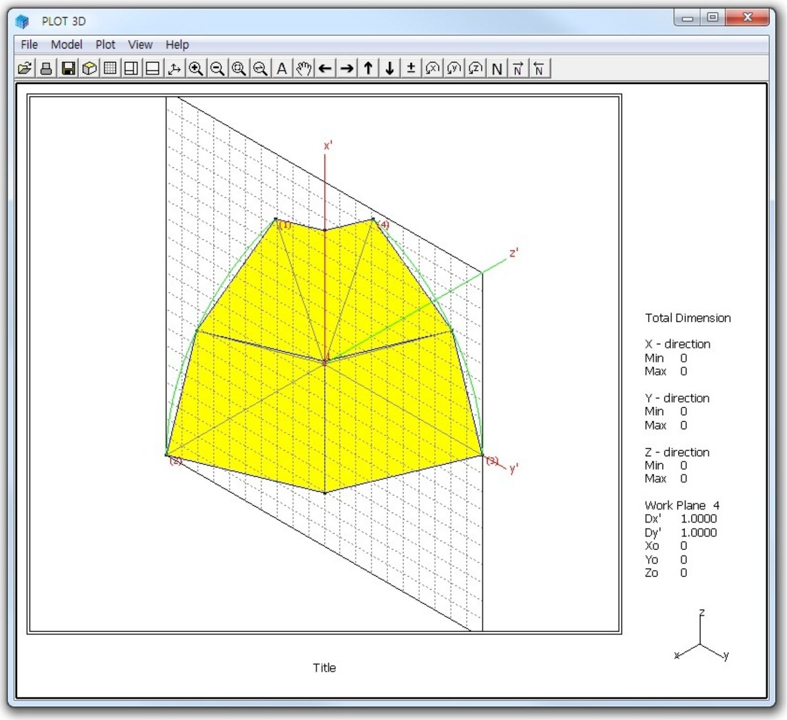


Figure 6.72 Completed quad block

14. Modify Title and Material & Element Generation Parameters in Block Editor dialog as shown in Figure 6.73.

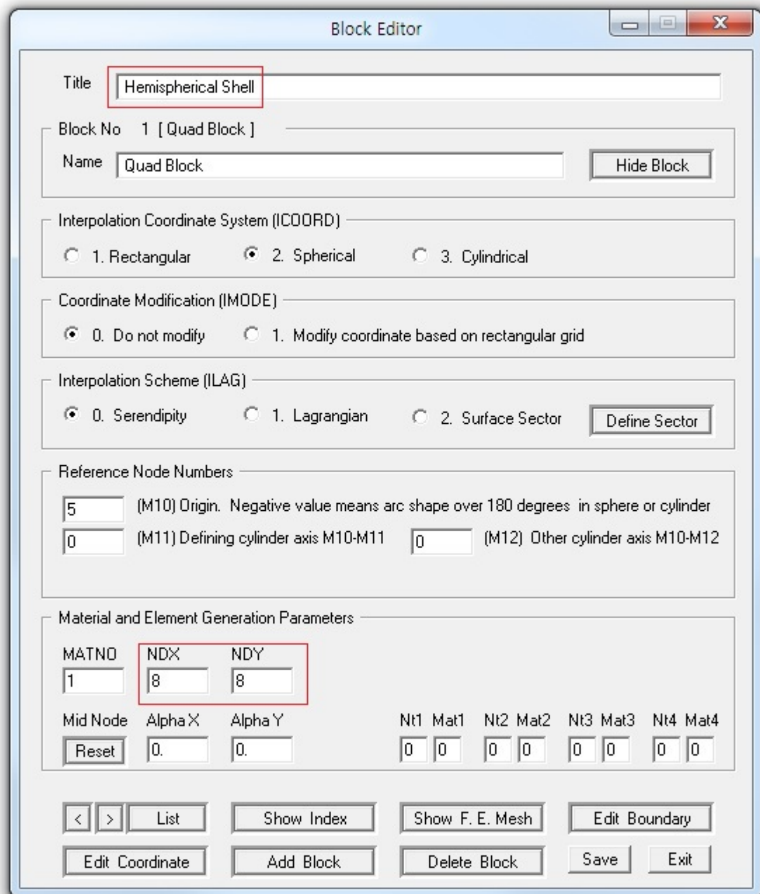


Figure 6.73 Block editor

Step 5: Edit Block Boundary Code

1. Click Edit Boundary in Figure 6.73.
2. Set the boundary codes as shown in Figure 6.74.
3. Click IBTYPE button to see description of boundary type in Fig. 6.75.
4. Click Update and then OK buttons.
5. Click Save in Figure 6.73 and type in file name as EX3.

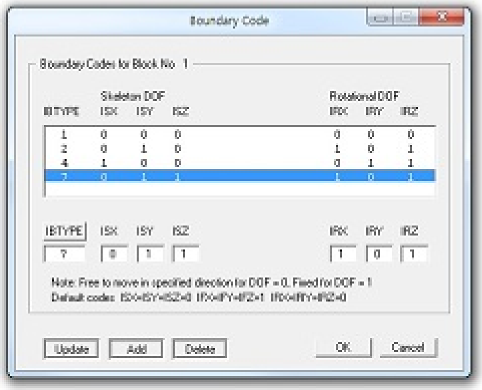


Figure 6.74 Boundary code editor

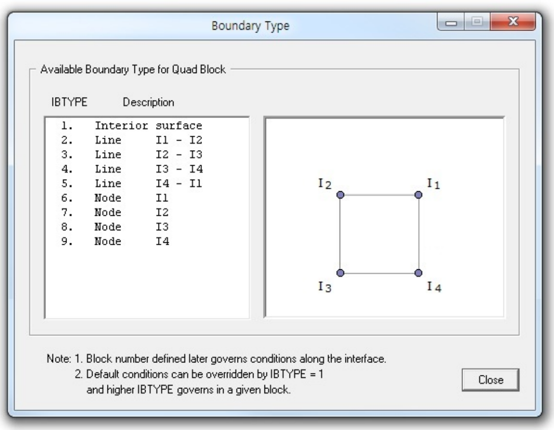


Figure 6.75 Boundary type for quad block

Step 6: View Skeleton and Rotation Boundary Code

1. Select View → General in PLOT-3D menu.
2. Select Skeleton Boundary Code and click OK in Figure 6.25.
3. Click Show Numbers toolbar as shown in Figure 6.26.
4. Skeleton boundary codes are shown in Figure 6.76.

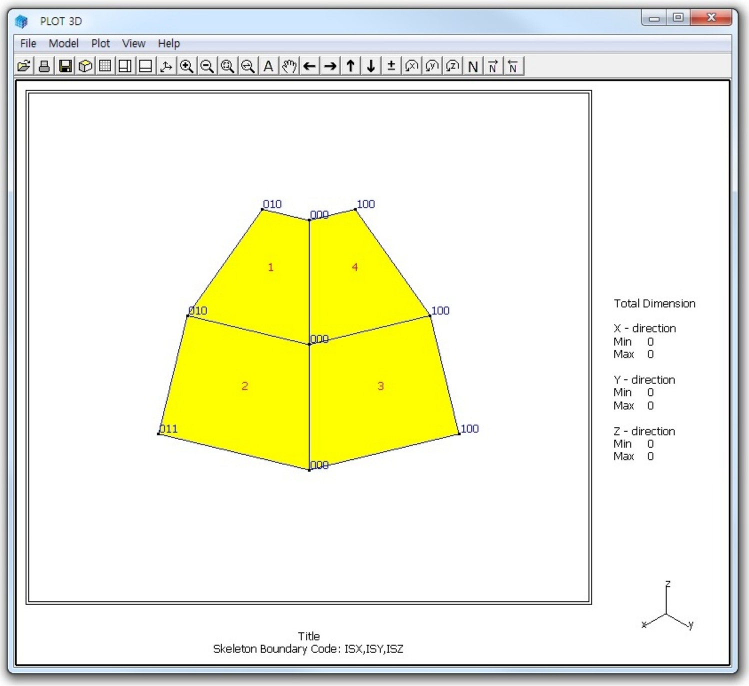


Figure 6.76 Skeleton boundary codes

5. Click Next Numbers toolbar twice in Figure 6.77.

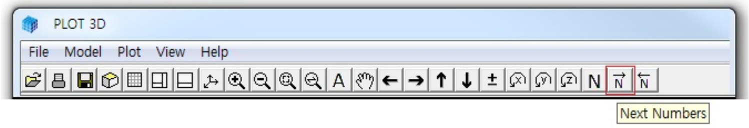


Figure 6.77 Next numbers toolbar

6. Rotation boundary codes are shown in Figure 6.78.

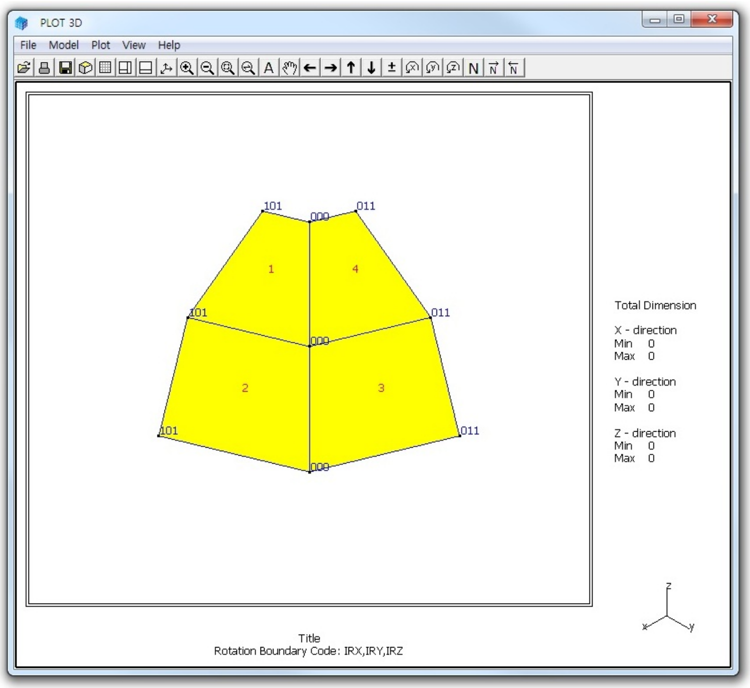


Figure 6.78 Rotation boundary codes

Step 7: Plot Finite Element Mesh

1. Click Show F. E. Mesh in Figure 6.73.
2. Select Model → Work Plane → Isometric Z-axis in Figure 6.51.
3. Finite element meshes are shown in Figure 6.79.

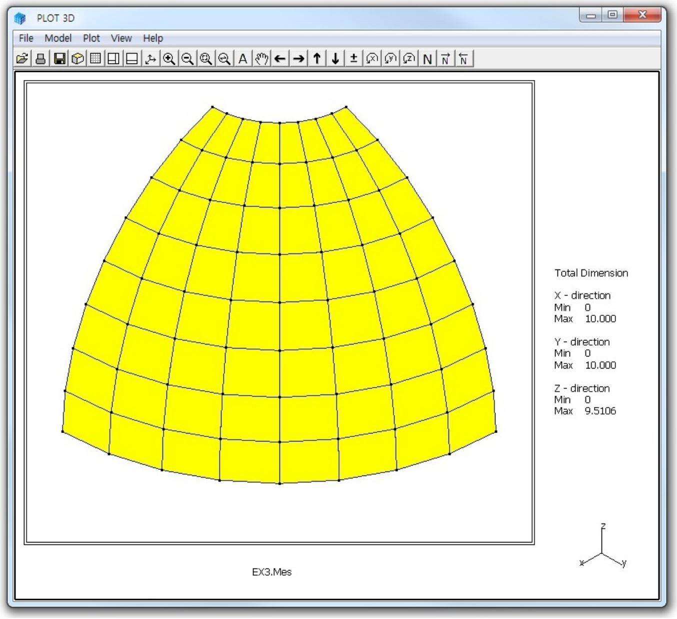


Figure 6.79 Finite element mesh

4. Follow the same procedure to plot boundary codes as in Step 6.
5. Skeleton and rotation boundary codes are shown in Figures 6.80 and 6.81, respectively.

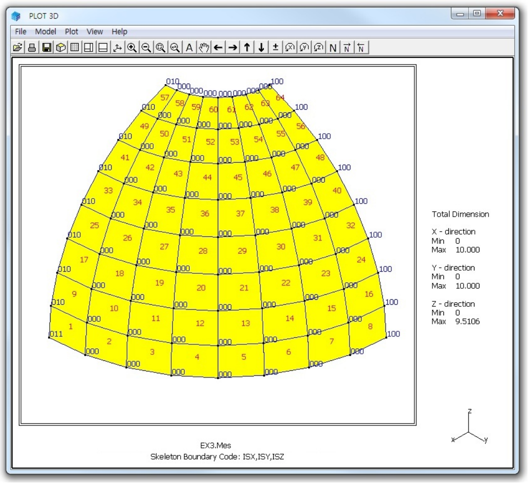


Figure 6.80 Skeleton boundary codes

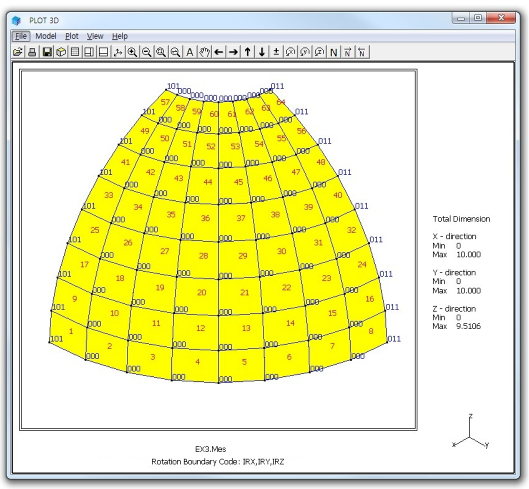


Figure 6.81 Rotation boundary codes

6.4 Horseshoe Tunnel

This example illustrates how to build block mesh for horseshoe tunnel with reinforced concrete lining as schematically shown in Figure 6.82.

This example involves following eight main steps:
1. Access block mesh generator
2. Set work plane
3. Build entities
4. Add work plane
5. Build blocks
6. Set global boundary
7. View selected material
8. Plot finite element mesh

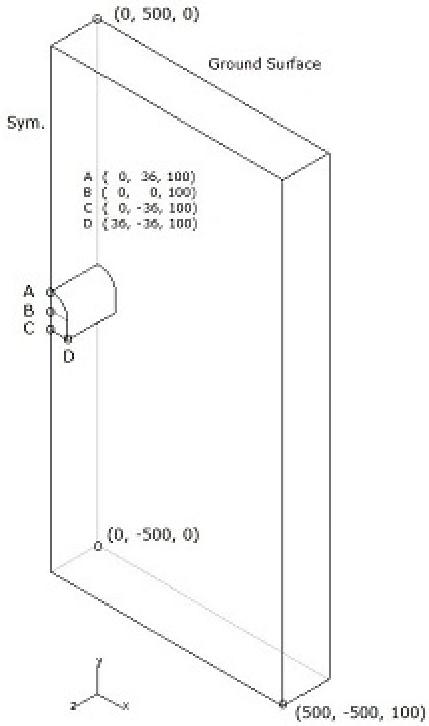


Figure 6.82 Schematic horseshoe tunnel section

Step 1: Access Block Mesh Generator (New)

Access Block Mesh Generator by selecting the following menu items in SMAP (Figure 6.2):

Run → Mesh Generator → Block Mesh → New

Step 2: Set Work Plane

1. Select Work Plane No 4 as shown in Figure 6.83.
2. Select Isometric for Reset Initial Global Coordinate Layout.
3. Set parameters for Grid Dimensions and Divisions.
4. Click Description to see layout of NQ = 8 in Figure 6.84.
5. Click Update.
6. Figure 6.85 shows isometric view of work plane.

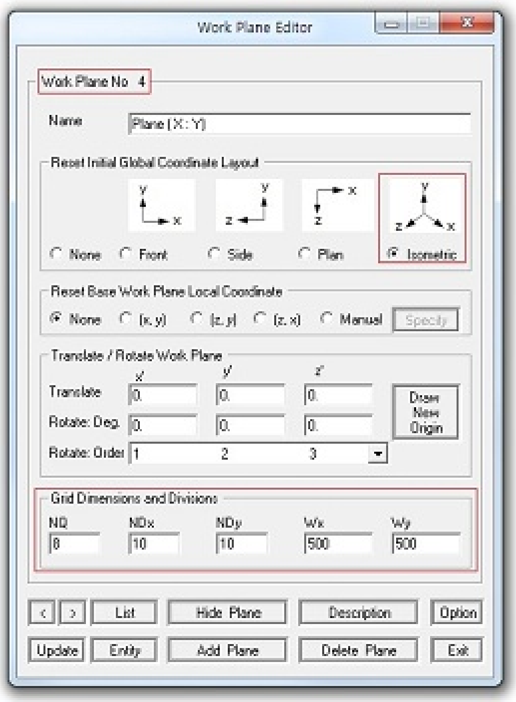


Figure 6.83 Work plane editor

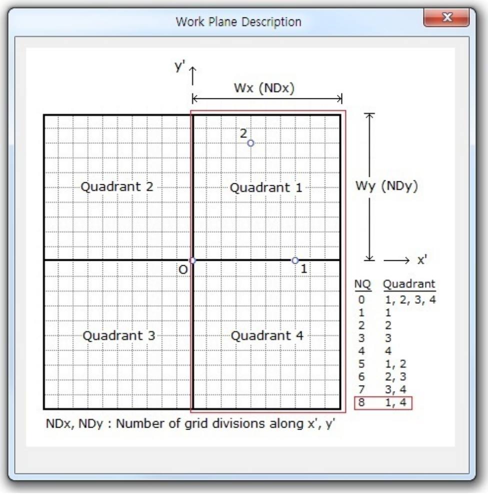


Figure 6.84 Work plane description (NQ=8)

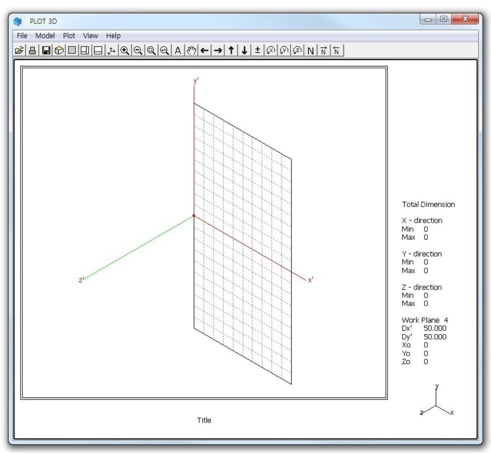


Figure 6.85 Work plane on drawing board

Step 3: Build Entities

Following five entities are used to make it easier to build blocks

- Cylinder entity for Upper Core
- Cube entity for Lower Core
- Cylinder entity for Around Upper Core
- Cube entity for Around Lower Core
- Cube entity for Outer Boundary

Upper Core by Cylinder Entity

1. Click Entity in Figure 6.83.
2. Click Add in Entity Editor dialog in Figure 6.88.
3. Click Cylinder in Figure 6.86 and click OK.
4. Set the geometric parameters as in Figure 6.87.
5. Click Draw Cylinder Entity and then click Finish.
6. Set option parameters as in Figure 6.88 and click Reset To Global.
7. Cylinder entity for upper core is shown in Figure 6.89.

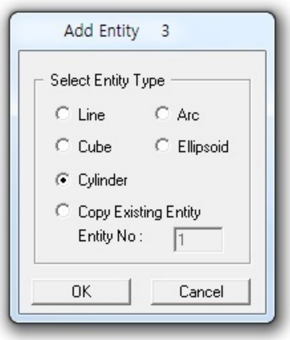


Figure 6.86 Entity type selection

Other Entities

8. Follow the same procedure as for upper core.

Lower Core:	Figures 6.90 - 6.92
Around Upper Core:	Figures 6.93 - 6.95
Around Lower Core:	Figures 6.96 - 6.98
Outer Boundary:	Figures 6.99 - 6.101

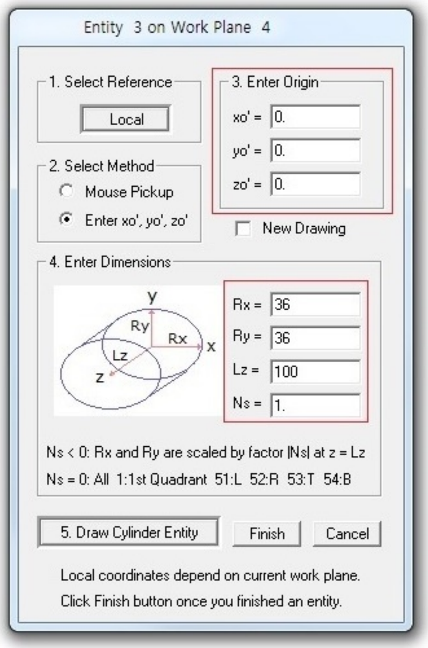


Figure 6.87 Cylinder entity for upper core

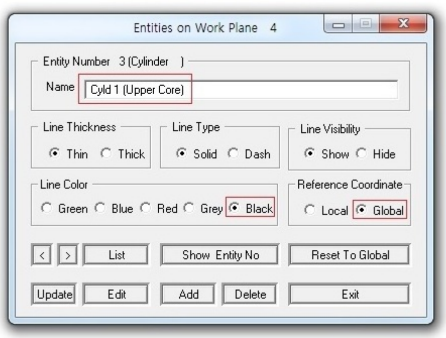


Figure 6.88 Entity editor

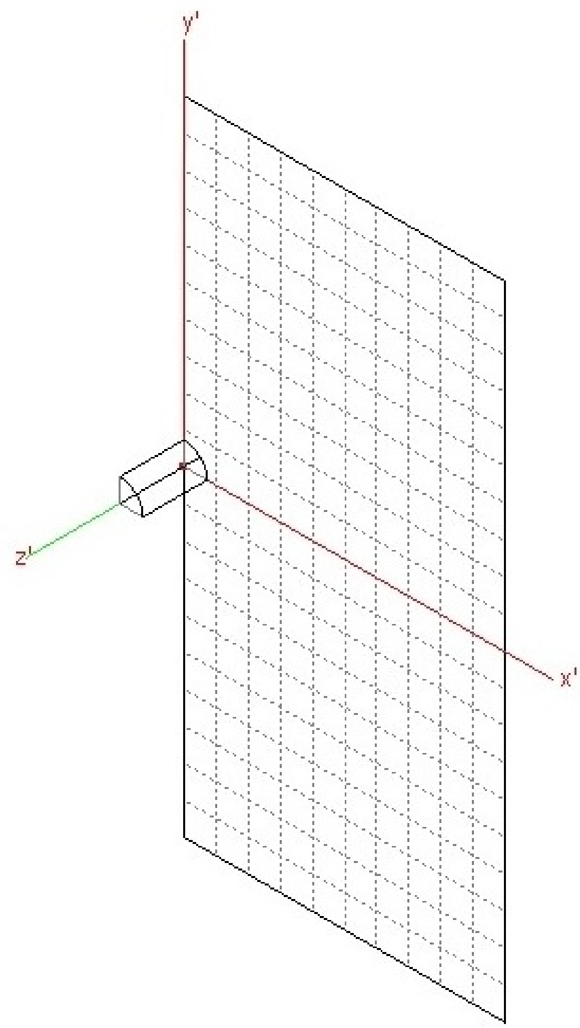


Figure 6.89 Upper core on drawing board

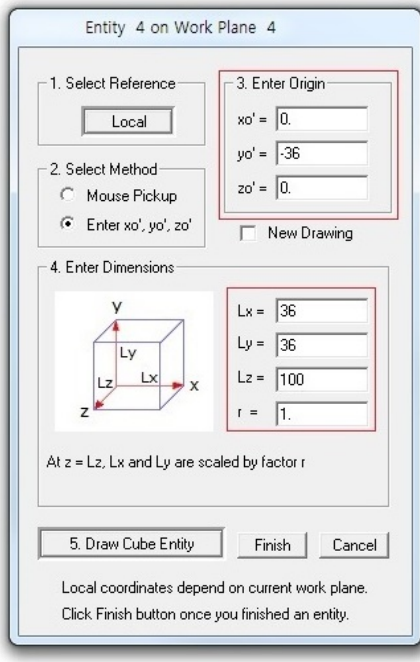


Figure 6.90 Cube entity for lower core

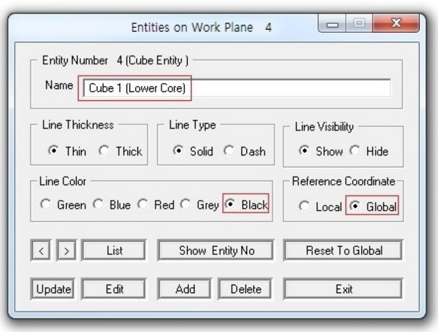


Figure 6.91 Entity editor

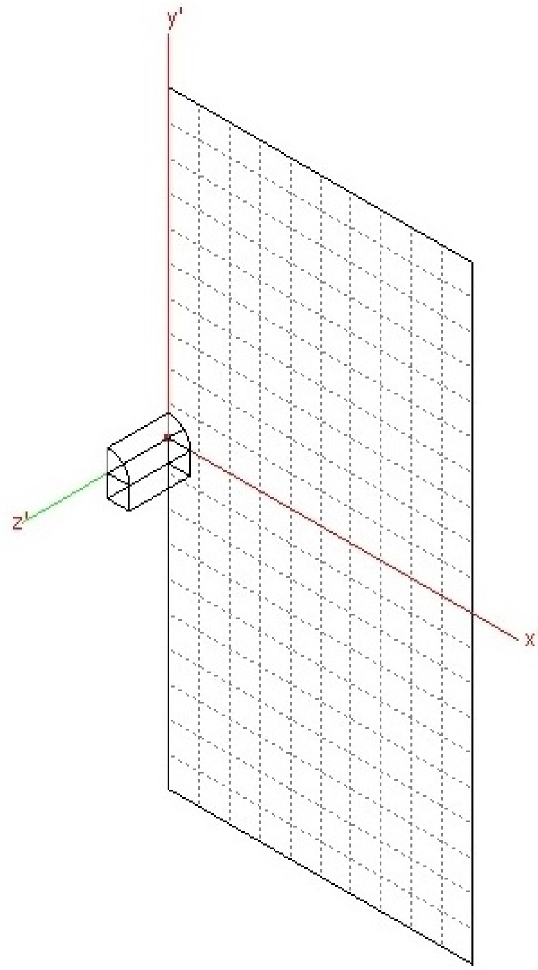


Figure 6.92 Lower core on drawing board

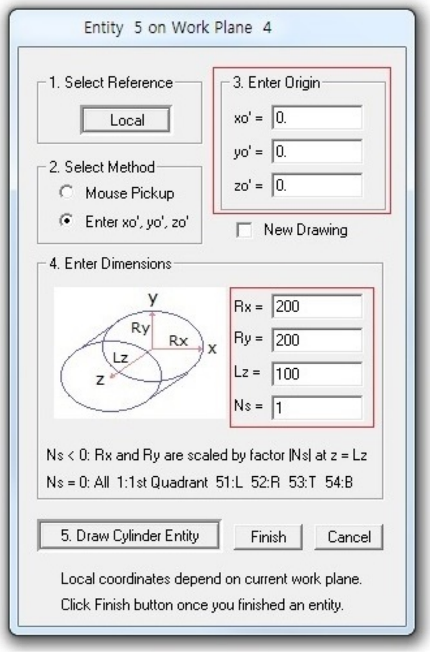


Figure 6.93 Cylinder entity for around upper core

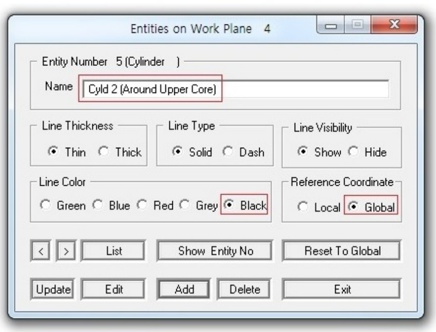


Figure 6.94 Entity editor

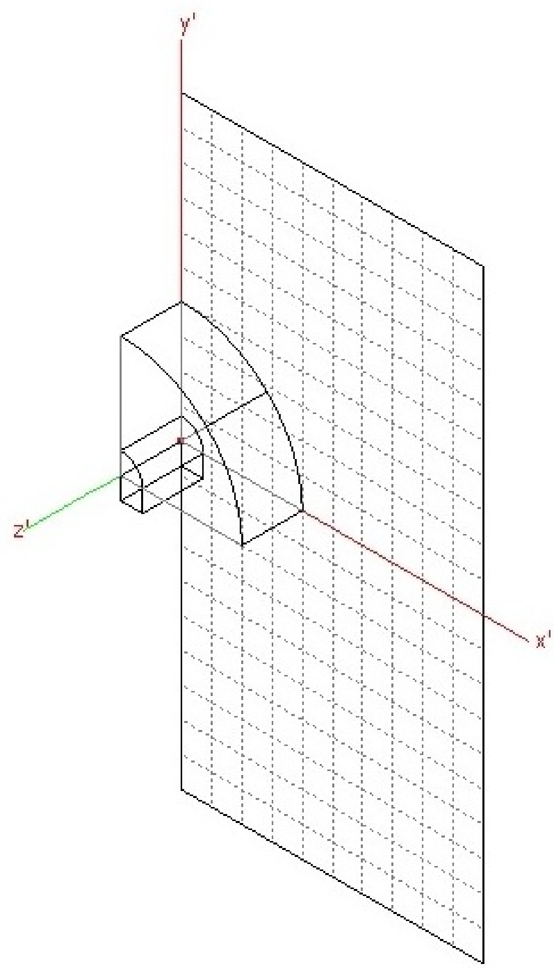


Figure 6.95 Around upper core on drawing board

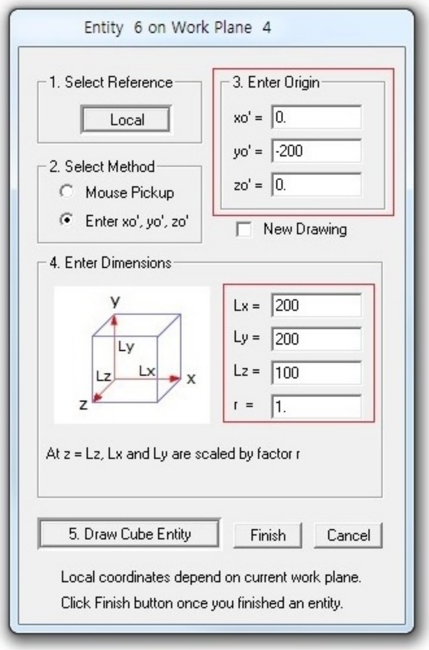


Figure 6.96 Cube entity for around lower core

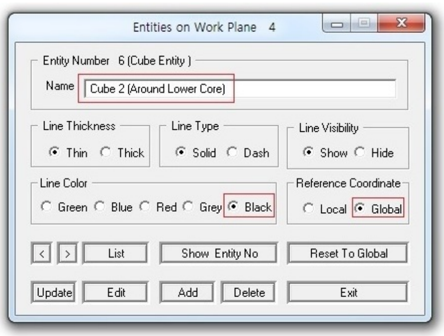


Figure 6.97 Entity editor

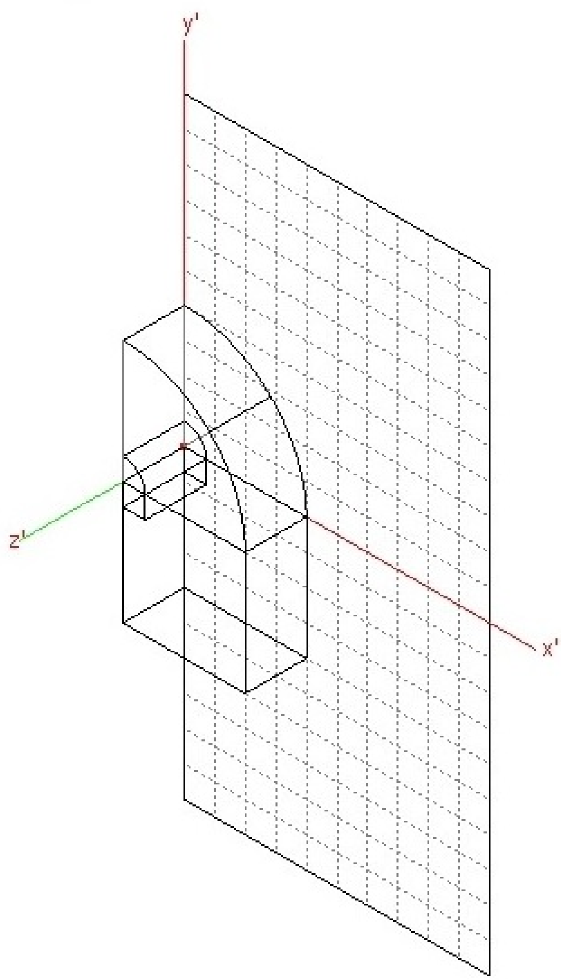


Figure 6.98 Around lower core on drawing board

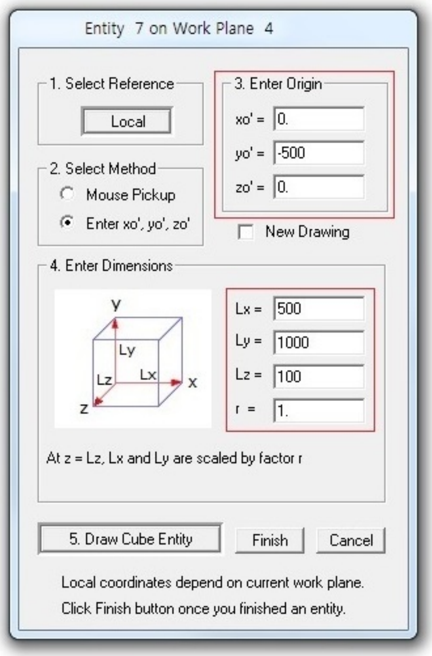


Figure 6.99 Cube entity for outer boundary

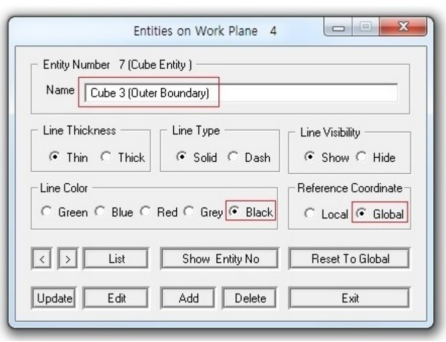


Figure 6.100 Entity editor

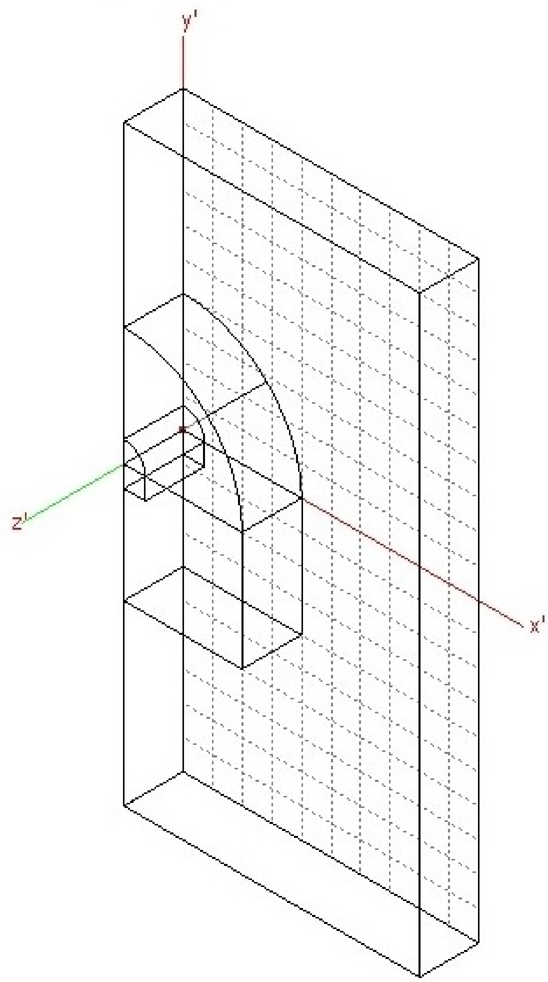


Figure 6.101 Outer boundary on drawing board

9. Available entities on work plane 4 are listed as shown in Figure 6.102 by clicking List button in the Entity Editor dialog.
10. Click OK in Figure 6.102.
11. Click Exit in Figure 6.100.

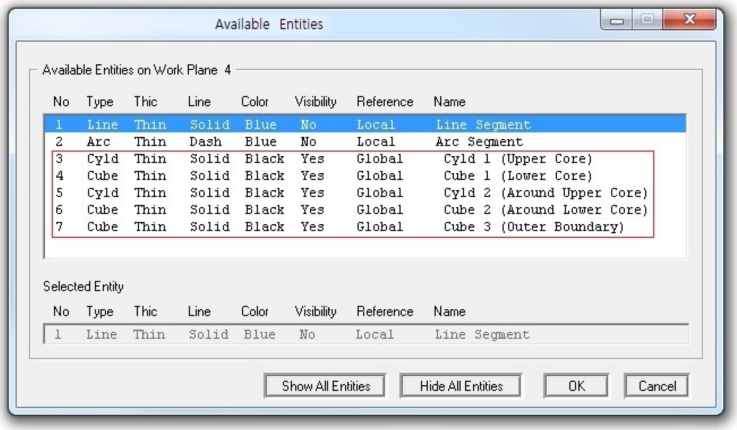


Figure 6.102 Available entities on work plane 4

Step 4: Add Work Plane

At Step 2, we set Work Plane No 4 which represents back surface.
At Step 3, we built 5 entities on this Work Plane No 4.

Here, we want to add new Work Plane No 5 in the following way:

- Copy Work Plane No 4 along with entities on it.
- Add this copied one as new Work Plane No 5.
- Modify such that it represents front surface.

Once we have this new Work Plane No 5, it will be much easier to build blocks since front and back surfaces of work planes can be accessed simply by one click of Back or Next button on Coordinates on Work Plane dialog in Figure 6.103.

Perform the following four steps:

1. Select Work Plane No 4 in Work Plane Editor dialog in Figure 6.83
2. Click Add Plane button in Figure 6.83
3. Modify Name and Translation as in Figure 6.104
4. Click Update in Figure 6.104

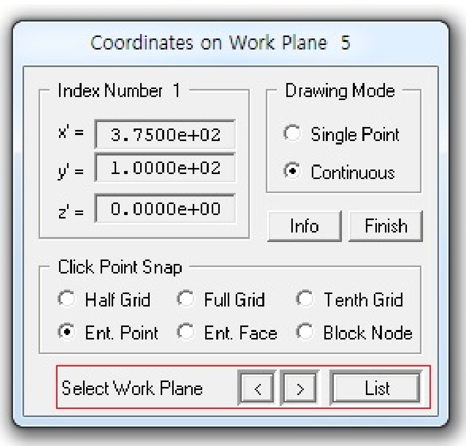


Figure 6.103 Coordinates on work plane

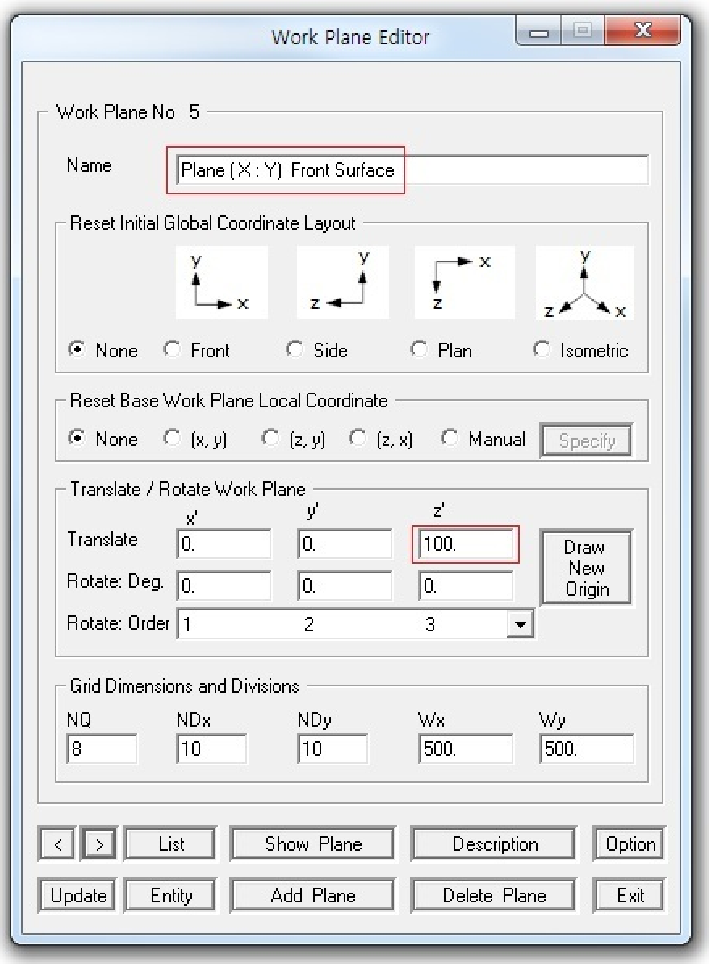


Figure 6.104 Work Plane No 5 representing front surface

Step 5: Build Blocks

Fourteen blocks are used to model the geometry of horseshoe tunnel as shown in Figures 6.105 and 6.106.

- 8 blocks for surrounding medium
- 2 blocks for tunnel core
- 4 blocks for tunnel lining as shell elements

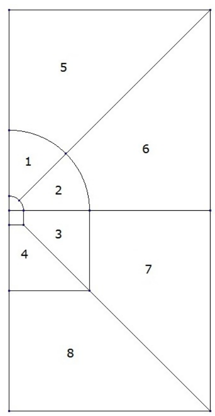


Figure 6.105 Block numbers for surrounding medium

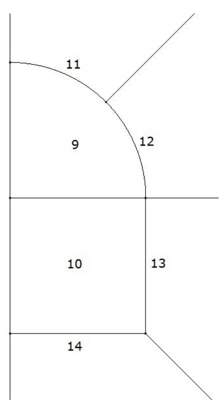


Figure 6.106 Block numbers for tunnel core and lining

Building Block No 1

1. Click Zoom toolbar to turn on in Figure 6.107.
2. Click the mouse at top-left point and release at bottom-right point as shown in Figure 6.108.
3. Figure 6.109 shows zoomed area around upper core.
4. Click Zoom toolbar again to turn off.
5. Click Block Editor toolbar in Figure 6.110.
6. Select Hexa, Cylindrical, and Z axis as shown in Figure 6.111.
7. Click OK.
8. Click Draw Index Number in Figure 6.112.
9. Select Ent. Point for Click Point Snap as in Figure 6.103.

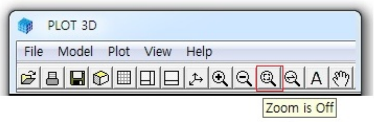


Figure 6.107 Zoom toolbar

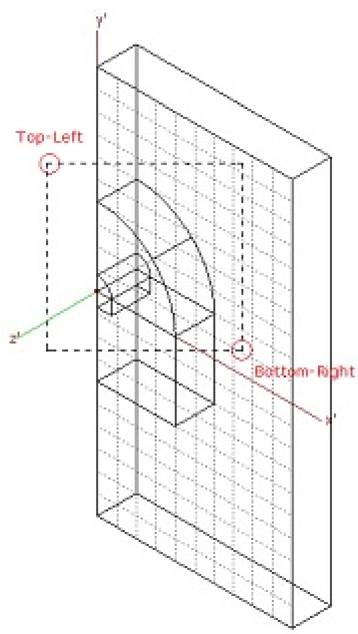


Figure 6.108 Zoom area selection

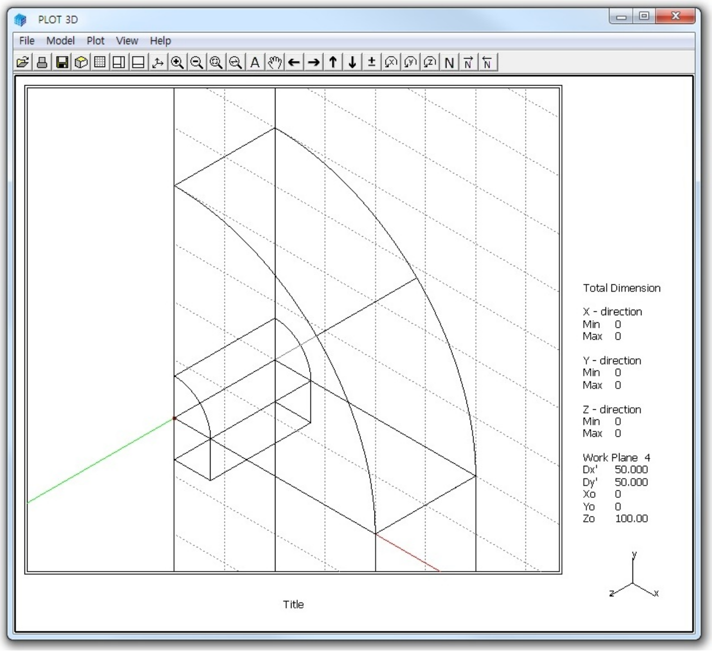


Figure 6.109 Zoomed area around upper core

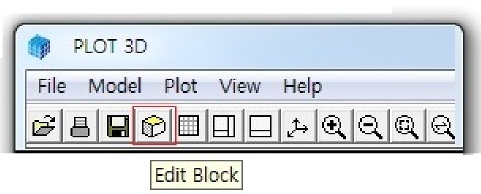


Figure 6.110 Block editor toolbar

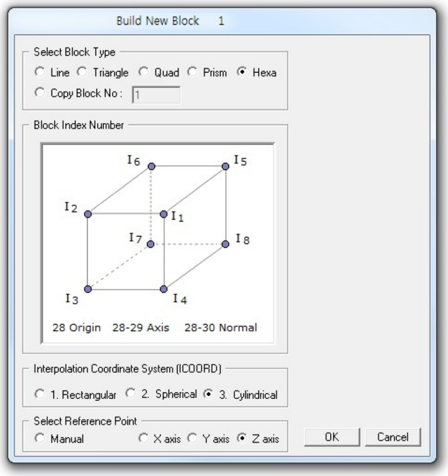


Figure 6.111 Block type selection

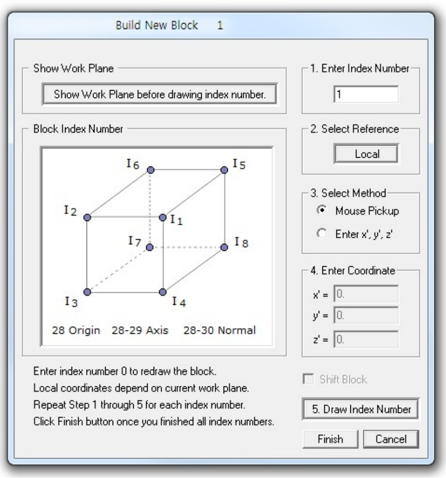


Figure 6.112 Hexa block

10. Select Work Plane No 5 for front surface using Back / Next button in Figure 6.103.
11. Click the points for index numbers on front surface as in Fig 6.113.

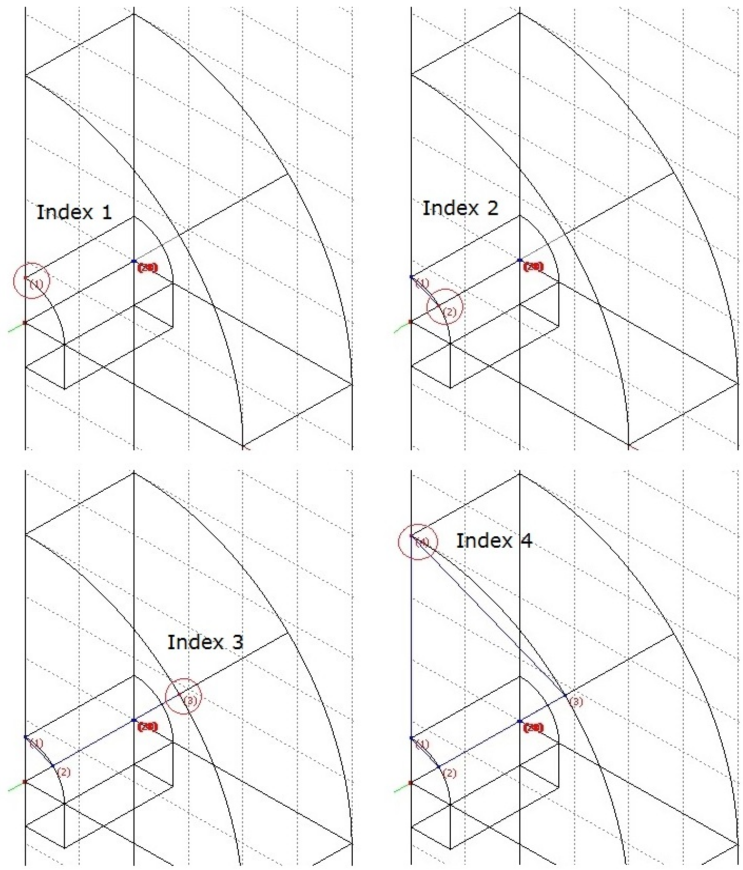


Figure 6.113 Index numbers on front surface

12. Select Work Plane No 4 for back surface using Back / Next button in Figure 6.103.
13. Click the points for index numbers on back surface as in Fig 6.114.

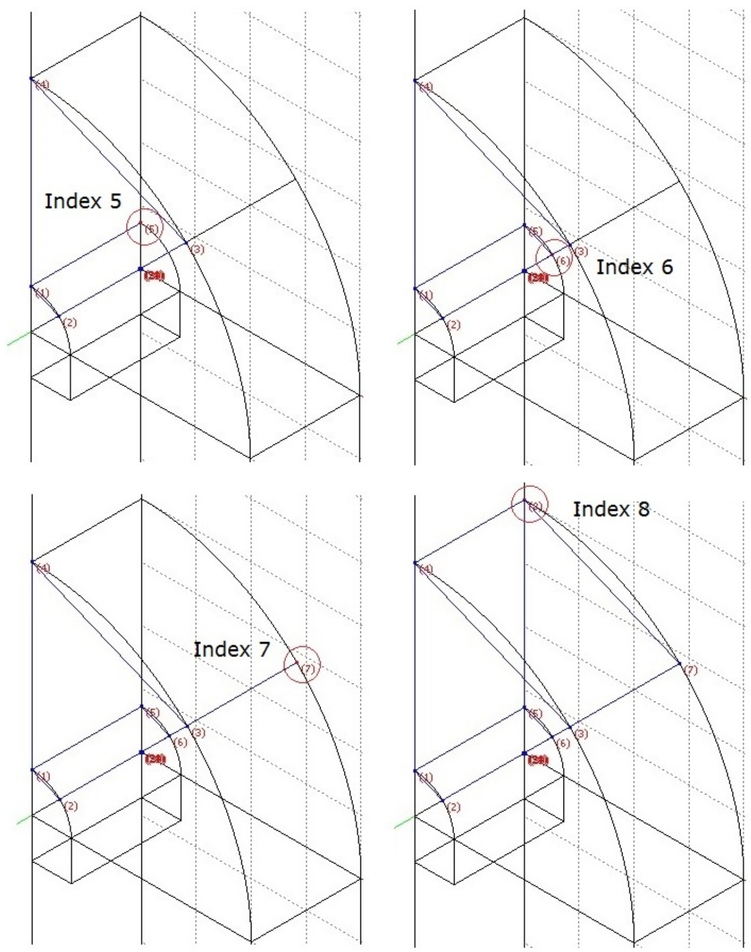


Figure 6.114 Index numbers on back surface

Now, the geometry of the first hexahedron block is completed.

14. Click Finish in Figure 6.103 and then click Finish in Figure 6.112.
15. Modify Title, Block Name and Material & Element Generation Parameters in Block Editor as shown in Figure 6.115.
16. Click Reset button.

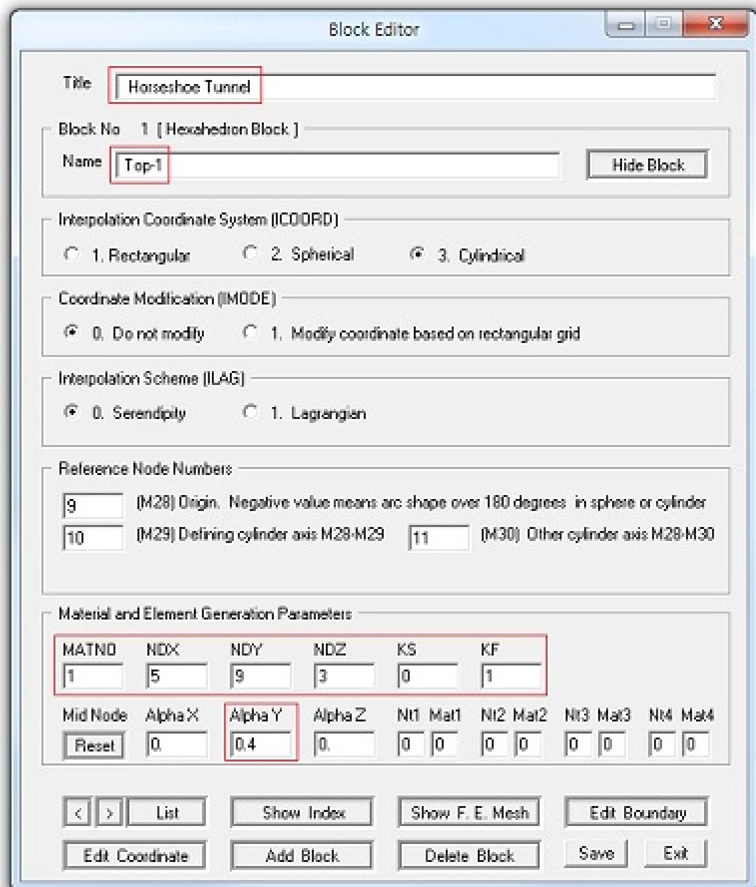


Figure 6.115 Block No 1

17. Figure 6.116 shows Block No 1 on drawing board.

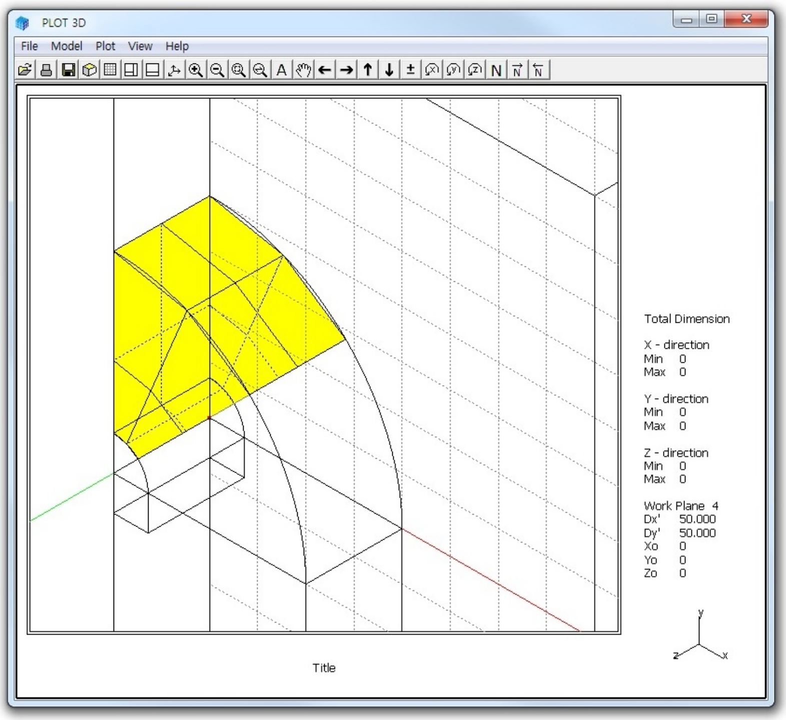


Figure 6.116 Block No 1 on drawing board

Building Other Blocks

18. Follow the same procedure as for Block No 1.

Block No 2	(Side-1):	Figures 6.117 - 6.118
Block No 3	(Side-2):	Figures 6.119 - 6.120
Block No 4	(Bottom-1):	Figures 6.121 - 6.122
Block No 5	(Top-2):	Figures 6.123 - 6.124
Block No 6	(Side-3):	Figures 6.125 - 6.126
Block No 7	(Side-4):	Figures 6.127 - 6.128
Block No 8	(Bottom-2):	Figures 6.129 - 6.130
Block No 9	(Core-1):	Figures 6.131 - 6.132
Block No 10	(Core-2):	Figures 6.133 - 6.134
Block No 11	(Liner-1):	Figures 6.135 - 6.136
Block No 12	(Liner-2):	Figures 6.137 - 6.138
Block No 13	(Liner-3):	Figures 6.139 - 6.140
Block No 14	(Liner-4):	Figures 6.141 - 6.142

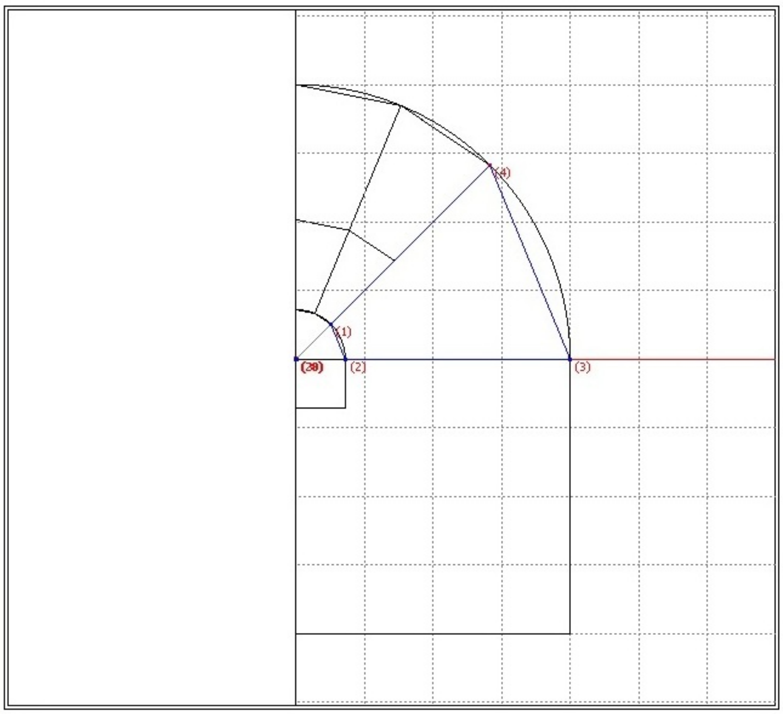


Figure 6.117 Index numbers on front surface (Block No 2)

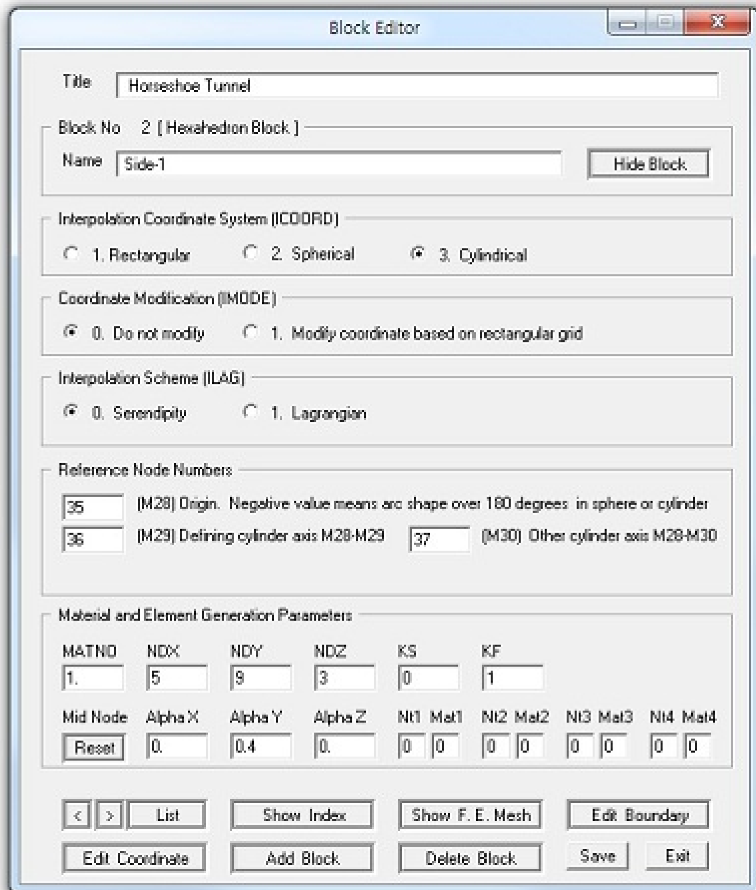


Figure 6.118 Block No 2 (Side-1)

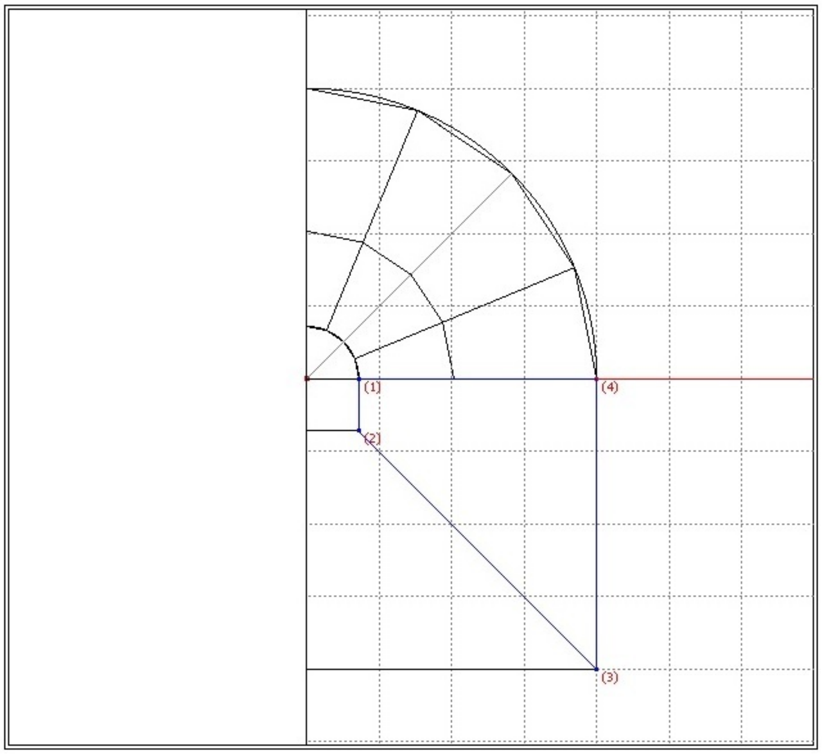


Figure 6.119 Index numbers on front surface (Block No 3)

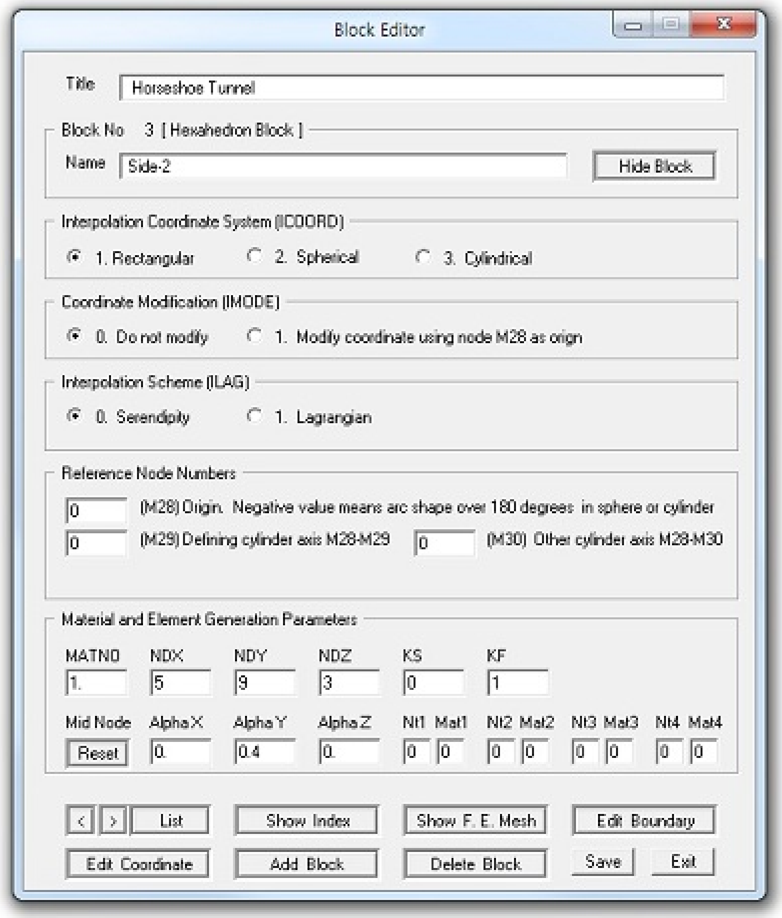


Figure 6.120 Block No 3 (Side-2)

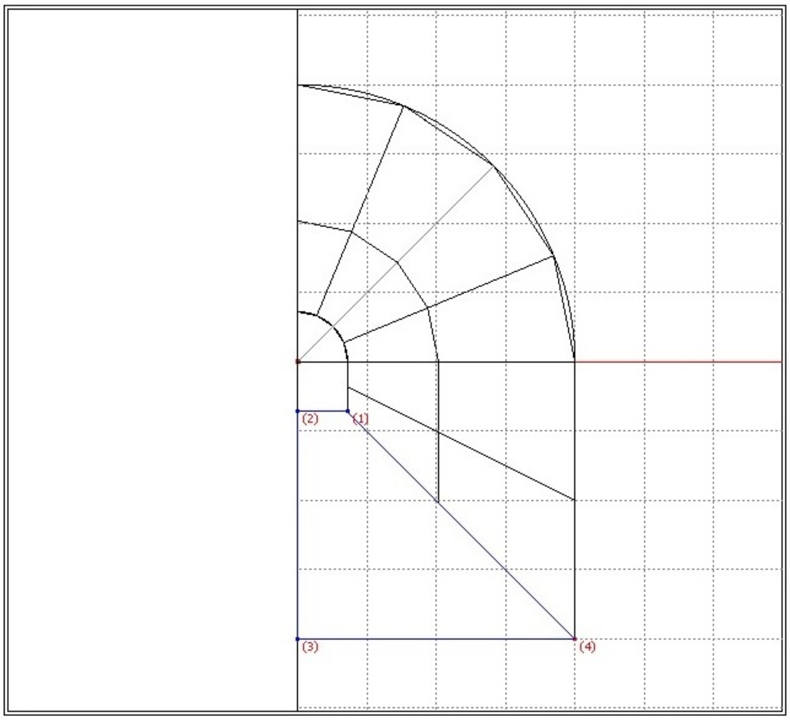


Figure 6.121 Index numbers on front surface (Block No 4)

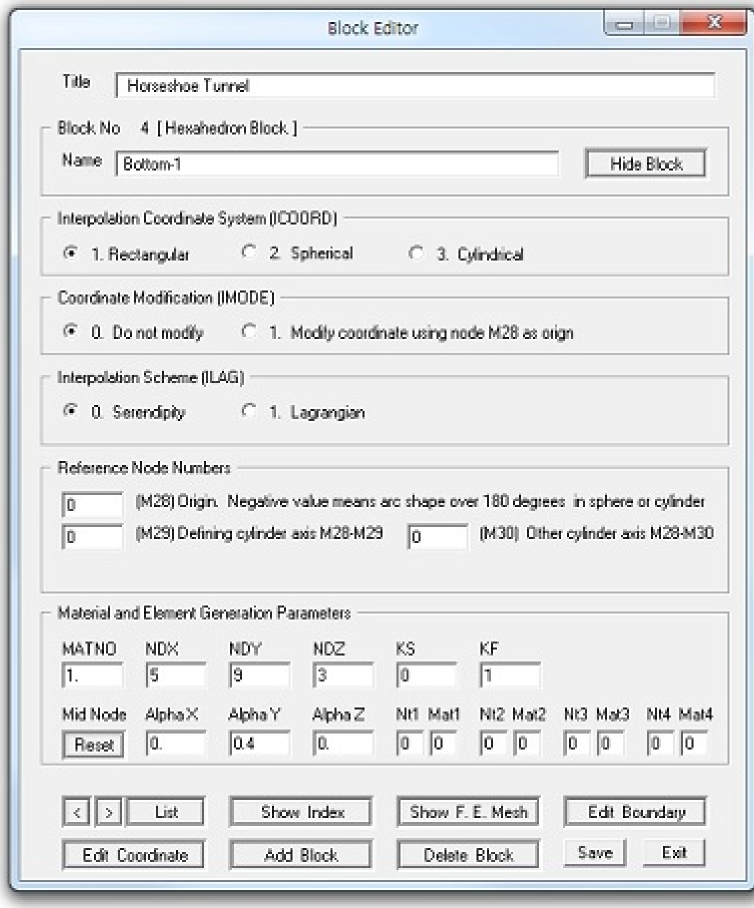


Figure 6.122 Block No 4 (Bottom-1)

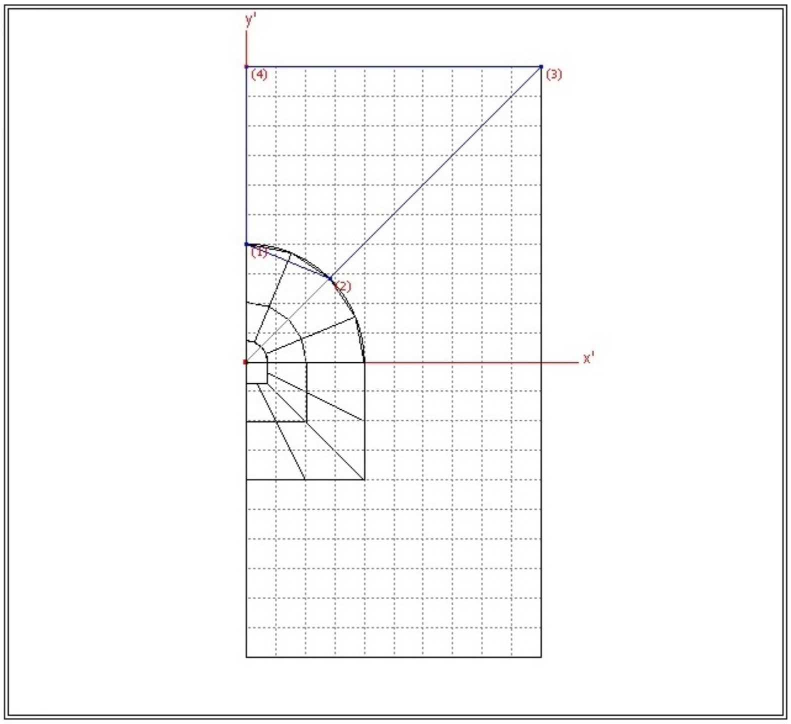


Figure 6.123 Index numbers on front surface (Block No 5)

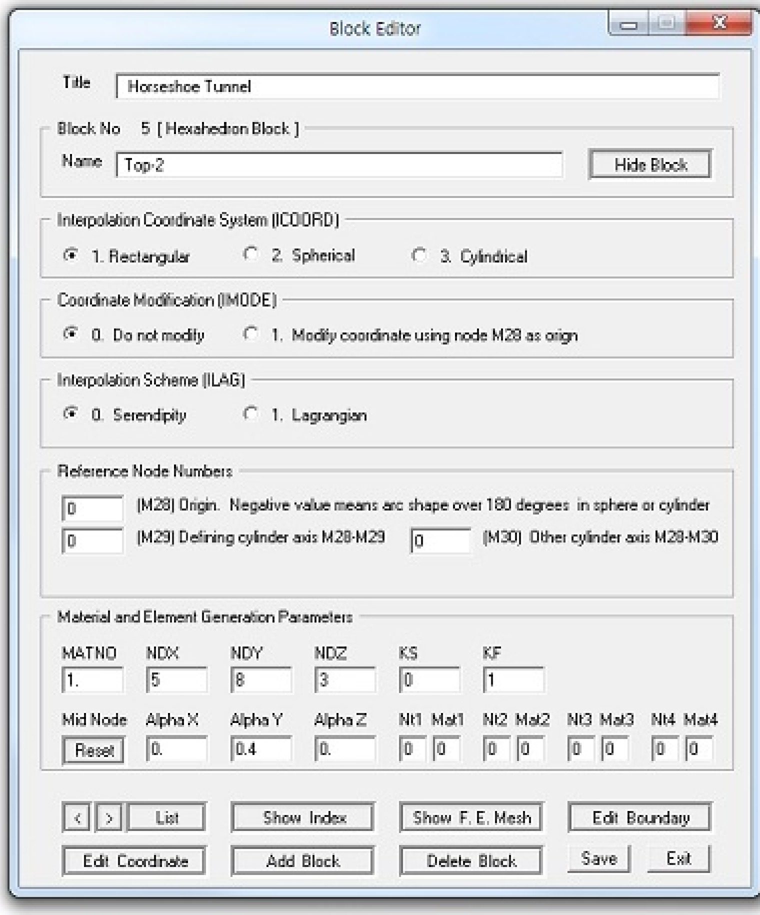


Figure 6.124 Block No 5 (Top-2)

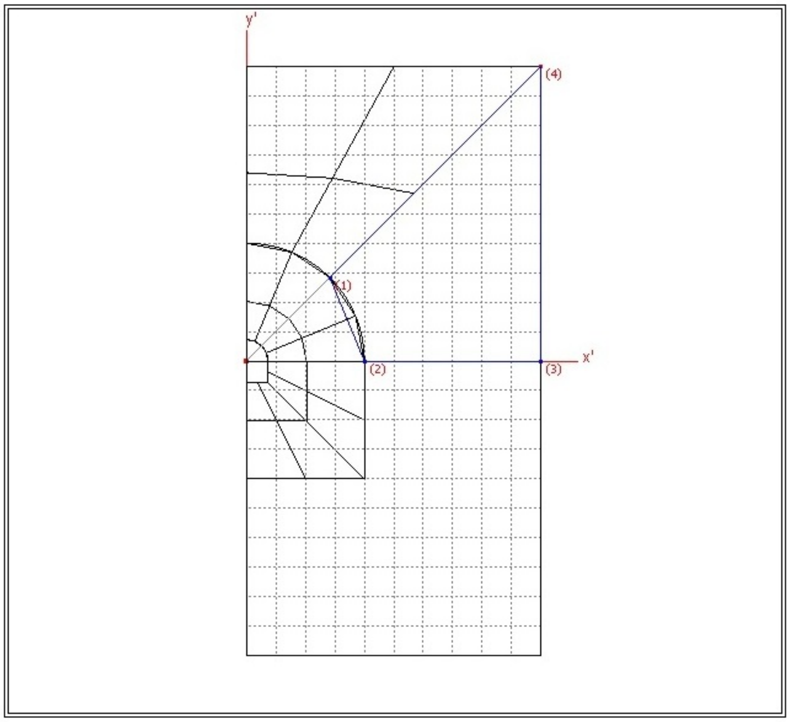


Figure 6.125 Index numbers on front surface (Block No 6)

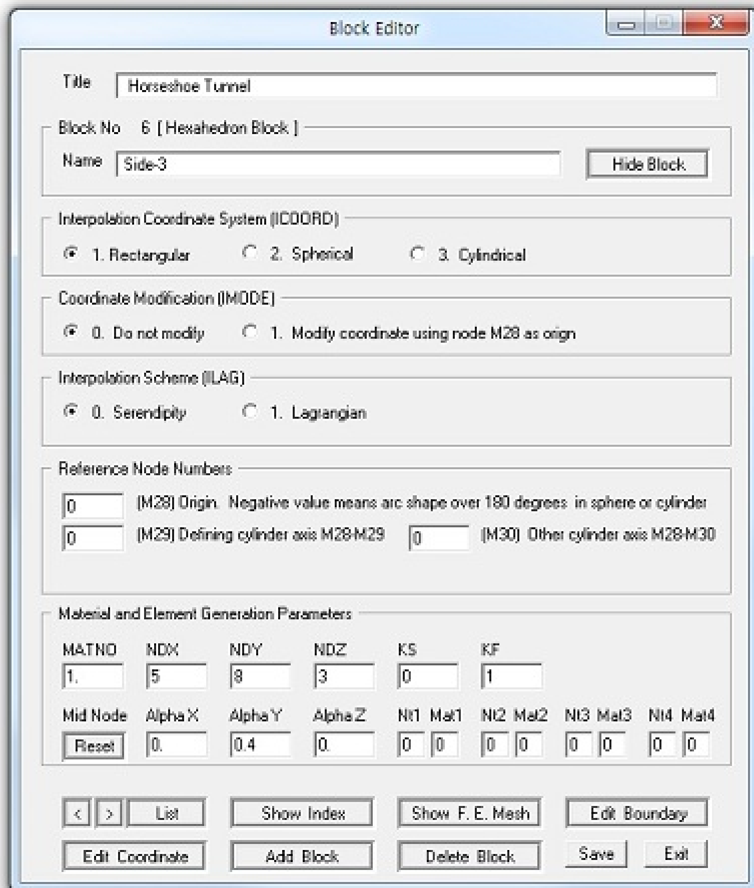


Figure 6.126 Block No 6 (Side-3)

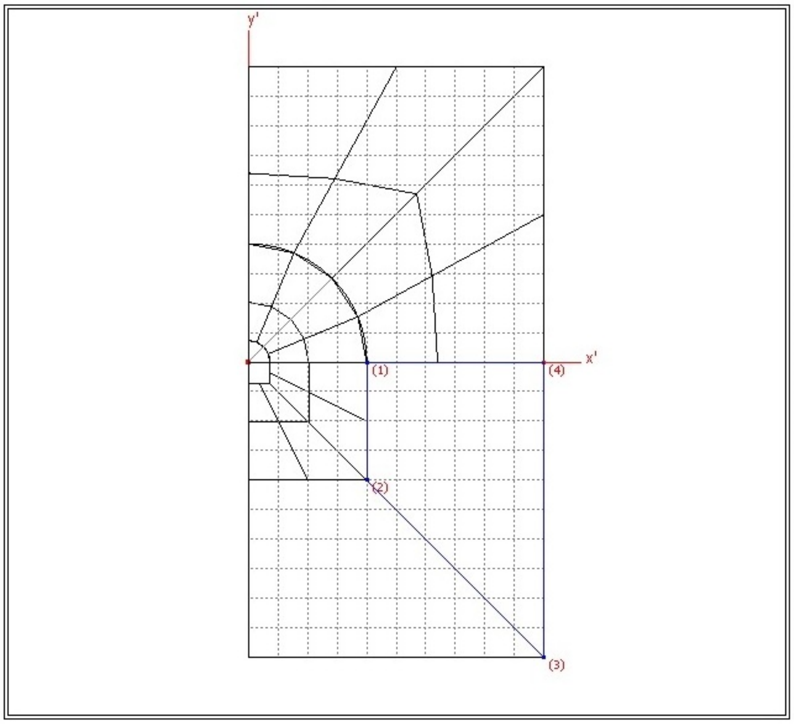


Figure 6.127 Index numbers on front surface (Block No 7)

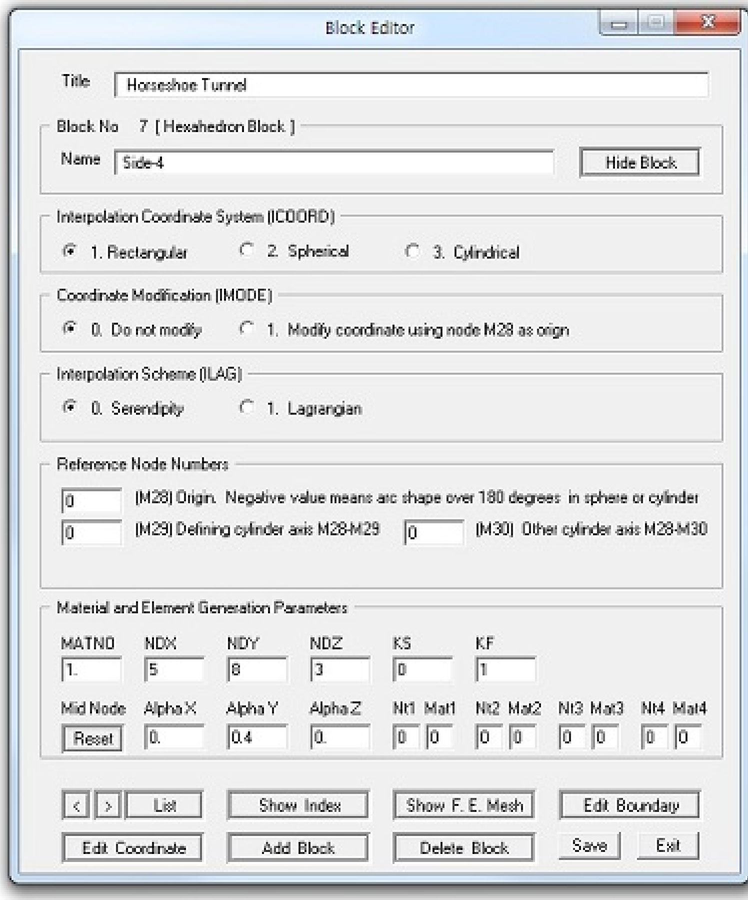


Figure 6.128 Block No 7 (Side-4)

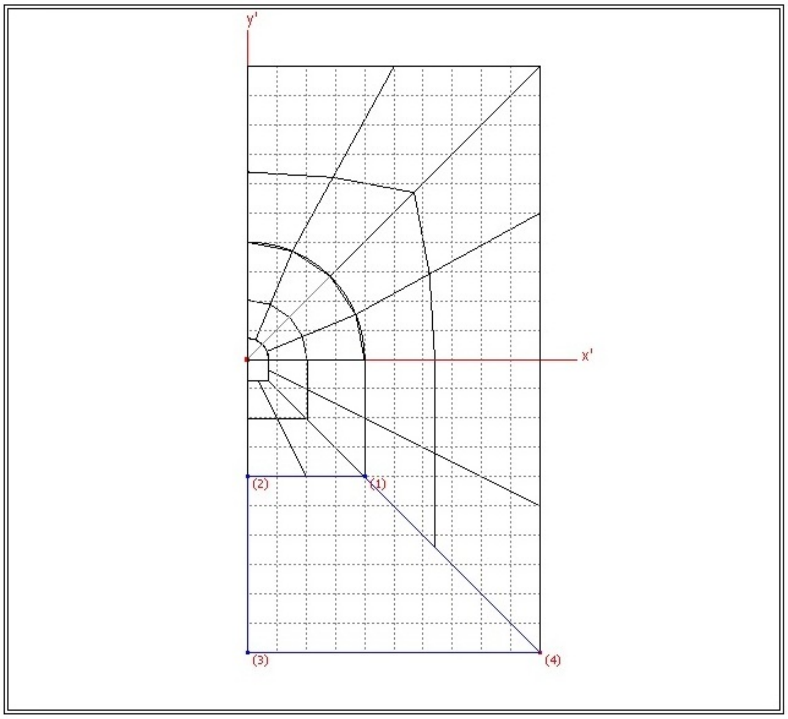


Figure 6.129 Index numbers on front surface (Block No 8)

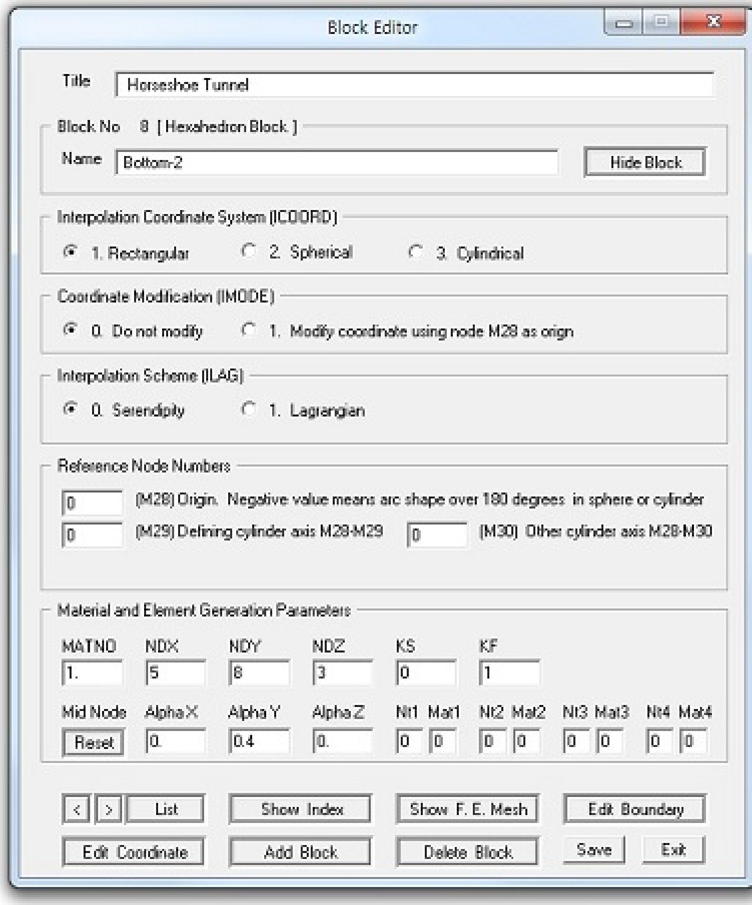


Figure 6.130 Block No 8 (Bottom-2)

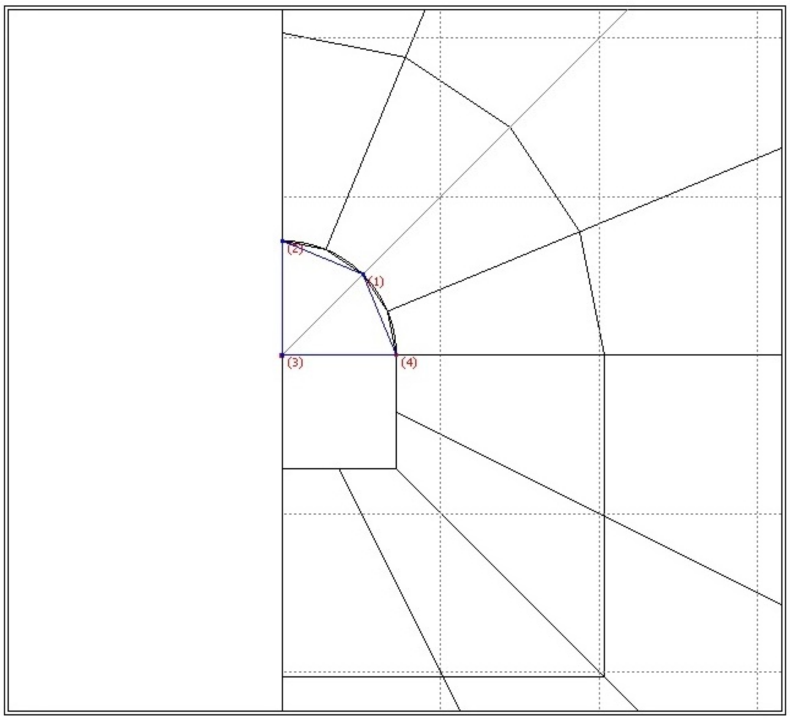


Figure 6.131 Index numbers on front surface (Block No 9)

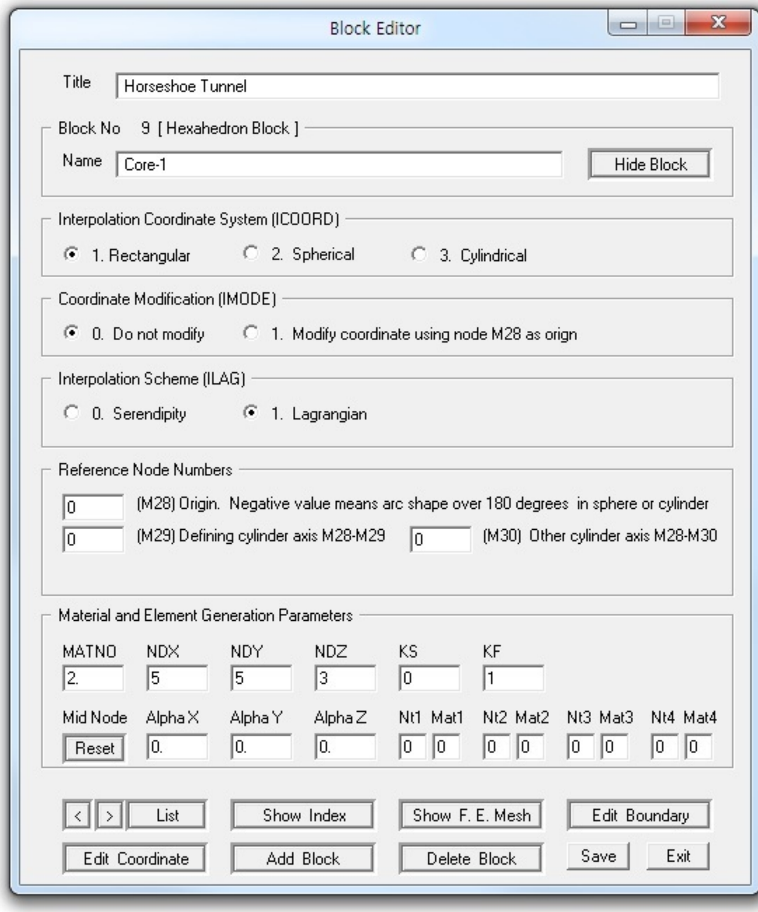


Figure 6.132 Block No 9 (Core-1)

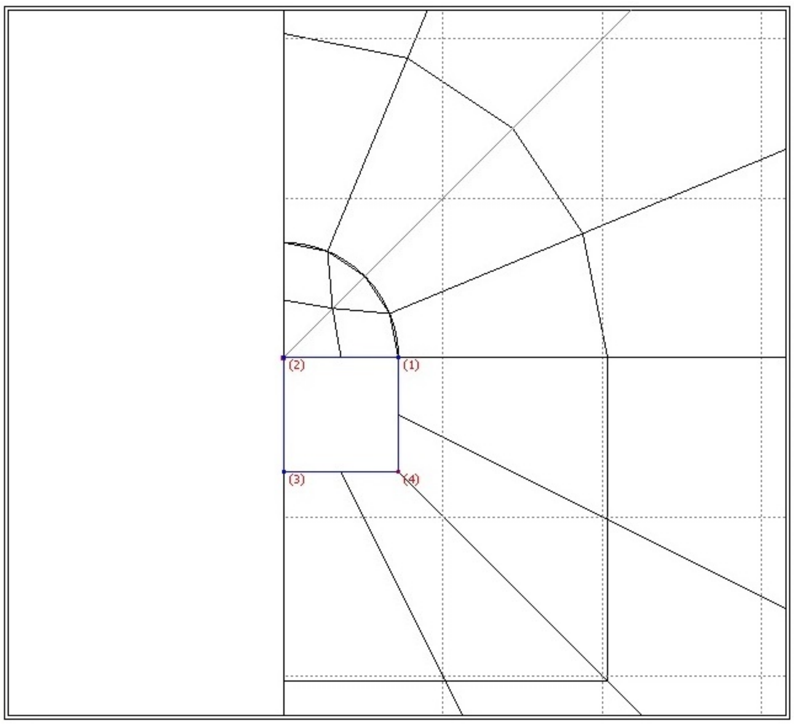


Figure 6.133 Index numbers on front surface (Block No 10)

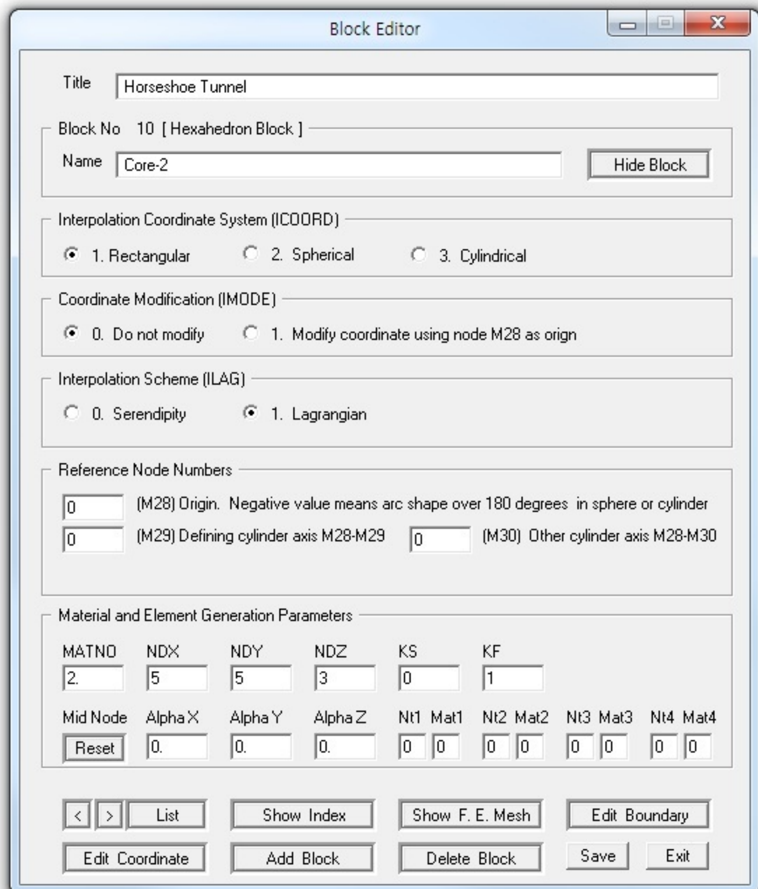


Figure 6.134 Block No 10 (Core-2)

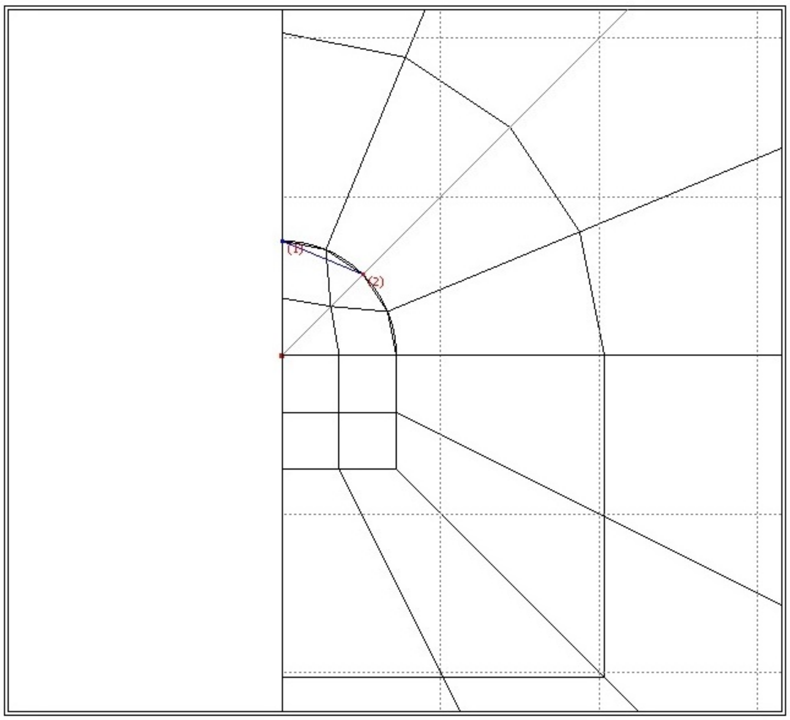


Figure 6.135 Index numbers on front surface (Block No 11)

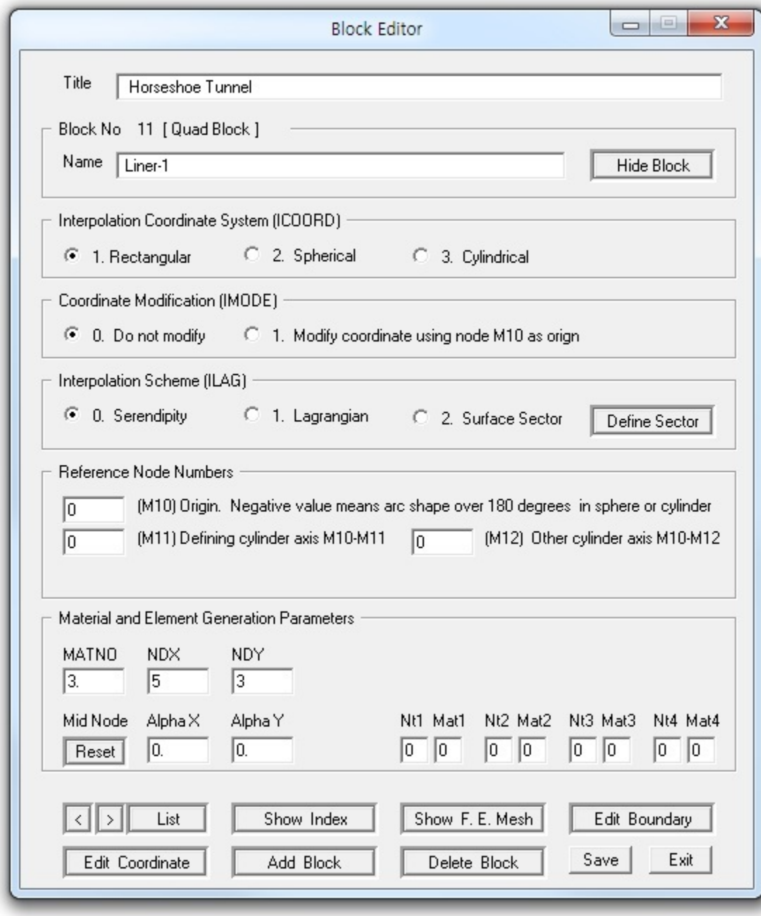


Figure 6.136 Block No 11 (Liner-1)

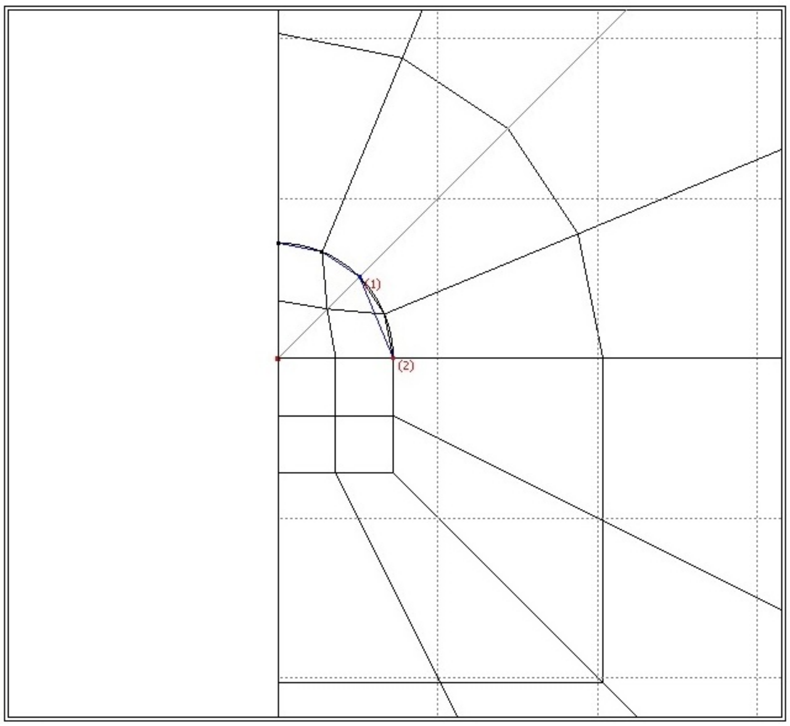


Figure 6.137 Index numbers on front surface (Block No 12)

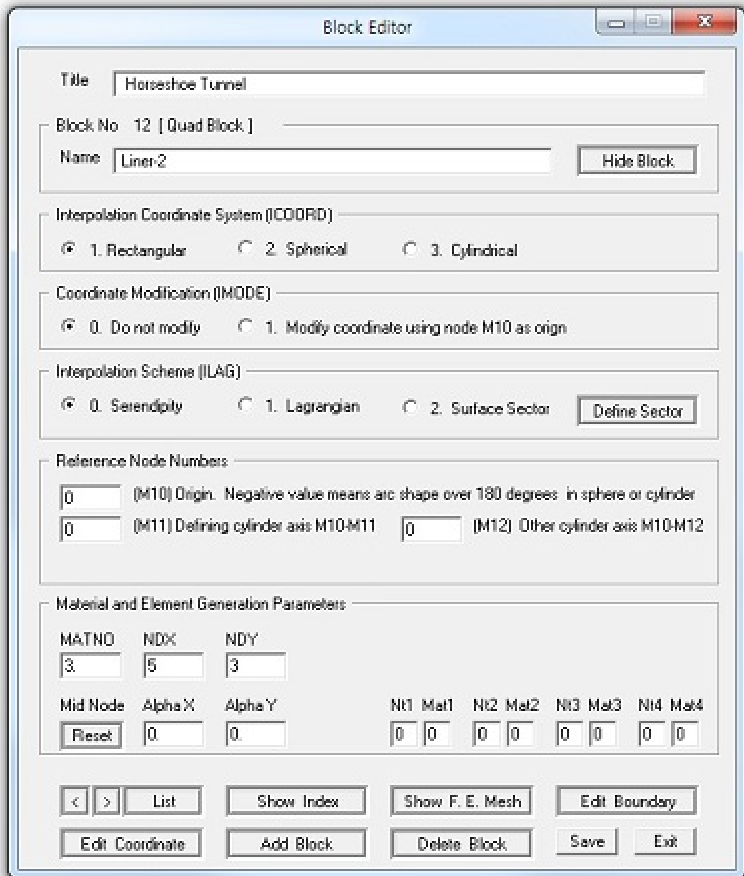


Figure 6.138 Block No 12 (Liner-2)

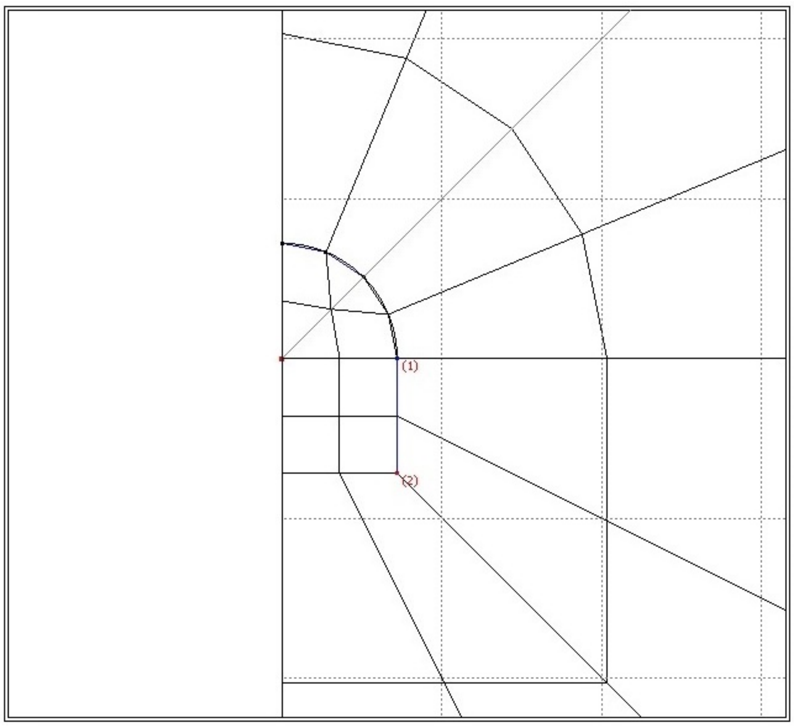


Figure 6.139 Index numbers on front surface (Block No 13)

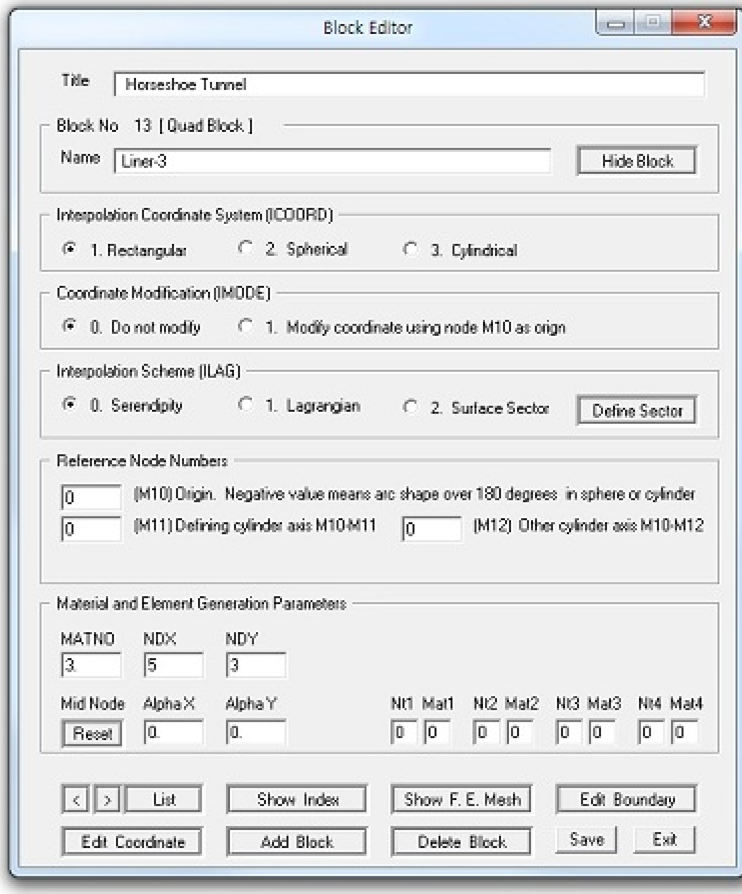


Figure 6.140 Block No 13 (Liner-3)

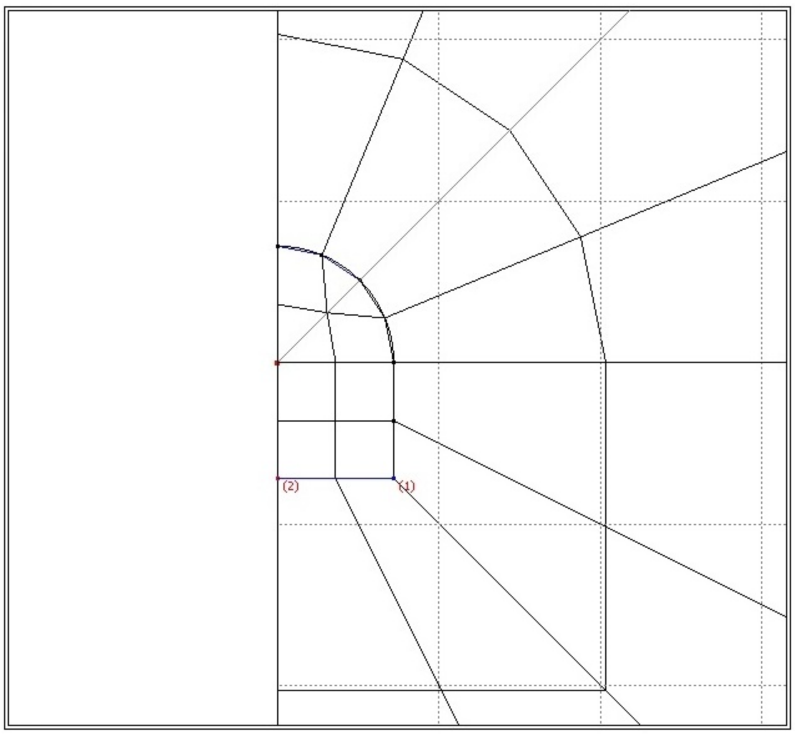


Figure 6.141 Index numbers on front surface (Block No 14)

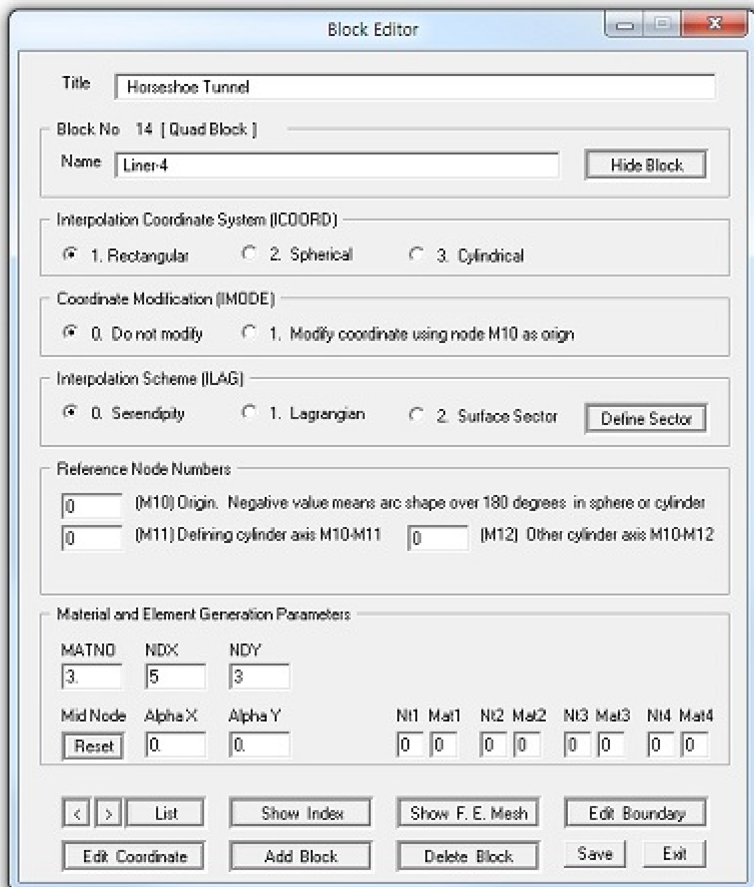


Figure 6.142 Block No 14 (Liner-4)

19. All blocks are listed as shown in Figure 6.143 by clicking List button in the Block Editor dialog.
20. Click OK.

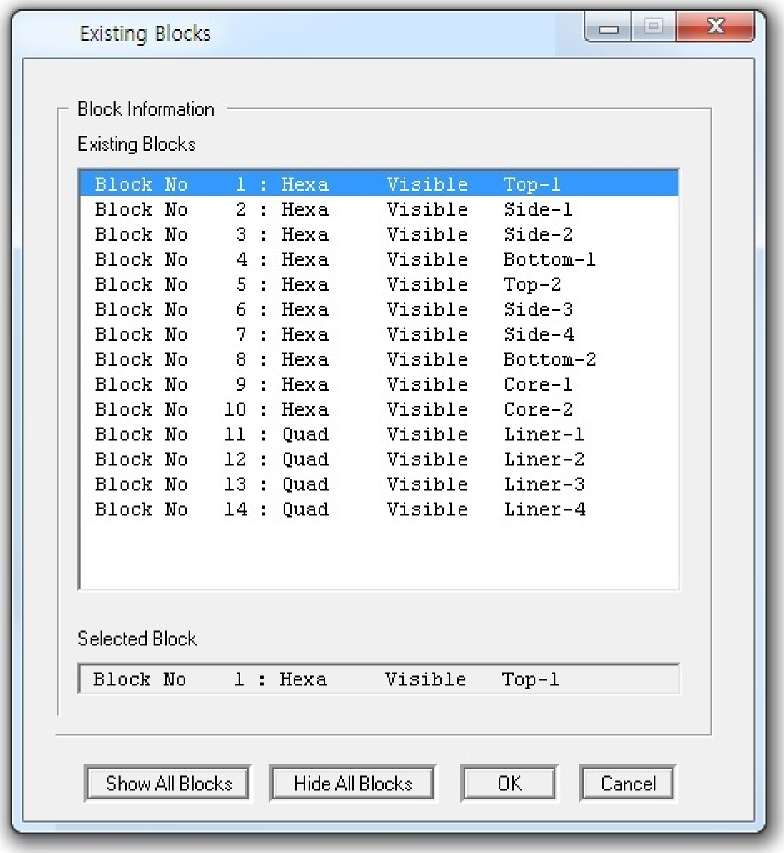


Figure 6.143 Listing of all of the blocks

Step 6: Set Global Boundary

1. Select Model → Edit Global Boundary in Figure 6.144.

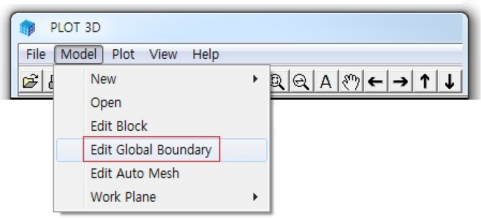


Figure 6.144 Edit global boundary menu

2. Set the boundary codes as shown in Figures 6.145 and 6.146.
3. Select Yes override block boundary.
4. Click Save and type in file name as EX4.

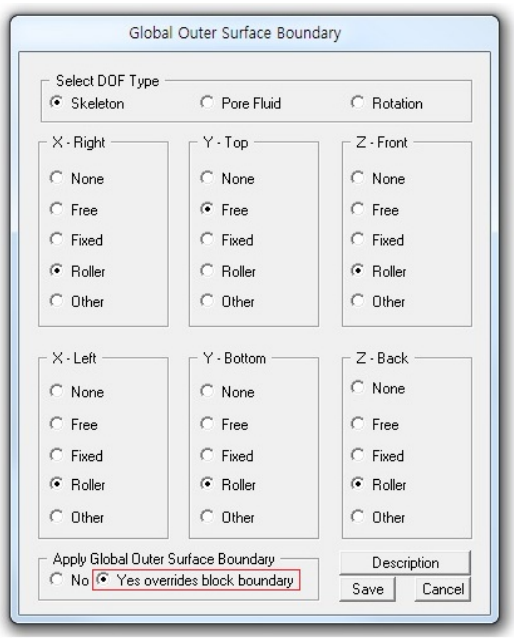


Figure 6.145 Global outer surface boundary (Skeleton)

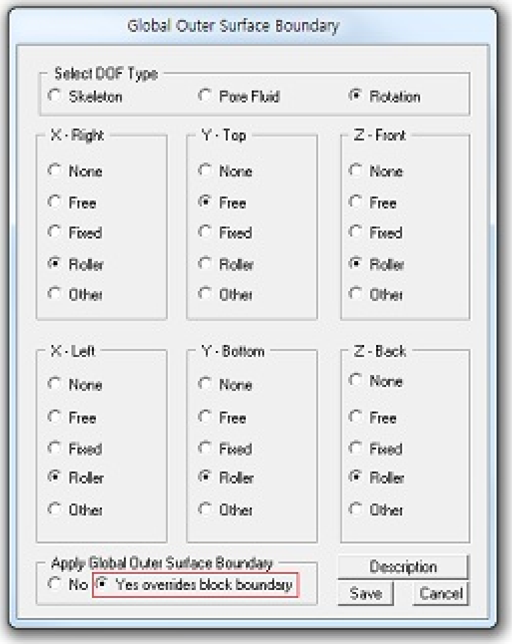


Figure 6.146 Global outer surface boundary (Rotation)

Step 7: View Selected Material

1. Select View → Mesh in PLOT-3D menu.
2. Select Only Selected One for Material Selection in Figure 6.147.
3. Click Number 3 in Available list.
4. Click OK.

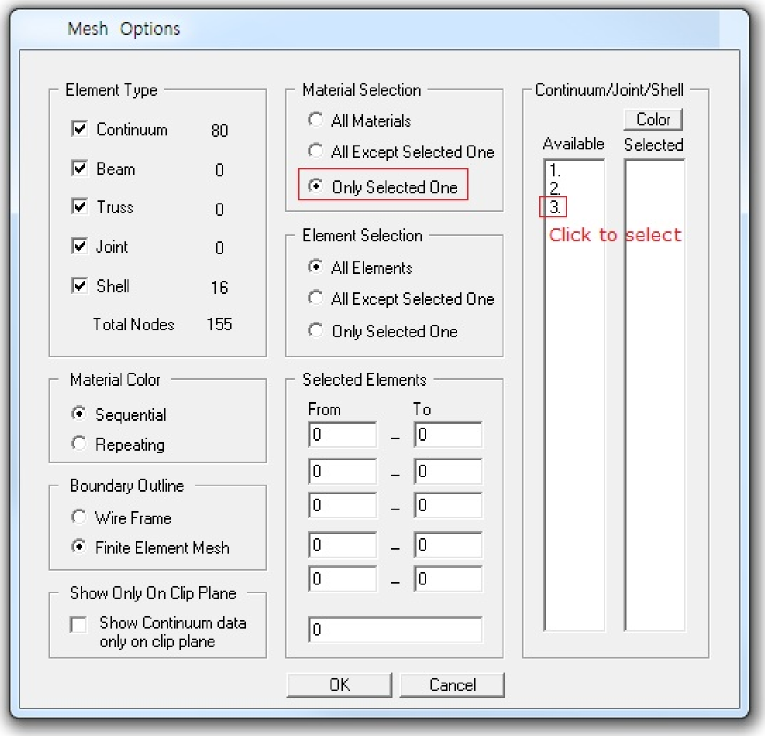


Figure 6.147 Mesh options

5. Figure 6.148 shows selected block meshes with material number 3 which represents tunnel lining.

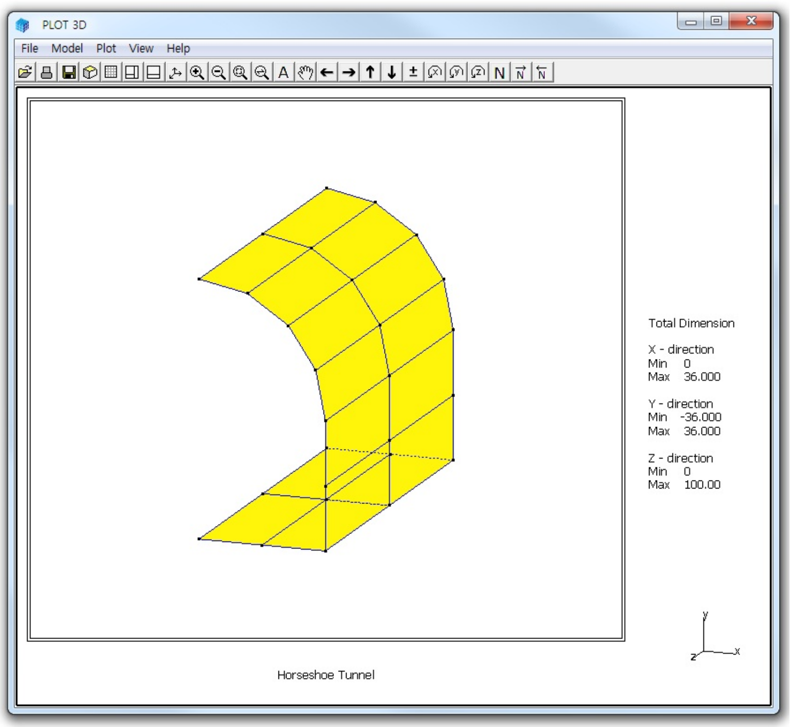


Figure 6.148 Block meshes representing tunnel lining

Step 8: Plot Finite Element Mesh

1. Click Show F. E. Mesh in Block Editor dialog.
2. Select Model → Work Plane → Isometric Y-axis in PLOT-3D menu.
3. Figure 6.149 shows finite element mesh.

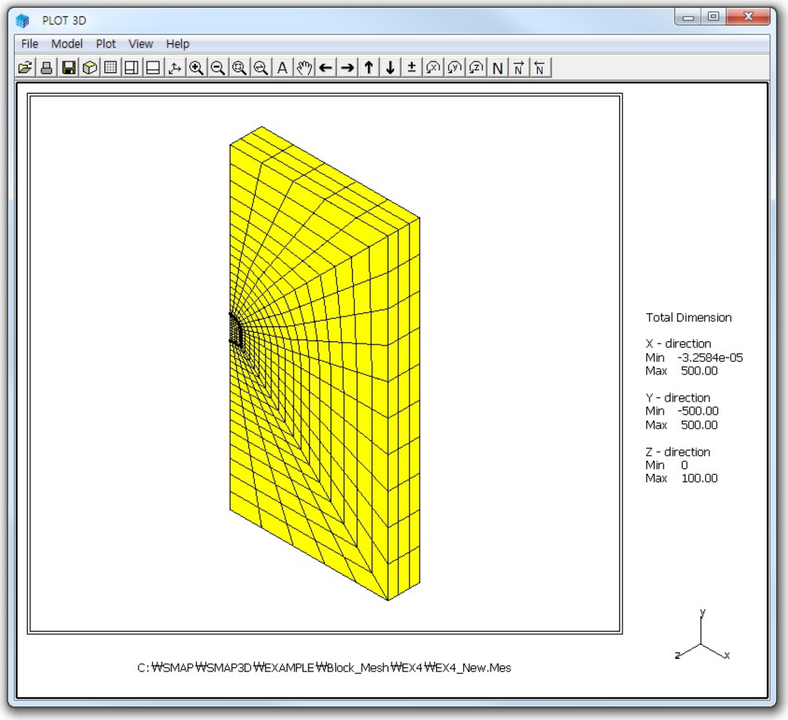


Figure 6.149 Finite element mesh

4. Follow the same procedure to view selected material as in Step 6.
5. Figure 6.150 shows finite element mesh with material number 3 which represents tunnel lining.

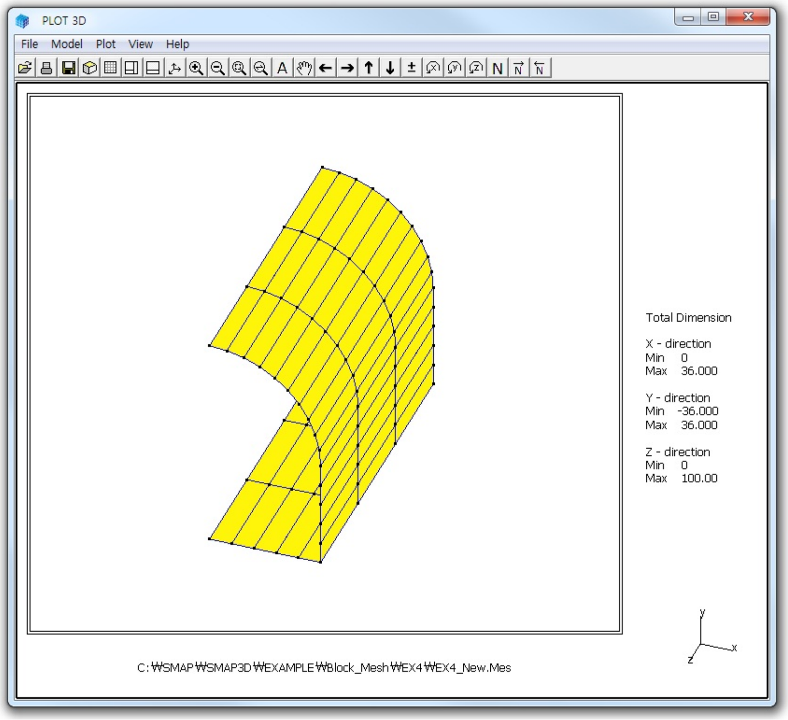


Figure 6.150 Finite element mesh representing tunnel lining

6.5 Space Truss

This example illustrates how to build directly finite element meshes for the space truss as schematically shown in Figure 6.151.

This example involves following nine main steps:

1. Access finite element mesh generator
2. Set work plane
3. Build cube entity
4. Add work plane
5. Build truss elements
6. Edit mesh title
7. Plot node and element numbers
8. Edit boundary codes
9. Plot skeleton boundary codes

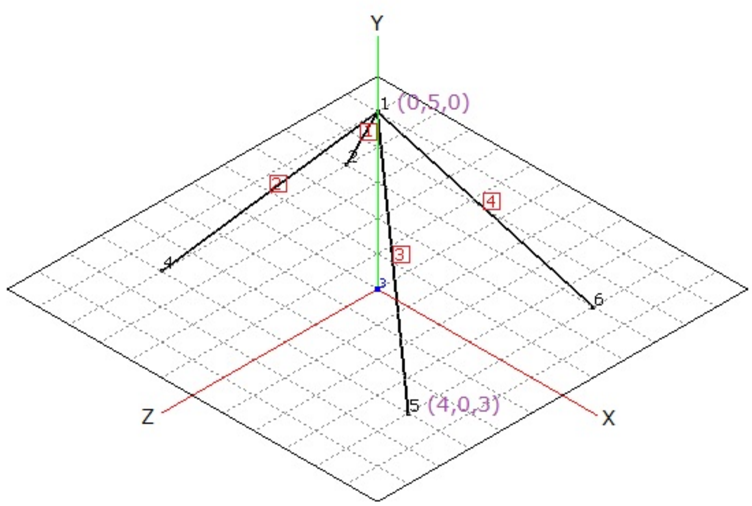


Figure 6.151 Space truss consisting of four elements

Step 1: Access Finite Element Mesh Generator

Access Finite Element Mesh Generator by selecting the following menu items in SMAP as in Figure 6.152:

Plot → Mesh → F. E. Mesh → New

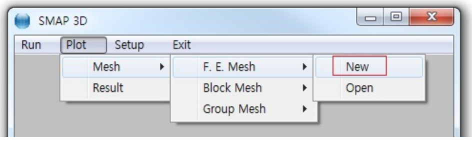


Figure 6.152 Access finite element mesh generator

Step 2: Set Work Plane

1. Select Work Plane No 4 and set parameters as shown in Fig. 6.153.
2. Click Update button.

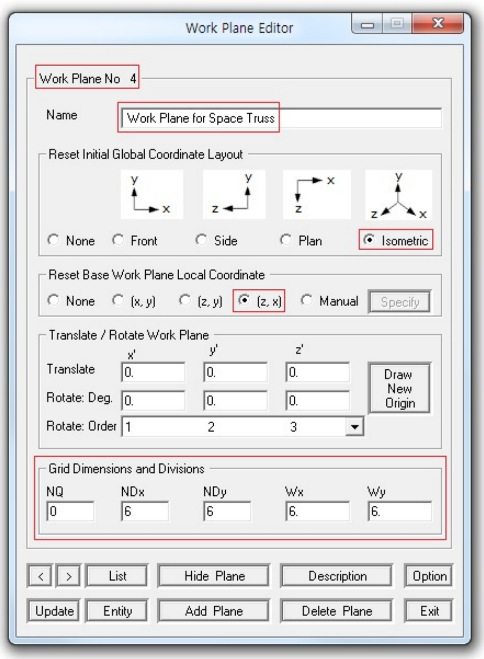


Figure 6.153 Work plane editor

3. Figure 6.154 shows work plane with Isometric Y-axis.

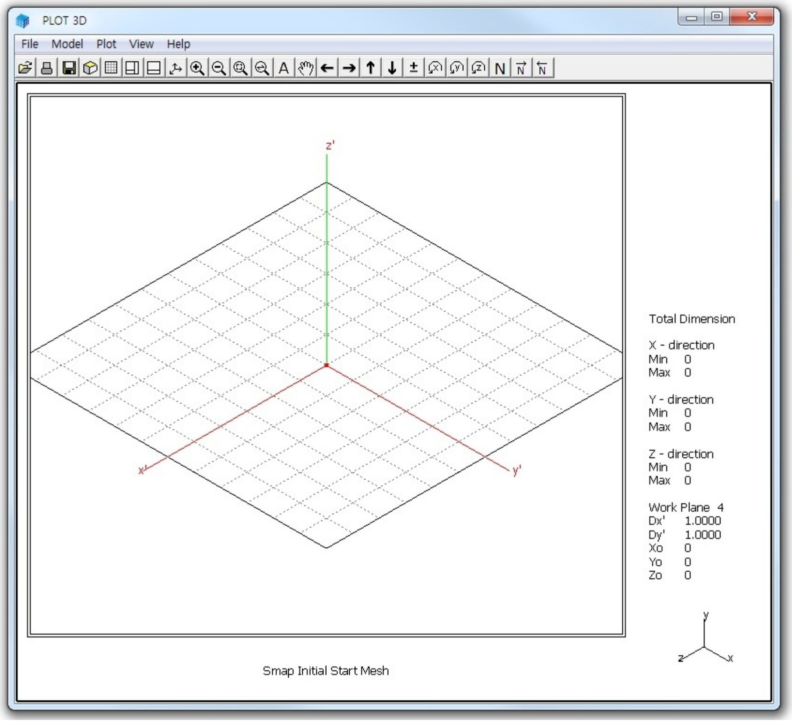


Figure 6.154 Work plane with Isometric Y-axis

Step 3: Build Cube Entity

1. Click Entity in Figure 6.153.
2. Click Add in Entity Editor dialog in Figure 6.155.

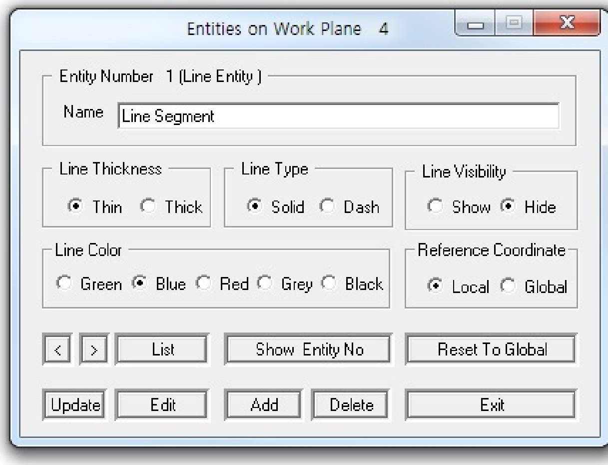


Figure 6.155 Entity editor

3. Select Cube in Entity Type Selection dialog in Figure 6.156.
4. Click OK.

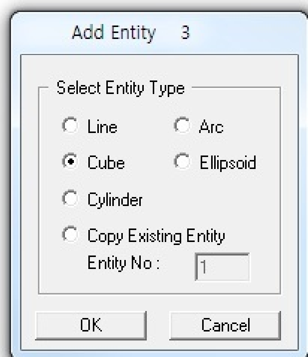


Figure 6.156 Entity type selection

5. Set geometric parameters of cube entity as shown in Figure 6.157.
6. Click Draw Cube Entity.
7. Click Finish.

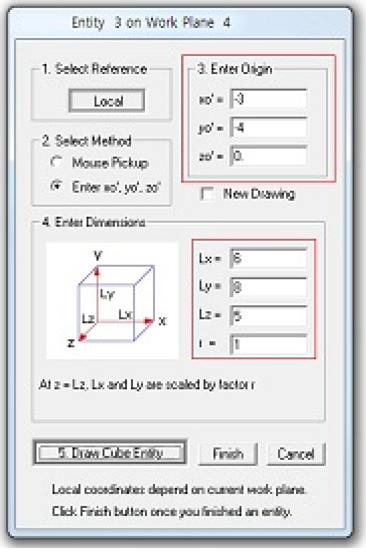


Figure 6.157 Cube entity

8. Set parameters of cube entity as shown in Figure 6.158.
9. Click Reset To Global and then click Exit.

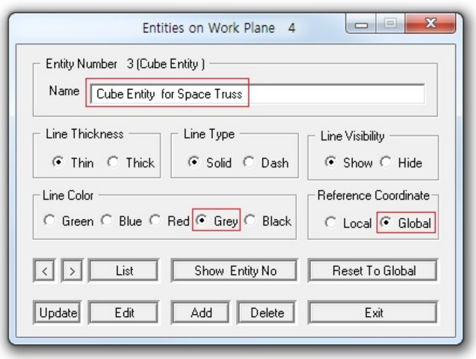


Figure 6.158 Entity editor

10. Figure 6.159 shows cube entity on drawing board.

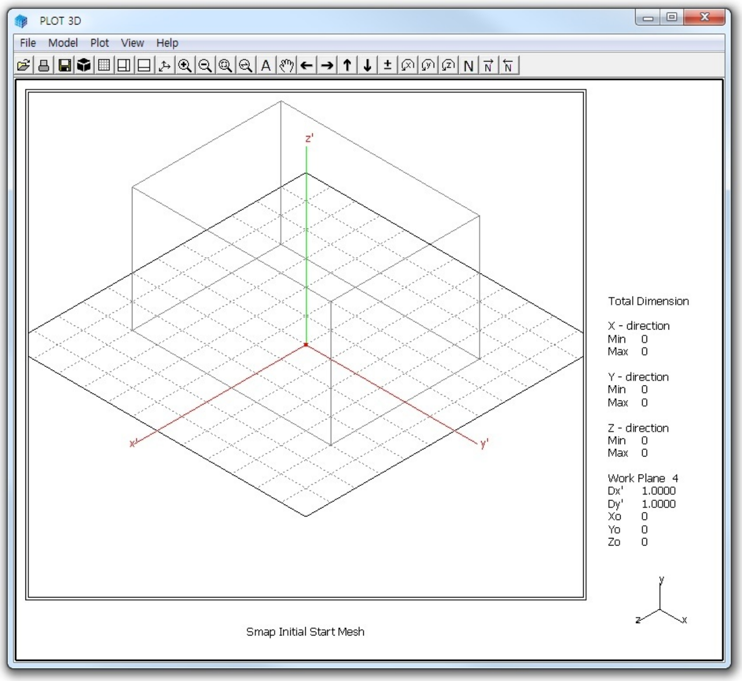


Figure 6.159 Cube entity on drawing board

Step 4: Add Work Plane

At Step 2, we set Work Plane No 4 which represents bottom surface. At Step 3, we built cube entity on this Work Plane No 4.

Here, we want to add new Work Plane No 5 in the following way:

- Copy Work Plane No 4 along with cube entity on it.
- Add this copied one as new Work Plane No 5.
- Modify such that it represents top surface.

Once we have this new Work Plane No 5, it will be much easier to build blocks since top and bottom surfaces of work planes can be accessed simply by one click of Back or Next button on Coordinates on Work Plane dialog in Figure 6.160.

Perform the following four steps:

1. Select Work Plane No 4 in Work Plane Editor dialog in Figure 6.153
2. Click Add Plane button in Figure 6.153
3. Modify Name and Translation as in Figure 6.161
4. Click Update in Figure 6.161

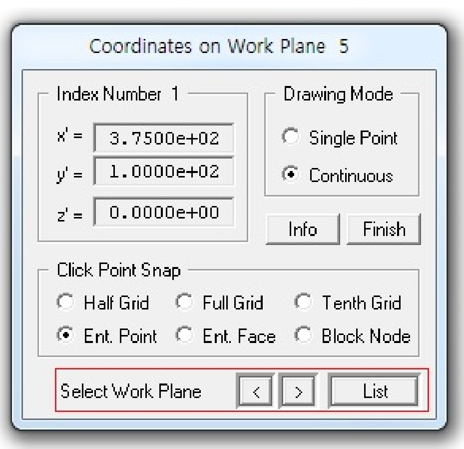


Figure 6.160 Coordinates on work plane

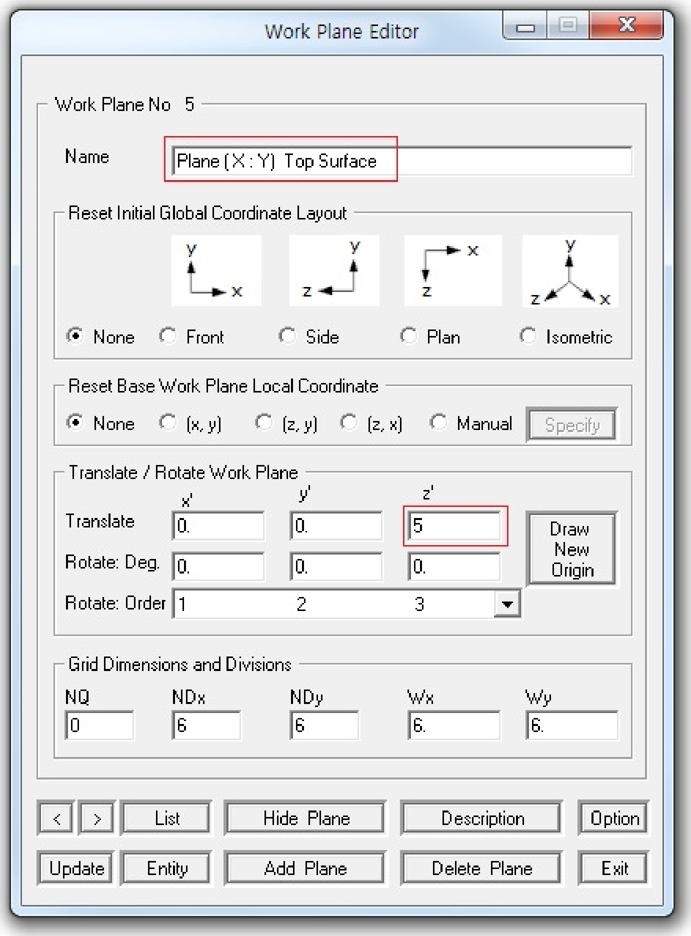


Figure 6.161 Work Plane No 5 representing top surface

Step 5: Build Truss Elements

Build Element 1

1. Click Edit Element toolbar in Figure 6.162.

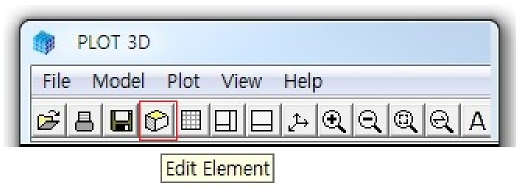


Figure 6.162 Edit element toolbar

Draw Index Numbers For Element 1

2. Select Line for element type in Figure 6.163.
3. Click OK.

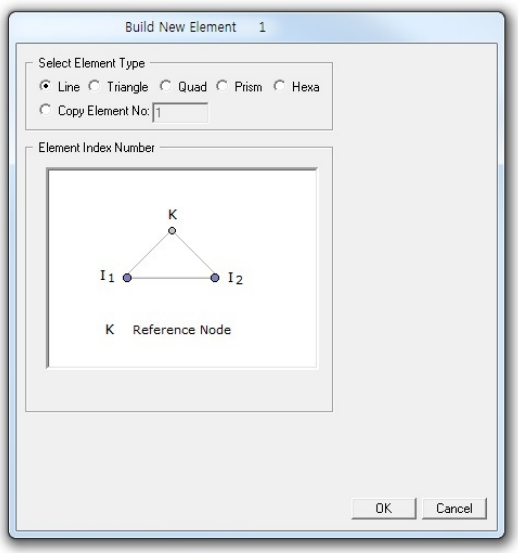


Figure 6.163 Element type selection

4. Type in 1 for MatNo and 1 for truss element as in Figure 6.164.
5. Click Draw Index Number.

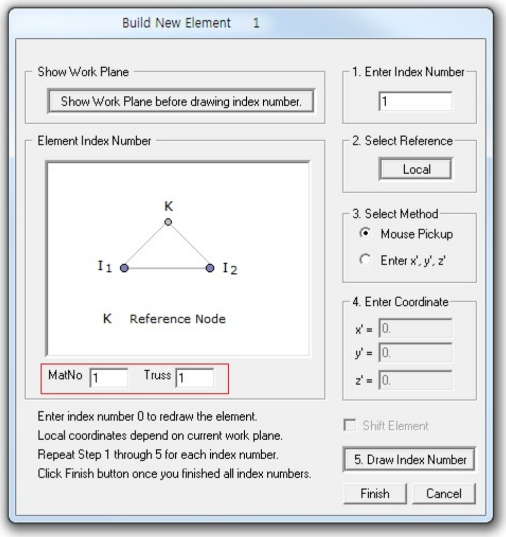


Figure 6.164 Line element

6. Select Full Grid for Click Point Snap as in Figure 6.165.

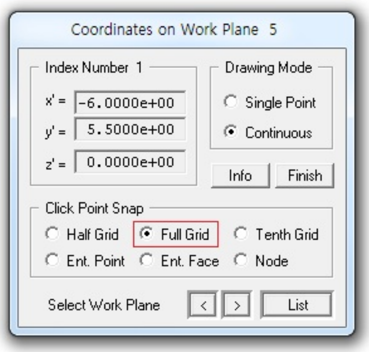


Figure 6.165 Coordinates on work plane

7. Select Work Plane No 5 for top surface using Back / Next button in Figure 6.165.
8. Click the point for index number 1 as in Figure 6.166.

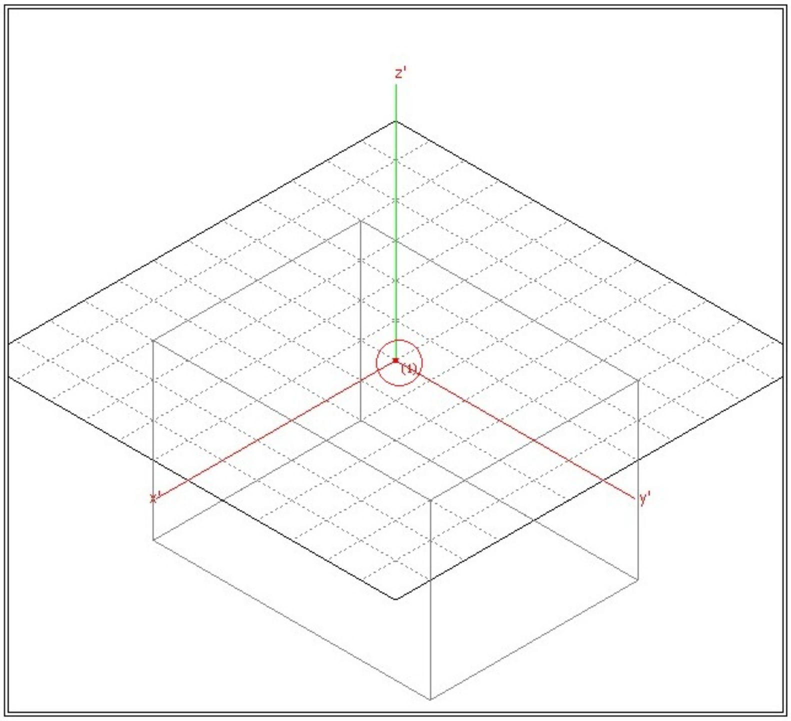


Figure 6.166 Index number 1 for Element 1

9. Select Work Plane No 4 for bottom surface using Back / Next button in Figure 6.165.
10. Click the points for index numbers 2 and 3 as in Figure 6.167.
11. Click Finish in Figure 6.165 and then click Finish in Figure 6.164.

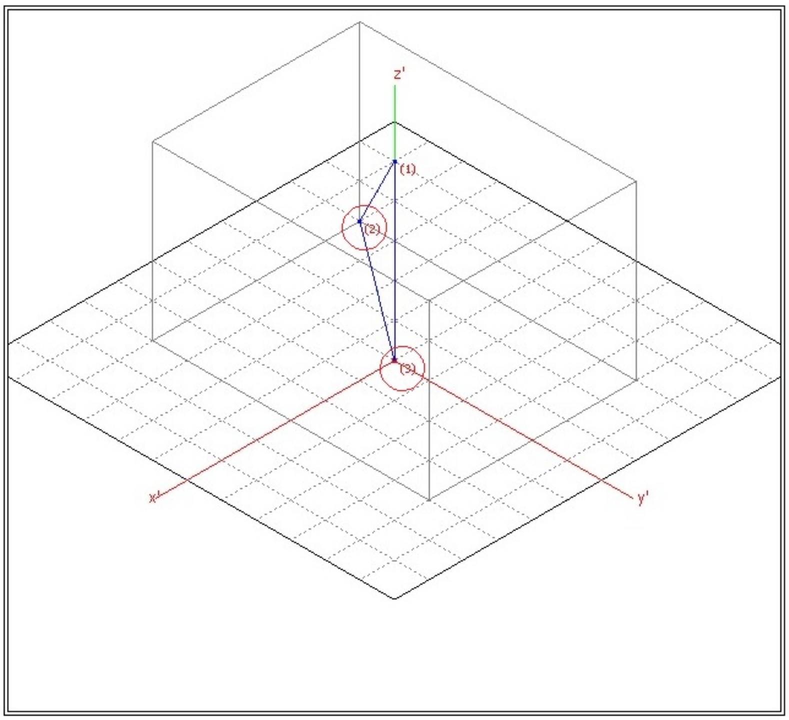


Figure 6.167 Index numbers 2 and 3 for Element 1

Build Element 2

12. Select Model → Edit Element in Figure 6.168.

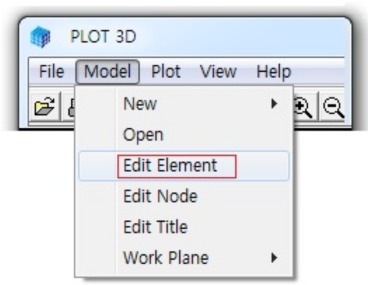


Figure 6.168 Edit element menu

13. Popup menu in Figure 6.169 is displayed by Shift + Right click.
14. Click Add menu.

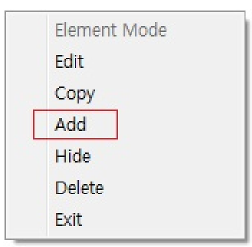


Figure 6.169 Popup menu for edit element

Draw Index Numbers For Element 2

15. Repeat steps 2 through 11 for Element 2 with MatNo = 2.
16. Figure 6.170 shows index numbers for Element 2.

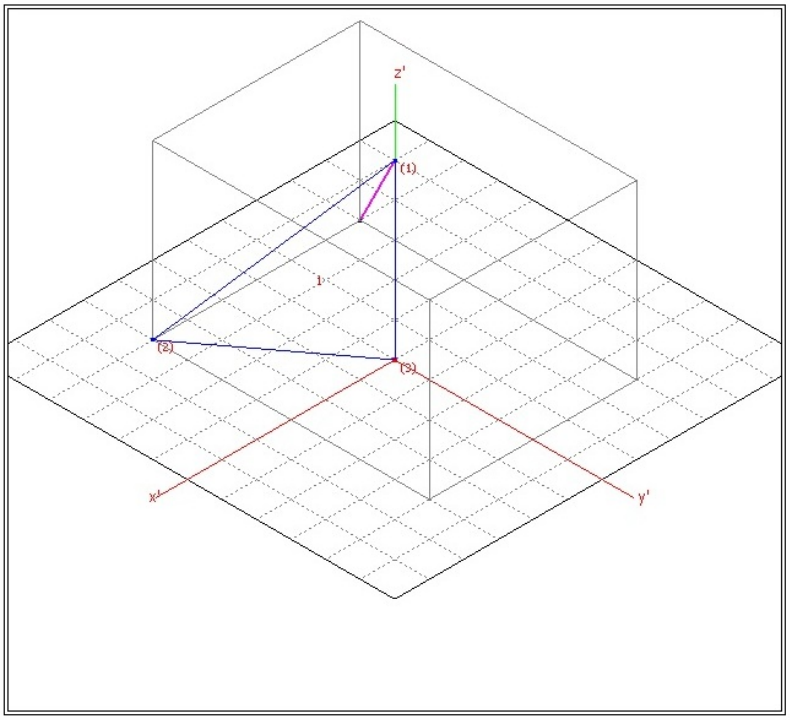


Figure 6.170 Index numbers for Element 2

Build Element 3

17. Get Popup menu in Figure 6.169 by Shift + Right click.
18. Click Add menu.

Draw Index Numbers For Element 3

19. Repeat steps 2 through 11 for Element 3 with MatNo = 3.
20. Figure 6.171 shows index numbers for Element 3.

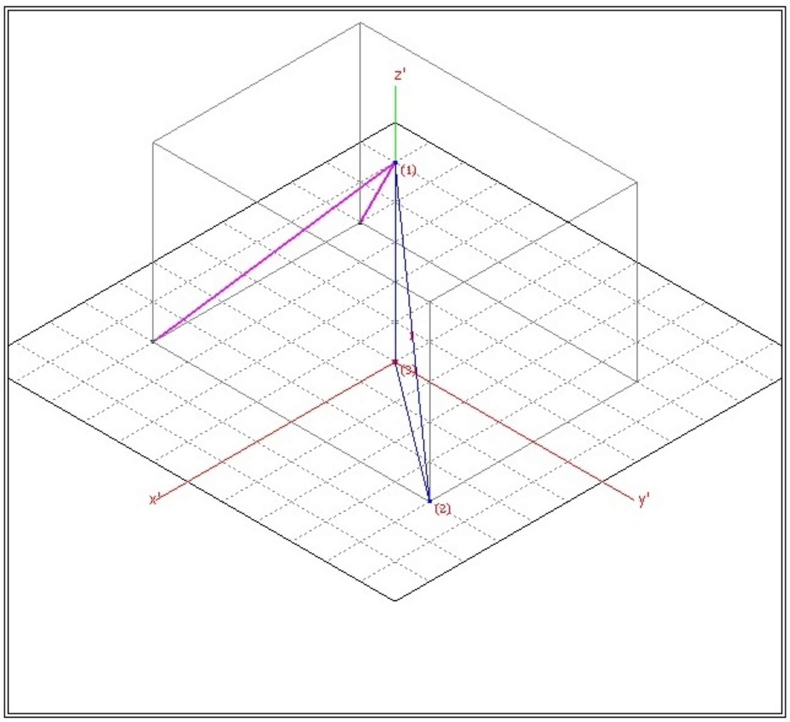


Figure 6.171 Index numbers for Element 3

Build Element 4

21. Get Popup menu in Figure 6.169 by Shift + Right click.
22. Click Add menu.

Draw Index Numbers For Element 4

23. Repeat steps 2 through 11 for Element 4 with MatNo = 4.
24. Figure 6.172 shows index numbers for Element 4.

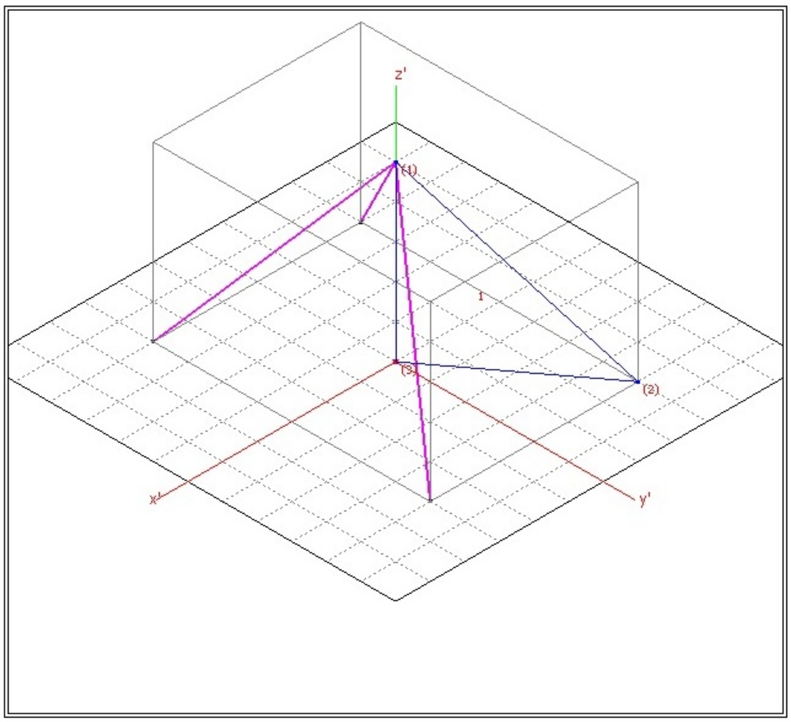


Figure 6.172 Index numbers for Element 4

Step 6: Edit Mesh Title

1. Select Model → Edit Title in Figure 6.168.
2. Type in new title in Mesh Title Editor dialog in Figure 6.173.
3. Click OK.

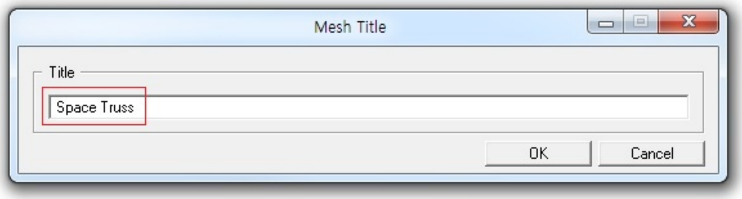


Figure 6.173 Mesh title editor

4. Click Save toolbar in Figure 6.174 and type file name as EX5.

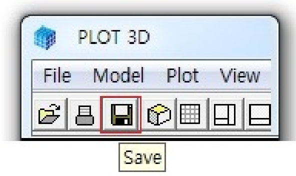


Figure 6.174 Save file toolbar

Step 7: Plot Node and Element Numbers

1. Click Show Numbers toolbar in Figure 6.175.

Figure 6.175 Show numbers toolbar

3. Click Next Numbers toolbar three times in Figure 6.176.

Figure 6.176 Next numbers toolbar

4. Fig 6.177 shows finite element mesh with node & element numbers.

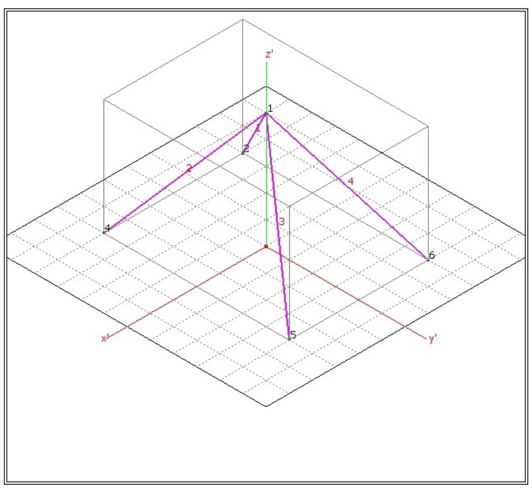


Figure 6.177 Node & element numbers

Step 8: Edit Boundary Codes

1. Select Model → Edit Node in Figure 6.178.

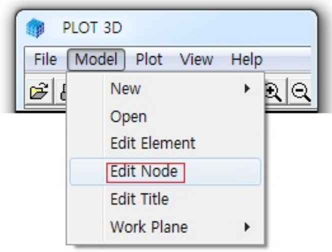


Figure 6.178 Edit node menu

2. Popup menu in Figure 6.179 is displayed by Shift + Right click.
3. Click Boundary menu.

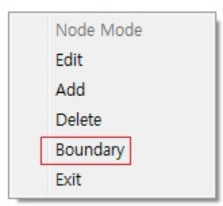


Figure 6.179 Popup menu for edit node

4. Set the boundary codes for Node 1 as shown in Figure 6.180.
5. Click Update button.

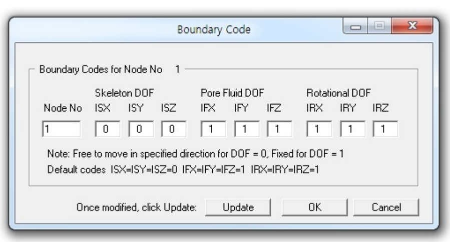


Figure 6.180 Boundary codes for Node 1

6. Set the boundary codes for Node 2 as shown in Figure 6.181.
7. Click Update button.

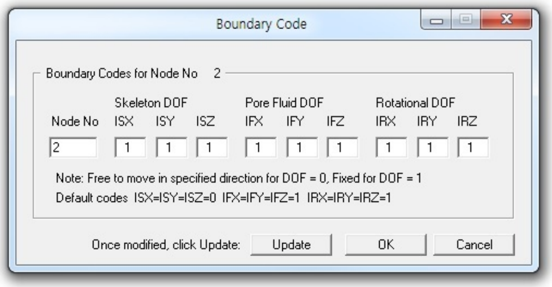


Figure 6.181 Boundary codes for Node 2

8. Repeat steps 6 and 7 for Nodes 3, 4, 5 and 6.
9. Click OK button.
10. Click Save toolbar in Figure 6.174.

Step 9: Plot Skeleton Boundary Codes

1. Click Show Numbers toolbar in Figure 6.175.
2. Click Next Numbers toolbar in Figure 6.176 until skeleton boundary.
3. Figure 6.182 shows skeleton boundary codes for the space truss.

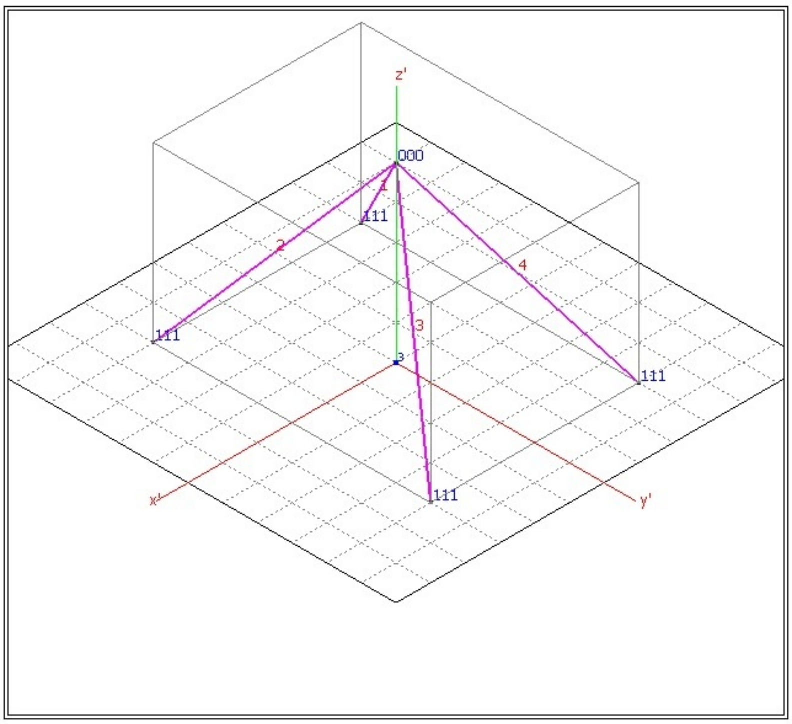


Figure 6.182 Skeleton boundary codes for space truss

PRESMAP
Example Problem

PRESMAP menu includes six Pre-Processing programs: PRESMAP-2D, NATM-2D, CIRCLE-2D, PRESMAP-3D, CROSS-3D, GEN-3D, and PRESMAP-GP. These Pre-Processing programs are mainly used to generate Mesh File described in Section 4.3 of User's Manual. Refer to SMAP-3D User's Manual:

- Section 5 for input parameters for PRESMAP programs.
- Section 3.2.2 for running PRESMAP programs.

7.1 PRESMAP-2D

PRESMAP-2D includes Model 1, 2, 3, and 4. Model 1 is basic pre-processor which can be applied to model various types of problem geometry.

Model 2 is the special pre-processor developed to model near-field around underground openings such as tunnels, culverts, etc. Model 3 is the special pre-processor developed to model triangular and rectangular shape geometry. Model 4 is the useful pre-processor to generate layered embankments having slope.

7.1.1 Model 1

A typical underground tunnel is chosen here to illustrate mesh generations using PRESMAP-2D Model 1 and 2. Figure 7.1 shows geological condition around tunnel consisting of four layers: weathered soil, weathered rock, soft rock, and hard rock. Figure 7.2 shows in detail tunnel cross section including shotcrete and rock bolt dimensions.

For convenience, the tunnel problem geometry is divided into three regions as shown in Figure 7.3; Core, Near-field, and Far-field regions. By symmetry, only right half of the tunnel geometry is considered. Model 1 is used to generate Core and Far-field region meshes. And Model 2 is used to generate Near-field region mesh. Near-field region mesh generation will be explained in the next section. And assembly of Core, Near-field, and Far-field regions will be explained in ADDRGN-2D Example Problems in Section 8.1.

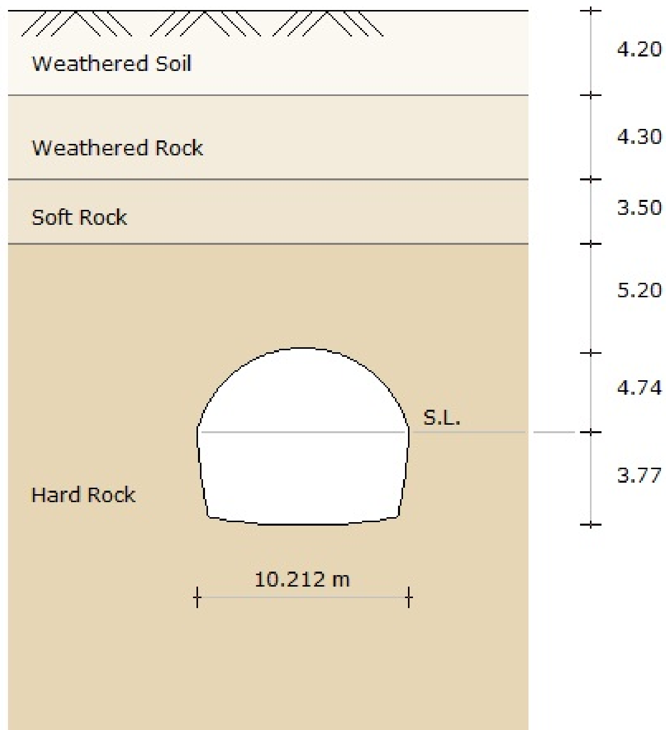
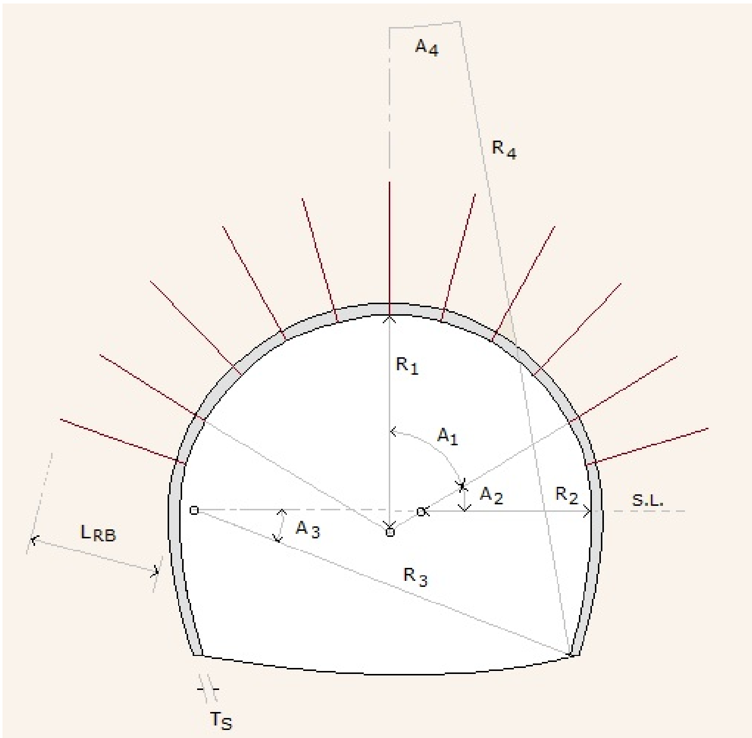


Figure 7.1 Geological condition



R_1 = 5.24 m A_1 = 60°
R_2 = 4.24 m A_2 = 30°
R_3 = 9.86 m A_3 = 19.781°
R_4 = 23.86 m

Number of Rock Bolts (NUMRB)	=	11
Length of Rock Bolts (LRB)	=	3.0 m
Spacing of Rock Bolts (TSPACING)	=	1.2 m
Thickness of Shotcrete (TS)	=	15 Cm
Thickness of Liner (TL)	=	30 Cm
Reinforcing Bar Area (ASI)	=	22 Cm^2
Reinforcing Bar Area (ASO)	=	22 Cm^2

Figure 7.2 PD-2 tunnel section detail

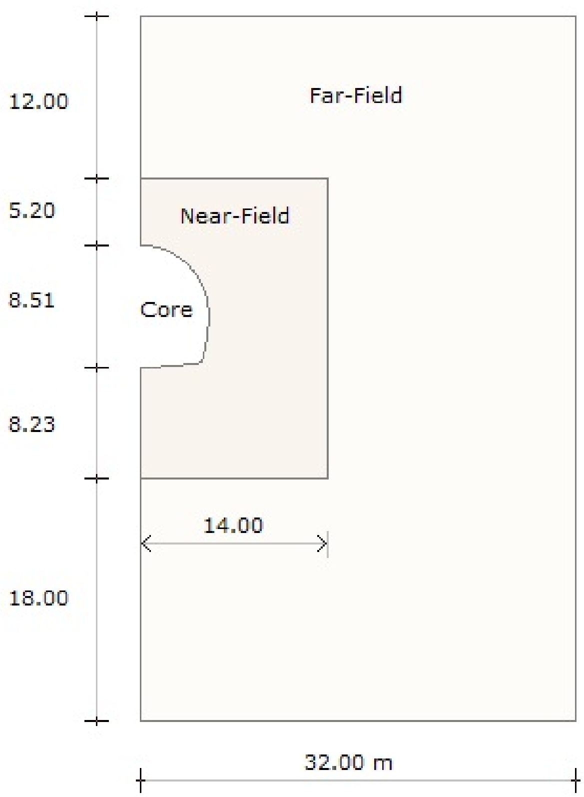


Figure 7.3 Core, Near-field, and Far-field regions

7.1.1.1 Core Region Mesh Generation

Figure 7.4 shows the block diagram for the Core region. Three blocks are used in the horizontal direction (NBX=3) and four blocks in the vertical direction (NBY=4). Block numbers should be in order from top to bottom and left to right. Top 9 blocks (Block numbers 1,2,3,5,6,7,9,10, and 11) represent upper half of tunnel core to be excavated first and bottom 3 blocks (Block numbers 4,8, and 12) represent lower half of tunnel core to be excavated later.

Each block can be consisted of 4 to 9 block nodes depending on whether you can include side and center block nodes. For those blocks facing the tunnel wall of the Core region, side block nodes are included to form the curve. Note that when the side block node is not specified, the straight line will be formed along that side.

Block index should be specified in counterclockwise. For example, the index of Block 4 can be written as I_1=11, I_2 =4, I_3=5, I_4=12, M_5=0, M_6=0, M_7 =7, M_8=0, M_9=0. Next, each block is further divided into elements. For example, Block 4 has 2 elements in the horizontal direction (NDX=2) and 6 elements in the vertical direction (NDY=6). It should be noted that to be compatible, the same number of divisions be specified along the two adjacent blocks. For example, Blocks 4, 8, and 12 have 6 elements in the vertical direction so that the generated elements can share the same nodal points along the boundaries of these blocks.

Since the tunnel is symmetry about y axis, the boundary condition along the y axis is specified as the roller which allows the displacement in the y direction and the boundary condition at all other nodes is specified to be free. And material number.4 representing hard rock is specified for all blocks since the Core region belongs to the hard rock layer as shown in Figure 7.1.

Table 7.1 shows the listing of input file, CORE.Rgn, which has been prepared according to the PRESMAP-2D Model 1 in Section 7.2.1 of User's Manual. Note that the format of the PRESMAP-2D output file is the same as that of Mesh File in SMAP-2D User's Manual. Graphical outputs are shown in Figure 7.5.

Table 7.1 Listing of input file CORE.Rgn

```
* INPUT DATA FOR PRESMAP-2D MODEL 1
* CARD 1.1
  PD-2 CORE REGION GENERATION
* CARD 1.2
* IP
  0
* CARD 1.3
* NBLOCK  NBNODE  NSNEL  CMFAC
    12      30      1      1.0
* CARD 1.4
* NBX  NBY  MIDX  MIDY  NF  NSNODE
   3    4     0     0    1    1
* CARD 2.1
* NODE  X         Y
   1   0.0      4.74
   2   0.0      3.16
   3   0.0      1.58
   4   0.0      0.0
   5   0.0     -3.77
   6   0.684    4.695
   7   0.76    -3.7579
   8   1.356    4.562
   9   1.488    2.819
   10  1.594    1.425
   11  1.702    0.0
   12  1.517   -3.722
   13  2.005    4.341
   14  2.273   -3.662
   15  2.62     4.038
   16  2.9204   2.4907
   17  3.157    1.273
   18  3.404    0.0
   19  3.025   -3.577
   20  3.19     3.66
   21  3.776   -3.47
   22  3.705    3.205
   23  4.157    2.69
   24  4.538    2.12
   25  4.783    1.623
   26  4.962    1.097
   27  5.07     0.5534
   28  5.106    0.0
   29  4.96    -1.693
   30  4.524   -3.337
```

```
* =====================================================
* CARD 3.1
* BLNAME
  BLOCK 1
* CARD 3.2
* IBLNO
    1
* CARD 3.3
* I1  I2  I3  I4  M5  M6  M7  M8  M9
   8   1   2   9   6   0   0   0   0
* CARD 3.4
* IBASE  IB1  IB2  IB3  IB4  IB5  IB6  IB7  IB8
    12    12   13   13   12   12   13   12   12
* CARD 3.5
* MATNO  NDX  NDY   KS   KF
    4     2    2    0    1
* CARD 3.6
* NFSIDE
    0
* =====================================================
  BLOCK 2
  2
  9   2   3   10  0   0   0   0   0
  12  12  13  13  12  12  13  12  12
  4   2   2   0   1
  0
* =====================================================
  BLOCK 3
  3
  10  3   4   11  0   0   0   0   0
  12  12  13  13  12  12  13  12  12
  4   2   2   0   1
  0
* =====================================================
  BLOCK 4
  4  3.337
  11  4   5   12  0   0   7   0   0
  12  12  13  13  12  12  13  12  12
  4   2   6   0   1
  0
* =====================================================
  BLOCK 5
  5
  15  8   9   16  13  0   0   0   0
  12  12  12  12  12  12  12  12  12
  4   2   2   0   1
  0
```

```
* =====================================================
  BLOCK 6
  6
  16 9  10 17 0  0  0  0  0
  12 12 12 12 12 12 12 12 12
  4  2  2  0  1
  0
* =====================================================
  BLOCK 7
  7
  17 10 11 18 0  0  0  0  0
  12 12 12 12 12 12 12 12 12
  4  2  2  0  1
  0
* =====================================================
  BLOCK 6
  6
  16 9  10 17 0  0  0  0  0
  12 12 12 12 12 12 12 12 12
  4  2  2  0  1
  0
* =====================================================
  BLOCK 7
  7
  17 10 11 18 0  0  0  0  0
  12 12 12 12 12 12 12 12 12
  4  2  2  0  1
  0
* =====================================================
  BLOCK 8
  8
  18 11 12 19 0  0  14 0  0
  12 12 12 12 12 12 12 12 12
  4  2  6  0  1
  0
* =====================================================
  BLOCK 9
  9
  22 15 16 24 20 0  0  23 0
  12 12 12 12 12 12 12 12 12
  4  2  2  0  1
  0
```

```
* =====================================================
  BLOCK 10
  10
  24 16 17 26 0  0  0  25 0
  12 12 12 12 12 12 12 12 12
  4  2  2  0  1
  0
* =====================================================
  BLOCK 11
  11
  26 17 18 28 0  0  0  27 0
  12 12 12 12 12 12 12 12 12
  4  2  2  0  1
  0
* =====================================================
  BLOCK 12
  12
  28 18 19 30 0  0  21 29 0
  12 12 12 12 12 12 12 12 12
  4  2  6  0  1
  0
* =====================================================
```

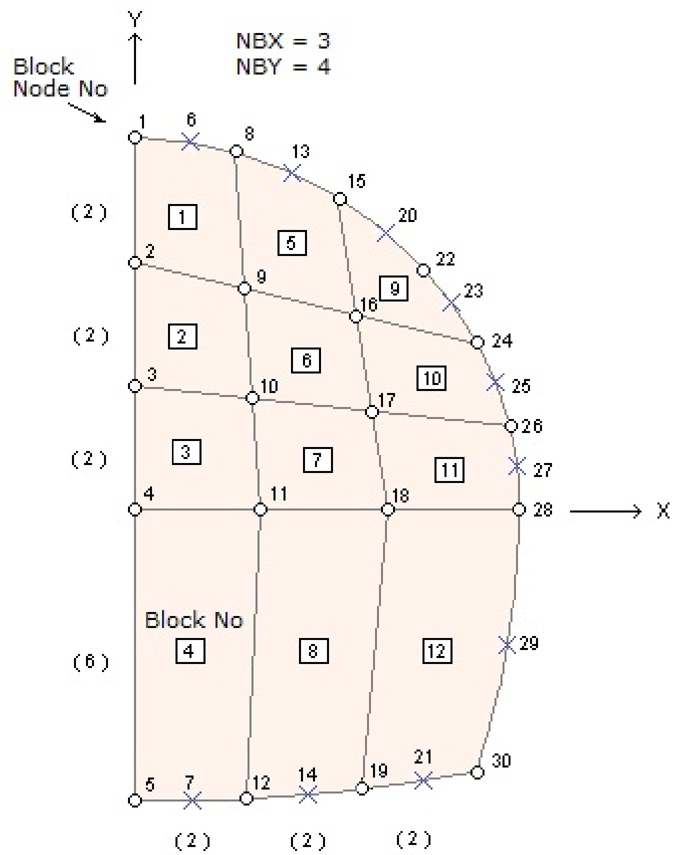


Figure 7.4 Core region block diagram

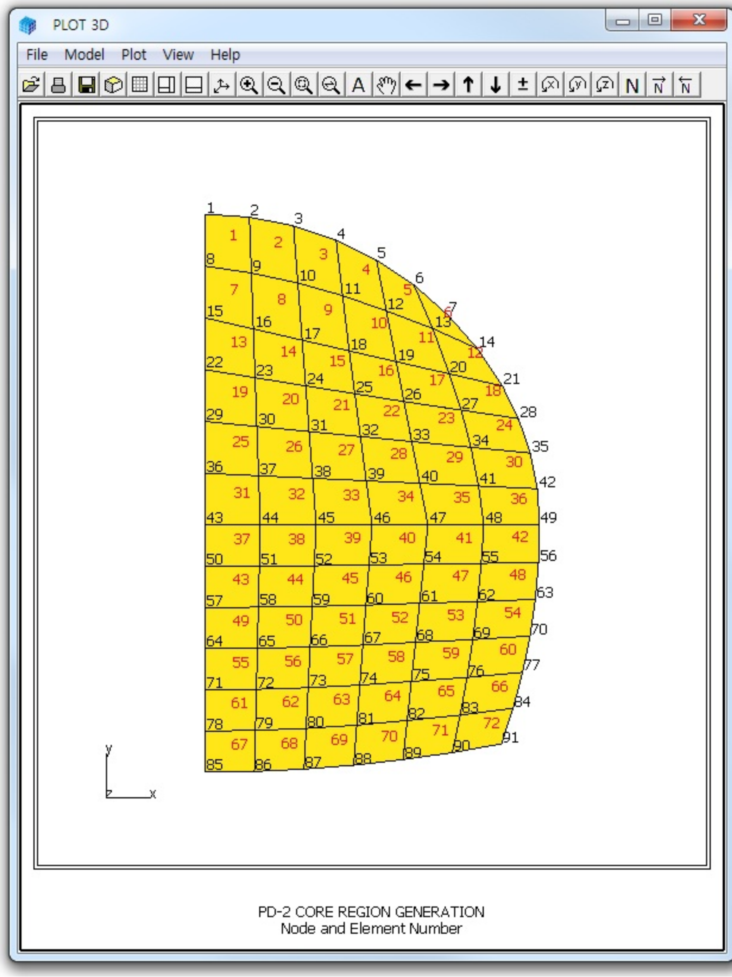


Figure 7.5 Generated element and node numbers for Core region

7.1.1.2 Far-Field Region Mesh Generation

Figure 7.6 shows the block diagram for the Far-field region. Two blocks are used in the horizontal direction (NBX=2) and 6 blocks in the vertical direction (NBY=6). Block numbers 1 and 7 represent weathered soil (MATNO=1). Block numbers 2 and 8 represent weathered rock (MATNO= 2). Block numbers 3 and 9 represent soft rock (MATNO=3). And the rest of blocks represent hard rock (MATNO=4) except Block numbers 4 and 5 (MATNO=0). Note that Block numbers 4 and 5 are void blocks. Elements in this void blocks are not generated in Far-field region, but will be generated in Core and Near-field regions.

You can specify the index of each block as for Core region. Side block nodes are used here to make element sizes bigger as the elements are away from the tunnel core. To simulate plane strain condition at the remote boundary, boundary conditions for the left, right, and bottom are specified as the roller.

Table 7.2 shows the listing of input file, FAR.Rgn, which has been prepared according to the PRESMAP-2D Model 1 in Section 7.2.1 of User's Manual. Generated element and node numbers are shown in Figure 7.7. Note that the Far-field element number starts from 337, considering that there are 336 elements in Core and Near-field regions.

Table 7.2 Listing of input file FAR.Rgn

```
* INPUT DATA FOR PRESMAP-2D MODEL 1
* CARD 1.1
  PD-2 FAR-FIELD REGION GENERATION
* CARD 1.2
* IP
  0
* CARD 1.3
* NBLOCK  NBNODE  NSNEL  CMFAC
   12       31     337    1.0
* CARD 1.4
* NBX  NBY  MIDX  MIDY  NF  NSNODE
   2    6    0     0    1     1
* CARD 2.1
* NODE   X      Y
   1    0.0    21.94
   2    0.0    17.74
   3    0.0    13.44
   4    0.0    9.94
   5    0.0    0.0
   6    0.0   -12.0
   7    0.0   -19.2
   8    0.0   -30.0
   9    14.0   21.94
   10   14.0   17.74
   11   14.0   13.44
   12   14.0   9.94
   13   14.0   0.0
   14   14.0  -12.0
   15   14.0  -19.2
   16   14.0  -30.0
   17   21.2   21.94
   18   21.2   17.74
   19   21.2   13.44
   20   21.2   9.94
   21   21.2   0.0
   22   21.2  -12.0
   23   21.2  -30.0
   24   32.0   21.94
   25   32.0   17.74
   26   32.0   13.44
   27   32.0   9.94
   28   32.0   0.0
   29   32.0  -12.0
   30   32.0  -19.2
   31   32.0  -30.0
```

```
* =====================================================
* CARD 3.1
* BLNAME
  BLOCK 1
* CARD 3.2
* IBLNO
    1
* CARD 3.3
* I1 I2 I3 I4 M5 M6 M7 M8 M9
   9  1  2 10  0  0  0  0  0
* CARD 3.4
* IBASE IB1 IB2 IB3 IB4 IB5 IB6 IB7 IB8
    12   12  13  13  12  12  13  12  12
* CARD 3.5
* MATNO  NDX  NDY  KS   KF
    1     6    1    0    1
* CARD 3.6
* NFSIDE
    0
* =====================================================
  BLOCK 2
  2
  10 2  3  11 0  0  0  0  0
  12 12 13 13 12 12 13 12 12
  2  6  1  0  1
  0
* =====================================================
  BLOCK 3
  3
  11 3  4  12 0  0  0  0  0
  12 12 13 13 12 12 13 12 12
  3  6  2  0  1
  0
* =====================================================
  BLOCK 4
  4
  12 4  5  13 0  0  0  0  0
  12 12 13 13 12 12 13 12 12
  0  6  6  0  1
  0
* =====================================================
  BLOCK 5
  5
  13 5  6  14 0  0  0  0  0
  12 12 13 13 12 12 13 12 12
  0  6  6  0  1
  0
```

```
* =====================================================
  BLOCK 6
  6
  14 6  8  16 0  7  0  15 0
  12 12 13 15 14 12 13 14 12
  4  6  4  0  1
  0
* =====================================================
  BLOCK 7
  7
  24 9  10 25 17 0  18 0  0
  12 13 12 12 13 12 12 12 13
  1  4  1  0  1
  0
* =====================================================
  BLOCK 8
  8
  25 10 11 26 18 0  19 0  0
  12 13 12 12 13 12 12 12 13
  2  4  1  0  1
  0
* =====================================================
  BLOCK 9
  9
  26 11 12 27 19 0  20 0  0
  12 13 12 12 13 12 12 12 13
  3  4  2  0  1
  0
* =====================================================
  BLOCK 10
  10
  27 12 13 28 20 0  21 0  0
  12 13 12 12 13 12 12 12 13
  4  4  6  0  1
  0
* =====================================================
  BLOCK 11
  11
  28 13 14 29 21 0  22 0  0
  12 13 12 12 13 12 12 12 13
  4  4  6  0  1
  0
```

```
* ====================================================
  BLOCK 12
  12
  29 14 16 31 22 15 23 30 0
  12 13 12 14 15 12 12 14 13
  4  4  4  0  1
  0
* ====================================================
* END OF DATA
```

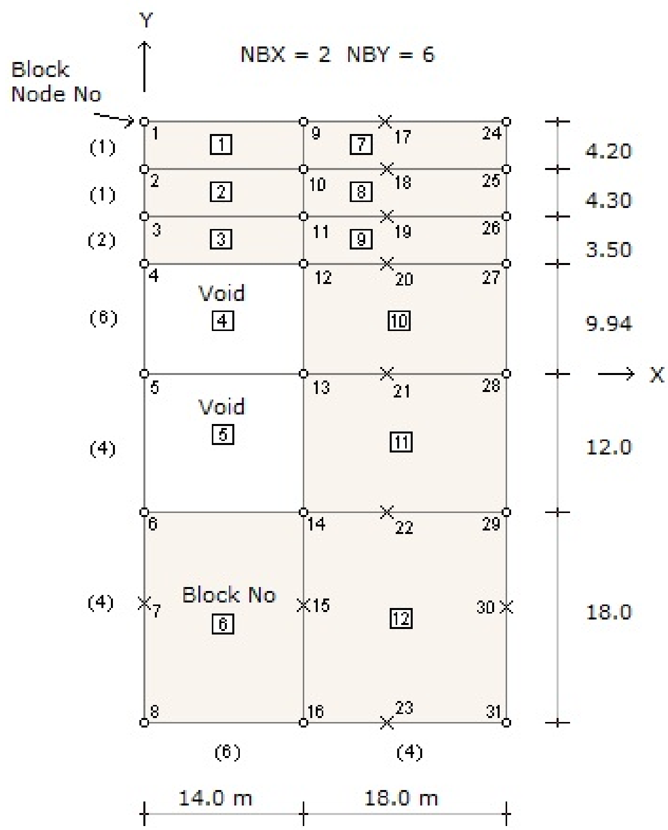


Figure 7.6 Far-field region block diagram

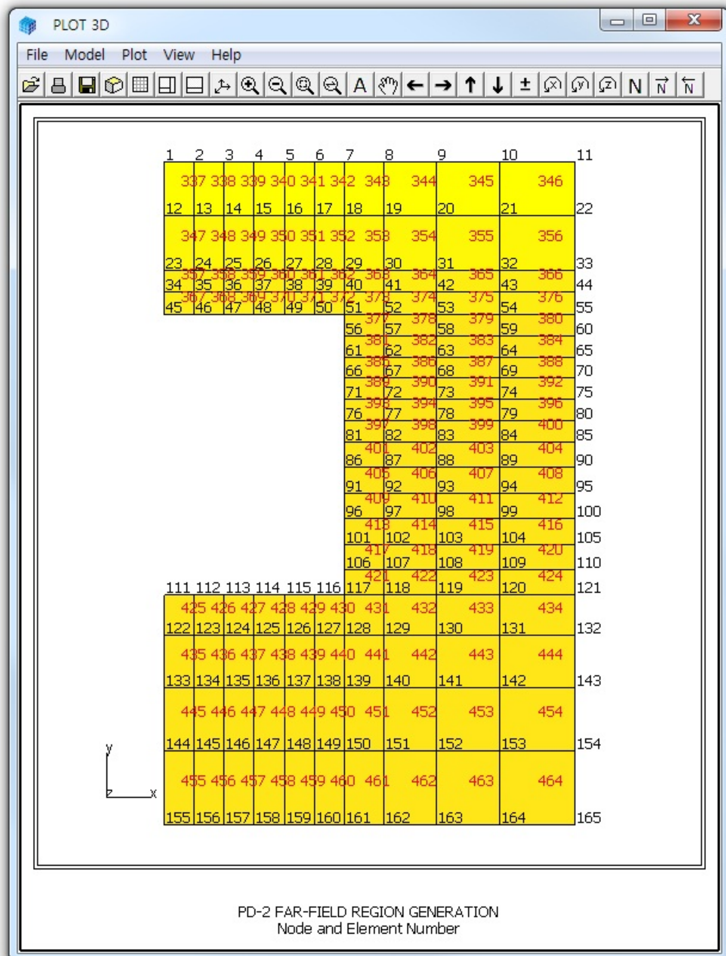


Figure 7.7 Generated element and node numbers for Far-field region

7.1.2 Model 2

Model 2 is the special pre-processor developed to model Near-field region around the underground openings. The Near-field region shown in Figure 7.3 is taken here as an example problem.

As shown in Figure 7.8, eight subregions are used to construct the Near-field region. And each subregion consists of three blocks. Then each block is further divided in radial and tangential directions. For example, Block number 5 in Subregion 2 has 5 elements in radial direction and 6 elements in the tangential direction. Note that element sizes in the third block increase gradually in the radial direction. Parameters specific to each subregion are tabulated in Table 7.3.

Table 7.4 shows the listing of input file, NEAR.Rgn, which has been prepared according to the PRESMAP-2D Model 2 in Section 7.2.2 of User's Manual. Generated element mesh is shown in Figure 7.9.

Table 7.3 Parameters specific in Near-field region

NSUBR = 8 NDRF = 2 NDRS = 5 NDRT = 4
DRF = 0.15 m DRS = 2.85 m

Subregion	ISBTYPE	LSFTYPE	NSEG
1	1	1	6
2	1	1	6
3	0	1	2
4	0	1	2
5	0	1	2
6	0	1	2
7	0	1	2
8	0	1	2

Global block numbers are in order from surface

to outer edge and counterclockwise.

Local block numbers in each subregion are in order
from surface to outer edge.

Example : In Subregion 2,
First block = 4 , Second block = 5, Third block = 6

Table 7.4 Listing of input file NEAR.Rgn

```
* INPUT DATA FOR PRESMAP-2D MODEL 2
* CARD 1.1
  PD-2 NEAR-FIELD MESH GENERATION
* CARD 1.2
* IP
  0
* CARD 1.3
* NSNEL  NSNODE  NF  CMFAC
   73      67    1   1.0
* CARD 1.4
* NSURB  NDRF  NDRS  NDRT  DRF   DRS
    8     2     5     4    0.15  2.85
* ====================================================
* CARD 2.1
* SUBNAME
  SUBREGION 1
* CARD 2.2
* ISUBNO
     1
* CARD 2.3
* ISBTYPE  LSFTYPE  NSEC
     1        1       6
* CARD 2.4.2 (LSFTYPE = 1)
* R      Xo   Yo     TA    TB
  23.86  0.0  20.09  270.  280.93
* (ISBTYPE = 1)
* CARD 2.5.3
* Xc    Yc    Xd    Yd
  0.0   -12.  14.0  -12.
* CARD 2.6
* IBASE1  IBASE2  IBASE3
    12      12      12
* IBb  IBa  IBc  IBd  IBab  IBac  IBcd  Ibbd
   12   13   13   12   12    13    12    12
* CARD 2.7
* MATNO1  KS1   KF1
      4    0     1
* MATNO2  KS2   KF2
      4    0     1
* MATNO3  KS3   KF3
      4    0     1
* CARD
* NFSIDE
      0
```

```
* =====================================================
      SUBREGION 2
      2
      1  1  6
      9.86  -4.754  0.0  340.22  360.
      14.0  -12.  14.0   0.0
      12 12 12
      12 12 12 12 12 12 12 12
      4  0  1
      4  0  1
      4  0  1
      0
* =====================================================
      SUBREGION 3
      3
      0  1  2
      4.24  0.866  0.0  0.0  15.0
      1
      14.0
      0
      14.0  3.31
      12 12 12
      12 12 12 12 12 12 12 12
      4  0  1
      4  0  1
      4  0  1
      0
* =====================================================
      SUBREGION 4
      4
      0  1  2
      4.24  0.866  0.0  15.0  30.0
      0
      14.0  3.31
      0
      14.0  6.63
      12 12 12
      12 12 12 12 12 12 12 12
      4  0  1
      4  0  1
      4  0  1
      0
```

```
* =====================================================
      SUBREGION 5
      5
      0   1   2
      5.24   0.0   -0.5   30.0   45.0
      0
      14.0   6.63
      0
      14.0   9.94
      12 12 12
      12 12 12 12 12 12 12 12
      4   0   1
      4   0   1
      4   0   1
      0
* =====================================================
      SUBREGION 6
      6
      0   1   2
      5.24   0.0   -0.5   45.0   60.0
      0
      14.0   9.94
      0
      9.33   9.94
      12 12 12
      12 12 12 12 12 12 12 12
      4   0   1
      4   0   1
      4   0   1
      0
* =====================================================
      SUBREGION 7
      7
      0   1   2
      5.24   0.0   -0.5   60.   75.0
      0
      9.33   9.94
      0
      4.67   9.94
      12 12 12
      12 12 12 12 12 12 12 12
      4   0   1
      4   0   1
      4   0   1
      0
```

```
* =====================================================
      SUBREGION 8
      8
      0  1  2
      5.24  0.0  -0.5  75.0  90.0
      0
      4.67  9.94
      0
      0.0   9.94
      12 12 12
      13 12 12 13 12 12 12 13
      4  0  1
      4  0  1
      4  0  1
      0
* =====================================================
```

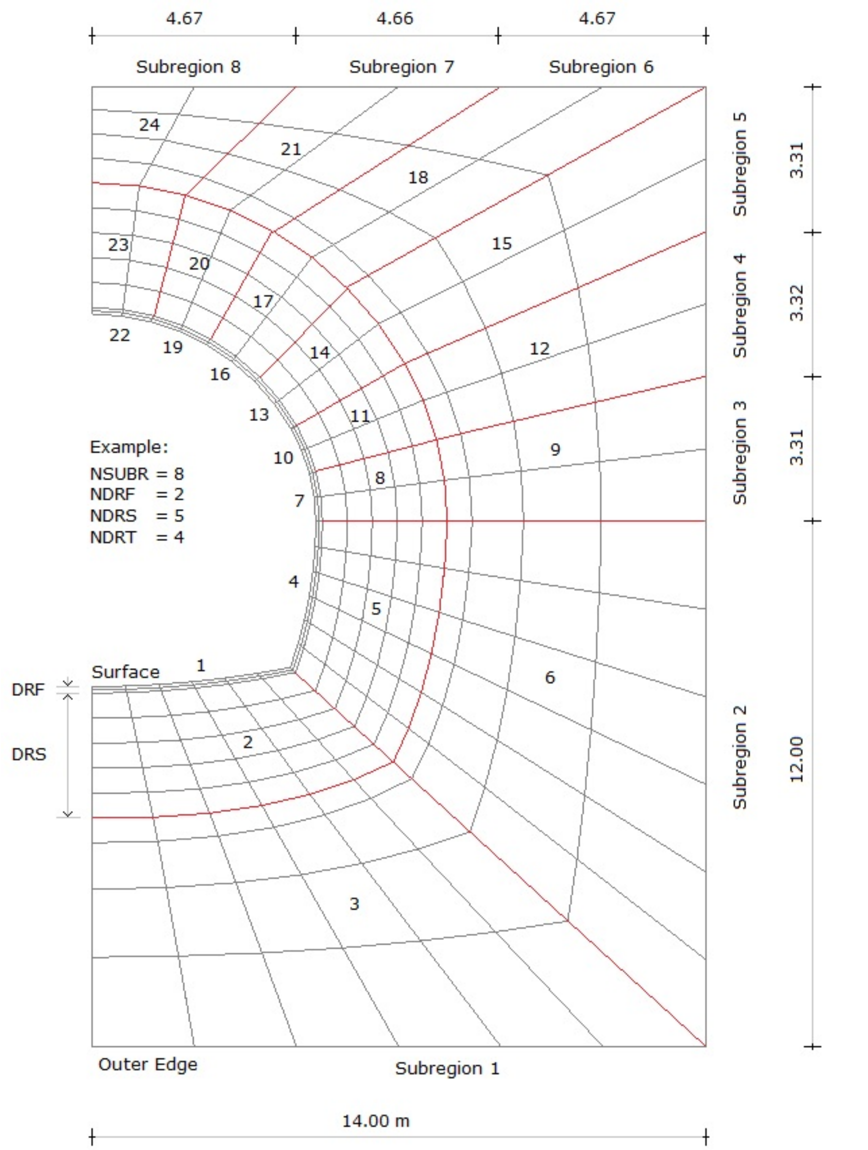


Figure 7.8 Subregions in Near-field region

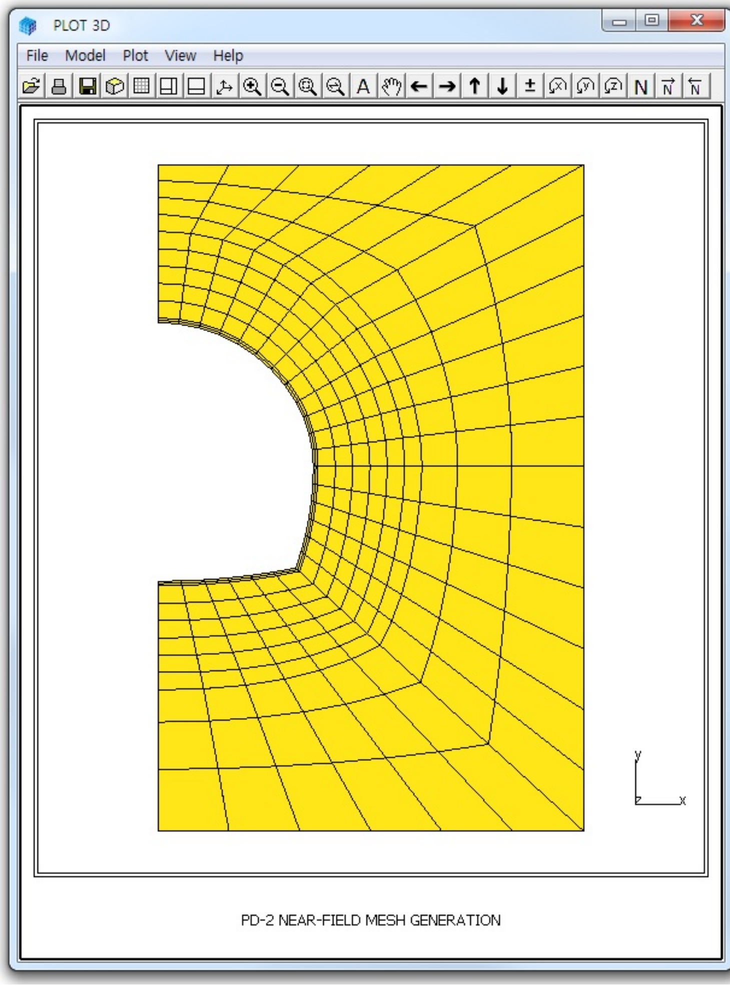


Figure 7.9 Generated mesh for Near-field region

7.1.3 Model 3

Model 3 is a useful pre-processor to generate triangular or rectangular meshes. It is much easier to use compared to Models 1 and 2.
But you have to specify the boundary codes manually.

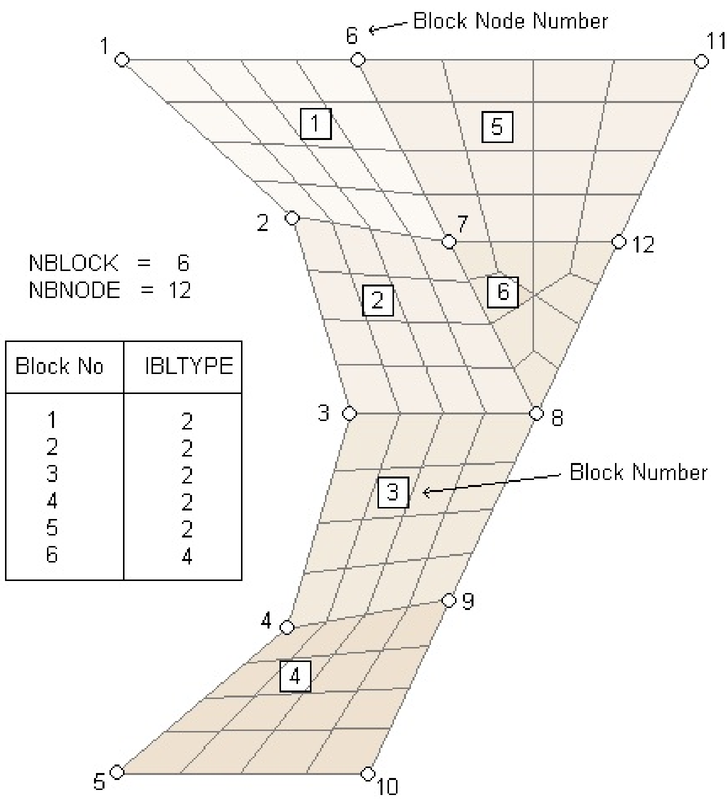
Figure 7.10 shows block diagram for Model 3 example problem.
Block numbers 1 to 5 are 4 x 4 rectangular shape and Block number 6 is the 9-element triangular shape.

Table 7.5 shows the listing of input file, GM3.Rgn, which has been prepared according to the PRESMAP-2D Model 3 in Section 7.2.3 of User's Manual. Generated element and node numbers are shown in Figure 7.11.

Table 7.5 Listing of input file GM3.Rgn

```
* INPUT DATA FOR PRESMAP-2D MODEL 3
* CARD 1.1
  MESH GENERATION SURROUNDING PIPE ( GM3 )
* CARD 1.2
* IP
  0
* CARD 1.3
* NBLOCK  NBNODE  NSNEL  NSNODE  CMFAC
     6      12     171    1       1.0
* CARD 2.1
* NODE     X           Y
   1  .324920E+02  .100000E+03
   2  .809020E+02  .587790E+02
   3  .100000E+03  .000000E+00
   4  .809020E+02 -.587790E+02
   5  .324920E+02 -.100000E+03
   6   100.         100.
   7   125.         50.
   8   150.         0.
   9   125.        -50.
   10  100.        -100.
   11  200.         100.
   12  175.         50.
* =======================================================
* CARD 3.1
* =======================================================
* IBLNO  IBLTYPE  MATNO  KS  KF
    1       2       2     0   1
* FOR IBLTYPE = 2
* I1 I2 I3 I4 M5 M6 M7 M8 M9 M10 M11 M12 M13 M14 M15 M16
   6  1  2  7  0  0  0  0  0   0   0   0   0   0   0   0
* =======================================================
* IBLNO  IBLTYPE  MATNO  KS  KF
    2       2       2     0   1
* FOR IBLTYPE = 2
* I1 I2 I3 I4 M5 M6 M7 M8 M9 M10 M11 M12 M13 M14 M15 M16
   7  2  3  8  0  0  0  0  0   0   0   0   0   0   0   0
* =======================================================
* IBLNO  IBLTYPE  MATNO  KS  KF
    3       2       2     0   1
* FOR IBLTYPE = 2
* I1 I2 I3 I4 M5 M6 M7 M8 M9 M10 M11 M12 M13 M14 M15 M16
   8  3  4  9  0  0  0  0  0   0   0   0   0   0   0   0
```

```
* =======================================================
* IBLNO  IBLTYPE  MATNO  KS  KF
    4       2       2     0   1
* FOR IBLTYPE = 2
* I1 I2 I3 I4 M5 M6 M7 M8 M9 M10 M11 M12 M13 M14 M15 M16
   9  4  5 10  0  0  0  0  0   0   0   0   0   0   0   0
* =======================================================
* IBLNO  IBLTYPE  MATNO  KS  KF
    5       2       2     0   1
* FOR IBLTYPE = 2
* I1 I2 I3 I4 M5 M6 M7 M8 M9 M10 M11 M12 M13 M14 M15 M16
  11  6  7 12  0  0  0  0  0   0   0   0   0   0   0   0
* =======================================================
* IBLNO  IBLTYPE  MATNO  KS  KF
    6       4       2     0   1
* FOR IBLTYPE = 2
* I1 I2 I3 M4 M5 M6 M7 M8 M9 M10 M11 M12
   7  8 12  0  0  0  0  0  0   0   0   0
* =======================================================
```



Block No	IBLTYPE
1	2
2	2
3	2
4	2
5	2
6	4

Figure 7.10 Block diagram for Model 3 example problem

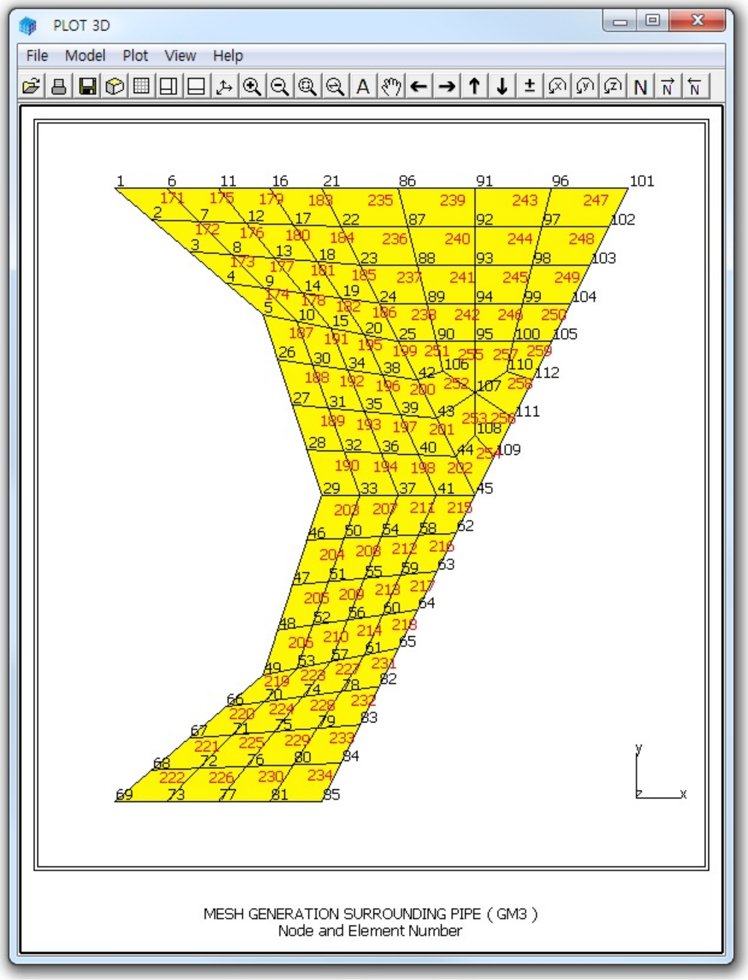


Figure 7.11 Generated element and node numbers for Model 3 example problem

7.1.4 Model 4

Model 4 is a useful pre-processor to generate horizontally layered dams or embankments. It is easy to use but the boundary codes should be specified manually.

As Model 4 example problem, an embankment with 3 layers is considered. Table 7.6 shows the listing of input file, GM4.Rgn, which has been prepared according to the PRESMAP-2D Model 4 in Section 7.2.4 of User's Manual. Generated element and node numbers are shown in Figure 7.12.

Table 7.6 Listing of input file GM4.Rgn

```
* CARD 1.1
* TITLE
  EXAMPLE PROBLEM FOR PRESMAP-2D MODEL 4
* CARD 1.2
* NLAYER  NDIV  ITRANGL
   3       3     1
* CARD 1.3
* NSNEL  NSNODE  CMFAC
   1      1       1.0
* CARD 2.1
* XB1  YB1  YB2  XB2
  0.0  3.0  0.0  12.
* CARD 3.1
* MATNO  KS  KF
   3     0   1
* END OF DATA
```

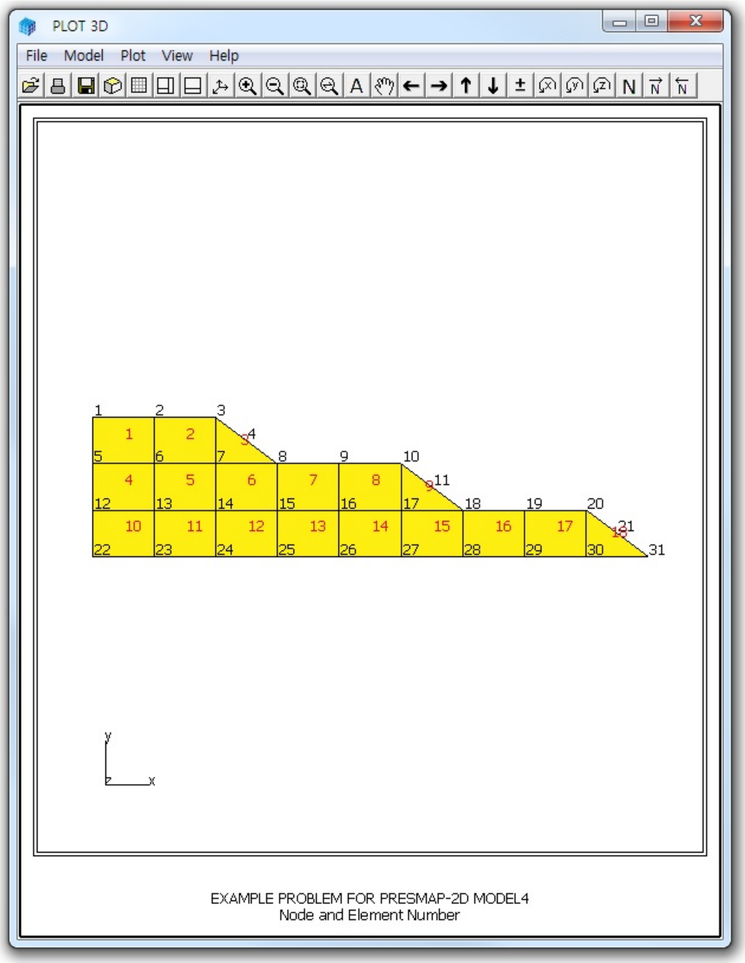


Figure 7.12 Generated element and node numbers for Model 4 example problem

7.2 NATM-2D

NATM-2D is the special pre-processing program to generate automatically two-dimensional finite element meshes and boundary conditions for NATM tunnels. NATM-2D has four different models:

Model 1	Single Tunnel (Half Section)
Model 2	Single Tunnel (Full Section)
Model 3	Two Tunnel (Symmetric Section)
Model 4	Two Tunnel (Unsymmetric Section)

Once you have executed NATM-2D, you will obtain following files:

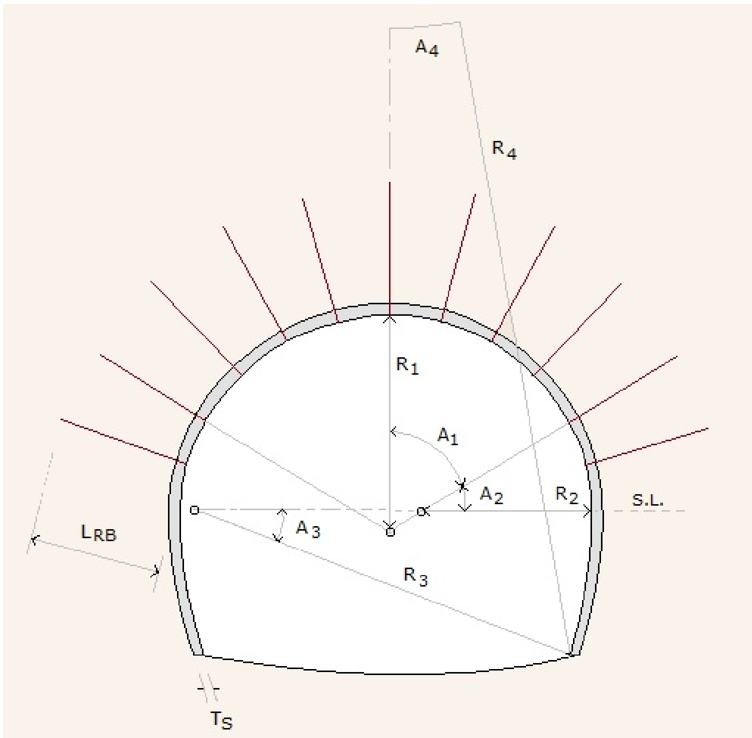
Output File	Mesh File including all elements (Continuum, Beam, and Truss). Output File is the user specified name.
BEAM.Dat	Mesh File including only beam elements.
TRUSS.Dat	Mesh File including only truss elements.
AD.Dat	Card Group 8 in Main File representing default element activities for upper and lower parts of Core, Shotcrete, and Rock Bolt including Joint and Lining elements.
LINING.Dat	Mesh File for Beam-Spring Lining Analysis. This file will be generated only for ILNCOUPL=1.

A typical PD2 tunnel shape is chosen here to illustrate mesh generation using NATM-2D as shown in Figure 7.13. For each model, we will present:

- Listing of input file
- Schematic tunnel section view
- Graphical output of finite element mesh

Table 7.7 Listing of input file PD2-1.Dat

```
* CARD 1.1
* TITLE
  NATM-2D MODEL 1 EXAMPLE PROBLEM
* CARD 1.2
* IUNIT
   2
* CARD 1.3
* MODEL  IGEN  IEXMESH   ILNCOUPL
   1      0     0                0
* CARD 2.1
* HT    HL   W     DELTAX  DELTAX  NDYMAX
  21.94 30.  20.  2.0     2.0     40
* CARD 3.1
* NLAYER
   4
* CARD 3.2
* LAYERNO   H     KF
   1        4.2   1
   2        4.3   1
   3        3.5   1
   4        39.94 1
* CARD 4.1
* R1    A1    R2    A2  R3    A3     R4    GR    GA
  5.24  60.   4.24  30. 9.86  19.781 23.86 1.0   0.5
* CARD 4.2
* INVSHOT  TS
  0        0.3
* CARD 4.3
* NUMRB  LRB    LSPACING  TSPACING  NSRB
  11     3.0    0.8       1.2        2
* CARD 5.1
* LDTYPE  DGW  GAMAW
  1       2.0  1.0
* END OF DATA
```



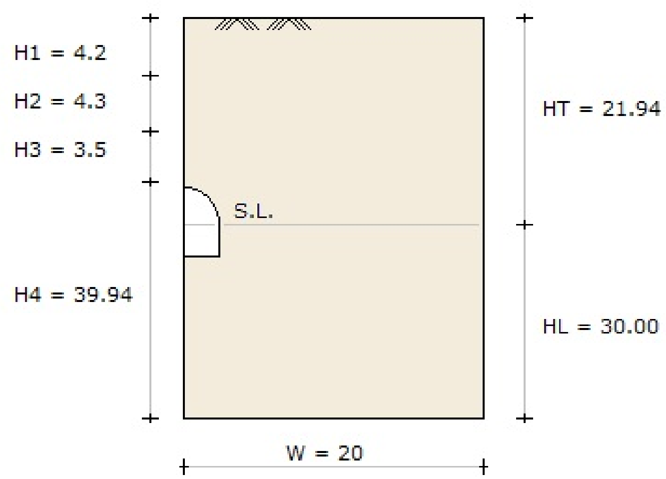
R_1 = 5.24 M
R_2 = 4.24 M
R_3 = 9.86 M
R_4 = 23.86 M

A_1 = 60°
A_2 = 30°
A_3 = 19.781°

Number of Rock Bolts (NUMRB)	=	11
Length of Rock Bolts (LRB)	=	3.0 M
Spacing of Rock Bolts (TSPACING)	=	1.2 M
Thickness of Shotcrete (TS)	=	15 Cm
Thickness of Liner (TL)	=	30 Cm
Reinforcing Bar Area (ASI)	=	22 Cm^2
Reinforcing Bar Area (ASO)	=	22 Cm^2

Figure 7.13 Tunnel dimensions used for example problem

MODEL=1 Single Tunnel (Half Section)



DELTAX = 2.0 DELTAY = 2.0 NDYMAX = 40

Figure 7.14 Schematic tunnel section view for Model 1 example

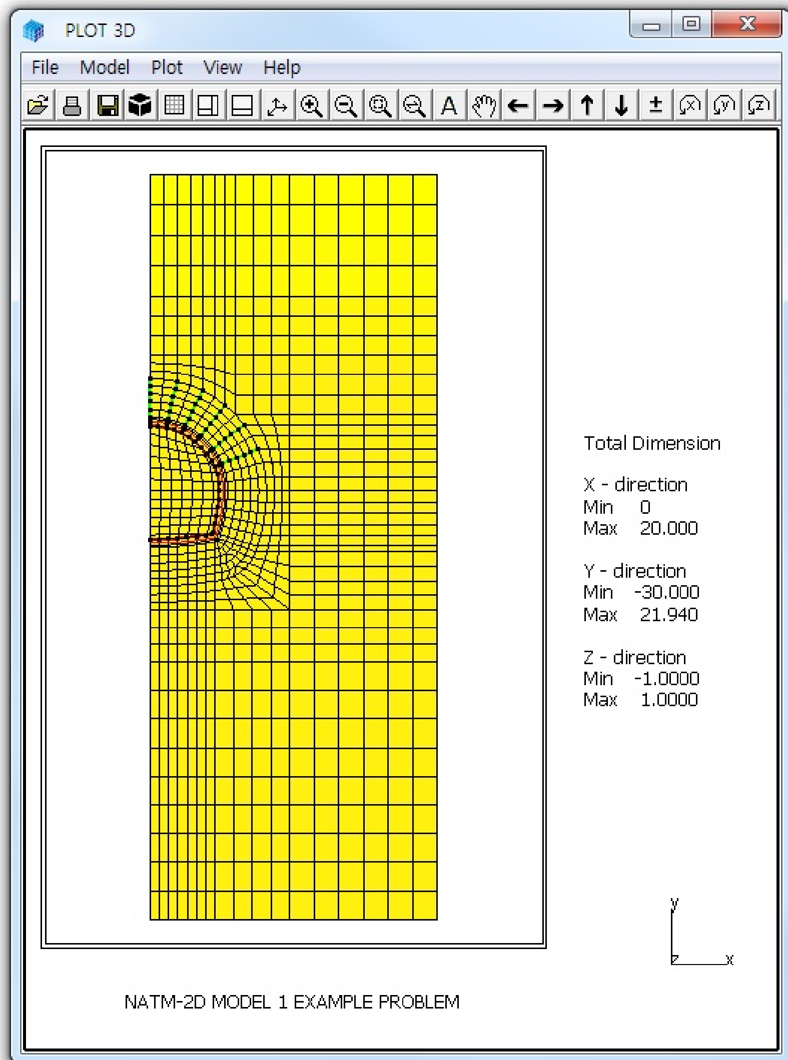


Figure 7.15 Generated finite element mesh for Model 1 example

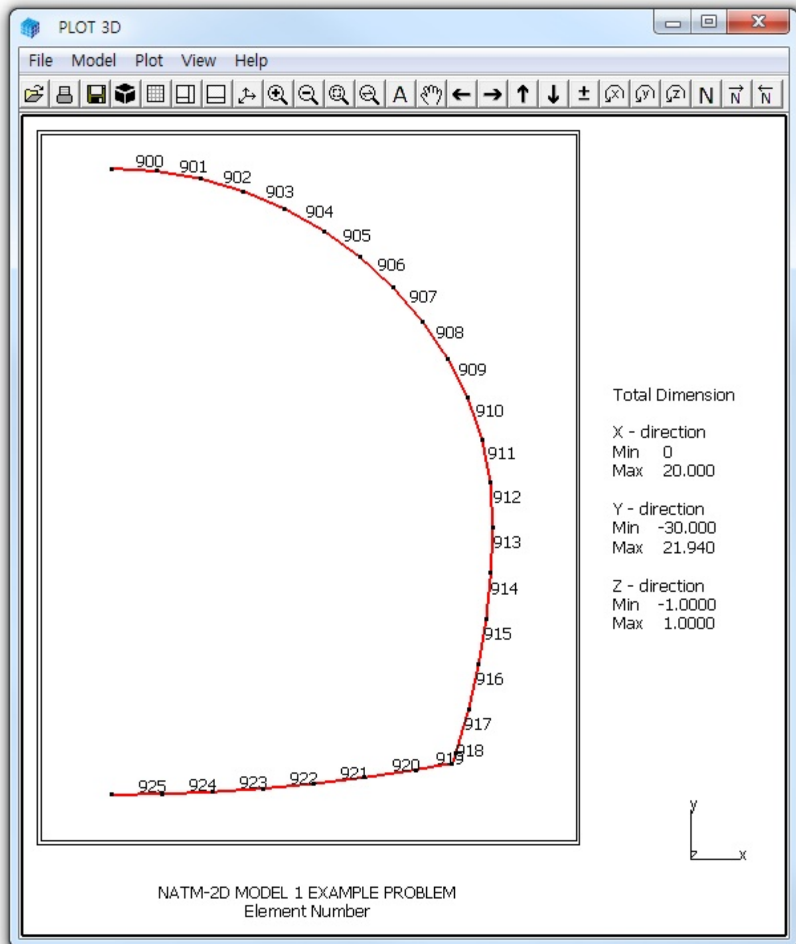


Figure 7.16 Generated beam element number for Model 1 example

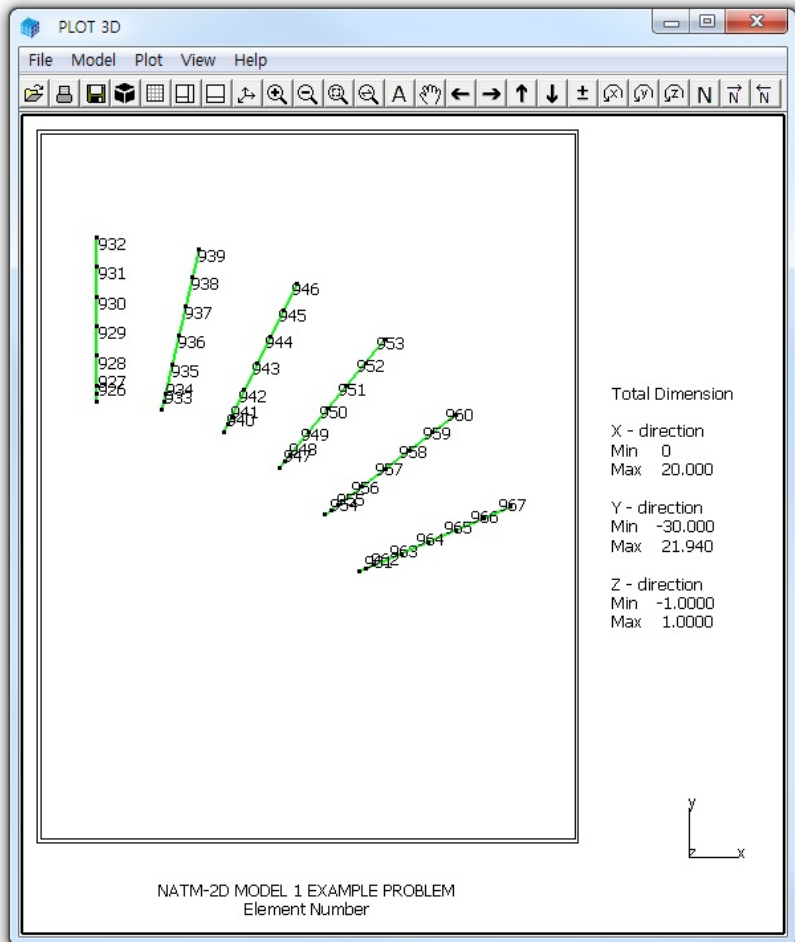
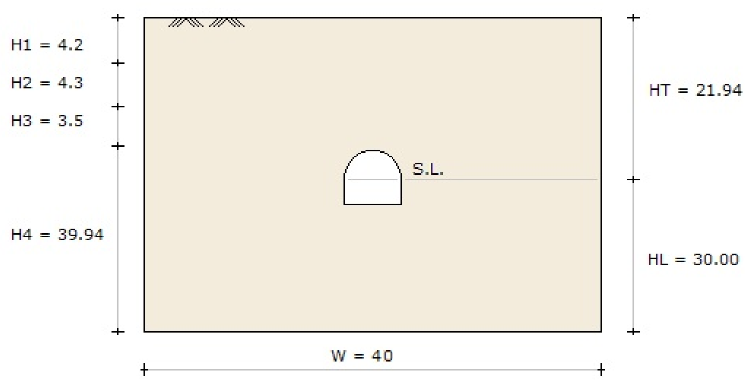


Figure 7.17 Generated truss element number for Model 1 example

Table 7.8 Listing of input file PD2-2.Dat

```
* CARD 1.1
* TITLE
  NATM-2D MODEL 2 EXAMPLE PROBLEM
* CARD 1.2
* IUNIT
   2
* CARD 1.3
* MODEL  IGEN  IEXMESH      ILNCOUPL
   2     0     0            0
* CARD 2.1
* HT    HL   W    DELTAX  DELTAX  NDYMAX
  21.94 30.  40.  2.0     2.0     40
* CARD 3.1
* NLAYER
   4
* CARD 3.2
* LAYERNO   H     KF
   1        4.2   1
   2        4.3   1
   3        3.5   1
   4        39.94 1
* CARD 4.1
* R1    A1    R2    A2  R3    A3     R4    GR    GA
  5.24  60.   4.24  30. 9.86  19.781 23.86 1.0   0.5
* CARD 4.2
* INVSHOT  TS
  0        0.3
* CARD 4.3
* NUMRB  LRB    LSPACING  TSPACING  NSRB
  11     3.0    0.8       1.2        2
* CARD 5.1
* LDTYPE  DGW  GAMAW
  1       2.0  1.0
* END OF DATA
```

MODEL=2 Single Tunnel (Full Section)



DELTAX = 2.0 DELTAY = 2.0 NDYMAX = 40

Figure 7.18 Schematic tunnel section view for Model 2 example

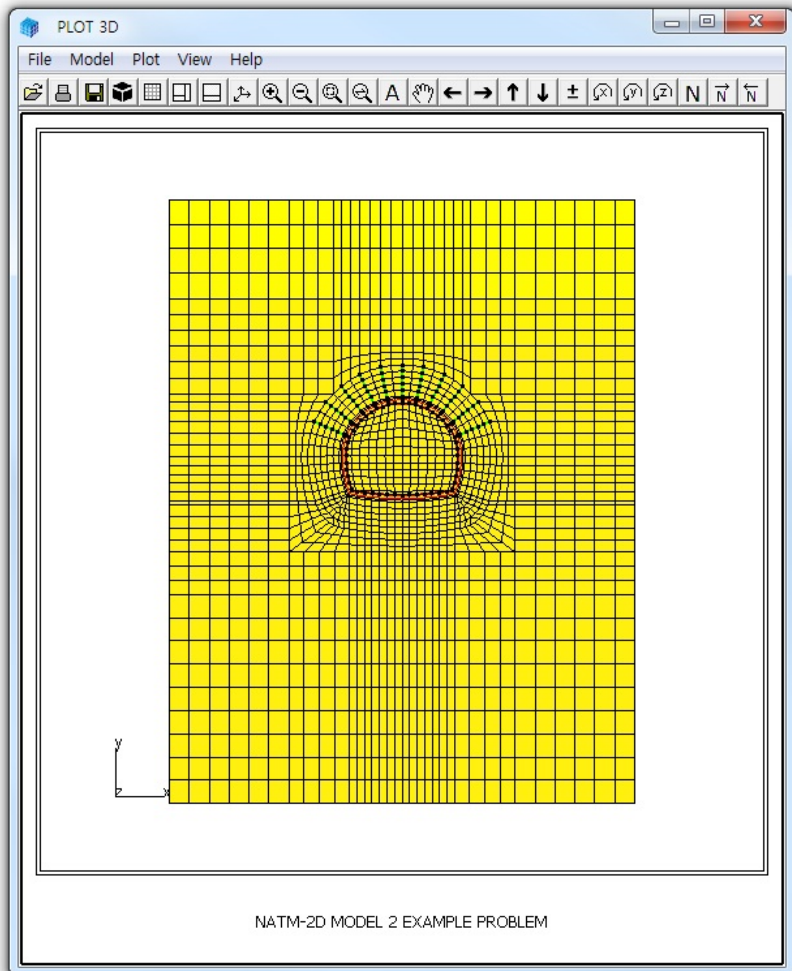
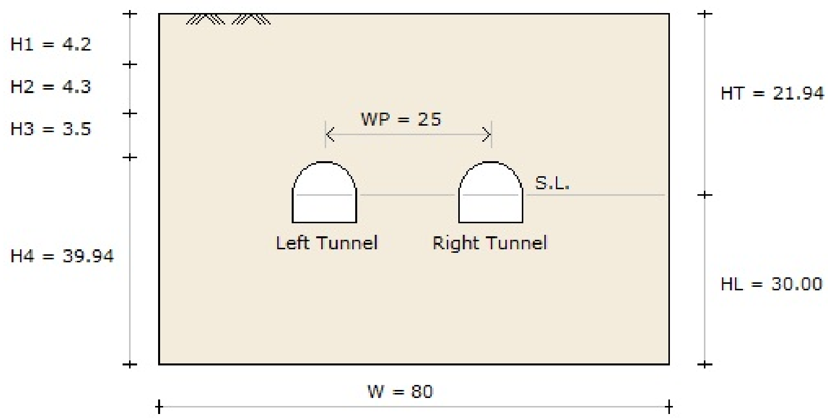


Figure 7.19 Generated finite element mesh for Model 2 example

Table 7.9 Listing of input file PD2-3.Dat

```
* CARD 1.1
* TITLE
  NATM-2D MODEL 3 EXAMPLE PROBLEM
* CARD 1.2
* IUNIT
   2
* CARD 1.3
* MODEL  IGEN  IEXMESH   ILNCOUPL
   3     0     0        0
* CARD 2.1
* HT    HL    W    WP    DELTAX  DELTAY  NDYMAX
  21.94 30.  80.  25.   2.0     2.0     40
* CARD 3.1
* NLAYER
   4
* CARD 3.2
* LAYERNO   H     KF
   1        4.2   1
   2        4.3   1
   3        3.5   1
   4        39.94 1
* CARD 4.1
* R1    A1    R2    A2  R3    A3     R4    GR    GA
  5.24  60.   4.24  30. 9.86  19.781 23.86 1.0   0.5
* CARD 4.2
* INVSHOT  TS
  0        0.3
* CARD 4.3
* NUMRB  LRB    LSPACING  TSPACING  NSRB
  11     3.0    0.8       1.2        2
* CARD 5.1
* LDTYPE  DGW  GAMAW
  1       2.0  1.0
* END OF DATA
```

MODEL=3 Two Tunnel (Symmetric Section)



DELTAX = 2.0   DELTAY = 2.0   NDYMAX = 40

Figure 7.20 Schematic tunnel section view for Model 3 example

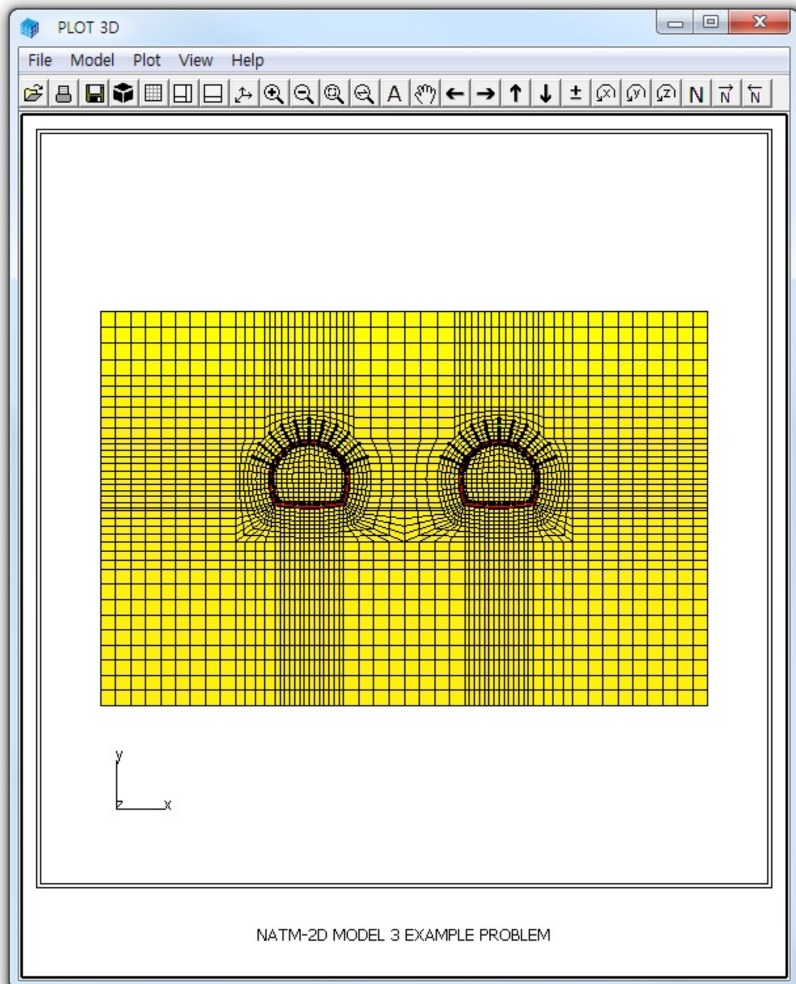
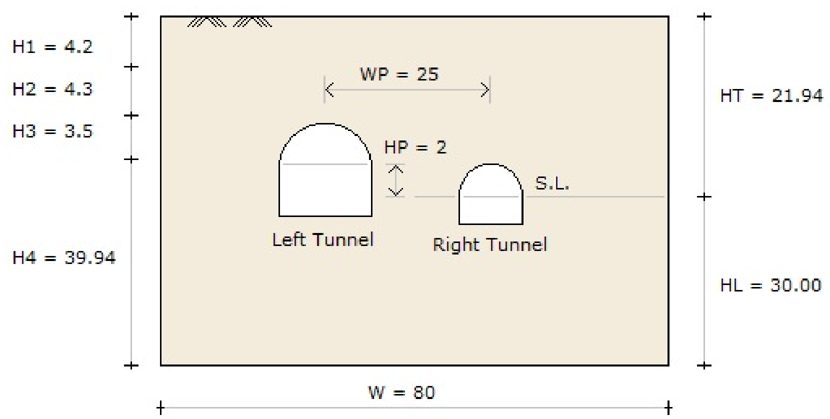


Figure 7.21 Generated finite element mesh for Model 3 example

Table 7.10 Listing of input file PD2-4.Dat

```
* CARD 1.1
* TITLE
  NATM-2D MODEL 4 EXAMPLE PROBLEM
* CARD 1.2
* IUNIT
   2
* CARD 1.3
* MODEL  IGEN  IEXMESH   ILNCOUPL
   4      0     0        0
* CARD 2.1
* HT    HL   W    WP    HP      DELTAX  DELTAY  NDYMAX
  21.94 30.  80.  25.   2.0     2.0     2.0     40
* CARD 3.1
* NLAYER
   4
* CARD 3.2
* LAYERNO   H     KF
   1        4.2   1
   2        4.3   1
   3        3.5   1
   4        39.94 1
* RIGHT TUNNEL
* CARD 4.1
* R1    A1    R2    A2  R3    A3     R4    GR    GA
  5.24  60.   4.24  30. 9.86  19.781 23.86 1.0   0.5
* CARD 4.2
* INVSHOT  TS
  0        0.3
* CARD 4.3
* NUMRB  LRB    LSPACING  TSPACING  NSRB
  11     3.0    0.8       1.2        2
* LEFT TUNNEL
* CARD 4.1
* R1    A1    R2    A2  R3    A3     R4    GR    GA
  7.24  60.   6.24  30. 11.86 21.781 25.86 1.0   0.5
* CARD 4.2
* INVSHOT  TS
  0        0.35
* CARD 4.3
* NUMRB  LRB    LSPACING  TSPACING  NSRB
  15     3.0    0.8       1.2        2
* CARD 5.1
* LDTYPE  DGW  GAMAW
  1       2.0  1.0
* END OF DATA
```

MODEL=4 Two Tunnel (Unsymmetric Section)



DELTAX = 2.0 DELTAY = 2.0

NDYMAX = 40

Right Tunnel Tunnel dimensions are shown in Figure 7.16

Left Tunnel R_1 = 7.24 M A_1 = 60°

R_2 = 6.24 M A_2 = 30°

R_3 = 11.86 M A_3 = 21.781°

R_4 = 25.86 M

Number of Rock Bolts (NUMRB)	=	15
Length of Rock Bolts (LRB)	=	3.0 M
Spacing of Rock Bolts (TSPACING)	=	1.2 M
Thickness of Shotcrete (TS)	=	35 Cm

Figure 7.22 Schematic tunnel section view for Model 4 example

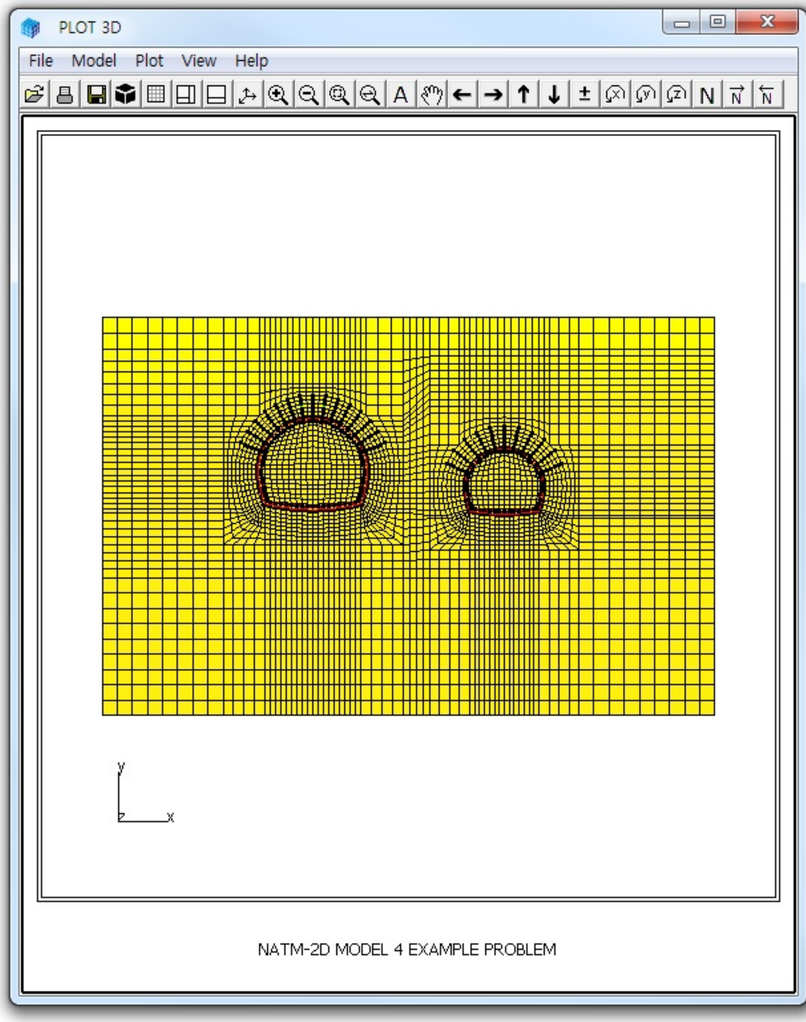


Figure 7.23 Generated finite element mesh for Model 4 example

Table 7.11 Listing of input file Shield.Dat

```
* CARD 1.1
* TITLE
  NATM-2D MODEL 2 FOR SEGMENT LINING
* CARD 1.2
* IUNIT
   2
* CARD 1.3
* MODEL  IGEN  IEXMESH    ILNCOUPL
   2     0     0          1
* CARD 2.1
* HT    HL   W    DELTAX  DELTAX  NDYMAX
  21.94 30.  40.  2.0     2.0     40
* CARD 3.1
* NLAYER
   4
* CARD 3.2
* LAYERNO   H     KF
   1        4.2   1
   2        4.3   1
   3        3.5   1
   4        39.94 1
* CARD 4.1
* R1    A1    R2    A2  R3    A3     R4    GR    GA
  5.3   60.   5.3   60. 5.3   30.    5.3   1.0   0.5
* CARD 4.2
* INVSHOT  TS   TL
  0        0.3  0.3
* NOTE: TUNNEL LINING RADIUS = R1 - TL = 5.3 - 0.3 = 5.0 M
* CARD 4.3
* NUMRB  LRB    LSPACING  TSPACING  NSRB
  11     3.0    0.8       1.2        2
* FOR FINE MESH, USE NSRB = 3
* CARD 5.1
* LDTYPE  DGW  GAMAW  HPRES  VPRES  SUBGK    ITSPR  NUMSJ
  1       2.0  1.0    20.    30.    1.0E+05  1      4
* CARD 5.2
* JOINT LOCATIONS (ANGLES FROM CROWN TOP)
* AJ1   AJ2   AJ3   AJ4
  0     60    120   180
* END OF DATA
```

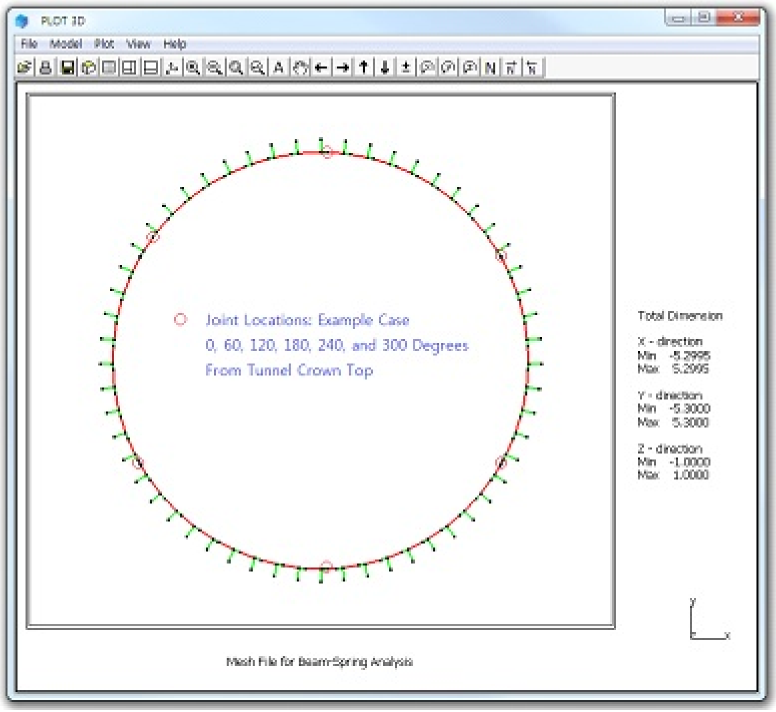


Figure 7.24 Finite element mesh for Model 2-1

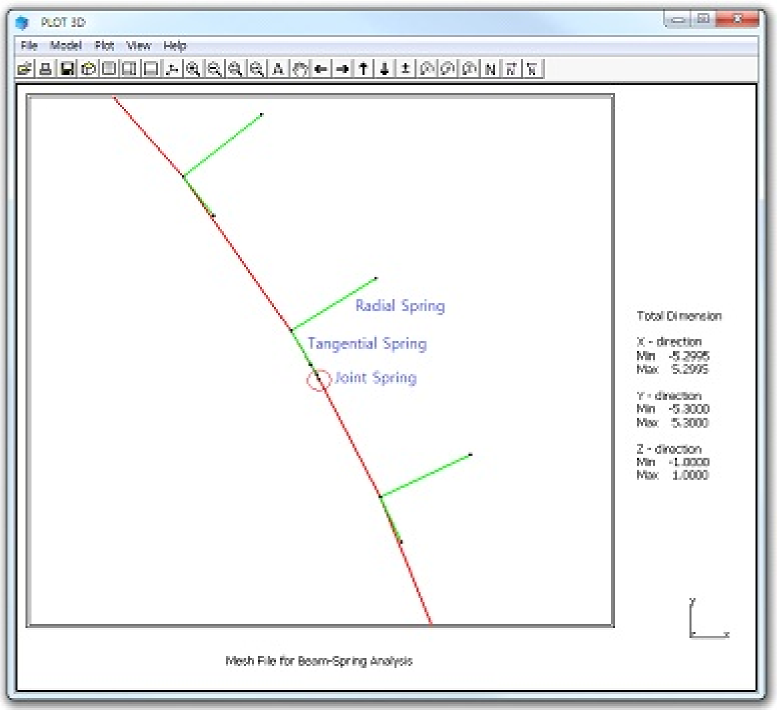


Figure 7.25 Detailed mesh around joint spring element

7.3 CIRCLE-2D

CIRCLE-2D is the special pre-processing program to generate automatically two-dimensional finite element meshes and boundary conditions for circular sections. CIRCLE-2D has three different models:

Model 1	Quarter	Section
Model 2	Half	Section
Model 3	Full	Section

CIRCLE-2D is described in Section 7.4 of User's Manual and can be selected in the following order:

Run → Mesh Generator → PreSmap → Circle 2D

When you finish the execution of CIRCLE-2D, select PLOT-3D to plot the generated finite element mesh.

Three example problems are presented here to show all three types of available models. Figure 7.26 shows schematic section views which are used for example problems.

For each model, we will present:

- Listing of input file
- Graphical output of finite element mesh

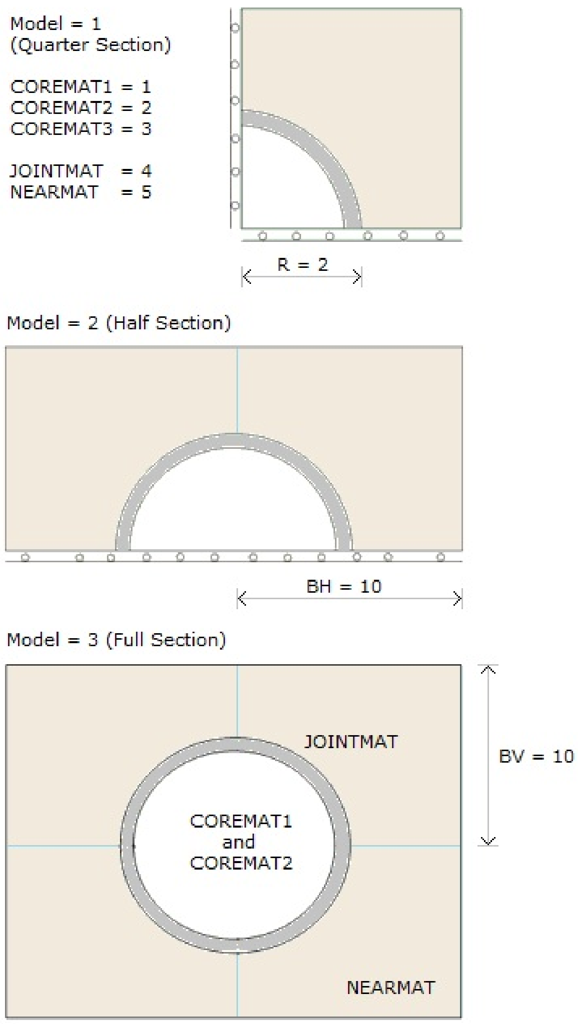


Figure 7.26 Schematic section views for CIRCLE-2D examples

Table 7.12 Listing of input file CIR1C_Q.Dat (MODEL = 1)

```
* CARD 1.1
* TITLE
  MODEL 1 (COARSE, ALL QUAD)
* CARD 1.2
* MODEL     NSNEL      NSNODE
  1         1          1
* CARD 2.1
* R         FINEMESH   NEARMESH    NDIV       BH       BV
  2.0       0          0           5          10.0     10.0
* CARD 3.1
* COREMAT1  COREMAT2   COREMAT2J   JOINTMAT   NEARMAT
  1         2          3           4          5
* END OF DATA
```

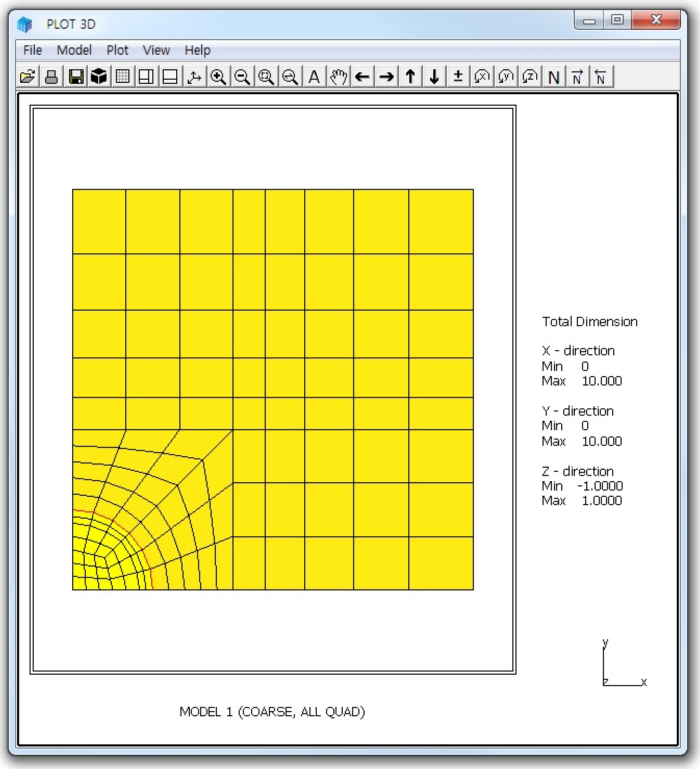


Figure 7.27 Generated finite element mesh for MODEL = 1

Table 7.13 Listing of input file CIR2C_Q.Dat (MODEL = 2)

```
* CARD 1.1
* TITLE
  MODEL 2 (COARSE, ALL QUAD)
* CARD 1.2
* MODEL     NSNEL       NSNODE
  2         1           1
* CARD 2.1
* R         FINEMESH    NEARMESH    NDIV        BH        BV
  2.0       0           0           5           10.0      10.0
* CARD 3.1
* COREMAT1  COREMAT2    COREMAT2J   JOINTMAT    NEARMAT
  1         2           3           4           5
* END OF DATA
```

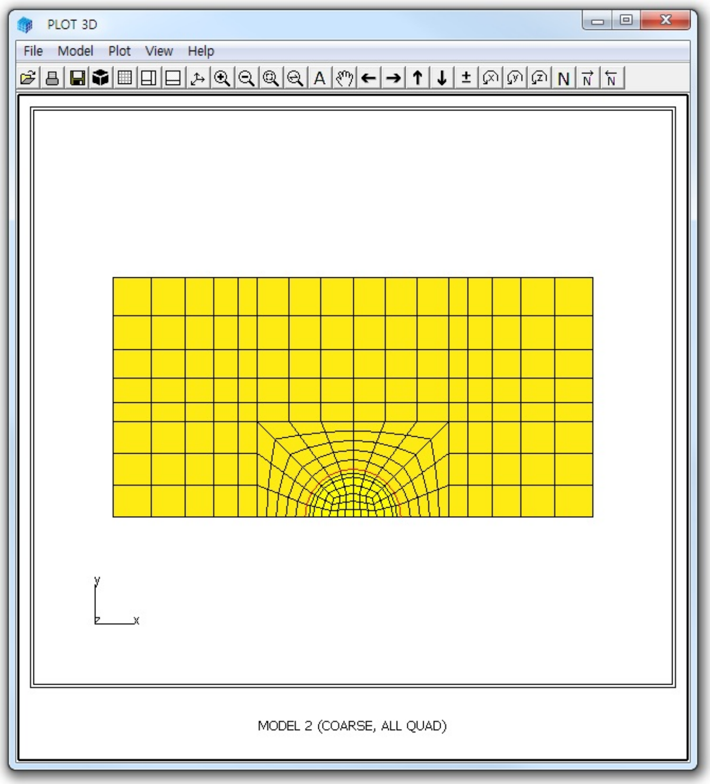


Figure 7.28 Generated finite element mesh for MODEL = 2

Table 7.14 Listing of input file CIR3C_Q.Dat (MODEL = 3)

```
* CARD 1.1
* TITLE
  MODEL 3 (COARSE, ALL QUAD)
* CARD 1.2
* MODEL     NSNEL      NSNODE
  3         1          1
* CARD 2.1
* R         FINEMESH   NEARMESH    NDIV       BH        BV
  2.0       0          0           5          10.0      10.0
* CARD 3.1
* COREMAT1  COREMAT2   COREMAT2J   JOINTMAT   NEARMAT
  1         2          3           4          5
* END OF DATA
```

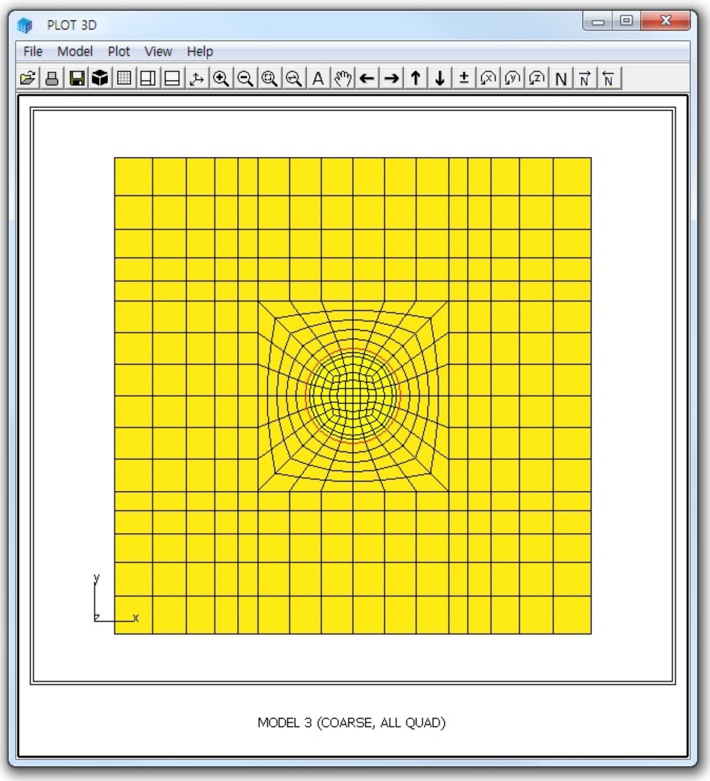


Figure 7.29 Generated finite element mesh for MODEL = 3

7.4 PRESMAP-3D

PRESMAP-3D is the basic pre-processor which can be applied to model various types of 3 dimensional geometries. Input parameters of PRESMAP-3D have been described in detail in Section 7.5 of User's Manual.

PRESMAP-3D can be selected in the following order:

Run → Mesh Generator → PreSmap → Presmap 3D

When you finish the execution of PRESMAP-3D, select PLOT-3D to plot the generated mesh.

7.4.1 Example 1

Figure 7.31 shows block nodes and block numbers for example 1. Detailed block information is listed in Table 7.15. There are 18 block nodes and 3 blocks. Both blocks 1 and 2 have 2 divisions in the x direction and only 1 division in y and z directions. Block 3 has 2 divisions in the z direction and only 1 division in x and y directions. To plot block diagram as shown in Figure 7.31, make the value of NBLOCK negative (example, NBLOCK=-3).

As boundary conditions, roller boundary is assumed along the left surface of blocks 1 and 2, fixed boundary along the right surface of block 3, and free boundary for the rest. Note that PRESMAP-3D generates fixed boundary for all rotational degrees of freedom; i.e., IRX=IRY=IRZ=1.

Graphical outputs are shown for:

- Node numbers in Figure 7.32
- Element numbers in Figure 7.33
- Boundary codes in Figure 7.34

Note that boundary codes in Figure 7.34 are expressed in 3 digits at nodal points. First digit represents ISX, second for ISY and third for ISZ.

Table 7.15 Listing of input file GM3D.Rgn

```
* CARD 1.1
* TITLE
  EXAMPLE 1
* CARD 1.2
* NBLOCK  NBNODE  NSNODE  NSNEL  CMFAC
  3        18      1       1      1.0
* CARD 2.1
* NODE   X     Y     Z
   1     0.0   4.0   5.0
   2     3.0   4.0   5.0
   3     4.0   4.0   5.0
   4     0.0   0.0   5.0
   5     3.0   0.0   5.0
   6     4.0   0.0   5.0
*
   7     0.0   4.0   2.0
   8     3.0   4.0   2.0
   9     4.0   4.0   2.0
  10     0.0   0.0   2.0
  11     3.0   0.0   2.0
  12     4.0   0.0   2.0
*
  13     0.0   4.0   0.0
  14     3.0   4.0   0.0
  15     4.0   4.0   0.0
  16     0.0   0.0   0.0
  17     3.0   0.0   0.0
  18     4.0   0.0   0.0
* ----------------------------------------------------------
* CARD 3.1
  BLOCK 1
* CARD 3.2
* ILAG
   0
* CARD 3.3
*  I1   I2   I3   I4   I5   I6   I7   I8
    2    1    4    5    8    7   10   11
*  M9  M10  M11  M12  M13  M14  M15  M16  M17  M18  M19  M20
    0    0    0    0    0    0    0    0    0    0    0    0
* M21  M22  M23  M24  M25  M26  M27
* CARD 3.4.1
* NBOUND
   2
```

```
* CARD 3.4.2
* IBTYPE  ISX  ISY  ISZ  IFX  IFY  IFZ
   1        0    0    0    0    0    0
   4        1    0    0    1    0    0
* CARD 3.5
* MATNO  NDX  NDY  NDZ  KS  KF
   1      2    1    1    0   0
* -------------------------------------------------------------
* CARD 3.1
  BLOCK 2
* CARD 3.2
* ILAG
   0
* CARD 3.3
*  I1   I2   I3   I4   I5   I6   I7   I8
    8    7   10   11   14   13   16   17
*  M9  M10  M11  M12  M13  M14  M15  M16  M17  M18  M19  M20
    0    0    0    0    0    0    0    0    0    0    0    0
* M21  M22  M23  M24  M25  M26  M27
* CARD 3.4.1
* NBOUND
   2
* CARD 3.4.2
* IBTYPE  ISX  ISY  ISZ  IFX  IFY  IFZ
   1        0    0    0    0    0    0
   4        1    0    0    1    0    0
* CARD 3.5
* MATNO  NDX  NDY  NDZ  KS  KF
   2      2    1    1    0   0
* -------------------------------------------------------------
* CARD 3.1
  BLOCK 3
* CARD 3.2
* ILAG
   0
* CARD 3.3
*  I1   I2   I3   I4   I5   I6   I7   I8
    3    2    5    6   15   14   17   18
*  M9  M10  M11  M12  M13  M14  M15  M16  M17  M18  M19  M20
    9    8   11   12    0    0    0    0    0    0    0    0
* M21  M22  M23  M24  M25  M26  M27
* CARD 3.4.1
* NBOUND
   2
* CARD 3.4.2
* IBTYPE  ISX  ISY  ISZ  IFX  IFY  IFZ
   1        0    0    0    0    0    0
   5        1    1    1    1    1    1
* CARD 3.5
* MATNO  NDX  NDY  NDZ  KS  KF
   3      1    1    2    0   1
```

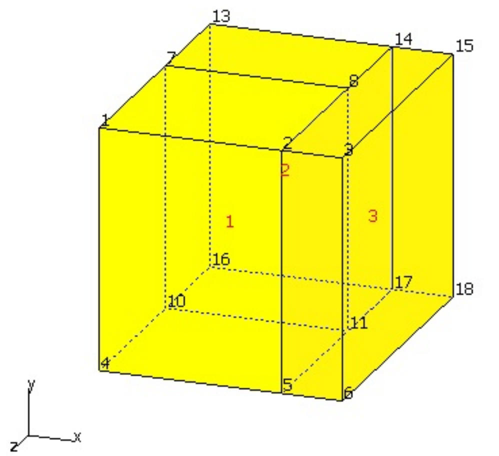


Figure 7.31 Block diagram for example 1

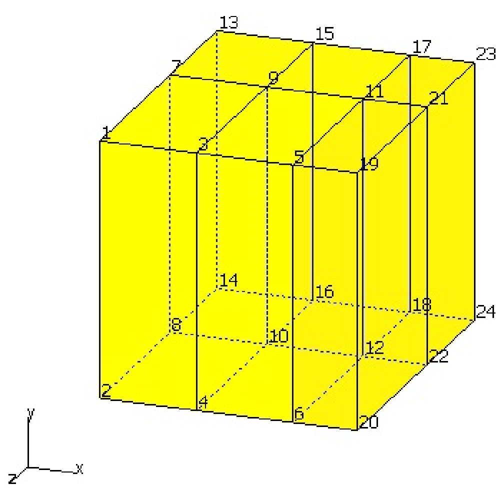


Figure 7.32 Generated node numbers for example 1

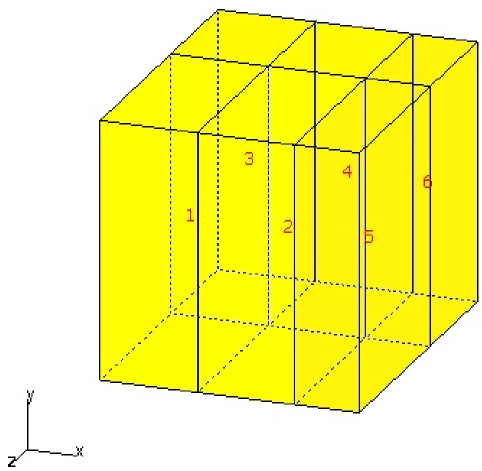


Figure 7.33 Generated element numbers for example 1

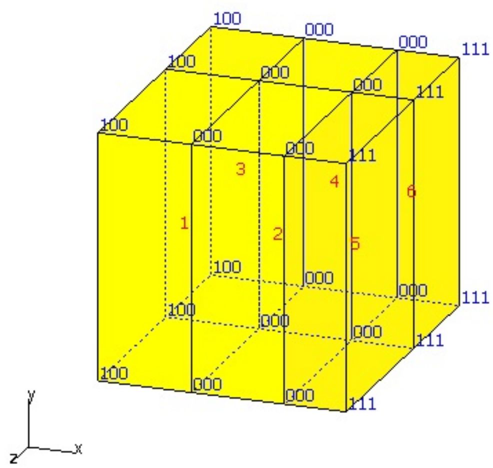


Figure 7.34 Generated boundary codes for example 1 (ISX, ISY, ISZ)

7.5 CROSS-3D

CROSS-3D is the special pre-processing program developed to generate automatically three-dimensional finite element meshes and boundary conditions for crossing tunnels. There are 3 models available for CROSS-3D. Model 1 represents identical size tunnels crossing at right angle at the same level. Model 2 represents large and small tunnels crossing at right angle at the same level. And Model 3 represents lower and upper tunnels crossing at right angle with some clearance. Input parameters of each model have been described in detail in Section 7.6 of User's Manual.

CROSS-3D can be selected in the following order:

Run → Mesh Generator → PreSmap → Cross 3D

When you finish the execution of CROSS-3D, select PLOT-3D to plot the generated mesh.

It should be noted that once you finished running CROSS-3D, you will obtain an intermediate file with file extension .Tmp in working directory. This intermediate file contains useful block information which is essentially the input data to the program PRESMAP-3D.

7.5.1 Model 1

Figure 7.35 shows the schematic view of identical two crossing tunnels for Model 1 example problem. Dimensions defining tunnel location are listed in Figure 7.36. Table 7.16 shows the listing of input file CROSS-M1.Dat.

The output file, CROSS-M1.Tmp in Table 7.17, from CROSS-3D contains block information for the program PRESMAP-3D. Block diagram is shown in Figures 7.37.

Generated finite element mesh is shown in Figure 7.38. Figure 7.39 shows finite element meshes around tunnel core sections.

Table 7.16 Listing of input file CROSS-M1.Dat

```
*
* CARD 1.1
* TITLE
 Identical two crossing tunnels (MODELNO = 1)
* CARD 1.2
* MODELNO  KF  NSNODE  NSNEL  CMFAC
    1       1    1       1     1.0
* CARD 2.1.1
* XL    YB    YT    ZL    t
 100.   50.   100.  100.  3.0
* CARD 2.1.2
* IPART  NDR  NTBND  NTOPN
    0      2   20     20
* CARD 2.1.3
* NTNODE
    9
* NODE  X       Y
    1   0.0     4.0
    2   2.8284  2.8284
    3   4.0     0.0
    4   4.0    -2.0
    5   0.0    -3.0
    6   1.53    3.7
    7   3.7     1.53
    8   4.0    -1.0
    9   2.0    -2.7
* CARD 3.1
* NBOUND
    6
* CARD 3.2
* IBTYPE ISX  ISY  ISZ  IFX  IFY  IFZ
    1     0    0    0    1    1    1
    2     0    0    1    1    1    1
    3     0    0    1    1    1    1
    4     1    0    0    1    1    1
    5     1    0    0    1    1    1
    7     0    1    0    1    1    1
* END OF DATA
```

Table 7.17 Listing of output file CROSS-M1.Tmp

```
* CARD 1.1
* TITLE
 Identical two crossing tunnels (MODELNO = 1)
* CARD 1.2
* NBLOCK NBNODE NSNODE  NSNEL          CMFAC
      25    150      1      1     .10000E+01
* CARD 2.1
* NODE   X-COORDINATE   Y-COORDINATE   Z-COORDINATE
    1     .00000E+00     .10000E+03     .10000E+03
    2     .00000E+00     .29125E+02     .10000E+03
    3     .00000E+00     .70000E+01     .10000E+03
    4     .00000E+00     .40000E+01     .10000E+03
    5     .00000E+00     .00000E+00     .10000E+03
    6     .00000E+00    -.30000E+01     .10000E+03
    7     .00000E+00    -.60000E+01     .10000E+03
    8     .00000E+00    -.18938E+02     .10000E+03
    9     .00000E+00    -.50000E+02     .10000E+03
   10     .26775E+01     .64750E+01     .10000E+03
   11     .15300E+01     .37000E+01     .10000E+03

    -

  139     .40000E+01    -.50000E+01     .26688E+02
  140     .40000E+01    -.50000E+02     .26688E+02
  141     .70000E+01     .00000E+00     .29125E+02
  142     .70000E+01    -.20000E+01     .29125E+02
  143     .70000E+01    -.50000E+01     .29125E+02
  144     .70000E+01    -.50000E+02     .29125E+02
  145     .10000E+03     .10000E+03     .27459E+02
  146     .10000E+03     .49497E+01     .27459E+02
  147     .10000E+03     .00000E+00     .29125E+02
  148     .10000E+03    -.20000E+01     .29125E+02
  149     .10000E+03    -.50000E+01     .29125E+02
  150     .10000E+03    -.50000E+02     .29125E+02
* ===========================================================
* CARD 3.1
  BLOCK  1
* CARD 3.2
* ILAG
    0
* CARD 3.3
*  I1   I2   I3   I4   I5   I6   I7   I8
   12    4    5   14   58   50   51   60
*  M9  M10  M11  M12  M13  M14  M15  M16  M17  M18  M19  M20
  136  129  130  137   11    0    0   13   57    0    0   59
```

```
* CARD 3.4.1
* NBOUND
    3
* CARD 3.4.2
* IBTYPE ISX   ISY   ISZ   IFX   IFY   IFZ
      1     0     0     0     1     1     1
      2     0     0     1     1     1     1
      4     1     0     0     1     1     1
* CARD 3.5
* MATNO  NDX   NDY   NDZ    KS    KF
      1     4     4     5     0     1
* ============================================================
* CARD 3.1
  BLOCK  2
* CARD 3.2
* ILAG
    0
* CARD 3.3
*  I1    I2    I3    I4    I5    I6    I7    I8
   25     3     4    12    68    49    50    58
*  M9   M10   M11   M12   M13   M14   M15   M16   M17   M18   M19   M20
  135   128   129   136    10     0    11     0    56     0    57     0
* CARD 3.4.1
* NBOUND
    3
* CARD 3.4.2
* IBTYPE ISX   ISY   ISZ   IFX   IFY   IFZ
      1     0     0     0     1     1     1
      2     0     0     1     1     1     1
      4     1     0     0     1     1     1
* CARD 3.5
* MATNO  NDX   NDY   NDZ    KS    KF
      2     4     2     5     0     1

    -
* ============================================================
* CARD 3.1
  BLOCK            24
* CARD 3.2
* ILAG
    0
* CARD 3.3
*  I1    I2    I3    I4    I5    I6    I7    I8
   43    29    30    44   123    72    73   124
*  M9   M10   M11   M12   M13   M14   M15   M16   M17   M18   M19   M20
  148   142   143   149    36     0    37     0    81     0    82     0
```

```
* CARD 3.4.1
* NBOUND
    3
* CARD 3.4.2
* IBTYPE ISX   ISY   ISZ   IFX   IFY   IFZ
      1    0     0     0     1     1     1
      2    0     0     1     1     1     1
      5    1     0     0     1     1     1
* CARD 3.5
* MATNO  NDX   NDY   NDZ    KS    KF
     10    5     2     5     0     1
* ==========================================================
* CARD 3.1
  BLOCK          25
* CARD 3.2
* ILAG
    0
* CARD 3.3
*  I1    I2    I3    I4    I5    I6    I7    I8
   44    30    32    46   124    73    75   126
*  M9   M10   M11   M12   M13   M14   M15   M16   M17   M18   M19   M20
  149   143   144   150    37    31    38    45    82    74    83   125
* CARD 3.4.1
* NBOUND
    4
* CARD 3.4.2
* IBTYPE ISX   ISY   ISZ   IFX   IFY   IFZ
      1    0     0     0     1     1     1
      2    0     0     1     1     1     1
      5    1     0     0     1     1     1
      7    0     1     0     1     1     1
* CARD 3.5
* MATNO  NDX   NDY   NDZ    KS    KF
     10    5     5     5     0     1
* ==========================================================
```

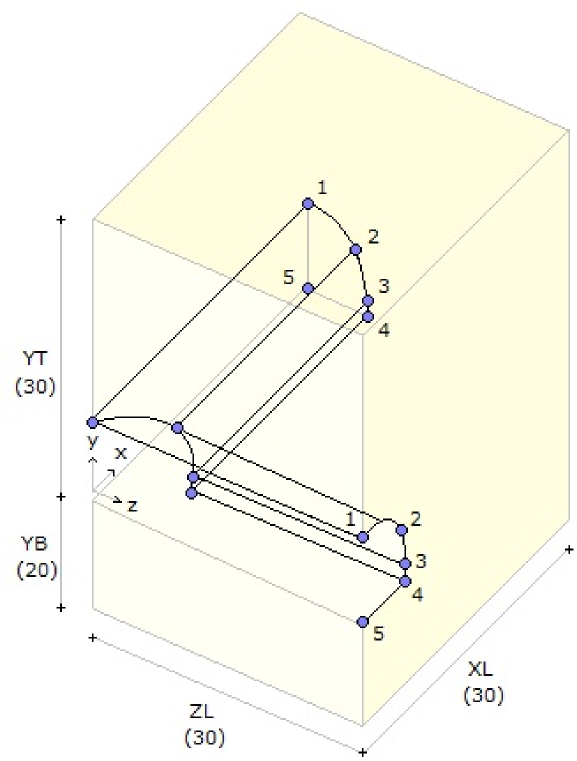


Figure 7.35 Schematic view of crossing tunnels for Model 1

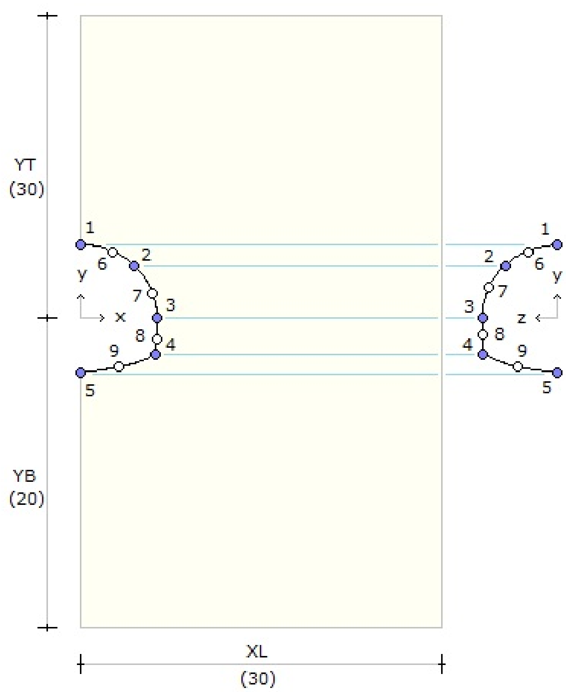


Figure 7.36 Tunnel shapes for Model 1

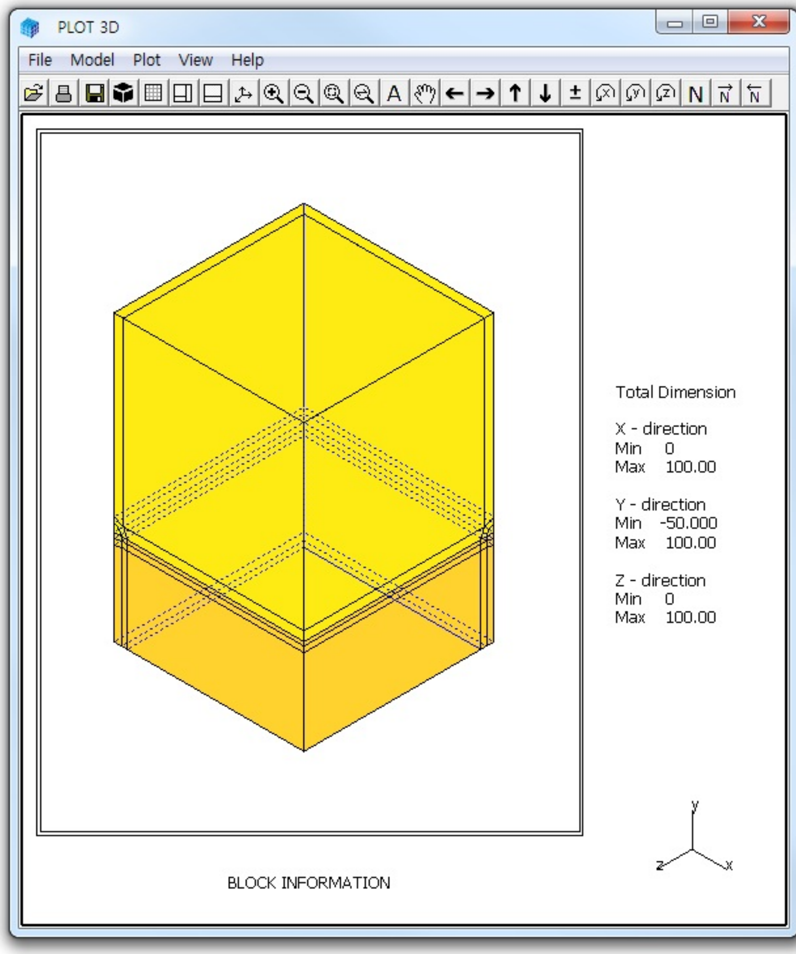


Figure 7.37 Block diagram for Model 1

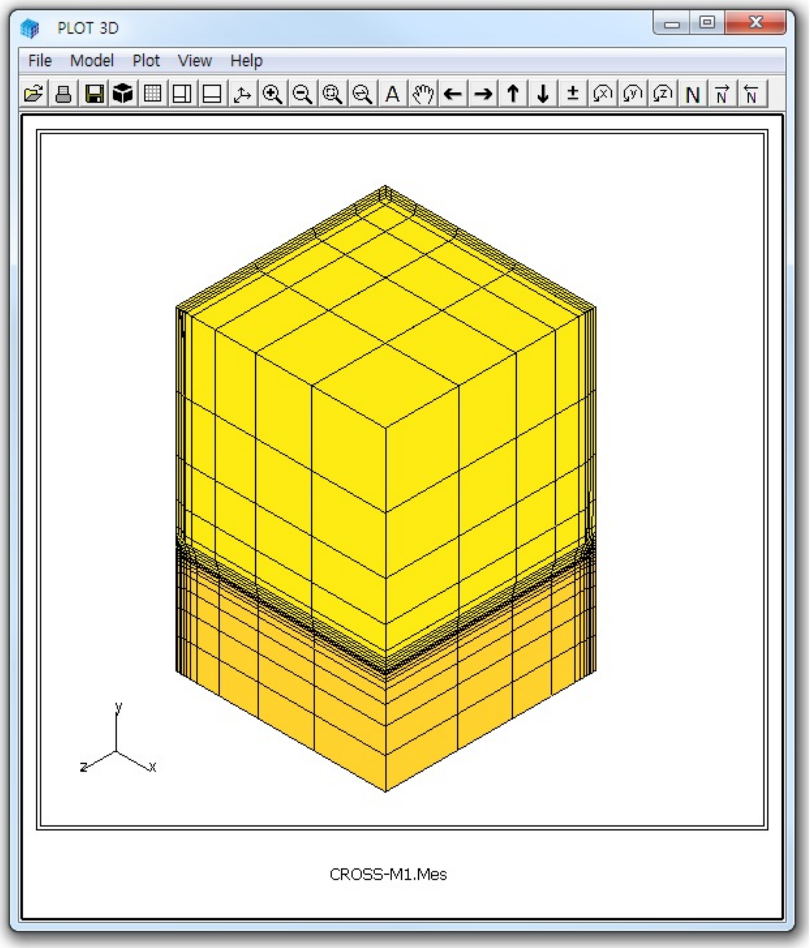


Figure 7.38 Generated finite element mesh for Model 1

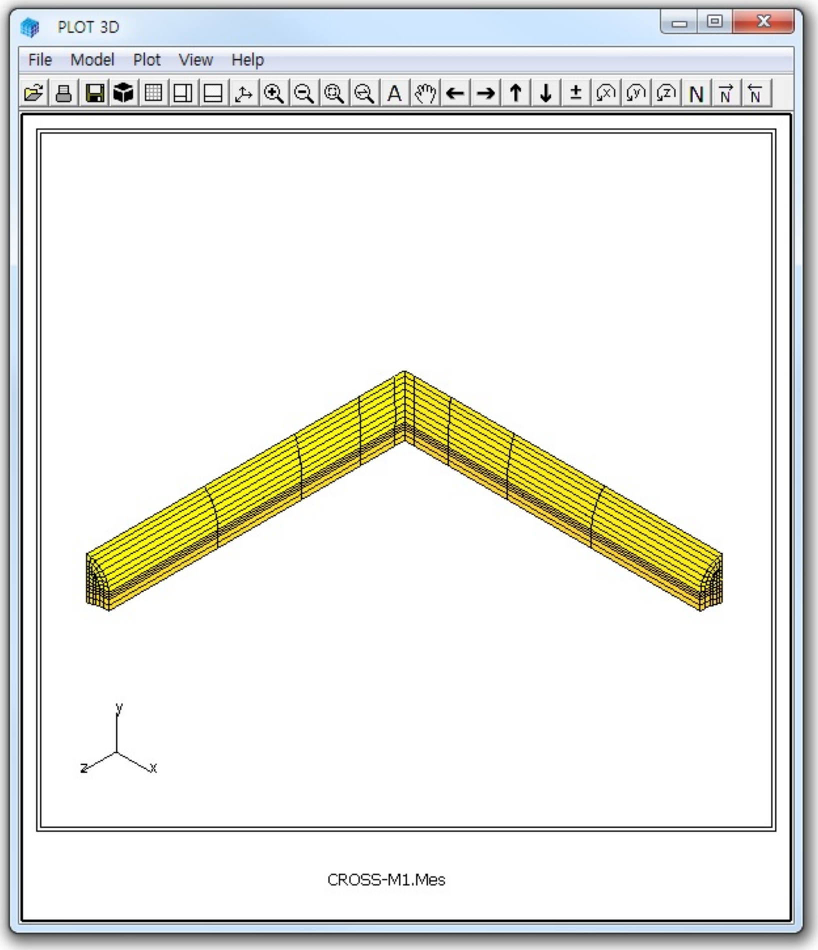


Figure 7.39 Generated mesh around tunnel core sections for Model 1

7.5.2 Model 2

Figure 7.40 shows the schematic view of large and small crossing tunnels for Model 2 example problem. Dimensions defining tunnel location are listed in Figure 7.41. Table 7.18 shows the listing of input file CROSS-M2.Dat.

The output file, CROSS-M2.Tmp in Table 7.19, from CROSS-3D contains block information for the program PRESMAP-3D. Block diagram is shown in Figures 7.42.

Generated finite element mesh is shown in Figure 7.43. Figure 7.44 shows the finite element meshes around tunnel core sections.

Table 7.18 Listing of input file CROSS-M2.Dat

```
* CARD 1.1
* TITLE
 Large and small crossing tunnels (MODELNO = 2)
* CARD 1.2
* MODELNO   KF  NSNODE  NSNEL  CMFAC
    2        1    1       1     1.0
* CARD 2.2.1
* XL     YB     YT     ZL     tl     ts
  30.    20.    30.    30.    3.0    3.0
* CARD 2.2.2
* IPART  NDR  NTBND  NTOPNL  NTOPNS
    0      2   20     20      14
* CARD 2.2.3
* NTLNODE
    9
* NODE   X       Y
    1   0.0     9.0
    2   4.3     7.0
    3   6.6     4.0
    4   7.2     2.0
    5   7.3     0.0
    6   7.3    -2.0
    7   0.0    -3.0
    8   2.3     8.3
    9   5.7     5.5
```

```
* CARD 2.3.3
* NTSNODE
    8
* NODE   Z        Y
    1   0.0      4.0
    2   3.5      2.0
    3   4.0      0.0
    4   4.0     -2.0
    5   0.0     -2.0
    6   2.6      3.0
    7   3.9      1.0
    8   4.0     -1.0
* CARD 3.1
* NBOUND
    6
* CARD 3.2
* IBTYPE ISX   ISY   ISZ   IFX   IFY   IFZ
    1      0     0     0     1     1     1
    2      0     0     1     1     1     1
    3      0     0     1     1     1     1
    4      1     0     0     1     1     1
    5      1     0     0     1     1     1
    7      0     1     0     1     1     1
* END OF DATA
```

Table 7.19 Listing of output file CROSS-M2.Tmp

```
* CARD 1.1
* TITLE
 Large and small crossing tunnels (MODELNO = 2)
* CARD 1.2
* NBLOCK NBNODE NSNODE  NSNEL          CMFAC
     34    201      1       1     .10000E+01
* CARD 2.1
* NODE   X-COORDINATE   Y-COORDINATE   Z-COORDINATE
    1      .00000E+00      .30000E+02      .30000E+02
    2      .49793E+01      .30000E+02      .30000E+02
    3      .81520E+01      .30000E+02      .30000E+02
    4      .30000E+02      .30000E+02      .30000E+02
    5      .00000E+00      .12000E+02      .30000E+02
    6      .49793E+01      .99021E+01      .30000E+02
    7      .81520E+01      .70000E+01      .30000E+02
    8      .30000E+02      .70000E+01      .30000E+02
    9      .00000E+00      .90000E+01      .30000E+02
   10      .43000E+01      .70000E+01      .30000E+02
   11      .66000E+01      .40000E+01      .30000E+02
    -
  193      .10300E+02    -.12500E+02      .70000E+01
  194      .30000E+02    -.12500E+02      .70000E+01
```

```
  199     .00000E+00     .00000E+00     .00000E+00
  200     .10212E+02     .17500E+01     .68250E+01
  201     .10300E+02    -.10000E+01     .70000E+01
* ============================================================
* CARD 3.1
  BLOCK  1
* CARD 3.2
* ILAG
    0
* CARD 3.3
*  I1   I2   I3   I4   I5   I6   I7   I8
   10    9   12   11   41   40   42   43
*  M9  M10  M11  M12  M13  M14  M15  M16  M17  M18  M19  M20
  190  166  167  177  112    0    0  113  124    0    0  125
* CARD 3.4.1
* NBOUND
    3
* CARD 3.4.2
* IBTYPE ISX  ISY  ISZ  IFX  IFY  IFZ
      1    0    0    0    1    1    1
      2    0    0    1    1    1    1
      4    1    0    0    1    1    1
* CARD 3.5
* MATNO  NDX  NDY  NDZ   KS   KF
      1    4    3    6    0    1
* ============================================================
* CARD 3.1
  BLOCK  2
* CARD 3.2
* ILAG
    0
* CARD 3.3
*  I1   I2   I3   I4   I5   I6   I7   I8
   11   12   13   14   43   42   22   23
*  M9  M10  M11  M12  M13  M14  M15  M16  M17  M18  M19  M20
  177  167  168  178    0    0    0  114    0  159    0  147
* CARD 3.4.1
* NBOUND
    3
* CARD 3.4.2
* IBTYPE ISX  ISY  ISZ  IFX  IFY  IFZ
      1    0    0    0    1    1    1
      2    0    0    1    1    1    1
    -
* CARD 3.5
* MATNO  NDX  NDY  NDZ   KS   KF
      1    4    4    6    0    1
```

```
* ============================================================
* CARD 3.1
  BLOCK          33
* CARD 3.2
* ILAG
    0
* CARD 3.3
*  I1   I2   I3   I4   I5   I6   I7   I8
   51   50   54   55   63   62   66   67
*  M9  M10  M11  M12  M13  M14  M15  M16  M17  M18  M19  M20
  144  185  186  145   91    0   92    0  156    0  157    0
* CARD 3.4.1
* NBOUND
    3
* CARD 3.4.2
* IBTYPE ISX  ISY  ISZ  IFX  IFY  IFZ
      1    0    0    0    1    1    1
      2    0    0    1    1    1    1
      5    1    0    0    1    1    1
* CARD 3.5
* MATNO  NDX  NDY  NDZ   KS   KF
     10    6    2    6    0    1
* ============================================================
* CARD 3.1
  BLOCK          34
* CARD 3.2
* ILAG
    0
* CARD 3.3
*  I1   I2   I3   I4   I5   I6   I7   I8
   55   54   58   59   67   66   72   73
*  M9  M10  M11  M12  M13  M14  M15  M16  M17  M18  M19  M20
  145  186  187  146   92   95   97   96  157  193  158  194
* CARD 3.4.1
* NBOUND
    4
* CARD 3.4.2
* IBTYPE ISX  ISY  ISZ  IFX  IFY  IFZ
      1    0    0    0    1    1    1
      2    0    0    1    1    1    1
      5    1    0    0    1    1    1
      7    0    1    0    1    1    1
* CARD 3.5
* MATNO  NDX  NDY  NDZ   KS   KF
     10    6    5    6    0    1
```

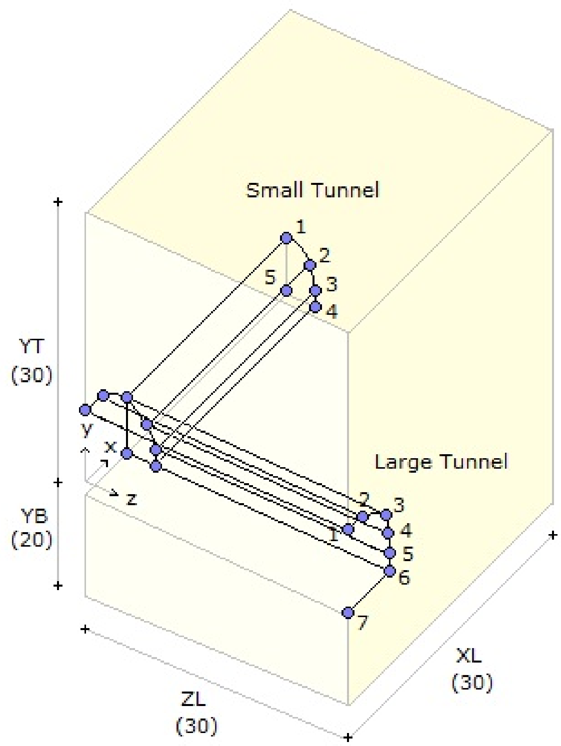


Figure 7.40 Schematic view of crossing tunnels for Model 2

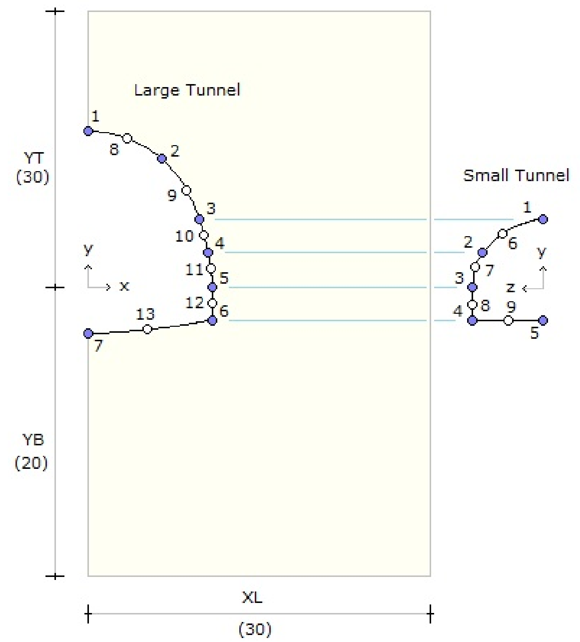


Figure 7.41 Tunnel shapes for Model 2

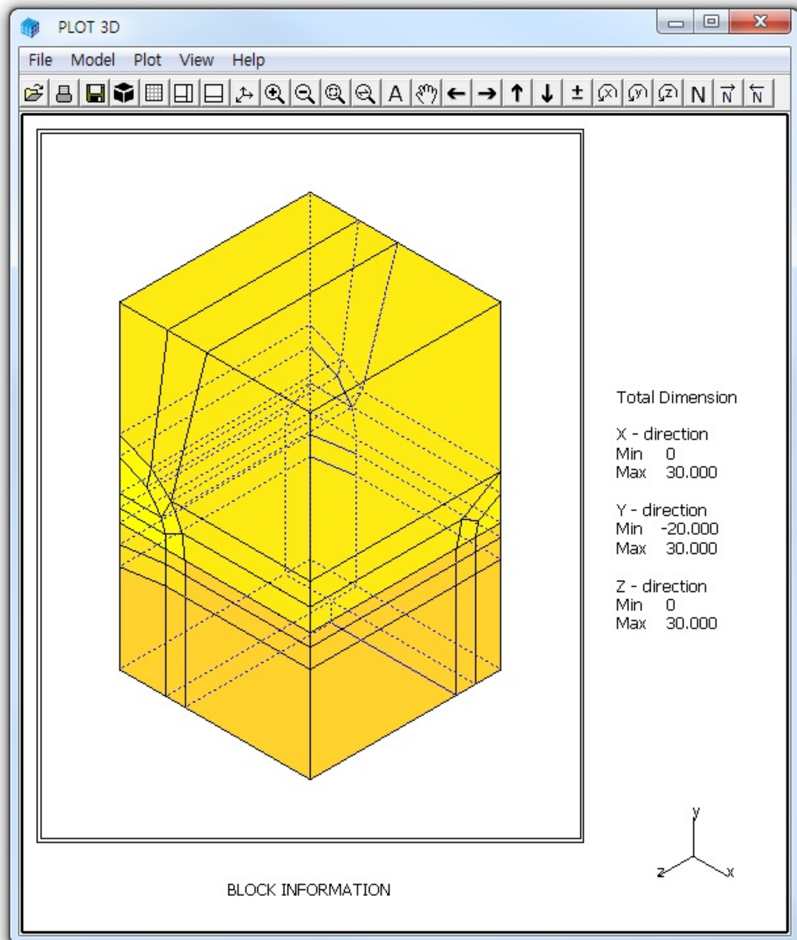


Figure 7.42 Block diagram for Model 2

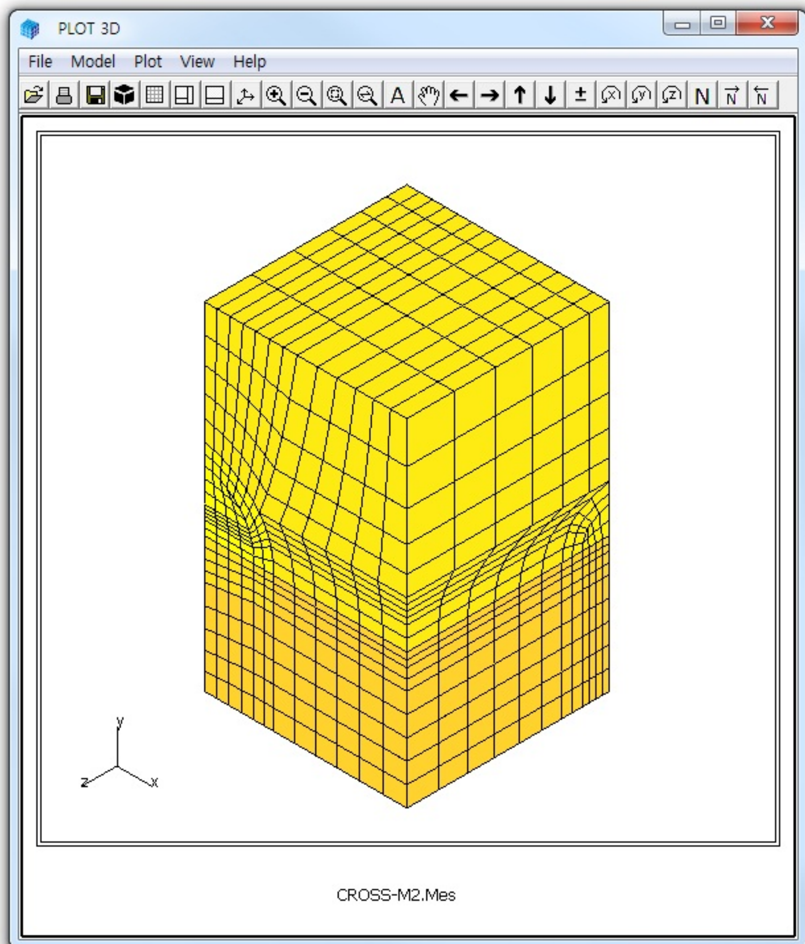


Figure 7.43 Generated finite element mesh for Model 2

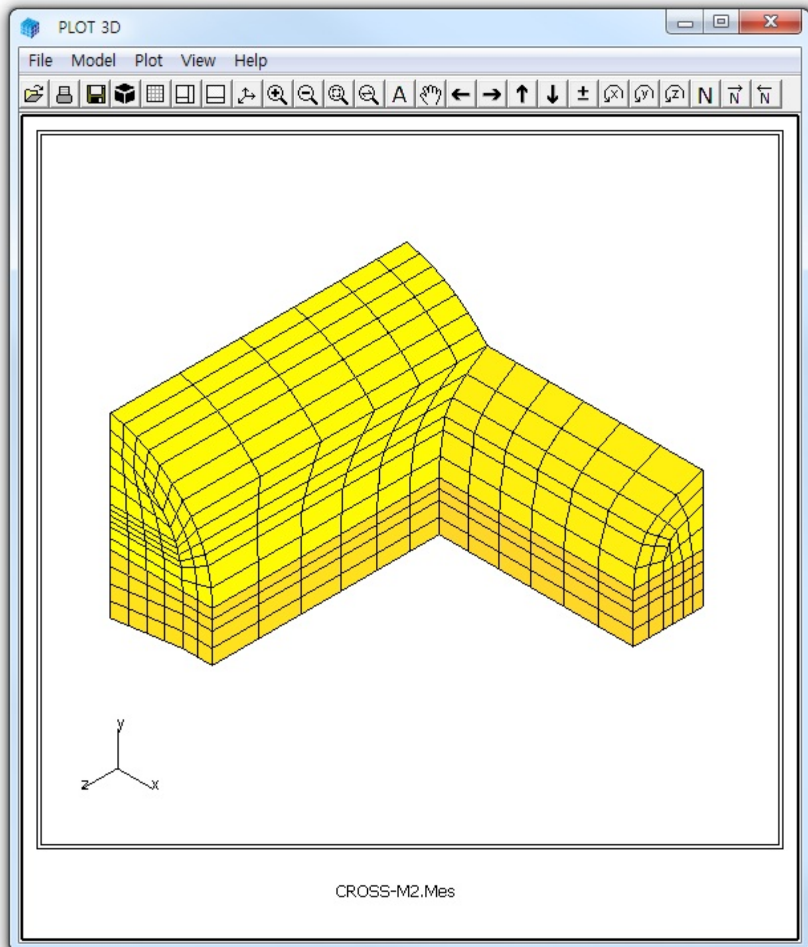


Figure 7.44 Generated mesh around tunnel core sections for Model 2

7.5.3 Model 3

Figure 7.45 shows the schematic view of crossing tunnels with clearance for Model 3 example problem. Dimensions defining tunnel location are listed in Figure 7.46. Table 7.20 shows the listing of input file CROSS-M3.Dat.

The output file, CROSS-M3.Tmp in Table 7.21, from CROSS-3D contains block information for the program PRESMAP-3D. Block diagram is shown in Figures 7.47.

Generated finite element mesh is shown in Figure 7.48. Figure 7.49 shows the finite element meshes around tunnel core sections.

Table 7.20 Listing of input file CROSS-M3.Dat

```
* CARD 1.1
* TITLE
  Crossing tunnels with clearance (MODELNO = 3)
* CARD 1.2
* MODELNO  KF  NSNODE  NSNEL  CMFAC
    3       1    1       1     1.0
* CARD 2.3.1
* XL    YB    YC    YT    ZL    tl    tu
  30.   20.   16.   20.   30.   3.0   3.0
* CARD 2.3.2
* NDRL   NDRU  NTBND  NTOPNL  NTOPNS
    2      2    20     14      14
* CARD 2.3.3
* NTLNODE
    9
* NODE  X       Y
    1   0.0     4.0
    2   2.828   2.828
    3   4.0     0.0
    4   2.828  -2.828
    5   0.0    -4.0
    6   1.531   3.7
    7   3.7     1.531
    8   3.7    -1.531
    9   1.531  -3.7
* CARD 2.3.3
* NTUNODE
    9
```

```
* NODE   Z        Y
    1   0.0      19.
    2   2.12     18.12
    3   3.0      16.
    4   2.12     13.88
    5   0.0      13.
    6   1.148    18.77
    7   2.77     17.148
    8   2.77     14.852
    9   1.148    13.23
* CARD 3.1
* NBOUND
    6
* CARD 3.2
* IBTYPE ISX   ISY   ISZ   IFX   IFY   IFZ
    1     0     0     0     1     1     1
    2     0     0     1     1     1     1
    3     0     0     1     1     1     1
    4     1     0     0     1     1     1
    5     1     0     0     1     1     1
    7     0     1     0     1     1     1
* END OF DATA
```

Table 7.21 Listing of output file CROSS-M3.Tmp

```
* CARD 1.1
* TITLE
  Crossing tunnels with clearance (MODELNO = 3)
* CARD 1.2
* NBLOCK NBNODE NSNODE  NSNEL           CMFAC
      44    226      1      1     .10000E+01
* CARD 2.1
* NODE   X-COORDINATE   Y-COORDINATE   Z-COORDINATE
    1      .30000E+02     .36000E+02     .00000E+00
    2      .30000E+02     .29313E+02     .00000E+00
    3      .30000E+02     .22000E+02     .00000E+00
    4      .30000E+02     .19000E+02     .00000E+00
    5      .30000E+02     .16000E+02     .00000E+00
    6      .30000E+02     .13000E+02     .00000E+00
    7      .30000E+02     .10000E+02     .00000E+00
    8      .30000E+02     .49490E+01     .00000E+00
    9      .30000E+02     .00000E+00     .00000E+00
   10      .30000E+02    -.49490E+01     .00000E+00
   11      .30000E+02    -.12459E+02     .00000E+00
    -
  218      .49490E+01    -.20000E+02     .14792E+02
```

```
  222     .30000E+02     .11760E+02     .14792E+02
  223     .30000E+02     .49490E+01     .14792E+02
  224     .30000E+02     .00000E+00     .14792E+02
  225     .30000E+02    -.49490E+01     .14792E+02
  226     .30000E+02    -.20000E+02     .14792E+02
* ============================================================
* CARD 3.1
  BLOCK  1
* CARD 3.2
* ILAG
    0
* CARD 3.3
*  I1   I2   I3   I4   I5   I6   I7   I8
  171  151  152  173  120  113  114  122
*  M9  M10  M11  M12  M13  M14  M15  M16  M17  M18  M19  M20
  213  164  165  215  170    0    0  172  119    0    0  121
* CARD 3.4.1
* NBOUND
    3
* CARD 3.4.2
* IBTYPE ISX  ISY  ISZ  IFX  IFY  IFZ
      1    0    0    0    1    1    1
      2    0    0    1    1    1    1
      4    1    0    0    1    1    1
* CARD 3.5
* MATNO  NDX  NDY  NDZ   KS   KF
      1    3    3    6    0    1
* ============================================================
* CARD 3.1
  BLOCK  2
* CARD 3.2
* ILAG
    0
* CARD 3.3
*  I1   I2   I3   I4   I5   I6   I7   I8
  120  113  114  122   39   54   55   41
*  M9  M10  M11  M12  M13  M14  M15  M16  M17  M18  M19  M20
    0    0    0    0  119    0    0  121   38    0    0   40
* CARD 3.4.1
* NBOUND
    3
* CARD 3.4.2
* IBTYPE ISX  ISY  ISZ  IFX  IFY  IFZ
      1    0    0    0    1    1    1
 -
```

```
* ============================================================
* CARD 3.1
  BLOCK 43
* CARD 3.2
* ILAG
    0
* CARD 3.3
*  I1    I2    I3    I4    I5    I6    I7    I8
  205   187   189   207    77   131   133    79
*  M9   M10   M11   M12   M13   M14   M15   M16   M17   M18   M19   M20
  225   217   218   226   196   188   197   206   143   132   144    78
* CARD 3.4.1
* NBOUND
    4
* CARD 3.4.2
* IBTYPE ISX   ISY   ISZ   IFX   IFY   IFZ
      1     0     0     0     1     1     1
      2     0     0     1     1     1     1
      5     1     0     0     1     1     1
      7     0     1     0     1     1     1
* CARD 3.5
* MATNO   NDX   NDY   NDZ    KS    KF
     11     6     5     6     0     1
* ============================================================
* CARD 3.1
  BLOCK 44
* CARD 3.2
* ILAG
    0
* CARD 3.3
*  I1    I2    I3    I4    I5    I6    I7    I8
   77   131   133    79    10    34    36    12
*  M9   M10   M11   M12   M13   M14   M15   M16   M17   M18   M19   M20
    0     0     0     0   143   132   144    78    21    35    22    11
* CARD 3.4.1
* NBOUND
    4
* CARD 3.4.2
* IBTYPE ISX   ISY   ISZ   IFX   IFY   IFZ
      1     0     0     0     1     1     1
      3     0     0     1     1     1     1
      5     1     0     0     1     1     1
      7     0     1     0     1     1     1
* CARD 3.5
* MATNO   NDX   NDY   NDZ    KS    KF
     11     6     5     3     0     1
```

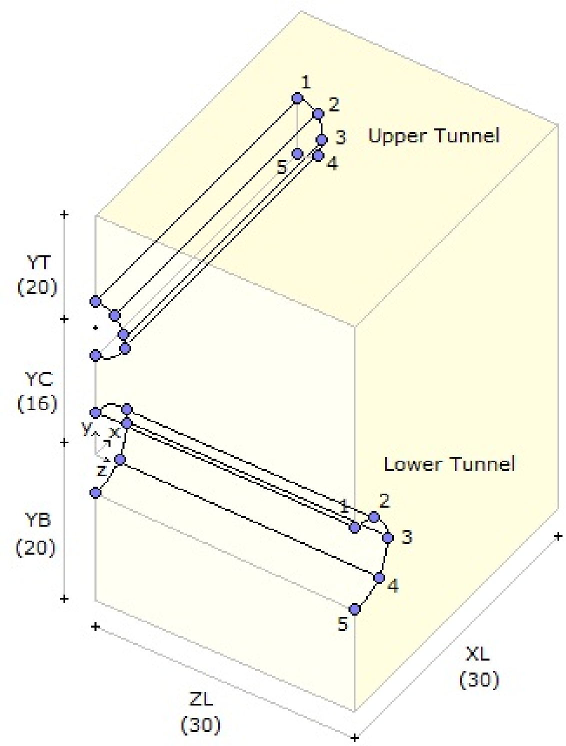


Figure 7.45 Schematic view of crossing tunnels for Model 3

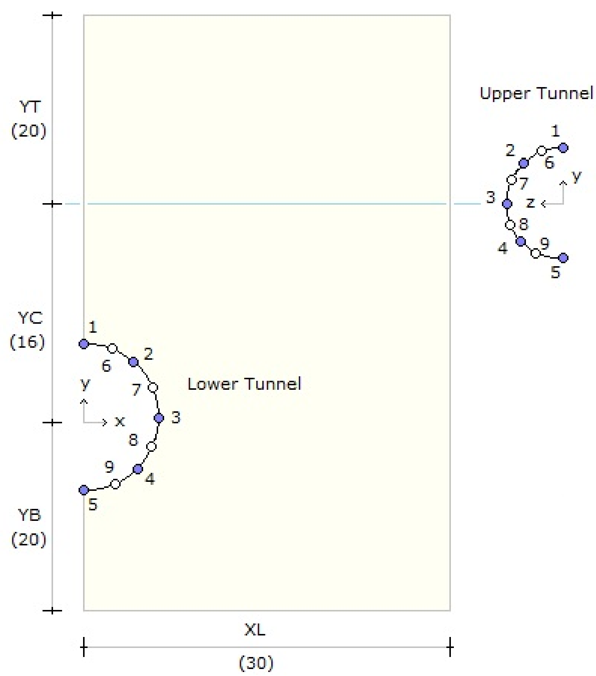


Figure 7.46 Tunnel shapes for Model 3

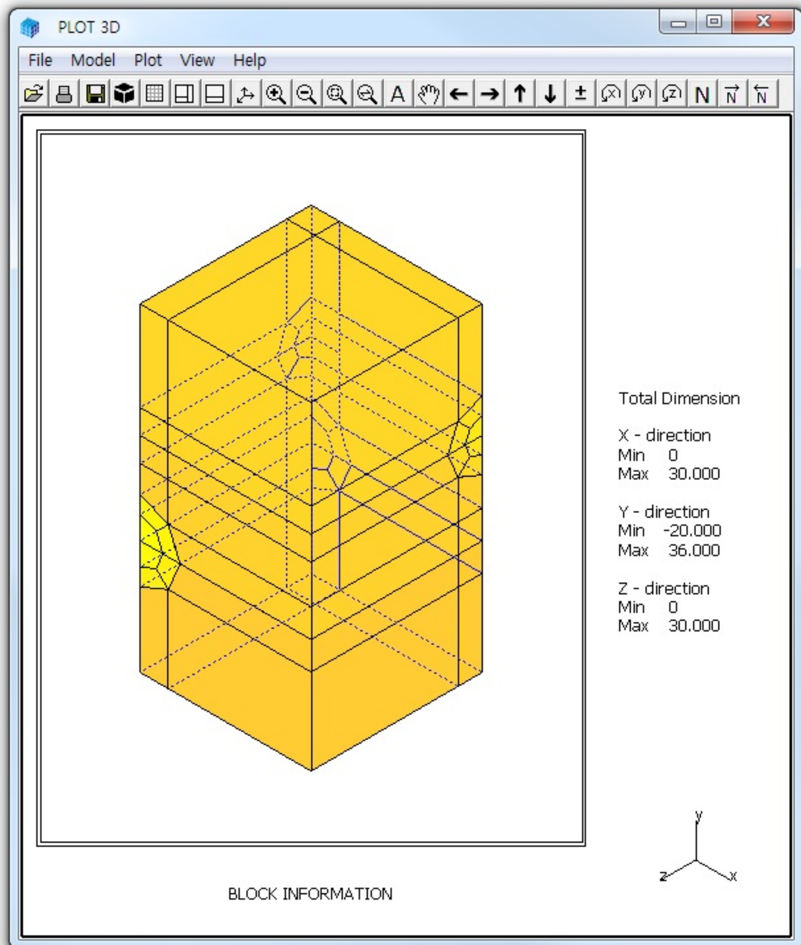


Figure 7.47 Block diagram for Model 3

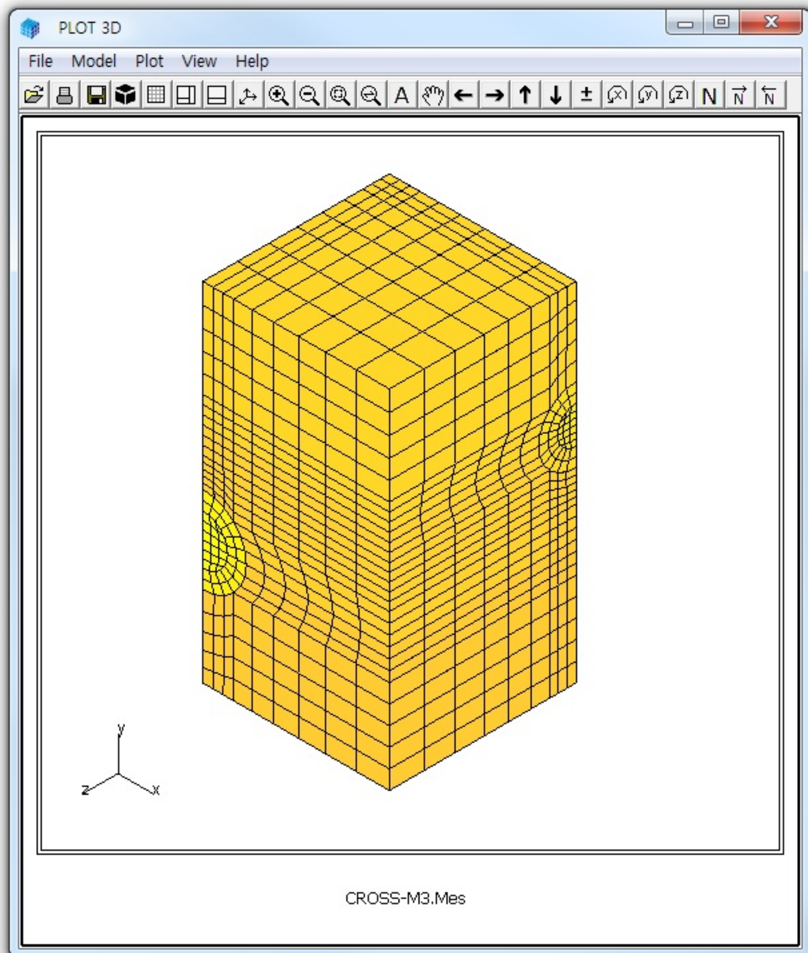


Figure 7.48 Generated finite element mesh for Model 3

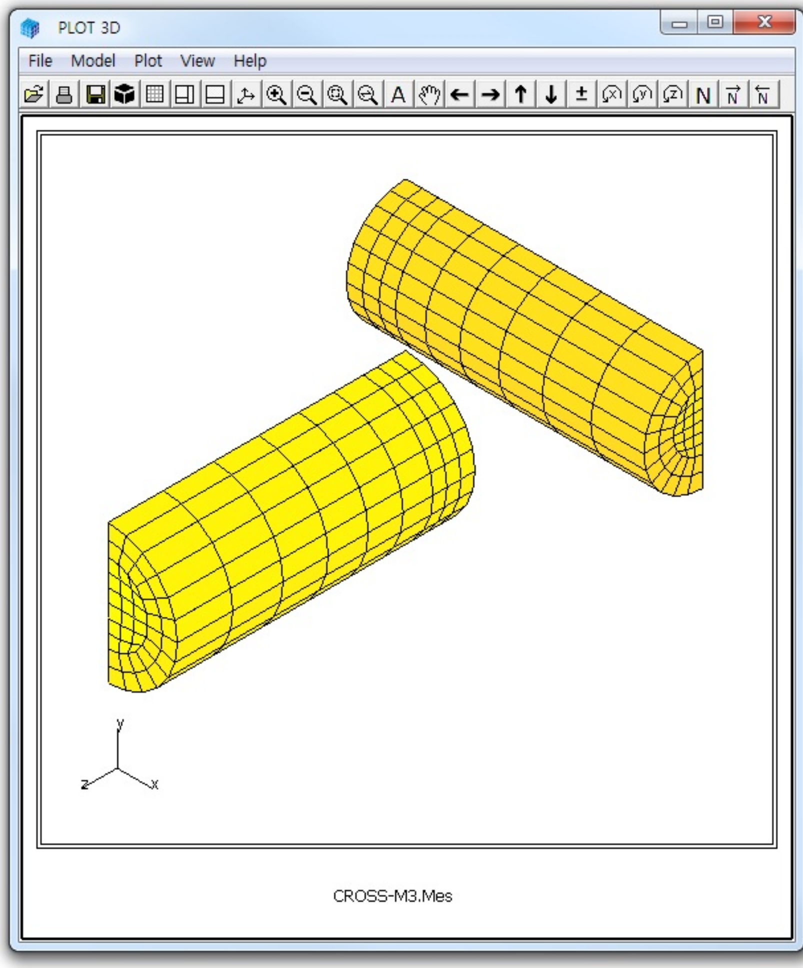


Figure 7.49 Generated mesh around tunnel core sections for Model 3

7.6 GEN-3D

GEN-3D is used to generate nodal coordinates, element indexes, boundary codes, external loads and transmitting boundaries in 3-dimensional coordinate system by extending typical 2-dimensional output from PRESMAP-2D, NATM-2D, CIRCLE-2D, or PRESMAP-GP. Input parameters of GEN-3D have been described in detail in Section 7.7 of User's Manual.

GEN-3D can be selected in the following order:

Run → Mesh Generator → PreSmap → Gen 3D

When you finish the execution of GEN-3D, select PLOT-3D to plot the generated mesh.

7.6.1 Example 1: 3-D Tunnel Mesh Generation

In Sections 7.1 and 8.1, a typical tunnel section having Core, Near-field, and Far-field regions has been constructed using PRESMAP-2D and ADDRGN-2D. Table 7.22 shows the listing of the file, Ex1-2D.Mes, containing this typical 2D section. Graphical output for this typical 2D section is presented in Figure 7.50.

In this example, we want to extend this typical 2D section in XY coordinate into 5 sections in Z coordinate using GEN-3D as schematically illustrated in Figure 7.51. The input file, Ex1.Dat in Table 7.23, has been prepared according to Section 7.7 of User's Manual.

Generated 3-dimensional finite element mesh is shown in Figures 7.52. The output file from GEN-3D contains nodal coordinates, element indexes, and boundary codes which are compatible to format of SMAP-3D Mesh File.

Table 7.22 Listing of typical 2D section Ex1-2D.Mes for Example 1

```
COMBINED REGION : Ex1-2D.Mes
NUMNP   NCONT   NBEAM   NTROSS
 506     464       0      0
NODAL COORDINATES
NODE  ISX  ISY  IFX  IFY  IRZ      XC           YC
  1    1    0    1    1    1  .000000E+00  .474000E+01
  2    0    0    1    1    1  .684000E+00  .469500E+01
  3    0    0    1    1    1  .135600E+01  .456200E+01
  4    0    0    1    1    1  .200500E+01  .434100E+01
  5    0    0    1    1    1  .262000E+01  .403800E+01
  6    0    0    1    1    1  .319000E+01  .366000E+01
  7    0    0    1    1    1  .370500E+01  .320500E+01
  8    1    0    1    1    1  .000000E+00  .395000E+01
  9    0    0    1    1    1  .714000E+00  .384225E+01
 10    0    0    1    1    1  .142200E+01  .369050E+01
 11    0    0    1    1    1  .210460E+01  .349792E+01
  -
  -
494    0    0    1    1    1  .261500E+02 -.241500E+02
495    1    0    1    1    1  .320000E+02 -.241500E+02
496    1    1    1    1    1  .000000E+00 -.300000E+02
497    0    1    1    1    1  .233333E+01 -.300000E+02
498    0    1    1    1    1  .466667E+01 -.300000E+02
499    0    1    1    1    1  .700000E+01 -.300000E+02
500    0    1    1    1    1  .933333E+01 -.300000E+02
501    0    1    1    1    1  .116667E+02 -.300000E+02
502    0    1    1    1    1  .140000E+02 -.300000E+02
503    0    1    1    1    1  .171500E+02 -.300000E+02
504    0    1    1    1    1  .212000E+02 -.300000E+02
505    0    1    1    1    1  .261500E+02 -.300000E+02
506    1    1    1    1    1  .320000E+02 -.300000E+02
ELEMENT INDEX
NEL   I1   I2   I3   I4   M5   M6   M7   M8 MATC KS KF INTR INTS   TBJWL
  1    2    1    8    9    0    0    0    0    4  0  1    2    2  .0000E+00
  2    3    2    9   10    0    0    0    0    4  0  1    2    2  .0000E+00
  3    4    3   10   11    0    0    0    0    4  0  1    2    2  .0000E+00
  4    5    4   11   12    0    0    0    0    4  0  1    2    2  .0000E+00
  5    6    5   12   13    0    0    0    0    4  0  1    2    2  .0000E+00
  6    7    6   13   14    0    0    0    0    4  0  1    2    2  .0000E+00
  -
  -
458  489  488  499  500    0    0    0    0    4  0  1    2    2  .0000E+00
459  490  489  500  501    0    0    0    0    4  0  1    2    2  .0000E+00
460  491  490  501  502    0    0    0    0    4  0  1    2    2  .0000E+00
461  492  491  502  503    0    0    0    0    4  0  1    2    2  .0000E+00
462  493  492  503  504    0    0    0    0    4  0  1    2    2  .0000E+00
463  494  493  504  505    0    0    0    0    4  0  1    2    2  .0000E+00
464  495  494  505  506    0    0    0    0    4  0  1    2    2  .0000E+00
```

Table 7.23 Listing of input file Ex1.Dat for Example 1

```
* CARD 1.1
* TITLE
  3-D TUNNEL MESH GENERATION
* CARD 1.2
* NBZ  NBNODE
   2     3
* CARD 1.3
* IBZ_base IBZ_front IBZ_back
   1        3         3
* CARD 2.1
* NODE   ZP     XP
   1     60.    0
   2     41.    0
   3      0.    0
* ===============================
* CARD 3.1
* BLNAME
  BLOCK1
* IBLNO
  1
* CARD 3.3
* I   J   LTYPE
  1   2   0
* CARD 3.4
* NDZ     ALPA
  2       0.4
* ===============================
* CARD 3.1
* BLNAME
  BLOCK2
* IBLNO
  2
* CARD 3.3
* I   J   LTYPE
  2   3   0
* CARD 3.4
* NDZ     ALPA
  3       0.4
* ===============================
* CARD 4.1
* ITRANB
  0
* END OF DATA
```

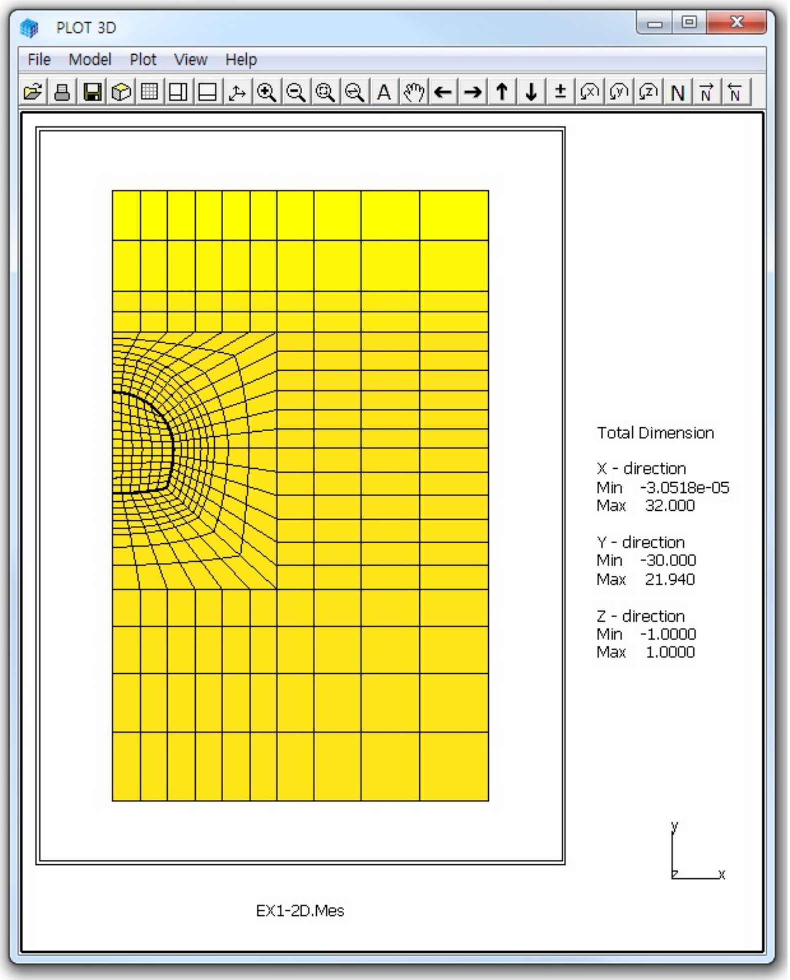
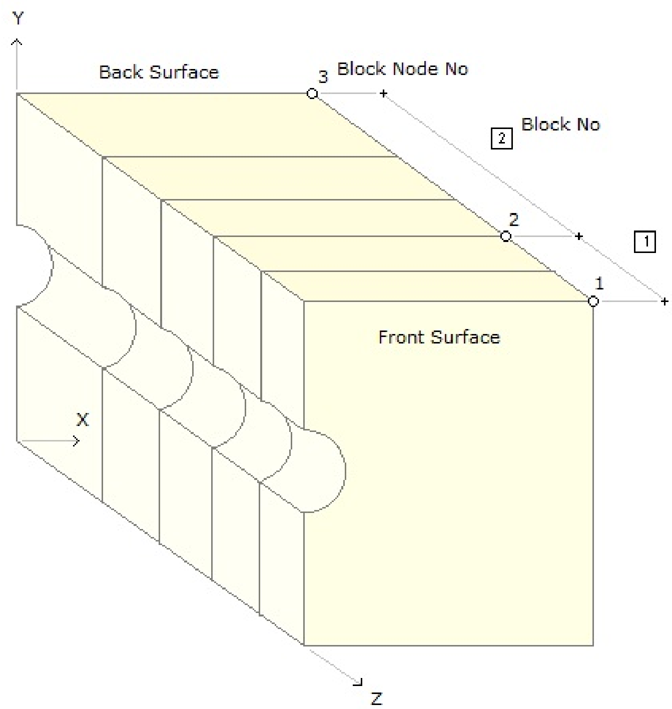


Figure 7.50 Finite element meshes representing typical 2D section



IPLANE = 0 (Input 2D section in X-Y plane)

For Block No 1,	I = 1	J = 2	NDZ = 2	α = 0.4
For Block No 2,	I = 2	J = 3	NDZ = 3	α = 0.4

Total Number of Blocks,	NBZ = 2
Total Number of Block Nodes,	NBNODE = 3

Figure 7.51 Schematic section view of 3D extension

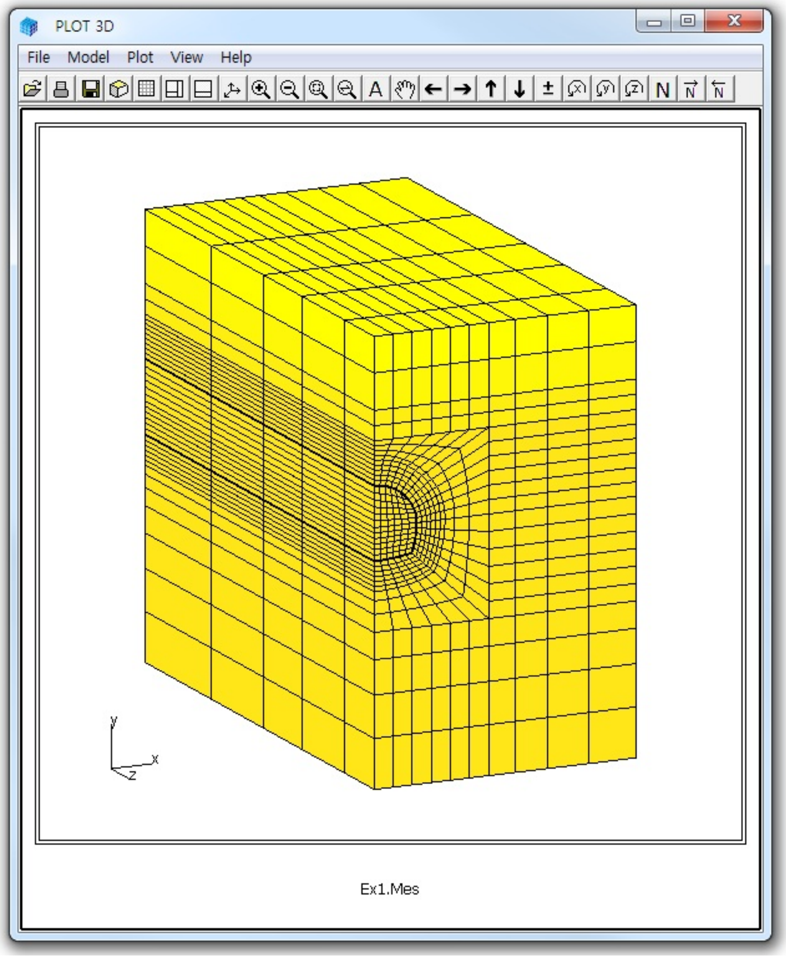


Figure 7.52 Generated 3D finite element mesh

7.6.2 Example 2: 3-D Curved Tunnel

Table 7.24 Listing of input file Ex2.Dat for Example 2

```
* CARD 1.1
* TITLE
  3-D CURVED TUNNEL
* CARD 1.2
* NBZ  NBNODE
   2     3
* CARD 1.3
* IBZ_base IBZ_front IBZ_back
   1        3         3
* CARD 2.1
* NODE   Zp     Xp
   1     0.0    16.30
   2    -6.238  15.06
   3    -42.60  0.00
*================================
* CARD 3.1
* BLNAME
  BLOCK1
* IBLNO
  1
* I   J   LTYPE
  1   2   1
* CARD 3.4
* NDZ,    ALPA
  4       0.5
* CARD 3.5
* Zo      Xo    R     Tb    Te
  0.0     0.0   16.3  0.0   22.5
*================================
* BLNAME
  BLOCK2
* IBLNO
  2
* I   J   LTYPE
  2   3   0
* NDZ     ALPA
  8       0.3
*================================
* CARD 4.1
* ITRANB
  0
* END OF DATA
```

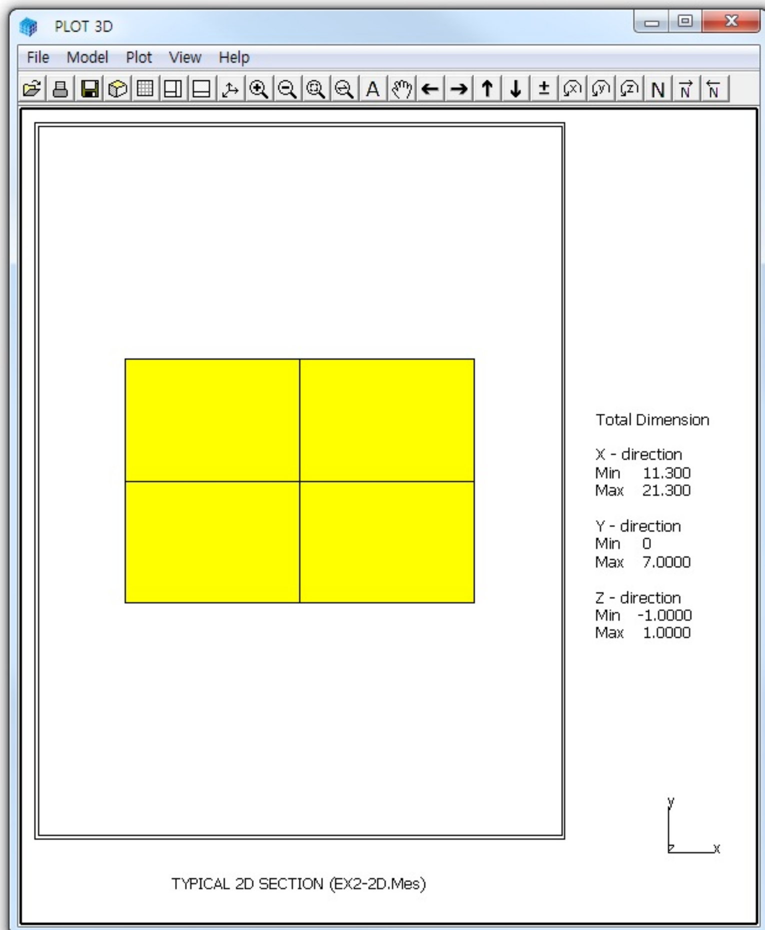


Figure 7.53 Typical 2D section for Example 2

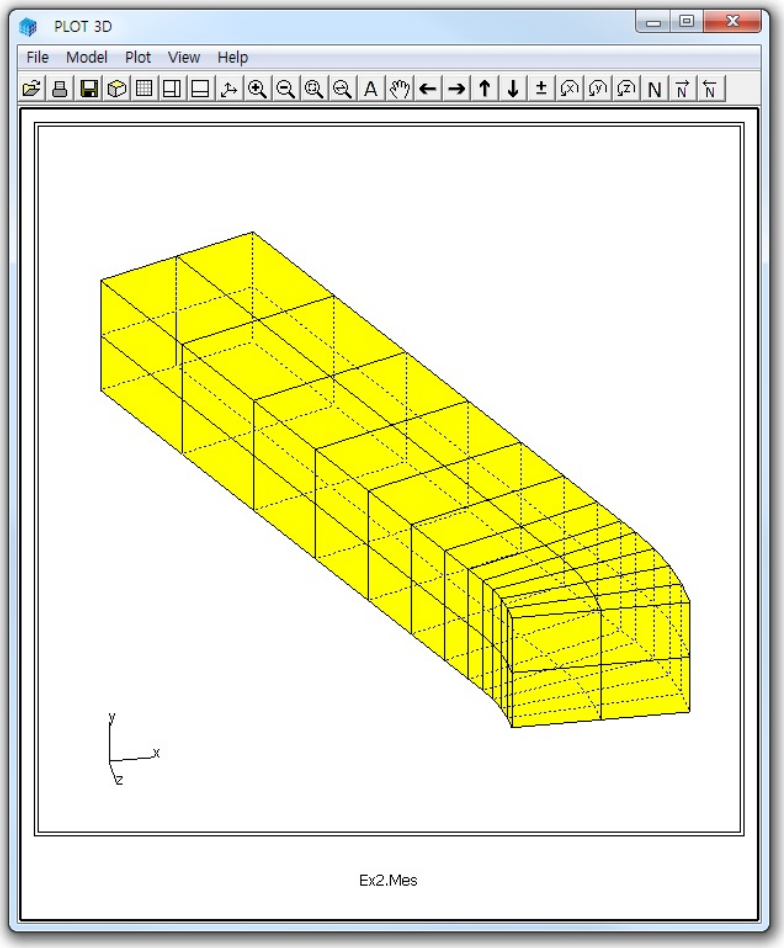


Figure 7.54 Generated 3D mesh for Example 2

7.6.3 Example 3: 3-D Tunnel with Prism Elements

Table 7.25 Listing of input file Ex3.Dat for Example 3

```
* CARD 1.1
* TITLE
  3-D TUNNEL WITH PRISM ELEMENTS
* CARD 1.2
* NBZ  NBNODE
   2     3
* CARD 1.3
* IBZ_base IBZ_front IBZ_back
   1        3         3
* CARD 2.1
* NODE    Zp     Xp
   1      0.0    16.30
   2    -6.238   15.06
   3    -42.60   0.00
*================================
* CARD 3.1
* BLNAME
  BLOCK1
* IBLNO
  1
* I   J   LTYPE
  1   2   1
* CARD 3.4
* NDZ,    ALPA
  4       0.5
* CARD 3.5
* Zo      Xo     R     Tb    Te
  0.0     0.0    16.3  0.0   22.5
*================================
* BLNAME
  BLOCK2
* IBLNO
  2
* I   J   LTYPE
  2   3   0
* NDZ     ALPA
  8       0.3
*================================
* CARD 4.1
* ITRANB
  0
* END OF DATA
```

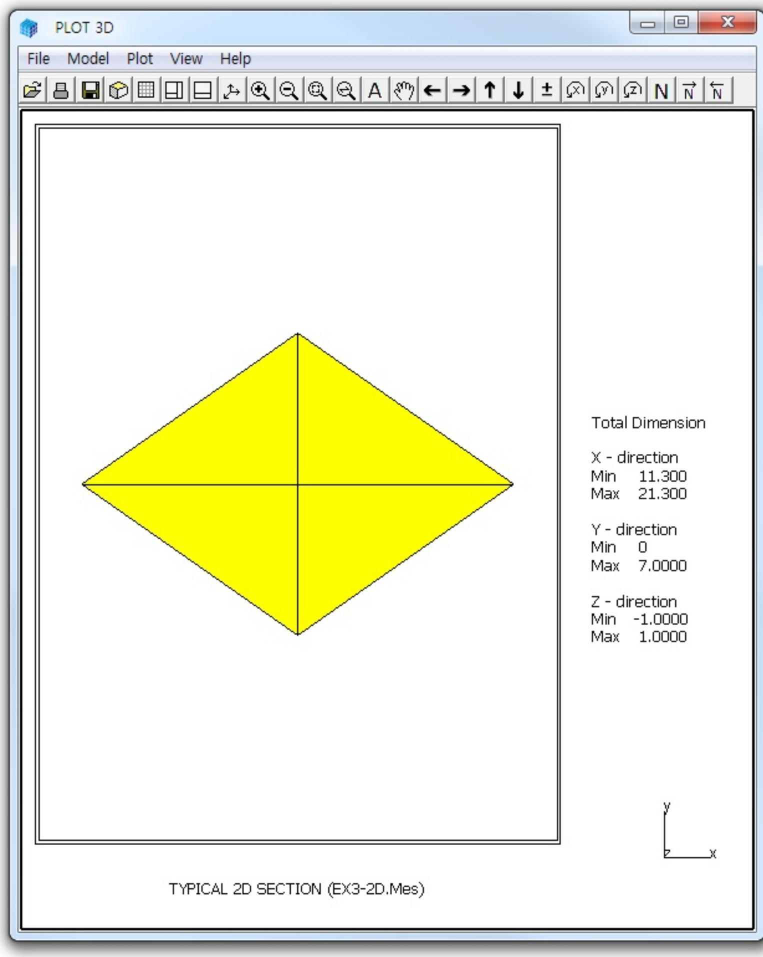


Figure 7.55 Typical 2D section for Example 3

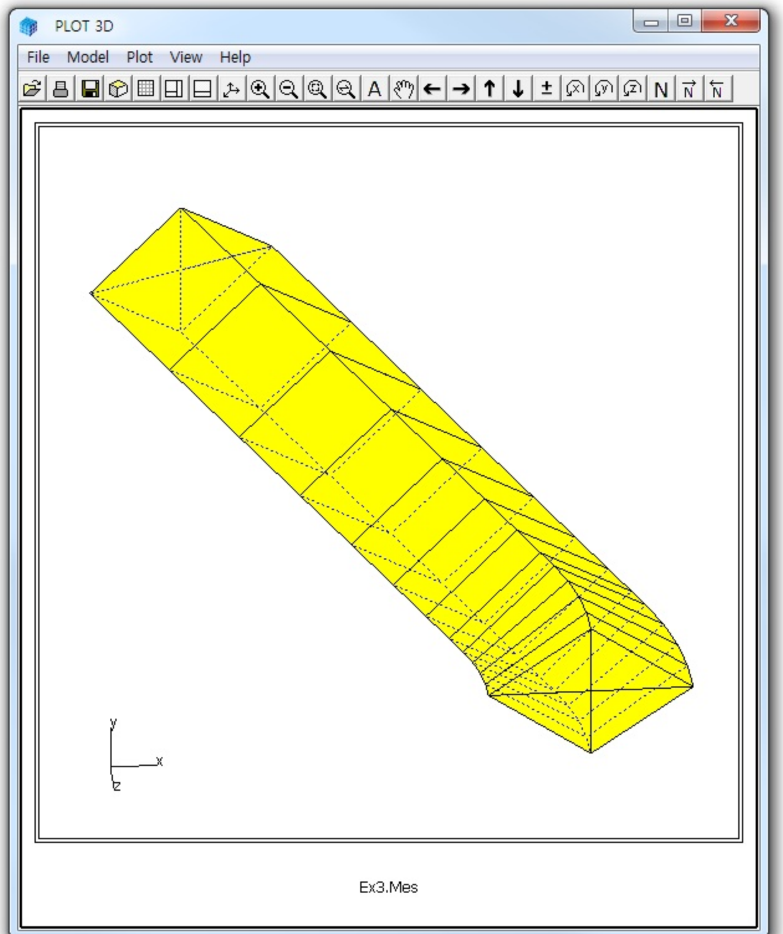


Figure 7.56 Generated 3D mesh for Example 3

7.6.4 Example 4: 3-D Shell Generation

Table 7.26 Listing of input file Ex4.Dat for Example 4

```
* CARD 1.1
* TITLE
  3-D SHELL GENERATION
* CARD 1.2
* NBZ    NBNODE  NSNODE  NSNEL  IBOUND  IPLANE  ICLOSE  CMFAC
  1      2       1       1      3       3       0       1.0
* CARD 1.2.1
* Xleft  Xright  Ybot    Ytop   Zback   Zfront
 -10     10     -10      10    -20      25
* CARD 1.2.2
* Xo     Yo     Zo
  0.0    0.0    0.0
* Xa     Ya     Za
  5.0    0.0   -2.0
* Xb     Yb     Zb
  0.0    10.0   0.0
* CARD 1.3
* IBZ_base IBZ_front IBZ_back
   1        3         3
* CARD 2.1
* NODE   Z      X
   1     30.    0
   2    -30.    0
*==================================
* CARD 3.1
* BLNAME
  BLOCK1
* IBLNO
  1
* CARD 3.3
* I   J   LTYPE
  1   2   0
* CARD 3.4
* NDZ     ALPA
  10      0.3
*==================================
* CARD 4.1
* ITRANB
  0
* END OF DATA
```

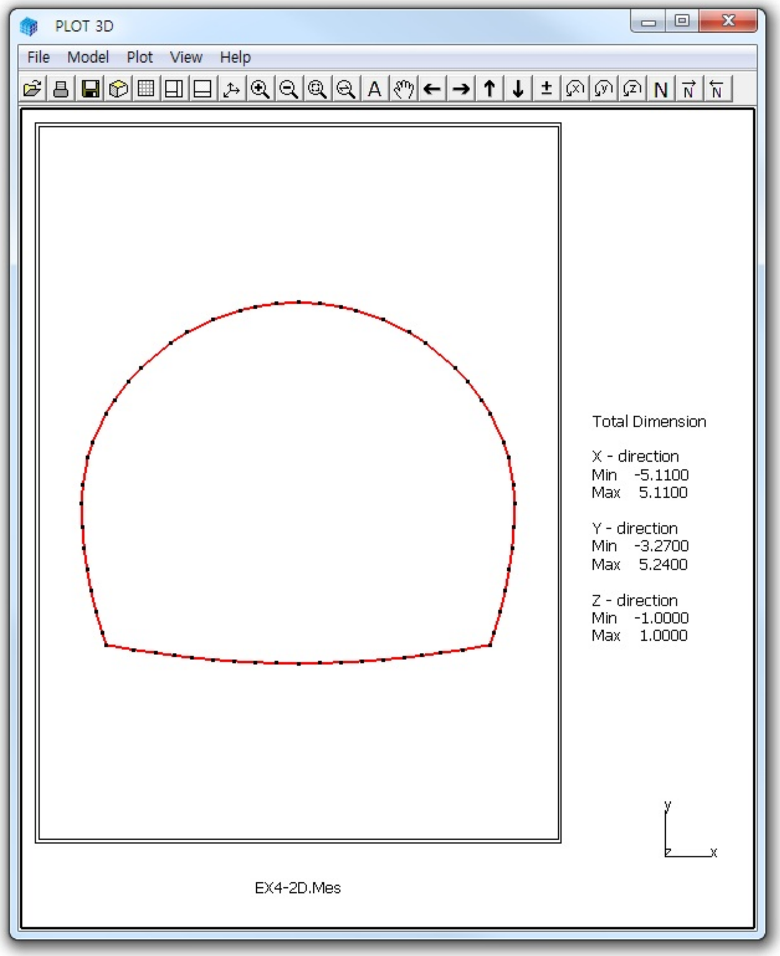


Figure 7.57 Typical 2D section for Example 4

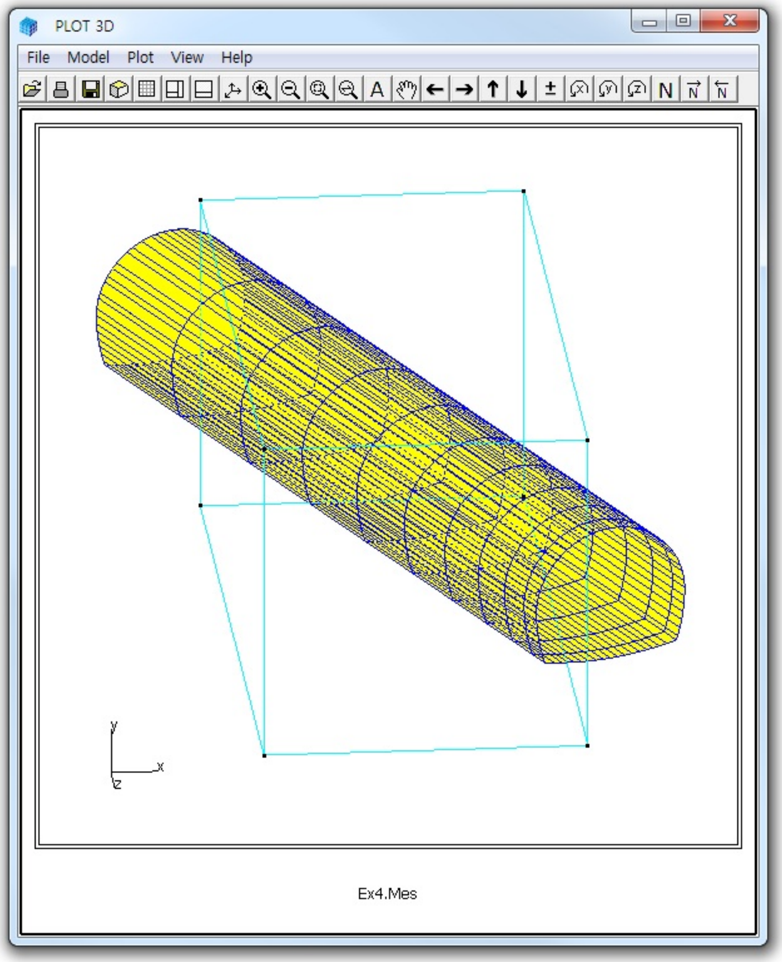


Figure 7.58 Generated 3D mesh for Example 4

7.6.5 Example 5: 3-D Pile Foundation

Table 7.27 Listing of input file Ex5.Dat for Example 5

```
* CARD 1.1
* TITLE
  3-D PILE FOUNDATION
* CARD 1.2
* NBZ  NBNODE  NSNODE  NSNEL  IBOUND  IPLANE  ICLOSE  CMFAC
   5     6       1       1       0       2      0       1.0
* IBZ_base IBZ_front IBZ_back
   1        1          3
* CARD 2.1
* NODE    Zp      Xp
   1     20.00    0
   2     19.50    0
   3     12.50    0
   4     12.25    0
   5     12.00    0
   6      0.00    0
*=======================================
* CARD 3.1
* BLNAME
  BLOCK1
* IBLNO
  1
* CARD 3.3
* I   J   LTYPE  IMATC  IMATB  IMATT
  1   2   0        0      0      0
* CARD 3.4
* NDZ     ALPA  MC1   MC2   MC3   MB   MT
   1      0.5   -1    -4    -5    0    0
*=======================================
* CARD 3.1
* BLNAME
  BLOCK2
* IBLNO
  2
* CARD 3.3
* I   J   LTYPE  IMATC  IMATB  IMATT
  2   3   0        0      0      0
* CARD 3.4
* NDZ     ALPA  MC1   MC2   MC3   MB   MT
   14     0.5    0     0     0    0    0
*=======================================
```

```
* CARD 3.1
* BLNAME
  BLOCK3
* IBLNO
  3
* CARD 3.3
* I   J   LTYPE  IMATC  IMATB  IMATT
  3   4   0       1      0       0
* CARD 3.4
* NDZ     ALPA  MC1  MC2  MC3  MB   MT
   1      0.5    3    4    5    0    0
*========================================
* CARD 3.1
* BLNAME
  BLOCK4
* IBLNO
  4
* CARD 3.3
* I   J   LTYPE  IMATC  IMATB  IMATT
  4   5   0       1      0       0
* CARD 3.4
* NDZ     ALPA  MC1  MC2  MC3  MB   MT
   1      0.5    3    4    0    0    0
*========================================
* CARD 3.1
* BLNAME
  BLOCK5
* IBLNO
  5
* CARD 3.3
* I   J   LTYPE  IMATC  IMATB  IMATT
  5   6   0       6      0       0
* CARD 3.4
* NDZ     ALPA  MC1  MC2  MC3  MB   MT
   8      0.3   -3   -5    0    0    0
*========================================
* CARD 4.1
* ITRANB
  0
* END OF DATA
```

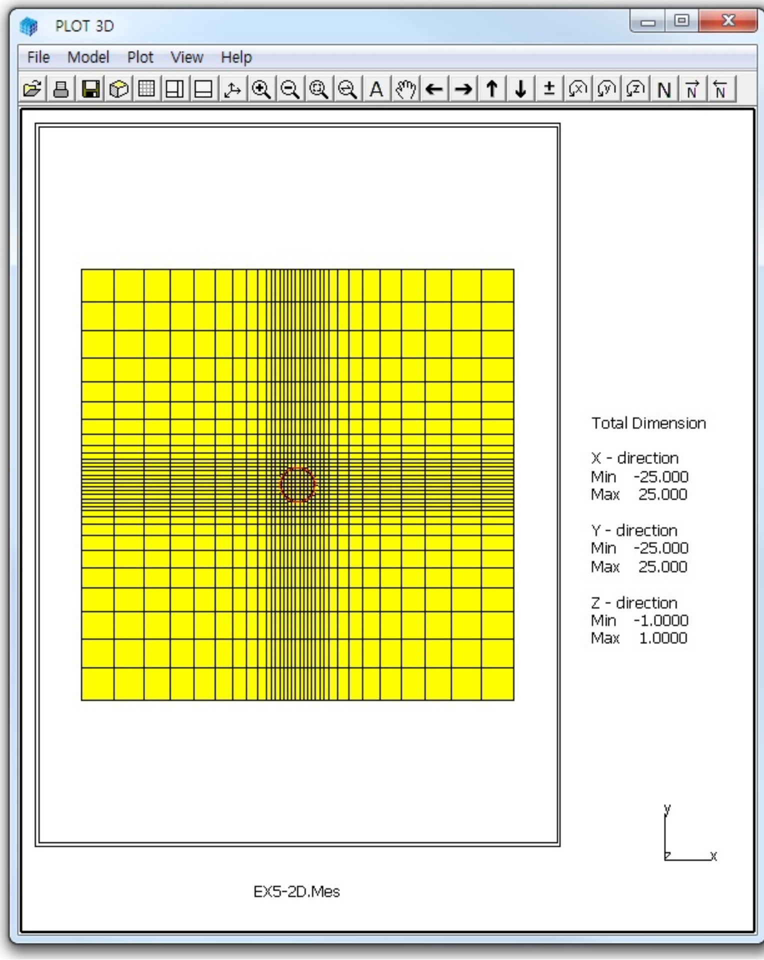


Figure 7.59 Typical 2D section for Example 5

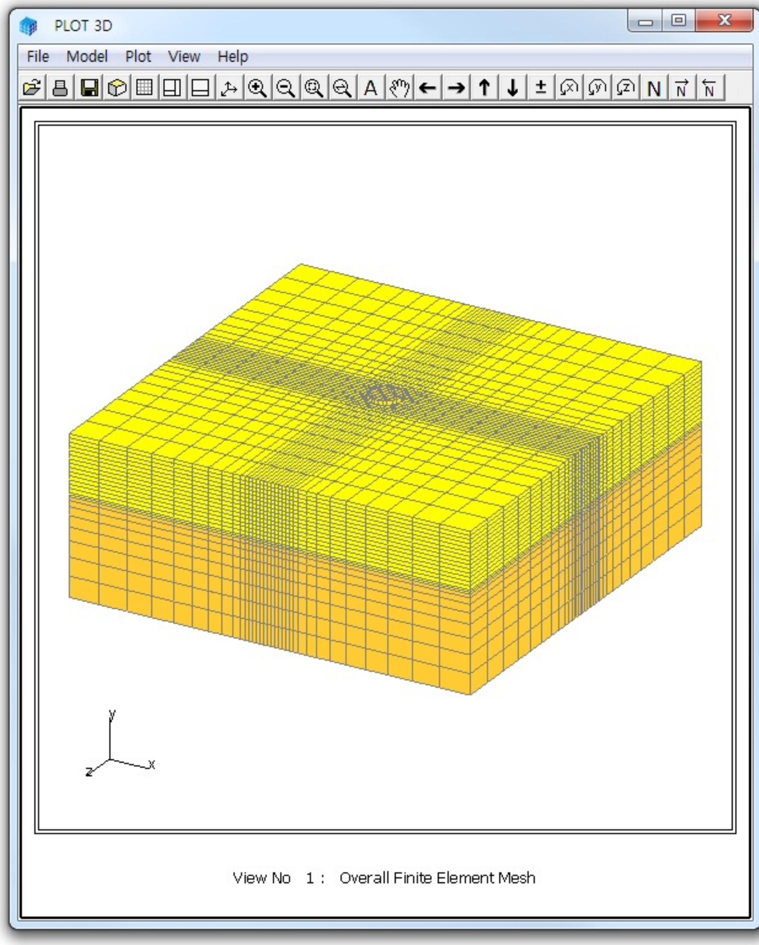


Figure 7.60 Generated 3D mesh for Example 5

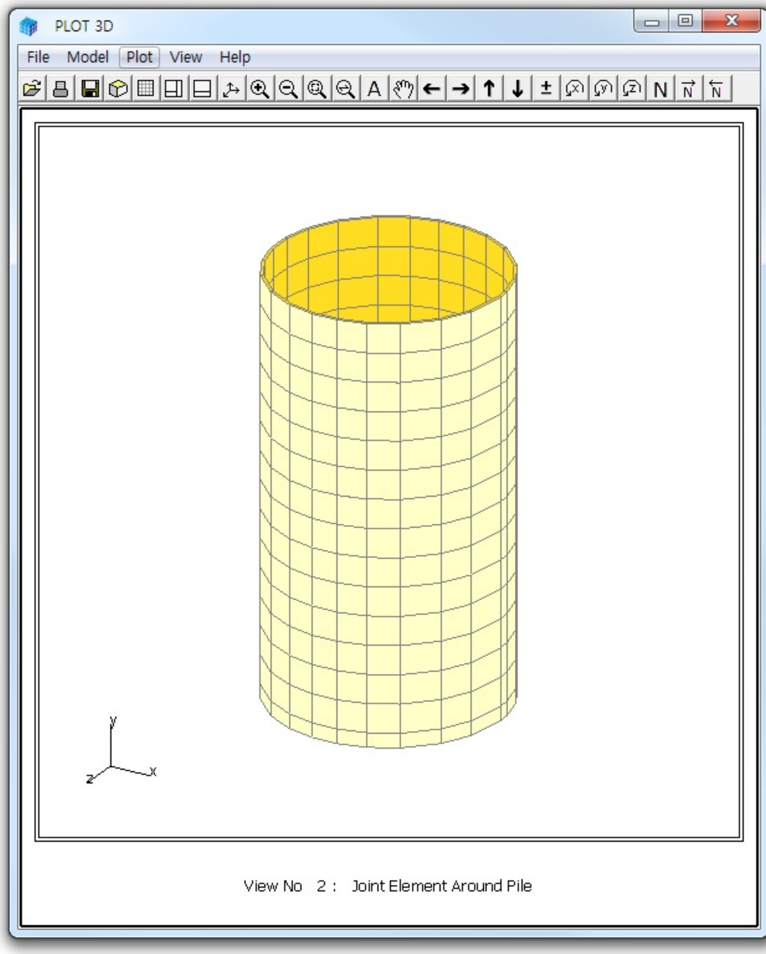


Figure 7.61 Generated joint element for Example 5

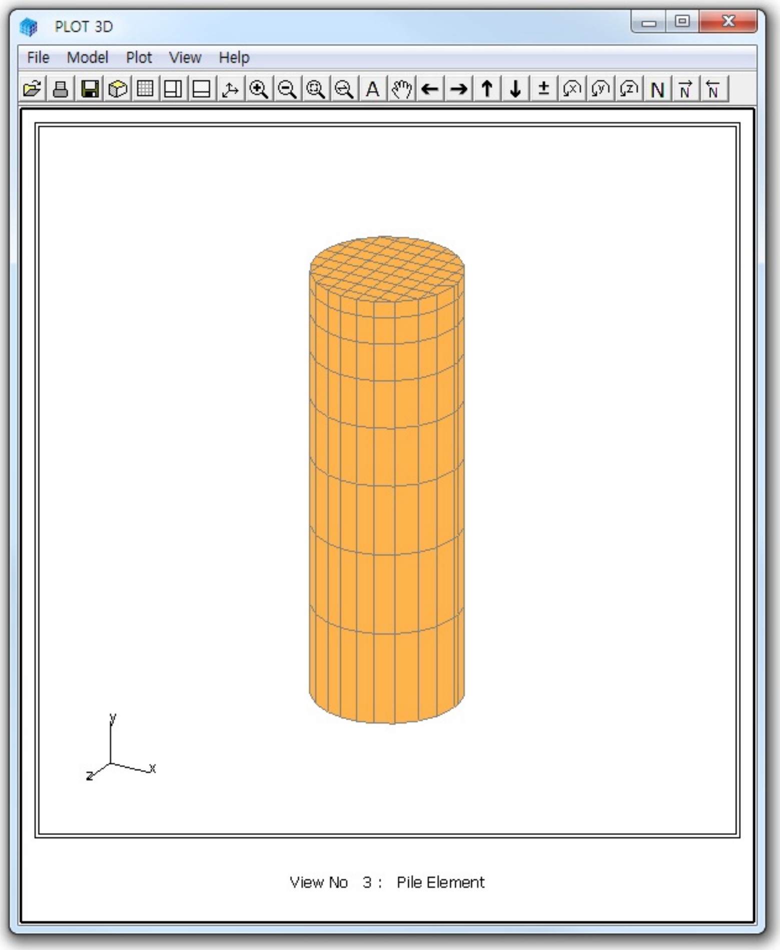


Figure 7.62 Generated pile element for Example 5

7.6.6 Example 6: 3-D Embedded Truss

Table 7.28 Listing of input file Ex6.Dat for Example 6

```
* CARD 1.1
* TITLE
  3-D EMBEDDED TRUSS
* CARD 1.2
* NBZ  NBNODE  NSNODE  NSNEL  IBOUND  IPLANE  ICLOSE  CMFAC
   1     2       29385  27982   0       0      1       1.0
* CARD 1.3
* IBZ_base IBZ_front IBZ_back
   1        3         3
* CARD 2.1
* NODE   Zp     Xp
   1     0.0    1.75
   2     0.0    1.75
*==============================
* CARD 3.1
* BLNAME
  BLOCK1
* IBLNO
  1
* CARD 3.3
* I   J   LTYPE
  1   2   1
* CARD 3.4
* NDZ, ALPA
  16   0.5
* CARD 3.5
* Zo    Xo    R     Tb    Te
  0.0   0.0   1.75  0.0   360.
*==============================
* END
  0
* END OF DATA
```

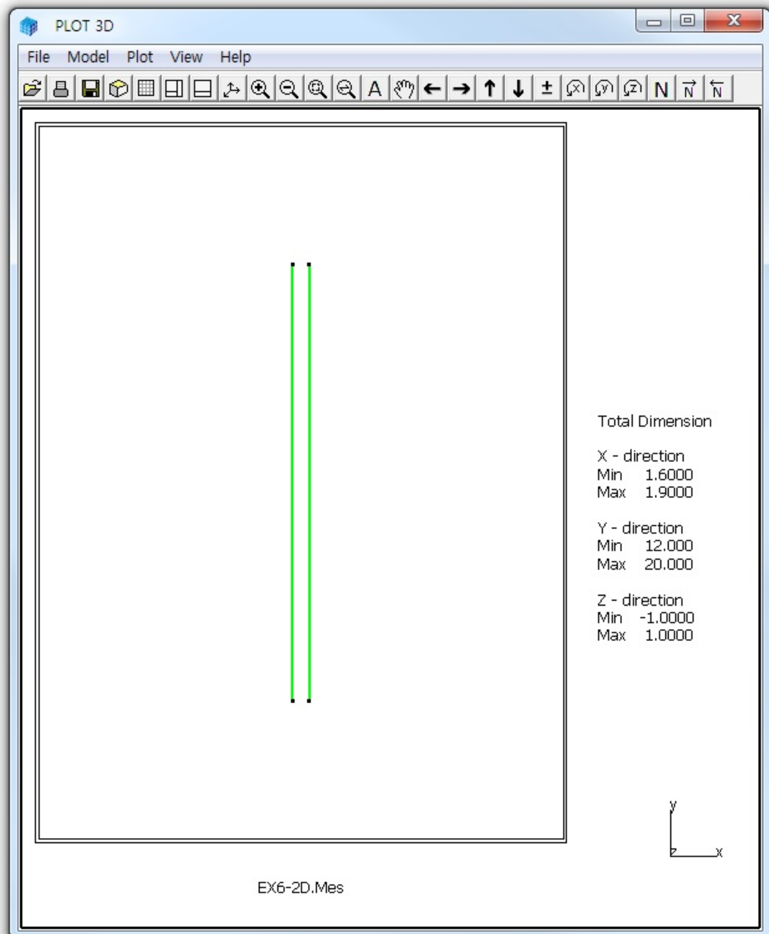


Figure 7.63 Typical 2D section for Example 6

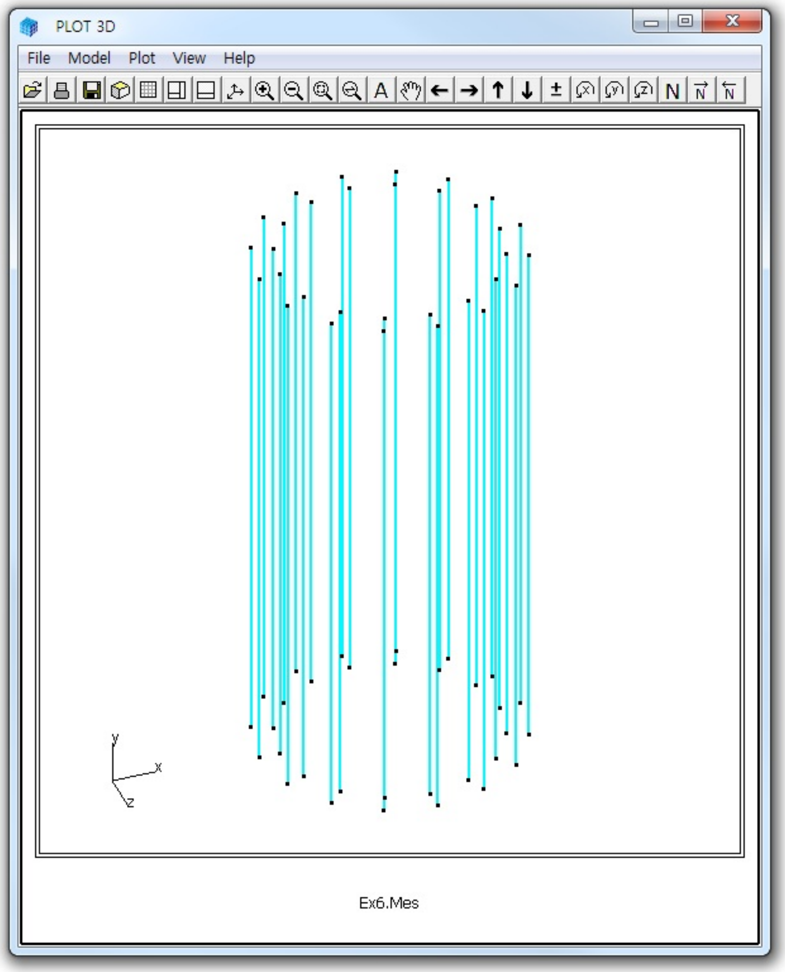


Figure 7.64 Generated 3D mesh for Example 6

7.6.7 Example 7: Pile Foundation Using CIRCLE-2D

Table 7.29 Listing of input file Ex7.Dat for Example 7

```
* CARD 1.1
* TITLE
  PILE FOUNDATION USING CIRCLE-2D (CIR2F_QT.MES)
* CARD 1.2
* NBZ  NBNODE  NSNODE  NSNEL  IBOUND  IPLANE  ICLOSE  CMFAC
   5     6       1       1      0       2       0      1.0
* IBZ_base IBZ_front IBZ_back
   1        1         3
* CARD 2.1
* NODE   Zp      Xp
   1     20.00   0
   2     19.50   0
   3     12.50   0
   4     12.25   0
   5     12.00   0
   6      0.00   0
*==============================================================
* CARD 3.1
* BLNAME
  BLOCK1
* IBLNO
  1
* CARD 3.3
* I   J   LTYPE  IMATC  IMATB  IMATT  NIXCH
  1   2   0       0      0      0      2
* CARD 3.4
* NDZ     ALPA  MC1  MC2  MC3  MB   MT
   1      0.5   -1   -5    0    0    0
* CARD 3.6
* MATNO   NEWNO  NI1  NI2  NI3  NI4  NI5   NI6   NI7   NI8
   3      2
   4      3
*==============================================================
```

```
* CARD 3.1
* BLNAME
  BLOCK2
* IBLNO
  2
* CARD 3.3
* I   J   LTYPE  IMATC  IMATB  IMATT  NIXCH
  2   3   0       0      0      0      3
* CARD 3.4
* NDZ     ALPA  MC1   MC2   MC3   MB    MT
   14     0.5    0     0     0     0     0
* CARD 3.6
* MATNO   NEWNO  NI1  NI2  NI3  NI4  NI5   NI6   NI7   NI8
   1      8
   3      2
   5      5
*==============================================================
* CARD 3.1
* BLNAME
  BLOCK3
* IBLNO
  3
* CARD 3.3
* I   J   LTYPE  IMATC  IMATB  IMATT  NIXCH
  3   4   0       0      0      0      4
* CARD 3.4
* NDZ     ALPA  MC1   MC2   MC3   MB    MT
   1      0.5    0     0     0     0     0
* CARD 3.6
* MATNO   NEWNO  NI1  NI2  NI3  NI4  NI5   NI6   NI7   NI8
   1      9      0    0    0    0    1010  1010  1010  1010
   3      2      0    0    0    0    0     0     0     0
   4      4      0    0    0    0    1010  1010  0     0
   5      6      0    0    0    0    1010  1010  1010  1010
*==============================================================
```

```
* CARD 3.1
* BLNAME
  BLOCK4
* IBLNO
  4
* CARD 3.3
* I   J   LTYPE  IMATC  IMATB  IMATT  NIXCH
  4   5   0        0      0       0      2
* CARD 3.4
* NDZ     ALPA  MC1   MC2   MC3   MB    MT
   1      0.5   -1    -4    -5    0     0
* CARD 3.6
* MATNO   NEWNO  NI1  NI2  NI3  NI4  NI5   NI6   NI7   NI8
   2      -1
   3      -1     0    0    0    0    25    25    0     0
*=============================================================
* CARD 3.1
* BLNAME
  BLOCK5
* IBLNO
  5
* CARD 3.3
* I   J   LTYPE  IMATC  IMATB  IMATT  NIXCH
  5   6   0       0       0       0      2
* CARD 3.4
* NDZ     ALPA  MC1   MC2   MC3   MB    MT
   8      0.3   -2    -3    -4    0     0
* CARD 3.6
* MATNO   NEWNO  NI1  NI2  NI3  NI4  NI5   NI6   NI7   NI8
   1      7
   5      7
*=============================================================
* CARD 4.1
* ITRANB
  0
* END OF DATA
```

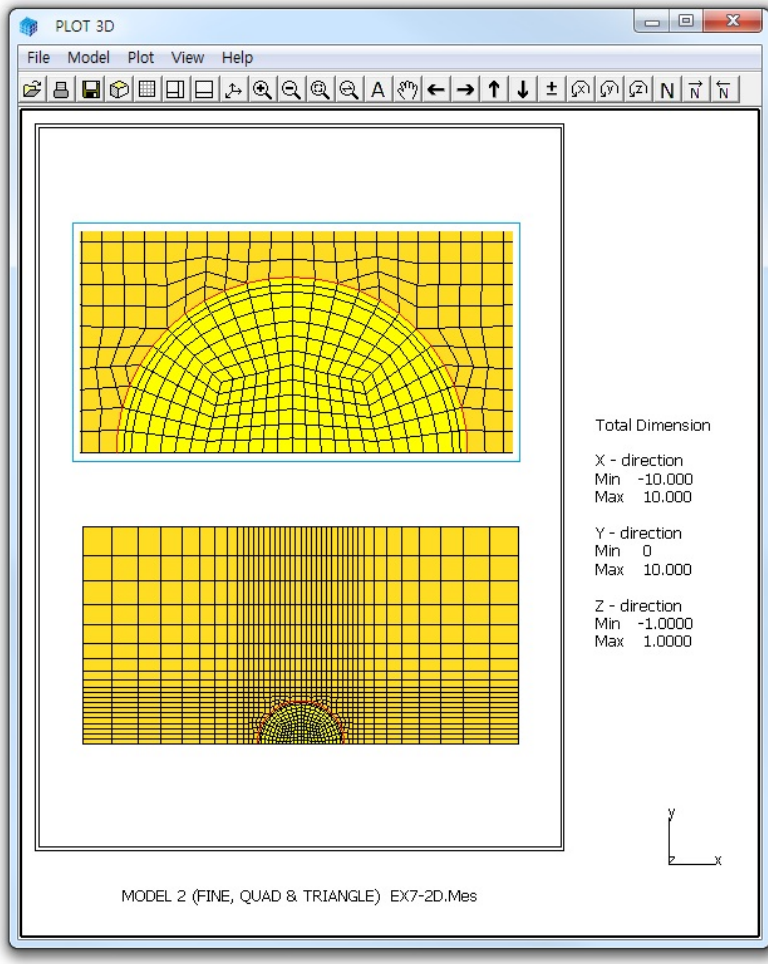


Figure 7.65 Typical 2D section for Example 7

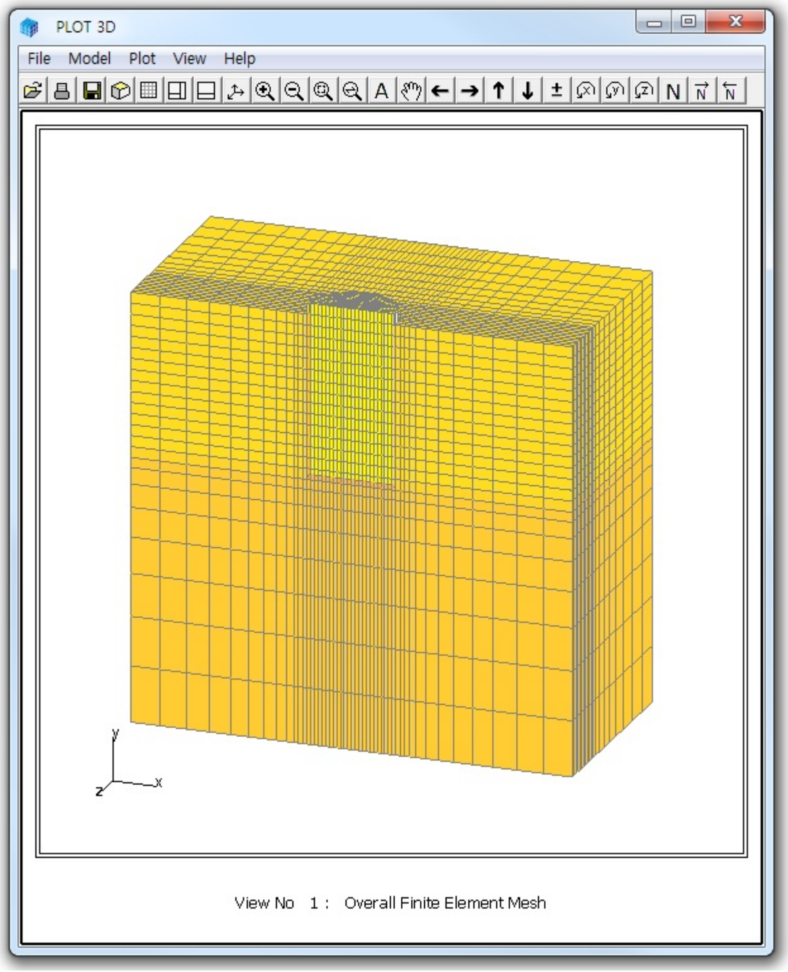


Figure 7.66 Generated 3D mesh for Example 7

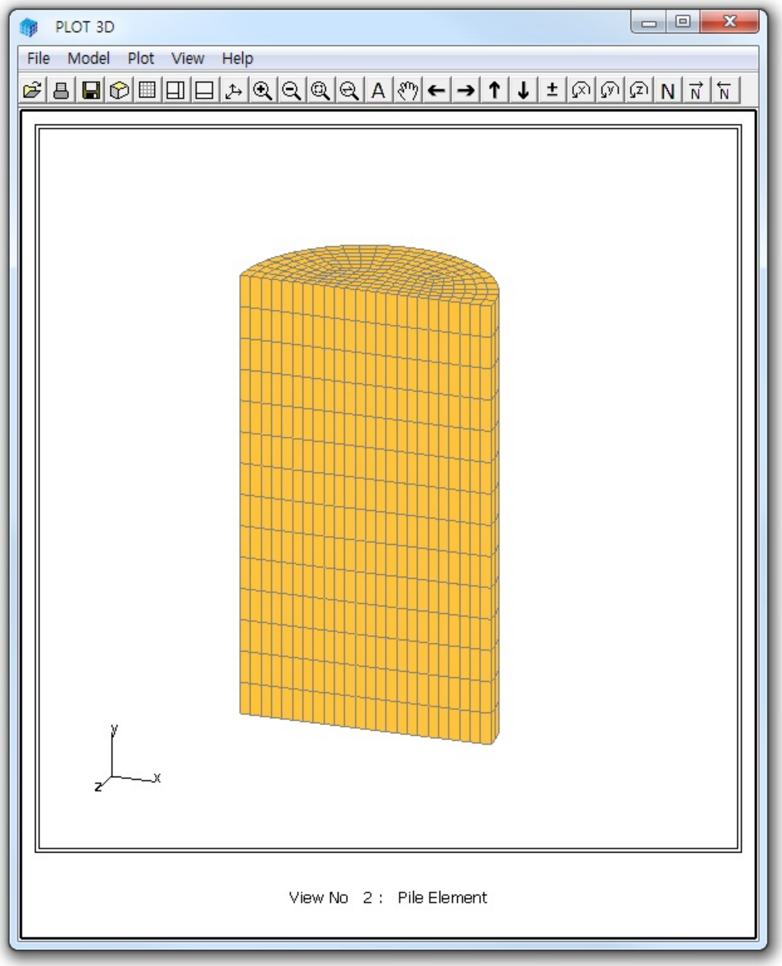


Figure 7.67 Generated pile element for Example 7

7.7 PILE-3D

PILE-3D is the special pre-processor which can be used to generate all input files required for pile foundations analysis. It can generate Concrete Pile with Anchor Bolts or Steel Pipe with Concrete Cap. Input parameters of PILE-3D have been described in detail in Section 7.8 of User's Manual.

Output files from PILE-3D include Pile3D.Dat, Pile3D.Mes, Pile3D.Man, and Pile3D.Pos. You can modify such generated files as you want.

PILE-3D can be selected in the following order.

Run → Mesh Generator → PreSmap → Pile 3D

When you finish the execution of PILE-3D, select PLOT-3D to plot the generated mesh.

7.7.1 Example 1: Concrete Pile with Anchor Bolts

Example 1 is to generate Concrete Pile with Anchor Bolts.
Detailed input data is listed in Table 7.30. Figure 7.68 shows schematic section view of concrete pile and soil profile around the pile foundation subjected to vertical and horizontal forces along with overturning moment.

Figure 7.69 shows generated finite element meshes around the pile foundation. Anchor bolts are modeled by truss elements. Interfaces between concrete pile and surrounding soils are modeled by joint elements which allows slippage and debonding.

Refer to other output files from PILE-3D:
Pile.Dat (Project), Pile.Mes (Mesh), Pile.Man (Main) and Pile.Pos (Post)

7.7.2 Example 2: Steel Pipe with Concrete Cap

Example 2 is to generate Steel Pile with Concrete Cap.
Input file is very similar to Example 1 except that steel pipe is used instead of concrete block. Figure 7.70 shows generated finite element meshes around the steel pipe foundation. Steel pipe is modeled by shell elements.

Table 7.30 Listing of input file EX1.dat

```
* Title
* Card 1.1
  Example 1
*
* Stitle
* Card 1.2
  Concrete Pile with Anchor Bolt
*
* Pile Dimension
* Card 2.1
* D          Ht         Hs         Hw          Nt
  1.5,       8,         0.2,       10,         20
*
* Steel Pipe
* Card 2.2
* Ep         Vp         tp
  2E+07,     0.3,       0.
*
* Reinforcing Bars
* Card 2.3
* Nr         dtop       dbot
  3,         10,        10
*
* Card 2.4-1
* Db         db         Nb
  32,        100,       43
* Card 2.4-2
* Db         db         Nb
  32,        300,       32
* Card 2.4-3
* Db         db         Nb
  32,        400,       26
*
* Concrete Property
* Card 2.5
* Ec         Vc         Phi        C           T          Gama
  2856759,   0.2,       30,        500,        300,       2.4
*
* Rebar Property
* Card 2.6
* Er         Sigy
  2E+07,     40000
*
```

```
* Pile Base Interface Property
* Card 2.7
* Eb          Gb         Phi        C           T          tb
  100000,     1000,      10,        0,          0.001,     0.1
*
* Soil/Rock Layers
*
* Card 3.1
* NLAYER
  4
*
* Card 3.2
* LayerNo     ModelNo    H          Gama        Ko
  1,          3,         2,         2,          0.46
* Pile Side Interface
* Card 3.3
* Ej          Gj         Phij       Cj          Tj         tj
  100000,     1000,      10,        0,          0.001,     0.1
* Card 3.4.3
* E           V          Phi        C           T
  1000,       0.3,       33,        0,          0.1
*
* Card 3.2
* LayerNo     ModelNo    H          Gama        Ko
  2,          3,         4,         2.2,        0.43
* Pile Side Interface
* Card 3.3
* Ej          Gj         Phij       Cj          Tj         tj
  100000,     1000,      10,        0,          0.001,     0.1
* Card 3.4.3
* E           V          Phi        C           T
  3000,       0.3,       35,        0,          0.1
*
* Card 3.2
* LayerNo     ModelNo    H          Gama        Ko
  3,          3,         3,         2.2,        0.34
* Pile Side Interface
* Card 3.3
* Ej          Gj         Phij       Cj          Tj         tj
  100000,     1000,      10,        0,          0.001,     0.1
* Card 3.4.3
* E           V          Phi        C           T
  7000,       0.25,      41,        0,          0.1
*
* Card 3.2
* LayerNo     ModelNo    H          Gama        Ko
  4,          3,         5,         2.2,        0.34
```

```
* Pile Side Interface
* Card 3.3
* Ej          Gj         Phij       Cj          Tj         tj
  100000,     1000,      10,        0,          0.001,     0.1
* Card 3.4.3
* E           V          Phi        C           T
  8000,       0.25,      41,        0,          0.1
*
* Card 4.1
* Fv          Fh         M          NumStep
  31,         54,        812,       5
*
* Anchor Bolt
* Card 5.1
* Da          da         La         Nbolt       Ea         Sigya
  50,         200,       4,         32,         2E+07,     40000
*
* Finite Element Mesh on Plan View
* Card 6.1
* FineMesh    NearMesh   Ndiv       BH          BV
  0,          1,         10,        15.,        15.
*
* End of Data
```

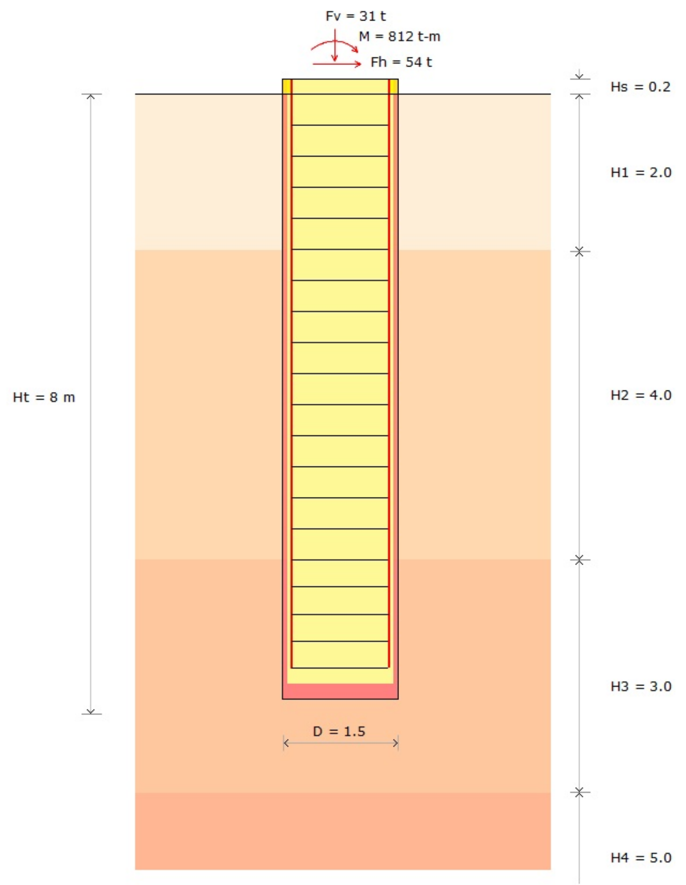


Figure 7.68 Schematic section view of concrete pile for Example 1

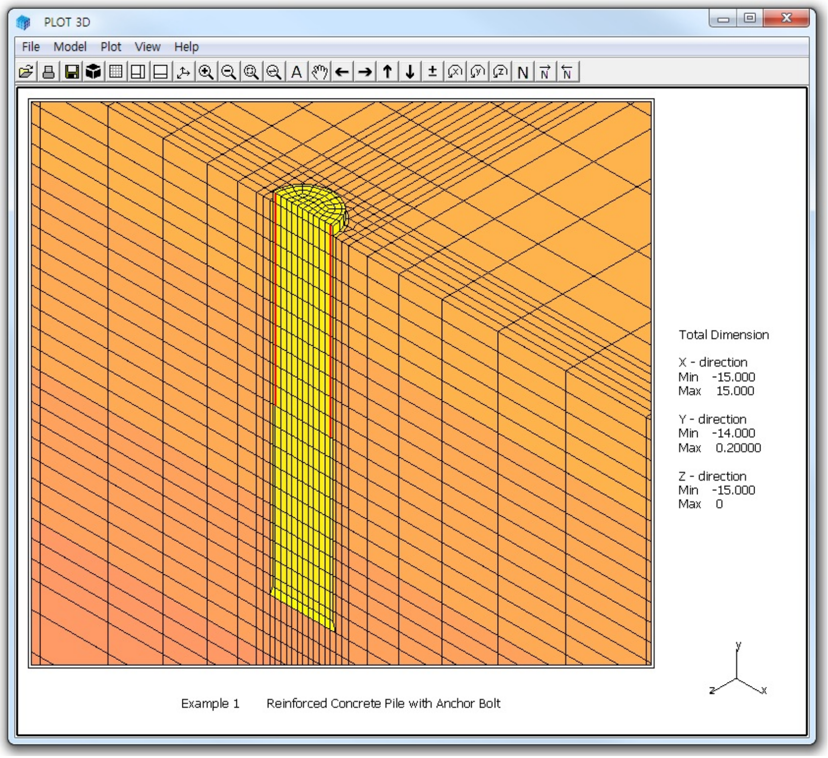


Figure 7.69 Generated finite element meshes for concrete pile

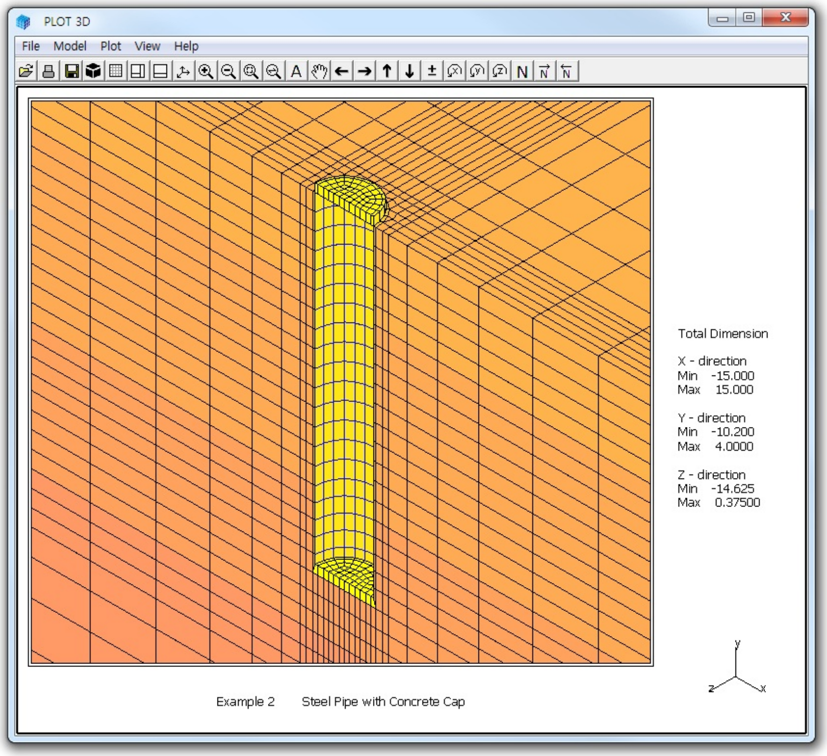


Figure 7.70 Generated finite element meshes for steel pipe

7.8 PRESMAP-GP

PRESMAP-GP is the general purpose pre-processor which can be used to generate coordinates, element indexes, and boundary codes of various geometries modeled by truss, beam, shell or continuum elements. Input parameters of PRESMAP-GP have been described in detail in Section 7.9 of User's Manual.

Input file for PRESMAP-GP is also called block mesh file which can be generated or modified by Block Mesh Generator described in Section 6 of User's Manual.

PRESMAP-GP can be selected in the following order.

Run → Mesh Generator → PreSmap → Presmap GP

When you finish the execution of PRESMAP-GP, select PLOT-3D to plot the generated mesh.

7.8.1 Example 1: 3-D Line/Surface/Volume Blocks

Example 1 shows you how Beam, Shell and Continuum elements are generated using various types of blocks. There are a total of 5 blocks consisting of a line block, a triangle surface block, and a quad surface block, a prism volume block, and a hexahedron volume block. Detailed block information is listed in Table 7.31.

Input block meshes and generated finite element meshes are presented in the following order:

Input Block Meshes
Figure 7.71 Node and block numbers
Figure 7.72 Block numbers for line and surface blocks
Figure 7.73 Material numbers for line and surface blocks
Figure 7.74 Block numbers for volume blocks
Figure 7.75 Material numbers for volume blocks
Figure 7.76 Skeleton boundary codes
Figure 7.77 Fluid boundary codes
Figure 7.78 Rotation boundary codes

Generated Finite Element Meshes

Figure 7.79 Node and element numbers
Figure 7.80 Element numbers for beam and shell elements
Figure 7.81 Material numbers for beam and shell elements
Figure 7.82 Element numbers for continuum elements
Figure 7.83 Material numbers for continuum elements
Figure 7.84 Skeleton boundary codes
Figure 7.85 Fluid boundary codes
Figure 7.86 Rotation boundary codes

Table 7.31 Listing of input file EX1.Meb

```
StartPresmap
VersionNo = 7.000
* CARD 1.1
* TITLE
  3-D LINE/SURFACE/VOLUME ELEMENT GENERATION
* CARD 1.2
* NBLOCK    NBNODE    NSNODE    NSNEL    IGBND    ISMAP    CMFAC    ICOMP
  5         12        1         1        0        3        1.000    1
*======================================================================
* CARD 1.3
* Global Outer Surface Boundary
* X - Right Boundary
* ISG  ISX  ISY  ISZ  IFG  IFX  IFY  IFZ  IRG  IRX  IRY  IRZ
  3    0    0    0    0    0    0    0    0    0    0    0
* X - Left Boundary
* ISG  ISX  ISY  ISZ  IFG  IFX  IFY  IFZ  IRG  IRX  IRY  IRZ
  3    0    0    0    0    0    0    0    0    0    0    0
* Y - Top Boundary
* ISG  ISX  ISY  ISZ  IFG  IFX  IFY  IFZ  IRG  IRX  IRY  IRZ
  4    1    1    0    4    1    1    1    0    0    0    0
* Y - Bottom Boundary
* ISG  ISX  ISY  ISZ  IFG  IFX  IFY  IFZ  IRG  IRX  IRY  IRZ
  3    0    0    0    4    1    0    1    0    0    0    0
* Z - Front Boundary
* ISG  ISX  ISY  ISZ  IFG  IFX  IFY  IFZ  IRG  IRX  IRY  IRZ
  3    0    0    0    0    0    0    0    4    0    1    0
* Z - Back Boundary
* ISG  ISX  ISY  ISZ  IFG  IFX  IFY  IFZ  IRG  IRX  IRY  IRZ
  3    0    0    0    0    0    0    0    4    1    0    1
*======================================================================
```

```
* CARD 1.4
* Automatic Finite Element Generation Control Parameters
* Min Length    Max Element
  1.000         10000
*=======================================================================
* CARD 2.1
* NODE  X        Y        Z
  1     4.0      6.5      0.0
  2     0.0      2.0      0.0
  3     5.9      0.8      0.0
  4     7.0      7.0      0.0
  5     7.0      1.0      0.0
  6     5.72     3.87     0.0
  7     4.0      6.5     -1.0
  8     0.0      2.0     -1.0
  9     5.9      0.8     -1.0
 10     7.0      7.0     -1.0
 11     7.0      1.0     -1.0
 12     5.72     3.87    -1.0
*=======================================================================
StartBlock
* CARD 3.0
* IBETYPE
  1
* CARD 3.1
* BLNAME
  BLOCK 1
* CARD 3.2
* ICOORD IMODE  ILAG
  1      0      0
* CARD 3.3
* I1     I2
  1      3
* M3
  0
* M4
  0
* M5     M6     M7
  0      0      0
* CARD 3.4.1
* NBOUND
  2
* CARD 3.4.2
* IBTYPE ISX    ISY    ISZ    IFX    IFY    IFZ    IRX    IRY    IRZ
  3      0      0      0      1      1      1      1      1      1
  4      0      0      1      1      1      1      1      0      0
```

```
* CARD 3.5
* MATNO  NDX
  1      4
EndBlock
*=======================================================================
StartBlock
* CARD 3.0
* IBETYPE
  -2
* CARD 3.1
* BLNAME
  BLOCK 2
* CARD 3.2
* ICOORD IMODE  ILAG
  1      0      1
* CARD 3.3
* I1     I2     I3
  1      2      3
* M4     M5     M6
  0      0      0
* M7
  0
* M8     M9     M10
  0      0      0
* CARD 3.4.1
* NBOUND
  4
* CARD 3.4.2
* IBTYPE ISX    ISY    ISZ    IFX    IFY    IFZ    IRX    IRY    IRZ
  1      0      0      0      0      0      0      1      1      1
  2      1      1      1      0      0      0      1      1      1
  3      0      1      1      1      1      1      0      0      0
  4      1      1      1      1      1      1      1      1      1
* CARD 3.5
* MATNO  NDXY
  4      4
EndBlock
*=======================================================================
StartBlock
* CARD 3.0
* IBETYPE
  2
* CARD 3.1
* BLNAME
  BLOCK 3
* CARD 3.2
* ICOORD IMODE  ILAG
  1      0      1
```

```
* CARD 3.3
* I1     I2     I3     I4
  4      1      3      5
* M5     M6     M7     M8
  0      0      0      0
* M9
  0
* M10    M11    M12
  0      0      0
* CARD 3.4.1
* NBOUND
  1
* CARD 3.4.2
* IBTYPE ISX    ISY    ISZ    IFX    IFY    IFZ    IRX    IRY    IRZ
  5      1      0      1      0      1      0      1      0      1
* CARD 3.5
* MATNO  NDX    NDY
  2      1      4
* NT1    NT2    NT3    NT4
  0      0      0      0
* MAT1   MAT2   MAT3   MAT4
  0      0      0      0
EndBlock
*=======================================================================
StartBlock
* CARD 3.0
* IBETYPE
  -3
* CARD 3.1
* BLNAME
  BLOCK 4
* CARD 3.2
* ICOORD IMODE  ILAG
  1      0      1
* CARD 3.3
* I1   I2   I3   I4   I5   I6
  1    2    3    7    8    9
* M7   M8   M9   M10  M11  M12  M13  M14  M15  M16  M17
  0    0    0    0    0    0    0    0    0    0    0
* M18  M19  M20  M21
  0    0    0    0
* M22  M23  M24
  0    0    0
* CARD 3.4.1
* NBOUND
  3
```

```
* CARD 3.4.2
* IBTYPE ISX     ISY     ISZ     IFX     IFY     IFZ     IRX     IRY     IRZ
  1      1       1       1       0       0       0       1       1       1
  3      0       0       0       1       1       1       0       0       0
  4      1       1       0       0       1       1       1       0       0
* CARD 3.5
* MATNO  NDXY    NDZ     KS      KF
  1      4       1       0       1
EndBlock
*======================================================================
StartBlock
* CARD 3.0
* IBETYPE
  3
* CARD 3.1
* BLNAME
  BLOCK 5
* CARD 3.2
* ICOORD IMODE  ILAG
  1      0       1
* CARD 3.3
* I1   I2   I3   I4   I5   I6   I7   I8
  4    1    3    5    10   7    9    11
* M9   M10  M11  M12  M13  M14  M15  M16  M17  M18  M19  M20
  0    0    0    0    0    0    0    0    0    0    0    0
* M21  M22  M23  M24  M25  M26  M27
  0    0    0    0    0    0    0
* M28  M29  M30
  0    0    0
* CARD 3.4.1
* NBOUND
  3
* CARD 3.4.2
* IBTYPE ISX     ISY     ISZ     IFX     IFY     IFZ     IRX     IRY     IRZ
  1      0       0       0       0       0       0       0       0       0
  2      0       0       0       1       1       1       0       0       0
  3      1       1       0       0       1       1       1       0       0
* CARD 3.5
* MATNO  NDX     NDY     NDZ     KS      KF
  3      1       4       1       0       1
* NT1    NT2     NT3     NT4
  0      0       0       0
* MAT1   MAT2    MAT3    MAT4
  0      0       0       0
EndBlock
*======================================================================
EndOfLastBlock
```

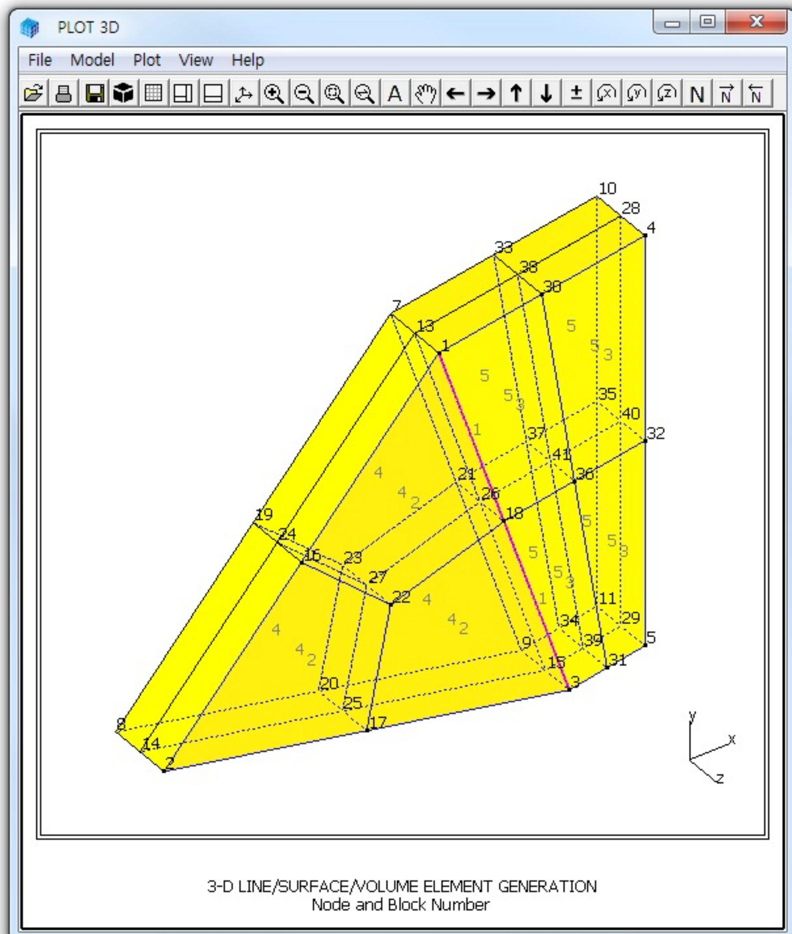


Figure 7.71 Node and block numbers for Example 1

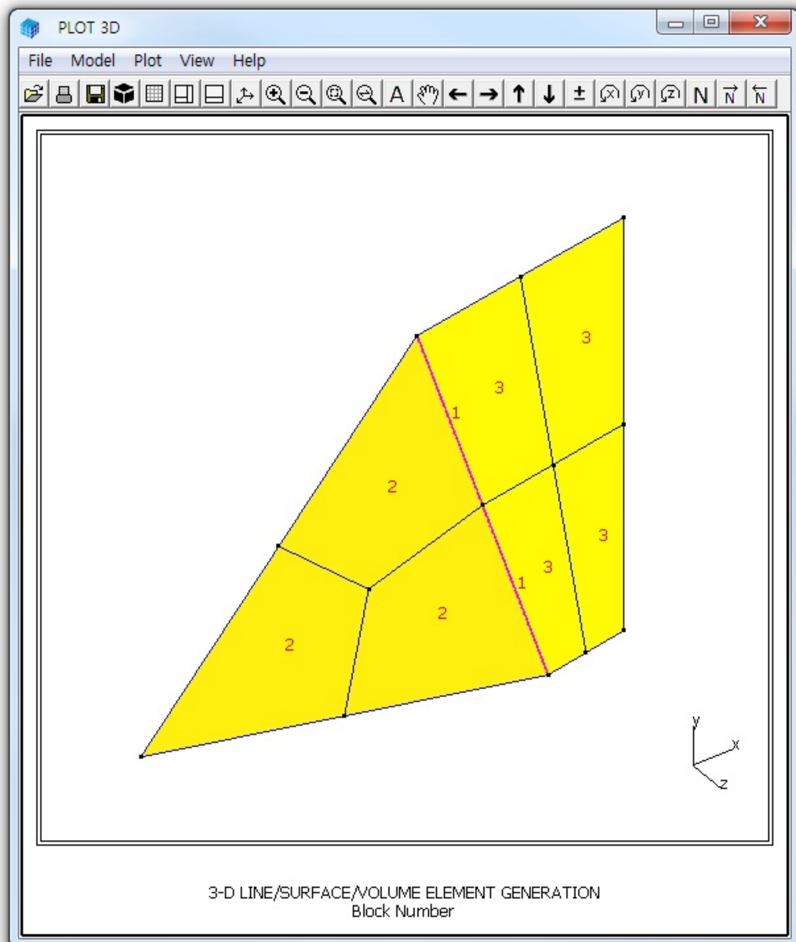


Figure 7.72 Block numbers for line and surface blocks

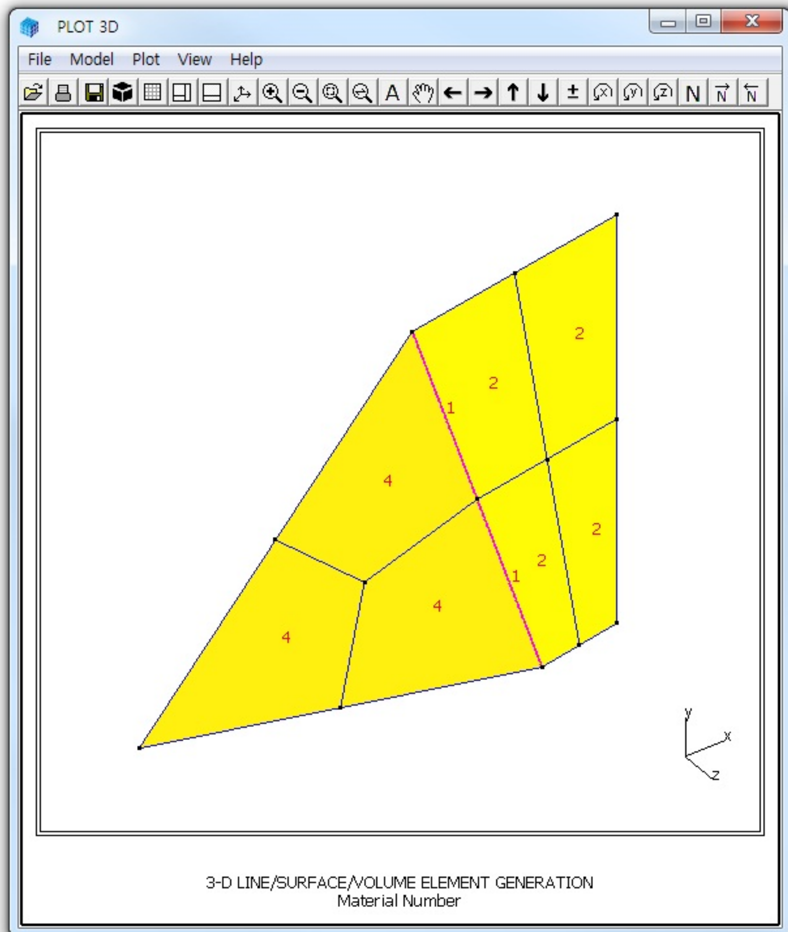


Figure 7.73 Material numbers for line and surface blocks

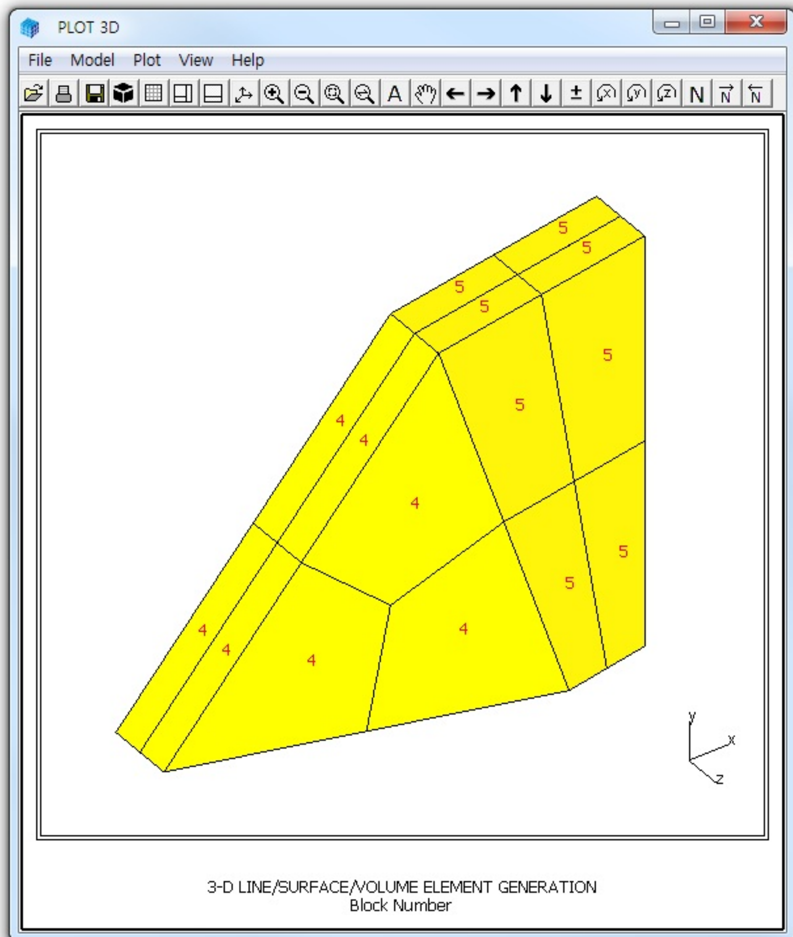


Figure 7.74 Block numbers for volume blocks

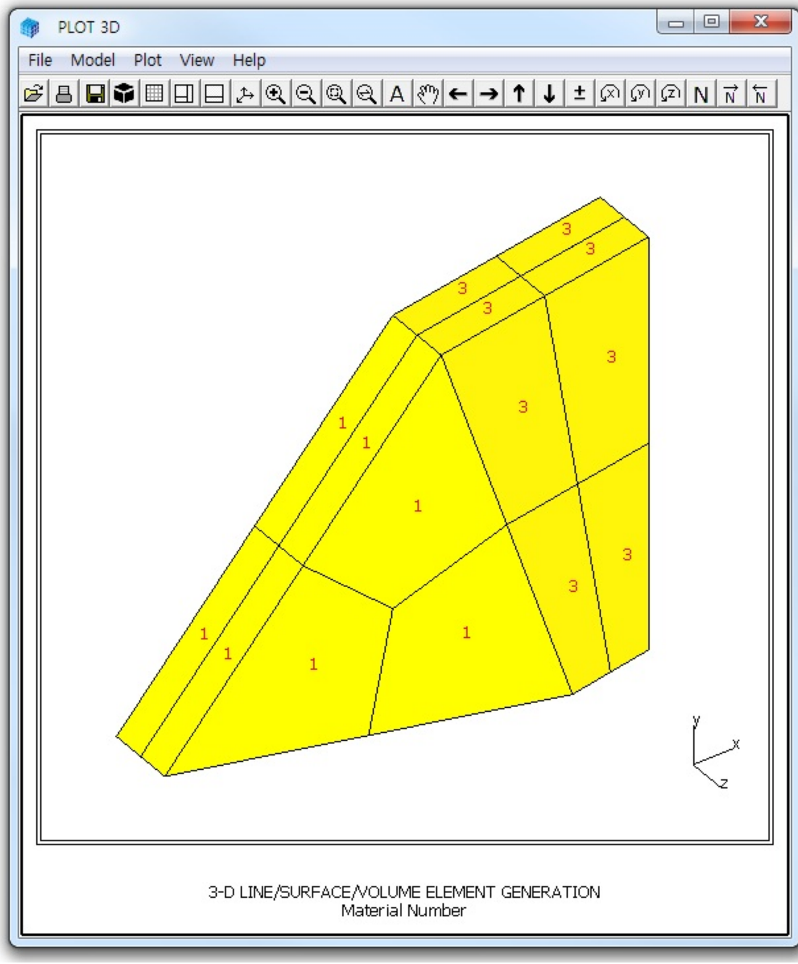


Figure 7.75 Material numbers for volume blocks

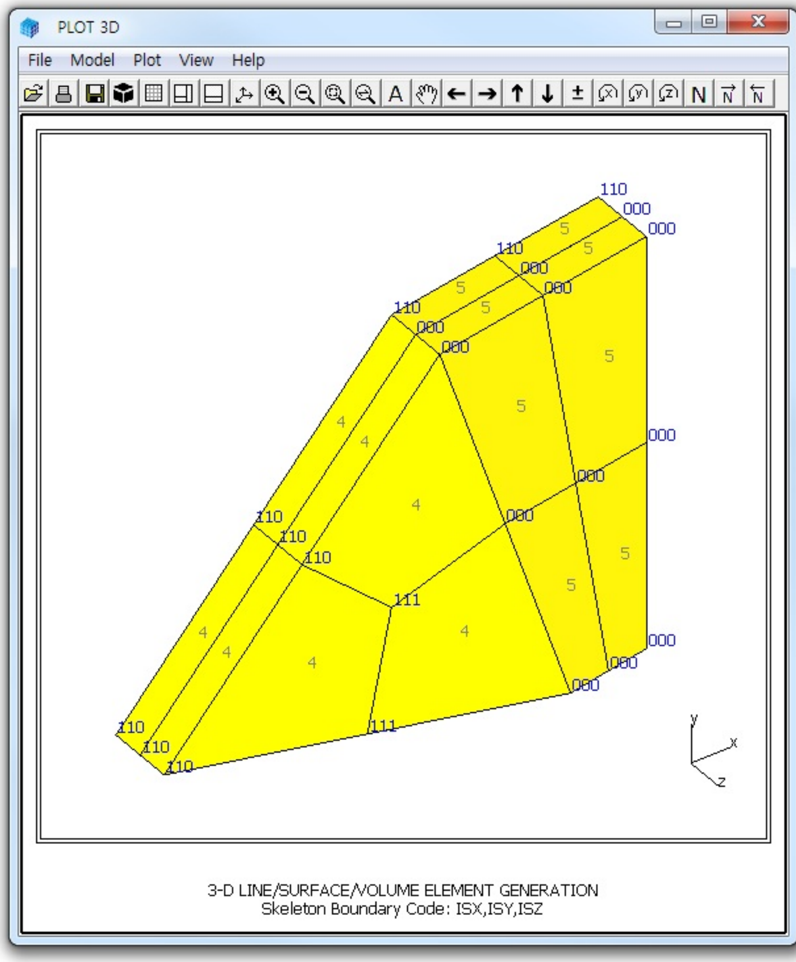


Figure 7.76 Skeleton boundary codes

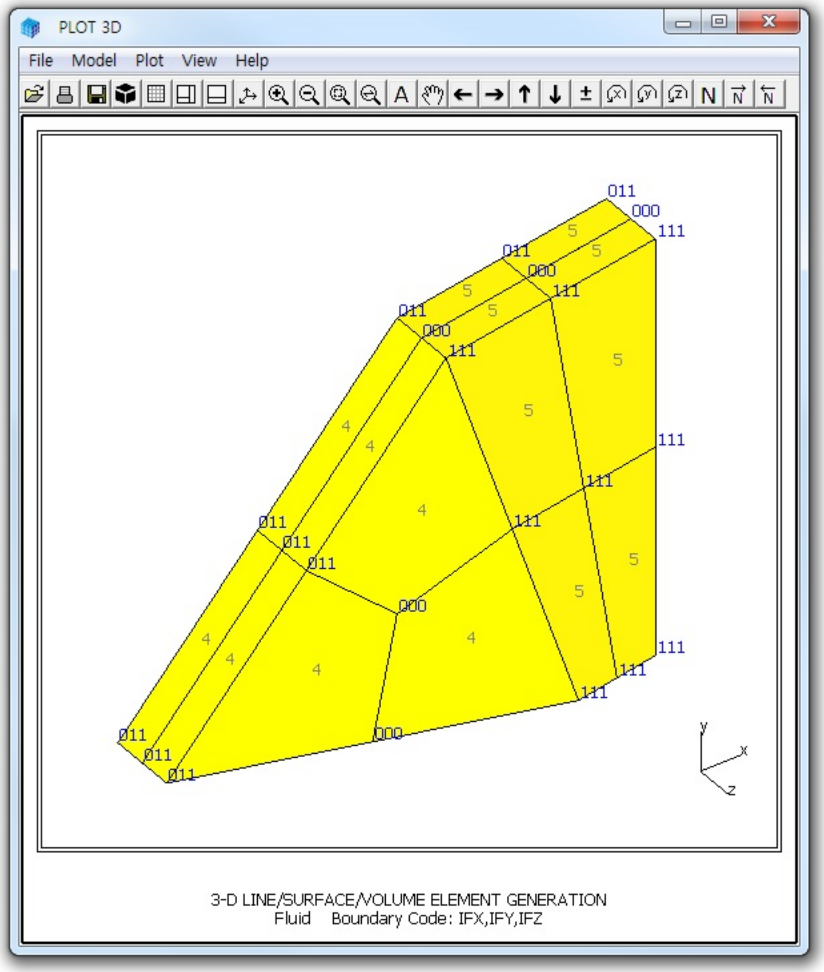


Figure 7.77 Fluid boundary codes

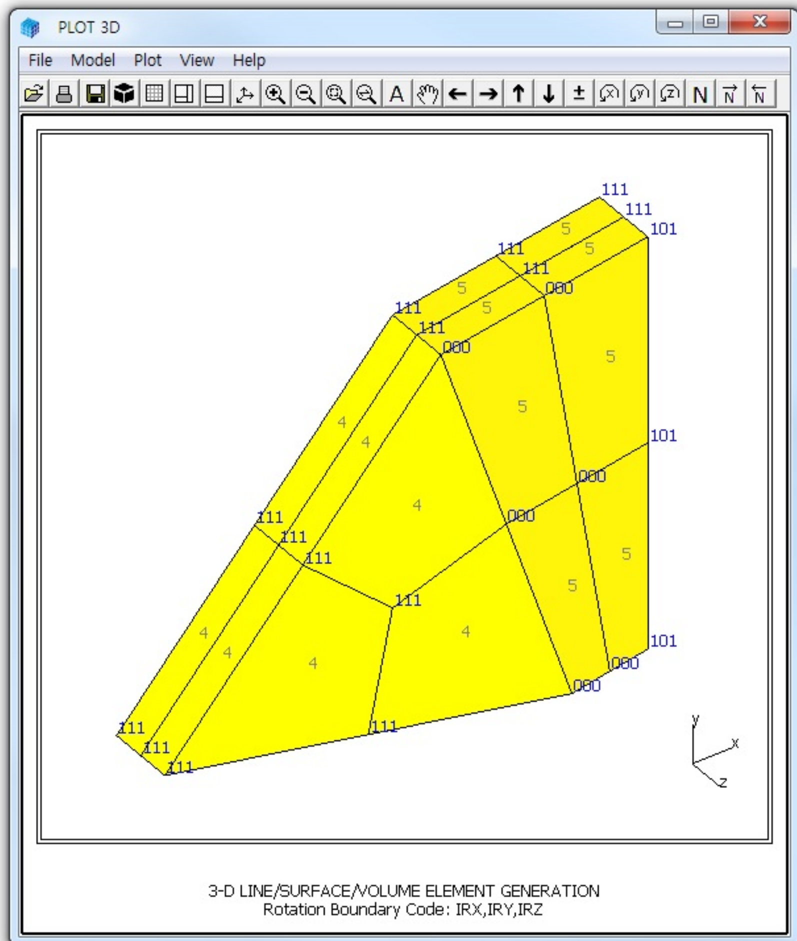


Figure 7.78 Rotation boundary codes

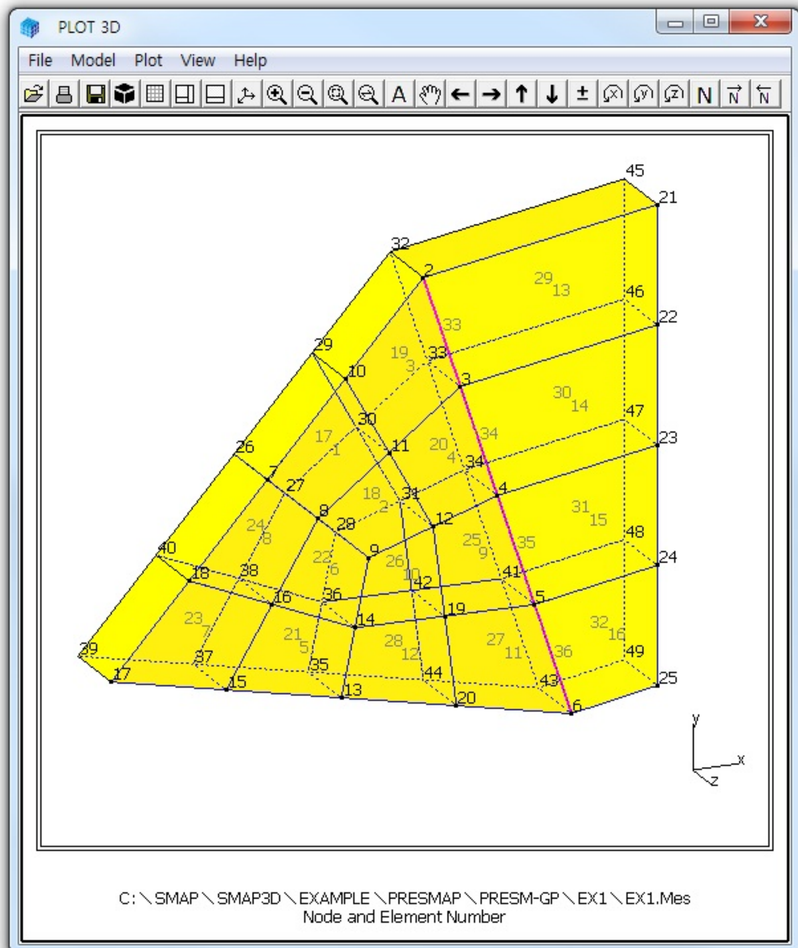


Figure 7.79 Node and element numbers for Example 1

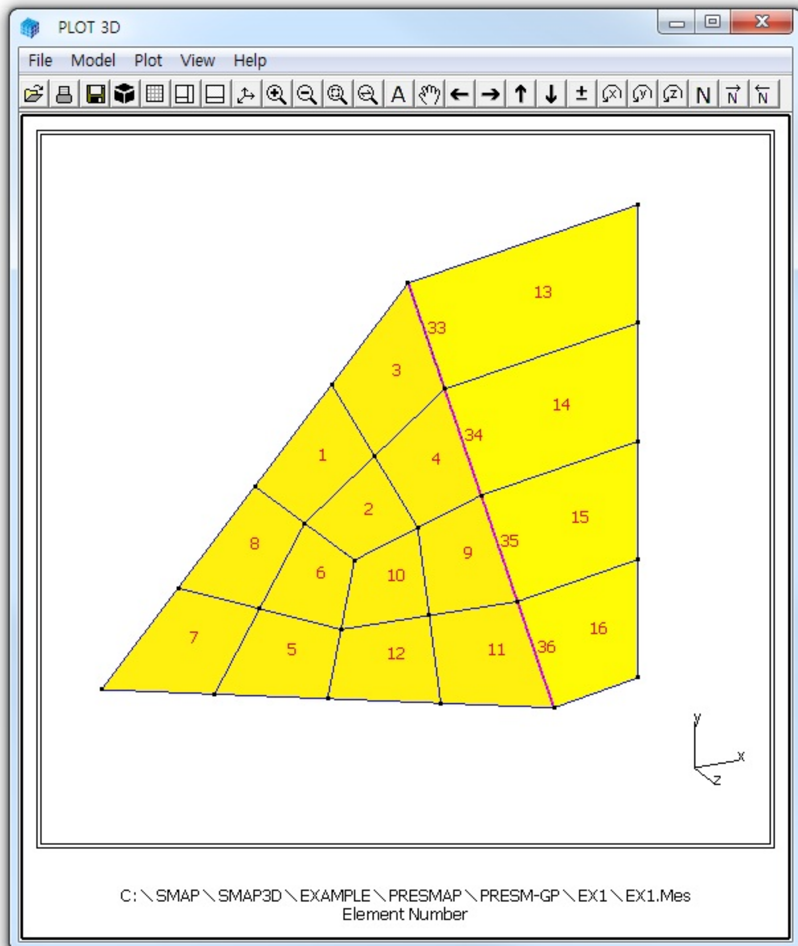


Figure 7.80 Element numbers for beam and shell elements

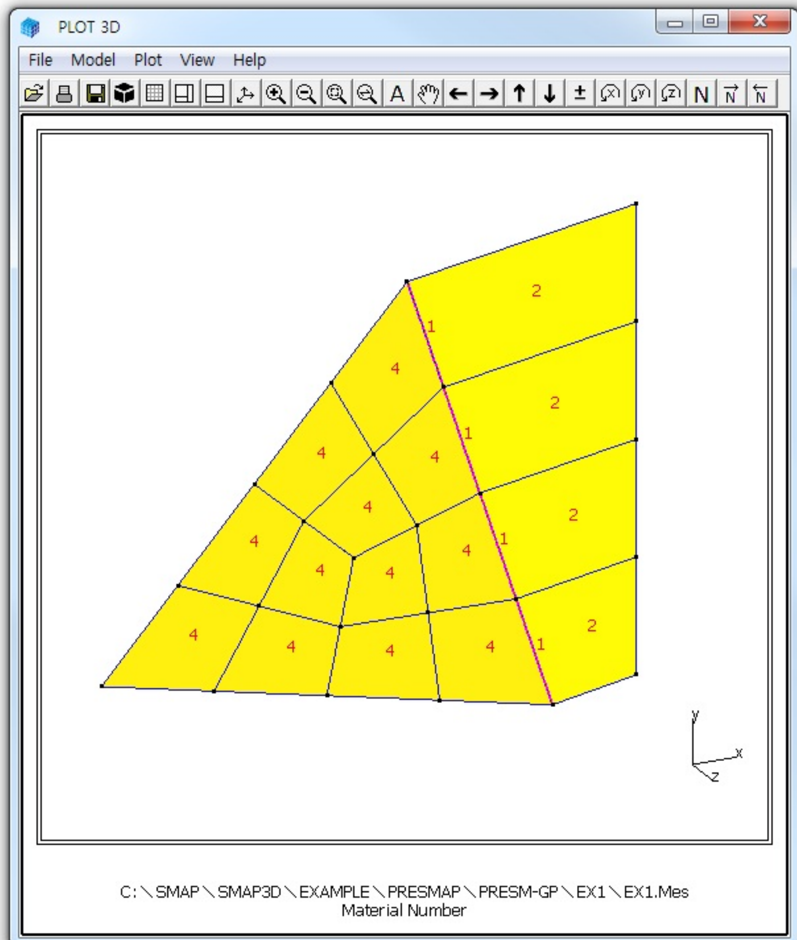


Figure 7.81 Material numbers for beam and shell elements

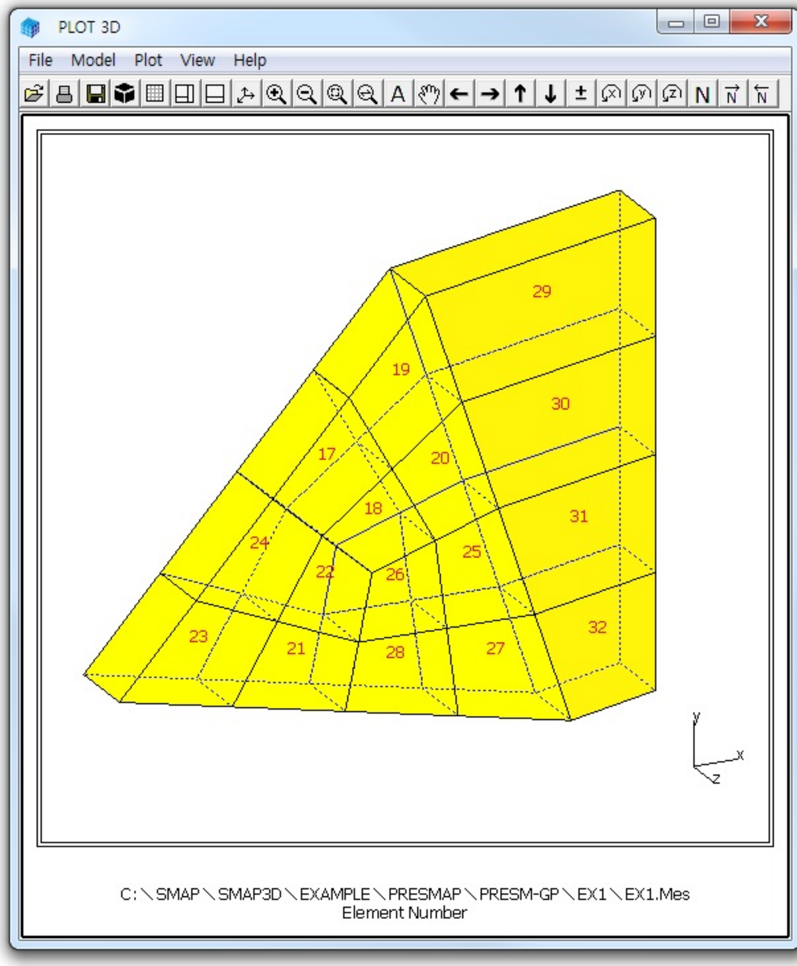


Figure 7.82 Element numbers for continuum elements

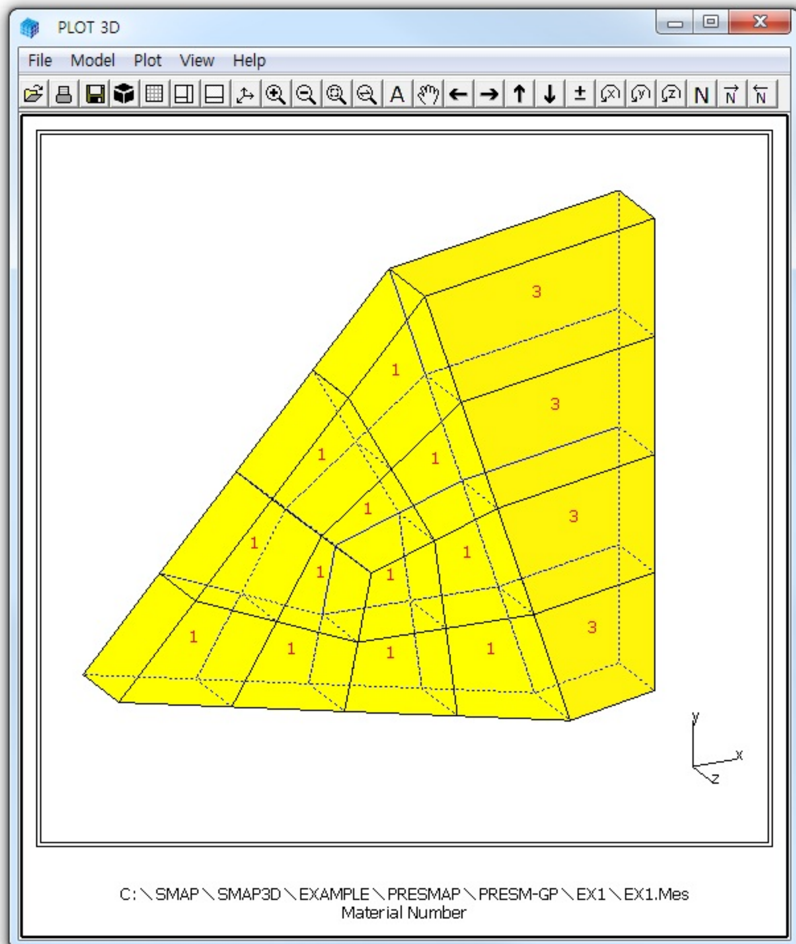


Figure 7.83 Material numbers for continuum elements

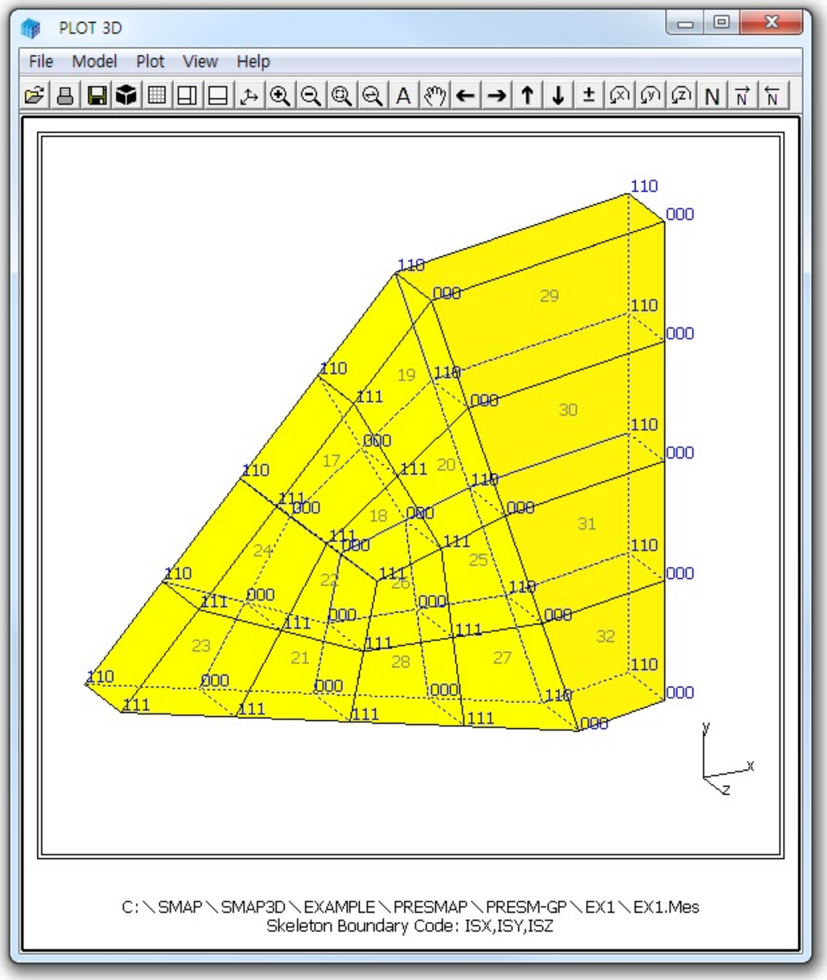


Figure 7.84 Skeleton boundary codes

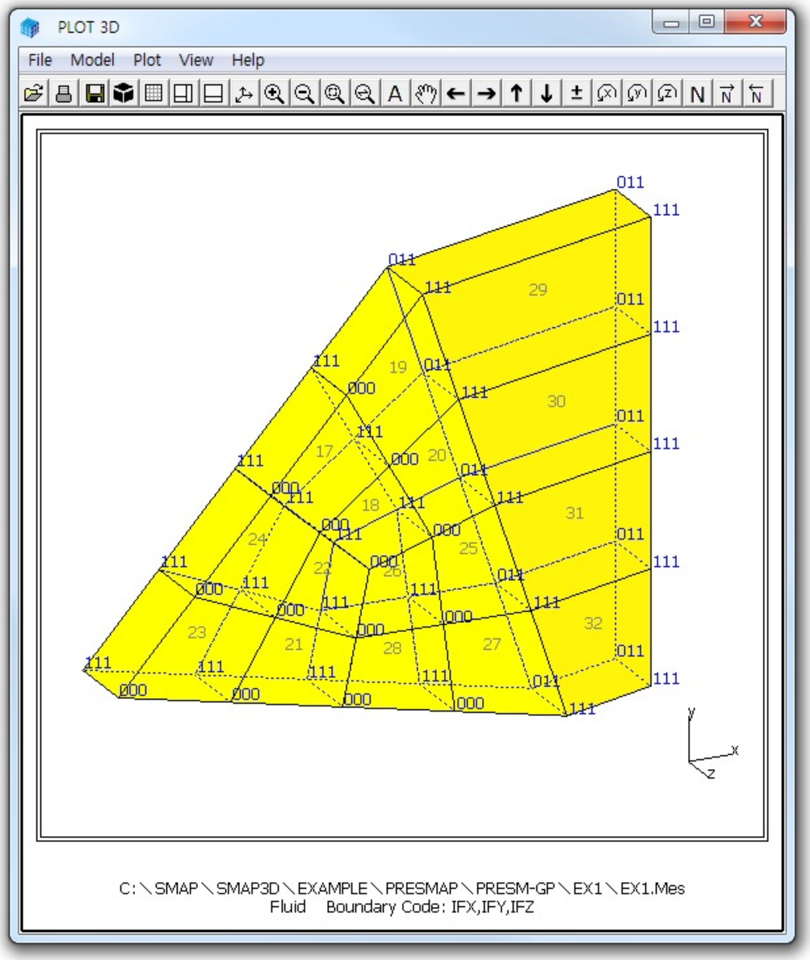


Figure 7.85 Fluid boundary codes

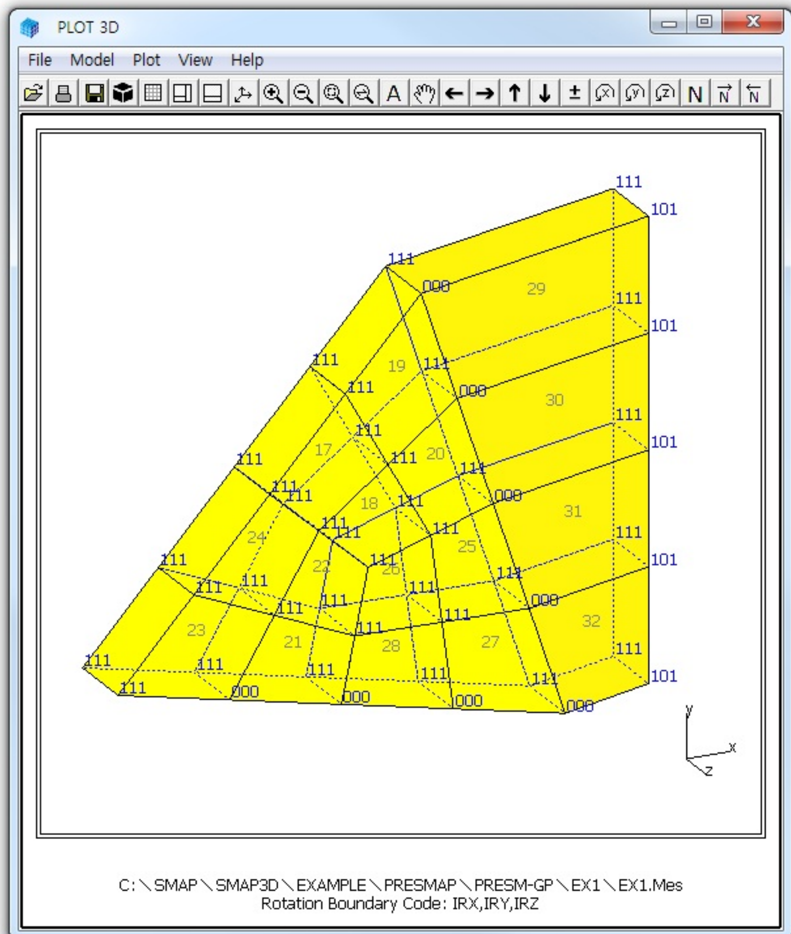


Figure 7.86 Rotation boundary codes

7.8.2 Example 2: Surface with Corner Triangles

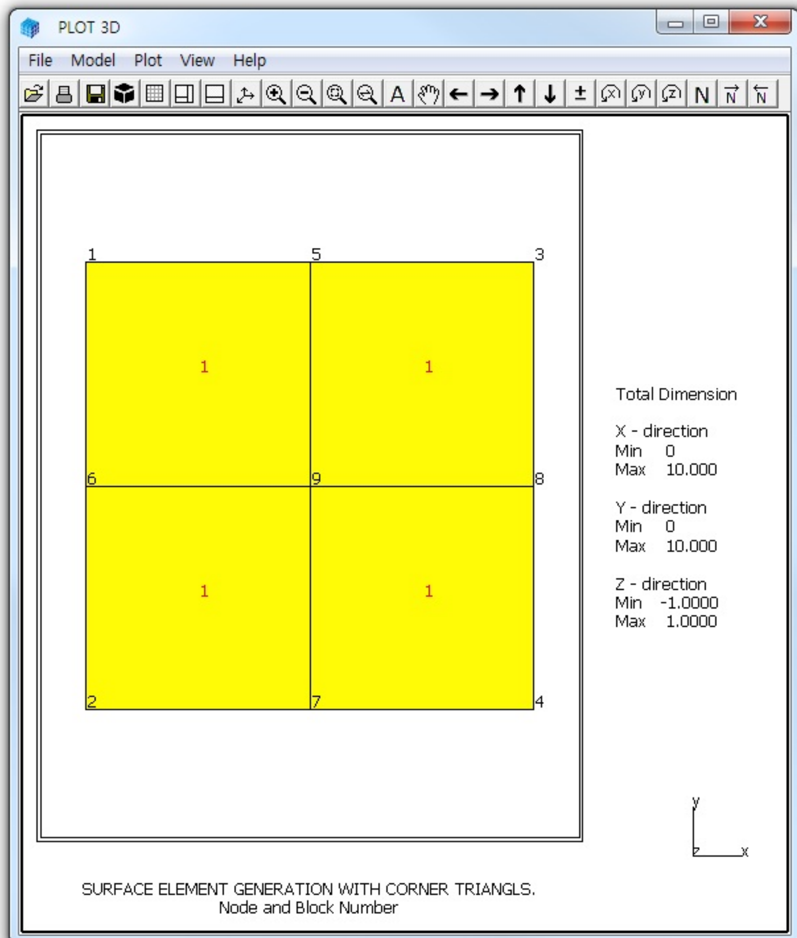


Figure 7.87 Block mesh for Example 2

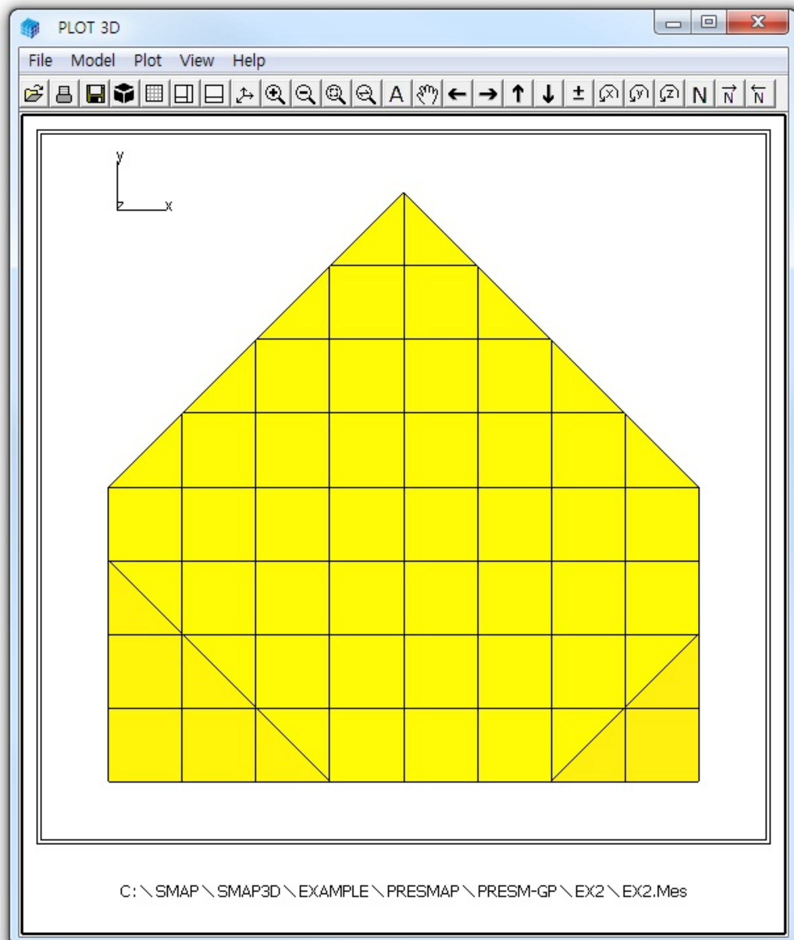


Figure 7.88 Finite element mesh for Example 2

7.8.3 Example 3: Circular Sector

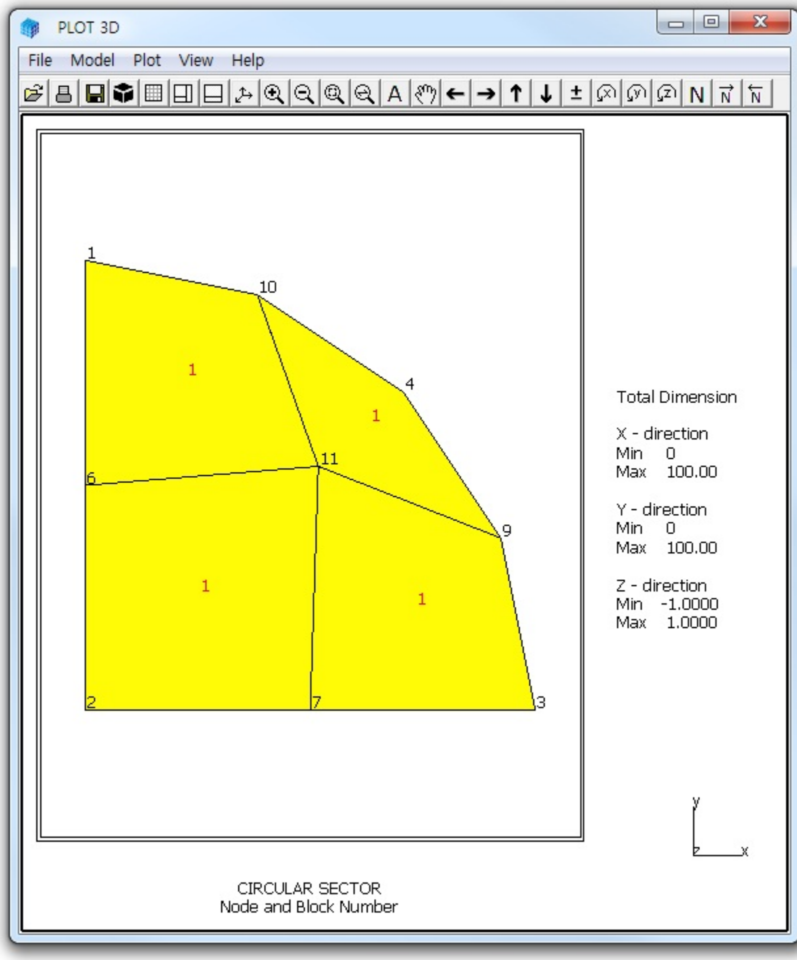


Figure 7.89 Block mesh for Example 3

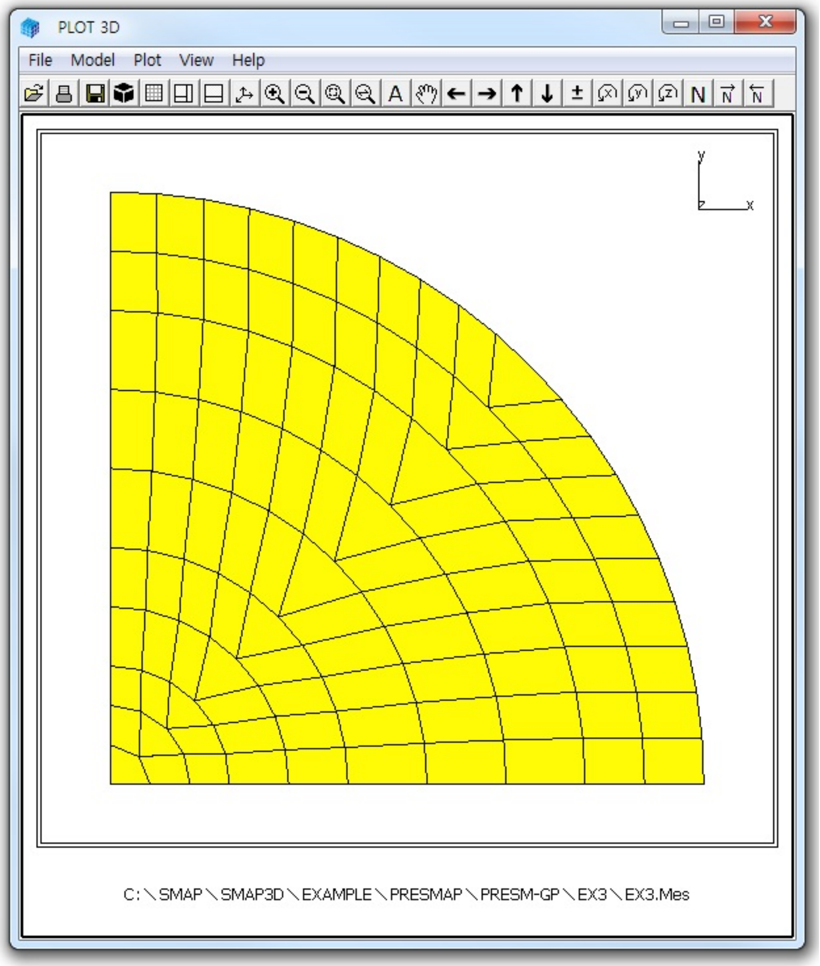


Figure 7.90 Finite element mesh for Example 3

7.8.4 Example 4: Straight Line Sector

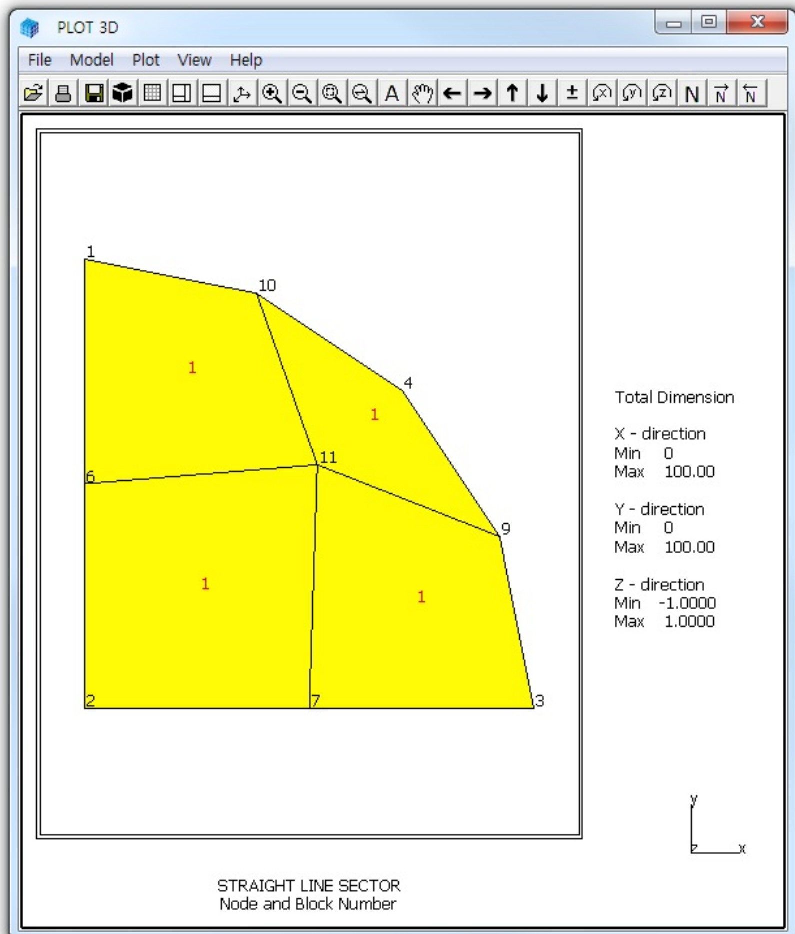


Figure 7.91 Block mesh for Example 4

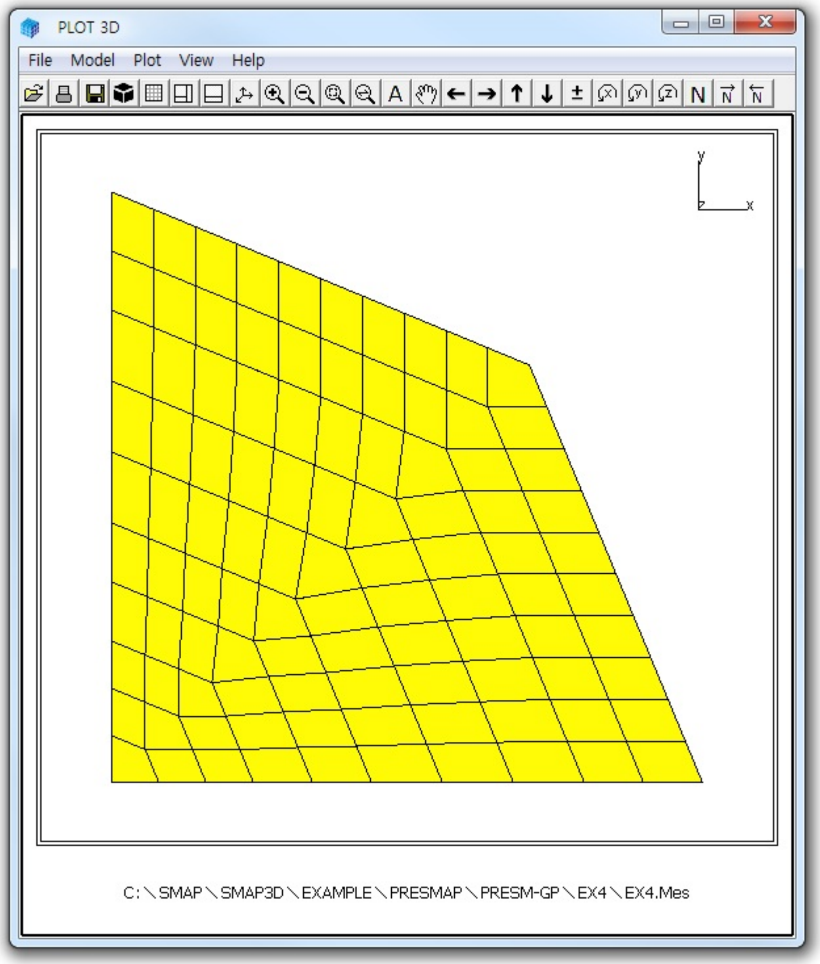


Figure 7.92 Finite element mesh for Example 4

7.8.5 Example 5: Surface and Line Element (1)

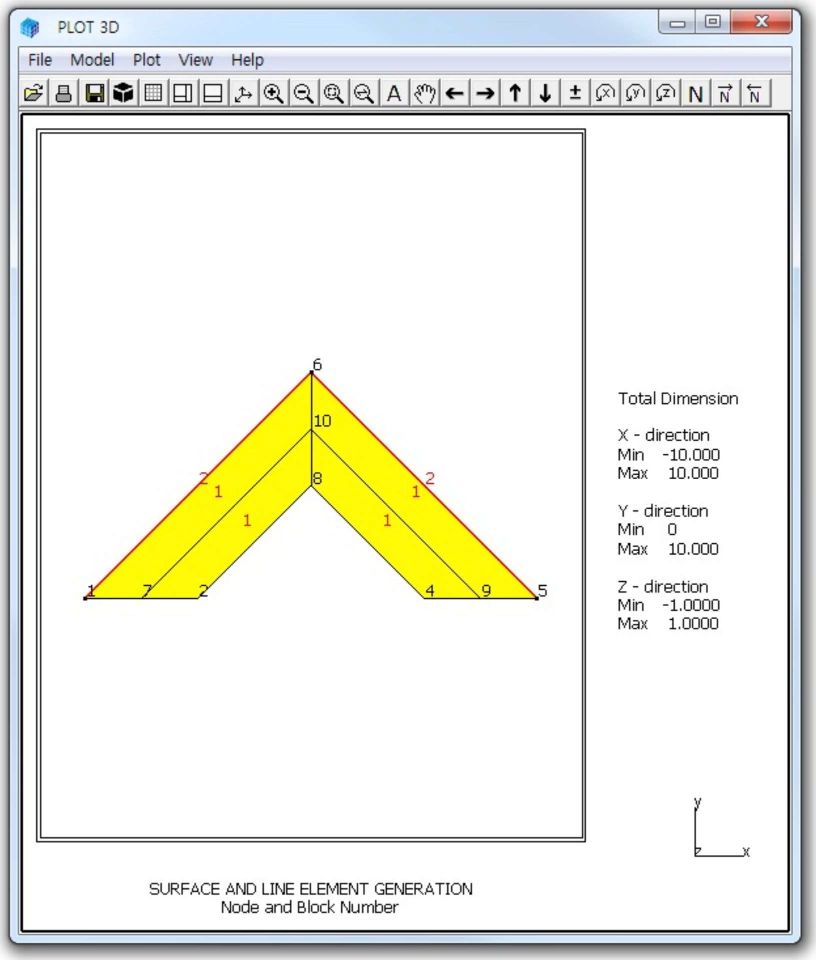


Figure 7.93 Block mesh for Example 5

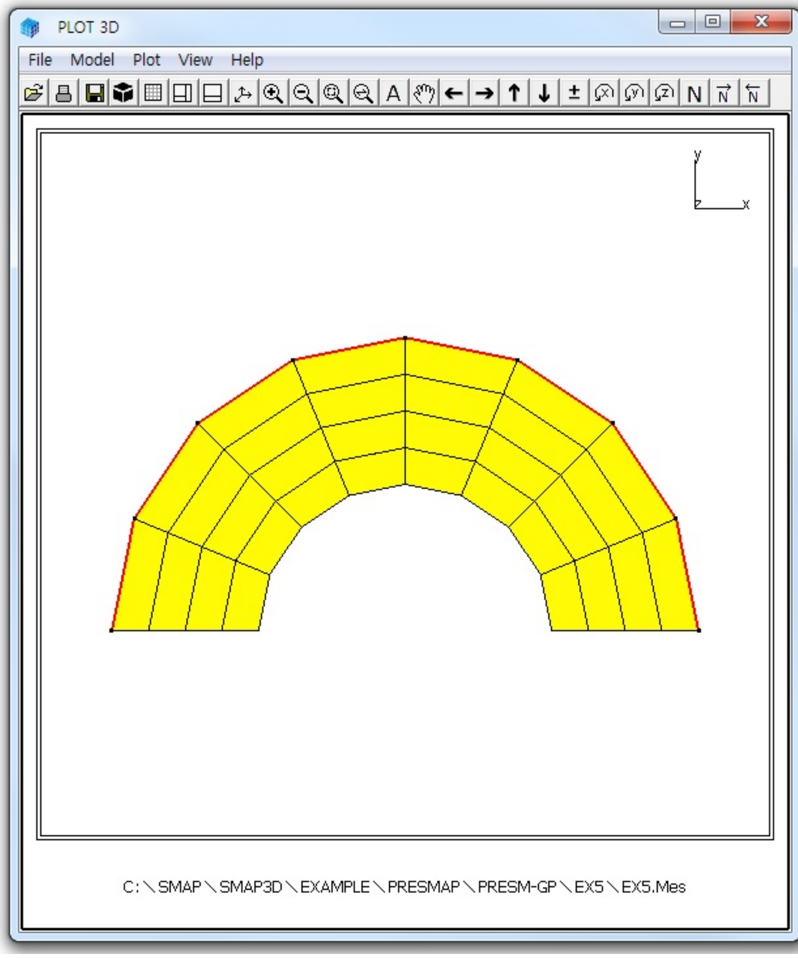


Figure 7.94 Finite element mesh for Example 5

7.8.6 Example 6: Surface and Line Element (2)

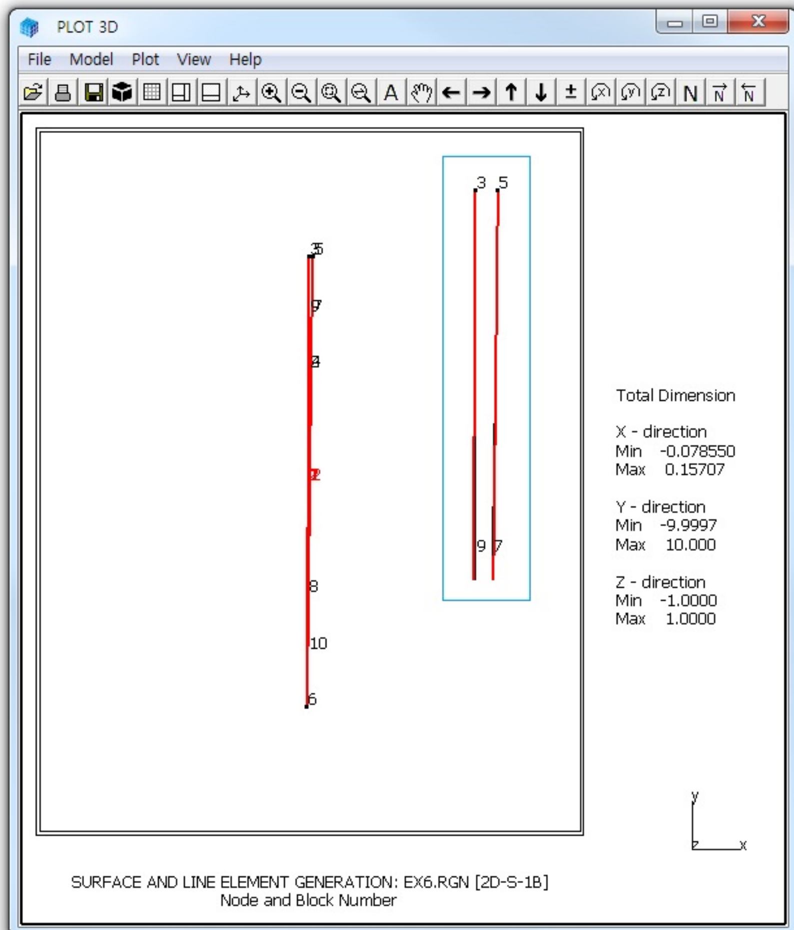


Figure 7.95 Block mesh for Example 6

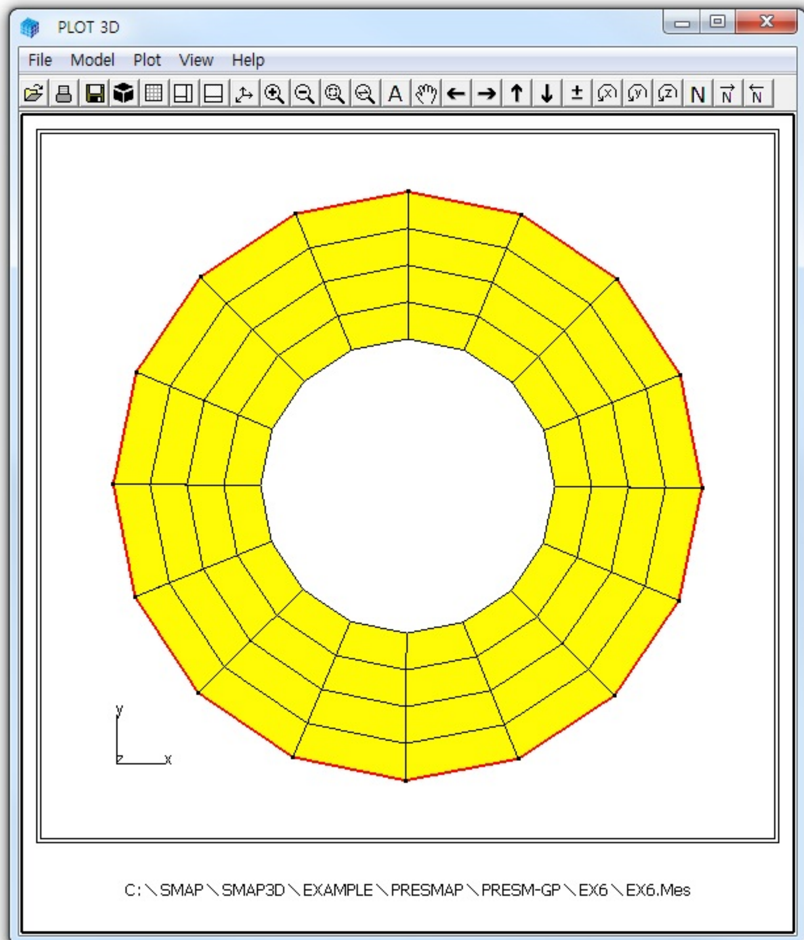


Figure 7.96 Finite element mesh for Example 6

7.8.7 Example 7: Surface and Line Element (3)

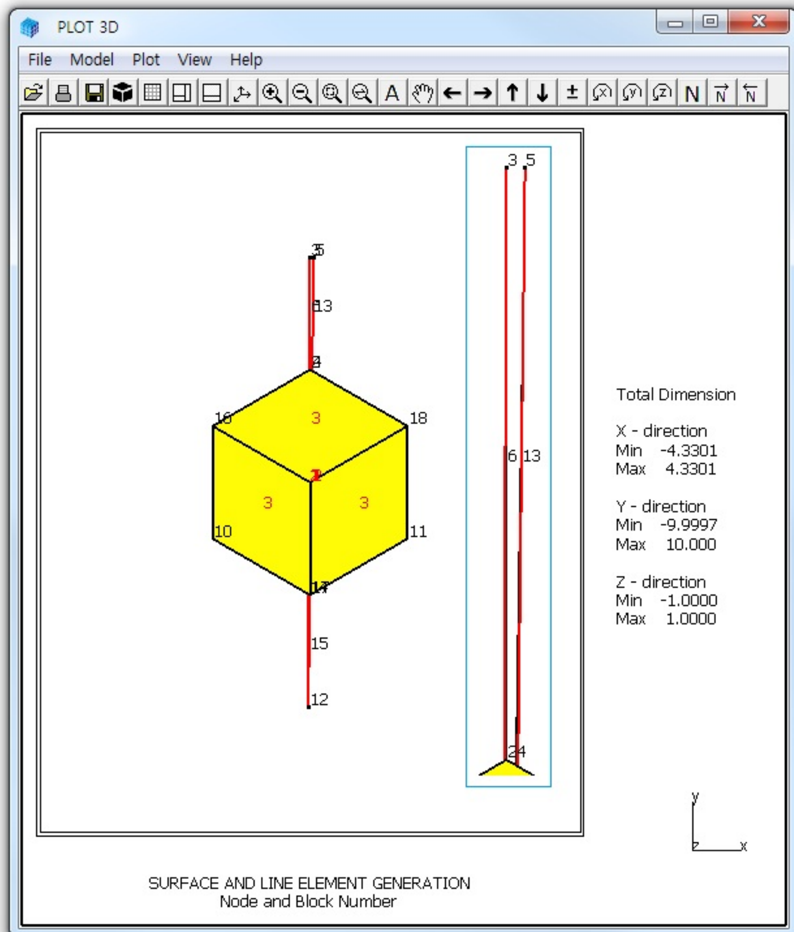


Figure 7.97 Block mesh for Example 7

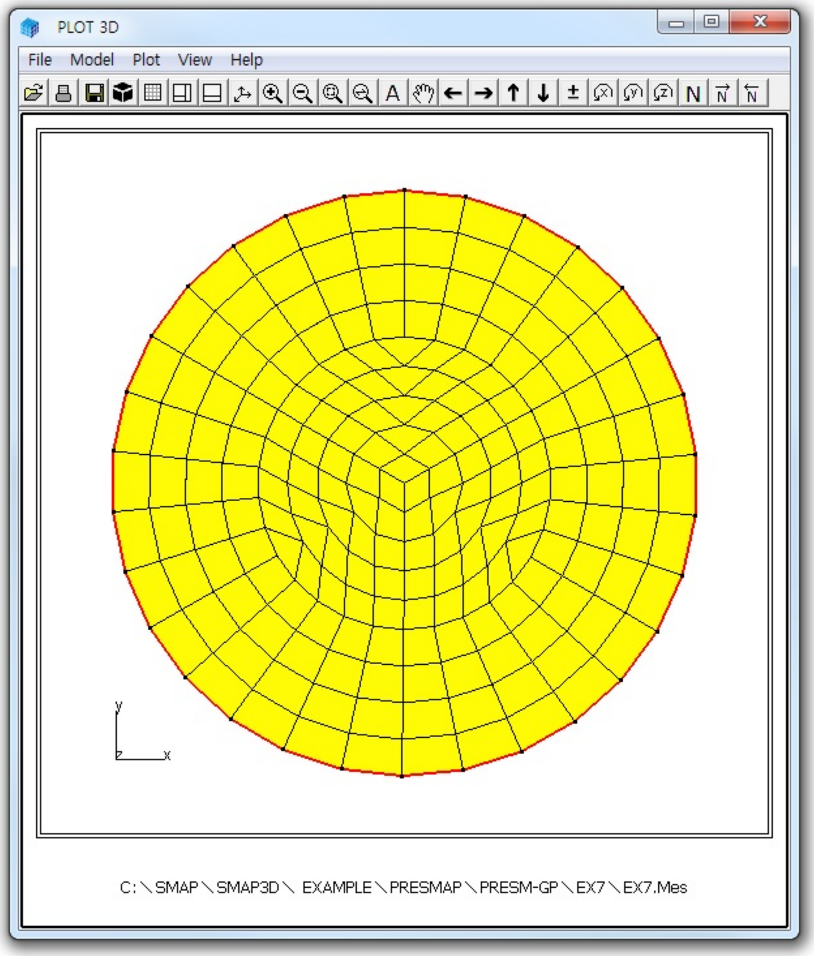


Figure 7.98 Finite element mesh for Example 7

7.8.8 Example 8: Cement-Soil Road

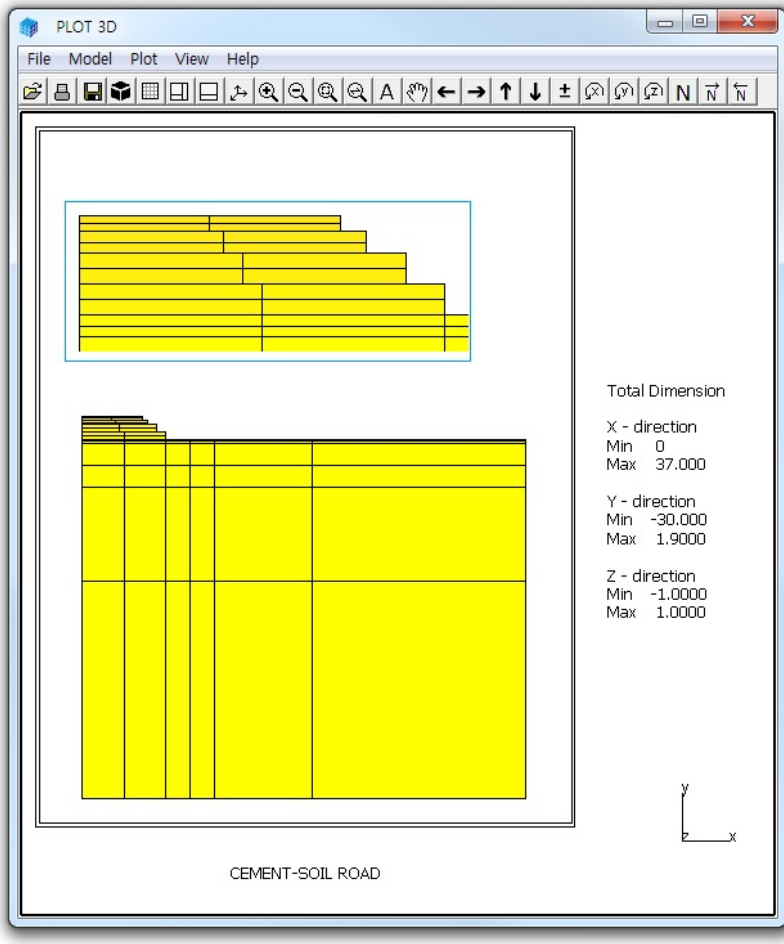


Figure 7.99 Block mesh for Example 8

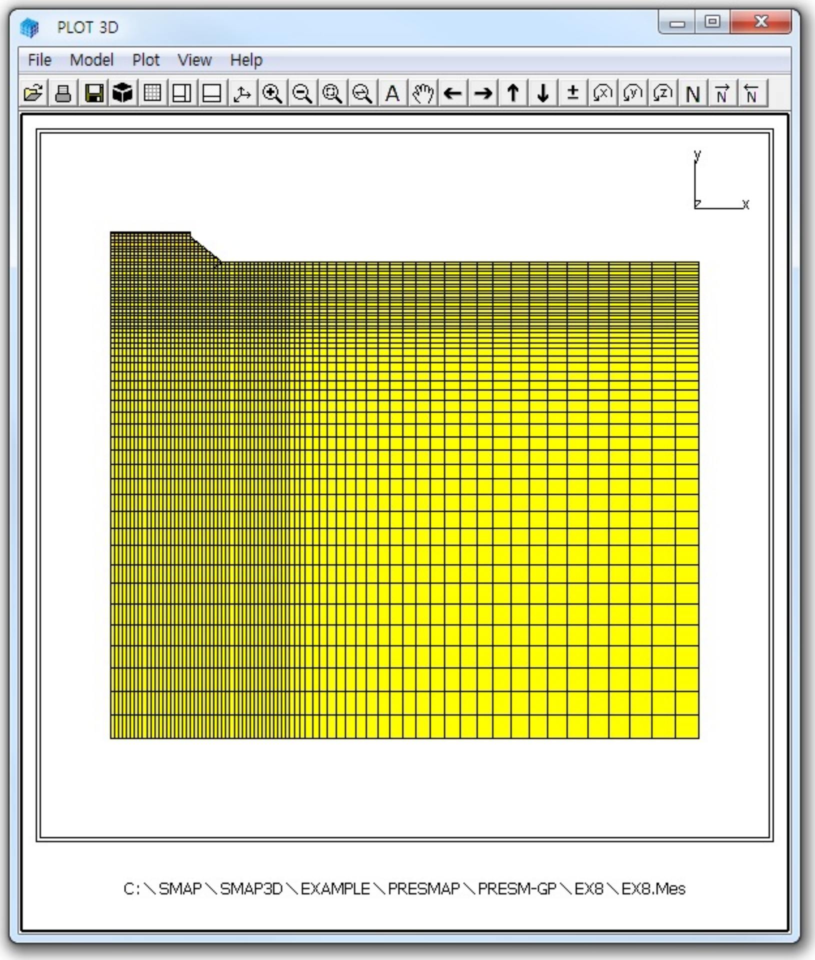


Figure 7.100 Finite element mesh for Example 8

7.8.9 Example 9: Tunnel in Spherical Geometry

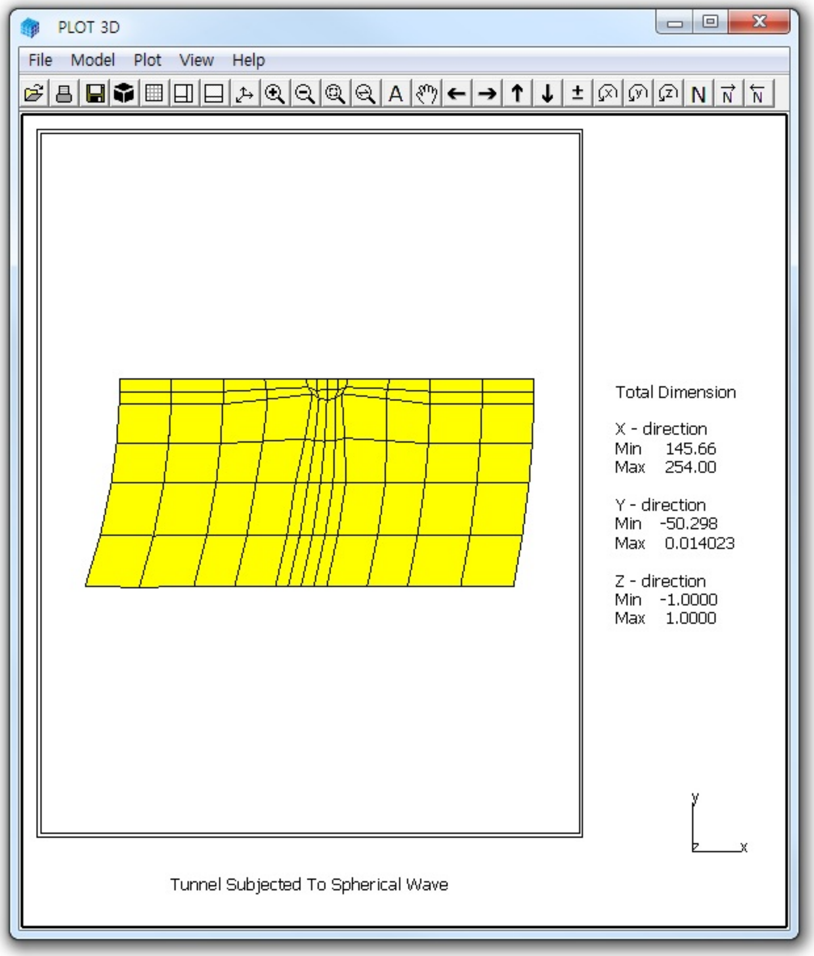


Figure 7.101 Block mesh for Example 9

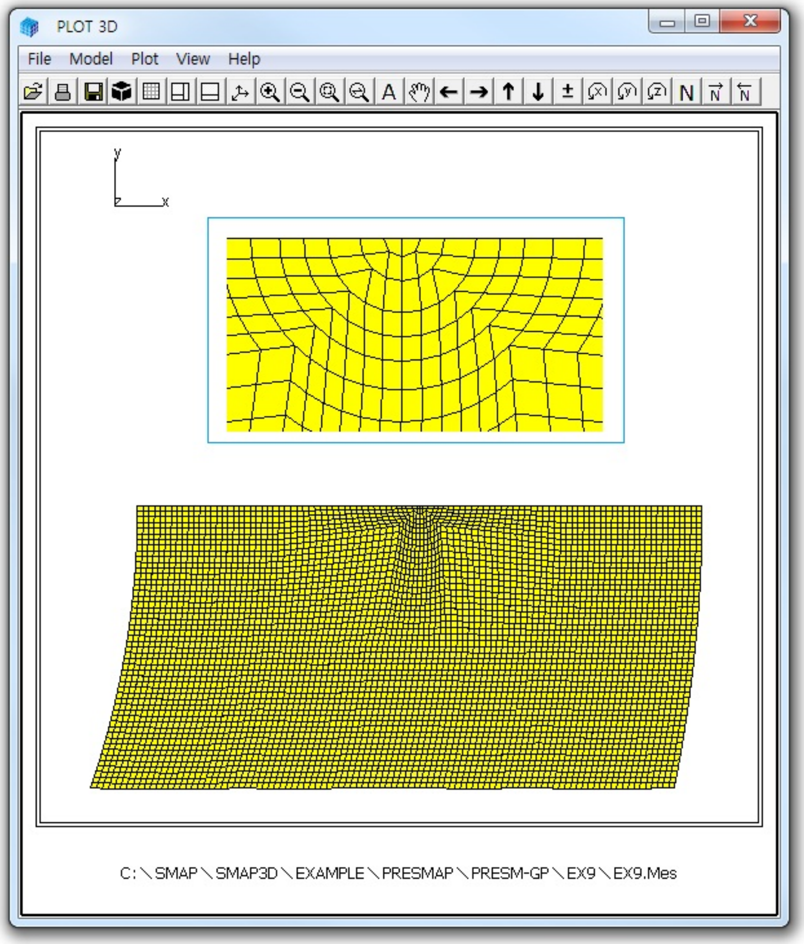


Figure 7.102 Finite element mesh for Example 9

7.8.10 Example 10: Horseshoe Tunnel

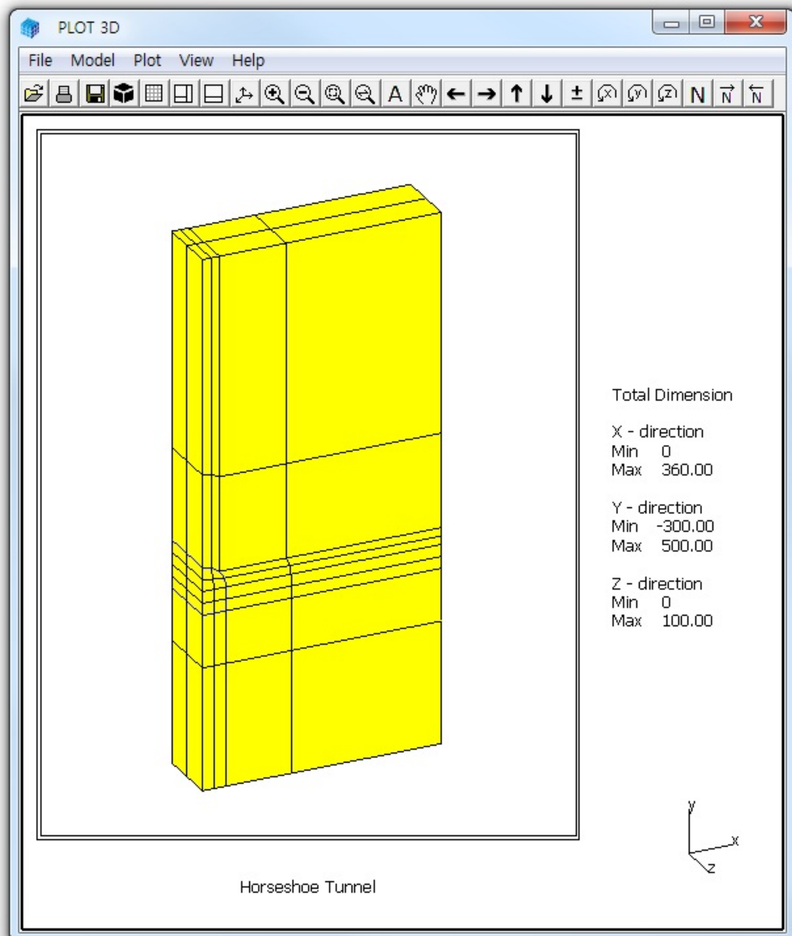


Figure 7.103 Block mesh for Example 10

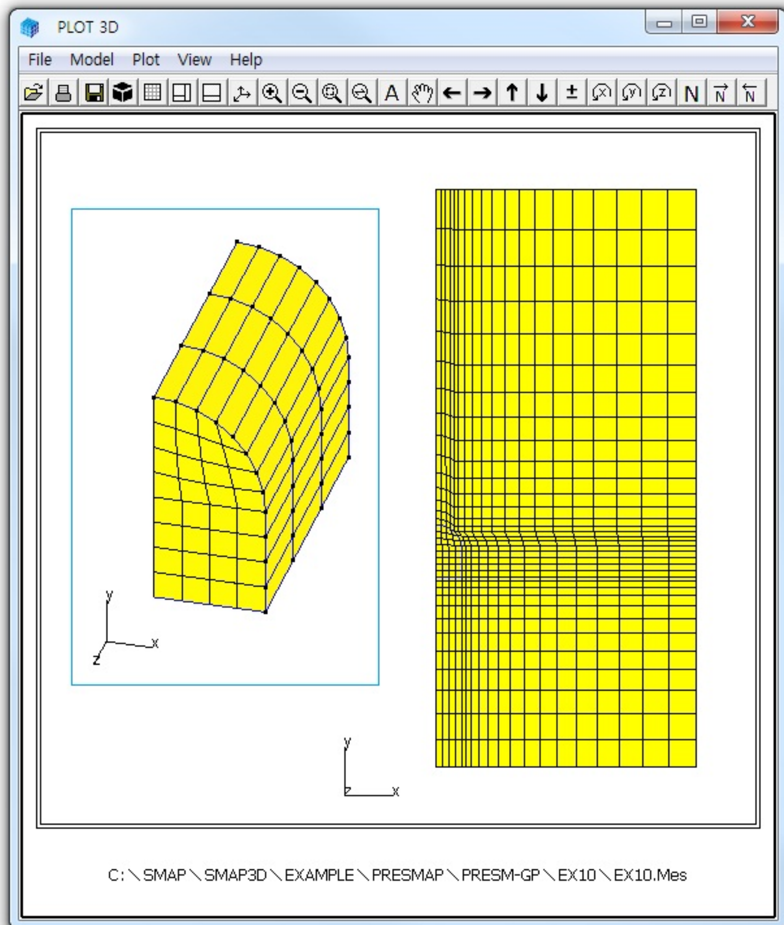


Figure 7.104 Finite element mesh for Example 10

7.8.11 Example 11: Wedge Volume and Surface Block

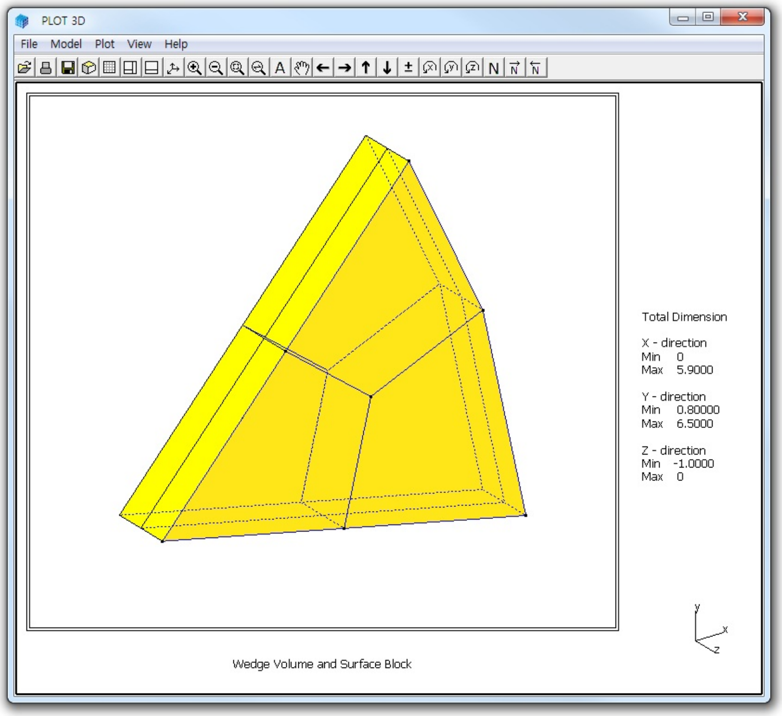


Figure 7.105 Block mesh for Example 11

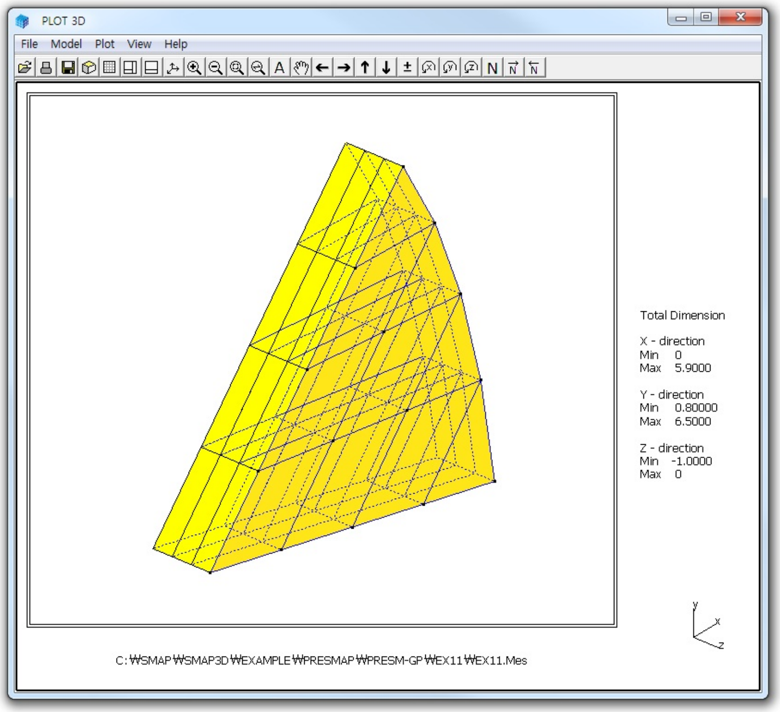


Figure 7.106 Finite element mesh for Example 11

7.9 JOINT-3D

JOINT-3D is the special pre-processor which can be used to generate jointed continuum finite element meshes given the conventional continuum SMAP-3D Mesh File input. For the jointed continuum analysis, each continuum finite element is surrounded by joint elements which allow slippage along the joint when reaching shear strength and debonding normal to the joint face when exceeding tensile strength.

JOINT-3D includes following features:

- Internal Joints within the specified group of materials
- Boundary Joints along the specified group of materials
- Surface Joints along the specified group of element surfaces

First, you need to prepare SMAP-3D Mesh File consisting of continuum elements. Copy C:\Smap\Ct\Ctdata\Joint-3D.dat into Working Directory and then modify input parameters as described in Section 7.10 of User's Manual.

JOINT-3D can be selected in the following order.

Run → Mesh Generator → PreSmap → Joint

Dialog for Input and Output File Names will be displayed as in Fig. 7.107.

7.9.1 Example 1: Horseshoe Tunnel

Example 1 shows you how jointed continuum elements are generated around the horseshoe tunnel. Figure 7.108 shows material numbers of continuum elements: 1 representing for Far Field, 2 for Tunnel Core and 3 for Near Field in the input mesh. Joint data is prepared to generate internal joints within the Near Field as listed in Table 7.32. Note that it also specify Outer Shell between Tunnel Core and and Near Field to generate tunnel liner.

Figure 7.109 shows generated jointed finite element meshes around tunnel core along with shell elements representing for tunnel liner.

Table 7.32 Listing of input file Joint.inp for Example 1

```
******************************************************************
* Jointed Continuum Generation                                   *
* ===============================================================*
*                                                                *
* Card 1.1                                                       *
* Title                                                          *
  Example 1: Horseshoe Tunnel
* Card 1.2                                                       *
* AllJoint                                                       *
*  = 0 Generate Joint Elements along all interfaces              *
*      between continuum elements.                               *
*      Cards 2, 3 and 4 are not used.                            *
*                                                                *
*  = 1 Generate Joint Elements for material numbers of           *
*      continuum elements as specified in Cards 2 and 3.         *
*      Card 4 is not used.                                       *
*                                                                *
*  = 2 Generate Joint Elements for element surface numbers of    *
*      continuum elements as specified in Card 4.                *
*      Cards 2 and 3 are ignored.                                *
*                                                                *
* ThicAJ  Joint Thickness Used For AllJoint = 0                  *
*                                                                *
*----------------------------------------------------------------*
*                                                                *
*  To Run JOINT-3D                                               *
*                                                                *
*  Method 1                                                      *
*  ========                                                      *
*                                                                *
*  SMAP-3D > Run > Mesh Generator > PreSmap > Joint              *
*  Specify input and output file names shown on the screen.     *
*                                                                *
*  Method 2                                                      *
*  ========                                                      *
*                                                                *
*  1. Select SMAP-3D > Setup > PLOT 3D                           *
*  Specify Joint Thickness View Factor greater than 0.0          *
*  Example: Joint Thickness View Factor = 1.0                    *
*                                                                *
*  2. Select SMAP-3D > Mesh > F.E. Mesh > Open
*
*  This wil open Mesh File of Continuum Elements.                *
*                                                                *
*  Input file Joint.inp should exist in Working Directory.      *
*  Output file JointedMesh.Mes is shown in Working Directory.   *
*                                                                *
```

```
* AllJoint  ThicAJ                                                 *
*------------------------------------------------------------------*
  1         0.03
*                                                                  *
********************************************************************
* Card 2                                                           *
* Internal Joint Generation By AllJoint = 1                        *
********************************************************************
*                                                                  *
* Card 2.1                                                         *
* NumIJ  (Number of Continuum Materials for Internal Joints)       *
* ThicIJ (Joint Thickness)                                         *
*                                                                  *
* NumIJ     ThicIJ                                                 *
*------------------------------------------------------------------*
  1         0.02
*                                                                  *
* Card 2.2                                                         *
* MatIJ (Material No of Continuum Element for Internal Joints)     *
*           InnerShell = 0: No  1: Includes Inner Shell            *
*           OuterShell = 0: No  1: Includes Outer Shell            *
* MatIJ     InnerShell  OuterShell                                 *
*------------------------------------------------------------------*
  3         0           1
*                                                                  *
********************************************************************
* Card 3                                                           *
* Boundary Joint Generation By AllJoint = 1                        *
********************************************************************
*                                                                  *
* Card 3.1                                                         *
* NumBJ  (Number of Continuum Materials for Boundary Joints)       *
* ThicBJ (Joint Thickness)                                         *
*         InterfaceJoint = 0: No  1: Includes Joint Element        *
* NumBJ     ThicBJ      InterfaceJoint                             *
*------------------------------------------------------------------*
* 3         0.03        1
  0         0.03        1
*                                                                  *
* Card 3.2                                                         *
* MatBJ (Material No of Continuum Element for Boundary Joints)     *
*        InnerShell = 0: No  1: Includes Inner Shell               *
*        OuterShell = 0: No  1: Includes Outer Shell               *
*                                                                  *
* MatBJ     InnerShell  OuterShell                                 *
*------------------------------------------------------------------*
* 1         1           1
* 2         1           1
* 3         1           1
*                                                                  *
********************************************************************
```

```
* Card 4                                                            *
* Surface Joint Generation By AllJoint = 2                          *
*********************************************************************
*                                                                   *
* Card 4.1                                                          *
* NumSJG (Number of Groups for Surface Joints)                      *
*                                                                   *
* NumSJG                                                            *
*-------------------------------------------------------------------*
  0
* 2
*                                                                   *
* Card 4.2                                                          *
* NumSJG [i]   : Number of Element Surfaces in Group i              *
* ThicSJG[i]   : Thickness of Surface Joint in Group i              *
*                                                                   *
* NumSJG [i]   ThicSJG[i]                                           *
*-------------------------------------------------------------------*
*  2           0.5
*  2           0.4
*********************************************************************
* Group (1)                                                         *
* ElementNo    SurfaceNo                                            *
*-------------------------------------------------------------------*
*  1           1
*  2           2
*********************************************************************
* Group (2)                                                         *
* ElementNo    SurfaceNo                                            *
*-------------------------------------------------------------------*
*  3           4
*  4           4
* End of Data                                                       *
*********************************************************************
```

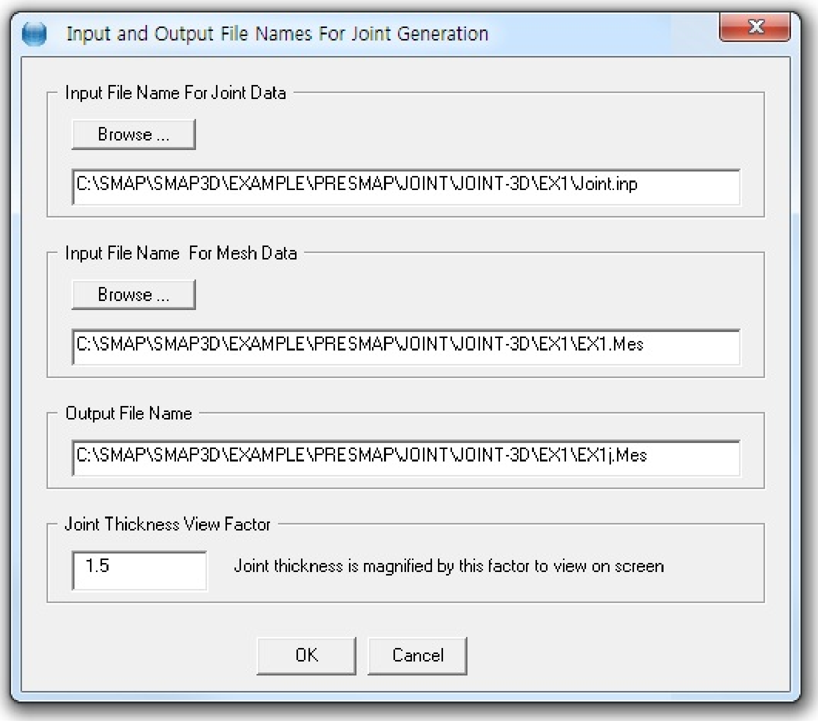


Figure 7.107 Dialog for input and output file names for joint generation

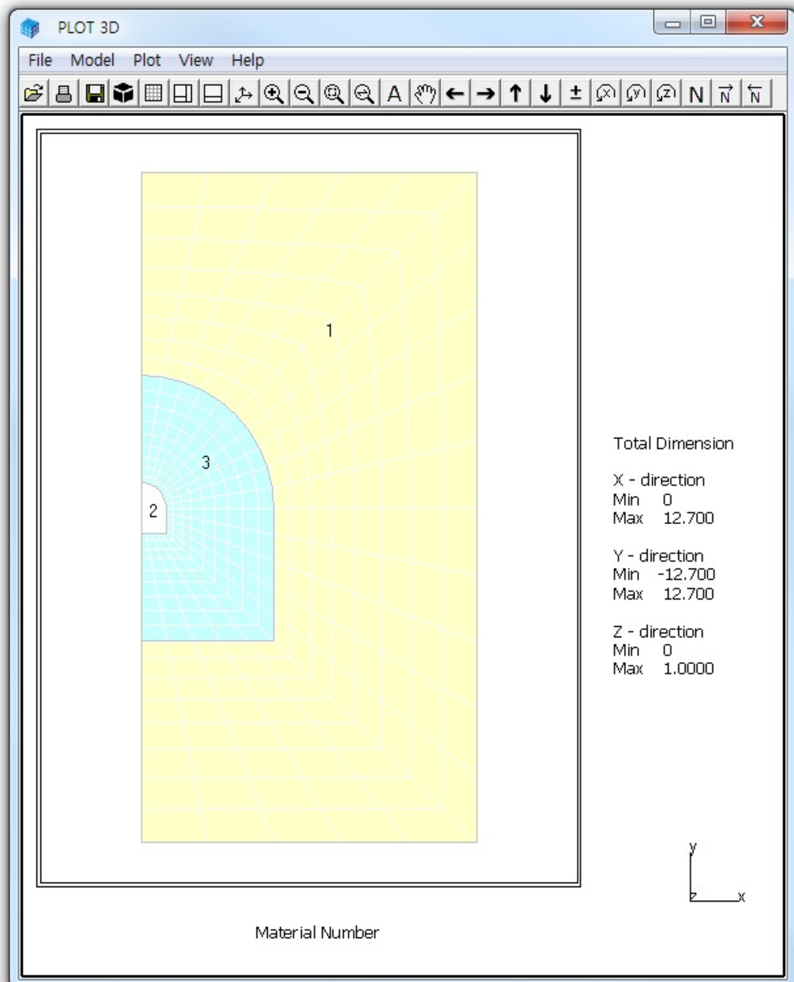


Figure 7.108 Material numbers in input mesh for Example 1

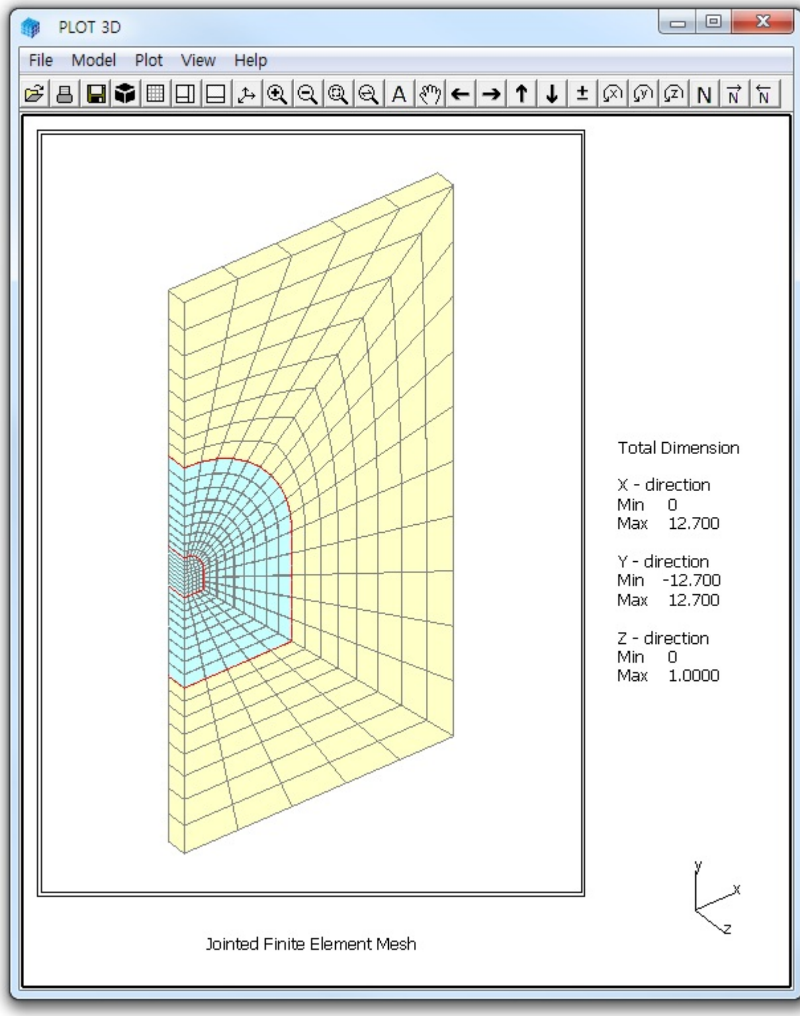


Figure 7.109 Generated jointed finite element mesh for Example 1

7.9.2 Example 2: Vertical Tank with Internal Joints

Example 2 shows jointed continuum elements which are generated within the vertical tank. Figure 7.110 shows material numbers of continuum elements: 1 to 3 representing for vertical tank in the input mesh. Joint data is prepared to generate internal joints within the vertical tank as listed in Table 7.33. Note that it also specify Inner and Outer Shells between vertical tank and surrounding soils.

Figure 7.111 shows generated jointed finite element meshes within vertical tank and shell elements along the boundary.

Table 7.33 Listing of input file Joint.inp for Example 2

```
****************************************************************
* Jointed Continuum Generation                                 *
* =============================================================*
*                                                              *
* Card 1.1                                                     *
* Title                                                        *
  Example 2: Vertical Tank with Internal Joints
* Card 1.2                                                     *
* AllJoint                                                     *
*  = 0 Generate Joint Elements along all interfaces            *
*      between continuum elements.                             *
*      Cards 2, 3 and 4 are not used.                          *
*                                                              *
*  = 1 Generate Joint Elements for material numbers of         *
*      continuum elements as specified in Cards 2 and 3.       *
*      Card 4 is not used.                                     *
*                                                              *
*  = 2 Generate Joint Elements for element surface numbers of  *
*      continuum elements as specified in Card 4.              *
*      Cards 2 and 3 are ignored.                              *
*                                                              *
* ThicAJ  Joint Thickness Used For AllJoint = 0                *
*                                                              *
*--------------------------------------------------------------*
*                                                              *
* AllJoint  ThicAJ                                             *
*--------------------------------------------------------------*
  1         0.03
*                                                              *
****************************************************************
```

```
* Card 2                                                         *
* Internal Joint Generation By AllJoint = 1                      *
******************************************************************
*                                                                *
* Card 2.1                                                       *
* NumIJ  (Number of Continuum Materials for Internal Joints)     *
* ThicIJ (Joint Thickness)                                       *
*                                                                *
* NumIJ     ThicIJ                                               *
*----------------------------------------------------------------*
  3         0.03
* Card 2.2                                                       *
* MatIJ (Material No of Continuum Element for Internal Joints)   *
*           InnerShell = 0: No  1: Includes Inner Shell          *
*           OuterShell = 0: No  1: Includes Outer Shell          *
* MatIJ     InnerShell  OuterShell                               *
*----------------------------------------------------------------*
  1         1           1
  2         1           1
  3         1           1
*                                                                *
******************************************************************
* Card 3                                                         *
* Boundary Joint Generation By AllJoint = 1                      *
******************************************************************
*                                                                *
* Card 3.1                                                       *
* NumBJ  (Number of Continuum Materials for Boundary Joints)     *
* ThicBJ (Joint Thickness)                                       *
*         InterfaceJoint = 0: No  1: Includes Joint Element      *
* NumBJ     ThicBJ      InterfaceJoint                           *
*----------------------------------------------------------------*
  0         0.03        1
*
******************************************************************
* Card 4                                                         *
* Surface Joint Generation By AllJoint = 2                       *
******************************************************************
*                                                                *
* Card 4.1                                                       *
* NumSJG (Number of Groups for Surface Joints)                   *
*                                                                *
* NumSJG                                                         *
*----------------------------------------------------------------*
  0
* End of Data                                                    *
******************************************************************
```

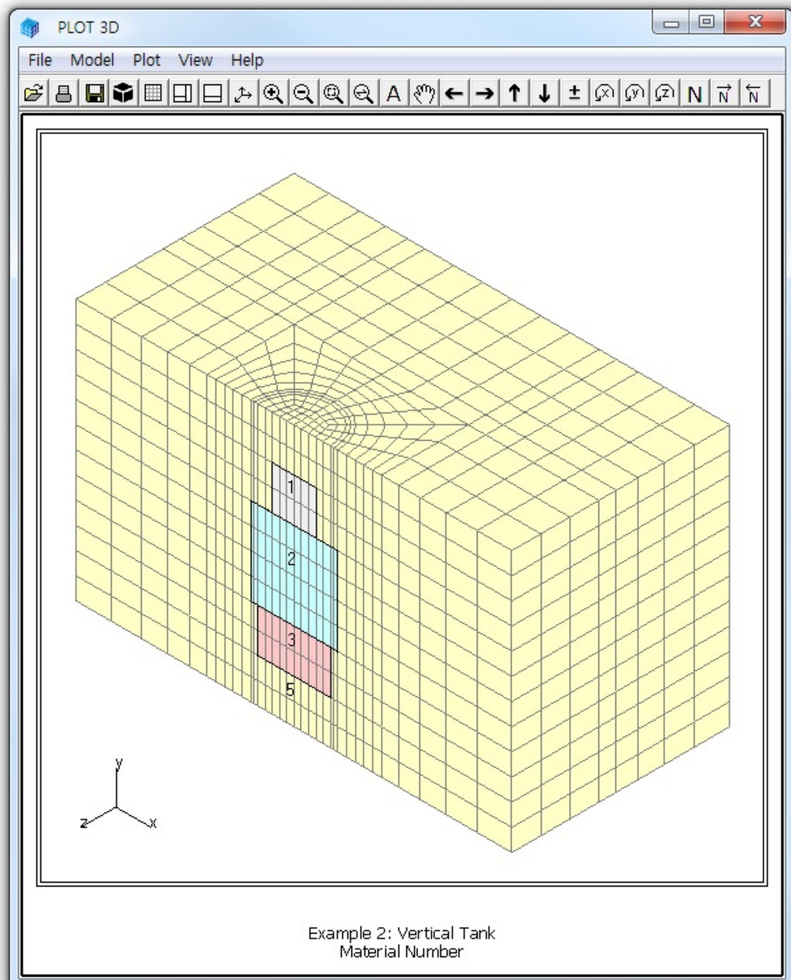


Figure 7.110 Material numbers in input mesh for Example 2

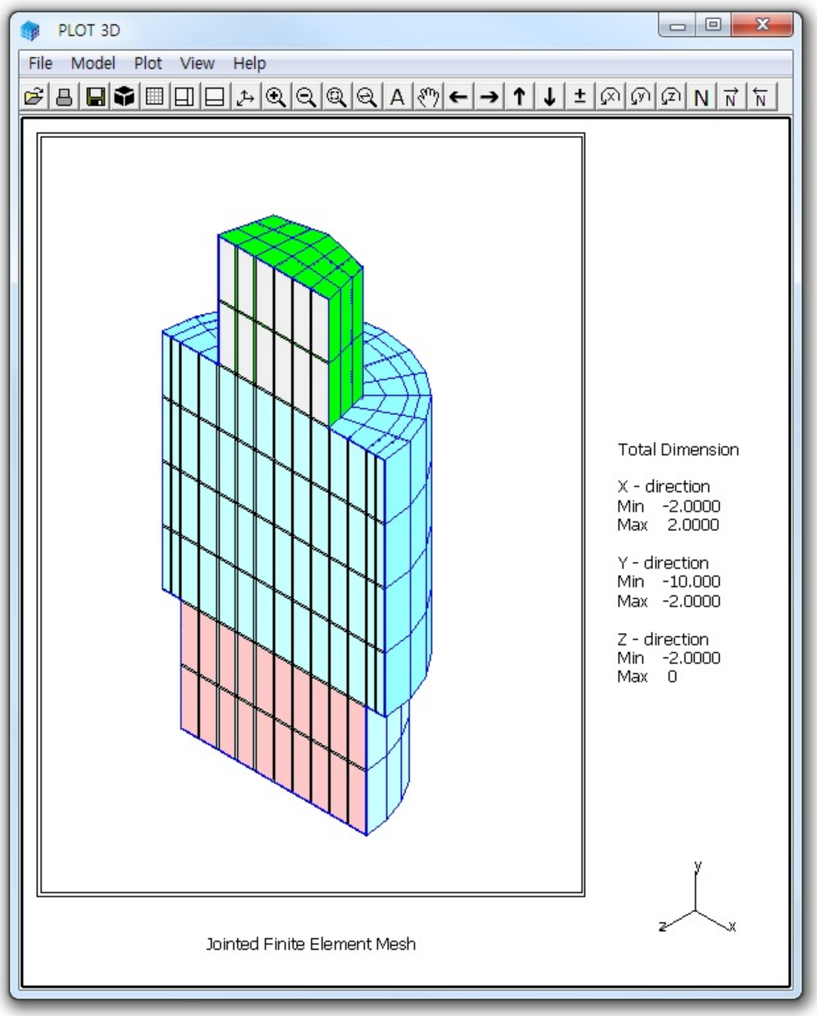


Figure 7.111 Generated jointed finite element mesh for Example 2

7.9.3 Example 3: Vertical Tank with Boundary Joints

Example 3 is the same as Example 2 except that it generates boundary joints along the interface between the vertical tank and surrounding soils. Joint data is prepared to generate boundary joints along the interface as listed in Table 7.34. Note that it also specify Inner and Outer Shells between vertical tank and surrounding soils.

Figure 7.112 shows generated boundary joint elements and shell elements along the interface between vertical tank and surrounding soils.

Table 7.34 Listing of input file Joint.inp for Example 3

```
*****************************************************************
* Jointed Continuum Generation                                  *
* ==============================================================*
*                                                               *
* Card 1.1                                                      *
* Title                                                         *
  Example 3: Vertical Tank with Boundary Joints
* Card 1.2                                                      *
* AllJoint                                                      *
*  = 0 Generate Joint Elements along all interfaces             *
*      between continuum elements.                              *
*      Cards 2, 3 and 4 are not used.                           *
*                                                               *
*  = 1 Generate Joint Elements for material numbers of          *
*      continuum elements as specified in Cards 2 and 3.        *
*      Card 4 is not used.                                      *
*                                                               *
*  = 2 Generate Joint Elements for element surface numbers of   *
*      continuum elements as specified in Card 4.               *
*      Cards 2 and 3 are ignored.                               *
*                                                               *
* ThicAJ  Joint Thickness Used For AllJoint = 0                 *
*                                                               *
*---------------------------------------------------------------*
*                                                               *
* AllJoint  ThicAJ                                              *
*---------------------------------------------------------------*
  1         0.03
*                                                               *
*****************************************************************
```

```
* Card 2                                                         *
* Internal Joint Generation By AllJoint = 1                      *
******************************************************************
*                                                                *
* Card 2.1                                                       *
* NumIJ  (Number of Continuum Materials for Internal Joints)     *
* ThicIJ (Joint Thickness)                                       *
*                                                                *
* NumIJ     ThicIJ                                               *
*----------------------------------------------------------------*
  0         0.03
*                                                                *
******************************************************************
* Card 3                                                         *
* Boundary Joint Generation By AllJoint = 1                      *
******************************************************************
*                                                                *
* Card 3.1                                                       *
* NumBJ  (Number of Continuum Materials for Boundary Joints)     *
* ThicBJ (Joint Thickness)                                       *
*         InterfaceJoint = 0: No  1: Includes Joint Element      *
* NumBJ     ThicBJ      InterfaceJoint                           *
*----------------------------------------------------------------*
  3         0.03        1
*                                                                *
* Card 3.2                                                       *
* MatBJ (Material No of Continuum Element for Boundary Joints)   *
*        InnerShell = 0: No  1: Includes Inner Shell             *
*        OuterShell = 0: No  1: Includes Outer Shell             *
*                                                                *
* MatBJ     InnerShell  OuterShell                               *
*----------------------------------------------------------------*
  1         1           1
  2         1           1
  3         1           1
*                                                                *
******************************************************************
* Card 4                                                         *
* Surface Joint Generation By AllJoint = 2                       *
******************************************************************
*                                                                *
* Card 4.1                                                       *
* NumSJG (Number of Groups for Surface Joints)                   *
*                                                                *
* NumSJG                                                         *
*----------------------------------------------------------------*
  0
* End of Data                                                    *
******************************************************************
```

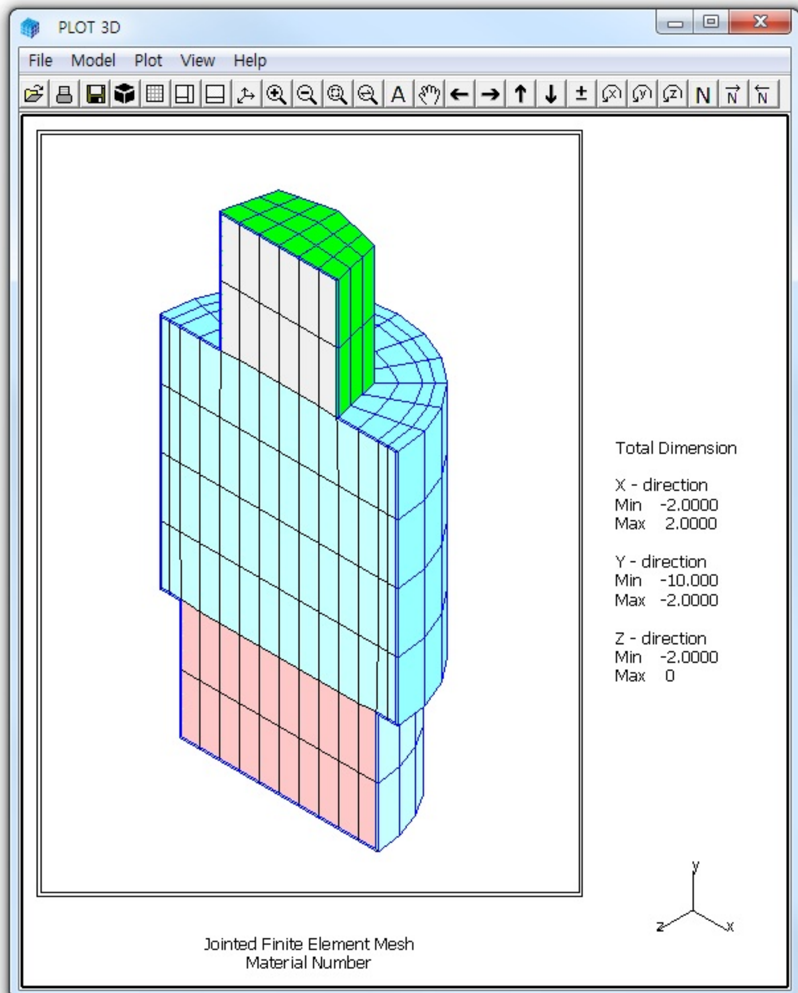


Figure 7.112 Generated jointed finite element mesh for Example 3

7.10 INTERSECTION

INTERSECTION programs are mainly used to compute the locations of the 3D surfaces crossing each other. These surfaces consist of Shell Elements with different materials. The computed coordinates of intersections can be used for the construction of complicated three-dimensional meshes. Refer to detailed descriptions in Section 7.11 of User's Manual.

There are two INTERSECTION programs provided in this manual; SHELL ELEMENT and TWO TUNNELS.

7.10.1 Example 1: Shell Element

SHELL ELEMENT is the basic program which can be applied to find the line of intersection of three-dimensional surfaces. First, you need to prepare a SMAP-3D mesh file composed of Shell Elements with different material numbers.

SHELL ELEMENT can be accessed by selecting the following menu
Run → Mesh Generator → PreSmap → Intersection → Shell Element

or
Setup → PLOT 3D → Compute Intersection → Yes
and then open mesh file Plot → Mesh → F. E. Mesh → Open

Example 1 shows input finite element meshes where a smaller rectangular plate crossing a large square plate at right angles as shown in Fig. 7.113.

Figure 7.114 shows computed intersection between two plates. Note that computed coordinates of intersections are represented by Truss Elements.

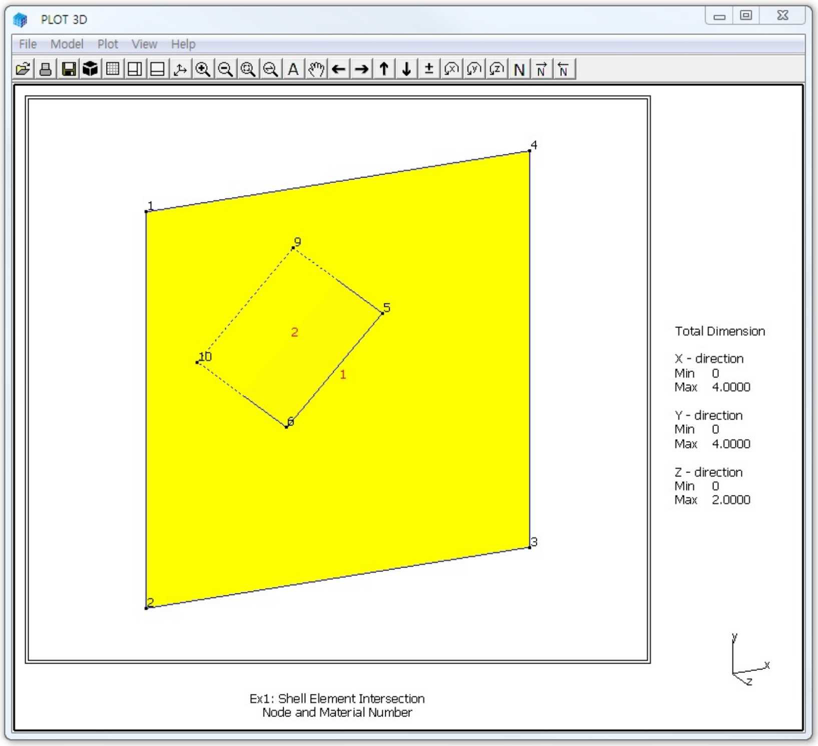


Figure 7.113 Input shell element intersection for Example 1

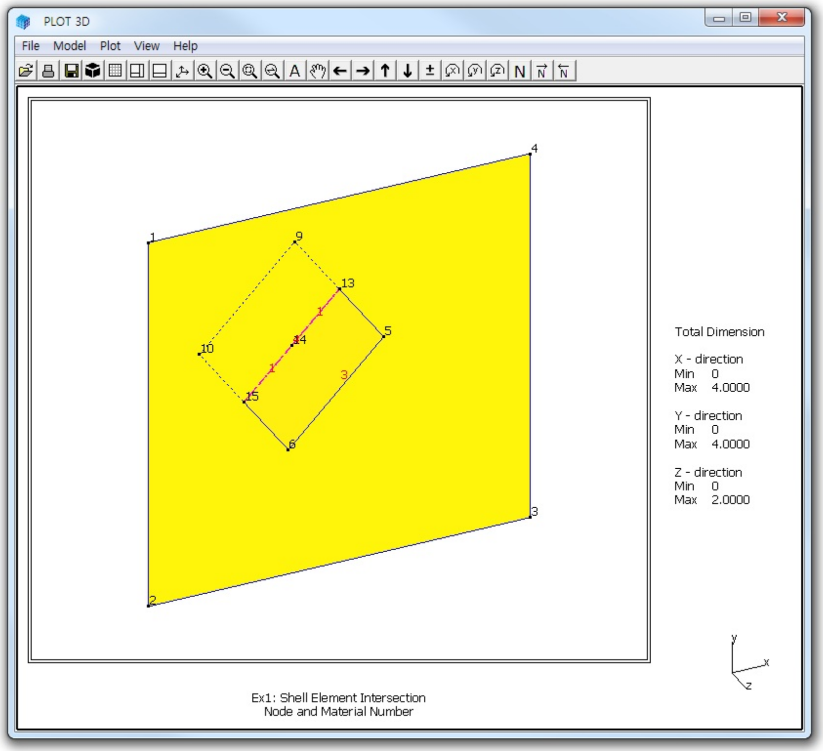


Figure 7.114 Output shell element intersection for Example 1

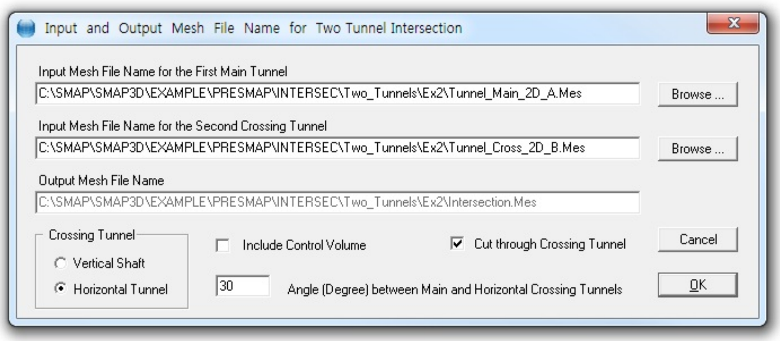
7.10.2 Example 2: Two Tunnels

TWO TUNNELS is the special program where the second crossing tunnel cuts through the first main tunnel at some angles. First, you need to prepare two SMAP-2D mesh files representing for cross sections of the first main and the second crossing tunnels. These cross sections are modeled by two-dimensional Beam Elements with different beam section numbers.

TWO TUNNELS can be accessed by selecting the following menu
Run → Mesh Generator → PreSmap → Intersection → Two Tunnels

Note that output file Intersection.Mes contains Shell Elements representing both main and crossing tunnels.

For best appearance of generated meshes, you need to
copy C:\SMAP\CT\CTDATA\DV-ADRGN.dat into Working Directory
and then modify control parameters in Intersection Calculation for PLOT as shown in Table 7.35.



Example 2 is to show computed intersection between the first main and second crossing tunnels at 30 degree angles. Figures 7.115 & 7.116 show two-dimensional cross sections of first main and second crossing tunnels, respectively, which are used as input meshes consisting of beam elements.

Figure 7.117 shows generated three-dimensional finite element meshes. And Figures 7.118 and 7.119 show the finite element meshes of the first main and second crossing tunnels, respectively, at the intersection points.

Table 7.35 Listing of input file ADDRGN.dat

```
**********************************************************
*                                                        *
*   USERS CAN CHANGE FOLLOWING VALUES TO CONTROL         *
*   ADDRGN-2D MESH GENERATION                            *
*                                                        *
*========================================================*
*                                                        *
*  A. COORDINATE COINCIDENCE:                            *
*     When distance between two adjacent coordinates     *
*     is less than RLMINV/RSMINV, those coordinates      *
*     are assumed to be coincident.                      *
*                                                        *
*   NF = 0: RLMINV is used for all cases.                *
*        1: RSMINV is used for all cases.                *
*        2: RSMINV for segment end points and            *
*           RLMINV for all other cases.                  *
*                                                        *
*   RSMINV = RSFAC * LMIN                                *
*     RSFAC : User input.                                *
*     LMIN  : Min. element length calculated by program  *
*                                                        *
*   NF  RLMINV   RSFAC   RDMINV                          *
*   --  ------   -----   ------                          *
    2   0.0001   0.05    0.005
*                                                        *
*========================================================*
*                                                        *
*  B. REMOVE SHORTER LINE ELEMENTS:                      *
*     When the length of line element (Na-Nb) is shorter *
*     than the specified value Smin = Amin x Save,       *
*     assign Nb as Na, remove such line elements,        *
*     adjust all associated element indexes, and         *
*     reorder element & node numbers in sqquence.        *
*                                                        *
*   NS = 0: Do not apply                                 *
*        1: Apply all line  elements                     *
*        2: Apply all beam  elements                     *
*        3: Apply all truss elements                     *
*        4: Apply specified beam  materials              *
*        5: Apply specified truss materials              *
*        6: Apply specified beam & truss materials       *
*                                                        *
*   bMat1, bMat2, bMat3: Specified beam  materials       *
*   tMat1, tMat2, tMat3: Specified truss materials       *
*                                                        *
```

```
*  C. MOVE NODAL COORDINATES:                                *
*     Node Nc moves along the line (Nc-Nr)                   *
*     NM : Number of nodes to be moved                       *
*     Nc : Current Node to be moved                          *
*     Nr : Reference Node                                    *
*     Ac : Percent movement from Nc to Nr                    *
*            = 0.5 moves half way to Node Nr                 *
*            = 1.0 merges to Node Nr                         *
*   Note : To use this method C                              *
*          1. Run the program with NM = 0 at first           *
*          2. Get the node numbers (Nc, Nr) from plot        *
*          3. Edit this file for NM, (Nc, Nr, Ac) set        *
*          4. Run the program again with data at step 3      *
*---------------------------------------------------------*
*   Standalone ADDRGN-2D                                     *
*   NS    Amin  bMat1  bmat2  bMat3  tMat1  tMat2  tMat3 *
*   ---   ----  -----  -----  -----  -----  -----  ----- *
    0     0.2   1      2      3      1      2      3
*   NM                                                       *
    0
*   Nc    Nr    Ac                                           *
*---------------------------------------------------------*
*                                                            *
*   Intersection Calculation for PLOT-3D                     *
*                                                            *
*   NS    Amin  bMat1  bmat2  bMat3  tMat1  tMat2  tMat3 *
*   ---   ----  -----  -----  -----  -----  -----  ----- *
    3     0.2   1      2      3      1      2      3
*                                                            *
*   NM                                                       *
    11
*   Nc    Nr    Ac                                           *
*   ----  ----  ----                                         *
    1236  1175  1.0
    1174  1175  1.0
    1113  1175  0.4
    1175  1237  0.4
    747   746   1.0
    626   686   1.0
    566   627   1.0
    567   566   0.5
    507   567   0.5
    456   455   1.0
    455   393   0.3
```

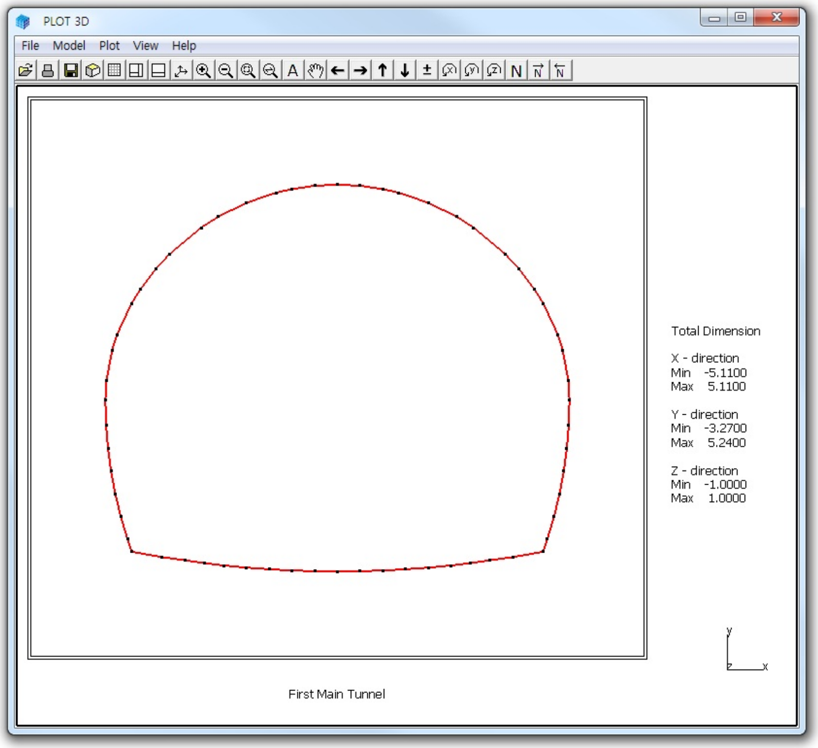


Figure 7.115 Input main tunnel cross section for Example 2

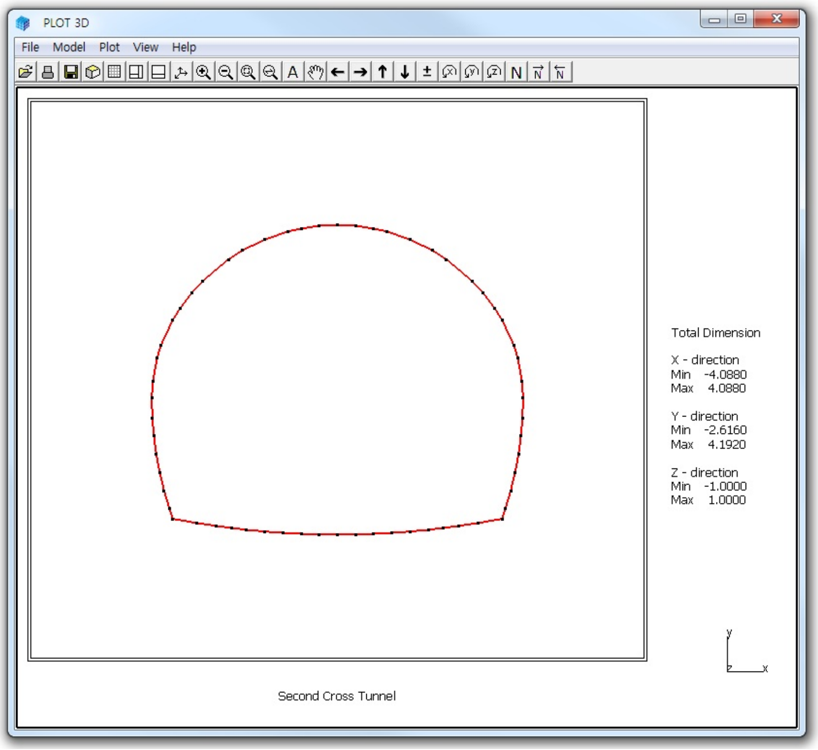


Figure 7.116 Input crossing tunnel cross section for Example 2

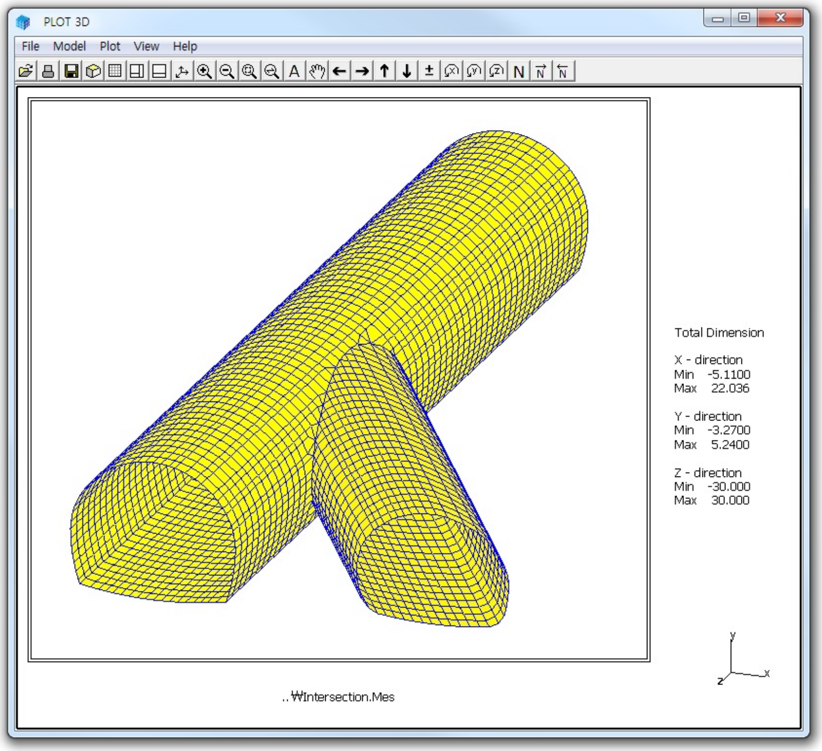


Figure 7.117 Generated finite element meshes for Example 2

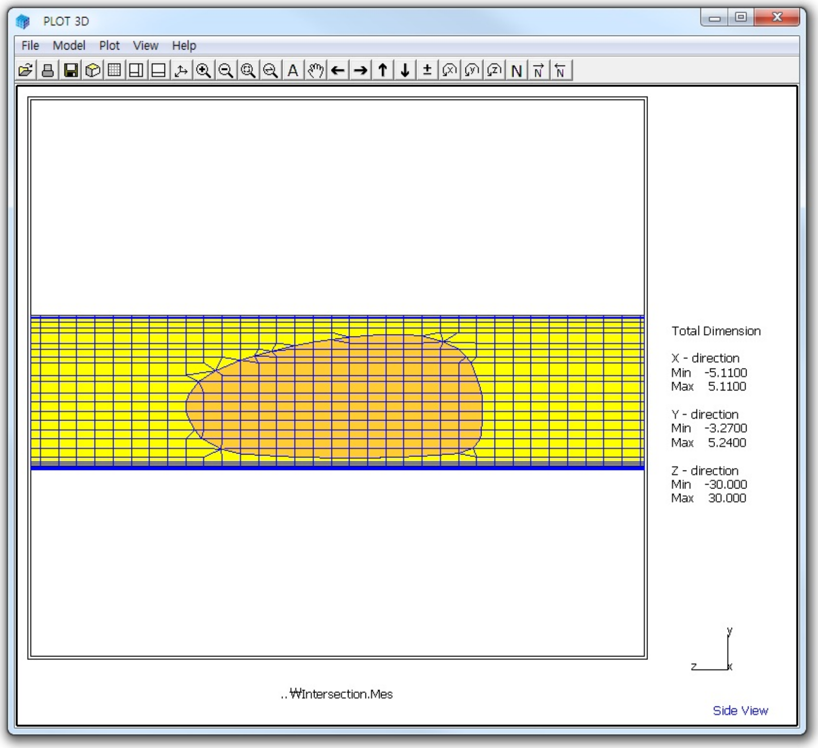


Figure 7.118 Generated main tunnel at intersection for Example 2

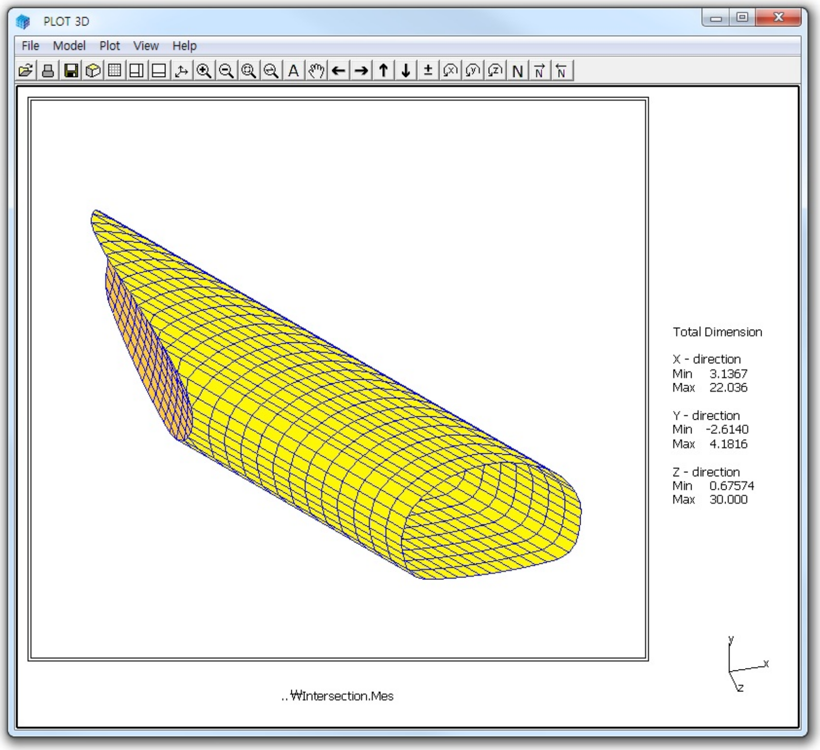


Figure 7.119 Generated crossing tunnel at intersection for Example 2

ADDRGN
Example Problem

ADDRGN is the pre-processing program which has the following functions:

- Combine two different meshes
- Modify existing meshes
  - Change coordinates
  - Change boundary codes
  - Cut elements
  - Change material numbers
- Generate finite element meshes (ADDRGN-2D)

Refer to SMAP-3D User's Manual:

- Section 5 for group mesh generation (ADDRGN-2D)
- Section 8 for input parameters

8.1 ADDRGN-2D

ADDRGN-2D is the two dimensional pre-processor which is used to combine, modify, or generate finite element meshes.

ADDRGN-2D can be selected in the following order:

Run → Mesh Generator → AddRgn → Addrgn 2D

When you finish the execution of ADDRGN-2D, select PLOT-3D to plot modified or generated mesh.

8.1.1 Combining Meshes

In the PRESMAP-2D Example Problem in Sections 7.1.1 and 7.1.2, three different regions (Core, Near-field, and Far-field) are generated using Models 1 and 2. Now, we want to combine all these different regions into one using ADDRGN-2D. Note that CORE.Mes, NEAR.Mes and FAR.Mes are the output files corresponding to the input file CORE.Rgn, NEAR.Rgn and FAR.Rgn respectively.

Element numbers 1 to 72 are assigned for CORE.Mes, 73 to 336 for NEAR.Mes and 337 to 464 for FAR.Mes. When we combine two regions, element numbers should be continuous through the regions. So, let's first add NEAR.Mes (called REGION B) to CORE.Mes (called REGION A) to make CONE.Mes (called COMBINED REGION). Next, let's add FAR.Mes (called REGION B) to CONE.Mes (called REGION A) to make the final mesh CNF.Mes (called COMBINED REGION). ADDRGN input files are listed in Tables 8.1 and 8.2.

Figure 8.1 shows the element meshes of combined region representing all three regions.

Table 8.1 Listing of input file ADD2D-1.Dat

```
* ADD2D-1.Dat
* CARD 1.1
* IMOD = 0 : ADD REGION B TO REGION A
  0
* CARD 2.1
* FILEA : Input file name containing REGION A
  CORE.Mes
* FILEB : Input file name containing REGION B
  NEAR.Mes
* FILEC : Output file name to store COMBINED REGION
  CONE.Mes
* CARD 2.2
* INTERFACE
  0
* END OF DATA
```

Table 8.2 Listing of input file ADD2D-2.Dat

```
* ADD2D-2.Dat
* CARD 1.1
* IMOD = 0 : ADD REGION B TO REGION A
  0
* CARD 2.1
* FILEA : Input file name containing REGION A
  CONE.Mes
* FILEB : Input file name containing REGION B
  FAR.Mes
* FILEC : Output file name to store COMBINED REGION
  CNF.Mes
* CARD 2.2
* INTERFACE
  0
* END OF DATA
```

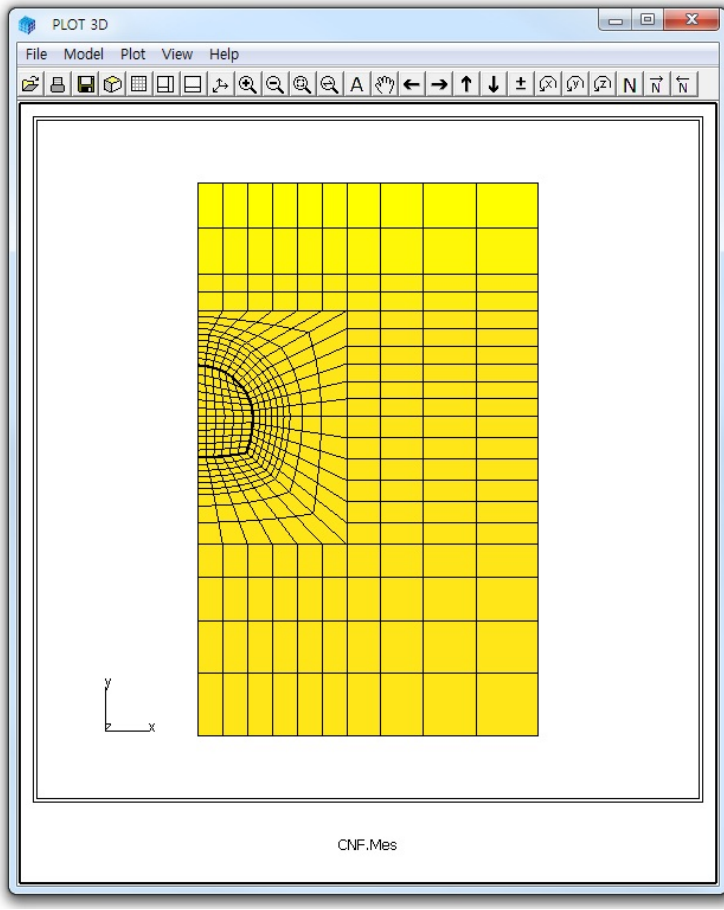


Figure 8.1 Final element meshes representing Core, Near-field, and Far-field regions, CNF.Mes

8.1.2 Modifying Mesh

In this example, we want to generate symmetric meshes using ADDRGN-2D. As the existing mesh, we take the CORE.Mes which has been generated using PRESMAP-2D Model 1 (refer to Section 7.1.1.2). Note that CORE.Mes represents the right side of the tunnel core. ADDRGN input file to generate Left Core is listed in Table 8.3. The output file LCORE.Mes contains Left Core whose graphical output is shown in Figure 8.2.

By combining both left and right core regions as instructed in Table 8.4, we can generate a whole core region, WCORE.Mes. Graphical output of WCORE.Mes is shown in Figure 8.3.

Table 8.3 Listing of input file ADD2D-3.Dat

```
* ADD2D-3.Dat
* CARD 1.1
* IMOD = 1 : MODIFY EXISTING MESH
  1
* CARD 3.1
* FILEA : Input file name to be modified
  CORE.Mes
* FILEM : Output file name to store modified mesh
  LCORE.Mes
* CARD 3.2
* NSNEL  NSNODE
  73     1
* CARD 3.3
* IEDIT = 0 : CHANGE COORDINATES
  0
* CARD 3.3.1.1
* Xo    Yo    Xonew   Yonew
  0.0   0.0   0.0     0.0
* CARD 3.3.1.2
* Xscale  Yscale
 -1.0      1.0
* END OF DATA
```

Table 8.4 Listing of input file ADD2D-4.Dat

```
* ADD2D-4.Dat
* CARD 1.1
* IMOD = 0 : ADD REGION B TO REGION A
  0
* CARD 2.1
* FILEA : Input file name containing REGION A
  CORE.Mes
* FILEB : Input file name containing REGION B
  LCORE.Mes
* FILEC : Output file name to store COMBINED REGION
  WCORE.Mes
* CARD 2.2
* INTERFACE
  0
* END OF DATA
```

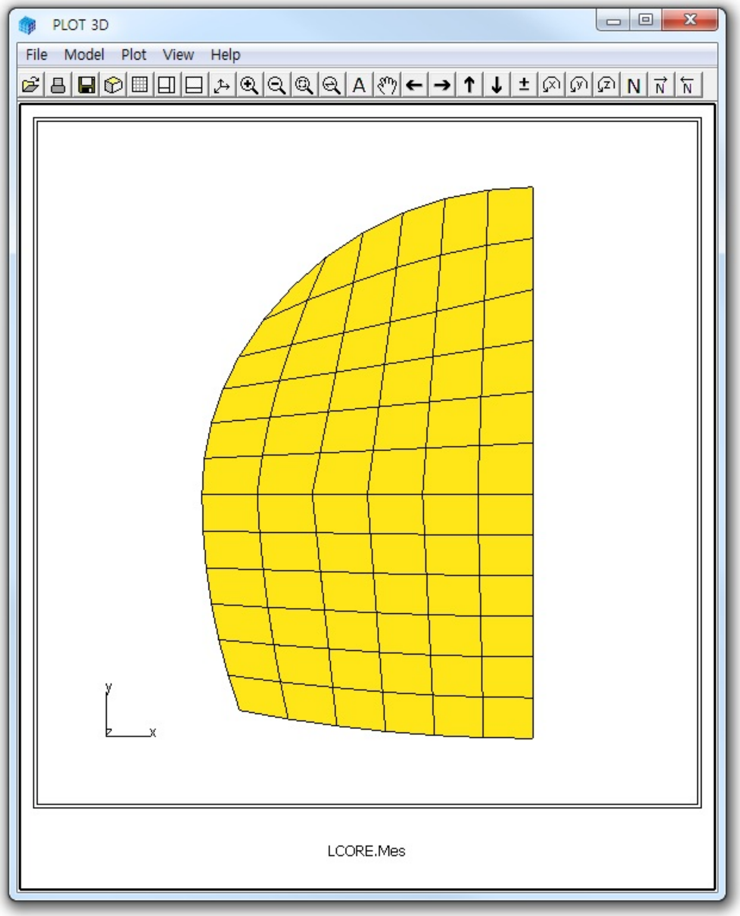


Figure 8.2 Left core mesh, LCORE.Mes

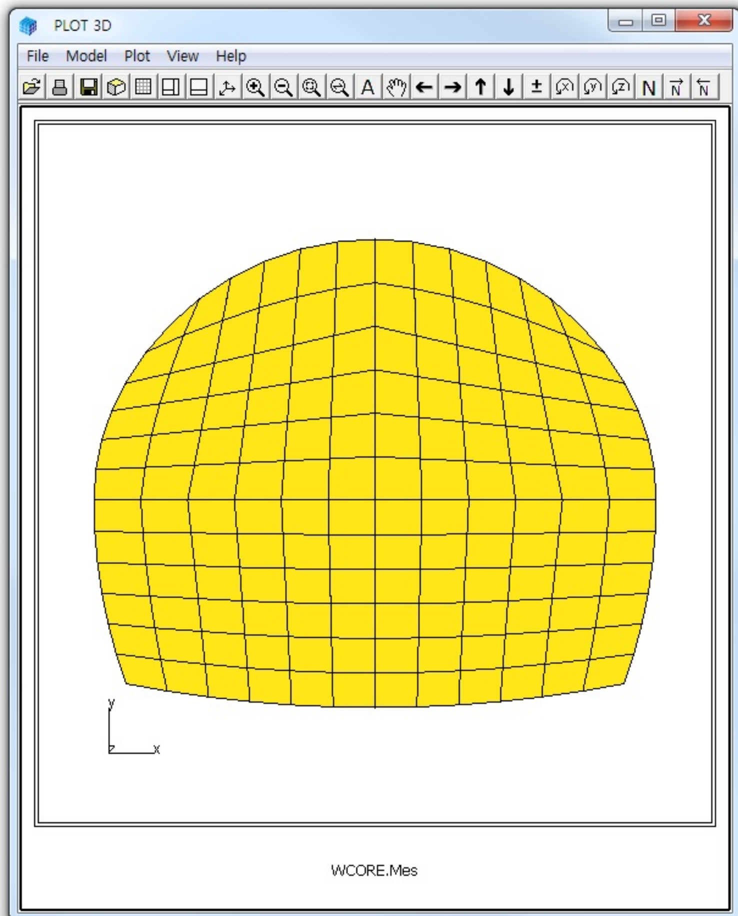


Figure 8.3 Combined whole core mesh, WCORE.Mes

8.1.3 Generating Mesh

This example is to show a powerful mesh generation feature using ADDRGN-2D. All you need to do is to specify the locations, dimensions and material numbers of structures along with few instructions for mesh generation. ADDRGN-2D will do the rest of the work to build the Mesh File.

As the first example, we take a simple problem as schematically shown in Figure 8.4. A utility tunnel with a diameter of 4 meters is located 6 meters below the ground surface. Table 8.5 shows the full listing of input file ADD2D-5.Dat. The base mesh consists of 3 blocks in the horizontal direction and 1 block in the vertical direction.

The first group represents soft rock underlying soil. And the second group represents the utility tunnel. Tunnel liner is modeled by beam element and the interface between the liner and the surrounding soil is modeled by joint element which will allow the slippage and separation. Finite element meshes generated by ADDRGN-2D are shown in Figures 8.5 and 8.6. It should be noted that the joint thickness in Figure 8.6 is exaggerated to show clearly both inner and outer joint faces. The real joint thickness is specified in material property card in Main File.

Table 8.5 Listing of input file ADD2D-5.Dat

```
* ADD2D-5.Dat
* CARD 1.1
* IMOD = 2 : GENERATE BASE MESH AND THEN MODIFY
* IMOD JK
   2    3
* CARD 4.1
* NBX  NBY
   3    1
* CARD 4.2
* XO    YO
  0.0   0.0
* CARD 4.3
* W     DX     ALPAX
  14.0  0.3   -0.3
  21.0  0.3    0.5
  11.0  0.3    0.3
* CARD 4.4
* H     DY     ALPAY
  20.0  0.3    0.5
* CARD 4.5
* IGMOD
  1
* -------------------------------------------------------------
* CARD 3.1
* FILEA
  BMESH.DAT
* FILEM
  ADD2D-5.Mes
* CARD 3.2
* NSNEL  NSNODE
      1       1
* CARD 3.3
* IEDIT = 4 : BUILD USER-SPECIFIED CURVES.
     4
* CARD 3.3.5.1
* NODE
     0
* CARD 3.3.5.2
* NOEL
     0
* CARD 3.3.5.3
* IBOUND
     0
```

```
* CARD 3.3.5.4
* NGROUP
    2
* XREF  YREF
  14.0  20.0
* --------------------- GROUP 1  ------------------------------
*
*                      SOFT ROCK
*
* CARD 3.3.5.4.1.1
* MTYPE
     3
* CARD 3.3.5.4.1.2
* MATNO  KF   LTPI  LMAT
     7   0     0     0
* CARD 3.3.5.4.2.1
* NPOINT  MOVE  IREF  XLO   YLO
     6     1     0    0.0   0.0
* CARD 3.3.5.4.2.2
* NP   X     Y
   1   0.0   0.0
   2  46.0   0.0
   3  46.0  13.0
   4  31.0  12.0
   5  19.0   8.0
   6   0.0   4.0
* CARD 3.3.5.4.3
* NSEGMENT
     6
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     1      1    0     3
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     2      1    0     3
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     3      1    0     2
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     4      1    0     2
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     5      1    0     2
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     6      1    0     2
```

```
* --------------------- GROUP 2 -------------------------------
*                  UTILITY TUNNEL
* CARD 3.3.5.4.1.1
* MTYPE
    -3
* CARD 3.3.5.4.1.2
* MATNO  KF  MATNOJT  KFJT  THICJT  LTPI, LMATI, LTPO, LMATO
     3   0     4       0     0.1     2     5      2     6
* CARD 3.3.5.4.2.1
* NPOINT  MOVE  IREF  XLO   YLO
     1      0    1    8.0  -6.0
* CARD 3.3.5.4.2.2
* NP   X     Y
   1   2.0   0.0
* CARD 3.3.5.4.3
* NSEGMENT
     1
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV IEND
     1      2     0   2
* CARD 3.3.5.4.3.2
* X0    Y0    RX   RY   THETA_B   THETA_E
  0.0   0.0   2.0  2.0  0.0       360.
* ------------------------------------------------------------
* END OF DATA
```

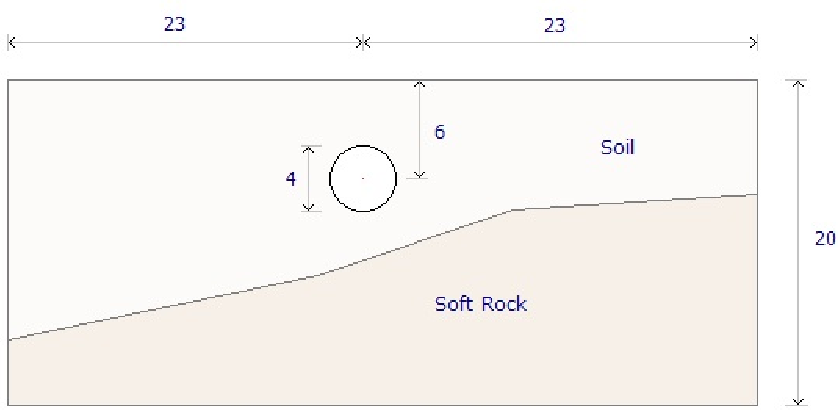


Figure 8.4 Schematic section view.

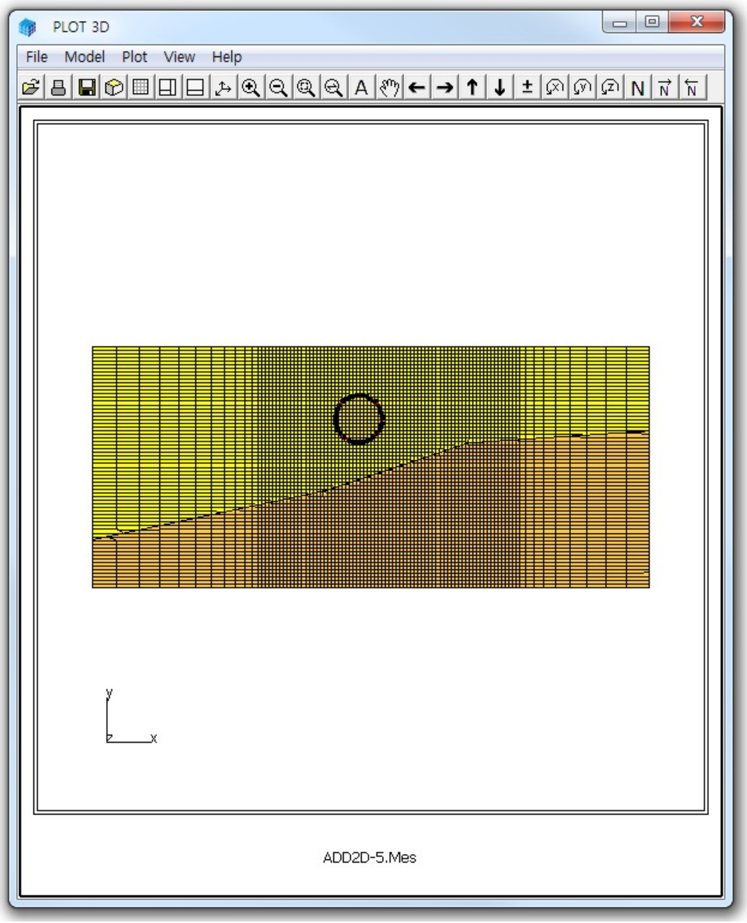


Figure 8.5 Overall finite element mesh

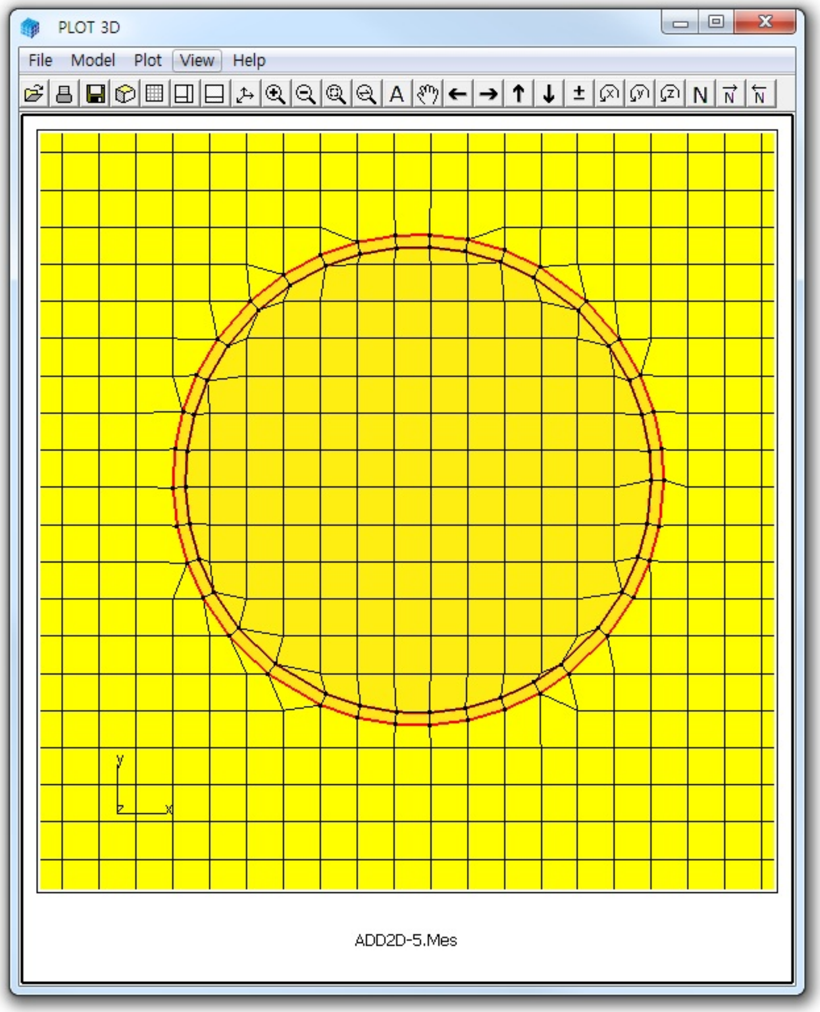


Figure 8.6 Finite element mesh around utility tunnel.

As the second example, we take a complex problem as schematically shown in Figure 8.7. The problem geometry includes different types of underground structures; strut, anchor bar, pile, utility tunnel, subway tunnel, rock bolt, foundation and fault zone. Table 8.6 shows the partial listing of input file ADD2D-6.Dat. The base mesh consists of 3 blocks in the horizontal direction and 2 blocks in the vertical direction.

For detailed description of input parameters, refer to Section 8.2 in SMAP-3D User's Manual. Joint elements are used to model the fault zone and the interfaces between surrounding medium and the structures such as pile and tunnels. Figure 8.8 shows overall finite element mesh generated by ADDRGN-2D. Detailed finite element meshes are shown in Figure 8.9 for the excavation zone and in Figure 8.10 for the tunnels and foundation. As in the previous example, the joint thickness in Figures 8.9 and 8.10 is exaggerated to show clearly both inner and outer joint faces. The real joint thickness is specified in material property card in Main File.

Table 8.6 Listing of input file ADD2D-6.Dat (Partial Listing)

```
* ADD2D-6.Dat
* CARD 1.1
* IMOD = 2 : GENERATE BASE MESH AND THEN MODIFY
* IMOD JK
   2    3
* CARD 4.1
* NBX  NBY
   3    2
* CARD 4.2
* XO    YO
  0.0   0.0
* CARD 4.3
* W     DX     ALPAX
  14.0  0.3   -0.3
  21.0  0.3    0.5
  11.0  0.3    0.3
* CARD 4.4
* H     DY     ALPAY
  23.0  0.3    0.5
  16.0  0.3    0.3
```

```
* CARD 4.5
* IGMOD
  1
* ------------------------------------------------------------
* CARD 3.1
* FILEA
  BMESH.DAT
* FILEM
  ADD2D-6.Mes
* CARD 3.2
* NSNEL  NSNODE
     1       1
* CARD 3.3
* IEDIT = 4 : BUILD USER-SPECIFIED CURVES.
     4
* CARD 3.3.5.1
* NODE
     0
* CARD 3.3.5.2
* NOEL
     0
* CARD 3.3.5.3
* IBOUND
     0
* CARD 3.3.5.4
* NGROUP
   22
* XREF  YREF
  14.0  39.0
* -------------------- GROUP 1  ------------------------------
*
*               MAKING GROUND SURFACE
*
* CARD 3.3.5.4.1.1
* MTYPE
    -1
* CARD 3.3.5.4.2.1
* NPOINT  MOVE  IREF  XLO   YLO
     8     1     0    0.0   0.0
* CARD 3.3.5.4.2.2
* NP   X     Y
   1  46.0  34.0
   2  39.0  34.0
   3  33.0  39.0
   4  18.0  39.0
   5  12.0  34.0
   6   0.0  34.0
   7   0.0   0.0
   8  46.0   0.0
```

```
* CARD 3.3.5.4.3
* NSEGMENT
     8
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     1      1     0     2
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEDN
     2      1     0     2
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     3      1     0     2
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEDN
     4      1     0     2
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     5      1     0     2
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEDN
     6      1     0     2
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     7      1     0     2
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEDN
     8      1     0     2
* --------------------- GROUP 2  ------------------------------
*
*                      SOFT ROCK
*
* CARD 3.3.5.4.1.1
* MTYPE
     3
* CARD 3.3.5.4.1.2
* MATNO  KF   LTPI  LMAT
     7   0     0     0
* CARD 3.3.5.4.2.1
* NPOINT  MOVE  IREF  XLO   YLO
     6     1     0    0.0   0.0
* CARD 3.3.5.4.2.2
* NP   X     Y
   1  46.0   0.0
   2  46.0  33.0
   3  31.0  32.0
   4  19.0  28.0
   5   0.0  24.0
   6   0.0   0.0
```

```
* CARD 3.3.5.4.3
* NSEGMENT
     6
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     1      1     0     3
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     2      1     0     3
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     3      1     0     0
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     4      1     0     0
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     5      1     0     0
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     6      1     0     3
* --------------------- GROUP 3  ------------------------------
*
*                        FAULT
*
* MTYPE
    -2
* CARD 3.3.5.4.1.2
* MATNOJT  KFJT  THICJT  LTPI, LMATI, LTPO, LMATO
    5        0   -0.1     0     0      0     0
* CARD 3.3.5.4.2.1
* NPOINT  MOVE  IREF  XLO   YLO
     4     1     0    0.0   0.0
* CARD 3.3.5.4.2.2
* NP    X      Y
   1  46.0  29.0
   2  29.0  19.0
   3  16.0  14.0
   4   0.0  10.0
* CARD 3.3.5.4.3
* NSEGMENT
     3
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     1      1     0     2
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEDN
     2      1     0     2
```

```
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     3      1     0     2
* --------------------- GROUP 4  ------------------------------
*
*                      FOUNDATION
*
* CARD 3.3.5.4.1.1
* MTYPE
     4
* CARD 3.3.5.4.1.2
* MATNO   KF  LTPI  LMAT
     2    0    0     0
* CARD 3.3.5.4.2.1
* NPOINT  MOVE  IREF  XLO   YLO
     8     1     0    0.0   0.0
* CARD 3.3.5.4.2.2
* NP   X     Y
   1  34.5  29.0
   2  34.5  30.0
   3  32.5  30.5
   4  32.5  39.0
   5  31.5  39.0
   6  31.5  30.5
   7  29.5  30.0
   8  29.5  29.0
* CARD 3.3.5.4.3
* NSEGMENT
     8
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     1      1     0     2
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     2      1     0     2
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     3      1     0     2
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     4      1     0     2
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     5      1     0     2
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     6      1     0     2
* CARD 3.3.5.4.3.1
```

```
* SEGNO  LTYPE  NDIV  IEND
     7      1    0     2
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     8      1    0     2
* --------------------- GROUP 5 -------------------------------
*
*                 LEFT UTILITY TUNNEL
*
* CARD 3.3.5.4.1.1
* MTYPE
    -3
* CARD 3.3.5.4.1.2
* MATNO  KF  MATNOJT  KFJT  THICJT  LTPI, LMATI, LTPO, LMATO
     3   0     4       0    -0.1     2     5      2     6
* CARD 3.3.5.4.2.1
* NPOINT  MOVE  IREF  XLO   YLO
     1      0    1    8.0  -6.0
* CARD 3.3.5.4.2.2
* NP   X     Y
   1   2.0   0.0
* CARD 3.3.5.4.3
* NSEGMENT
     1
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV IEND
     1      2     0   2
* CARD 3.3.5.4.3.2
* X0    Y0    RX   RY   THETA_B   THETA_E
  0.0   0.0   2.0  2.0  0.0       360.
* --------------------- GROUP 6 -------------------------------
*
*                 RIGHT UTILITY TUNNEL
*
* CARD 3.3.5.4.1.1
* MTYPE
    -3
* CARD 3.3.5.4.1.2
* MATNO  KF  MATNOJT  KFJT  THICJT  LTPI, LMATI, LTPO, LMATO
     3   0     4       0    -0.1     2     5      2     6
* CARD 3.3.5.4.2.1

---
---
```

```
* --------------------- GROUP 22 ------------------------------
*
*                       SUBWAY TUNNEL
*
* CARD 3.3.5.4.1.1
* MTYPE IGPOST OVERLAY GCOLOR GLTYPE GLTHIC GHIDE
  -3     0      0       0      0      0      0
* Card 3.3.5.4.1-1
* MAT  KF  MATj KFj  THICj  LTi LMi LTo LMo
    3   0    4   0  -0.100   2   5   2   6
* CARD 3.3.5.4.2.1
* NPOINT  MOVE  IREF  XLO   YLO
    4      1     1    0.0   0.0
* CARD 3.3.5.4.2.2
* NP   X     Y
   1   26.  24.
   2   20.  24.
   3   20.  20.
   4   26.  20
* CARD 3.3.5.4.3
* NSEGMENT
     4
* CARD 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV  IEND
     1      2    0     2
* Card 3.3.5.4.3.1-1
* Xo     Yo     Rx    Ry    Qb    Qe
  23.    24.    3.0   3.0   0.0   180.
* Card 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV   IEND
    2     1       0     2
* Card 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV   IEND
    3     1       0     2
* Card 3.3.5.4.3.1
* SEGNO  LTYPE  NDIV   IEND
    4     1       0     2
* ------------------------------------------------------------
* END OF DATA
```

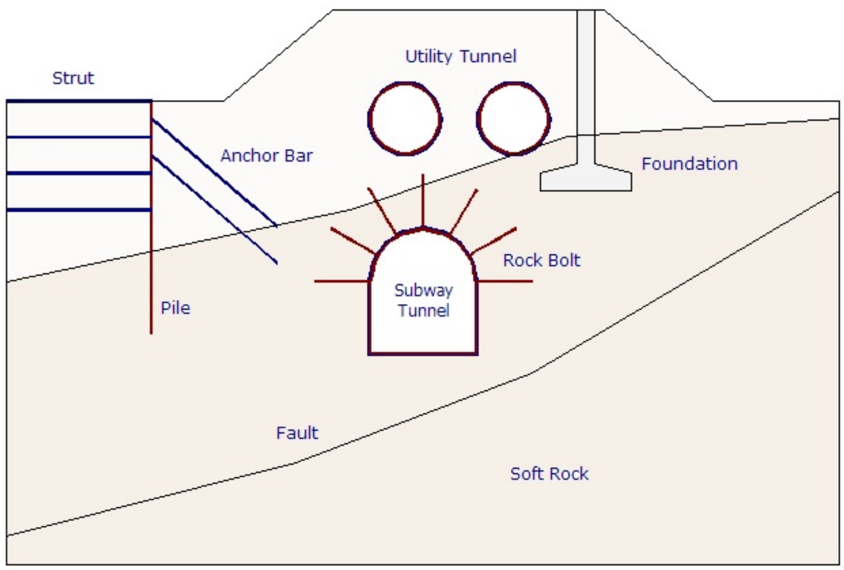


Figure 8.7 Schematic section view

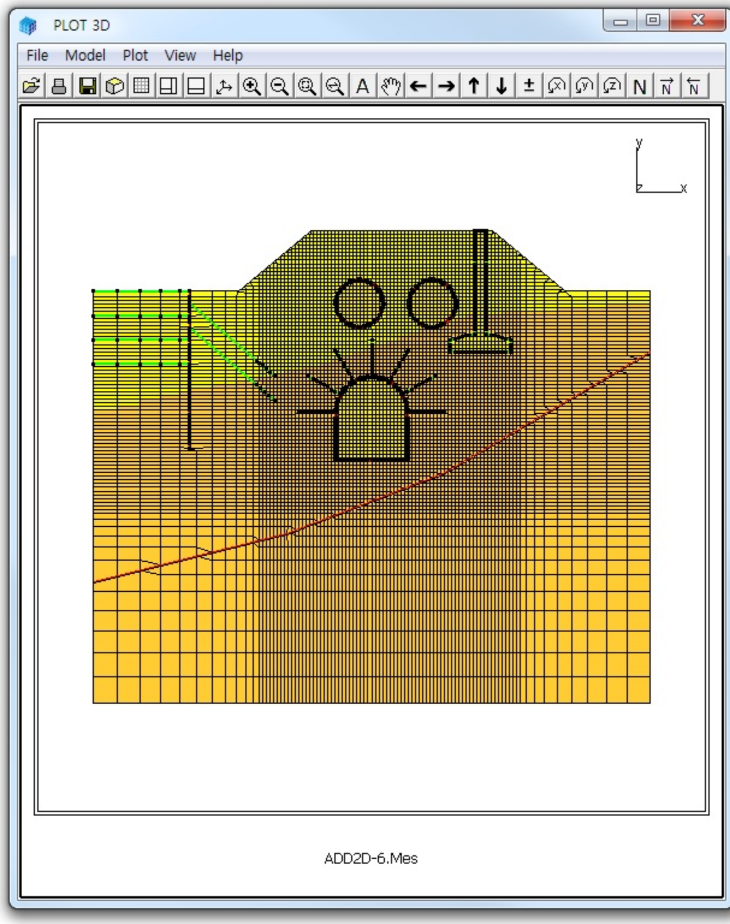


Figure 8.8 Overall finite element mesh

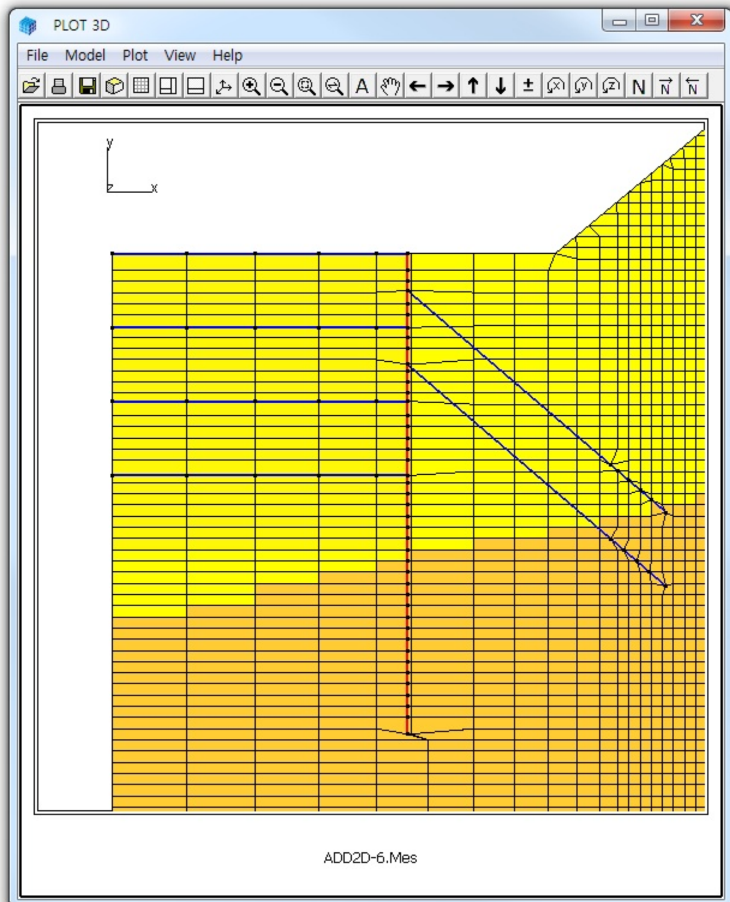


Figure 8.9 Finite element mesh around excavation zone

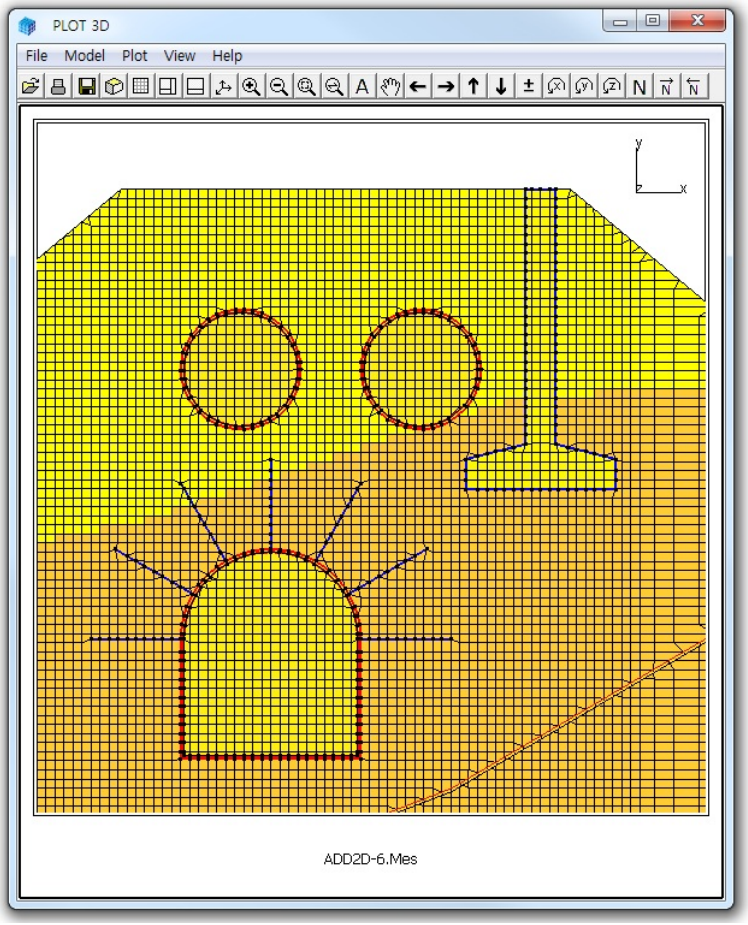


Figure 8.10 Finite element mesh around tunnels and foundation

8.2 ADDRGN-3D

ADDRGN-3D is mainly used to combine or modify the existing three dimensional continuum meshes.

ADDRGN-3D can be selected in the following order:

Run → Mesh Generator → AddRgn → Addrgn 3D

When you finish the execution of ADDRGN-3D, select PLOT-3D to plot the combined or modified mesh.

8.2.1 Combining Meshes

In this example, ADDRGN-3D is used to combine two different regions; FARA3D and FARB3D. FARA3D has 63 nodes and 24 elements as shown in Figure 8.11. FARB3D has 84 nodes and 36 elements as shown in Figure 8.12.

Both FARA3D and FARB3D have the common plane at y=10 where both regions share the identical nodal coordinates. Element numbers 1 to 24 are assigned for FARA3D and 25 to 60 for FARB3D. Note that element numbers should be continuous when combining two regions. The output file, FAR3D.Mes, is obtained by adding FARB3D.Mes (called REGION B) to FARA3D.Mes (called REGION A) as specified in input file ADD3D-1.Dat in Table 8.7. Graphical output for the combined region FAR3D is presented in Figure 8.13.

This example also demonstrates that ADDRGN-3D combines not only meshes but also nodal damping constants for transmitting boundary. Combined nodal damping constants are listed at the end of element indexes in the output file FAR3D.Mes.

Table 8.7 Listing of input file ADD3D-1.Dat

```
* ADD3D-1.Dat
* CARD 1.1
* IMOD = 0 : ADD REGION B TO REGION A
  0
* CARD 2.1
* FILEA : Input file name containing REGION A
  FARA3D.Mes
* FILEB : Input file name containing REGION B
  FARB3D.Mes
* FILEC : Output file name to store COMBINED REGION
  FAR3D.Mes
* END OF DATA
```

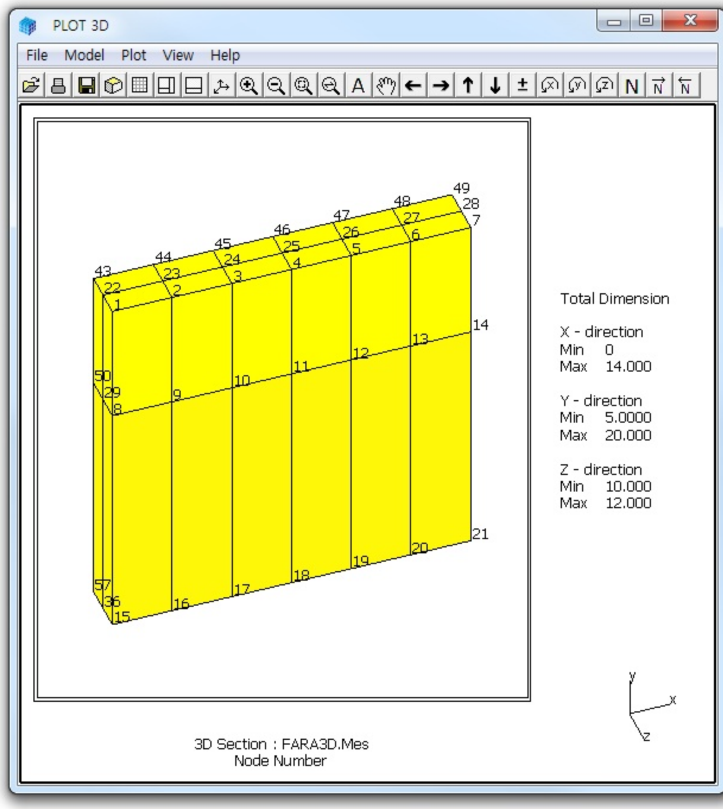


Figure 8.11 Node numbers for FARA3D.Mes

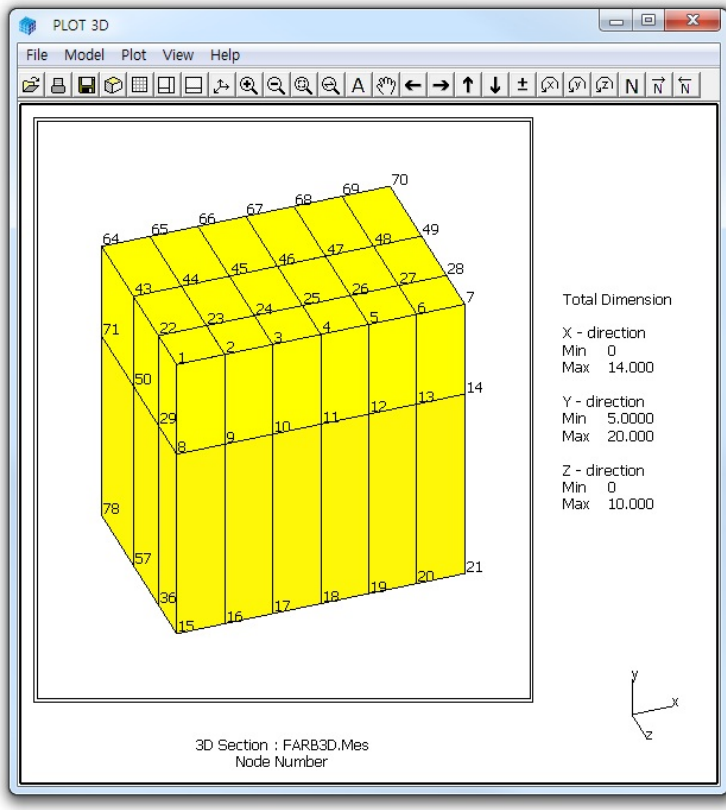


Figure 8.12 Node numbers for FARB3D.Mes

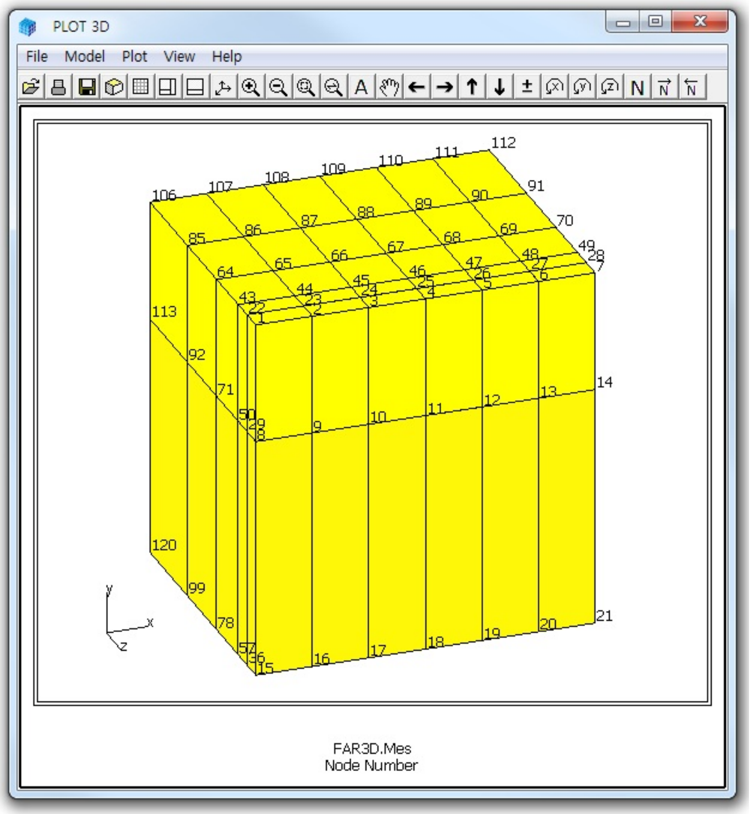


Figure 8.13 Node numbers for generated output FAR3D.Mes

8.2.2 Modifying Mesh

In this example, we want to generate symmetric meshes using ADDRGN-3D. As the existing mesh, we take the CROSS-3D Model 1 example problem, but generates only top half by specifying IPART=1 in card Group 2.1.2. Table 8.8 shows the listing of CROSS-3D input file CRM1-TOP.Dat. Graphical output is shown in Figure 8.14.

To generate bottom half which is symmetric about the plane at y=0, you can execute the input file ADD3D-2.Dat in Table 8.9. Graphical output is shown in Figure 8.15.

By combining both top and bottom regions, we can generate a whole region WCRM1.Mes. Input file ADD3D-3.Dat in Table 8.10 is used to build the combined mesh. Graphical output of this combined region is shown in Figure 8.16.

Table 8.8 Listing of CROSS-3D input file CRM1-TOP.Dat

```
* CARD 1.1
* TITLE
  Identical two crossing tunnels (MODELNO = 1)
* CARD 1.2
* MODELNO  KF  NSNODE  NSNEL  CMFAC
    1       1    1       1     1.0
* CARD 2.1.1
* XL    YB    YT    ZL    t
 100.   50.   100.  100.  3.0
* CARD 2.1.2
* IPART  NDR  NTBND  NTOPN
    1      2   20     20
* CARD 2.1.3
* NTNODE
    9
* NODE  X       Y
    1   0.0     4.0
    2   2.8284  2.8284
    3   4.0     0.0
    4   4.0    -2.0
    5   0.0    -3.0
    6   1.53    3.7
    7   3.7     1.53
    8   4.0    -1.0
    9   2.0    -2.7
* CARD 3.1
* NBOUND
    6
* CARD 3.2
* IBTYPE ISX  ISY  ISZ  IFX  IFY  IFZ
    1     0    0    0    1    1    1
    2     0    0    1    1    1    1
    3     0    0    1    1    1    1
    4     1    0    0    1    1    1
    5     1    0    0    1    1    1
    7     0    1    0    1    1    1
* END OF DATA
```

Table 8.9 Listing of input file ADD3D-2.Dat

```
* ADD3D-2.Dat
* CARD 1.1
* IMOD = 1 : MODIFY EXISTING MESH
  1
* CARD 3.1
* FILEA : Input file name to be modified
  CRM1-TOP.Mes
* FILEM : Output file name to store modified mesh
  CRM1-BOT.Mes
* CARD 3.2
* NSNEL  NSNODE
  746    1
* CARD 3.3
* IEDIT = 0 : CHANGE COORDINATES
  0
* CARD 3.3.1.1
* Xo    Yo    Zo      Xonew   Yonew   Zonew
  0.0   0.0   0.0     0.0     0.0     0.0
* CARD 3.3.1.2
* Xscale  Yscale  Zscale
  1.0     -1.0     1.0
* END OF DATA
```

Table 8.10 Listing of input file ADD3D-3.Dat

```
* ADD3D-3.Dat
* CARD 1.1
* IMOD = 0 : ADD REGION B TO REGION A
  0
* CARD 2.1
* FILEA : Input file name containing REGION A
  CRM1-TOP.Mes
* FILEB : Input file name containing REGION B
  CRM1-BOT.Mes
* FILEC : Output file name to store COMBINED REGION
  WCRM1.Mes
* END OF DATA
```

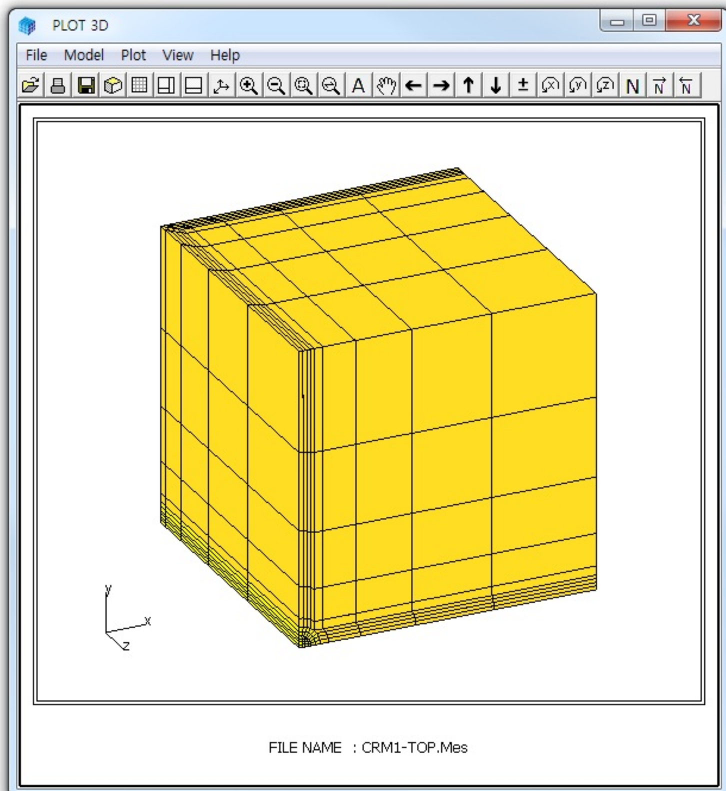


Figure 8.14 Finite element mesh for top half region

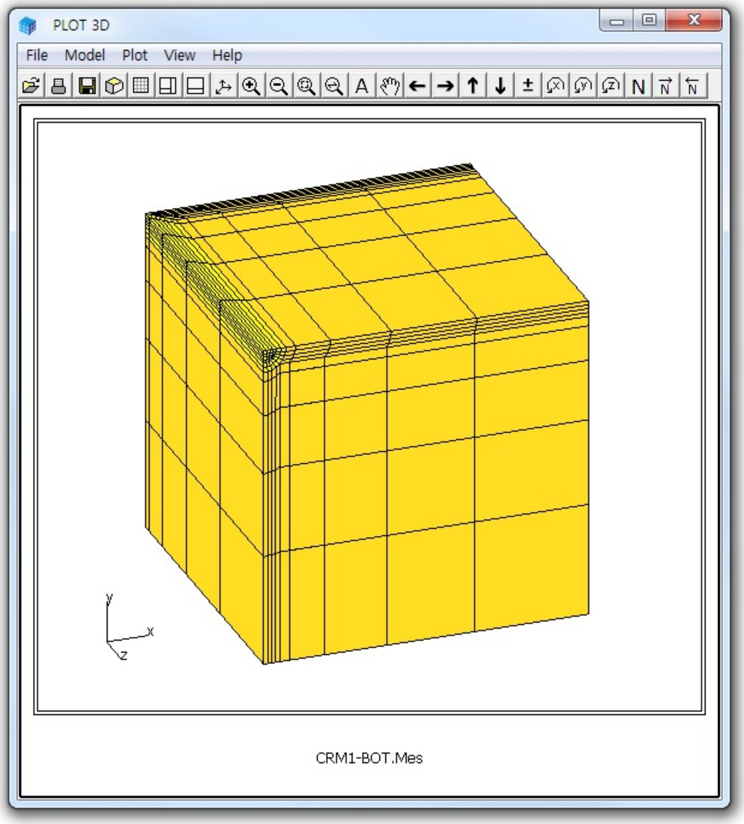


Figure 8.15 Finite element mesh for bottom half region

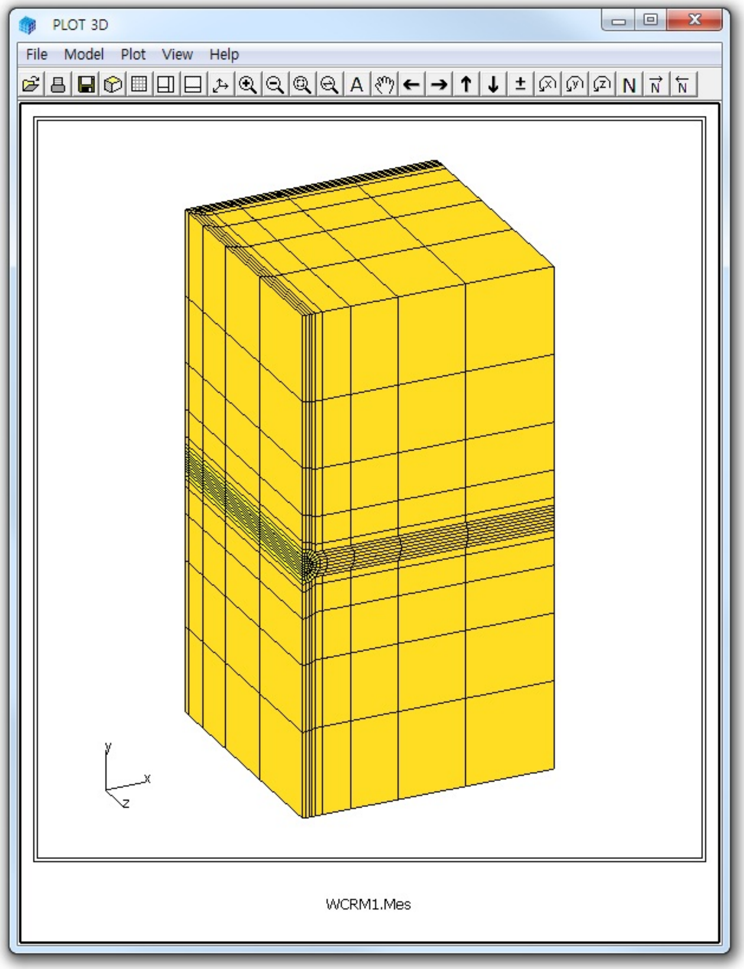


Figure 8.16 Finite element mesh for combined region

SUPPLEMENT
Example Problem

SUPPLEMENT Menu contains supporting programs which are useful to prepare input data for pre- and main-processing programs of SMAP-3D.

Running SUPPLEMENT is described in Section 3.2.6 of User's Manual and can be selected in the following order:

Run → Mesh Generater → Supplement → Edit, XY. Cards or Shrink File

EDIT is used to run text editor.

XY computes coordinates of mid points, cross points, or normal points.

CARDS generates element activity data in Card 8 in Section 4.4 Main File.

SHRINK FILE removes extra blank spaces before carriage return.
This will reduce the size of the file.

9.1 XY Example Problem

XY is the supporting program which computes coordinates of mid points, cross points, or normal points. Full description of XY is presented in Section 9.3 of User's Manual.

As an example, we select NF=6 which computes coordinates of point normal to the circular arc as shown in Figure 9.1.

Table 9.1 illustrates options available to the program XY and the user inputs specific to NF=6. Computed coordinates of the normal point are stored in the output file XY.Out and are listed in bottom part of Table 9.l.

Table 9.1 XY Example Problem

Type file name to store output: XY.Out

NF =	0	END OF COMPUTATION.
	1	COMPUTE MIDPOINT ON STRAIGHT LINE.
	2	COMPUTE MIDPOINT ON CIRCULAR ARC.
	3	COMPUTE INTERSECTION POINT OF TWO STRAIGHT LINES.
	4	COMPUTE INTERSECTION POINT OF CIRCULAR ARC AND STRAIGHT LINE.
	5	COMPUTE POINTS NORMAL TO STRAIGHT LINE.
	6	COMPUTE POINTS NORMAL TO CIRCULAR ARC.

NF= **6**

R,	Xo,	Yo,	TA
5.0	**0.0**	**0.0**	**0.0**
TAC,	CD		
45.0	**3.0**		

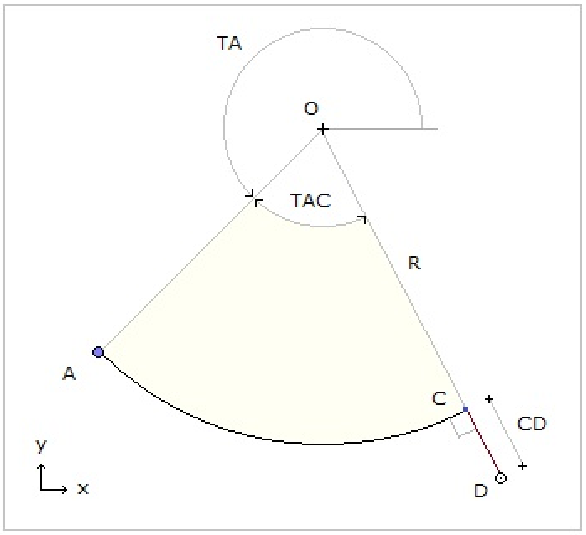
User inputs are **bold**.

Output file contains following information:

COMPUTED POINTS NORMAL TO CIRCULAR ARC

R	= 5.000000		
Xo	= 0.000000E+00	Yo	= 0.000000E+00
TA	= 0.000000E+00		
TAC	= 45.000000	CD	= 3.000000
XC	= 3.535527	YC	= 3.535540
XD	= 5.656844	YD	= 5.656865

NF = 6 Compute Points Normal to Circular Arc



INPUT:
R, X_o, Y_o, TA
TAC, CD

R	=	5.0
X_o, Y_o	=	0.0, 0.0
TA	=	0.0
TAC	=	45.0
CD	=	3.0

Figure 9.1 XY example problem

9.2 CARDS Example Problem

CARDS is the supporting program which is written to aid the preparation of SMAP-3D input cards. Currently, there is only one routine available to generate element activity data in Card Group 8.2 of Users Manual.

Table 9.2 shows user inputs for the example problem. Generated element activity data is stored in the output file, CARDS.Out, which is listed in Table 9.3.

Table 9.2 User inputs for CARDS example problem

CARD NO =	0	EXIT
	8.2	ELEMENT ACTIVITY

CARD NO = **8.2**
Type file name to store output: **CARDS.OUT**

NF =	0	END OF GENERATION
	1	GENERATE ELEMENT ACTIVITY/ DEACTIVITY

NF = **1**
NEL (start), NEL (end), NAC, NDAC
101 120 0 6

NF =	0	END OF GENERATION
	1	GENERATE ELEMENT ACTIVITY/ DEACTIVITY

NF = **1**
NEL (start), NEL (end), NAC, NDAC
121 130 3 50

NF =	0	END OF GENERATION
	1	GENERATE ELEMENT ACTIVITY/ DEACTIVITY

NF = **0**

User inputs are **bold**.

Table 9.3 Listing of output file CARDS.Out

```
* NEL   NAC  NDAC
*
  101     0     6
  102     0     6
  103     0     6
  104     0     6
  105     0     6
  106     0     6
  107     0     6
  108     0     6
  109     0     6
  110     0     6
  111     0     6
  112     0     6
  113     0     6
  114     0     6
  115     0     6
  116     0     6
  117     0     6
  118     0     6
  119     0     6
  120     0     6
*
  121     3    50
  122     3    50
  123     3    50
  124     3    50
  125     3    50
  126     3    50
  127     3    50
  128     3    50
  129     3    50
  130     3    50
* NFAD =    30
```

LOAD
Example Problem

10.1 LOAD-2D

LOAD-2D is the pre-processing program which can be used to generate external force (pressure), specified velocity, initial velocity, acceleration, and transmitting boundary. For the detailed description of input parameters, refer to section 11 of User's Manual.

LOAD-2D can be selected in the following order:

Run → Load Generator → Load 2D

When you select LOAD-2D, Load Generation Dialog will be displayed as in Figure 10.1. You need to specify input file names for Load and Mesh Data.

10.1.1 Example 1

Example 1 is to show the pressure load generation along the surfaces of elements 1, 2, 3 and 4 as schematically shown in Figure 10.2. Triangular pressure loads are acting on the surfaces of elements 1, 2 and 3. Right surfaces of elements 3 and 4 are subjected to the uniformly distributed pressure of 1.0. Two different load time histories, as shown in Figure 10.3, are considered.

Mesh Data contains information for nodal coordinates and element indexes. Mesh2D.Mes represents Mesh Data graphically shown in Figure 10.4 along with listing in Table 10.1. Load Data contains information for loads to be generated. Load2D.Dat in Table 10.2, has been prepared according to LOAD-2D User's Manual.

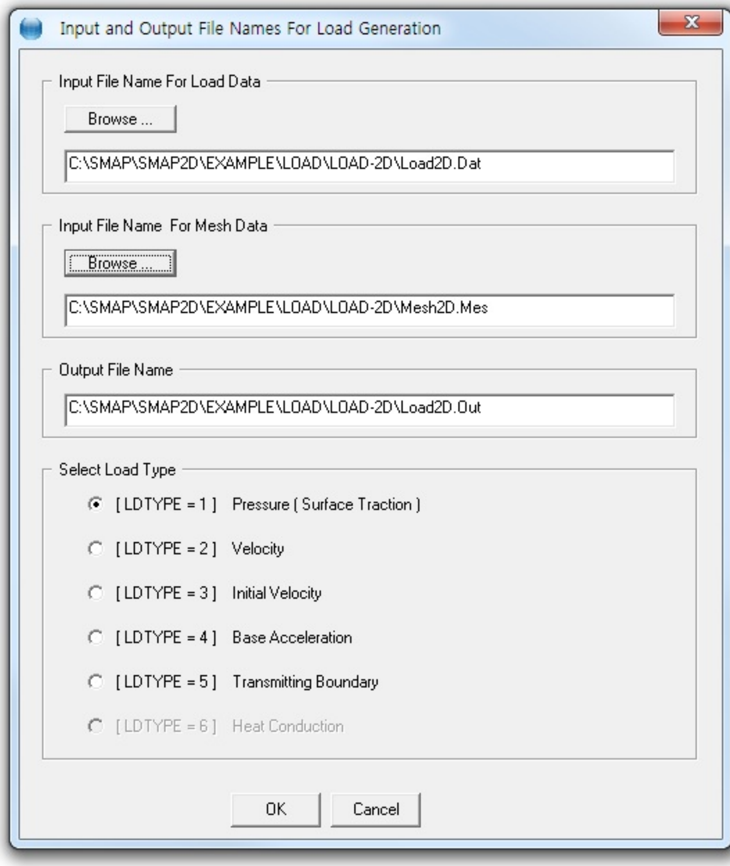
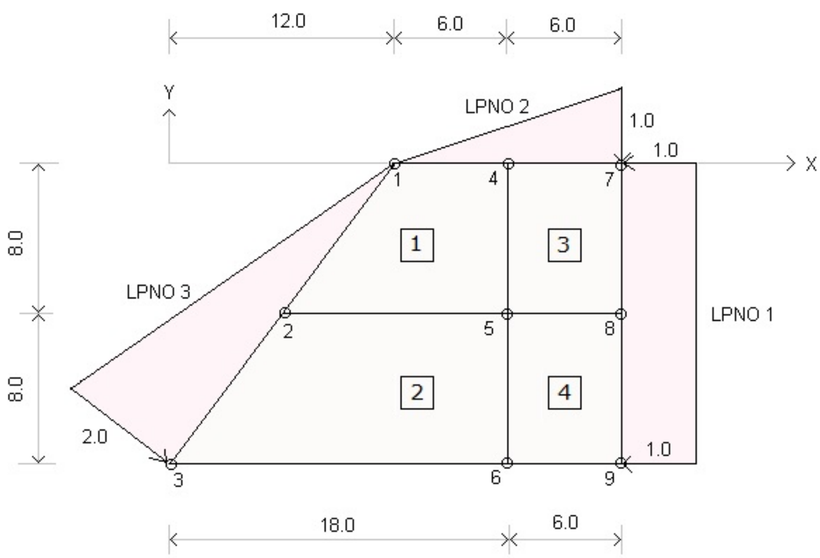


Figure 10.1 Load generation dialog



Loading Surface 1 (LSNO = 1) consists of nodes 9, 7, 8

Loading Surface 2 (LSNO = 2) consists of nodes 7, 4, 1

Loading Surface 3 (LSNO = 3) consists of nodes 1, 2, 3

Pressure Function 1 (LPNO = 1) $P_x = -1.0$

Pressure Function 2 (LPNO = 2) $P_Y = 1.0 - (1/12)X$

Pressure Function 3 (LPNO = 3) $P_n = -0.125Y$

Note: Mesh is axially symmetric about Y axis

Figure 10.2 Schematic view of pressure loads for Example 1

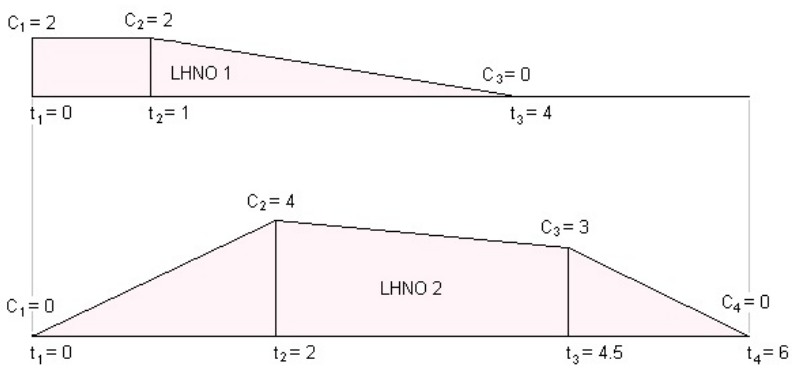


Figure 10.3 Load time histories for Example 1

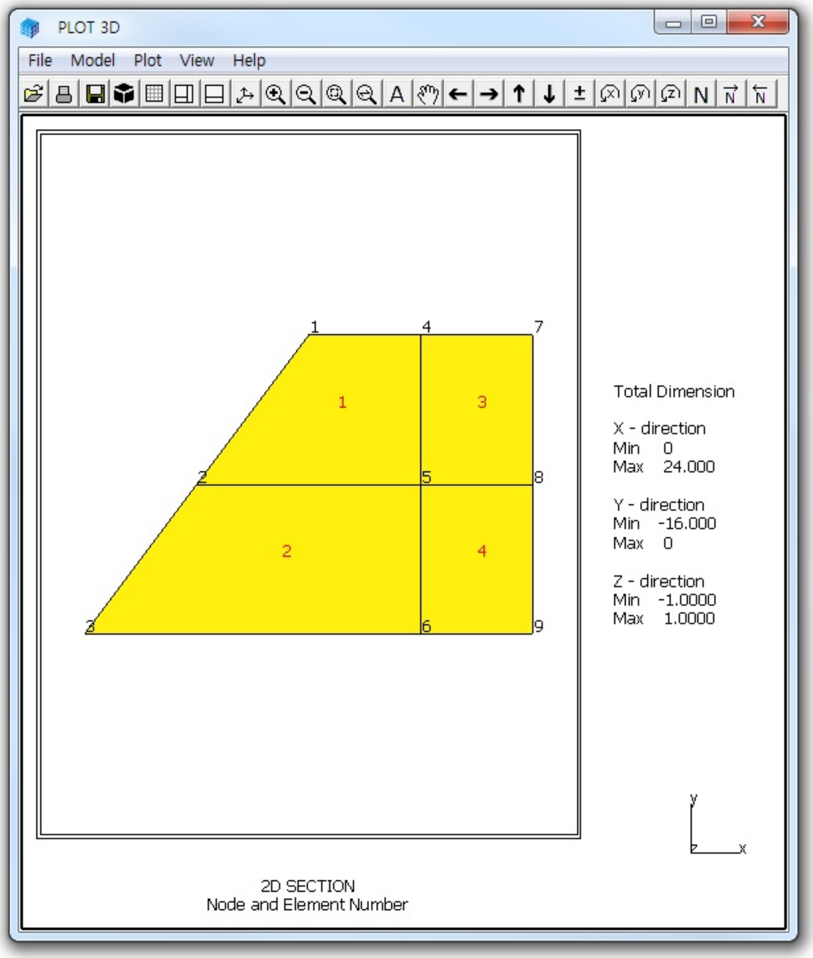


Figure 10.4 Finite element mesh for Example 1

Table 10.1 Listing of mesh data input file Mesh2D.Mes for Example 1

```
2D SECTION
NUMNP   NCONT   NBEAM  NTRUS
   9      4       0      0
NODAL COORDINATES
NODE  ISX  ISY  IFX  IFY  IRZ      XC           YC
  1    1    0    1    1    1      12.            0.
  2    0    0    1    1    0       6.           -8.
  3    0    0    1    1    0       0.          -16.
  4    0    0    1    1    0      18.            0.
  5    0    0    1    1    0      18.           -8.
  6    0    0    1    1    0      18.          -16.
  7    0    0    1    1    0      24.            0.
  8    0    0    1    1    0      24.           -8.
  9    0    0    1    1    0      24.          -16.
ELEMENT INDEX
NEL   I1   I2   I3   I4   M5   M6   M7   M8 MATC KS KF IR IS TBJWL
  1    4    1    2    5    0    0    0    0    4  0  1 2  2  .000E+00
  2    5    2    3    6    0    0    0    0    4  0  1 2  2  .000E+00
  3    7    4    5    8    0    0    0    0    4  0  1 2  2  .000E+00
  4    8    5    6    9    0    0    0    0    4  0  1 2  2  .000E+00
```

Table 10.2 Listing of load data inut file Load2D.Dat for Example 1

```
*
* LOAD-2D  INPUT
*
* CARD 1.1
* TITLE
  EXAMPLE 1  LOAD-2D Pressure [LDTYPE = 1]
* -----------------------------------------
* CARD 1.2
* NCTYPE
   0
* =========================================
* CARD 2.1
* NUMLS
   3
* -----------------------------------------
* CARD 2.2.1
* LSNO
   1
* CARD 2.2.2
* NUMNODE
   3
* CARD 2.2.3
* LISTING OF NODES
   9, 7, 8
* -----------------------------------------
* CARD 2.2.1
* LSNO
   2
* CARD 2.2.2
* NUMNODE
   3
* CARD 2.2.3
* LISTING OF NODES
   7, 4, 1
* -----------------------------------------
* CARD 2.2.1
* LSNO
   3
* CARD 2.2.2
* NUMNODE
   3
```

```
* CARD 2.2.3
* LISTING OF NODES
  1, 2, 3
* =========================================
* CARD 3.1
* NUMLF
   3
* -----------------------------------------
* CARD 3.2.1
* LFNO   LPTYPE
   1      0
* CARD 3.2.2
* A-X0   A-XX   A-XY
  -1.,   0.0,   0.0
* CARD 3.2.3
* A-Y0   A-YX   A-YY
  0.0,   0.0,   0.0
* CARD 3.2.4
* A-N0   A-NX   A-NY
  0.0,   0.0,   0.0
* -----------------------------------------
* CARD 3.2.1
* LFNO   LPTYPE
   2      0
* CARD 3.2.2
* A-X0   A-XX   A-XY
  0.0,   0.0,   0.0
* CARD 3.2.3
* A-Y0   A-YX   A-YY
  1.0,-0.083333,0.0
* CARD 3.2.4
* A-N0   A-NX   A-NY
  0.0,   0.0,   0.0
* -----------------------------------------
* CARD 3.2.1
* LFNO   LPTYPE
   3      1
* CARD 3.2.2
* A-X0   A-XX   A-XY
  0.0,   0.0,   0.0
* CARD 3.2.3
* A-Y0   A-YX   A-YY
  0.0,   0.0,   0.0
```

```
* CARD 3.2.4
* A-N0    A-NX    A-NY
  0.0,    0.0,   -0.125
* =========================================
* CARD 4.1
* NUMLH
   2
* -----------------------------------------
* CARD 4.2.1
* LHNO
   1
* CARD 4.2.2
* NUMTP
   3
* CARD 4.2.3
* T1    T2    T3
  0.0   1.0   4.0
* CARD 4.2.4
* C1    C2    C3
  2.0   2.0   0.0
* -----------------------------------------
* CARD 4.2.1
* LHNO
   2
* CARD 4.2.2
* NUMTP
   4
* CARD 4.2.3
* T1    T2    T3    T4
  0.0   2.0   4.5   6.0
* CARD 4.2.4
* C1    C2    C3    C4
  0.0   4.0   3.0   0.0
* =========================================
* CARD 5.1
* LSNO  LFNO  LHNO
   1,    1,    1
   2,    2,    1
   3,    3,    2
   0,    0,    0
* END OF INPUT DATA
```

The output file, Load2D.Out listed in Table 10.3, contains generated concentrated nodal forces and load time histories. Figure 10.5 shows time history curves for each load history number. The format of the generated load output is compatible to the format of Card Group 9 in SMAP-2D main input.

Table 10.3 Listing of load output file Load2D.Out for Example 1

```
* CARD 9.2.1
* NUMLP
   12
* LOAD HISTORY NO:  1
* CARD 9.2.2
* NODE   IDOF   LHNO   CINT
    1      2      1   -.74998E+01
    4      2      1   -.56999E+02
    7      1      1   -.96000E+02
    7      2      1   -.55500E+02
    8      1      1   -.19200E+03
    9      1      1   -.96000E+02
* LOAD HISTORY NO:  2
* CARD 9.2.2
* NODE   IDOF   LHNO   CINT
    1      1      2    .12000E+02
    1      2      2   -.90000E+01
    2      1      2    .40000E+02
    2      2      2   -.30000E+02
    3      1      2    .12000E+02
    3      2      2   -.90000E+01
* END OF LOAD HISTORY
* CARD 9.2.3.1
* NTFUN  NUMLH
    0      2
* CARD 9.2.3.2
* NUMTP  NTYPE   DTXX
    6      1    .00000E+00
* CARD 9.2.3.3
* LISTING OF TIME POINTS
  .0000E+00  .10000E+01  .20000E+01  .40000E+01  .4500E+01  .6000E+01
* CARD 9.2.3.4
* LISTING OF LOAD FOR HISTORY NO:  1
  .2000E+01  .20000E+01  .13333E+01 -.59605E-07  .0000E+00  .0000E+00
* CARD 9.2.3.4
* LISTING OF LOAD FOR HISTORY NO:  2
  .0000E+00  .20000E+01  .40000E+01  .32000E+01  .3000E+01  .0000E+00
* END OF LOAD DATA
```

Generated load vectors for concentrated forces can be plotted graphically. Refer to the step by step procedure in the file Running LOAD-2D.pdf.
The effect of LPTYPE (Effective vs Actual Surface) is described in the file LOAD-2D Example.pdf.

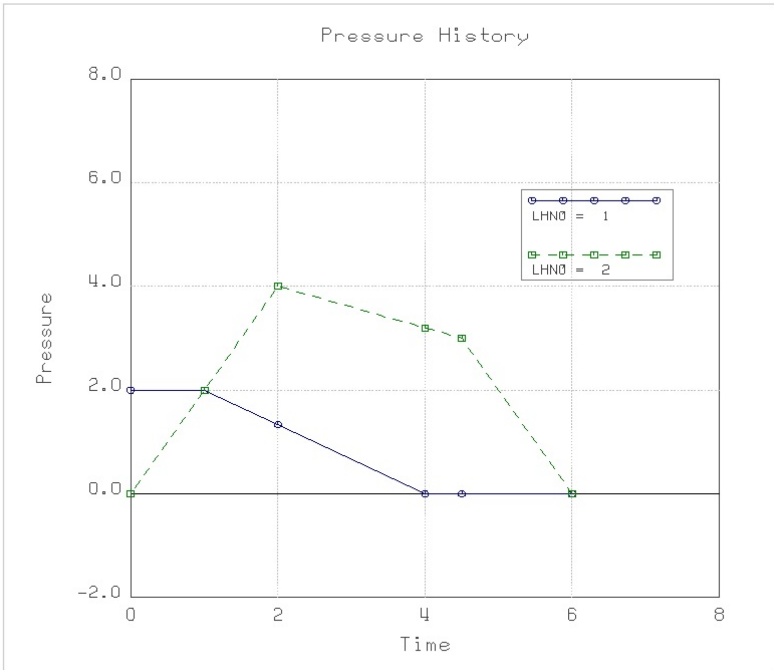


Figure 10.5 Generated load time histories for Example 1

10.2 LOAD-3D

LOAD-3D is the pre-processing program which can be used to generate external force (pressure), specified velocity, initial velocity, acceleration, and transmitting boundary. For the detailed description of input parameters, refer to section 11 of User's Manual.

LOAD-3D can be selected in the following order:

Run → Load Generator → Load 3D

When you select LOAD-3D, Load Generation Dialog will be displayed as in Figure 10.6. You need to specify input file names for Load and Mesh Data.

10.2.1 Example 1

Example 1 is to show the pressure load generation along the surfaces of elements 1 and 2 as schematically shown in Figure 10.7. Triangular pressure loads are acting on the right surfaces of elements 1 and 2. Top and rear surfaces of element 2 are subjected to the uniformly distributed pressures of 0.5 and 1.0, respectively. Three different load time histories, as shown in Figure 10.8, are considered.

Mesh Data contains information for nodal coordinates and element indexes. Mesh3D.Mes represents Mesh Data graphically shown in Figure 10.9 along with listing in Table 10.3. Load Data contains information for loads to be generated. Load3D.Dat in Table 10.4, has been prepared according to LOAD-3D User's Manual.

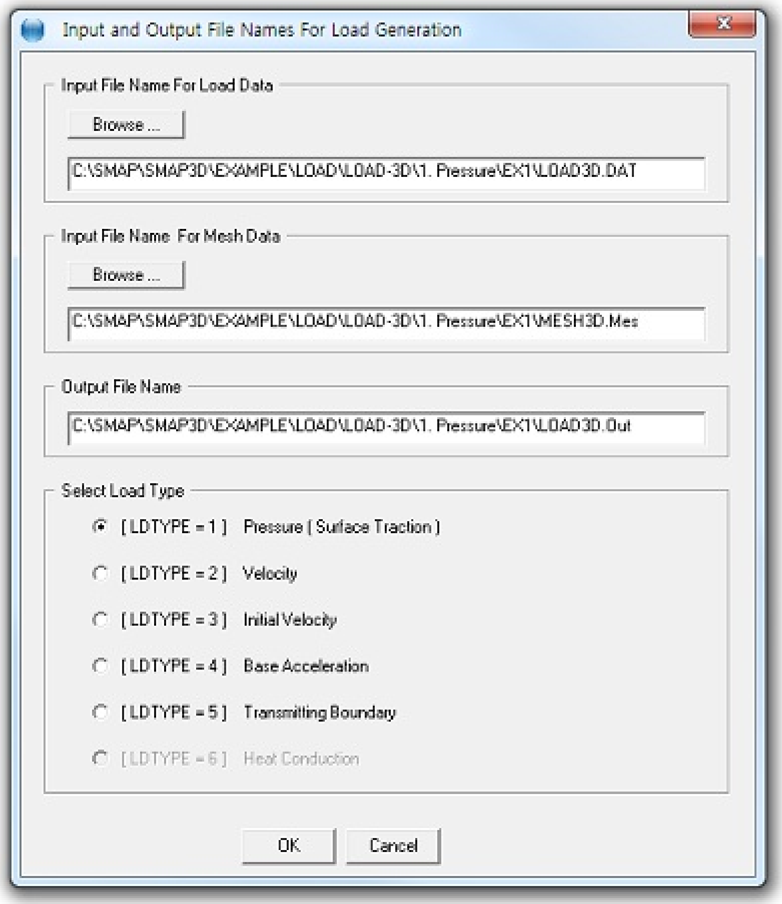
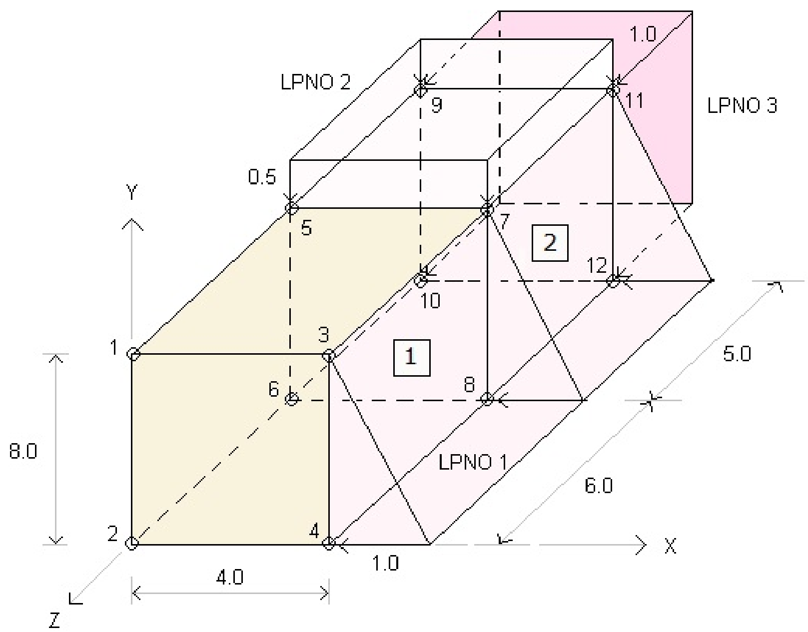


Figure 10.6 Load generation dialog



Loading Surface 1 (LSNO = 1) consists of nodes: 12, 9, 10, 11

Loading Surface 2 (LSNO = 2) consists of nodes: 11, 12, 3, 4, 7, 8

Loading Surface 3 (LSNO = 3) consists of nodes: 5, 7, 11, 9

Pressure Function 1 (LPNO = 1): $P_x = -1.0 + 0.125\ Y$

Pressure Function 2 (LPNO = 2): $P_Y = -0.5$

Pressure Function 3 (LPNO = 3): $P_n = 1.0$

Figure 10.7 Schematic view of pressure loads for Example 1

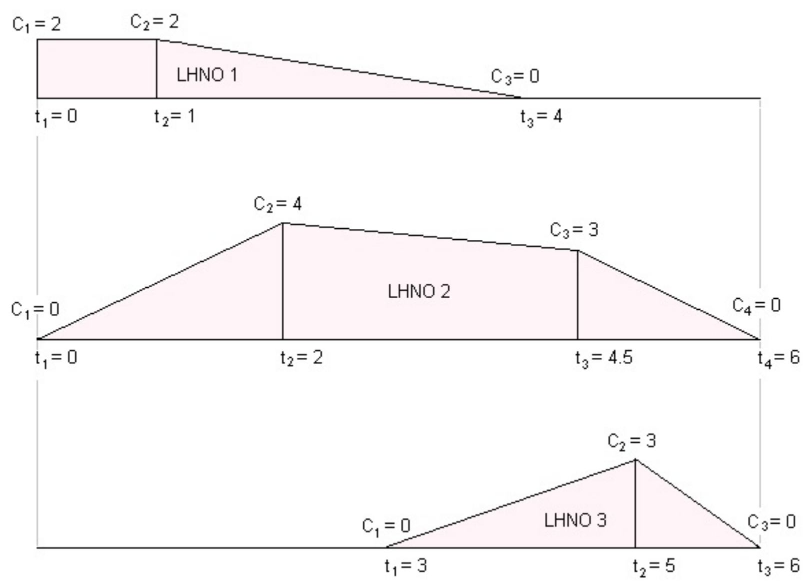


Figure 10.8 Load time histories for Example 1

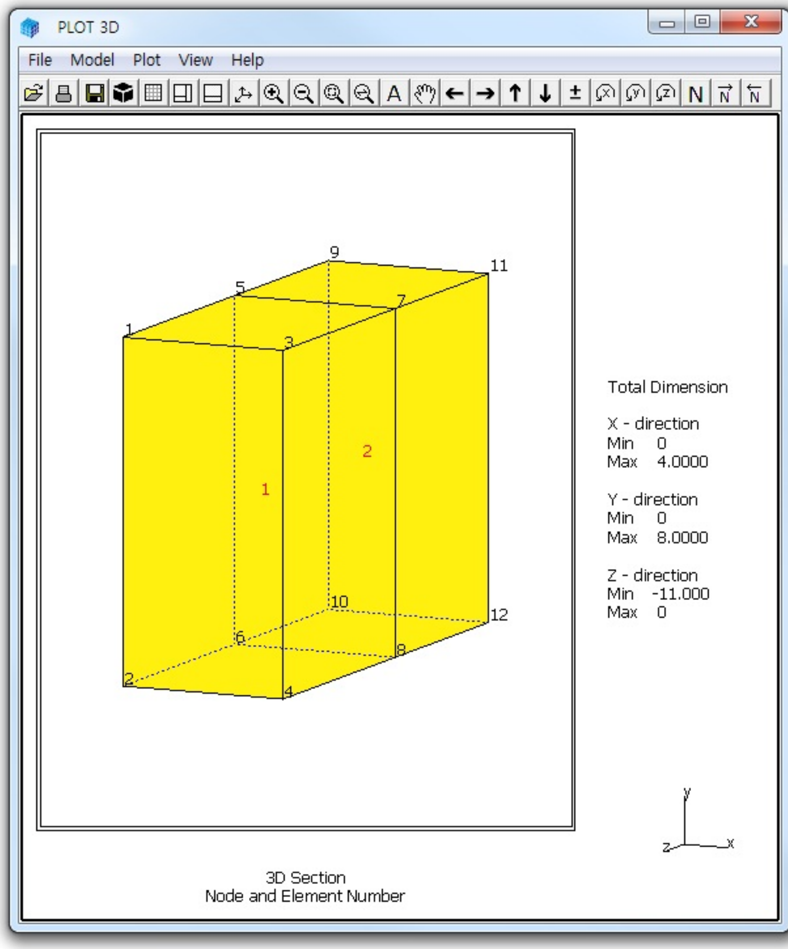


Figure 10.9 Finite element mesh for Example 1

Table 10.4 Listing of mesh input file Mesh3D.Mes for Example 1

```
3D Section
NUMNP    NCONT    NBEAM  NTRUS
  12       2      0       0
NODAL      BOUNDARY CONDITIONS      &            COORDINATES
NODE  ISX   ISY   ISZ   IFX   IFY   IFZ   IRX   IRY   IRZ XC    YC    ZC
  1    1     0     1     1     1     1     1     1     1  0.0   8.0   0.0
  2    0     0     1     1     1     1     1     1     0  0.0   0.0   0.0
  3    0     0     1     1     1     1     1     1     0  4.0   8.0   0.0
  4    0     0     1     1     1     1     1     1     0  4.0   0.0   0.0
  5    0     0     1     1     1     1     1     1     0  0.0   8.0  -6.0
  6    0     0     1     1     1     1     1     1     0  0.0   0.0  -6.0
  7    0     0     1     1     1     1     1     1     0  4.0   8.0  -6.0
  8    0     0     1     1     1     1     1     1     0  4.0   0.0  -6.0
  9    0     0     1     1     1     1     1     1     0  0.0   8.0  -11.0
 10    0     0     1     1     1     1     1     1     0  0.0   0.0  -11.0
 11    0     0     1     1     1     1     1     1     0  4.0   8.0  -11.0
 12    0     0     1     1     1     1     1     1     0  4.0   0.0  -11.0
             ELEMENT  INDEX
NEL    I1    I2    I3    I4    I5    I6    I7    I8   MATC  KS  KF  IR  IS  IT  TBJWL
  1     3     1     2     4     7     5     6     8     4   0   1   2   2   2.  0E+0
  2     7     5     6     8    11     9    10    12     4   0   1   2   2   2.  0E+0
```

Table 10.5 Listing of load input file Load3D.Dat for Example 1

```
*
* LOAD-3D  INPUT
*
* CARD 1.1
* TITLE
  EXAMPLE 1  LOAD-3D Pressure [LDTYPE = 1]
* ========================================
* CARD 2.1
* NUMLS
   3
* ----------------------------------------
* CARD 2.2.1
* LSNO
   1
* CARD 2.2.2
* NUMNODE
   4
* CARD 2.2.3
* LISTING OF NODES
  12, 9, 10, 11
* ----------------------------------------
* CARD 2.2.1
* LSNO
   2
* CARD 2.2.2
* NUMNODE
   6
* CARD 2.2.3
* LISTING OF NODES
  11, 12, 3, 4, 7, 8
* ----------------------------------------
* CARD 2.2.1
* LSNO
   3
* CARD 2.2.2
* NUMNODE
   4
* CARD 2.2.3
* LISTING OF NODES
  5, 7, 11, 9
```

```
* =========================================
* CARD 3.1
* NUMLF
   3
* -----------------------------------------
* CARD 3.2.1
* LFNO    LPTYPE
   1      0
* CARD 3.2.2
* A-X0    A-XX    A-XY    A-XZ
  -1.,    0.0,    0.125,  0.0
* CARD 3.2.3
* A-Y0    A-YX    A-YY    A-YZ
  0.0,    0.0,    0.0,    0.0
* CARD 3.2.4
* A-Z0    A-ZX    A-ZY    A-ZZ
  0.0,    0.0,    0.0,    0.0
* CARD 3.2.5
* A-N0    A-NX    A-NY    A-NZ
  0.0,    0.0,    0.0,    0.0
* -----------------------------------------
* CARD 3.2.1
* LFNO    LPTYPE
   2      0
* CARD 3.2.2
* A-X0    A-XX    A-XY    A-XZ
  0.0,    0.0,    0.0,    0.0
* CARD 3.2.3
* A-Y0    A-YX    A-YY    A-YZ
 -0.5,    0.0,    0.0,    0.0
* CARD 3.2.4
* A-Z0    A-ZX    A-ZY    A-ZZ
  0.0,    0.0,    0.0,    0.0
* CARD 3.2.5
* A-N0    A-NX    A-NY    A-NZ
  0.0,    0.0,    0.0,    0.0
* -----------------------------------------
* CARD 3.2.1
* LFNO    LPTYPE
   3      1
```

```
* CARD 3.2.2
* A-X0   A-XX   A-XY   A-XZ
  0.0,   0.0,   0.0,   0.0
* CARD 3.2.3
* A-Y0   A-YX   A-YY   A-YZ
  0.0,   0.0,   0.0,   0.0
* CARD 3.2.4
* A-Z0   A-ZX   A-ZY   A-ZZ
  0.0,   0.0,   0.0,   0.0
* CARD 3.2.5
* A-N0   A-NX   A-NY   A-NZ
  1.0,   0.0,   0.0,   0.0
* =========================================
* CARD 4.1
* NUMLH
   3
* -----------------------------------------
* CARD 4.2.1
* LHNO
   1
* CARD 4.2.2
* NUMTP
   3
* CARD 4.2.3
* T1   T2   T3
  0.0  1.0  4.0
* CARD 4.2.4
* C1   C2   C3
  2.0  2.0  0.0
* -----------------------------------------
* CARD 4.2.1
* LHNO
   2
* CARD 4.2.2
* NUMTP
   4
* CARD 4.2.3
* T1   T2   T3   T4
  0.0  2.0  4.5  6.0
* CARD 4.2.4
* C1   C2   C3   C4
  0.0  4.0  3.0  0.0
```

```
* -----------------------------------------
* CARD 4.2.1
* LHNO
   3
* CARD 4.2.2
* NUMTP
   3
* CARD 4.2.3
* T1    T2    T3
  3.0   5.0   6.0
* CARD 4.2.4
* C1    C2    C3
  0.0   3.0   0.0
* =========================================
* CARD 5.1
* LSNO  LFNO  LHNO
   1,    3,    1
   2,    1,    3
   3,    2,    2
   0,    0,    0
* END OF INPUT DATA
```

The output file, Load3D.Out listed in Table 10.6, contains generated concentrated nodal forces and load time histories. Figure 10.10 shows time history curves for each load history number. The format of the generated load output is compatible to the format of Card Group 9 in SMAP-3D main input.

Generated load vectors for concentrated forces can be plotted graphically. Refer to the step by step procedure in the file Running LOAD-3D.pdf.

Table 10.6 Listing of load output file Load3D.Out for Example 1

```
* CARD 9.2.1
* NUMLP
   14
* LOAD HISTORY NO:  1
* CARD 9.2.2
* NODE   IDOF   LHNO   CINT
    9      3      1    .80000E+01
   10      3      1    .80000E+01
   11      3      1    .80000E+01
   12      3      1    .80000E+01
* LOAD HISTORY NO:  2
* CARD 9.2.2
* NODE   IDOF   LHNO   CINT
    5      2      2   -.25000E+01
    7      2      2   -.25000E+01
    9      2      2   -.25000E+01
   11      2      2   -.25000E+01
* LOAD HISTORY NO:  3
* CARD 9.2.2
* NODE   IDOF   LHNO   CINT
    3      1      3   -.40000E+01
    4      1      3   -.80000E+01
    7      1      3   -.73333E+01
    8      1      3   -.14667E+02
   11      1      3   -.33333E+01
   12      1      3   -.66667E+01
* END OF LOAD HISTORY
* CARD 9.2.3.1
* NTFUN  NUMLH
    0      3
```

```
* CARD 9.2.3.2
* NUMTP  NTYPE   DTXX
    8      1    .00000E+00
* CARD 9.2.3.3
* LISTING OF TIME POINTS
  .00000E+00  .10000E+01  .20000E+01  .30000E+01  .40000E+01
  .45000E+01  .50000E+01  .60000E+01
* CARD 9.2.3.4
* LISTING OF LOAD FOR HISTORY NO:  1
  .20000E+01  .20000E+01  .13333E+01  .66667E+00 -.59605E-07
  .00000E+00  .00000E+00  .00000E+00
* CARD 9.2.3.4
* LISTING OF LOAD FOR HISTORY NO:  2
  .00000E+00  .20000E+01  .40000E+01  .36000E+01  .32000E+01
  .30000E+01  .20000E+01  .00000E+00
* CARD 9.2.3.4
* LISTING OF LOAD FOR HISTORY NO:  3
  .00000E+00  .00000E+00  .00000E+00  .00000E+00  .15000E+01
  .22500E+01  .30000E+01  .00000E+00
* END OF CONCENTRATED LOAD DATA
```

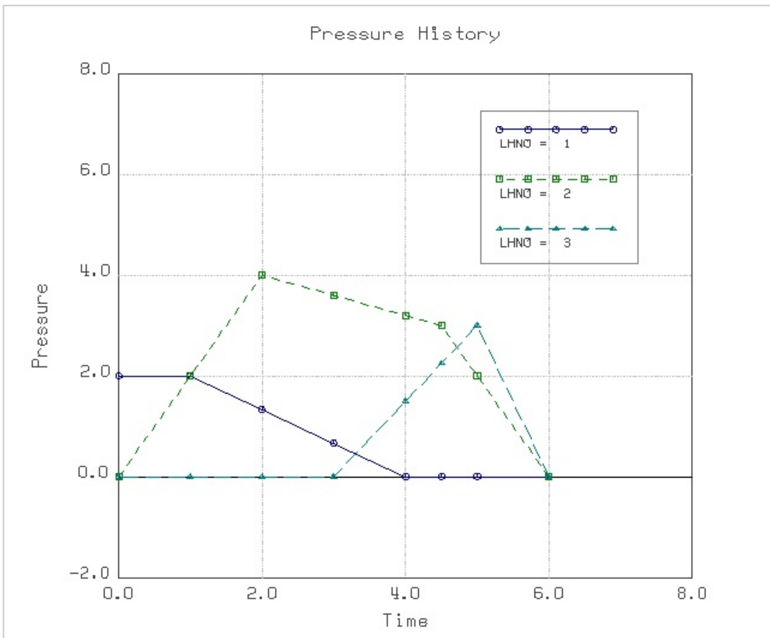


Figure 10.10 Generated load time histories for Example 1

XY Graph
Example Problem

XY Graph is a two-dimensional graph consisting of lines connecting each pair of data points, which can be plotted by PLOT XY or EXCEL. XY Graph User's Manual describes all the basic functions associated with XY graph creation and modifications.

Two example problems are presented:

1. New Graph

   Shows step by step procedure to create and modify XY graph.
   Main actions:
- Access XY graph
- Edit initial Draft XY
- Modify XY graph by Edit dialog
- Open XY graph on Excel Spreadsheet

2. SMAP Result

   Plots SMAP results specified in Card Group 12 in SMAP Post File.
   Main actions:
- Execute SMAP-3D example
- Access SMAP result
- Access PLOT XY in Plot menu
- Modify XY graph by Edit dialog
- Open XY graph on Excel Spreadsheet

11.1 New Graph

The main objective of this first example is to show the step by step procedure to create and modify XY graph.

This example consists of the following main actions:

- Access XY graph
- Edit initial Draft XY
- Modify XY graph by Edit dialog
- Open XY graph on Excel Spreadsheet

Step 1: Access XY Graph (New)

Access XY Graph by selecting following items in SMAP (Figure 11.1):
Plot → XY → PLOT XY → New

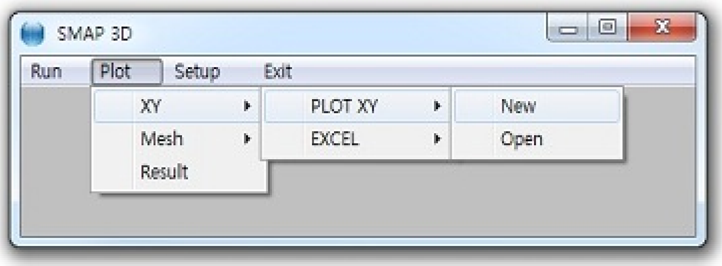


Figure 11.1 Accessing XY graph (New)

Step 2: Edit Initial Draft XY

Once selected, initial default file XY.dat will be opened by Notepad as listed in Table 11.1.

Edit the first plot in this default file as listed in Table 11.2.
And then save and exit.

Modified graph will be displayed on PLOT XY drawing board as shown in Figure 11.2.

Table 11.1 Draft XY Data (Initial Default File XY.dat)

```
Plot No. 1
Sub Title 1
XLabel-1
YLabel-1
 0        10
 100      20
 .000000E+00  .123456E+06
Curve 1
Legend
10,       20
90,       30
 .000000E+00  .123456E+06
Curve 2
Legend
 .000000E+00  .987654E+06
Plot No. 2
Sub Title 2
XLabel-2
YLabel-2
0         100
1000      200
 .000000E+00  .123456E+06
Curve 1
Legend
100       200
900       300
.000000E+00  .123456E+06
Curve 2
Legend
.000000E+00  .987654E+06
Plot No. 3
Sub Title 3
XLabel-3
YLabel-3
0          100
1000       200
.000000E+00  .123456E+06
Curve 1
Legend
200,      200
900,      300
.000000E+00  .123456E+06
Curve 2
Legend
.000000E+00  .987654E
```

Table 11.2 Modified Draft XY Data (File XY.dat)

```
Example 1
Stress History
Time (Sec)
Stress (MPa)
 0          10
 100        20
 .000000E+00   .123456E+06
Vertical
Stress
0           20
100         30
 .000000E+00   .123456E+06
Horizontal
Stress
 .000000E+00   .987654E+06
Plot No. 2
Sub Title 2
XLabel-2
YLabel-2
0           100
1000        200
 .000000E+00   .123456E+06
Curve 1
Legend
100         200
900         300
.000000E+00   .123456E+06
Curve 2
Legend
.000000E+00   .987654E+06
Plot No. 3
Sub Title 3
XLabel-3
YLabel-3
0           100
1000        200
.000000E+00   .123456E+06
Curve 1
Legend
200,        200
900,        300
.000000E+00   .123456E+06
Curve 2
Legend
.000000E+00   .987654E
```

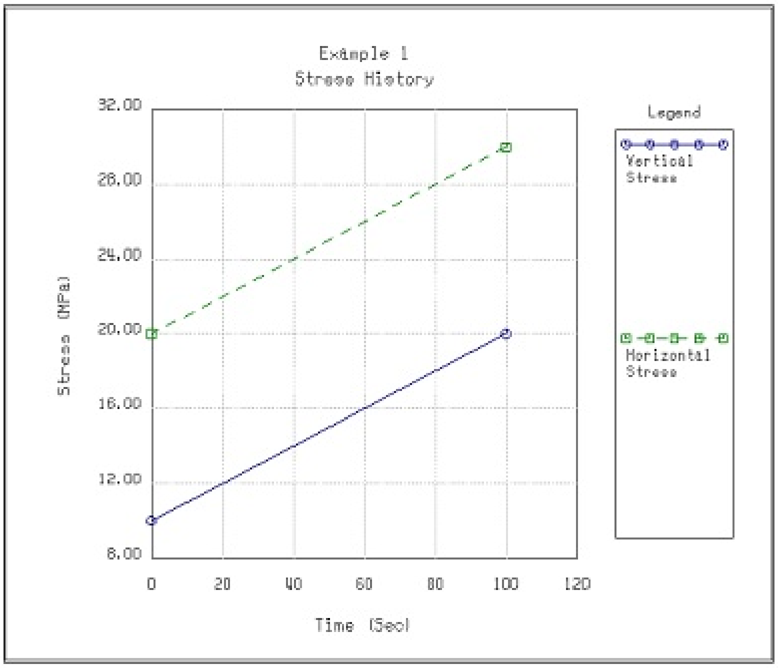


Figure 11.2 Modified graph on PLOT XY

Step 3: Modify XY Graph by Edit Dialog

Access Edit dialog by clicking the Edit menu in PLOT XY (Figure 11.3):

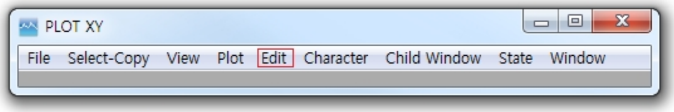


Figure 11.3 Edit menu in PLOT XY

Edit dialog will be displayed as shown in Figure 11.4.

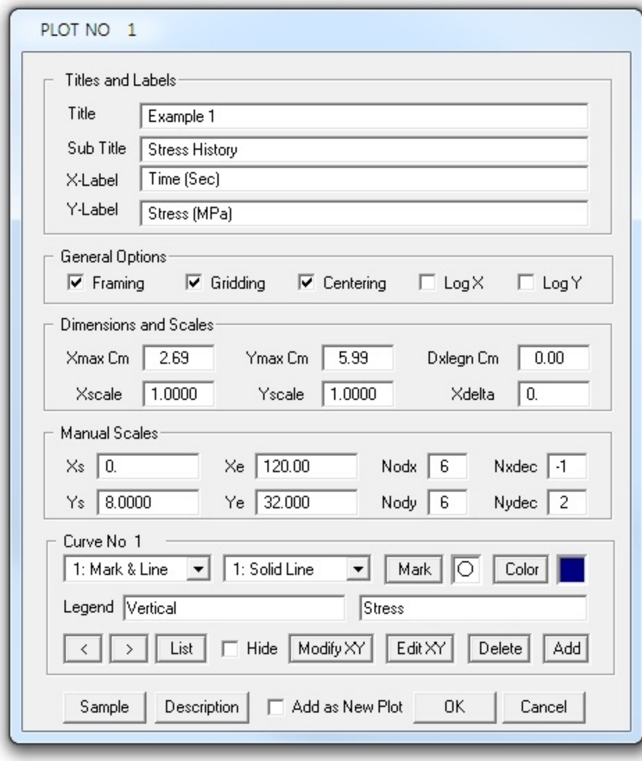


Figure 11.4 Edit dialog

There are many different options available for changing view of XY graphas described in detail in Section 12.3 in XY Graph User's Manual.

Here, change the color of the first curve into Red.
Click Color button and select Red from Color Palette dialog.
Then Red color will be updated for first curve as shown in Figure 11.5.

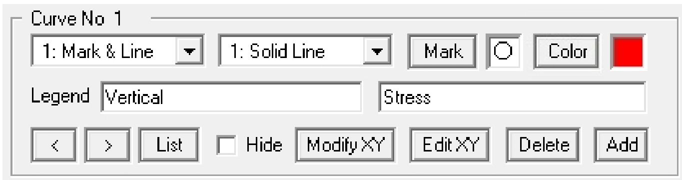


Figure 11.5 Updated red color in Edit dialog

Click OK button in Edit dialog.
Then updated plot will be displayed on PLOT XY as in Figure 11.6.

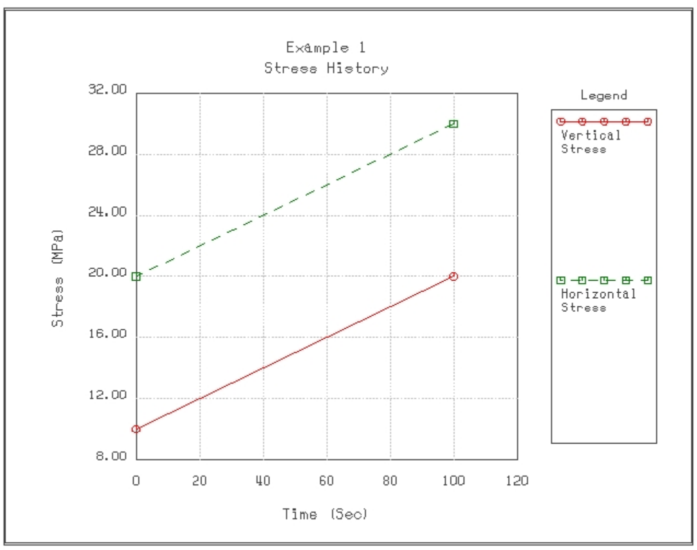


Figure 11.6 Updated first curve on PLOT XY

Step 4: Open XY Graph on Excel Spreadsheet

Access XY Graph by selecting following items in SMAP (Figure 11.7):
Plot → XY → EXCEL → Open

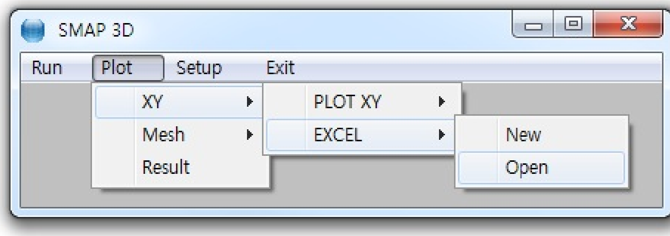


Figure 11.7 Accessing XY graph on Excel (Open)

Open XY.dat in the current working directory.
XY graph will be displayed on Excel Spreadsheet as in Figure 11.8.

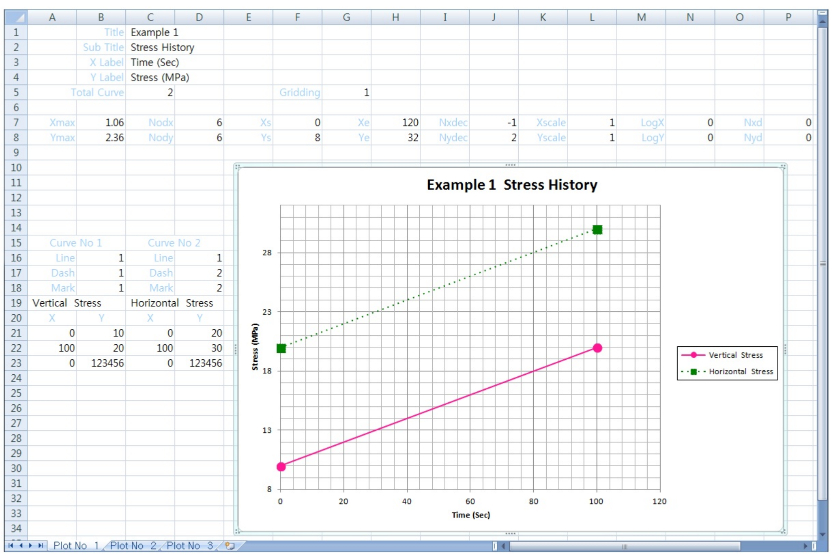


Figure 11.8 XY graph on Excel spreadsheet

11.2 SMAP Result

The main objective of this second example is to show the step by step procedure to plot SMAP results specified in Card Group 12 in SMAP Post File. This example involves SMAP-3D Example Problem 5 (Laminated Beam with Slip Interface).

This example consists of the following main actions:

- Execute SMAP-3D example
- Access SMAP result
- Access PLOT XY in Plot menu
- Modify XY graph by Edit dialog
- Open XY graph on Excel Spreadsheet

Step 1: Execute SMAP-3D Example

Execute SMAP-3D by selecting the following menu items in SMAP (Figure 11.9): Run → Smap → Execute

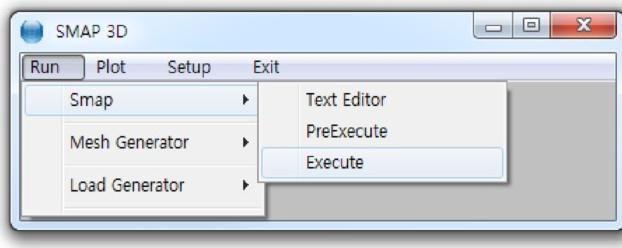


Figure 11.9 Execute SMAP-3D example problem

Note that SMAP-3D Example Problem 5 includes XY graph specified in Card Group 12 in SMAP Post File Vp5.Pos as listed in Table 11.3

Step 2: Access SMAP Result

Access SMAP Result by selecting the following menu items in SMAP : Plot → Result

Table 11.3 SMAP-3D post file (File Vp5.Pos)

```
* Card 11.1
* NPTYPE IHOR IVER
  0      0    0
* P L O T - X Y
* Card 12.1
* IPTYPE
  2
* Card 12.3.1
* IPLOT
  1
* Card 12.3.2
* NODE
  1
* LIST1,  LIST2,  ...
  4
* Card 12.3.4
* NDPQ
  1
* Card 12.3.5
* KX       KY
  1,       3
* Card 12.3.6
* TMFAC   SND    SNV   SNA   NC  ANGLE
  0.018  -100    1     1     0   0
* Card 12.3.7
* TITLE / X-LABEL / Y-LABEL
  Laminated Beam
  Applied Load (t)
  Displacement (Cm)
* Card 12.1
* IPTYPE
  0
* End of Data
```

Step 3: Access PLOT XY in Plot Menu

Select PLOT XY in Plot Menu dialog in Figure 11.10.

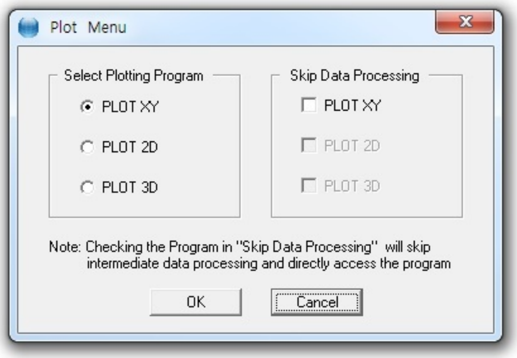


Figure 11.10 Plot menu dialog

Select PLOT XY in Select Plotting Program dialog in Figure 11.11. Click OK button.

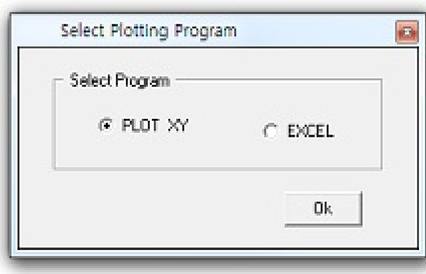


Figure 11.11 Select plotting program dialog

Step 4: Modify XY Graph by Edit Dialog

Once XY graph is displayed on PLOT XY, access Edit dialog by clicking the Edit menu in PLOT XY as shown in Figure 11.12

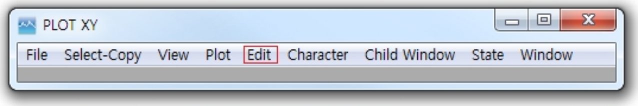


Figure 11.12 Edit menu in PLOT XY

Modify Edit dialog as shown in Figure 11.13.
The main modification is to plot the XY graph in log scales.
Click OK button in Edit dialog.

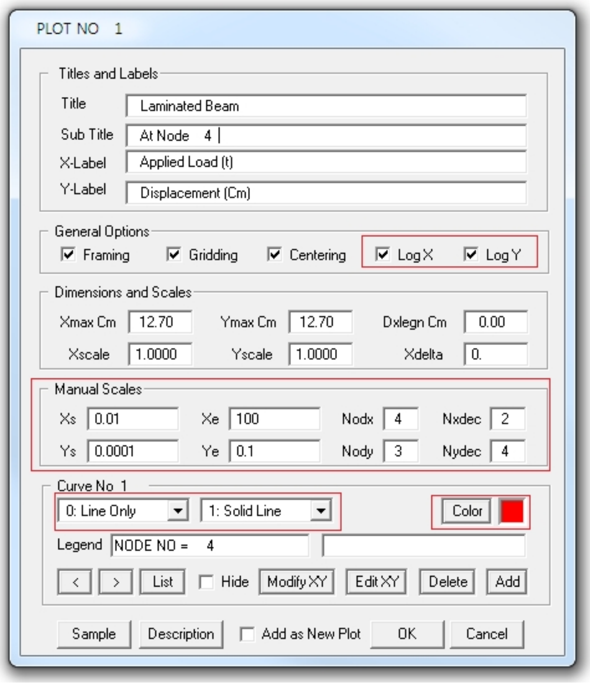


Figure 11.13 Edit dialog

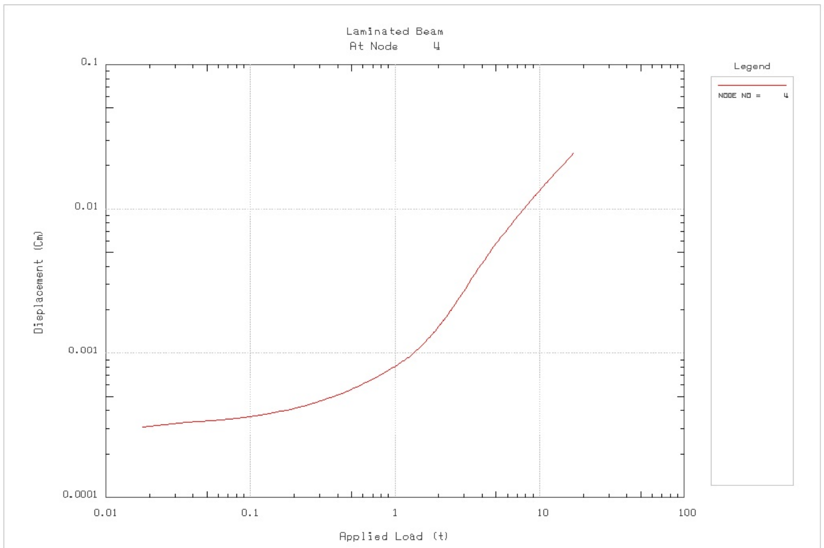


Figure 11.14 XY graph in log scales on PLOT XY

Refer to more samples in the following directory:
C:\Smap \Smap3D \Example \XY_Graph \PLOT XY Graph Sample.pdf

Step 5: Open XY Graph on Excel Spreadsheet

Access XY Graph by selecting following items in SMAP (Figure 11.15):
Plot → XY → EXCEL → Open

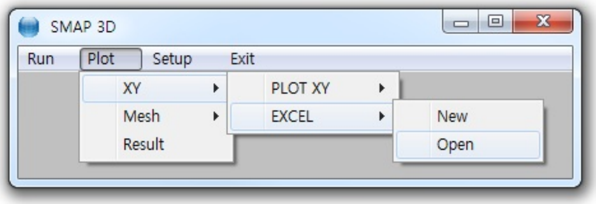


Figure 11.15 Accessing XY graph on Excel (Open)

Open PlotXy.dat in the current working directory.
XY graph will be displayed on Excel Spreadsheet as in Figure 11.16.

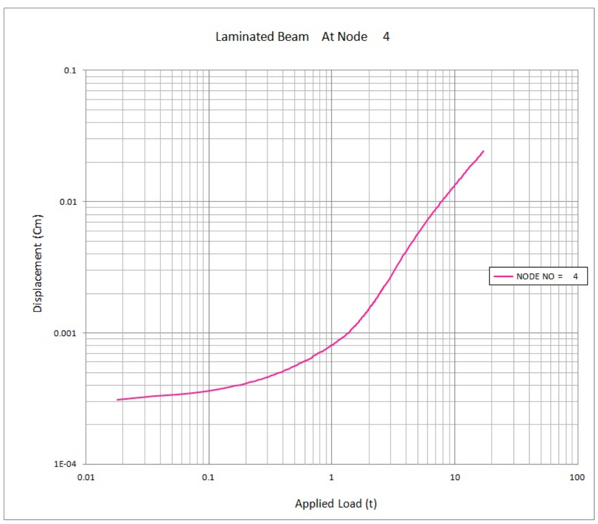


Figure 11.16 XY graph on Excel spreadsheet

Refer to more samples in the following directory:
C:\Smap \Smap3D \Example \XY_Graph \Excel XY Graph Sample.pdf

Go to Edit > Preferences > Page Display > Uncheck Enhance Thin Lines

SMAP ® - 3D

Structure Medium Analysis Program

3-D Static, Consolidation and Dynamic
Analysis for Dry, Saturated and
Partially Saturated Soils
and Rock Mass

Theory

Printed in the United States of America.

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Contents

Introduction

1.1 Introduction

SMAP-3D, which is an upgraded follow-on to the original MPDAP (Multi-Phase Dynamic Analysis Program), is a three-dimensional finite element computer program which has been continuously improved based on theoretical and experimental works since 1982. The program has been used to study fundamental mechanics of saturated porous medium. The program can be a powerful tool for the geomechanical analysis since it can solve static, consolidation and dynamic problems in dry, partially saturated or fully saturated soils and porous rock mass. The program considers material, geometric and boundary condition nonlinearities. Next two sections describe theoretical backgrounds of program SMAP-3D.

Section 2 describes theoretical formulations of nonlinear two-phase medium. Nonlinear compressibility equations are derived in detail for grains, saturated pore water, and partially saturated pore water. Field equations representing fundamental mechanics of two-phase medium are presented.

These field equations include effective stress law, constitutive equation for skeleton deformation, continuity equation of pore fluid, equation of motion for the bulk mixture, and equation of motion for pore fluid. Then, these field equations are discretized in space and expressed in incremental forms. Finally, global equilibrium equations are derived by principle of virtual work and then linearized to be solved by linear equation solver.

Section 3 describes constitutive relations of various nonlinear material models available in the program SMAP-3D. These nonlinear models include; Generalized Hoek and Brown Model, Single Hardening Elasto-Plastic Model, JWL High Explosive Model, Modified Cam Clay Model with Creep, Engineering Model, Joint Model, and Generalized Decoupled Hyperbolic Model.

To see the validation of the computational algorithms of the computer program SMAP-3D, refer to SMAP-3D Example Problems.

Finite Element Formulation of Nonlinear Two-Phase Medium

2.1 Introduction

Biot introduced fundamental analytical work describing the behavior of saturated porous media in a series of papers extending over many years (e.g. 1956, 1962a and 1962b). Other investigators have applied Biot's analytic results using techniques which approximate his equations with varying degrees of accuracy and sophistication (e.g. Ghaboussi and Wilson 1972, Mengi and McNiven, 1977). Theoretical formulations incorporated in the code SMAP-3D are the extension of Biot's two-phase theory to nonlinear region. These nonlinear two-phase theories have been developed over a decade under the sponsorship of Air Force Office of Scientific Research (e.g. Kim and Blouin 1984, Kim et al. 1986, 1987 and 1988).

In this section, the fundamental equations implemented in the code SMAP-3D are described. First the individual material components compressibility models are described in Subsection 2.2 for the solid grain, Subsection 2.3 for the pore water and Subsection 2.4 for the partially saturated water. Nonlinear material models of the skeleton are presented in Section 3. Field equations described in Subsection 2.5 include effective stress law, constitutive equation for skeleton deformation, continuity equation of pore fluid flow, equation of motion for the bulk mixture and equation of motion for pore fluid.

These field equations are described in terms of nodal values and expressed in incremental form in Subsection 2.6. Finally, global equilibrium equations for the two-phase medium are formulated in Subsection 2.7 and linearized to be solved by linear equation solver in Subsection 2.8.

2.2 Grain Model

To model the nonlinear response of the solid grains to both the applied pore pressure and effective stress, analytic expressions for the deformation of solids at high pressure are employed. High pressure data for many rocks and minerals show a linear relationship between loading wave velocity and particle velocity (e.g. Allen, 1967).
The loading wave velocity can be expressed as:

$$c_L = c_o + S\, v_p \tag{2.1}$$

where:

c_L	=	Loading wave velocity
c_o	=	The initial wave velocity at relatively low pressure
v_p	=	Peak particle velocity
S	=	Experimentally determined constant relating c_L to v_p (generally equal to about 1.5 for most dense rocks and minerals)

Conservation of mass and momentum on either side of the wave front yields the familiar relationships:

$$\sigma_p = \rho_o\, c_L\, v_p \tag{2.2}$$

$$M = \rho_o\, c_L^2 \tag{2.3}$$

where:

σ_p = Peak axial stress

ρ_o = Initial material density

M = Constrained secant modulus = σ_p / ε_p

ε_p = Peak axial strain corresponding to the peak stress σ_p

Substitution of Equation 2.1 into 2.2 gives:

$$\sigma_p = \rho_o c_o v_p + \rho_o S v_p^2 \tag{2.4}$$

and solving for peak particle velocity as a function of peak stress yields

$$v_p = \frac{f(\sigma_p)}{2 \rho_o S} \tag{2.5}$$

where

$$f(\sigma_p) = (\rho_o^2 c_o^2 + 4 \rho_o S \sigma_p)^{1/2} - \rho_o c_o \tag{2.6}$$

Substitution of Equation 2.1, 2.5, and 2.6 into Equation 2.3 gives:

$$M = F(\sigma_p) = \rho_o c_o^2 + c_o f(\sigma) + \frac{f^2(\sigma)}{4 \rho_o} \tag{2.7}$$

The tangent constrained modulus, M_t, used in the numerical model is defined as the slope of the stress strain curve by:

$$M_t = \frac{d\sigma}{d\varepsilon} \tag{2.8}$$

From Equation 2.7 and the definition of constrained modulus, M:

$$\varepsilon_p = \frac{\sigma_p}{F(\sigma_p)} \tag{2.9}$$

Differentiating Equation 2.9 with respect to σ_p and inverting gives the tangent constrained modulus as

$$M_t = \frac{F^2(\sigma_p)}{F(\sigma_p) - \sigma_p F'(\sigma_p)} \tag{2.10}$$

Differentiating Equations 2.6 and 2.7 with respect to σ_p yields:

$$F'(\sigma_p) = c_o f'(\sigma_p) + \frac{f(\sigma_p) f'(\sigma_p)}{2 \rho_o} \tag{2.11}$$

and

$$f'(\sigma_p) = \frac{2 \rho_o S}{(\rho_o^2 c_o^2 + 4 \rho_o S \sigma_p)^{1/2}} \tag{2.12}$$

Hence, Equations 2.5 through 2.12 can be used to define high pressure constrained stress strain and modulus relationships for the solid grains.

For two phase, coupled calculations, the volumetric relationships for the solid grains should be specified in terms of the bulk modulus, K_g, rather than in terms of the constrained modulus. At high pressures, the shear strength of the grain materials becomes insignificant compared to the applied stress and the materials tend to behave like fluids. At these pressures, the tangent bulk modulus equals the tangent constrained modulus with Poisson's ratio equal to 0.5. Beneath some threshold pressure, p_b, Poisson's ratio begins to decrease from 0.5 at p_b to an initial value of Poisson's ratio, v_o, at a low value of mean stress. We have used a simple relationship to approximate the influence of mean stress on Poisson's ratio for the solid grains:

$$K_g = g(p) M_t \tag{2.13}$$

The ratio of the bulk modulus to the tangent constrained modulus, g(p) at pressures less than p_b is given by:

$$g(p) = \frac{2}{3}\frac{(1-2v_o)}{(1-v_o)}\frac{p}{p_b} + \frac{(1+v_o)}{3(1-v_o)} \quad (2.14)$$

For pressures greater than p_b;

$$g(p) = 1 \quad (2.15)$$

Poisson's ratio can be computed as a function of the modulus ratio at a given pressure as:

$$v = \frac{3\,g(p) - 1}{1 + 3\,g(p)} \quad (2.16)$$

2.3 Pore Water Model

The model for the nonlinear, elastic compressibility of the pore water is derived from an equation of state reported by Ahrens (1988) and attributed to Bakanova, et. al. (1976). This equation relates the shock velocity in water to the peak particle velocity. In the lower pressure regime, a quadratic relation is used while a linear relation is used in the higher pressure regime. The transition point between the two regimes is defined in terms of a peak particle velocity at the transition, V_{pt}. Bakanova's equations can be expressed as:

$$\begin{aligned} &v_s \le v_{pt}: \\ &c = c_1 + S_1\,v_p + S_2\,v_p^2 \end{aligned} \quad (2.17)$$

$$\begin{aligned} &v_s > v_{pt}: \\ &c = c_2 + S_3\,v_p \end{aligned} \quad (2.18)$$

where:

c	=	Shock propagation velocity in the fluid
v_p	=	Peak fluid particle velocity
c_1, S_1, S_2	=	Constants used to fit data below the transition
c_2, S_3	=	Constants used to fit data above the transition

Equation 2.18 can also be expressed in terms of the shock velocity at the transition point, c_t. Substituting v_{pt} into Equation 2.18 yields:

$$c_2 = c_t - S_3 v_{pt} \tag{2.19}$$

Substituting 2.19 into 2.18 produces this expression for the shock velocity above the transition:

$$\begin{aligned} &v_s > v_{pt} : \\ &c = c_t + S_3 (v_p - v_{pt}) \end{aligned} \tag{2.20}$$

where:

c_t	=	Shock velocity at the transition
v_{pt}	=	Peak particle velocity at the transition (Model constant)

At the transition point, the shock velocity from Equations 2.17 and 2.20 should be equal to preserve continuity. Setting Equations 2.17 and 2.20 equal at $v_p = v_{pt}$ gives:

$$c_t = c_1 + S_1 v_{pt} + S_2 v_{pt}^2 \tag{2.21}$$

thereby defining c_t in terms of the model constants. Equations 2.17, 2.20, and 2.21 (with the constants c_1, S_1, S_2, and S_3) define the shock velocity as a function of peak particle velocity.

To derive a bulk modulus for water as a function of pressure, we first need an expression for peak particle velocity as a function of pressure.

Conservation of mass and momentum on either side of the wave front yields the familiar relationship from shock physics:

$$\pi_p = \rho_o \, c \, v_p \tag{2.22}$$

where:

π_p = Pore fluid pressure
ρ_o = Mass density of fluid

Substitution of Equation 2.17 into 2.22 yields an expression for the transition fluid pressure (π_{pt}):

$$\pi_{pt} = \rho_o \, v_{pt} \, (c_1 + S_1 \, v_{pt} + S_2 \, v_{pt}^2) \tag{2.23}$$

For water, the transition pressure is greater than 30,000 MPa. Below the transition pressure, substitution of Equation 2.17 into 2.22 will give:

$$v_p^3 + \frac{S_1}{S_2} v_p^2 + \frac{c_1}{S_2} v_p - \frac{\pi_p}{\rho_o \, S_2} = 0 \tag{2.24}$$

This cubic equation can be solved to yield an expression for v_p as a function of fluid pressure below the transition pressure π_{pt} :
where:

$$v_p = m \cos\left[\frac{1}{3} \cos^{-1}\left(\frac{3\,\beta}{\alpha \, m}\right) + \frac{4\,\pi}{3}\right] - \frac{S_1}{3\,S_2} \tag{2.25}$$

where

$$\alpha = \frac{c_1}{S_2} - \frac{1}{3}\left(\frac{S_1}{S_2}\right)^2 \tag{2.26}$$

$$\beta = \frac{-\pi_p}{\rho_o S_2} - \frac{1}{3}\left(\frac{S_1}{S_2}\right)\left(\frac{c_1}{S_2}\right) + \frac{2}{27}\left(\frac{S_1}{S_2}\right)^3 \tag{2.27}$$

$$m = 2\sqrt{\frac{-\alpha}{3}} \tag{2.28}$$

Above the transition pressure, substitution of Equation 2.20 into 2.22 yields a quadratic equation:

$$v_p^2 + \left(\frac{c_t - S_3 v_{pt}}{S_3}\right) v_p - \frac{\pi_p}{\rho_o S_3} = 0 \tag{2.29}$$

Solving this equation for v_p as a function of fluid pressure gives v_p for pressures above the transition pressure π_{pt} :

$$v_p = -\left(\frac{c_t - S_3 v_{pt}}{2S_3}\right) + \left[\left(\frac{c_t - S_3 v_{pt}}{2S_3}\right)^2 + \frac{\pi_p}{\rho_o S_3}\right]^{\frac{1}{2}} \tag{2.30}$$

The elastic bulk modulus of water (K_w) is defined as:

$$K_w = \frac{d\pi_p}{d\varepsilon_v} = \frac{d\pi_p \;/\; dv_p}{d\varepsilon_v \;/\; dv_p} \tag{2.31}$$

where ε_v is the volume strain corresponding to the pressure π_p.
Taking the derivative of Equation 2.22:

$$\frac{d\pi_p}{dv_p} = \rho_o \,(c' \, v_p + c) \tag{2.32}$$

The volume strain is given by:

$$\varepsilon_v = \frac{v_p}{c} \tag{2.33}$$

and taking the derivative yields:

$$\frac{d\varepsilon_v}{dv_p} = \frac{c - v_p\, c'}{c^2} \tag{2.34}$$

Substitution of Equations 2.32 and 2.34 into 2.31 gives an expression for the bulk modulus in terms of the shock and peak particle velocities:

$$K_w = \frac{\rho_o\, c^2\, (c + v_p\, c')}{c - v_p\, c'} \tag{2.35}$$

The derivatives of the shock velocity with respect to the peak particle velocity are given by:

$$\pi_p \le \pi_{pt} : \qquad c' = S_1 + 2\, S_2\, v_p \tag{2.36}$$

$$\pi_p > \pi_{pt} : \qquad c' = S_3 \tag{2.37}$$

The material constant values for this model are given in Table 2.1 for fresh water and sea water. The fresh water values are from Bakanova, et. al. (1976) as reported by Ahrens (1988). Parameters for sea water were fit to compressibility data described by Kim, et. al. (1986) and attributed to Britt (1985).

2.4 Partially Saturated Pore Water Model

When rock or soil is unsaturated, compression of the pore water and solid grains is nearly insignificant when compared with the compression of pore air. Under these conditions, material behavior is governed mostly by the skeleton model. With sufficient compression, the pore air gets squeezed out and the material becomes saturated. Rischbieter, et. al. (1977) demonstrated that even a minute amount of entrapped air drastically alters the pore pressure response in multiphase porous materials. To simulate this behavior, the pore fluid model is modified to account for the compressibility of pore air and converges to a saturated condition. Note that this model is invoked only when the initial saturation is less than 100%.

The compressibility of the air-water mixture, C_{aw}, is defined as:

$$C_{aw} = \frac{d\varepsilon_{v,aw}}{d\pi_p} \tag{2.38}$$

where π_p is the fluid pressure. The volumetric strain in the air-water mixture, $\varepsilon_{v,aw}$, is the sum of volume strain in the air and water. Using the definition of the initial saturation, it can be shown that:

$$\varepsilon_{v,aw} = (1 - S_o)\, \varepsilon_{v,a} + S_o\, \varepsilon_{v,w} \tag{2.39}$$

where:

$\varepsilon_{v,aw}$ = Volume strain of air-water mixture
$\varepsilon_{v,a}$ = Volume strain of air bubbles
$\varepsilon_{v,w}$ = Volume strain of water (from Equation 2.33)
S_o = Initial saturation

From Equations 2.38 and 2.39 we can get an expression for the compressibility of the air-water mixture:

$$C_{aw} = (1 - S_o)\ C_a + S_o\ C_w \tag{2.40}$$

Since the compressibility is the inverse of the bulk modulus, Equation 2.40 can be expressed as:

$$\frac{1}{K_{aw}} = \frac{1 - s_o}{K_a} + \frac{s_o}{K_w} \tag{2.41}$$

where:

K_{aw} = Bulk modulus of air-water mixture
K_a = Equivalent bulk modulus of air bubbles in the fluid
K_w = Bulk modulus of water (from Equation 2.35)

The volume strain and the equivalent bulk modulus of the air bubbles in the pore fluid are derived here using the adiabatic ideal gas law (γ-law). The model has been shown to be applicable when the degree of pore water saturation is above approximately 85% where the pore air is thought to exist as small bubbles within the fluid (occluded state).

The model is derived from the adiabatic ideal gas law:

$$\pi_a \cdot V_a^{\gamma} = \pi_{ao} \cdot V_{ao}^{\gamma} \tag{2.42}$$

where

π_{ao} Initial air pressure (absolute pressure)
π_a Current air pressure (absolute pressure)
V_{ao} Initial air volume
V_a Current air volume
γ Ratio of heat capacity (c_p/c_v)

The volume strain of air can be defined in terms of engineering strain:

$$\varepsilon_{v,a} = 1 - \left(\frac{V_a}{V_{ao}}\right) \tag{2.43}$$

Substituting Equation 2.42 into Equation 2.43, we can express the volume strain of air bubble in terms of air pressure:

$$\varepsilon_{v,a} = 1 - \left[\frac{\pi_{ao}}{\pi_a}\right]^{\frac{1}{\gamma}} \tag{2.44}$$

Neglecting the influence of surface tension,

$$\pi_a = \pi + p_a \tag{2.45}$$

where

π	Current pore water pressure (gage pressure)
P_a	Reference atmospheric pressure

Substitution of Equation 2.45 into Equation 2.44 yields

$$\varepsilon_{v,\,a} = 1 - \left(\frac{\pi_a}{\pi + P_a}\right)^{\frac{1}{\gamma}} \tag{2.46}$$

Tangent bulk modulus of air bubbles can be defined as

$$K_a = \frac{d\pi_a}{d\varepsilon_{v,a}} \tag{2.47}$$

Differentiating Equation 2.46 with respect to π,

$$\frac{d\varepsilon_{v,a}}{d\pi_a} = \frac{1}{\gamma \cdot \pi_{ao}}\left(\frac{\pi_{ao}}{\pi + P_a}\right)^{\left(1 + \frac{1}{\gamma}\right)} \tag{2.48}$$

Substitution of Equation 2.48 into Equation 2.47 yields

$$K_a = \gamma \cdot \pi_{ao} \left[\frac{\pi + P_a}{\pi_{ao}} \right]^{\left(1 + \frac{1}{\gamma}\right)} \tag{2.49}$$

Equations 2.35 and 2.49, when substituted into Equation 2.41, define the compressibility of the pore air-water mixture. The model does not employ an explicit expression for the saturation point, where the air bubbles no longer exist. However, Equation 2.49 implies that the stiffness of the pore air increases with the pressure. As the pressure increases, the contribution of the air to the net compressibility of the mixture becomes insignificant when compared to the compressibility of the water. This, in essence, results in fully saturated behavior but with a smooth model transition during collapse of the air bubbles. An example pressure-volume curve for water with an initial air content of 5% is shown in Figure 2.1. Notice that the mixture becomes pressure saturated at a volume strain of about 5%.

Table 2.1 Fluid compressibility model constants
(See Section 2.3 for definitions of constants)

Parameter	Unit	Fresh Water	Sea Water
ρ_0	kg /m^3	1002.8	1026
c_1	m/s	1500	1522
S_1	-	2.00	1.97
S_2	s/m	-1.07×10^{-4}	-0.898×10^{-4}
S_3	-	1.144	1.123
v_{pt}	m/s	4000	4573
c_t	m/s	7788	8653
π_{pt}	MP$_a$	31,240	40,600

2.5 Field Equations

Effective Stress Law

Terzaghi's effective stress equation is fundamental to the development of the fully coupled model. It relates the total applied stress, σ, to the pore pressure, π, and the effective stress, σ', according to

$$\sigma_{ij} = \sigma'_{ij} + \delta_{ij}\,\pi \tag{2.50}$$

where

σ_{ij} = Total stress
σ'_{ij} = Effective stress
δ_{ij} = Kronecker's delta
δ_{ij} = 0 if $i \neq j$
δ_{ij} = 1 if $i = j$

Constitutive Equation for Skeleton Deformation

The deformation of the porous skeleton is related to the applied effective stress and the pore pressure acting on the solid grains. The stress-strain relationship is given by

$$\{d\sigma'\} = [D^{ep}]\left(\{d\varepsilon\} - \frac{1}{3\,K_g}\{1\}\,d\pi\right) \tag{2.51}$$

The last term in Equation 2.51 is the strain in the skeleton resulting from compression of the solid grains by the pore pressure.

Continuity Equation of Pore Fluid Flow

The continuity equation for pore fluid flow is derived from mass conservation relationships. The volumetric strain of the pore fluid, ε_f, is given by

$$d\varepsilon_f = -\frac{d\rho_f}{\rho_f} = C_f\,d\pi \tag{2.52}$$

where

C_f = Pore fluid compressibility

π = Pore fluid pressure

The volume strain of the solid grains, ε_g , is give by

$$d\varepsilon_g = -\frac{d\rho_g}{\rho_g} = C_g\, d\pi + \frac{C_g}{1-n}\, dp' \tag{2.53}$$

where

C_g = Bulk compressibility of solid grains

p' = Effective mean pressure

The dry density, ρ_d , is given by

$$\rho_d = \frac{m_g}{V_t} = (1-n)\,\rho_g \tag{2.54}$$

where m_g is the mass of the solid grains in skeleton volume V_t . The change in dry density is given by

$$d\rho_d = -\rho_d\, d\varepsilon_v \tag{2.55}$$

where ε_v is the volumetric strain of the skeleton. Differentiating Equation 2.54 with respect to n and ρ_g gives

$$d\rho_d = (1-n)\, d\rho_g - \rho_g\, dn \tag{2.56}$$

Equating 2.55 and 2.56 yields

$$d\varepsilon_v = \frac{dn}{1-n} - \frac{d\rho_g}{\rho_g} \tag{2.57}$$

Conservation of mass for the pore fluid within a specified initial volume of saturated porous material is given by

$$n \, \rho_f \, V_t \;=\; \bar{n} \; \bar{\rho}_f \; \bar{V}_t \tag{2.58}$$

where as illustrated in Figure 2.2, the terms to the left of the equal sign represent the fluid mass under the initial conditions and the terms to the right represent the same fluid mass under deformed conditions.

Equation 2.58 may be expressed in infinitesimal incremental form as

$$n \, \rho_f \, V_t \;=\; (n + dn) \, (\rho_f + d\rho_f) \, (1 + d\epsilon_F) \, V_t \tag{2.59}$$

where

ϵ_F = Volumetric diffusion of pore fluid as depicted in Figure 2.2

Solving Equation 2.59 for $d\epsilon_F$ and discarding second order terms yields

$$d\epsilon_F \;=\; -\frac{dn}{n} - \frac{d\rho_f}{\rho_f} \tag{2.60}$$

Equation 2.60 is combined with Equation 2.57 by elimination of dn to yield

$$(1-n)\, d\epsilon_v + n\, d\epsilon_F + (1-n)\frac{d\rho_g}{\rho_g} + n\frac{d\rho_f}{\rho_f} \;=\; 0 \tag{2.61}$$

Combining Equations 2.52 and 2.53 with 2.61 gives

$$n\,(d\epsilon_F - d\epsilon_v) + d\epsilon_v - \frac{1}{K_m}\, d\pi - c_g \, dp' \;=\; 0 \tag{2.62}$$

where K_m is the bulk modulus of the solid/fluid mixture which is expressed by

$$K_m \;=\; \frac{1}{n\, C_f + (1-n)\, C_g} \tag{2.63}$$

The change in effective mean pressure is given by

$$dp' = K_s \left(d\epsilon_v - C_g \, d\pi\right) \tag{2.64}$$

Substituting Equation 2.64 into 2.62 gives

$$n \left(d\epsilon_F - d\epsilon_v\right) + \left(1 - C_g K_s\right) d\epsilon_v + \left(C_g^2 K_s - \frac{1}{K_m}\right) d\pi = 0 \tag{2.65}$$

or

$$\begin{aligned} n \left(d\epsilon_F - d\epsilon_v\right) &= \left(\alpha - \frac{C_g^2}{9} \{1\}^T [D^{ep}] \{1\}\right) d\pi \\ &\quad - \left(\{1\}^T - \frac{C_g}{3} \{1\}^T [D^{ep}]\right) \{d\epsilon\} \end{aligned} \tag{2.66}$$

Equation 2.66 can be expressed in the following convenient form:

$$d\pi = \bar{m}_2 \cdot d\epsilon_v + \bar{m} \cdot n \left(d\epsilon_F - d\epsilon_v\right) \tag{2.67}$$

where

$$\bar{m} = \frac{1}{\left[\dfrac{1}{K_m} - \dfrac{K_s^{ep}}{K_g^2}\right]} \tag{2.68}$$

$$\bar{m}_2 = \left[1 - \frac{K_s^{ep}}{K_g}\right] \cdot \bar{m} \tag{2.69}$$

Equation of Motion for the Bulk Mixture

The differential equation of motion governing the bulk mixture is expressed by equating the stress gradient to the inertial resistance as

$$\sigma_{ij,\,j} \;=\; (1-n)\,\rho_s\,\ddot{u}_i \;+\; n\,\rho_f\,\ddot{U}_i \qquad (2.70)$$

$\sigma_{ij,j}$ is the total stress gradient applied to an infinitesimal element of saturated material at some given time. $\sigma_{ij,j}$ is expressed in tensor and represents the stress gradient in each of three mutually perpendicular coordinates (e.g. see Mendleson, 1968). For instance, in the x direction,

$$\sigma_{xj,\,j} \;=\; \frac{\partial\sigma_{xx}}{\partial x} + \frac{\partial\sigma_{xy}}{\partial y} + \frac{\partial\sigma_{xz}}{\partial z} \;=\; (1-n)\,\rho_s\,\ddot{u}_x \;+\; n\,\rho_f\,\ddot{U}_x \qquad (2.71)$$

The term (1-n) ρ_s is the mass of the soil skeleton per unit volume of saturated material, where n is the porosity and ρ_s is the mass density of the solid grains. u_i is the displacement of the skeleton in the i direction and $\ddot{u}$ is the acceleration of the skeleton in the i direction. The term n ρ_f is the mass of pore fluid per unit volume of saturated material where ρ_f is the mass density of the pore fluid. U_i is the absolute displacement of the pore fluid in the i direction.

The bulk mass density of the saturated material, ρ, is given by

$$\rho \;=\; (1-n)\,\rho_s \;+\; n\,\rho_f \qquad (2.72)$$

Substitution of the value for (1-n) ρ_s from Equation 2.72 into Equation 2.70 gives

$$\sigma_{ij,\,j} \;=\; (\rho \;-\; n\,\rho_f)\,\ddot{u}_i \;+\; n\,\rho_f\,\ddot{U}_i \qquad (2.73)$$

A term w_i is introduced which is the apparent fluid displacement in the i direction relative to the soil skeleton and is given by

$$w_i \;=\; n\,(U_i \;-\; u_i) \qquad (2.74)$$

In seepage problems, w_i , is referred to as the discharge displacement. It describes the discharge of fluid through a soil mass of unit area. The discharge velocity, or apparent relative velocity, $\dot{w}_I$, between the soil particles and pore water is the velocity of water in a discharge duct of unit area needed to maintain the actual relative velocity in the porous soil of the same unit area. The actual relative velocity between the skeleton and the pore water is given by $\dot{w}_i / n$. Finally, $\ddot{w}_i$ is the apparent relative acceleration between the soil skeleton and pore water given by

$$\ddot{w}_i = n\,(\ddot{U}_i - \ddot{u}_i) \tag{2.75}$$

Equation 2.73 can be expressed in terms of the apparent relative fluid acceleration as simply

$$\sigma_{ij,\,j} = \rho\,\ddot{u}_i + \rho_f\,\ddot{w}_i \tag{2.76}$$

Equation of Motion for Pore Fluid

The finite element code SMAP-3D is capable of calculating the flow of pore fluid between elements. The flow of fluid with respect to the skeleton is controlled by Forchheimer's permeability model as described in a series of reports to the Air Force Office of Scientific Research (Kim, et. al., 1986, 1987, 1988; Blouin et. al., 1990, 1991). The Forchheimer model, as described by Kim, et. al. (1988) can be expressed as:

$$\pi_{,i} = \frac{\rho_f\,g}{k}\,\dot{w}_i + \frac{\beta_f}{k^{1/2}}\,\dot{w}_i^2 + \rho_f\,\ddot{U}_i \tag{2.77}$$

where

$\pi_{,i}$	=	Pore pressure gradient
g	=	Acceleration of gravity
ρ_f	=	Mass density of pore fluid
k	=	Darcy's coefficient of permeability (function of skeleton and fluid properties)
β_f	=	Ward's turbulent flow coefficient (function of skeleton and fluid properties)
$\dot{w}$	=	Apparent flow velocity relative to the skeleton
$\ddot{U}$	=	Absolute acceleration of pore fluid

The first term in Equation 2.77 is simply Darcy's law while the velocity squared term was apparently first proposed by Forchheimer (1901). The first two terms represent the frictional component of the pressure gradient while the last term accounts for the inertial effect of fluid flow.

Equation 2.77 can also be written in the form:

$$\pi_{,i} = \frac{\rho_f\, g}{k'}\, \dot{w}_i + \rho_f\, \ddot{U}_i \qquad (2.78)$$

where k' represents an equivalent permeability coefficient given by:

$$k' = \frac{k}{1 + \dfrac{\beta_f}{\rho_f\, g}\sqrt{k}\,|\dot{w}_i|} \qquad (2.79)$$

Hence, the flow of pore fluid in the soil skeleton is governed by Equations 2.78 and 2.79 and the flow coefficients k and β_f which can be determined from laboratory test data. Using the Equation 2.75, Equation 2.78 can be expressed in terms of skeleton and apparent relative fluid motions given by

$$\pi_{,i} = \frac{\rho_f}{n}\, \ddot{w}_i + \rho_f\, \ddot{u}_i + k'\, \dot{w}_i \qquad (2.80)$$

More recently, Blouin and his coworkers (1991) have proposed a refined expression of the Forchheimer model that, while remaining equivalent to Equation 2.77, clarifies the distinction between fluid-related and skeleton-related permeability properties. This new expression is:

$$\pi_{,i} = \frac{\mu}{\alpha}\, \dot{w}_i + \frac{\rho_f}{\beta}\, \dot{w}_i^2 + \rho_f\, \ddot{U}_i \qquad (2.81)$$

where

μ	=	Dynamic viscosity of the fluid
α, β	=	Flow coefficients that are properties of the porous skeleton only

The conversions between the different permeability parameters are obtained from equating the corresponding terms of Equations 2.77 and 2.81 to obtain:

$$k = \frac{\alpha \rho_f g}{\mu} \tag{2.82}$$

$$\beta_f = \frac{k^{\frac{1}{2}} \rho_f}{\beta} \tag{2.83}$$

While the parameters α and β and Equation 2.81 form the preferred expression for the permeability model, the current implementation of the model in our numerical codes follow the form of Equations 2.77 through 2.80.

2.6 Spatial Discretization and Incremental Relationships of Field Variables

Within each element, field variables can be discretized into element nodal values.

$$\begin{aligned} \{\Delta w\} &= [N]\,\{\Delta \underline{w}\}_e \\ \{\Delta \varepsilon\} &= [B]\,\{\Delta \underline{u}\}_e \\ \Delta w_{i,i} &= \{1\}^T\,[B]\,\{\Delta \underline{w}\}_e \\ \{\Delta u\} &= [N]\,\{\Delta \underline{u}\}_e \end{aligned} \tag{2.84}$$

Stress vector at time step n can be expressed as:

$$\{\sigma_n\} = \{\sigma_{n-1}\} + \{\Delta\sigma'\} + \{1\}\,\Delta\pi \tag{2.85}$$

Combining Equations 2.50, 2.51, 2.67 and 2.84 yields

$$\begin{aligned}\{\Delta\sigma\} = {} & ([D^{ep}]\,[B] + \bar{m}_1\,\{1\}\,\{1\}^T\,[B])\,\{\Delta\underline{u}\} \\ & + \bar{m}_2\,\{1\}\,\{1\}^T\,[B]\,\{\Delta\underline{w}\}\end{aligned} \tag{2.86}$$

where

$$\bar{m}_1 = \left[1 - \frac{K_s^{ep}}{K_g}\right]^2 \cdot \bar{m} \tag{2.87}$$

Equation 2.67 can be rewritten in incremental form as:

$$\Delta\pi = \bar{m}_2 \cdot \Delta u_{i,i} + \bar{m} \cdot \Delta w_{i,i} \tag{2.88}$$

2.7 Global Equilibrium Equations

Two global equilibrium equations are derived, first in terms of field variables and then discretized using nodal variables.

The first equates the total internal stresses plus the inertia forces to the applied boundary traction. Letting the solid skeleton movement be the virtual displacement, δu, the following global equilibrium equation for the bulk mixture is established:

$$\begin{aligned}\int_v \{\delta\varepsilon\}^T\,\{\sigma\}\,dv = {} & \int_s \{\delta u\}^T\,\{T\}\,ds - \int_v \{\delta u\}^T\,\rho\,\{\ddot{u}\}\,dv \\ & - \int_v \{\delta u\}^T\,\rho_f\,\{\ddot{w}\}\,dv\end{aligned} \tag{2.89}$$

where

$\delta\varepsilon$ is the virtual strain corresponding to virtual displacement δu.

The second equates the applied pore pressure on the boundary to the internal pore pressure plus the flow resistance force plus the inertia force on the pore fluid. Taking the apparent relative fluid movement as the virtual displacement, δw, the internal virtual work done by the pore pressure should be equal to the external virtual work. That is,

$$\int_v (\delta w_{i,i})^T \pi \cdot dv = \int_s \{\delta w\}^T \hat{n}\, ds - \int_v \{\delta w\}^T [r] \cdot \{\dot{w}\}\, dv - \int_v \{\delta w\}^T \rho_f \{\ddot{u}\}\, dv - \int_v \{\delta w\}^T \frac{1}{n} \rho_f \{\ddot{w}\}\, dv \qquad (2.90)$$

Replacing the field variables in Equation 2.89 and 2.90 by the discretized nodal variables using Equation 2.84 gives the following global equilibrium equation at time step n:

$$\begin{bmatrix} M_m & M_c \\ M_c^T & M_f \end{bmatrix} \begin{Bmatrix} \ddot{\underline{u}}_n \\ \ddot{\underline{w}}_n \end{Bmatrix} + \begin{bmatrix} 0 & 0 \\ 0 & H \end{bmatrix} \begin{Bmatrix} \dot{\underline{u}}_n \\ \dot{\underline{w}}_n \end{Bmatrix} + \begin{bmatrix} K_t + EE & C \\ C^T & E \end{bmatrix} \begin{Bmatrix} \Delta \underline{u}_n \\ \Delta \underline{w}_n \end{Bmatrix} = \begin{Bmatrix} F_n \\ G_n \end{Bmatrix} - \begin{Bmatrix} R^s_{n-1} + R^f_{n-1} \\ R^f_{n-1} \end{Bmatrix} \qquad (2.91)$$

where

$$M_m = \Sigma \int_v [N]^T \rho\, [N]\, dv$$

$$M_c = \Sigma \int_v [N]^T \rho_f\, [N]\, dv$$

$$M_f = \Sigma \int_v [N]^T \frac{1}{n} \rho_f\, [N]\, dv$$

$$H = \Sigma \int_v [r]\, [N]^T\, [N]\, dv$$

$$K_t = \Sigma \int_v [B]^T\, [D^{ep}]\, [B]\, dv$$

$$EE = \Sigma \int_v \bar{m}_1 [B]^T \{1\} \{1\}^T [B] \, dv$$

$$C = \Sigma \int_v \bar{m}_2 [B]^T \{1\} \{1\}^T [B] \, dv$$

$$F_n = \Sigma \int_s [N]^T \{T\} \, ds + \Sigma \int_v [N]^T \rho \{b\} \, dv$$

$$E = \Sigma \int_v \bar{m} [B]^T \{1\} \{1\}^T [B] \, dv$$

$$G_n = \Sigma \int_s [N]^T \hat{\pi}_n \, ds + \Sigma \int_v [N]^T \rho_f \{b\} \, dv$$

$$R^s_{n-1} = \Sigma \int_v [B]^T \{\sigma'_{n-1}\} \, dv$$

$$R^f_{n-1} = \Sigma \int_v [B]^T \{1\} \pi_{n-1} \, dv$$

$[r]$ = Inverse of permeability matrix

$\{b\}$ = Component of body force vector

Equation 2.91 can be rewritten in the simpler form:

$$[M] \{\ddot{\underline{d}}_n\} + [D] \{\dot{\underline{d}}_n\} + [K] \{\Delta \underline{d}_n\} = \{P_n\} - \{R_{n-1}\} \tag{2.92}$$

2.8 Linearized Global Equilibrium Equations

Introducing a time integration method which incorporates both Newmark's β method and Wilson's θ method, the generalized acceleration vector is expressed as

$$\{ \ddot{\underline{d}}_n \} = C_1 \{ \Delta \underline{d}_n \} + C_2 \{ \dot{\underline{d}}_{n-1} \} + C_3 \{ \ddot{\underline{d}}_{n-1} \} \tag{2.93}$$

where

$$\begin{aligned} C_1 &= \frac{1}{\beta\, \theta^3\, \Delta t^2} \\ C_2 &= -\frac{1}{\beta\, \theta^2\, \Delta t} \\ C_3 &= 1 - \frac{1}{2\, \beta\, \theta} \end{aligned} \tag{2.94}$$

and the generalized velocity vector is expressed as

$$\{ \dot{\underline{d}}_n \} = B_1 \{ \Delta \underline{d}_n \} + B_2 \{ \dot{\underline{d}}_{n-1} \} + B_3 \{ \ddot{\underline{d}}_{n-1} \} \tag{2.95}$$

where

$$\begin{aligned} B_1 &= \frac{\gamma}{\beta\, \theta^3\, \Delta t} \\ B_2 &= 1 - \frac{\gamma}{\beta\, \theta^2} \\ B_3 &= \Delta t - \frac{\gamma}{2\, \beta\, \theta}\, \Delta t \end{aligned} \tag{2.96}$$

Substituting Equations 2.93 and 2.95 into Equation 2.92 and rearranging, we can obtain the following linearized global equilibrium equations which can be solved simultaneously at each step:

$$[\tilde{K}]\,\{\Delta \underline{d}_n\} = \{\tilde{P}_n\} \tag{2.97}$$

where the generalized stiffness matrix is given by

$$[\tilde{K}] = C_1\,[M] + B_1\,[D] + [K] \tag{2.98}$$

and the generalized force vector is given by

$$\begin{aligned}\{\tilde{P}_n\} = \{P_n\} - \{R_{n-1}\} &- [M]\,(C_2\,\{\dot{\underline{d}}_{n-1}\} + C_3\,\{\ddot{\underline{d}}_{n-1}\}) \\ &- [D]\,(B_2\,\{\dot{\underline{d}}_{n-1}\} + B_3\,\{\ddot{\underline{d}}_{n-1}\})\end{aligned} \tag{2.99}$$

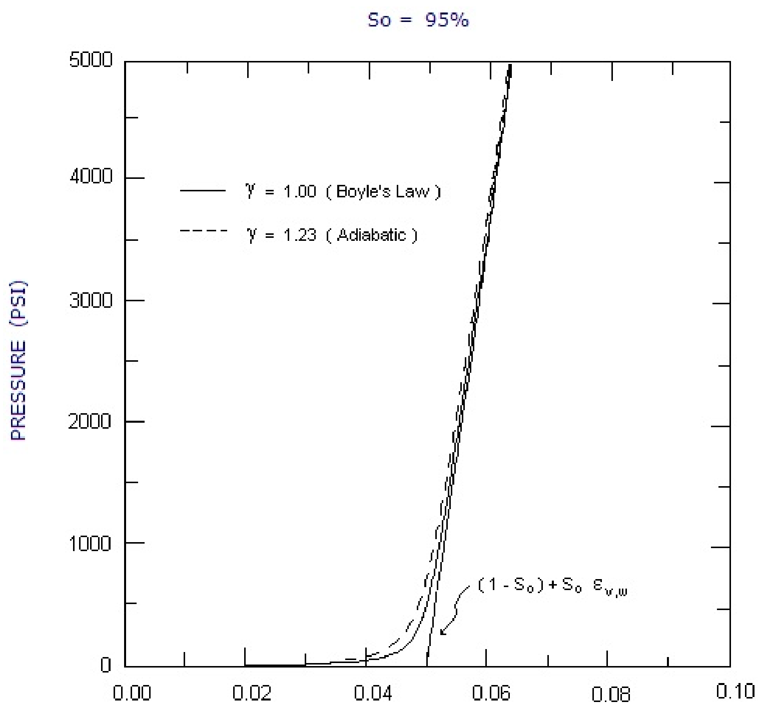


Figure 2.1 Prediction of air-water compressibility

Conservation of Fluid Mass

$$n \ \ \rho_f \, v_t = n' \, \rho_f' \, v_t'$$

v_t	=	Apparent fluid volume before compression
v_t'	=	$(1+ \varepsilon_F)\, v_t$: apparent fluid volume after compression
ε_v	=	Volumetric strain of porous skeleton
ε_F	=	Volumetric diffusion of pore fluid

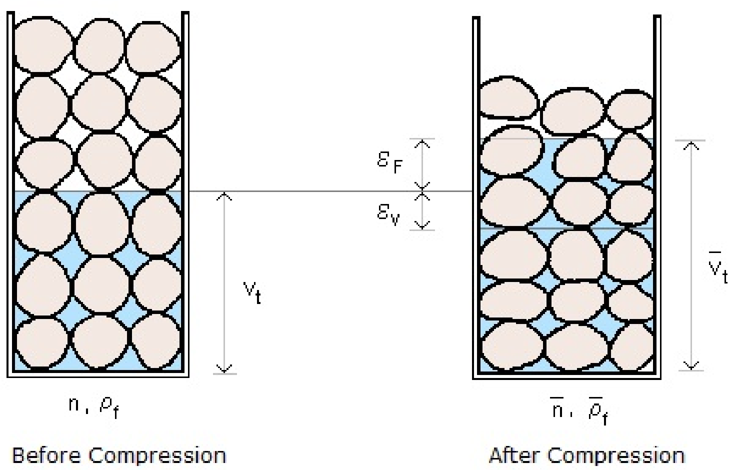


Figure 2.2 Schematic illustration of conservation of pore fluid mass in saturated porous materials

Nonlinear Material Models

3.1 Generalized Hoek and Brown Model

3.1.1 Introduction

Generalized Hoek and Brown Model represents the skeleton constitutive relations of soils or porous materials. In its generalized form, the model includes the empirically based Hoek and Brown failure equation as well as the classical Von Mises, Mohr-Coulomb, and Drucker-Prager failure equations. As one of the useful features, the model can use empirical data base for the strength of in situ rock mass when the in situ strength data are not available.

In this section, the 3-dimensional elasto-plastic matrix is derived for the Generalized Hoek and Brown Model. The model is elastic below the failure surface and perfectly plastic along the failure surface with the volumetric and deviatoric behaviors dependent upon one another once the failure surface is reached.

And the failure equation is expressed in terms of the alternate stress invariant (p, q, and θ) given by

$$p = \frac{1}{3}\sigma_{ii}$$

$$
\begin{aligned}
S_{ij} &= \sigma_{ij} - p \cdot \delta_{ij} \\
J_2 &= \frac{1}{2} S_{ij} S_{ij} \\
J_3 &= \frac{1}{3} S_{ij} S_{jk} S_{ki} \\
q &= \sqrt{3J_2} \\
\theta &= \frac{1}{3} \operatorname{Sin}^{-1}\left(-\frac{27}{2} \frac{J_3}{q^3} \right)
\end{aligned}
\tag{3.1}
$$

where σ_{ij} is the total stress tensor and S_{ij} is the deviatoric stress tensor.

3.1.2 Elastic Stress-Strain Relationship

The incremental elastic constitutive law can be expressed in the following matrix form:

$$\{d\sigma\} = [D^e]\{d\varepsilon^e\} \tag{3.2}$$

where

$\{d\sigma\}$	Stress increment
$[D^e]$	Elastic stress-strain matrix
$\{d\varepsilon^e\}$	Elastic strain increment

3.1.3 Failure Surface

The failure surface is described by the following equation:

$$F(p, q, \theta) = q - ((\alpha + \beta p)^n + \kappa) R(\theta) = 0 \tag{3.3}$$

The expression for R(θ) in Equation 3.3 is given by

$$R(\theta) = \frac{x\,(\sqrt{3}\,\cos\theta + \sin\theta) + (2k-1)\,[(2+\cos2\theta+\sqrt{3}\,\sin2\theta)\,x + 5k^2-4k]^{1/2}}{[\,x\,(2+\cos2\theta+\sqrt{3}\,\sin2\theta) + (1-2k)^2\,]} \tag{3.4}$$

where

$$\left(-\frac{\pi}{6} \le \theta \le \frac{\pi}{6}\right)$$

x = $(1\text{-}k^2)$

k = the ratio of the shear strength in triaxial extension to the shear strength in triaxial compression at the same mean pressure

The function R(θ) describes the shape of the yield surface, as projected in the π plane (octahedral plane). Figure 3.1 and 3.2 show the influence of the parameter k on the shape of the yield surface. k is the ratio of the shear strength in triaxial extension to the shear strength in triaxial compression at the same mean pressure. k is a measure of the influence of the intermediate principal stress on the yield surface and can vary from 0.5 to 1.0. When k is equal to unity, R(θ) is circular, indicating a Drucker-Prager or Von Mises failure model. When k is less than unity, R(θ) is a smooth cornered approximation to the Mohr-Coulomb failure envelope.

The parameter n in Equation 3.3 determines the shape of the yield surface in the p-q plane. For n=0, the shear strength is constant with respect to the mean pressure and the strength envelope reduces to the Von Mises or Tresca yield surface. For n=1/2, the strength envelope represents Hoek and Brown (1982) failure surface. This nonlinear failure model is a multidimensional generalization of the original one-dimensional axisymmetric Hoek and Brown model which is based on extensive laboratory and field data (Kim, Piepenburg and Merkle, 1986).

For n = 1, shear strength is linearly proportional to the mean pressure and the strength envelope in the p-q plane is representative of the Drucker-Prager or Mohr-Coulomb failure surface.

The parameters α, β and κ of Equation 3.3 define the failure envelope in the p-q plane. They can be determined from laboratory tests. Recommended relationships for determining these parameters for Von Mises, Hoek and Brown and Mohr-Coulomb type materials are listed in Table 3.1. The empirical material parameters for n=1/2 are tabulated in Table 3.2 for several different rock types as a function of rock quality. Detailed description of rock quality is shown in Table 3.3.

3.1.4 Flow Rule

A variable dilatancy potential function, G, is defined such as

$$\frac{\partial G}{\partial p} = \left(\frac{\partial F}{\partial p}\right) r$$

$$\frac{\partial G}{\partial q} = \frac{\partial F}{\partial q} \qquad (3.5)$$

$$\frac{\partial G}{\partial \theta} = \frac{\partial F}{\partial \theta}$$

where r is a dilatancy parameter ($0 \le r \le 1$)

r = 0 No plastic volume change
= 1 Associated flow

Thus, in general,

$$\{d\varepsilon^p\} = d\lambda \, \{g\} \qquad (3.6)$$

where

$$\{g\} = \left\{\frac{\partial G}{\partial \sigma}\right\}$$

3.1.5 Consistency Equation

During yielding , the consistency equation forces the stress to move along the failure surface

$$dF = \{a\}^T \{d\sigma\} = 0 \tag{3.7}$$

where

$$\{a\} = \left\{\frac{\partial F}{\partial \sigma}\right\} \tag{3.8}$$

3.1.6 Incremental Elasto-Plastic Constitutive Law

Total strain is defined as the sum of elastic and plastic strains

$$\{d\varepsilon\} = \{d\varepsilon^e\} + \{d\varepsilon^p\} \tag{3.9}$$

Substituting Equation 3.9 into 3.2, we have

$$\{d\sigma\} = [D^e] (\{d\varepsilon\} - \{d\varepsilon^p\}) \tag{3.10}$$

From the flow rule defined in Equation 3.6, we can rewrite Equation 3.10 as

$$\{d\sigma\} = [D^e] \{d\varepsilon\} - d\lambda [D^e] \{g\} \tag{3.11}$$

Substituting Equation 3.11 into 3.7 and solving for dλ, we obtain

$$d\lambda = \frac{\{a\}^T [D^e] \{d\varepsilon\}}{\{a\}^T [D^e] \{g\}} \tag{3.12}$$

Back substituting Equation 3.12 into Equation 3.11, the stress increment is directly related to the total strain increment as follows:

$$\{d\sigma\} = [D^{ep}]\{d\varepsilon\} \tag{3.13}$$

where

$$[D^{ep}] = [D^{e}] - \frac{[D^{e}]\{g\}\{a\}^T[D^{e}]}{\{a\}^T[D^{e}]\{g\}} \tag{3.14}$$

3.1.7 Calculation of {a}

Differentiating the yield function with respect to p, q, and θ, we have

$$\begin{aligned}
\frac{\partial F}{\partial p} &= -n(\alpha+\beta p)^{n-1}\cdot\beta\cdot R(\theta)\\
\frac{\partial F}{\partial q} &= 1\\
\frac{\partial F}{\partial \theta} &= -\{(\alpha+\beta p)^n+\kappa\}\frac{\partial R(\theta)}{\partial \theta}
\end{aligned} \tag{3.15}$$

where

$$\begin{aligned}
\frac{\partial R}{\partial \theta} &= \frac{1}{R_D}\left[\frac{\partial R_N}{\partial \theta} - R(\theta)\frac{\partial R_D}{\partial \theta}\right]\\
R_N &= x(\sqrt{3}\cos\theta+\sin\theta)+(2k-1)\left[(2+\cos 2\theta+\sqrt{3}\sin 2\theta)x+5k^2-4k\right]^{1/2}\\
R_D &= x(2+\cos 2\theta+\sqrt{3}\sin 2\theta)+(1-2k)^2\\
\frac{\partial R_N}{\partial \theta} &= x(\cos\theta-\sqrt{3}\sin\theta)+\frac{x(2k-1)(\sqrt{3}\cos 2\theta-\sin 2\theta)}{\left[x(2+\cos 2\theta+\sqrt{3}\sin 2\theta)+5k^2-4k\right]^{1/2}}\\
\frac{\partial R_D}{\partial \theta} &= 2x(\sqrt{3}\cos 2\theta-\sin 2\theta)
\end{aligned} \tag{3.16}$$

The derivative of the yield function with respect to stress can be written in general 3-dimensional condition as

$$\{a\} = \frac{\partial F}{\partial p}\left\{\frac{\partial p}{\partial \sigma}\right\} + \frac{\partial F}{\partial q}\left\{\frac{\partial q}{\partial \sigma}\right\} + \frac{\partial F}{\partial \theta}\left\{\frac{\partial \theta}{\partial \sigma}\right\} \tag{3.17}$$

where

$$\left\{\frac{\partial p}{\partial \sigma}\right\} = \frac{1}{3} < 1\ 1\ 1\ 0\ 0\ 0 >^T$$

$$\left\{\frac{\partial \theta}{\partial \sigma}\right\} = \frac{9}{2q^3 \cos 3\theta}\left(\frac{3J_3}{q}\left\{\frac{\partial q}{\partial \sigma}\right\} - \left\{\frac{\partial J_3}{\partial \sigma}\right\}\right)$$

$$\left\{\frac{\partial q}{\partial \sigma}\right\} = \frac{3}{2q} < S_x\ \ S_y\ \ S_z\ \ 2\,\sigma_{xy}\ \ 2\,\sigma_{yz}\ \ 2\,\sigma_{xz} >^T$$

$$\left\{\frac{\partial J_3}{\partial \sigma}\right\} = \left\{\begin{array}{l} S_y S_z - \sigma_{yz}^2 + \frac{1}{9}q^2 \\ S_x S_z - \sigma_{xz}^2 + \frac{1}{9}q^2 \\ S_x S_y - \sigma_{xy}^2 + \frac{1}{9}q^2 \\ 2\,(-S_z\sigma_{xy} + \sigma_{yz}\sigma_{xz}) \\ 2\,(-S_x\sigma_{yz} + \sigma_{xz}\sigma_{xy}) \\ 2\,(-S_y\sigma_{xz} + \sigma_{xy}\sigma_{yz}) \end{array}\right\}$$

$$\{\sigma\}^T = < \sigma_x\ \ \sigma_y\ \ \sigma_z\ \ \sigma_{xy}\ \ \sigma_{yz}\ \ \sigma_{xz} >$$

$$\{\varepsilon\}^T = < \varepsilon_x\ \ \varepsilon_y\ \ \varepsilon_z\ \ \gamma_{xy}\ \ \gamma_{yz}\ \ \gamma_{xz} >$$

$$\gamma_{xy} = 2\,\varepsilon_{xy} \qquad \gamma_{yz} = 2\,\varepsilon_{yz} \qquad \gamma_{xz} = 2\,\varepsilon_{xz}$$

Table 3.1 Material Constants in Grenerlized Hoek and Brown Model

	n = 0 Von Mises or Tresca	n = 1/2 Hoek and Brown	n = 1 Mohr-Coulomb or Drucker-Prager
α	N/A	$\left(\frac{m^2}{36} + s\right) \sigma_c^2$	1000
β	N/A	$m\ \sigma_c$	$\frac{6 \sin\phi}{(3 - \sin\phi)}$
κ	$q' - 1$	$\frac{1}{6}\ m\ \sigma_c$	$\frac{3(1 - \sin\phi)}{(3 - \sin\phi)} \sigma_c - 1000$

q' = $\sigma_1 - \sigma_3$
where σ_1 and σ_3 are major and minor pricipal stresses at failure.

σ_c = Unconfined compressive strength

ϕ = Internal friction angle

m,s = Hoek and Brown's material constants as tabulated in Table 3.2.

Table 3.2 Hoek and Brown Material Parameters (m, s)

Rock Type / Rock Quality	Dolomite, Limestone & Marble	Mudstone, Siltstone, Shale and Slate (normal to cleavage)	Sandstone and Quartzite	Andesite, Dolerite & Rhyolite	Amphibolite, Gabbro, Gneiss, Norite and Quartz-Diorite
Intact CSIR rating = 100 NGI rating = 150	m = 7 s = 1	10.0 1.0	15.0 1.0	17.0 1.0	25.0 1.0
Very Good Quality CSIR rating = 85 NGI rating = 100	3.5 0.1	5.0 0.1	7.5 0.1	8.5 0.1	12.5 0.1
Good Quality CSIR rating = 65 NGI rating = 10	0.7 0.004	1.0 0.004	1.5 0.004	1.7 0.004	2.5 0.004
Fair Quality CSIR rating = 44 NGI rating = 1	0.14 0.001	0.20 0.0001	0.3 0.0001	0.34 0.0001	0.5 0.0001
Poor Quality CSIR rating = 23 NGI rating = 0.1	0.04 0.00001	0.05 0.00001	0.08 0.00001	0.09 0.00001	0.13 0.00001
Very Poor Quality CSIR rating = 3 NGI rating = 0.01	0.007 0.0	0.01 0.0	0.015 0.1	0.017 0.0	0.025 0.0

Table 3.3 Description of Rock Quality in Table 3.2

Intact Rock Samples	Laboratory size specimens free from joints
Very Good Quality Rock Mass	Tightly interlocking undisturbed rock with unweathered joints at 1 to 3m
Good Quality Rock Mass	Fresh to slightly weathered rock, slightly disturbed with joints at 1 to 3m
Fair Quality Rock Mass	Several sets of moderately weathered joints spaced at 0.3 to 1m
Poor Quality Rock Mass	Numerous weathered joints at 30 to 500mm with sane gouge. Clean compacted waste rock
Very Poor Quality Rock Mass	Numerous heavily weathered joints spaced < 50m with gouge. Waste rock with fines

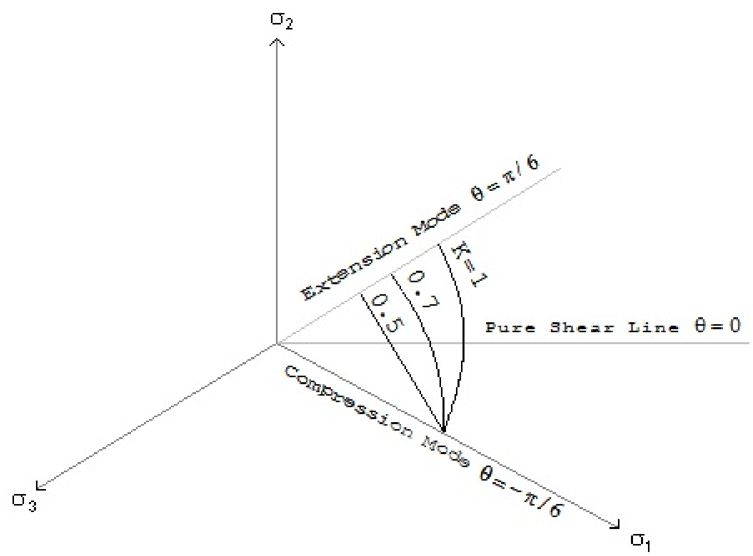


Figure 3.1 Shape of strength envelope, R(θ), on octahedral plane

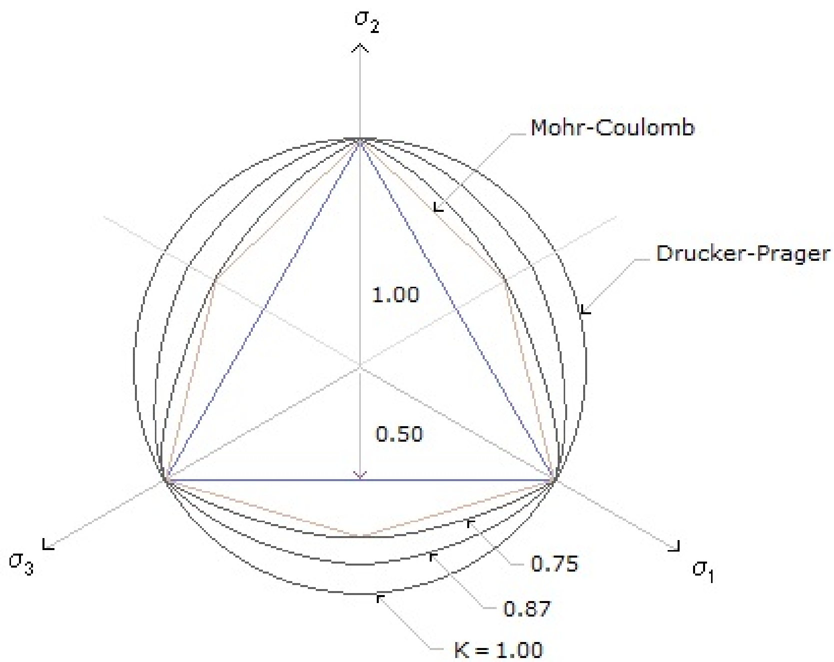


Figure 3.2 Shape of strength envelope, R(θ), on octahedral plane

3.2 Single Hardening Elasto-Plastic Model

3.2.1 Introduction

The Single Hardening Elasto-Plastic Model is a 3 invariant, single hardening surface, material model representing the drained response of the porous skeleton. The Single Hardening Model is the simplified version of the existing three invariant model which has been continuously upgraded since 1985 and reported by Merkle and Dass (1985), Dass and Merkle (1986), and Blouin, Chitty, Rauch, and Kim (1990). The major change from the existing three invariant model is the replacement of two hardening yield surfaces by the single hardening yield surface developed by Lade (1990). The advantages of this new model over the three invariant model include requirement of a lesser number of material constants, simple procedures of material parameter determination, and computational efficiency.

The Single Hardening Elasto-Plastic Model is a non-associated, isotropic, work hardening, elasto-plastic model with a single hardening yield surface bounded by a failure envelope, as shown in Figure 3.3. The yield surface has the shape of a teardrop with its pointed apex at the origin in principal stress space. The failure surface is a hyperboloid with its apex on the hydrostatic axis in the principal stress space. The shape of both yield and failure surfaces in the π-plane, perpendicular to the hydrostatic axis (see Figure 3.3), is a triple ellipse in polar coordinates.

3.2.2 Notations

Positive signs are used throughout this section to represent compression. Only those symbols which are not explicitly defined in the main text will be described below.

P_a	Atmospheric pressure
$\{\varepsilon\}$	Total strain vector
$\{\varepsilon_e\}$	Elastic strain vector
$\{\varepsilon_p\}$	Plastic strain vector associated with yield surface
$\{\varepsilon_u\}$	Plastic strain vector associated with failure surface

$\{\sigma\}$	Stress vector
σ_{oct}	Octahedral normal stress
τ_{oct}	Octahedral shear stress
ν	Poisson's ratio

3.2.3 Total Strain Formulation

In general, the total strain is comprised of the elastic component and two plastic components related to the yield and failure surfaces as given respectively by:

$$\{d\varepsilon\} = \{d\varepsilon_e\} + \{d\varepsilon_p\} + \{d\varepsilon_u\} \tag{3.18}$$

3.2.4 Elastic Response

At stress states inside the yield surfaces, the skeleton response is treated as nonlinear elastic and governed by the previous maximum peak stress. Two options are available for modeling the elastic response within the framework of the three invariant skeleton model: the modified elastic model and the Lade and Nelson elastic model. In both options, Poisson's ratio is assumed to remain constant.

Modified elastic model

During virgin unloading, the elastic bulk modulus is given by:

$$K = \frac{K_{ur}\, P_a}{3(1-2\nu)} \left[\frac{\sigma_{oct}}{P_a} \right]^n \geq K_i \tag{3.19}$$

where K_{ur} and n are material constants obtained in the parameter fitting. K_i represents the initial bulk modulus at low pressures and is necessary for modeling the behavior of rock-type materials that have a definite initial elastic behavior. In uncemented soils, K_i can be taken as a very small value. The initial bulk modulus is also used to determine the initial position of the yield surface by defining the initial elastic range.

During unloading or reloading, the skeleton modulus is described by one of two segments as depicted in Figure 3.4. Between the previous peak mean stress, $\sigma_{oct,max}$, and the transition into the nonlinear segment at $\sigma_{oct,\,b}$, the elastic bulk modulus is constant and is given by:

$$K = K_1 = \frac{K_{ur}\,P_a}{3(1-2\nu)}\left[\frac{\sigma_{oct,\,max}}{P_a}\right]^n \tag{3.20}$$

The transition into the nonlinear segment occurs at:

$$\sigma_{oct,b} = \lambda\,\sigma_{oct,\,max} \tag{3.21}$$

where λ is a model parameter. At mean stresses less than $\sigma_{oct,\,b}$ the nonlinear bulk modulus is given by:

$$K = K_1\left[1 - \gamma\left[\frac{\beta}{\gamma}\right]^{\frac{\sigma_{oct}}{\sigma_{oct,b}}}\right] \tag{3.22}$$

Referring to Figure 3.4, the model parameters γ and β are given by:

$$\gamma = 1 - \frac{K_o}{K_1} \tag{3.23}$$

where K_0 is the bulk modulus at zero pressure and

$$\beta = 1 - \frac{K_*}{K_1} \tag{3.24}$$

where K_* is the bulk modulus at one quarter of the transition pressure $\sigma_{oct,b}$. While this formulation allows for relatively accurate curve fitting of observed soil response, the model has three disadvantages:

1. For certain closed-loop stress/strain paths, the model may violate the energy conservation principle;
2. Unloading at low pressures could potentially generate expansive volumetric strains; and
3. At the transition pressure, $\sigma_{oct,b}$ the modulus is not continuous.

Lade and Nelson elastic model

The second elastic model option is based on a relationship derived by Lade and Nelson (1987). This formulation is continuous and was derived from the energy conservation principle. Lade and Nelson's model can be expressed as:

$$K = \frac{K_{ur} P_a}{3^{n+1}(1-2\nu)} \left[\left[\frac{3\sigma_{oct}}{P_a} \right]^2 + \frac{6(1+\nu)}{1-2\nu} \frac{J_2^1}{P_a^2} \right]^{\frac{n}{2}} \geq K_i \tag{3.25}$$

where the parameters K_{ur}, n, and K_i are the same as used in Equation 3.19. Since this model is fit strictly using the slope of an initial unload curve, it can be difficult to closely match the observed characteristics of an unload cycle.

Fitting

Poisson's ratio (ν) for a given material can be determined in a number of ways using unload/reload data which represent the elastic response of the skeleton. Lade and Nelson (1987) recommended obtaining Poisson's ratio directly from strain measurements in triaxial compression unload/reload cycles, right after stress reversal at hydrostatic conditions where:

$$\nu = -\frac{\varepsilon_r}{\varepsilon_a} = \frac{1}{2}\left[1 - \frac{\varepsilon_v}{\varepsilon_a} \right] \tag{3.26}$$

where

ε_a	Axial strain
ε_r	Radial strain
ε_v	Volume strain

In addition, since the elastic response is completely defined by any two independent elastic parameters, Poisson's ratio can be obtained from the bulk modulus, (K), measured in a hydrostatic compression unload, and any other elastic modulus. For example, a triaxial compression unload yields the shear modulus (G), an unconfined compression unload gives the

Young's modulus (E), and an uniaxial strain unload produces the constrained modulus (M). Any one of these parameters can be used with the bulk modulus to obtain Poisson's ratio:

$$\nu = \frac{3K - 2G}{2(3K + G)} \tag{3.27}$$

$$\nu = \frac{3K - E}{6K} \tag{3.28}$$

$$\nu = \frac{3K - M}{3K + M} \tag{3.29}$$

To obtain the elastic model parameters K_{ur} and n, Equation 3.19 is rewritten in the form:

$$\log\left[\frac{3K(1-2\nu)}{P_a}\right] = \log K_{ur} + n \log\left[\frac{\sigma_{oct}}{P_a}\right] \tag{3.30}$$

Values of K and σ_{oct} from the initial unloading response at various pressures in the hydrostatic compression test, are then plotted as log $(3K(I-2v)/P_a)$ versus log (σ_{oct}/P_a). A least squares linear regression is then applied in log-log space. The parameter n is the slope of this line, while K_{ur} is the intercept where (σ_{oct}/P_a) is 1.0. The parameters λ, γ, and β for the modified elastic unload model are determined from a single unload/reload cycle in the hydrostatic compression test as depicted in Figure 3.4. The parameters are computed using Equations 3.21, 3.23, and 3.24.

3.2.5 Failure Surface

The failure surface is a hyperboloid with its apex on the hydrostatic axis in principal stress space as shown in Figure 3.3. The shape of the failure surface in the π -plane, perpendicular to the hydrostatic axis is a triple ellipse in polar coordinates.

The failure criteria are given by:

$$f_u = \frac{\tau_{oct}}{R(\theta)}\left(\frac{m}{P_a} + \frac{1}{\bar{\sigma}_{oct}}T\right) - \eta_1 = 0 \qquad (3.31)$$

where

$$R(\theta) = \frac{2K}{(1 + K) + (1 - K)\,\mathrm{Sin}3\theta} \qquad (3.32)$$

$$\bar{\sigma}_{oct} = \sigma_{oct} + T \qquad (3.33)$$

σ_{oct}	Octahedral normal stress
τ_{oct}	Octahedral shear stress
θ	Lode angle
T	Tensile strength
K	The ratio of extensive to compressive strength at given mean pressure

m and η_1 are the failure constants which can be determined from the following fitting procedure. In triaxial compression mode, $R(\theta) = 1$ and Equation 3.31 reduces to:

$$\frac{\bar{\sigma}_{oct}}{\tau_{oct}} = \frac{1}{\eta_1} + \frac{m}{\eta_1}\left(\frac{\bar{\sigma}_{oct}}{P_a}\right) \qquad (3.34)$$

By plotting the failure stress points from each triaxial compression test in terms of $\bar{\sigma}_{oct}/\tau_{oct}$ versus $\bar{\sigma}_{oct}/P_a$, a straight line fit will yield an intercept of $1/\eta_1$ and a slope of m/η_a. Then the parameter η_1 is obtained simply by taking the inverse value of intercept and the parameter m is obtained by multiplying the slope by η_1.

3.2.6 Plastic Response Related to Yield Surface

Both yield and potential equations are based on Lade's single hardening model (Lade, 1990) which replaces previous two yield surface model (Lade, 1977).

To be consistent with the failure equation described in the previous subsection, however, Lade's equations were modified such that the shape of both yield and potential surfaces in the π-plane consists of triple ellipse given by Equation 3.32.

The yield equation is composed of the stress function (f_p') and the hardening function (f_p'').

$$f_p = f_p'(I_1, J_2, \theta) - f_p''(W_p) = 0 \tag{3.35}$$

The stress function is given by:

$$f_p' = \left(\psi_1 \frac{\bar{I}_1^{\,3}}{\bar{I}_3} - \frac{\bar{I}_1^{\,2}}{\bar{I}_2}\right)\left[\frac{\bar{I}_1}{p_a}\right]^h - e^q \tag{3.36}$$

where the stress quantities I_1, I_2, and I_3 are defined by:

$$\bar{I}_1 = I_1 + 3T \tag{3.37}$$

$$\bar{I}_2 = \left(\frac{J_2}{R(\theta)^2}\right) - \frac{\bar{I}_1^{\,3}}{3} \tag{3.38}$$

$$\bar{I}_3 = 2\left[\frac{J_2}{3R(\theta)^2}\right]^{\frac{3}{2}} - \frac{\bar{I}_1}{3}\left(\frac{J_2}{R(\theta)^2}\right) + \frac{\bar{I}_1^{\,3}}{27} \tag{3.39}$$

Note that I_1 is the first invariant of the total stress tensor, J_2 is the second invariant of deviatoric stress tensor and $R(\theta)$ is given by Equation 3.32. The variable q in Equation 3.36 is related to the shear stress level S as:

$$q = \frac{\alpha \cdot s}{1 - (1 - \alpha) \cdot s} \tag{3.40}$$

The shear stress level is defined as:

$$S = \frac{\frac{\tau_{oct}}{R_{(\theta)}}\left(\frac{m}{P_a} + \frac{1}{\bar{\sigma}_{oct}}\right)}{\eta_1} \tag{3.41}$$

The parameter q has the value of zero along the hydrostatic axis and unity along the failure surface. Thus, the material constants which are specific to the stress function are ψ_1, h and α. Determination of these material constants will be described at the end of this subsection.

The hardening function is given by:

$$f_p'' = \left[\frac{W_p}{D \cdot P_a}\right]^{\frac{h}{p}} \tag{3.42}$$

where the plastic work is expressed as:

$$W_p = \int \{\sigma\}^T \{d\epsilon_p\} \tag{3.43}$$

and the constant D is related to the isotropic hardening constants (C and P) as:

$$D = \frac{C}{(27\psi_1 + 3)^{P/h}} \tag{3.44}$$

The constant ψ_1 in Equation 3.36 and 3.44 is assumed to depend on the type of material.

Material Type	ψ_1
Sand	0.018
Clay	0.006
Mortar	0.004
Sandstone	0.0013
Concrete	0.0015
Reinforced Concrete	0.0007

It should be noted that the values of ψ_1 in the above table are based on Lade's data (Kim and Lade, 1988) but ψ_1 does not have any influence on the shape of yield surfaces on the π-plane.

Isotropic hardening constants (C and P) can be determined by fitting to the isotropic compression test. For the isotropic compression loading, Equation 3.35 reduces to:

$$\frac{W_p}{P_a} = C\left[\frac{I_1}{P_a}\right]^p \qquad (3.46)$$

Taking the logarithm of both sides of Equation 3.46 yields

$$\log_{10}\left(\frac{W_p}{P_a}\right) = \log_{10}C + P\log_{10}\left(\frac{I_1}{P_a}\right) \qquad (3.47)$$

so that the parameters C and P can be found from a log-log plot of (W_p/P_a) versus (I_1/P_a).

Yield constants (h and α) can be obtained by fitting to the triaxial compression test data. Along the isotropic and triaxial compression stress paths, Equation 3.35 has the same form as the Lade's single hardening yield equation so that the same procedure as described by Lade and Kim, 1988 can be used to determine yield constants (h and α).

The yield constant h can be obtained from:

$$h = \frac{\log_{10}\left(\frac{\left(\Psi_1 \frac{I_{1F}^3}{I_{3F}} - \frac{I_{1F}^2}{I_{2F}}\right)e}{27\psi_1 + 3}\right)}{\log_{10}\left(\frac{I_{1H}}{I_{1F}}\right)} \tag{3.48}$$

where I_{IF}, I_{2F}, and I_{3F} are the first, second and third invariant of the total stress tensor, respectively, at the failure point of triaxial compression test; I_{1H} is the first invariant of the total stress tensor in the hydrostatic compression test, measured at the same plastic work as for the failure point of triaxial compression test.

The yield constant α can be determined from

$$\alpha = \frac{1}{4}\frac{q_{80}}{1 - q_{80}} \tag{3.49}$$

Where q_{80} is value of q at the stress level S= 0.8 and is obtained from

$$q = l_n \frac{\left[\frac{W_p}{D P_a}\right]^{\frac{h}{p}}}{\left(\psi_1 \frac{I_1^{\,3}}{I_3} - \frac{I_1^{\,2}}{I_2}\right)\left[\frac{I_1}{P_a}\right]^h} \tag{3.50}$$

The potential equation is expressed in terms of stress invariants as

$$g_p = \left(\psi_1 \frac{\bar{I}_1^{\,3}}{\bar{I}_3} - \frac{\bar{I}_1^{\,2}}{\bar{I}_2} + \psi_2\right)\left[\frac{\bar{I}_1}{P_a}\right]^{\mu} \tag{3.51}$$

Material constants (ψ_2 and μ) which are specific to the potential surface can be determined in the same way as for Lade's single hardening model (Kim and Lade, 1988).

For the triaxial compression test, the potential constants (ψ_2 and μ) are related to the stresses and the plastic strains as

$$\xi_y = \frac{1}{\mu}\,\xi_x - \psi_2 \tag{3.52}$$

where

$$\xi_x = \psi_1 \frac{I_1^{\,3}}{I_3} - \frac{I_1^{\,2}}{I_2} \tag{3.53}$$

$$\xi_y = \frac{1}{1+\upsilon_p}\left(\frac{I_1^{\,3}}{I_2^2}(\sigma_a + \sigma_r + 2\upsilon_p\sigma_r) + \psi_1 \frac{I_1^{\,4}}{I_3^{\,2}}(\sigma_a\sigma_r + \upsilon_p\sigma_r^2)\right)$$

$$- 3\Psi_1 \frac{I_1^{\,3}}{I_3} + 2\frac{I_1^{\,2}}{I_2} \tag{3.54}$$

and

$$\upsilon_p = -\frac{\varepsilon_r^{\,p}}{\varepsilon_a^{\,p}} \tag{3.55}$$

Note that σ_a and σ_r are the axial and radial stress, respectively, and ε_a^p and ε_r^p are the axial and radial plastic strain, respectively. As described in Equation 3.52, the constants ψ_2 and μ now can be determined by the least square fit of a series of ξ_x and ξ_y data set.

The potential surface in Equation 3.51 is mainly used to compute the direction of the plastic strain increment during yielding.

$$\{d\varepsilon_p\} = d\lambda_p\left\{\frac{\partial g_p}{\partial \sigma_{ij}}\right\} \tag{3.56}$$

where $d\lambda_p$ is the scalar quantity.

3.2.7 Plastic Response Along the Failure Surface

When materials are hardening along the failure surface, the yield surface is crossing over the failure surface so that the state of stresses should satisfy both yield and failure equations. For simplicity, it has been assumed that there are no plastic volume changes and no strain softenings associated with the failure surface. However, there will be plastic volume changes associated with the yield surface along the failure surface. The exact method to derive the elasto-plastic stress-strain matrix $[D_{ep}]$ is presented by Merkle and Dass (1985).

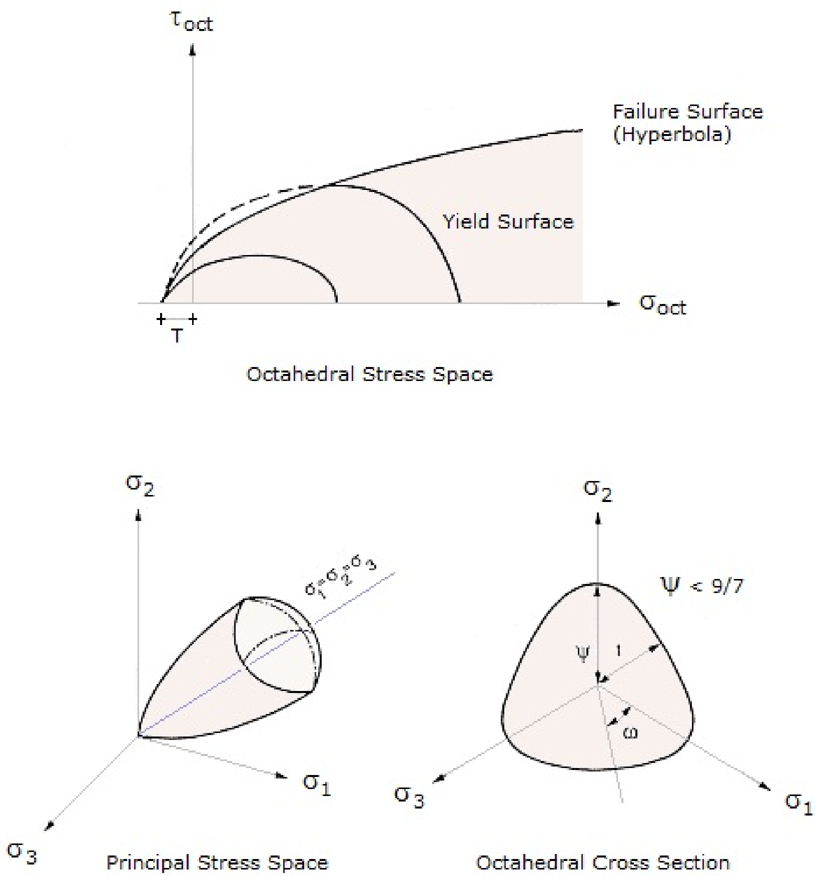


Figure 3.3 Yield and Failure Surfaces used in the Single Hardening Elasto-Plastic Model

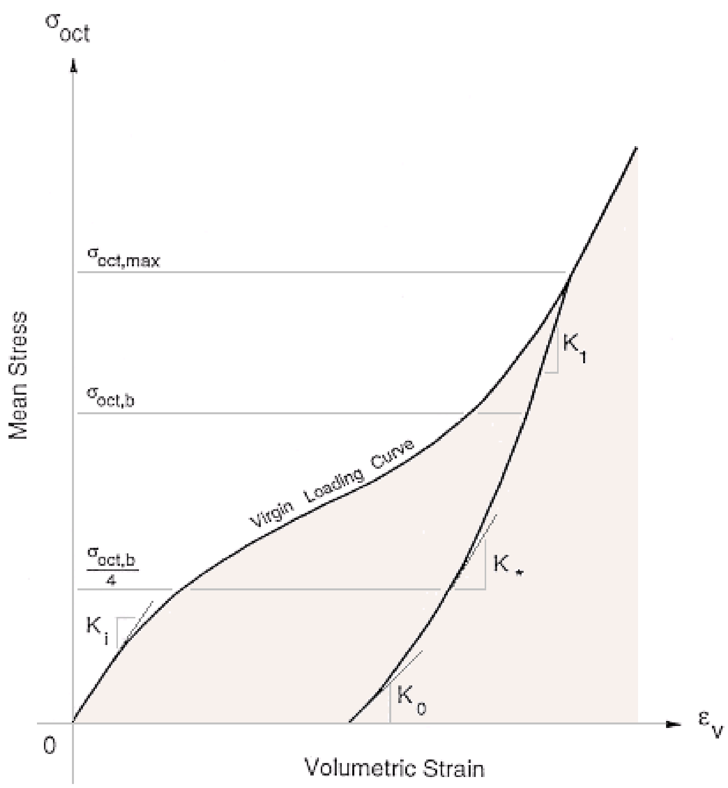


Figure 3.4 Fitting of modified elastic formulation to unload/ reload hydrostatic compression response

3.3 JWL High Explosive Model

JWL High Explosive Model represents JWL equation of state (Jones, Wilkens, and Lee, 1968).

JWL is a relatively simple pressure-volume-energy equation of state developed to predict the adiabatic expansion of detonation products. JWL equation of state is given by:

$$\bar{P} = A\left(1-\frac{\omega}{R_1 V}\right)e^{-R_1 V} + B\left(1-\frac{\omega}{R_2 V}\right)e^{-R_1 V} + \frac{\omega E}{V} \tag{3.57}$$

where $\bar{p}$ is the pressure, V is the relative volume (ρ_o/ρ), and E is the internal energy density. And A, B, R_1, R_2, and ω are material constants.

To simulate progress of chemical reaction, Burn Fraction (BF) is used.

$$BF = \frac{(t - t_b)\, C_d}{B_s\, \ell} \tag{3.58}$$

where

t	Current time
t_b	Detonation time
C_d	Detonation velocity
B_s	Constant used to spread the detonation front (usually 2.5)
ℓ	Element characteristic length

The value of Burn Fraction is limited as follows:

For $t \le t_b$,

$$BF = 0 \tag{3.59}$$

For $t > t_b + B_s \ell / C_d$,

$$BF = 1 \tag{3.60}$$

Thus, the actual pressure (P) developing in the element is obtained by combining Equations 3.57 and 3.58. That is

$$P = BF \cdot \bar{P} \tag{3.61}$$

The following JWL model parameters represents the properties of typical ANFO:

A	20 GPa
B	0.2 GPa
R_1	3.7
R_2	0.9
ω	0.2
E	7.08 GPa (Initial chemical energy)
C_d	3048 m/s
ρ_o	830 Kg/m^3 (Initial density)

3.4 Modified Cam Clay Model with Creep

3.4.1 Introduction

Long term deformations of embankments on saturated clay soils are generally associated with consolidation and creep effects. Consolidation settlements are primarily due to the expulsion of pore water while creep strains are time-dependent deformations taking place under constant stress.

This section presents detailed derivation of constitutive relations when both plastic and creep strains develop. Modified Cam Clay Model is used to represent the elasto-plastic behavior of clay soils. In computing creep strains, averaging scheme is introduced such that volumetric scaling governs at low stress ratio and deviatoric scaling governs at high stress ratio.

In this section, the elasto-plastic matrix is derived for the Modified Cam Clay Model incorporating Singh-Mitchell creep equations.

3.4.2 Yield and Failure Equations

The failure surface (critical state line) of the Modified Cam Clay Model is given by

$$q = M P' \tag{3.62}$$

where M is the failure constant and P′ and q are the alternate stress invariants given by

$$\begin{aligned} P' &= \frac{1}{3}\sigma'_{ij}\ \delta_{ij} \\ q &= \sqrt{3J_2} \\ J_2 &= \frac{1}{2} S_{ij}\ S_{ij} \\ S_{ij} &= \sigma'_{ij} - P'\ \delta_{ij} \end{aligned} \tag{3.63}$$

The yield surface of the Modified Cam Clay Model is given by

$$F = \frac{q^2}{M^2} + P'\,(P' - P'_o) = 0 \tag{3.64}$$

where P_o' is the preconsolidation pressure which grows not only with plastic volumetric strain but also with time.

3.4.3 Elastic Stress-Strain Relationship

The elastic tangent bulk modulus is based on the recompression or swelling response in isotropic compression test.

$$B_k = \frac{2.3\,(1 + e_o)}{C_r} P' \tag{3.65}$$

Where

e_o	Initial void ratio
C_r	Recompression or swelling index

Assuming the constant Poisson's ratio (υ), the elastic tangent shear modulus would be

$$G = 3.45 \frac{(1-2v)}{(1+v)} \frac{(1+e_o)}{C_r} p' \qquad (3.66)$$

With these effective mean pressure dependent elastic tangent moduli, the incremental elastic constitutive law can be expressed in the following matrix form:

$$\{d\sigma'\} = [D^e]\{d\varepsilon^e\} \qquad (3.67)$$

where

$\{d\sigma'\}$	Effective stress increment
$[D^e]$	Elastic stress-strain matrix
$\{d\varepsilon^e\}$	Elastic strain increment

3.4.4 Plastic Strain Increment

Plastic strain increment is assumed to be normal to the yield surface.

$$\{d\varepsilon^p\} = d\lambda_p \left\{ \frac{\partial F}{\partial \sigma'} \right\} \qquad (3.68)$$

where

$\{d\varepsilon^p\}$	Plastic strain increment
$d\lambda_p$	Proportional constant for plastic strain
$\{\partial F/\partial \sigma'\}$	Derivative of yield surface with respect to stress

3.4.5 Creep Strain Increment

Creep strain increment is assumed to be normal to the equivalent yield surface.

$$\{d\varepsilon^c\} = d\lambda_c \left\{ \frac{\partial F_e}{\partial \sigma'} \right\} dt \tag{3.69}$$

where

$\{d\varepsilon^c\}$	Creep strain increment
$d\lambda_c$	Proportional constant for creep strain
$\{\partial F_e / \partial \sigma'\}$	Derivative of equivalent yield surface with respect to stress
dt	Time increment

Note that the equivalent yield surface is defined as

$$F_e = \frac{q^2}{M^2} + P'(P' - P_e') = 0 \tag{3.70}$$

where

$$P_e' = \frac{q^2}{M^2 P'} + P' \tag{3.71}$$

The proportional constant for creep strain can be computed by averaging both volumetric and deviatoric scaling.

$$d\lambda_c = \frac{\eta}{M} d\lambda_d + \left(1 - \frac{\eta}{M}\right) d\lambda_v \tag{3.72}$$

where

η	Stress ratio (q/p')
$d\lambda_v$	Volumetric scaling factor
$d\lambda_d$	Deviatoric scaling factor

The volumetric scaling factor is based on the secondary consolidation curve.

$$d\lambda_v = \frac{C_\alpha}{2.3\,(1+e_o)\;t_v}\left(\frac{\partial F_e}{\partial P'}\right)^{-1} \tag{3.73}$$

The volumetric age (t_v) in Equation 3.73 is given by

$$t_v = t_{vi}\left(\frac{P'_o}{P'_e}\right)^{\frac{C_c - C_r}{C_\alpha}} \tag{3.74}$$

where

t_{vi}	Reference volumetric time
C_c	Virgin compression index
C_α	Secondary compression coefficient

The deviatoric scaling factor is based on Singh-Mitchell creep equation (1968).

$$d\lambda_d = \sqrt{\frac{3}{2}}\;A e^{\alpha\eta}\left(\frac{t_{di}}{t}\right)^m\left(\frac{\partial F_e}{\partial \sigma'_{ij}}\frac{\partial F_e}{\partial \sigma'_{ij}} - \frac{1}{3}\frac{\partial F_e}{\partial P'}\right)^{-1/2} \tag{3.75}$$

Where

t_{di}	Reference deviatoric time
t	Current time
A	Singh-Mitchell creep parameter
α	Singh-Mitchell creep parameter
m	Singh-Mitchell creep parameter

3.4.6 Total Strain Increment

Total strain increment consists of elastic, plastic and creep strains.

$$\{d\varepsilon\} = \{d\varepsilon^e\} + \{d\varepsilon^p\} + \{d\varepsilon^c\} \tag{3.76}$$

3.4.7 Consistency Equation

During the subsequent yielding, the equation forces the stress increment to move on the subsequent yield loci.

$$dF = \left\{\frac{\partial F}{\partial \sigma'}\right\}^T \{d\sigma'\} + \frac{\partial F}{\partial P_o'} dP_o' = 0 \tag{3.77}$$

The preconsolidation pressure increment (dP_o') is related to the plastic volumetric strain increment ($d\varepsilon_v^p$) and the time increment (dt).

$$dP_o' = \frac{2.3\,(1 + e_o)}{(c_c - c_r)} p_o' \, d\varepsilon_v^p + \frac{c_\alpha}{(c_c - c_r)} \frac{p_o'}{t_v} dt \tag{3.78}$$

From Equation 3.68 the plastic volumetric strain increment ($d\varepsilon_v^p$) can be expressed in terms of $d\lambda_p$

$$d\varepsilon_v^p = d\lambda_p \frac{\partial F}{\partial P'} \tag{3.79}$$

3.4.8 Evaluation of $d\lambda_p$

The elastic strain increment in Equation 3.67 can be expressed in terms of $d\lambda_p$ by combining Equations 3.76 and 3.67.

$$\{d\sigma'\} = [D^e] \left(\{d\varepsilon\} - d\lambda_p \left\{\frac{\partial F}{\partial \sigma'}\right\} - \{d\varepsilon^c\} \right) \tag{3.80}$$

Substituting Equations 3.78, 3.79 and 3.80 into the Consistency Equation 3.77 and solving for $d\lambda_p$.

$$d\lambda_p = \frac{\left\{\frac{\partial F}{\partial \sigma'}\right\}^T [D^e] (\{d\varepsilon\} - \{d\varepsilon^c\}) + p_n}{\left\{\frac{\partial F}{\partial \sigma'}\right\}^T [D^e] \left\{\frac{\partial F}{\partial \sigma'}\right\} - p_d} \tag{3.81}$$

Where

$$P_n = \frac{\partial F}{\partial P_o'} \frac{P_o'}{t_v} \frac{c_\alpha}{(c_c - c_r)} dt$$

$$P_d = \frac{\partial F}{\partial P_o'} \frac{\partial F}{\partial P'} \frac{(1 + e_o)}{2.3(c_c - c_r)} P_o'$$

3.4.9 Effective Stress Increment

The effective stress increment can be obtained by backsubstituting Equation 3.81 into 3.80.

$$\{d\sigma'\} = [D^{ep}] \{d\varepsilon\} - \{d\sigma_c'\} \tag{3.82}$$

where the incremental elasto-plastic matrix is expressed as

$$[D^{ep}] = [D^e] - \frac{[D^e] \left\{\frac{\partial F}{\partial \sigma'}\right\} \left\{\frac{\partial F}{\partial \sigma'}\right\}^T [D^e]}{\left\{\frac{\partial F}{\partial \sigma'}\right\}^T [D^e] \left\{\frac{\partial F}{\partial \sigma'}\right\} - P_d} \tag{3.83}$$

and the stress increment associated with creep is given by

$$\{d\sigma_c^{/}\} = [D^{ep}]\{d\varepsilon^c\} + \frac{[D^e]\left\{\frac{\partial F}{\partial \sigma^{/}}\right\} P_n}{\left\{\frac{\partial F}{\partial \sigma^{/}}\right\}^T [D^e]\left\{\frac{\partial F}{\partial \sigma^{/}}\right\} - P_d} \tag{3.84}$$

3.4.10 Evaluation of Derivatives

$$\frac{\partial F}{\partial P^{/}} = 2P^{/} - P_o^{/}$$

$$\left\{\frac{\partial P^{/}}{\partial \sigma^{/}}\right\} = \frac{1}{3} <1\ 1\ 1\ 0\ 0\ 0>^T$$

$$\frac{\partial F}{\partial q} = \frac{2q}{M^2}$$

$$\left\{\frac{\partial q}{\partial \sigma^{/}}\right\} = \frac{3}{2q}\{S_{ij}\}$$

$$\frac{\partial F}{\partial P_o^{/}} = -P^{/}$$

3.5 Engineering model

3.5.1 Introduction

The Engineering Model is hypoelastic-perfectly plastic in shear and hypoelastic in compression. A hypoelastic material is one for which the stress increments are homogeneous linear functions of the strain increments. In general, the coefficients in the linear functions depend on the stress. The principal advantages of the Engineering Model are ease of fitting to laboratory or in situ test data, simplicity of shear plasticity formulation, and the simple form of compressive hysteresis, which most soils exhibit. Its principal disadvantages are lack of hysteresis in pure shear at constant volume below the failure surface, and lack of dilatancy because the plastic strain increments are assumed to be normal to the hydrostatic axis. The Engineering Model is completely described by a pressure-volume strain curve for hydrostatic compression and a two-invariant failure surface.

3.5.2 Hydrostatic Response

The hydrostatic response is represented by the incremental elastic (hypoelastic) bulk modulus as a function of current compressive volumetric strain (ε_v), maximum past compressive volumetric strain (ε_{vm}) and compressive volumetric strain increment ($d\varepsilon_v$) as shown in Figure 3.5a.

$$K = K(\varepsilon_v, \varepsilon_{vm}, d\varepsilon_v) \qquad (3.85)$$

Poisson's ratio is also defined for each hydrostat segment.

$$\upsilon = \upsilon(\varepsilon_v, \varepsilon_{vm}, d\varepsilon_v) \qquad (3.86)$$

The corresponding hypoelastic constrained compression and shear moduli are then computed from the following expressions respectively:

$$M = \frac{3K(1 - \upsilon)}{(1 + \upsilon)} \qquad (3.87)$$

and

$$G = \frac{3K(1 - 2\upsilon)}{2(1 + \upsilon)} \qquad (3.88)$$

3.5.3 Plastic Shear Response

The failure surface is composed of three segments of conical surfaces as shown in Figure 3.5b, each having an equation of the form:

$$f(I_1, \sqrt{J_2}) = \sqrt{J_2} - (a + bI_1) = 0 \qquad (3.89)$$

The material is assumed to behave incrementally elastically when the stress point lies below the failure surface. When the stress point moves along the failure surface, the material response is assumed to be hypoelastic-perfectly plastic in shear. The plastic strain increments are assumed to be normal to the hydrostatic axis so that there would be no plastic volume changes associated with the failure surface.

The derivation of elasto-plastic stress-strain matrix $[D_{ep}]$ is given by Merkle and Dass (1985).

3.5.4 Parameter Determination

The Engineering Model parameters can be obtained by fitting a series of straight lines to shear strength, hydrostatic compression, and constrained compression or K_0 test data.

The parameters of shear strength envelope shown in Figure 3.5b can be determined by fitting to shear strength data in drained triaxial compression tests which arc expressed as functions of I_1, and $\sqrt{J_2}$.

Note that the values of I_1 and $\sqrt{J_2}$ at the failure points of triaxial compression are computed in terms of σ_{af} and σ_r.

$$I_1 = \sigma_{af} + 2\sigma_r \qquad (3.90)$$

and

$$\sqrt{J_2} = \frac{|\sigma_{af} - \sigma_r|}{\sqrt{3}} \tag{3.91}$$

where σ_{af} is the axial stress at failure and σ_r is the confining stress.

The hypoelastic bulk modulus and Poisson's ratio in Equations 3.85 and 3.86 respectively can be determined from hydrostatic and constrained compression tests. Then the Poisson's ratio is computed from Equation 3.87.

$$\upsilon = \frac{3K - M}{3K + M} \tag{3.92}$$

When K_o test data are available from constrained compression tests conducted in a triaxial cell, in which the confining stress is measured, Poisson's ratio can be directly computed from:

$$\upsilon = \frac{K_o}{1 + K_o} \tag{3.93}$$

where

$$K_o = \frac{d\sigma_r}{d\sigma_a} \tag{3.94}$$

When only uniaxial compression test data are available, it is customary to assume Poisson's ratio and to compute bulk modulus from

$$K = \frac{M(1 + \upsilon)}{3(1 - \upsilon)} \tag{3.95}$$

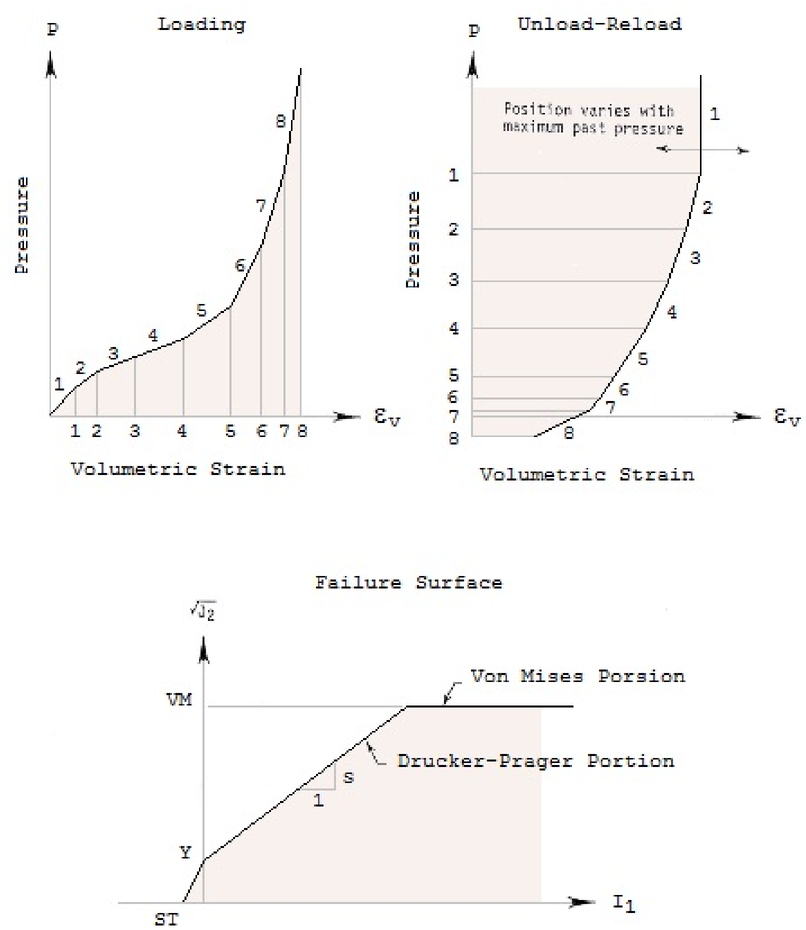


Figure 3.5 Engineering Model

3.6 Joint Model

3.6.1 Introduction

Joint Model is often used to represent rock joints, faults, and interfaces. Along the joint face, slipping takes place when the shear stress exceeds shear strength and debonding occurs when adjacent two blocks are not in contact.

Joint Model is to be used as a material model for the joint element (see Figure 3.6) as described in Card 5.4 of SMAP-3D User's Manual.

3.6.2 Strain-Displacement Relation

Strains in the joint local coordinate are

$$\{\Delta\varepsilon\} = \begin{Bmatrix} \Delta\gamma'_{zx} \\ \Delta\gamma'_{zy} \\ \Delta\varepsilon'_{zz} \end{Bmatrix} \tag{3.96}$$

where

$\Delta\gamma_{zx}'$	Shear strain increment in the plane $z'x'$
$\Delta\gamma_{zy}'$	Shear strain increment in the plane $z'y'$
$\Delta\varepsilon_{zz}'$	Normal strain increment

Local displacement increment, $\{\Delta u'\}$, is related to the global displacement increment, $\{\Delta u\}$, as follows:

$$\{\Delta u'\} = [\beta]\,[\Delta u] \tag{3.97}$$

where

$$\{\Delta u'\} = \begin{Bmatrix} \Delta u_x' \\ \Delta u_y' \\ \Delta u_z' \end{Bmatrix} \qquad \{\Delta u\} = \begin{Bmatrix} \Delta u_x \\ \Delta u_y \\ \Delta u_z \end{Bmatrix}$$

$[\beta]$ Coordinate transformation matrix

Strain-displacement relation in the local coordinate is given by

$$\{\Delta\epsilon'\} = \frac{1}{\delta}\{\Delta u'\} \tag{3.98}$$

where δ is the thickness of joint. And global displacement increment can be expressed in terms of global nodal displacement increment, $\{\Delta\bar{u}\}$, using the shape function matrix, [h], as

$$\{\Delta u\} = [h]\{\Delta\bar{u}\} \tag{3.99}$$

Now, Substituting Equations 3.97 and 3.99 into the Equation 3.98, we obtain

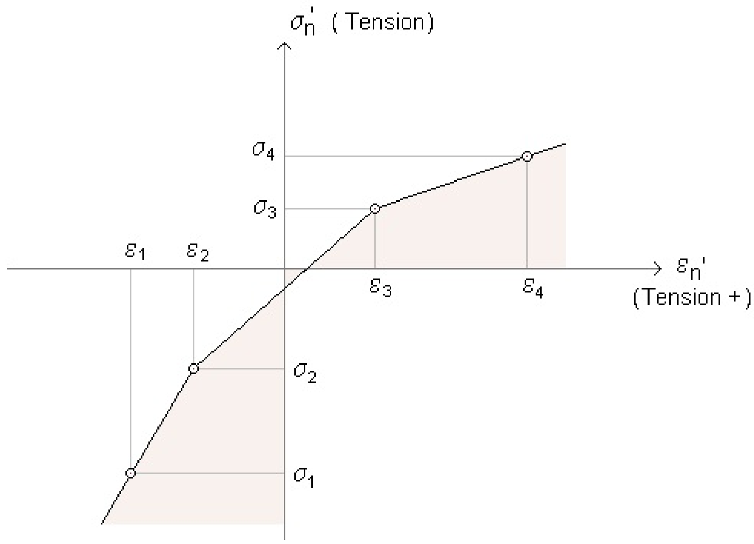
$$\{\Delta\epsilon'\} = [B]\{\Delta\bar{u}\} \tag{3.100}$$

where

$$[B] = \frac{1}{\delta}[\beta][h] \tag{3.101}$$

3.6.3 Normal Stress-Strain Relation

Normal Stress-strain relation is assumed to be nonlinear elastic as shown below



Thus, Young's modulus (E) is computed as follows:

For $\varepsilon_n' < \varepsilon_2$

$$E = \frac{\sigma_2 - \sigma_1}{\varepsilon_2 - \varepsilon_1}$$

For $\varepsilon_2 \le \varepsilon_n' < \varepsilon_3$

$$E = \frac{\sigma_3 - \sigma_2}{\varepsilon_3 - \varepsilon_2} \tag{3.102}$$

For $\varepsilon_n' \ge \varepsilon_3$

$$E = \frac{\sigma_4 - \sigma_3}{\varepsilon_4 - \varepsilon_3}$$

3.6.4 Shear Stress-Strain Relation

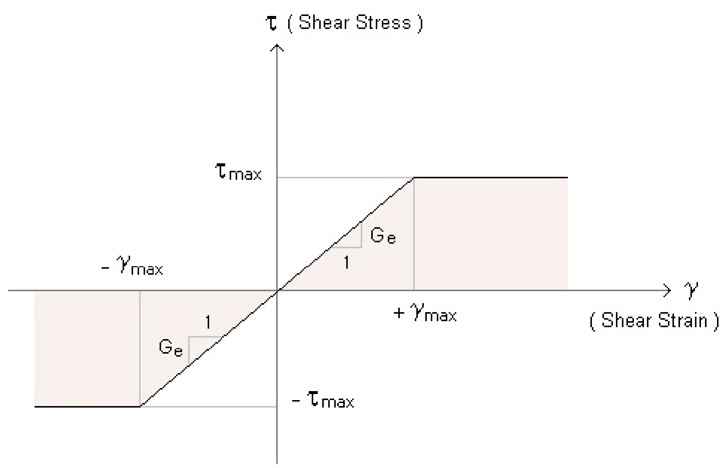
The shear strength of joint is assumed to follow Mohr-Coulomb failure criterion.

$$\tau_{max} = C - \sigma_n' \tan \varphi \qquad (3.103)$$

where

τ_{max}	Maximum shear stress
C	Cohesion
φ	Friction angle
σ_n'	Normal stress (Tension is positive)

Shear stress-strain relation is assumed to be elastic below the strength envelope and perfectly plastic along strength envelope as shown below:



Thus, shear modulus (G) is computed as follow:

$$\text{For } |\gamma| < \gamma_{max} \quad G = G_e \qquad \text{For } |\gamma| \geq \gamma_{max} \quad G = 0 \qquad (3.104)$$

Note that G_e is the elastic shear modulus.

3.6.5 Element Stiffness Matrix

Joint stress-strain relation can be given by

$$\{\Delta\sigma'\} = [C'] \{\Delta\varepsilon'\} \tag{3.105}$$

where

$$\{\Delta\sigma'\} = \begin{Bmatrix} \Delta\tau'_{zx} \\ \Delta\tau'_{zy} \\ \Delta\sigma'_{zz} \end{Bmatrix}$$

$$[C'] = \begin{bmatrix} G & 0 & 0 \\ 0 & G & 0 \\ 0 & 0 & E \end{bmatrix}$$

Note that both volumetric and shear responses are assumed to be decoupled.

Following element stiffness matrix, [K], can be derived using the principle of virtual work:

$$[K] = \int_v [B]^T [C'] [B] \, dv \tag{3.106}$$

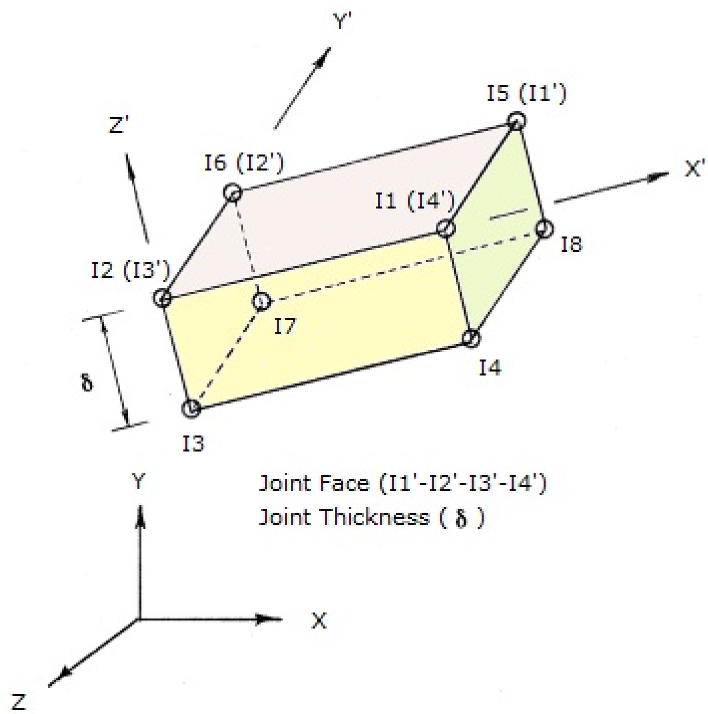


Figure 3.6 Local and Global Coordinates of Joint Element

3.7 Generalized Decoupled Hyperbolic Model

3.7.1 Introduction

GDHM (Generalized Decoupled Hyperbolic Model) is the decoupled material model which is the generalized form of the original hyperbolic model (Duncan and Chang, 1970). Main features of GDHM include:

- Hoek and Brown in situ rock strength model and
- Strength envelope expressed as a function of Lode angle on octahedral plane

3.7.2 Stress-Strain Relation

It is assumed that volumetric behavior is not coupled with deviatoric behavior. Thus, we have two independent equations:

$$\begin{aligned} dp &= K \cdot d\varepsilon_v \\ d\tau_{oct} &= 2 \cdot G \cdot d\gamma_{oct} \end{aligned} \tag{3.107}$$

where

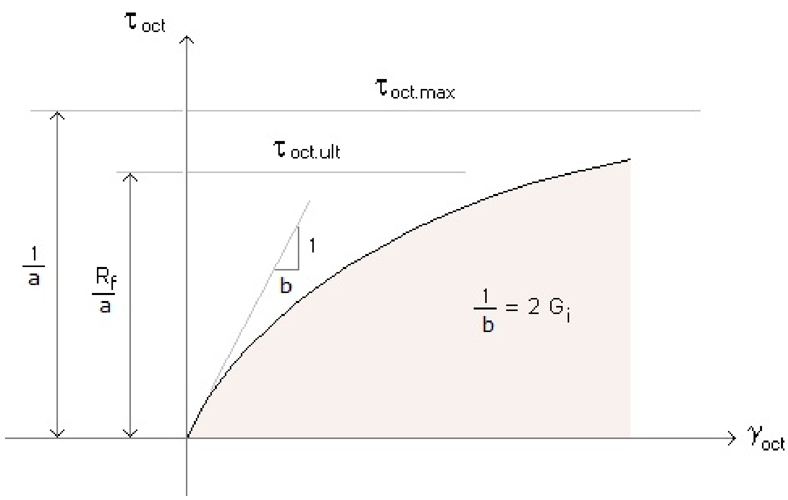
p	Mean pressure
ε_v	Volumetric strain
τ_{oct}	Octahedral shear stress
γ_{oct}	Octahedral shear strain
K	Bulk modulus
G	Shear modulus

At constant mean pressure and constant Lode angle, the shear stress-strain relation is assumed to be hyperbolic.

That is

$$\tau_{oct} = \frac{\gamma_{oct}}{b + a\,\gamma_{oct}} \tag{3.108}$$

As shown in the following figure, τ_{oct} approaches to the maximum shear stress, $\tau_{oct.max}$, as γ_{oct} goes to infinity. And the slope of the Equation 3.109 at $\gamma_{oct} = 0$ is equal to 1/b. Most triaxial compression tests, however, show that ultimate shear strength, $\tau_{oct.ult}$, is reached at finite value of γ_{oct} and $\tau_{oct.ult}$ is approximately 70~90% of $\tau_{oct.max}$.



That is,

$$\frac{1}{a} = \tau_{oct.max} = \frac{1}{R_f}\tau_{oct.ult} \qquad (3.109)$$

$$\frac{1}{b} = 2\ G_i \qquad (3.110)$$

where

R_f	Material constant (0.7~0.9)
G_i	Initial shear modulus

Differentiating Equation 3.109 with respect to γ_{oct},

$$\frac{d\tau_{oct}}{d\gamma_{oct}} = \frac{b}{(b + a\gamma_{oct})^2} \tag{3.111}$$

Solving for γ_{oct} from Equation 3.108,

$$\gamma_{oct} = \frac{b\,\tau_{oct}}{(1 - a\,\tau_{oct})} \tag{3.112}$$

Now, substituting Equation 3.113 into 3.112, we obtain the following loading shear modulus:

$$G = G_i \left(1 - \frac{\tau_{oct}}{\left(\frac{1}{a}\right)}\right)^2 \tag{3.113}$$

3.7.3 Shear Strength Equation

Kim (1984) presented the following generalized form of ultimate shear strength equation:

$$\tau_{oct.ult} = \frac{\sqrt{2}}{3}\,[(\alpha + \beta\,p)^n + \kappa]\;R(\theta) \tag{3.114}$$

The strength parameters (n, α, β, κ) are tabulated in Table 3.1 and R(θ) is given by the Equation 3.4. Note that the strength equation represents Von Mises Model when n =1, Hoek and Brown In Situ Rock Model when n=1/2, and Mohr-Coulomb or Drucker-Prager Model when n=1.

Substituting Equation 3.110 into 3.115,

$$\frac{1}{a} = \frac{1}{R_f}\,\frac{\sqrt{2}}{3}\,[(\alpha + \beta p)^n + \kappa]\;R(\theta) \tag{3.115}$$

Now, combining Equations 3.114, 3.115 and 3.116, the generalized loading shear modulus, G_i is given by

$$G = G_i \left[1 - \frac{R_f \, \tau_{oct}}{\tau_{oct.ult}} \right]^2 \tag{3.116}$$

The initial shear modulus, G_i, in Equation 3.117 may be obtained from the following empirical equations:

For cohesive soil (Hardin and Black, 1968)

$$G_i = 1230 \frac{(2.973 - e)^2}{(1 + e)} \bar{\sigma}_{oct}^{1/2} \; OCR^K \tag{3.117}$$

where

e	Void ratio
OCR	Overconsolidation ratio
K	Parameter expressed as a function of plasticity index. Note that stress unit is psi

For cohesionless soil (Janbu, 1963)

$$G_i = \frac{1}{2(1 + \nu)} K_{ur} \, P_a \left(\frac{\sigma_3}{P_a} \right)^n \tag{3.118}$$

where

ν	Poisson's ratio
P_a	Atmospheric pressure
K_{ur}, n	Material constants
σ_3	Confining pressure

It should be noted that Equation 3.117 can be degenerated to the original Duncan and Chang's hyperbolic model when Mohr-Coulomb strength envelope is specified along the triaxial compression mode.

That is,

$$\tau_{oct} = \frac{\sqrt{2}}{3}(\sigma_1 - \sigma_3) \tag{3.119}$$

$$\tau_{oct.ult} = \frac{\sqrt{2}}{3}(\sigma_1 - \sigma_3)_{ult}$$

$$= \frac{6\ \mathrm{Sin}\varphi}{(3 - \mathrm{Sin}\varphi)} P + \frac{6\ \mathrm{Cos}\varphi}{(3 - \mathrm{Sin}\varphi)} C \tag{3.120}$$

where

$$P = \frac{1}{3}(\sigma_1 + 2\ \sigma_3) \tag{3.121}$$

Substituting Equation 3.122 into 3.121 and solving for σ_1, we obtain

$$\sigma_1 = \frac{(1 + \mathrm{Sin}\varphi)}{(1 - \mathrm{Sin}\varphi)}\sigma_3 + \frac{2\ \mathrm{Cos}\varphi}{(1 - \mathrm{Sin}\varphi)} C \tag{3.122}$$

Backsubstituting σ_1 in Equation 3.123 into the right hand side of Equation 3.121,

$$\tau_{oct.ult} = \frac{\sqrt{2}}{3}\frac{(2\ \mathrm{Sin}\varphi\ \sigma_3 + 2\ \mathrm{Cos}\varphi\ C)}{(1 - \mathrm{Sin}\varphi)} \tag{3.123}$$

Now, substituting Equations 3.120 and 3.124 into 3.117, we can obtain the following Duncan and Chang Hyperbolic Model (1970):

$$G = G_i \left[1 - \frac{R_f (1 - \mathrm{Sin}\varphi)(\sigma_1 - \sigma_3)}{2\ \mathrm{Sin}\varphi\ \sigma_3 + 2\ \mathrm{Cos}\varphi\ C} \right]^2 \tag{3.124}$$

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